

3-D Problems in Elasticity

most problems are defined in displacements.

Most 3-D problems require more mathematics and that is what we will do.

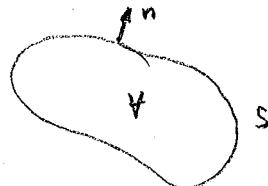
Review: Green's theorem

$$\int_V \nabla \cdot \underline{A} dV = \int_S \underline{A} \cdot \underline{n} ds \quad \text{Let: } \underline{A} = g \nabla h \quad g = g(x, y, z) \\ h = h(x, y, z)$$

$$\nabla \cdot (g \nabla h) = \nabla g \cdot \nabla h + g \nabla^2 h$$

$$\underline{A} \cdot \underline{n} = g \underline{n} \cdot \nabla h = g \frac{\partial h}{\partial n} \quad \text{remember } \frac{\partial f}{\partial k} \text{ (deriv of } f \text{ along any line whose unit vector is } \underline{k}) = \nabla f \cdot \underline{k} \text{ when } \underline{k} \perp \underline{n}$$

\underline{A} must be C_1 in V & on S .



\therefore Green's first identity

$$\int_V (g \nabla^2 h + \nabla g \cdot \nabla h) dV = \int_S g \frac{\partial h}{\partial n} ds \quad (1)$$

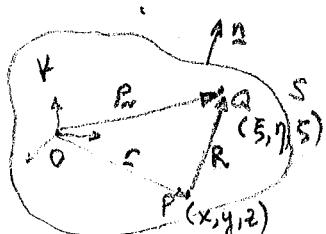
$$\text{if we let } \underline{A} = h \nabla g \text{ then } \int_V (h \nabla^2 g + \nabla h \cdot \nabla g) dV = \int_S h \frac{\partial g}{\partial n} ds \quad (2)$$

Green's 2nd identity: (1) - (2) or Green's theorem.

$$\int_V (g \nabla^2 h - h \nabla^2 g) dV = \int_S (g \frac{\partial h}{\partial n} - h \frac{\partial g}{\partial n}) ds$$

g, h must be C^2 everywhere in V
single valued & continuous w/ their
first partial derivs on boundary.

$$g(P) = \frac{1}{4\pi} \left[\int_S \left\{ \frac{1}{R(P, Q)} \frac{\partial g}{\partial n} - g \frac{\partial}{\partial n} \left(\frac{1}{R(P, Q)} \right) \right\} ds - \int_V \frac{1}{R(P, Q)} \nabla^2 g dV \right]$$



$$\rho^2 = \xi^2 + \eta^2 + \zeta^2 = \rho(\xi, \eta, \zeta) \quad \text{where } Q \in V \text{ or } S \\ r^2 = x^2 + y^2 + z^2 = r(x, y, z) \quad \text{where } P \in V \text{ only}$$

$$R^2 = (x-\xi)^2 + (y-\eta)^2 + (z-\zeta)^2$$

$$R = R(P, Q) = R(x, y, z, \xi, \eta, \zeta)$$

choose $h = \frac{1}{R}$ in green's 2nd identity where $h(Q) = \frac{1}{R(P,Q)}$ where P is fixed

Using green's 2nd ident

$$\int_V \left(g \nabla^2 \frac{1}{R} - \frac{1}{R} \nabla^2 g \right) dV = \int_S \left(g \frac{\partial(\frac{1}{R})}{\partial n} - \frac{1}{R} \frac{\partial g}{\partial n} \right) ds$$

to find $\nabla^2(\frac{1}{R})$: look at $\nabla^2(R^m)$ and this requires $\nabla(R^m)$

$$\nabla(R^m) = \frac{d}{dR}(R^m) \cdot \nabla R = mR^{m-1} \frac{\underline{R}}{|R|} = mR^{m-1} \frac{\underline{R}}{R} = mR^{m-2} \underline{R} \text{ where } \underline{R} \text{ is a unit vector in } \underline{R} \text{ direction}$$

$$\nabla \cdot \nabla(R^m) = \nabla \cdot (mR^{m-2} \underline{R}) = mR^{m-2} \nabla \cdot \underline{R} + m(m-2) R^{m-4} \frac{\underline{R} \cdot \underline{R}}{R^2} \text{ using the above}$$

now $\underline{P} \cdot \underline{R} = 3$ since $\underline{R} = (x-5, y-7, z-5)$

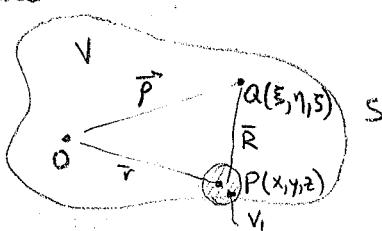
$$\therefore \nabla^2(R^m) = R^{m-2}(m+1)m \quad \text{when } m=1 \quad \nabla^2(\frac{1}{R}) = 0 \quad w/R \neq 0$$

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Quick recap

$$g(P) = \frac{1}{4\pi} \left[\int_S \left(\frac{1}{R(P,Q)} \frac{\partial g}{\partial n} - g \frac{\partial}{\partial n} \left(\frac{1}{R(P,Q)} \right) \right) ds - \int_V \frac{1}{R(P,Q)} \nabla^2 g dV \right]$$

where



$$\text{we also notice that } \int_V \left[g \left(\nabla^2 \frac{1}{R} - \frac{1}{R} \nabla^2 g \right) dV = \int_S \left(g \frac{\partial(\frac{1}{R})}{\partial n} - \frac{1}{R} \frac{\partial g}{\partial n} \right) ds \right]$$

we also showed that $\nabla^2(\frac{1}{R}) = 0$ for $R \neq 0$; ie we can draw a sphere of volume V_1 around point P and define $V = V_1$, where $V_1 = \frac{4\pi \epsilon^3}{3}$

$$\text{then } \int_{V-V_1} g \nabla^2 \left(\frac{1}{R} \right) dV = 0 \quad \text{now } \int_V \nabla^2 \left(\frac{1}{R} \right) dV = \int_V \nabla \cdot \left(\nabla \frac{1}{R} \right) dV = \int_V \nabla \frac{1}{R} \cdot \underline{n} ds$$

where \underline{n} is normal to surface S_1 : but $\underline{n} \cdot \underline{v} = \frac{\partial}{\partial n} = \frac{\partial}{\partial r}$ on S_1

$$\text{thus } \int_{S_1} \frac{\partial}{\partial n} \left(\frac{1}{R} \right) ds = \int_{S_1} \frac{\partial}{\partial r} \left(\frac{1}{R} \right) ds = - \int_{S_1} \frac{1}{R^2} ds = - \int_{S_1} \frac{1}{R^2} \cdot R^2 d\Omega = -4\pi$$

where $d\Omega$ is the solid angle $\therefore \int_{S_1} \frac{\partial}{\partial n} \left(\frac{1}{R} \right) ds = -4\pi$

$$\therefore \text{in general } \int_V \nabla^2 \left(\frac{1}{R}\right) dV = \begin{cases} 0 & \{R=0\} \notin V \\ -4\pi & \{R=0\} \in V \end{cases}$$

$$\therefore \int_V = \int_{V-V_1} + \int_{V_1} = -4\pi$$

thus we can then say that $\nabla^2 \left(\frac{1}{R}\right) = -4\pi \delta(R)$ where $\delta(R) = \begin{cases} 0 & R \neq 0 \\ 1 & R=0 \end{cases}$

$$\text{or } \nabla^2 \left(\frac{1}{|x-x'|}\right) = -4\pi \delta(x-x') \quad \text{and } \int_V g(x) \delta(x-a) dV = g(a)$$

$$\therefore \int_{V_1} g \nabla^2 \left(\frac{1}{R}\right) dV = -4\pi g(P) \int_{V_1} \delta(R) dV = -4\pi g(P) \quad \text{now take } V_1 \rightarrow 0 \Rightarrow V-V_1 \rightarrow V$$

another interesting property of the dirac delta fn is $\int_V g(x) \delta'(x-a) dV = -g'(a)$

$$\therefore \int_V \left[g \nabla^2 \left(\frac{1}{R}\right) - \frac{1}{R} \nabla g \right] dV = \int_S \left(g \frac{\partial(\frac{1}{R})}{\partial n} - \frac{1}{R} \frac{\partial g}{\partial n} \right) ds$$

$$- 4\pi g(P) - \int_V \frac{1}{R} \nabla^2 g dV = \int_S (\quad) ds$$

$$\text{or } g(P) = \frac{1}{4\pi} \int_V \left[\frac{1}{R} \frac{\partial g}{\partial n} - g \left(\frac{\partial(\frac{1}{R})}{\partial n} \right) \right] ds - \frac{1}{4\pi} \int_V \frac{1}{R} \nabla^2 g dV \quad *$$

Dirac Delta fn are generalized functions

Now if g and $\frac{\partial g}{\partial n} = 0$ on the surface the first integral = 0

$$\therefore g(P) = -\frac{1}{4\pi} \int_V \frac{1}{R(P,Q)} \nabla^2 g dV$$

in most cases: $g \rightarrow 0$ and $\frac{\partial g}{\partial n} \rightarrow 0$ as $R \rightarrow \infty$ and suppose $\nabla^2 g = f$ in V then

$$g(P) = -\frac{1}{4\pi} \int_V \frac{f}{R(P,Q)} dV$$

are these solutions unique? look at b.c. and (*)

Boundary cond. $g|_S = g_0$ Dirichlet

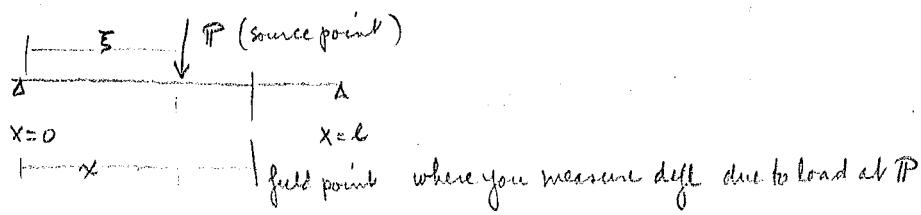
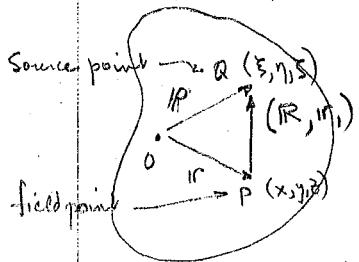
$\frac{\partial g}{\partial n}|_S = f_0$ Neumann

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Review

$$\text{Green's theorem: } \int_S (g \frac{\partial h}{\partial n} - h \frac{\partial g}{\partial n}) ds = \int_V (g \nabla^2 h - h \nabla^2 g) dV$$

$$\text{Green's formula: } g(P) = \frac{1}{4\pi} \left[\int_S \left\{ \frac{1}{R(P,Q)} \frac{\partial g(Q)}{\partial n} - g(Q) \frac{\partial}{\partial n} \frac{1}{R(P,Q)} \right\} ds - \int_V \frac{1}{R(P,Q)} \nabla^2 g(Q) dV \right]$$



Poisson's Equation is the general problem we need to solve $\nabla^2 g = f$

$$\text{BC: a) } g = g_0 \text{ on } S \quad (\text{Dirichlet Condition})$$

$$\text{b) } \frac{\partial g}{\partial n} = r_0 \text{ on } S \quad (\text{Neumann condition})$$

Prescribe one but not both
then get unique soln to Poisson
Eq. (b) is to w/ arbitrary
const.

Proof of Uniqueness

Assume g_1, g_2 satisfy DE & BC then $g_1 - g_2 = u$ is also a soln

$$\therefore \nabla^2 u = 0 \text{ and } u|_S = 0 \text{ or } \frac{\partial u}{\partial n}|_S = 0$$

$$\text{Remember } \int_V h \nabla^2 g dV = - \int_V \nabla g \cdot \nabla h dV + \int_S h \frac{\partial g}{\partial n} ds \quad \text{if } u=h=g \text{ then } \Rightarrow \int_V |\nabla u|^2 dV = 0$$

$$\text{or } \nabla u = 0, \quad u = \text{constant} \text{ but since } u|_S = 0 \Rightarrow u = 0 \text{ everywhere } \therefore g_1 = g_2$$

Better way for 1st bc: is since u must attain its max/min on bdy $\Rightarrow u = 0$ everywhere

$$\therefore g_1 = g_2$$

$$\textcircled{2} \text{ if } g \rightarrow 0 \text{ & } \frac{\partial g}{\partial n} \rightarrow 0 \text{ as } p \rightarrow \infty \Rightarrow g(P) = - \frac{1}{4\pi} \int_V \frac{\nabla^2 g}{R} dV$$

HELMHOLTZ RESOLUTION

Suppose we define $G = \nabla \phi + \nabla \times H$ where ϕ is scalar & H is vector pointfn
lamellar solenoidal

also pick $(\nabla \cdot H = 0)$ to remove the indeterminacy since ϕ, H_1, H_2, H_3 are 4 scalars (of H)

but G_1, G_2, G_3 are 3 fns (of G)

to determine ϕ, H

$$\text{take } \nabla \cdot G = \nabla^2 \phi + \nabla \cdot \nabla \times H = \nabla^2 \phi \quad \text{since we can rearrange s.t. } \nabla \cdot H \times \nabla = 0$$

we can thus use in $\mathbf{g}(\mathbf{r}) = -\frac{1}{4\pi} \int_V \frac{\nabla^2 g}{R} dV \Rightarrow \text{if } g = \phi \Rightarrow \nabla^2 g = r^2 \phi = \nabla \cdot \mathbf{G}$

$$\boxed{\phi = -\frac{1}{4\pi} \int_V \frac{\nabla \cdot \mathbf{G}}{R} dV \quad \text{if } \mathbf{G} \text{ vanishes at } \infty}$$

take $\nabla \times \mathbf{G} = \nabla \times \nabla \phi + \nabla \times \nabla \times \mathbf{H} = \nabla \nabla \cdot \mathbf{H} - \nabla^2 \mathbf{H} = -\nabla^2 \mathbf{H}$ since $\nabla \cdot \mathbf{H} = 0$

\therefore use again in the above formula to get

$$\boxed{\mathbf{H} = \frac{1}{4\pi} \int_V \frac{\nabla \times \mathbf{G}}{R} dV \quad \text{if } \mathbf{G} \text{ vanishes at } \infty}$$

rewrite to get

$$\mathbf{G} = \nabla \phi + \nabla \times \mathbf{H} = -\frac{1}{4\pi} \left[\nabla \int_V \frac{\nabla \cdot \mathbf{G}}{R} dV + \nabla \times \int_V \frac{\nabla \times \mathbf{G}}{R} dV \right]$$

where $\mathbf{G} = G(x, y, z)$

$R = R(x, y, z, \xi, \eta, \zeta)$

$\nabla = \nabla(x, y, z)$

$dV = dV(\xi, \eta, \zeta)$

Remember that the displacement equations of equilib for an isotropic, homog, linear elastic body is

$$(\lambda + \mu) \nabla \nabla \cdot \mathbf{u} + \mu \nabla^2 \mathbf{u} + \mathbf{f} = 0$$

or $\nabla^2 \mathbf{u} + \left(\frac{\lambda + \mu}{\mu}\right) \nabla \nabla \cdot \mathbf{u} + \frac{\mathbf{f}}{\mu} = 0$

recall also $\nu = \frac{\lambda}{2(\lambda + \mu)}$ $1 - 2\nu = \frac{\mu}{\lambda + \mu}$ where μ is the shear modulus

so $\nabla^2 \mathbf{u} + \left(\frac{1}{1-2\nu}\right) \nabla \nabla \cdot \mathbf{u} + \frac{\mathbf{f}}{\mu} = 0$

Introduce displacements for 3-D elasticity to solve the above. This was done by Brinson, Papkovitch, Neuber (1930-1932)

thus

$$\boxed{\mathbf{u} = \mathbf{B} - \frac{\nabla}{4(1-\nu)} [\mathbf{r} \cdot \mathbf{B} + \beta] \quad \text{position vector of field point P}}$$

where $\nabla^2 \beta = -\frac{\mathbf{f}}{\mu}$ and $\nabla^2 \beta = \frac{\mathbf{r} \cdot \mathbf{f}}{\mu}$ position vector of field point where \mathbf{f} exists

To prove their validity

$$\text{Let } u_1 = \nabla \phi + \nabla \times H \quad \text{with } \nabla \cdot H = 0$$

$$\therefore \nabla^2 u_1 + (1-2\nu) \nabla \cdot \nabla u_1 + \frac{f}{\mu} = \nabla^2 (\nabla \phi + \nabla \times H) + \frac{1}{(1-2\nu)} [\nabla \cdot \nabla \cdot \nabla \phi + \nabla \cdot \nabla \times H] + \frac{f}{\mu} = 0$$

$$\nabla^2 (\nabla \phi + \nabla \times H) + \frac{1}{1-2\nu} (\nabla \cdot \nabla^2 \phi) + \frac{f}{\mu} = \nabla^2 (\alpha \nabla \phi + \nabla \times H) = -\frac{f}{\mu} = \nabla^2 \mathbf{B}$$

$$\text{where } \alpha = 1 + \frac{1}{1-2\nu} = \frac{2(1-\nu)}{1-2\nu} \quad \text{and } \mathbf{B} = \alpha \nabla \phi + \nabla \times H$$

$$\text{now define } \beta \text{ s.t. } \phi = \frac{1}{2\alpha} (\mathbf{r} \cdot \mathbf{B} + \beta) = \frac{1}{2\alpha} (x_j B_j + \beta)$$

$$\begin{aligned} 2\alpha \nabla \phi &= \epsilon_i (x_j B_j + \beta)_{,i} = \epsilon_i (x_{j,i} B_j + x_j B_{j,i} + \beta_i) = \epsilon_i (\delta_{ji} B_j + x_j B_{j,i} + \beta_i) \\ &= \epsilon_i (B_i + x_j B_{j,i} + \beta_i) \end{aligned}$$

$$\begin{aligned} 2\alpha \nabla \cdot \nabla \phi &= \epsilon_i \cdot \epsilon_i (B_i + x_j B_{j,i} + \beta_i)_{,i} = (B_{i,i} + B_{i,i} + x_j B_{j,ii} + \beta_{ii}) \\ &= 2 \nabla \cdot \mathbf{B} + \mathbf{r} \cdot \nabla^2 \mathbf{B} + \nabla^2 \beta \end{aligned}$$

Review

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Green's formula says - value of f_n for interior point depends on $\nabla^2 f_n$ in the volume and f_n and $\frac{\partial f_n}{\partial n}$ on the boundary.

We are working to show

$$u_1 = \mathbf{B} - \frac{\nabla}{4(1-\nu)} [\mathbf{r} \cdot \mathbf{B} + \beta] \quad \text{where } \nabla^2 \mathbf{B} = -\frac{f}{\mu} \\ \nabla^2 \beta = \frac{\mathbf{r} \cdot f}{\mu}$$

to prove this we use the Helmholtz resolution

$$u_1 = \nabla \phi + \nabla \times H \quad \text{with } (\nabla \cdot H) = 0$$

and by manipulation

$$\phi = \frac{1}{2\alpha} (\mathbf{r} \cdot \mathbf{B} + \beta) = \frac{1}{2\alpha} (x_j B_j + \beta) \quad \alpha = \frac{2(1-\nu)}{1-2\nu}$$

$$\text{we also showed } 2\alpha \nabla \phi = \epsilon_i (B_i + x_j B_{j,i} + \beta_i) \quad (*)$$

$$\text{and } 2\alpha \nabla^2 \phi = 2 \nabla \cdot \mathbf{B} + \mathbf{r} \cdot \nabla^2 \mathbf{B} + \nabla^2 \beta$$

$$\text{remember } \mathbf{B} = \alpha \nabla \phi + \nabla \times H \quad \therefore \nabla \cdot \mathbf{B} = \alpha \nabla^2 \phi + 0$$

$$\text{and then } 2 \nabla \cdot \mathbf{B} = 2 \nabla \cdot \mathbf{B} + \mathbf{r} \cdot \nabla^2 \mathbf{B} + \nabla^2 \beta \quad \therefore \nabla^2 \beta = -\mathbf{r} \cdot \nabla^2 \mathbf{B}$$

$$\text{but } \nabla^2 \beta = \frac{\mathbf{r} \cdot f}{\mu} \quad \text{since } \nabla^2 \mathbf{B} = -\frac{f}{\mu}$$

$$\text{also } \mathbf{B} - \alpha \nabla \phi = \nabla \times H = \mathbf{B} - \frac{1}{2} \nabla (\mathbf{r} \cdot \mathbf{B} + \beta) \quad \text{using } (*)$$

$$\therefore u_1 = \nabla \times H + \nabla \phi = \mathbf{B} - \alpha \nabla \phi + \nabla \phi \quad \text{using the preceding result for } \nabla \times H$$

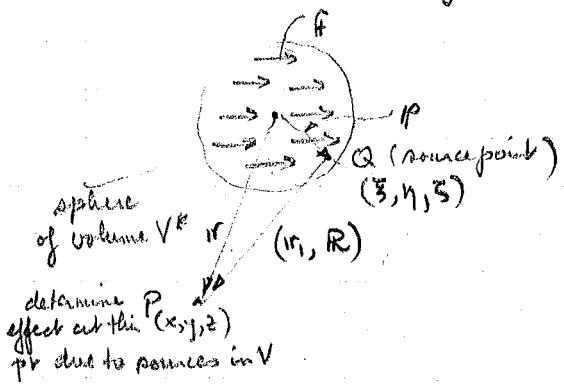
$$u = \mathbf{B} + \frac{1-\alpha}{2\alpha} \nabla (\mathbf{r} \cdot \mathbf{B} + \beta) \quad \frac{1-\alpha}{2\alpha} = -\frac{1}{4(1-\alpha)} \quad \text{using } \nabla \phi = \frac{1}{2\alpha} \nabla (\mathbf{r} \cdot \mathbf{B} + \beta)$$

$$u = \mathbf{B} - \frac{1}{4(1-\alpha)} \nabla (\mathbf{r} \cdot \mathbf{B} + \beta) \quad \text{hence if we can find } \mathbf{B}, \beta \text{ we can determine } u$$

Let's apply this to Lord Kelvin's Problem.

Kelvin's Problem = stress distribution in an ∞ elastic body subject to concentrated force.

To define Concentrated force: distribute forces over a δV let $\delta V \rightarrow 0$ in such a manner that magnitude of the forces remain const.



$$P = \lim_{V^* \rightarrow 0} \int_V f dV$$

$$\nabla^2 \mathbf{B} = -f/\mu \quad \text{for points inside } V^*$$

$$\nabla^2 \mathbf{B} = 0 \quad \text{" " outside } V^*$$

since we know $\nabla^2 \mathbf{B}$ we can use green's formula as we will use simpler formula w/o surface integral. Then plugging back into more general to see if omission is justified

use Green's formula w/o surface integral: if $\int_R \frac{\partial \mathbf{B}}{\partial n} ds \rightarrow 0$, $\mathbf{B} \rightarrow 0$ as $R \rightarrow \infty$

$$\mathbf{B} = -\frac{1}{4\pi} \int_V \frac{\nabla^2 \mathbf{B}(x, y, z)}{R} dxdydz = -\frac{1}{4\pi} \int_{V-V^*} -\frac{1}{4\pi} \int_{V^*} = -\frac{1}{4\pi} \int_{V^*} \text{ since } \nabla^2 \mathbf{B} = 0 \text{ in } V-V^*$$

More general form is

$$g(\mathbf{P}) = \frac{1}{4\pi} \left\{ \int_S \left[\frac{1}{R} \frac{\partial g}{\partial n} - g \frac{\partial \left(\frac{1}{R} \right)}{\partial n} \right] ds - \int_{V^*} \frac{1}{R} \nabla^2 g dV \right\}$$

$$\text{as } V^* \rightarrow 0, R \leftarrow \infty \quad \text{but } \mathbf{B} = \frac{1}{4\pi} \int_V \frac{f}{R} dV = \frac{\mathbf{P}}{4\pi \mu r}$$

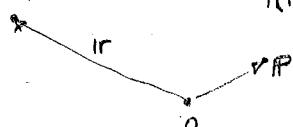
$$\text{Note: as } r \rightarrow \infty \mathbf{B} \rightarrow 0 \quad \frac{\partial \mathbf{B}}{\partial n} = \frac{\partial \mathbf{B}}{\partial r} \approx \frac{1}{r^2} \quad \therefore \int_R \frac{1}{R} \frac{\partial \mathbf{B}}{\partial n} ds \leq \frac{1}{R} \cdot \frac{1}{R^2} \cdot R^2 \rightarrow 0$$

$$\text{remember } \beta = -\frac{1}{4\pi} \int_{V^*} \frac{\nabla^2 \beta}{R} dxdydz \quad \text{but } \nabla^2 \beta = \frac{\rho}{\mu} \cdot \frac{f}{R} \quad \text{since we are inside } V^*$$

$$\beta = \frac{1}{4\pi \mu} \int_{V^*} \frac{\rho \cdot f}{R} dV = -\frac{1}{4\pi \mu} \int_{V^*} \frac{\rho \cdot f}{R} dV \quad \text{since } \rho = \text{position vector of point inside } V^* \quad \text{since we are inside } V^* \quad \rho = \frac{\mathbf{P}}{r}$$

$$\text{as } R \rightarrow \infty \quad dV \rightarrow 0 \quad \rho \rightarrow 0 \quad \therefore \beta = -\frac{1}{4\pi \mu} \cdot \frac{0}{\infty} \mathbf{P} = 0$$

$$\text{thus } u = \mathbf{B} - \frac{1}{4(1-\alpha)} \nabla (\mathbf{r} \cdot \mathbf{B} + \beta) \quad \mathbf{B} = \frac{\mathbf{P}}{4\pi \mu r} \quad \beta = 0$$



For a double force

use superposition // for each we know the answer // take their solution and take limit as $\epsilon \rightarrow 0$

We can also get problem that produces a couple

and we can also do this in all 3 planes.

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For 1st Homework find stresses & strains from u_i for the Kelvin Problem.

Continuing: $u_i = B_i - \frac{1}{4(1-\nu)} (x_j B_j)_{,i} = B_i - \frac{1}{4(1-\nu)} (B_j \delta_{ij} + x_j B_{j,i})$

Let $P = Pe_3 \Rightarrow B_1 = B_2 = 0 \quad B_3 = \frac{P}{4\pi\mu r} \quad \text{since } B = B_i e_i = \frac{P}{4\pi\mu r} = \frac{Pe_3}{4\pi\mu r}$

thus $u_1 = -\frac{1}{4(1-\nu)} [x_3 B_{3,1}] \quad B_{3,1} = \frac{\partial B_3}{\partial r} \frac{\partial r}{\partial x_1} = -\frac{P}{4\pi\mu r^2} \cdot \frac{x_1}{r} = -\frac{Px_1}{4\pi\mu r^3}$

$$\boxed{u_1 = \frac{1}{4(1-\nu)} \frac{x_1 x_3 P}{4\pi\mu r^3}}$$

Similarly $\boxed{u_2 = \frac{Px_2 x_3}{16\pi\mu(1-\nu)r^3}} \quad \text{since } B_1 = B_2 = 0$

$$\boxed{u_3 = \frac{P}{4\pi\mu r} - \frac{1}{4(1-\nu)} \left[\frac{P}{4\pi\mu r} + \frac{x_3^2 P}{4\pi\mu r^3} \right] \quad \text{since } B_3 \neq 0}$$

Alternate Derivation of Papkovitch-Boussinesq - Neuber Functions

Start w/ Equil in u_i

$$\mu \nabla^2 u_i + \frac{\mu}{(1-2\nu)} \nabla (\nabla \cdot u_i) + f_i = 0 \quad (1)$$

Now define $u_i = \nabla \phi + \nabla \times H_i \quad w/ (\nabla \cdot H = 0) \quad (2) \quad \text{Helmholtz Resolution}$

Put (2) into (1) to get $\mu \nabla^2 (\alpha \nabla \phi + \nabla \times H) + f_i = 0 \quad (3) \quad \text{where } \alpha = \frac{2(1-\nu)}{(1-2\nu)}$

define $B = \alpha \nabla \phi + \nabla \times H \quad (4)$

Put (4) into (3) to get $\boxed{\nabla^2 B = -\frac{f}{\mu}} \quad (5)$

Put (4) into (2) to get $u_i - fB = (1-\alpha) \nabla \phi \quad (6)$

take $\nabla \cdot (4)$ to get $\nabla \cdot B = \alpha \nabla^2 \phi \quad \text{since } \nabla \cdot H = 0 \quad (7)$

$$\text{Consider next } \nabla^2(r \cdot \mathbf{B}) = 2\mathbf{B} \cdot \nabla + r \cdot \nabla^2 \mathbf{B} \quad (8)$$

$$= 2\alpha \nabla^2 \phi - r \frac{\mathbf{f}}{\mu} \quad (8')$$

define $\boxed{\nabla^2 \beta = \frac{r \cdot \mathbf{f}}{\mu}}$ so that all terms of (8') have ∇^2 (9)

Using (9) $\nabla^2(r \cdot \mathbf{B}) = 2\alpha \nabla^2 \phi - \nabla^2 \beta$ integrate to get

$$\nabla(r \cdot \mathbf{B} + \beta) = 2\alpha \nabla \phi \quad \text{taking constant of integration to be zero} \quad (10)$$

$$\text{Put (10) into (6)} \quad u_1 - \bar{B} = \frac{1-\alpha}{2\alpha} \nabla(r \cdot \mathbf{B} + \beta) \quad \text{and define } \frac{1-\alpha}{2\alpha} = -\frac{1}{4(1-\alpha)}$$

then $\boxed{u_1 = \bar{B} + \frac{1}{4(1-\alpha)} \nabla(r \cdot \mathbf{B} + \beta)}$

We will now solve Kelvin's Problem using green's functions (shades of Point forces & BIE)

if $L u = f$ where L is a differential operator relating the displacement to the forcing function, then

$$L^{-1}f = u \quad \text{where } L^{-1} \text{ is the green's function}$$

We can use green's functions only for linear problems because of superposition principle. Hence if we can get the solution to

$\boxed{\text{Point load problem } \Rightarrow G}$ then $\boxed{\text{Load distribution } \Rightarrow \int_{\text{load}}^{\text{body}} \delta(\mathbf{r} - \mathbf{r}') dP} = G * f \text{ where } f \text{ is the load distribution function.}$

since $\sigma_{ij} = C_{ijkl} E_{kl}$ and $C_{ijkl} = C_{jikl} = C_{ijlk} = C_{klij}$ then
 $\sigma_{ij} = C_{ijkl} u_{k,l}$

Now the Equil Eqns. $\sigma_{ij,j} + f_i = C_{ijkl} u_{k,l,j} + f_i = 0$

for isotropy $C_{ijkl} = \lambda \delta_{ij} \delta_{kl} + \mu (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk})$
 $\sigma_{ij} = \lambda \epsilon_{kk} \delta_{ij} + 2\mu \epsilon_{ij}$

Define the displacement $T\mathbf{T}_{ij}(r) = \text{displacement } u_i(r) \text{ produced by unit force in } x_j \text{ direction acting at origin for } \infty \text{ body}$

If you substitute into Equil Eqn.

$$C_{ijkl} T\mathbf{T}_{km,lj}(r) + \delta_{im} \delta(r) = 0 \quad \text{where } f_i = \text{unit force} = \begin{cases} 1 & i=m \\ 0 & i \neq m \end{cases}$$

Kronecker Dirac δ fun.

4/16/79

Greens Tensor Function to Solve Kelvin Problem

$$c_{ijkl} u_{k,lj} + f_i = 0$$

for an infinite body

$$c_{ijkl} \bar{U}_{km,ij}^{(r)} + \delta_{im} \delta(r) = 0 \quad (\#) \quad \delta_{im} \text{ is needed} \Rightarrow \text{force acts only in one direction}$$

B.C. $U_{ij} \rightarrow 0$ as $r \rightarrow \infty$

Note: $\delta(r) = \delta(x_1)\delta(x_2)\delta(x_3)$

take equil. $c_{ijkl} u_{k,lj} = -f_i(r) = - \int f_i(r') \delta(r-r') dV' = - \int \delta_{im} \delta(r-r') f_m(r') dV'$

\therefore substitutes for $\delta_{im} \delta(r)$, $-c_{ijkl} \bar{U}_{km,ij}(r)$

$$\therefore c_{ijkl} u_{k,lj} = \int c_{ijkl} \bar{U}_{km,ij}(r-r') f_m(r') dV'$$

$$\therefore U_k(r) = \int \bar{U}_{km}(r-r') f_m(r') dV' \quad \begin{array}{l} \text{throw out constants (since they represent rigid body} \\ \text{rotation & displ.) from integration & divide by} \\ \text{elasticity tensor.} \end{array}$$

This is solution to the general problem with body forces distributed in V'

Determination of Greens tensor function by means of Fourier transform.

def: $\bar{U}_{km}(ik) = \int U_{km}(r) e^{-irk} dV_r$ dV_r is the space where r is measured

$$U_{km}(r) = \frac{1}{(2\pi)^3} \int \bar{U}_{km}(ik) e^{irk} dV_{ik} \quad dV_{ik} \text{ " " " " } ik \text{ is measured}$$

$r = x_i \otimes i$

Consider $c_{ijkl} K_i K_j \bar{U}_{km}(ik) = c_{ijkl} \int K_i K_j U_{km}(r) e^{-irk} dV_r$

integrate twice to get (neglecting surface terms since they go to 0 as $S \rightarrow \infty$)

$$= -c_{ijkl} \int \bar{U}_{km,ij}(r) e^{-irk} dV_r = \int \delta_{im} \delta(r) e^{-irk} dV = \delta_{im} \quad \text{using Equil. Eqs above (\#)}$$

Up to now we used anisotropic results - now specify to isotrop.

$$\text{Now } c_{ijkl} K_i K_j \bar{U}_{km}(ik) = (\lambda + \mu) K_i K_m \bar{U}_{km} + \mu K^2 \bar{U}_{lm} = \delta_{im} \quad (\ast\#)$$

Mult by K_i and sum to give

$$(\lambda + 2\mu) K^2 \bar{U}_{km} = K_m \quad (\#)$$

$$\begin{bmatrix} 1 \\ 1 \\ 0_3 \end{bmatrix} + \begin{bmatrix} 3 \\ 0_{-1} \end{bmatrix} = \begin{bmatrix} 2 \\ 0_2 \end{bmatrix} \Rightarrow \begin{cases} \uparrow \\ \downarrow \end{cases} \begin{bmatrix} 2 \\ 0_3 \end{bmatrix} = \begin{cases} \uparrow \\ \downarrow \end{cases} \begin{bmatrix} 1 \\ 1 \\ 0_3 \end{bmatrix}$$

Sec 131 | Southwell first solved the problem in 1926.

For stresses w/o cavity

$$\sigma_{zz} = T \quad \text{all others are zero}$$

hence form of the R, θ, ϕ coords

$$\Phi_R \cdot \Phi_\theta = T \cos \phi \quad \Phi_\theta \cdot \Phi_\phi = -T \sin \phi$$

$$\text{thus } \sigma_{RR} = T \cos^2 \phi \quad \sigma_{R\theta} = -T \sin \phi \cos \phi \quad \sigma_{\theta\theta} = T \sin^2 \phi$$

$$\sigma_{R\phi} = 0 \quad \text{everywhere since } \Phi_\theta \cdot \Phi_\phi = 0$$

The stress system necessary is made up of

1. Double force in the Z direction
2. center of compression
3. Stress for $\bar{T} = C_2 (r^2 + z^2)^{-\frac{3}{2}}$
4. Simple tension

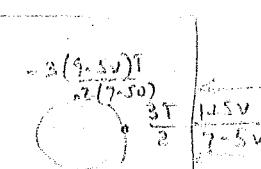
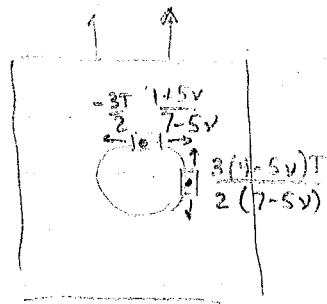
$$\text{This gives rise to } B_1, B_2 = 0 \quad B_3 = A \frac{\partial}{\partial z} \left(\frac{1}{R} \right)$$

$$\beta = \frac{P}{R} + \frac{\partial^2}{\partial z^2} \left(\frac{C}{R} \right)$$

$$\sigma_{\theta\theta} = \frac{a^3 T}{2(7-5v) R^5} \left\{ 3a^2 (3-7\cos^2 \phi) + R^2 [(4-8v) + 5(1-2v)\cos^2 \phi] + 2(7-5v) R^6 \sin^2 \phi \right\}$$

$$\text{when } \phi = 0 \quad \sigma_{\theta\theta} = \frac{a^3 T}{2(7-5v) R^5} \left\{ 12a^2 + R^2 [9-15v] \right\} = -12a^2 + 9a^2 - 15v a^2 = -3a^2 - 15v a^2 = -3a^2 (1+5v)$$

$$\sigma_{\theta\theta} \Big|_{\phi=0, \theta=\pi} = -\frac{3a^2 T (1+5v)}{2(7-5v) R^5} = -\frac{3T}{2} \frac{(1+5v)}{7-5v}$$



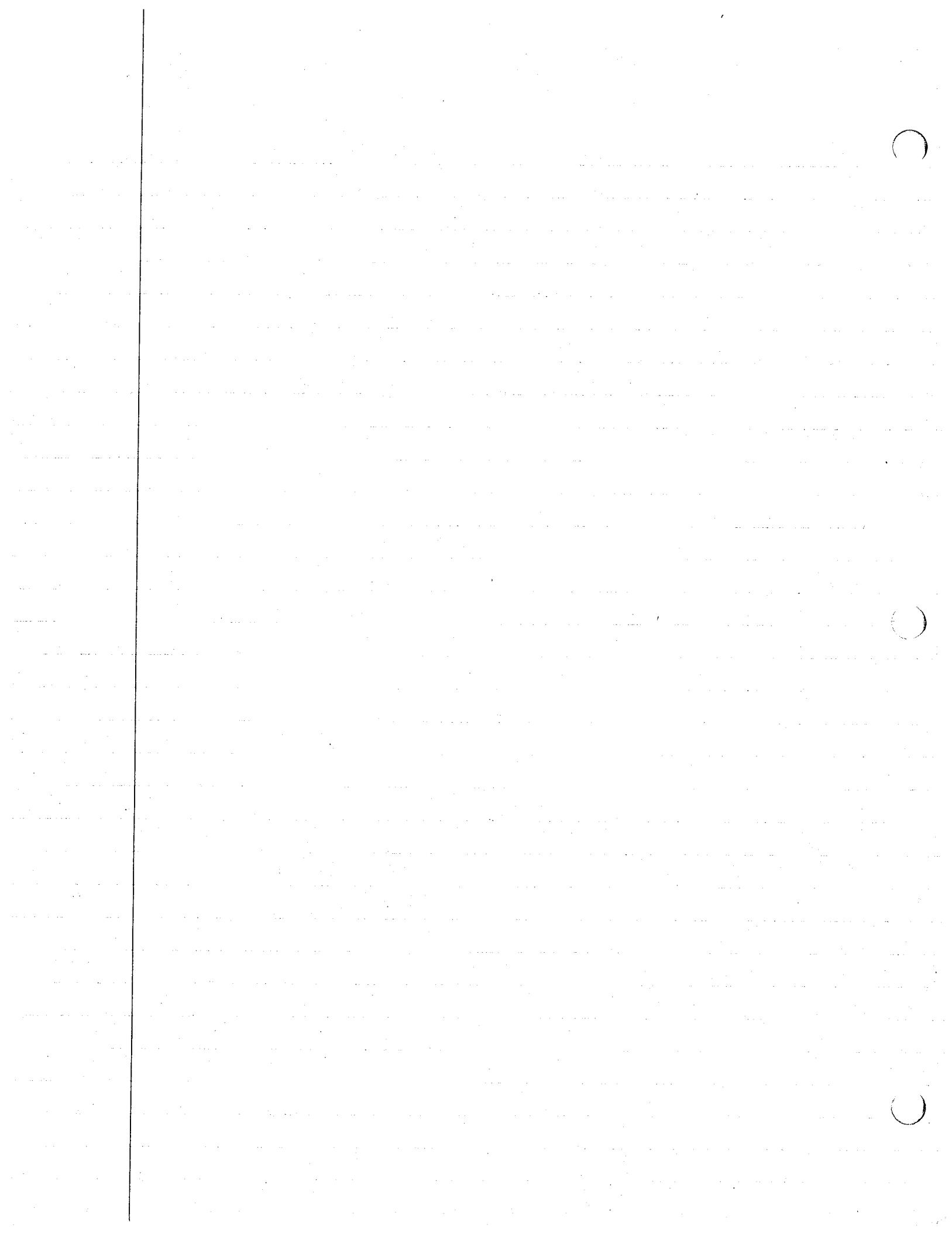
$$-\frac{3T}{2} \frac{(1+5v) + 4(1+5v)}{2(7-5v)} = \frac{15T}{7-5v}$$

$$\frac{3T}{2} \frac{(8-10v)}{7-5v} = \frac{15T}{7-5v}$$

$$-\frac{3}{2} \frac{(1+5v)T}{7-5v} + \frac{3}{2} \frac{(8-10v)}{7-5v} = \frac{3}{2} \frac{3(1+5v)T}{7-5v}$$

$$-\frac{3}{2} \frac{(1+5v)}{7-5v} + \frac{3}{2} \frac{(8-10v)}{7-5v} = \frac{3}{2} \frac{3(8-10v)}{7-5v}$$

$$\frac{3}{2} \frac{3(8-10v)}{7-5v}$$



Elementary $k_{\text{in}} U_{\text{lim}}$ using (*) & (*ii) to obtain

$$\frac{\lambda + \mu}{\lambda + 2\mu} = \frac{k_1 k_m}{k^2} \Rightarrow \mu k^2 = \bar{U}_{im} = \delta_{im}$$

Solve for \bar{U}_{im} (or \bar{U}_{km})

$$\bar{U}_{km} = \frac{1}{\mu} \left[\frac{\delta_{km}}{k^2} - \frac{\lambda_0 + \mu}{\lambda_0 + 2\mu} \frac{k_k k_m}{k^4} \right]$$

to get $U_{km}(r)$ we take

$$U_{km}(1k) = \frac{1}{(2\pi)^3} \int_{V_{km}} U(1k) e^{i k x + ik^2 d} dV_{km}$$

the only integral we need is

$$\frac{1}{\pi r^2} \int \frac{e^{ikr + ir}}{k^4} dV_K z = r$$

show this to be true

Differentiate both sides w/r k_m , k_m to get

$$\frac{e^{\lambda}}{\pi^2} \int_{-\infty}^{\infty} k e^{ikm} e^{-\frac{(k-k')^2}{4}} dk = \delta_{km}$$

$$\text{if } k_m \text{ then } \frac{1}{\pi^2} \int \frac{e^{ik_m r}}{k^2} = r_{pp} \quad \text{Laplace inversion w/ } k_m \text{ in } k^2$$

$$U_{km}(r) = \frac{1}{8\pi\mu} \left[\delta_{km} r_{pp} - \frac{\lambda + \mu}{\lambda + 2\mu} r_{km} \right]$$

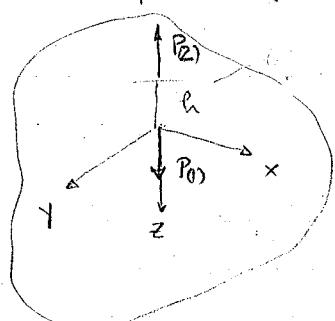
This is the greatest forest in Africa

Show this is the same as the result we had worked out by means of the potentials - see p.g. 4 above; U_1, U_2, U_3 w/p.

We will now construct solutions to some problems : NUCLEI OF STRAIN - SINGULARITIES

RICH KING Th 2-4 Fr 12:30-2:30. (Rm 264 Durand.)

Given an infinite body - Double Force



define the strength of singularity of ℓ : $P\ell = A$

We want to find the stress field for a double force w/out moment.

$$U_{12} = - \left[U_1 + \frac{2m_{\text{light}}}{\lambda^2} + \text{h.o.t.} \right] \quad \text{Since } U_{12} \text{ is in opposite dir}$$

$$\text{Total solution} \quad \frac{u_1}{p} = u_{1(1)} + u_{1(2)} = -\frac{\partial u_1}{\partial z}, \quad$$

$$\lim_{n \rightarrow \infty} u_1 = - A \frac{\partial u_{10}}{\partial t} \quad \text{using the Kelvin solution if forces are tensile } A > 0 \\ \text{compressive } A < 0$$

Now what is in fact is the dipole field

From Kelvin Prob

$$\mathbf{B} = \frac{\mathbf{P}}{4\pi\mu r} ; \beta = 0$$

$$B_z = \frac{P}{4\pi\mu r} \quad B_x = B_y = \beta = 0$$

$$u_{(1)} = \left[\mathbf{B} - \frac{1}{4(1-\nu)} \nabla (\mathbf{r} \cdot \mathbf{B}) \right] \frac{1}{r} \quad (\text{we divide by } P \text{ since we want a unit force})$$

$$= \left[\frac{P e_z}{4\pi\mu r} - \frac{1}{4(1-\nu)} \nabla \left(\frac{zP}{4\pi\mu r} \right) \right] \frac{1}{r} = \frac{1}{4\pi\mu} \left[\frac{e_z}{r} - \frac{1}{4(1-\nu)} \nabla \frac{z}{r} \right]$$

$$\frac{\partial u_{(1)}}{\partial z} = \frac{1}{4\pi\mu} \left[-\frac{e_z z}{r^2} - \frac{1}{4(1-\nu)} \nabla \left(\frac{1}{r} - \frac{z^2}{r^3} \right) \right]$$

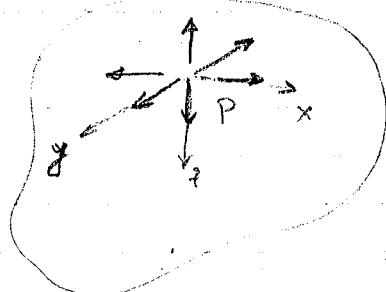
$$\text{In general } u = \mathbf{B} - \frac{1}{4(1-\nu)} \nabla (\mathbf{r} \cdot \mathbf{B} + \beta)$$

$$\text{here: } \mathbf{B} = \frac{Az e_z}{4\pi\mu r^3} \quad \beta = -\frac{A}{4\pi\mu r} \quad \mathbf{r} \cdot \mathbf{B} = \frac{Az^2}{4\pi\mu r^3}$$

Double
force
center

$$u = \frac{A}{4\pi\mu} \left[\frac{z e_z}{r^3} - \frac{1}{4(1-\nu)} \nabla \left(\frac{z^2}{r^3} - \frac{1}{r} \right) \right] = -A \frac{\partial u_{(1)}}{\partial z}$$

Suppose we look at Center of Dilatation (3 double forces along x, y, z axes) in an isotropic medium where each of the forces are of strength P



By superposition, we obtain

$$\mathbf{B} = A \left[\frac{ze_z + xe_x + ye_y}{4\pi\mu r^3} \right] = \frac{A \mathbf{r}}{4\pi\mu r^3}$$

$$\beta = -\frac{3A}{4\pi\mu r}$$

$$\text{thus } \mathbf{r} \cdot \mathbf{B} = \frac{Ar^2}{4\pi\mu r^3} = \frac{A}{4\pi\mu r}$$

$$u = \frac{Air}{4\pi\mu r^3} - \frac{1}{4(1-\nu)} \nabla \left[\frac{A}{4\pi\mu r} - \frac{3A}{4\pi\mu r} \right] \quad \nabla \left(\frac{1}{r} \right) = -\frac{1}{r^2}$$

$$= \frac{Air}{4\pi\mu r^3} - \frac{1}{4(1-\nu)} \cdot \frac{-2A}{4\pi\mu} \cdot -\frac{1}{r^3}$$

$$= \frac{Air}{4\pi\mu r^3} \left[1 - \frac{2}{4(1-\nu)} \right] = \frac{Air}{4\pi\mu r^3} \left[1 - \frac{1}{2(1-\nu)} \right] = \frac{CTR}{4(1-\nu)r^3}$$

Center of
Dilatation

$$u = \frac{-(1-2\nu)}{8(1-\nu)\pi\mu} \nabla \left(\frac{1}{r} \right)$$

$$\nabla \left(\frac{1}{r} \right) = -\frac{1}{r^2} \frac{\partial r}{\partial x_i} ; \frac{\partial r}{\partial x_i} = \frac{1-2x_i}{2r} \quad \nabla \left(\frac{1}{r} \right) = -\frac{1-2x_i}{2r}$$

$$\text{if } C = \frac{(1-\nu)}{2\pi\mu} A$$

Strength of center of dilation

$$\left\{ C \sim 1.3 = \frac{1b \cdot 1}{1m^2 / 12} \right\}$$

$$u_r = -\frac{1}{\epsilon(1-\nu)} \nabla \left(\frac{C}{r} \right) = O\left(\frac{1}{r} + \frac{1}{r^3}\right) = O\left(\frac{1}{r}\right)$$

$$\text{hence using the Papovitch Number} \Rightarrow B=0 \quad \beta = \frac{C}{r}$$

hence we note that the same displ field is obtained from $B=0 \quad \beta = \frac{C}{r}$

$$\text{and } B = \frac{A\mu r}{4\pi\mu r^3} \quad \beta = -\frac{3A}{4\pi\mu r^2}$$

hence soln is really not unique

Note that u is not dependent on orientation of axes ∇ and hence is isotropic

Also the field is divergence free everywhere since

$$\nabla \cdot u = \frac{C}{\epsilon(1-\nu)} \nabla^2 \left(\frac{1}{r} \right) = 0 \quad \text{because } \nabla^2 \left(\frac{1}{r} \right) = 0 \quad r \neq 0.$$

Since it is indep of orientation we can take the spherical coord system

$$u_r = \frac{C R}{4(1-\nu)R^3} \quad \text{using } \nabla \left(\frac{1}{R} \right) = -\frac{R}{R^3} \quad \text{where } R = R\theta_R$$

$$u_r = \frac{C}{4(1-\nu)R^3}; \quad u_\theta = 0 \quad u_\phi = 0 \quad \text{Dipole in } R, \theta, \phi \text{ system.}$$

$$\epsilon_{RR} = \frac{\partial u_r}{\partial R} = -\frac{C}{2(1-\nu)R^3} \quad \epsilon_{R\phi} = 0 \quad \epsilon_{R\theta} = 0 \quad \epsilon_{\phi\phi} = 0$$

$$\epsilon_{\theta\theta} = \epsilon_{\phi\phi} = \frac{u_r}{R} = -\frac{C}{4(1-\nu)R^3}$$

$$\Omega = \lambda \nabla \cdot u I + \mu (\nabla u I + u \nabla I)$$

$$= 0 + 2\mu I$$

$$\therefore \Omega = \begin{pmatrix} 2\mu \epsilon_{RR} & 0 & 0 \\ 0 & 2\mu \epsilon_{\theta\theta} & 0 \\ 0 & 0 & 2\mu \epsilon_{\phi\phi} \end{pmatrix} = \begin{pmatrix} \Omega_{RR} & 0 & 0 \\ 0 & \Omega_{\theta\theta} & 0 \\ 0 & 0 & \Omega_{\phi\phi} \end{pmatrix}$$

$$\therefore \Omega_{RR} = -\frac{C\mu}{(1-\nu)R^3} \quad \Omega_{\theta\theta} = \frac{C\mu}{2(1-\nu)R^3} = \Omega_{\phi\phi} \quad \Omega_{\theta\phi} = \Omega_{\phi R} = \Omega_{R\theta} = 0$$

Problem: Given a Spherical cavity under pressure P_i ; assume cavity is small w.r.t body

Since P_i is same in all directions assume at large R

the hydrostatic pressure (P_i, P_j)

$\Omega_{RR} = -P_i$ at $R=a$ $\Omega_{R\theta} = \Omega_{\theta\phi} = 0$

we can use center of dilation soln since bc are isotropic

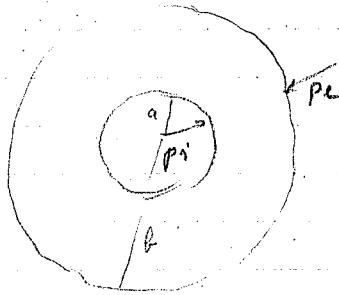
$$\text{take } \Omega_{RR} = -\frac{C\mu}{(1-\nu)a^3} = -P_i \quad \text{thus } C = P_i a^3 \frac{1-\nu}{\mu}$$

$$\text{then } \sigma_{RR} = -\frac{\rho_i a^3}{R^3} \quad \sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{\rho_i a^3}{2R^3} \quad \sigma_{\theta\phi} = \sigma_{\phi R} = \sigma_{\phi\theta} = 0$$

$$u_R = \frac{\rho_i a^3}{4\mu R^2} \quad u_\theta = u_\phi = 0 \quad w, u_R = \frac{CR \cdot \rho_i}{4(1-\nu) R^3} \quad C = \frac{\rho_i (1-\nu)}{\mu} a^3$$

4/20/79

Hollowsphere solutions



$$\sigma_{RR}|_{R=a} = -\rho_i \quad \sigma_{r\phi} = \sigma_{r\theta} = 0, \text{ at } r=a, r=b$$

$$\sigma_{rr}|_{R=b} = -p_e$$

$$\text{For center of dilation } \sigma_{RR} = -\frac{A\mu}{(1-\nu)R^3} \quad \sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{A\mu}{2(1-\nu)R^3}$$

$$\sigma_{R\phi} = \sigma_{R\theta} = \sigma_{\phi\theta} = 0$$

since this soln has only one variable A we cannot satisfy both bc. Superpose a uniform state of hydrostatic tension $\sigma_{RR} = \sigma_{\theta\theta} = \sigma_{\phi\phi} = B$ since $p_e \neq p_i$ are const. we can also do this by putting uniform compress at a & superposing on cavity soln at $r=a$. $\sigma_{RR}|_{R=a} = -\rho_i + p_e$

$$\text{bc. } -\frac{A\mu}{(1-\nu)a^3} + B = -\rho_i \quad A = \frac{(1-\nu)a^3 b^3 (p_e - \rho_i)}{\mu (a^3 - b^3)}$$

$$-\frac{A\mu}{(1-\nu)b^3} + B = -p_e \quad B = \frac{b^3 p_e - a^3 \rho_i}{a^3 - b^3}$$

$$\therefore \sigma_{RR} = (\rho_i - p_e) \frac{a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 p_e - a^3 \rho_i}{a^3 - b^3}$$

$$\sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{a^3 b^3 (p_e - \rho_i)}{2(a^3 - b^3) R^3} + \frac{b^3 p_e - a^3 \rho_i}{a^3 - b^3}$$

for a very thin shell $a \approx b$

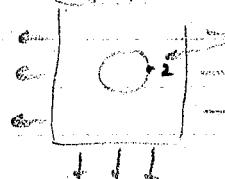
$$\text{for } \rho_i = 0 \text{ then } \sigma_{RR} = \frac{p_e b^3 (R^3 - a^3)}{(a^3 - b^3) R^3} \quad \sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{b^3 p_e (2R^3 + a^3)}{2(a^3 - b^3) R^3}$$

$$\text{if } a \ll b \text{ then } \sigma_{RR} \rightarrow p_e (-1 + (\frac{a}{R})^3) \quad \sigma_{RR} = p_e \frac{(1 - (\frac{a}{R})^3)}{-(1 - (\frac{a}{R})^3)} \rightarrow p_e (-1 + (\frac{a}{R})^3)$$

$$\sigma_{\theta\theta} = \sigma_{\phi\phi} = -[1 + \frac{1}{2}(\frac{a}{R})^3] p_e$$

if we let $b \rightarrow \infty$ then we have an ∞ body w/ uniform stress w/o applied stresses on $R=a$. then for $R=a$ $\sigma_{RR}=0$ $\sigma_{\theta\theta}=\sigma_{\phi\phi}=-\frac{3}{2} p_e$

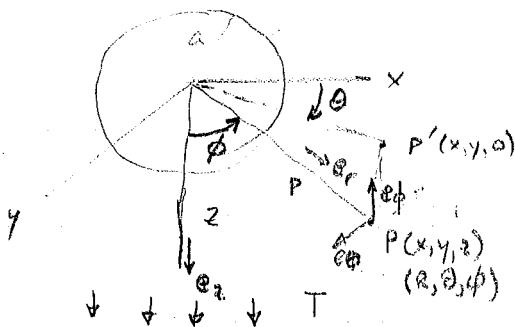
for 2d case



$$\sigma_{\theta\theta} = 2p_e$$

Suppose we have an infinite body w/ Spherical Cavity
uniaxial Remote uniform tension

$\uparrow \uparrow \uparrow \uparrow T$



For body w/o cavity ('Atate)

$$\sigma_{zz}' = T \quad \sigma_{xx}' = \sigma_{yy}' = \sigma_{xy}' = \sigma_{yx}' = \sigma_{xz}' = \sigma_{zx}' = 0$$

transform σ_3' in ('state) on cavity in R, θ, ϕ coords.

given $\sigma = T \epsilon_x \epsilon_z$ we need $\sigma_{122}', \sigma_{12\phi}', \sigma_{\phi\phi}'$

$$= \epsilon_R \cdot \sigma' \cdot \epsilon_R = \sigma_{RR} \quad \text{thus we need } \epsilon_R \cdot \epsilon_z = \cos \phi \quad \sigma_{RR} = T \cos^2 \phi$$

$$\epsilon_R \cdot \sigma' \cdot \epsilon_\phi = \sigma_{R\phi} \quad \epsilon_z \cdot \epsilon_z = \cos^2 \phi, \quad \epsilon_\phi \cdot \epsilon_z = -\sin \phi \quad \sigma_{R\phi} = -T \sin \phi \cos \phi$$

$$\epsilon_\phi \cdot \sigma' \cdot \epsilon_\phi = \sigma_{\phi\phi} \quad \text{thus} \quad \sigma_{\phi\phi} = +T \sin^2 \phi$$

$$= \epsilon_R \cdot \sigma' \cdot \epsilon_\phi = \sigma_{R\phi} = 0 \quad \text{throughout the body} \quad \text{since } \epsilon_z \cdot \epsilon_\phi = 0 \quad \& \quad \sigma_{\phi\phi} = 0$$

\therefore on $R=a$ $\sigma_{RR}, \sigma_{R\phi}$ are $\neq 0$ \therefore we must ^{superpose} a soln. \Rightarrow

$$\sigma_{RR}'' = -T \cos^2 \phi \quad \sigma_{R\phi}'' = T \sin \phi \cos \phi \quad \text{on } R=a \quad \text{and} \quad \sigma_{ij}'' \rightarrow 0 \quad \text{as } R \rightarrow \infty$$

Continuing the above - if you try $B_1, B_2 \geq 0$ $B_3 = A \frac{\partial}{\partial z} \left(\frac{1}{R} \right)$ 4/23

$$\beta = \frac{D}{R} + \frac{\partial^2}{\partial z^2} \left(\frac{C}{R} \right)$$

This is a solution that gives the following stresses $(B, \beta) \Rightarrow u_i \Rightarrow \epsilon_i \Rightarrow \sigma$

$$\sigma_{RR}'' = \frac{\mu \cos^2 \phi}{(1-\nu) R^3} \left[(5-\nu) A - \frac{18C}{R^2} \right] + \frac{\mu}{(1-\nu) R^3} \left[-\nu A - D + \frac{6C}{R^2} \right]$$

$$\sigma_{R\phi}'' = \frac{\mu \sin \phi \cos \phi}{(1-\nu) R^3} \left[(1+\nu) A - \frac{12C}{R^2} \right]$$

$$\sigma_{\phi\phi}'' = 0$$

use the method of superposition

1. Use uniform stress in infinite body soln.
2. Find σ 's on $R=a$
3. Superpose soln. \Rightarrow σ 's on $R \neq a$ are canceled

BC Equating result at $r=a$ gives

$$\left. \tau_{RR}'' \right|_{R=a} = \frac{\mu v^2 d}{(1-v)a^3} \left[(5-v)A - \frac{18C}{a^2} \right] + \frac{\mu}{(1-v)a^3} \left[-vA - D + \frac{6C}{a^2} \right] = -T \cos^2 \phi$$

$$\Rightarrow -vA - D + \frac{6C}{a^2} = 0 \quad \text{and} \quad \frac{\mu}{(1-v)a^3} \left[(5-v)A - \frac{18C}{a^2} \right] = -T$$

$$\left. \tau_{R\phi}'' \right|_{R=a} = \frac{\mu \cos \phi \sin \phi}{(1-v)a^3} \left[(1+v)A - \frac{12C}{a^2} \right] = T \sin \phi \cos \phi$$

$$\Rightarrow \frac{\mu}{(1-v)a^3} \left[(1+v)A - \frac{12C}{a^2} \right] = T$$

Solving for A, C, D :

$$A = \frac{5(1-v)a^3 T}{\mu(7-5v)} \quad D = -\frac{(1-v)(6-5v)a^3 T}{\mu(7-5v)} \quad C = -\frac{(1-v)a^5 T}{\mu(7-5v)}$$

$\tau_{\phi\phi}$ will create stress concentration along the rim thus

$$\tau_{\phi\phi} = \frac{a^3 T}{2(7-5v)R^5} \left\{ 3a^2(3-7\cos^2 \phi) + R^2(4-5v) + 5(1-2v)\cos^2 \phi \right\} + \frac{2(7-5v)}{a^3} R^5 \sin^2 \phi$$

$\tau_{\phi\phi}$ max occurs @ $R=2a$ and $\phi = \pi/2$

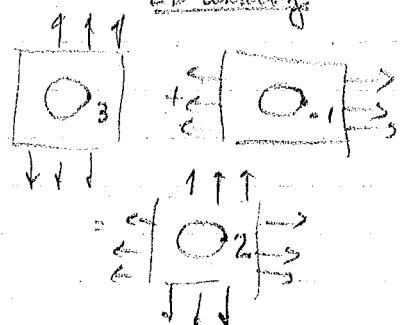
$$\tau_{\phi\phi \text{ max}} = \frac{3(9-5v)}{2(7-5v)} T = K T \quad \text{for } v=0.3 \quad K=2$$

In plane problem most are independent of v

except 1. Multiply connected region where load does not self equil on the same boundary

2D analogy:

find 3D analogy for all around tension or comp. to

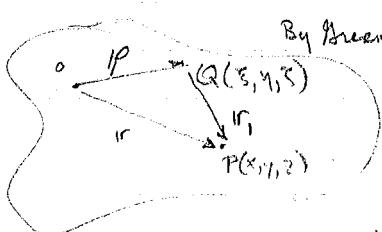


4/25/79

Half-Space Problems

If we want solutions in the near field of applied loads we can assume the body is semi infinite - solve problem & get solution in the near field - we then care not what bc or thickness of the actual problem.

Consider



$$\text{By Green's 2nd ident: } \int_S g \frac{\partial h}{\partial n} - h \frac{\partial g}{\partial n} ds - \int_V (g \nabla^2 h - h \nabla^2 g) dV = 0$$

$$\text{Pick } g \rightarrow \nabla^2 g = 0 \text{ in } V$$

$$g = \chi_{r_1} \text{ on } S.$$

$$\therefore \int_S \left(g \frac{\partial h}{\partial n} - h \frac{\partial g}{\partial n} \right) ds - \int_V g \nabla^2 h dV = 0 \quad (1)$$

$$\text{We had also shown that } 4\pi h = \int_S \left[\frac{1}{r_1} \frac{\partial h}{\partial n} - h \frac{\partial \left(\chi_{r_1} \right)}{\partial n} \right] ds - \int_V \frac{1}{r_1} \nabla^2 h dV \quad (2)$$

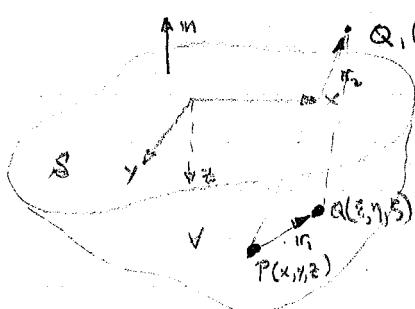
here we need $h, \frac{\partial h}{\partial n}$ on S and $\nabla^2 h$ in V

Now sub in (1) from (2)

$$4\pi h = \int_S h \frac{\partial}{\partial n} \left(g - \frac{1}{r_1} \right) ds + \int_V \left(g - \frac{1}{r_1} \right) \nabla^2 h dV$$

$$\text{Let } g = \frac{1}{r_1} = -G$$

$$4\pi h = - \left[\int_S h \frac{\partial}{\partial n} G ds + \int_V G \nabla^2 h dV \right] \quad \text{this requires only the value of } \nabla^2 h \text{ in the volume and } h \text{ on the surface.}$$

Note: $\nabla^2 G = 0$ in V and $G=0$ on S 

(mirror image of Q)

$$r_1 = |r_1| = \sqrt{(x-s)^2 + (y-\eta)^2 + (z-\zeta)^2}$$

$$r_2 = |r_2| = \sqrt{(x-s)^2 + (y-\eta)^2 + (z+\zeta)^2}$$

$$\nabla^2 \left(\frac{1}{r_2} \right) = 0$$

$$\text{on } S \quad r_1 = r_2 = r_0 = \sqrt{(x-s)^2 + (y-\eta)^2 + z^2}$$

$$\text{We now set } g = \frac{1}{r_2}, \quad G = \frac{1}{r_1} - \frac{1}{r_2} \quad \therefore \quad \nabla^2 G = 0 \text{ in } V \text{ and } G=0 \text{ on } S'$$

$$\text{For the half space} \quad \frac{\partial}{\partial n} = -\frac{\partial}{\partial z} \quad \therefore \quad -\frac{\partial G}{\partial n} = +\frac{\partial}{\partial z} \left(\frac{1}{r_1} - \frac{1}{r_2} \right)$$

$$= \frac{z-\eta}{r_0^3} + \frac{z+\zeta}{r_0^3}$$

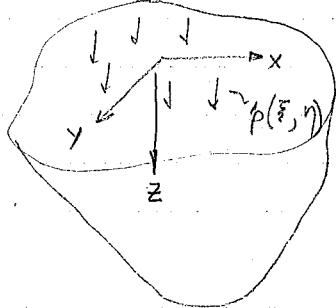
$$\text{but } -\frac{\partial G}{\partial n} \Big|_{\substack{\text{Surf} \\ z=0}} = \frac{z-\eta}{r_0^3} + \frac{z+\zeta}{r_0^3} = \frac{2z}{r_0^3} = -2 \frac{\partial}{\partial z} \left(\frac{1}{r_0} \right)$$

$$\therefore 4\pi h(x, y, z) = -2 \frac{\partial}{\partial z} \int_S \frac{h(\xi, \eta, 0)}{r_0} d\xi d\eta - \int_V \left(\frac{1}{r_1} - \frac{1}{r_2} \right) \nabla^2 h(\xi, \eta, 0) d\xi d\eta d\zeta \quad (*)$$

$\therefore G = G(x, y, z, \xi, \eta, \zeta) = \text{green's fn.}$

Example: of the results above

1. Distributed load on the half space



$$\text{B.C. on } z \quad \sigma_{zx} = \sigma_{zy} = 0 \quad \sigma_{zz} = -p(\xi, \eta)$$

$$\text{let } B_x = 0 = B_y \quad B_z \neq 0 \quad \beta \neq 0$$

$$u = B_z \varphi_z - \frac{1}{4(1-\nu)} \nabla \left[z B_z + \beta \right]$$

$$\text{since } f=0 \Rightarrow \nabla^2 B_z = 0 \quad \nabla^2 \beta = 0$$

$$\sigma_{zx} = \frac{\mu(1-2\nu)}{2(1-\nu)} \frac{\partial B_z}{\partial x} - \frac{\mu}{2(1-\nu)} \left(z \frac{\partial^2 B_z}{\partial x \partial z} + \frac{\partial^2 \beta}{\partial x \partial z} \right)$$

$$\sigma_{zy} = \frac{\mu(1-2\nu)}{2(1-\nu)} \frac{\partial B_z}{\partial y} - \frac{\mu}{2(1-\nu)} \left(z \frac{\partial^2 B_z}{\partial y \partial z} + \frac{\partial^2 \beta}{\partial y \partial z} \right)$$

$$\sigma_{zz} = \mu \frac{\partial B_z}{\partial z} - \frac{\mu}{2(1-\nu)} \left(z \frac{\partial^2 B_z}{\partial z^2} + \frac{\partial^2 \beta}{\partial z^2} \right)$$

$$\text{on Boundary} \quad \sigma_{zx} = 0 \Rightarrow \frac{\partial B_z}{\partial x} - \left(z \frac{\partial^2 B_z}{\partial x \partial z} + \frac{\partial^2 \beta}{\partial x \partial z} \right) = 0$$

since at surface $z=0$

$$\sigma_{zy} = 0 \Rightarrow (1-2\nu) \frac{\partial B_z}{\partial y} - \left(z \frac{\partial^2 B_z}{\partial y \partial z} + \frac{\partial^2 \beta}{\partial y \partial z} \right) = 0$$

$$\sigma_{zz} = 2(1-\nu) \frac{\partial B_z}{\partial z} - \left(z \frac{\partial^2 B_z}{\partial z^2} + \frac{\partial^2 \beta}{\partial z^2} \right) = -2(1-\nu) \frac{\partial \beta}{\partial z}$$

$$\text{try } h = (1-2\nu) \frac{\partial B_z}{\partial x} - \frac{\partial^2 \beta}{\partial x \partial z} \quad \text{since } h=0 \text{ on Bdy \& } \nabla^2 B_z, \nabla^2 \beta = 0 \text{ in V} \Rightarrow \nabla^2 h = 0 \text{ in V}$$

$\Rightarrow h=0$ everywhere: trivial result

$$\text{if } h = (1-2\nu) \frac{\partial B_z}{\partial y} - \frac{\partial^2 \beta}{\partial y \partial z} \text{ same trivial result}$$

$$\therefore \text{take } h = 2(1-\nu) \frac{\partial B_z}{\partial z} - \frac{\partial^2 \beta}{\partial z^2} \quad \text{now } \nabla^2 h = 0 \text{ in V but } h(\xi, \eta, 0) \neq 0$$

$$\therefore 4\pi h(x, y, z) = -2 \frac{\partial}{\partial z} \int_S -2(1-\nu) \frac{P}{r_0} d\xi d\eta$$

using (*) above

$$\text{or } h(x, y, z) = \frac{4(1-\nu)}{4\pi \mu} \frac{\partial}{\partial z} \int_S \frac{P(\xi, \eta)}{r_0} d\xi d\eta$$

$$\therefore 2(1-\nu) \frac{\partial B_2}{\partial z} - \frac{\partial^2 \beta}{\partial z^2} = \frac{(1-\nu)}{\pi \mu} \frac{2}{r_0} \int_S \frac{p(\xi, \eta)}{r_0} d\xi dy \quad (1)$$

Integrate BC. w/r to x, y

$$(1-2\nu) B_2 - \frac{\partial \beta}{\partial z} = f(y, z) \text{ - rigid body displ. set } = 0$$

$$(2) \quad (1-2\nu) B_2 - \frac{\partial \beta}{\partial z} = f(x, z) = " " " \text{ set } = 0$$

$$\therefore (3) \therefore 2(1-\nu) B_2 - \frac{\partial \beta}{\partial z} = \frac{1-\nu}{\pi \mu} \int_S \frac{p(\xi, \eta)}{r_0} d\xi dy + f(x, y) \text{ - rigid body disp. set } = 0$$

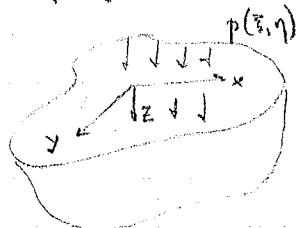
Solving we get from (2) + (3)

$$B_2 = \frac{1-2\nu}{\pi \mu} \int \frac{p(\xi, \eta)}{r_0} d\xi dy \quad r_0 = \sqrt{(x-\xi)^2 + (y-\eta)^2 + z^2}$$

$$\frac{\partial \beta}{\partial z} = \frac{(1-2\nu)(1-\nu)}{\pi \mu} \int \frac{p(\xi, \eta)}{r_0} d\xi dy \Rightarrow \beta = \frac{(1-2\nu)(1-\nu)}{\pi \mu} \int \left[\int \int \frac{p(\xi, \eta)}{r_0} d\xi dy \right] dz$$

4/27/79

Review of half space problem:



$$\text{B.C. } \sigma_{zx} = \sigma_{zy} = 0$$

$$\sigma_{zz} = -p \\ 4\pi h(x, y, z) = -2 \frac{\partial}{\partial z} \int_S \frac{\partial h(\xi, \eta, 0)}{r_0} d\xi dy = \int_V \left(\frac{1}{r_1} - \frac{1}{r_2} \right) \nabla^2 h d\xi dy dz$$

$$r_0 = \sqrt{(x-\xi)^2 + (y-\eta)^2 + z^2}$$

using Papkovitch-Neuber func w/ $B_2 \neq 0$, $\beta \neq 0$
we obtained

$$(*) \quad 2(1-\nu) \frac{\partial B_2}{\partial x} - \frac{\partial^2 \beta}{\partial x \partial z} = 0$$

$$(**) \quad 2(1-\nu) \frac{\partial B_2}{\partial y} - \frac{\partial^2 \beta}{\partial y \partial z} = -2 \frac{(1-\nu)}{\mu} p$$

$$(1-2\nu) \frac{\partial B_2}{\partial z} - \frac{\partial^2 \beta}{\partial z^2} = 0$$

$$\text{and } \nabla^2 B_2 = 0 \quad \nabla^2 \beta = 0$$

since there are no body forces

$$\therefore \text{take } \nabla^2 \text{ of } (*) \quad w/h = 2(1-\nu) \frac{\partial B_2}{\partial z} - \frac{\partial^2 \beta}{\partial z^2}$$

$$B_2 = \frac{1-\nu}{\pi \mu} \int \frac{p(\xi, \eta)}{r_0(x, y, z, \xi, \eta)} d\xi dy$$

$$\beta = \frac{(1-2\nu)(1-\nu)}{\pi \mu} \int \frac{p(\xi, \eta)}{r_0} d\xi dy dz$$

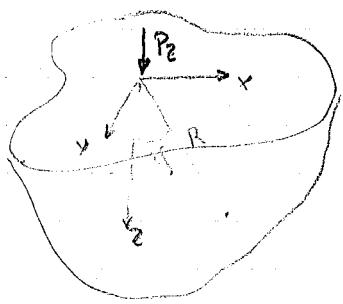
$$r_0 = \sqrt{(x-\xi)^2 + (y-\eta)^2 + z^2} \rightarrow \sqrt{x^2 + y^2 + z^2} \\ \text{when } \xi, \eta \rightarrow 0$$

$$\therefore B_2 = \frac{1-\nu}{\pi \mu} \frac{P_2}{R}$$

$$\beta = \frac{(1-2\nu)(1-\nu)}{\pi \mu} P_2 \int \frac{dz}{R}$$

$$= (1-2\nu)(1-\nu) P_2 \ln(R+z)$$

Boundary value problem - Point force at origin



$$P_2 = \lim_{A \rightarrow \infty} \int_A P dA$$

Consider now Distributed Normal Loads

$$u_2 = B_2 - \frac{1}{4(1-\nu)} \frac{\partial}{\partial z} \left(z B_2 + \beta \right)$$

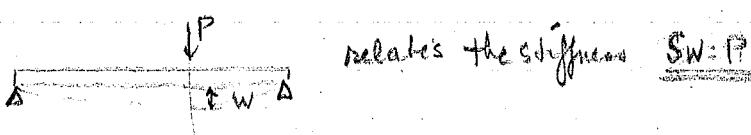
$$= B_2 - \frac{1}{4(1-\nu)} \left[B_2 + z \frac{\partial B_2}{\partial z} + \beta_z \right]$$

using (2,3) page 10 front side

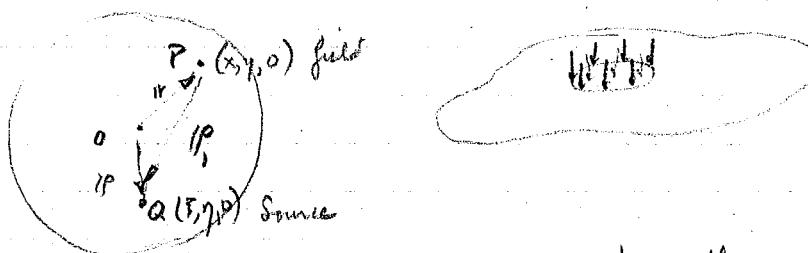
$$\text{we also showed that } \frac{\partial \beta}{\partial z} = (1-2\nu) B_2 \quad \text{for } \gamma_{xz} = \gamma_{yz} = 0$$

$$= B_2 - \frac{1}{2} \left[z \frac{\partial B_2}{\partial z} \right] \quad w = B_2 = \frac{1-\nu}{2\pi\mu} \int_{r_0}^{\infty} \frac{P(\xi, \eta)}{\xi} d\xi d\eta$$

$$\text{define } W = u_2(x, y, 0) \quad \therefore \quad W = \frac{B_2}{2} = \frac{1-\nu}{2\pi\mu} \int \frac{P(\xi, \eta)}{\sqrt{(x-\xi)^2 + (y-\eta)^2}} d\xi d\eta$$

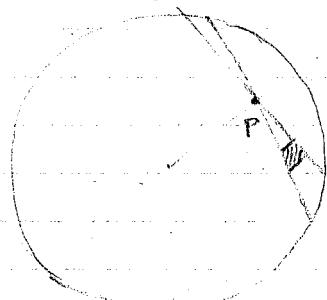


Load distributed over a circle



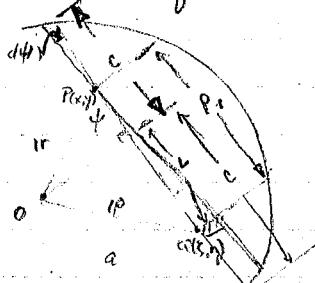
we will keep P fixed and sweep out Q - to do this we draw a line thru P intersecting circle and then sweep line through 180° to get entire circle

like the discrete continuous analog of the beam



1/30/29

Contouring of circular normal load on half space



$$Y = P_1 + C \cos \psi$$

$$C^2 = a^2 - r^2 \sin^2 \psi \quad (1)$$

$$r^2 = Y^2 + r^2 \sin^2 \psi \quad (2)$$

$$dA = P_1 dP_1 d\psi \quad dP_1 = dY \Big|_{\text{const } \psi} \quad -c \leq Y \leq a$$

$$P_1 = \sqrt{(x-\xi)^2 + (y-\eta)^2} \quad -\pi/2 \leq \psi \leq \pi/2$$

$$w(x,y) = \frac{1-\nu}{2\pi\mu} \int_{-\pi/2}^{\pi/2} \int_{-c}^c p(v,\phi) dv d\phi = \frac{1-\nu}{2\pi\mu} \int_{-\pi/2}^{\pi/2} \int_{-c}^c P(\xi, \eta) p_1 d\eta d\phi$$

Case 1 $p = \text{const.} = p_0 \Rightarrow \int_{-c}^c dv = 2c$

$$w(x,y) = \frac{1-\nu}{\pi\mu} \int_{-\pi/2}^{\pi/2} p_0 c d\phi = \frac{2(1-\nu)p_0}{\pi\mu} \int_0^{\pi} \sqrt{a^2 - r^2 \sin^2 \phi} d\phi$$

valid under load only

Now $E(\eta/a) = \int_0^{\pi/2} \sqrt{1 - (\eta/a)^2 \sin^2 \phi} d\phi$ complete elliptic integral of the 2nd kind.
 $w/E(0) = \eta/a$ $E(1) = 1$



$$w(0,0,0) = \frac{(1-\nu)}{\mu} p_0 a$$

$$w(a,0,0) = \frac{2(1-\nu)p_0 a}{\pi\mu}$$

Case 2 $p = \frac{N}{2\pi a \sqrt{a^2 - p^2}}$ load like a heavy stamp over the area of a small circle.

Total force

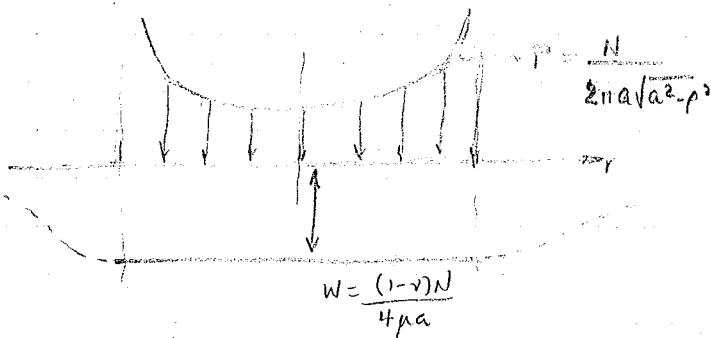
$$\int p dA = \int_0^{2\pi} \int_0^a \frac{N p d\phi dr}{2\pi a \sqrt{a^2 - p^2}} = \frac{N}{2\pi a} \int_0^{2\pi} \left[\sqrt{a^2 - p^2} \right]_0^a d\theta = N$$

$$w = \frac{(1-\nu)N}{4\pi^2 \mu a} \int_{-\pi/2}^{\pi/2} \int_{-c}^c \frac{dv d\phi}{\sqrt{a^2 - p^2}}$$

$$a^2 - p^2 = c^2 - v^2 \quad \text{using (1)+(2)}$$

$$w = \frac{(1-\nu)N}{4\pi^2 \mu a} \int_{-\pi/2}^{\pi/2} \left[\sin^{-1} \frac{v}{a} \right]_c^c d\phi = \frac{1-\nu}{4\pi^2 \mu a} (\pi/2 + \pi/2) \int_{-\pi/2}^{\pi/2} d\phi = \frac{(1-\nu)N}{4\mu a} = \text{constant}$$

$$0 \leq r \leq a$$



This is like a die applied to an area where the depth is kept constant but leads to some stress distribution which is same as that shown.

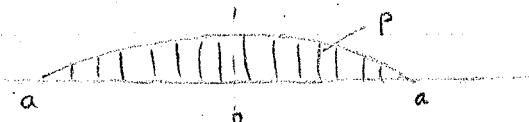
might use this locally on a punch problem of a finite thickness medium if $t \gg 2a$

Case 3

$$P = \frac{3N}{2\pi a^2} \sqrt{a^2 - r^2} \quad \text{load/unit area}$$

$$\text{Total force: } P dA = \int_0^{2\pi} \int_0^a \frac{3N}{2\pi a^2} \sqrt{a^2 - r^2} r dr d\theta = N$$

$$w = \frac{3(1-\nu)N}{4\pi^2 \mu a^3} \int_{-R_x}^{R_x} \int_{-c}^c \sqrt{c^2 - r^2} dr df = \frac{3(1-\nu)N}{16\mu a^3} (2a^2 - r^2)$$

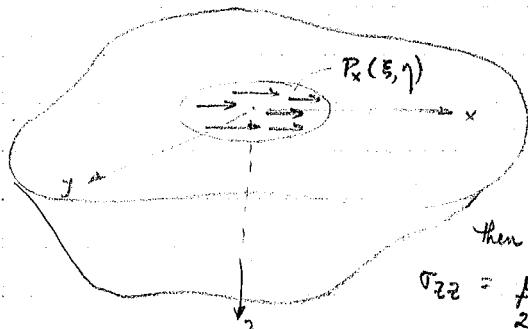


center is twice rim

$$w(0) = 2 \times w(a)$$

5/2/79

Tangential tractions over surface of half space



$$\text{B.C. } @ z=0 \quad \sigma_{zz} = \sigma_{zy} = 0 \quad \tau_{zx} = -p_x$$

$$T_x = p_x = \sigma_{xz} n_z = -\sigma_{xz} = -\sigma_{zx} \quad \text{hence}$$

$$\text{let } \sigma_{yz} = 0$$

then

$$\sigma_{zz} = \frac{\mu}{2(1-\nu)} \left\{ 2(1-\nu) \frac{\partial \sigma_z}{\partial z} + 2\nu \frac{\partial \sigma_x}{\partial x} - \left(z \frac{\partial^2 \sigma_z}{\partial z^2} + x \frac{\partial^2 \sigma_x}{\partial z^2} + \frac{\partial^2 \beta}{\partial z^2} \right) \right\}$$

$$\sigma_{yz} = \frac{\mu}{2(1-\nu)} \left\{ (1-2\nu) \frac{\partial \sigma_y}{\partial y} - \left(z \frac{\partial^2 \sigma_z}{\partial y \partial z} + x \frac{\partial^2 \sigma_x}{\partial y \partial z} + \frac{\partial^2 \beta}{\partial y \partial z} \right) \right\}$$

$$\sigma_{zx} = \frac{\mu}{2(1-\nu)} \left\{ (1-2\nu) \left(\frac{\partial \sigma_z}{\partial x} + \frac{\partial \sigma_x}{\partial z} \right) - \left(z \frac{\partial^2 \sigma_z}{\partial x \partial z} + x \frac{\partial^2 \sigma_x}{\partial x \partial z} + \frac{\partial^2 \beta}{\partial x \partial z} \right) \right\}$$

Using the theorem relating values on surface to those inside volume, we find we must try several functions to find something that works. (top of pg 10)

$$\text{Consider } H = \frac{\partial \sigma_{zx}}{\partial y} - \frac{\partial \sigma_{yz}}{\partial x} = \frac{\mu}{2(1-\nu)} \left\{ (1-2\nu) \frac{\partial^2 \sigma_x}{\partial y \partial z} + \frac{\partial^2 \sigma_x}{\partial y \partial z} \right\} = \mu \frac{\partial^2 \sigma_x}{\partial y \partial z}$$

$$\text{For } z=0 \quad \sigma_{yz}=0 \Rightarrow \frac{\partial \sigma_{zy}}{\partial x} = 0 \quad \text{thus } H(x, y, 0) = \frac{\partial \sigma_{zx}}{\partial y} = -\frac{\partial p_x}{\partial y}$$

$$\text{But } H(x, y, 0) = -\frac{\partial p_x}{\partial y} = \mu \frac{\partial^2 \sigma_x}{\partial y \partial z} \Big|_{z=0}$$

integrate wrt y $\therefore -p_x = \mu \frac{\partial \sigma_x}{\partial z} \Big|_{z=0} (+ \text{f.m.})$ (which only gives rise to rigid body rot + trans. b.c. = 0)

Recall $\nabla^2 \sigma_x = 0$ since we have no body forces
(since $\nabla^2 \sigma = -P/\mu$)

$$\therefore \frac{\partial \nabla^2 \sigma_x}{\partial z} = \nabla^2 \frac{\partial \sigma_x}{\partial z} = 0$$

Use theorem that $4\pi h(x, y, z) = -2 \frac{\partial}{\partial z} \int_S \frac{h(\xi, \eta, 0)}{r_0} d\xi d\eta - \int_V (\frac{1}{r_1} - \frac{1}{r_2}) \nabla^2 h(\xi, \eta, z) d\xi d\eta dz$ ④

with $h = \frac{\partial B_x}{\partial z}$. Then $\frac{\partial B_x}{\partial z}(x, y, z) = -\frac{1}{2\pi} \frac{\partial}{\partial z} \int_{r_0} \frac{\frac{\partial B_x}{\partial z}|_{z=0}}{r_0} ds$ since $\nabla^2 \frac{\partial B_x}{\partial z} = 0$

$$\therefore \frac{\partial B_x}{\partial z} = -\frac{1}{2\pi \mu} \frac{\partial}{\partial z} \int_{r_0} \frac{-B_x(\xi, \eta)}{r_0} ds$$

$$\text{Now integrate } B_x = \frac{1}{2\pi \mu} \int_{r_0} \frac{B_x(\xi, \eta)}{r_0} d\xi d\eta$$

$$\text{Consider } h'_1 = (1-2\nu) \frac{\partial B_z}{\partial y} - \left(x \frac{\partial^2 B_x}{\partial y \partial z} + \frac{\partial^2 \beta}{\partial y \partial z} \right)$$

$$h'_1(x, y, 0) = 0 \text{ since this is essentially } \sigma_{yz}|_{z=0} = 0$$

$$\text{now } \nabla^2 h'_1 = (1-2\nu) \frac{\partial}{\partial y} \nabla^2 B_z = \left(x \frac{\partial^2}{\partial y \partial z} \nabla^2 B_x + \nabla^2 B_x \cdot \frac{\partial^2 B_x}{\partial y \partial z} + 1 \cdot \frac{\partial^3 B_x}{\partial x \partial y \partial z} + \frac{\partial^2}{\partial y \partial z} \nabla^2 \beta \right) \\ = -\frac{\partial^3 B_x}{\partial x \partial y \partial z} \text{ since } f \neq 0 \quad \text{since } f \neq 0 \quad = 0 \text{ since } f = 0$$

Trick to make h'_1 zero everywhere we will use the theorem again by adding $z \frac{\partial^2 B_x}{\partial x \partial y}$ to h'_1

$$\therefore h_1 = h'_1 + z \frac{\partial^2 B_x}{\partial x \partial y} \\ \text{Consider } h_1 = (1-2\nu) \frac{\partial B_z}{\partial y} = \left(-2 \frac{\partial^2 B_x}{\partial x \partial y} + x \frac{\partial^2 B_x}{\partial y \partial z} + \frac{\partial^2 \beta}{\partial y \partial z} \right) \quad (1)$$

$$h_1(x, y, 0) = 0 \text{ and } \nabla^2 h_1 = 0 \text{ then by ④ } h_1(x, y, z) = 0$$

$$\text{Consider } h'_2 = 2(1-\nu) \frac{\partial B_z}{\partial z} + 2\nu \frac{\partial B_x}{\partial x} = \left(x \frac{\partial^2 B_x}{\partial z^2} + \frac{\partial^2 \beta}{\partial z^2} \right)$$

$$h'_2(x, y, 0) = 0 \text{ since this is essentially } \sigma_{zz}|_{z=0} = 0$$

now $\nabla^2 h'_2 = -\frac{\partial B_x}{\partial x \partial z^2}$ as before; to make h'_2 zero everywhere again by the theorem add $z \frac{\partial^2 B_x}{\partial x \partial z}$

$$\text{then } h_2 = h'_2 + z \frac{\partial^2 B_x}{\partial x \partial z}$$

$$\therefore h_2 = 2(1-\nu) \frac{\partial B_z}{\partial z} + 2\nu \frac{\partial B_x}{\partial x} = \left(-2 \frac{\partial^2 B_x}{\partial x \partial z} + x \frac{\partial^2 B_x}{\partial z^2} + \frac{\partial^2 \beta}{\partial z^2} \right) \quad (2)$$

$$\therefore h_2(x, y, 0) = 0 \text{ w/ } \nabla^2 h_2 = 0 \Rightarrow h_2(x, y, z) = 0$$

Integrate (1) wrt y

$$\int h_1 dy = (1-2\nu) B_z = \left(-2 \frac{\partial B_x}{\partial x} + x \frac{\partial B_x}{\partial z} + \frac{\partial \beta}{\partial z} \right) = 0 \quad (3) \quad (+ \text{ fn of } (x, y) \text{ r.b.r or t.m})$$

differentiate (3) wrt z

$$\therefore (1-2\nu) \frac{\partial B_z}{\partial z} = \left(x \frac{\partial^2 B_x}{\partial z^2} + \frac{\partial^2 \beta}{\partial z^2} - \frac{\partial B_x}{\partial x} - 2 \frac{\partial^2 B_x}{\partial x \partial z} \right) = 0 \quad (4)$$

Subtract (2) from (4) to give

$$-2B_z + 2\nu \frac{\partial B_x}{\partial z} - \frac{\partial B_x}{\partial x} = 0 \quad (5)$$

$$\text{or } \frac{\partial B_z}{\partial z} = (1-2v) \frac{\partial B_x}{\partial x}$$

from (3) solve for $\frac{\partial \beta}{\partial z}$

$$\frac{\partial \beta}{\partial z} = (1-2v) B_z - x \frac{\partial B_x}{\partial z} + z \frac{\partial B_x}{\partial x}$$

One Hour midterm - Friday May 11 Open Notes - no books
HW : Due May 16.

Knowing B_x we can get $\frac{\partial B_x}{\partial x} \Rightarrow$ we know $\frac{\partial B_z}{\partial z} \Rightarrow$ integrate to get B_z

knowing B_x we can get $\frac{\partial B_x}{\partial x}, \frac{\partial B_x}{\partial z}$. Knowing B_z also \Rightarrow we know $\frac{\partial \beta}{\partial z} \Rightarrow$ integrate to get β .

5.4.79

if we have tangential forces in a half space then

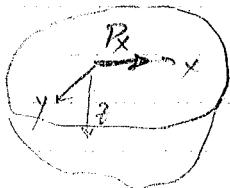
if $P_x(\xi, \eta)$ is given at $z=0$

$$B_x = \frac{1}{2\pi\mu} \int_{r_0}^{\infty} P_x(\xi, \eta) d\xi d\eta$$

$$\frac{\partial B_z}{\partial z} = (1-2v) \frac{\partial B_x}{\partial x}; \quad B_y = 0$$

$$\frac{\partial \beta}{\partial z} = (1-2v) B_z - x \frac{\partial B_x}{\partial z} + z \frac{\partial B_x}{\partial x}$$

Concentrated force in tangential direction on a half space (Cerruti's Problem)



$$r_0^2 = (x-\xi)^2 + (y-\eta)^2 + z^2$$

$$P_x = \lim_{A \rightarrow 0} \int_A P_x d\xi d\eta \quad r_0^2 \rightarrow R^2 = x^2 + y^2 + z^2$$

$$\therefore B_x = \frac{P_x}{2\pi\mu R}$$

$$\therefore \frac{\partial B_x}{\partial x} = -\frac{P_x}{2\pi\mu R^3} x$$

$$\frac{\partial B_x}{\partial z} = -\frac{P_x}{2\pi\mu R^3} z$$

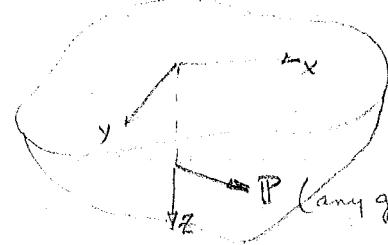
$$\frac{\partial B_z}{\partial z} = -\frac{(1-2v) P_x}{2\pi\mu R^3} x$$

$$\Rightarrow B_z = -\frac{(1-2v) P_x}{2\pi\mu R (R+z)} x$$

$$\text{after subst: } \frac{\partial \beta}{\partial z} = \frac{(1-2v)^2 P_x}{2\pi\mu R (R+z)} x$$

$$\Rightarrow \beta = -\frac{(1-2v)^2 P_x}{2\pi\mu (R+z)} x$$

Caution: All these integrals give integrating functions which we hope involve only rigid body rotation/ translation & can be dropped.

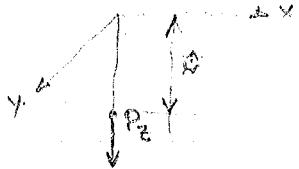


Mindlin Problem: A concentrated force at any point in the interior of a half-space

\mathbf{e}_{xy} is the unit vector in the xy plane

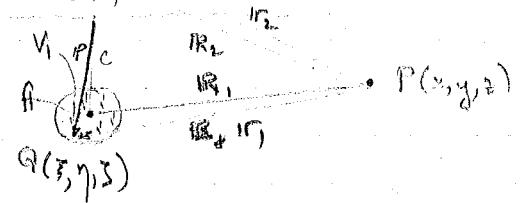
$$\text{while } \mathbf{P} = P_z \mathbf{e}_z + P_{x-y} \mathbf{e}_{x-y}$$

Look at P_z portion of problem



Assume like a Kelvin solution with a distributed body force f in a small volume V w/surface area S

mirrorimage $(0,0;c)$



$$r_1^2 = (x - \xi)^2 + (y - \eta)^2 + (z - \zeta)^2$$

$$r_2^2 = (x - \xi)^2 + (y - \eta)^2 + (z + \zeta)^2$$

$$R_1^2 = (x^2 + y^2) + (z - c)^2$$

$$R_2^2 = (x^2 + y^2) + (z + c)^2$$

$$\rho^2 = \xi^2 + \eta^2 + \zeta^2 \quad \text{and} \quad f = f_z \mathbf{e}_z$$

as $V_1 \rightarrow 0$, $r_1 \rightarrow R_1$, $r_2 \rightarrow R_2$, $\rho \rightarrow c$
define limit as $V_1 \rightarrow 0$ $\int f_z dS dy dz = P_z$

Let $B_x, B_y = 0$ and $B_z, \beta \neq 0$: But $\nabla^2 B_z = -\frac{f}{\mu} = -\frac{f_z}{\mu}$ in V_1
 $\int_0^c f_z dS dy dz \neq P_z$

also $\nabla^2 \beta = \begin{cases} 2f_z/\mu & \text{in } V_1 \\ 0 & \notin V_1 \end{cases}$

Calculate stresses

$$\sigma_{zz} = \frac{\mu}{2(1-\nu)} \left\{ 2(1-\nu) \frac{\partial \beta_z}{\partial z} - z \frac{\partial^2 \beta_z}{\partial z^2} - \frac{\partial^2 \beta}{\partial z^2} \right\}$$

$$\sigma_{zy} = \frac{\mu}{2(1-\nu)} \left\{ (1-2\nu) \frac{\partial \beta_z}{\partial y} - z \frac{\partial^2 \beta_z}{\partial y \partial z} - \frac{\partial^2 \beta}{\partial y \partial z} \right\}$$

$$\sigma_{zx} = \frac{\mu}{2(1-\nu)} \left\{ (1-2\nu) \frac{\partial \beta_z}{\partial x} - z \frac{\partial^2 \beta_z}{\partial x \partial z} - \frac{\partial^2 \beta}{\partial x \partial z} \right\}$$

on bdy $z=0$, $\sigma_{zz}=0$, $\sigma_{zy}=0$, $\sigma_{zx}=0$

$$\text{Let } h = 2(1-\nu) \frac{\partial B_2}{\partial z} - \frac{\partial^2 \beta}{\partial z^2} \Rightarrow \sigma_{zz} = 0$$

$$\text{Now } \nabla^2 h = -2(1-\nu) \frac{\partial f_2}{\partial z} - \frac{1}{\mu} \frac{\partial^2}{\partial z^2} (z f_2) \quad \text{in } V_1$$

now $h=0$ at $z=0$ and $\nabla^2 h = 0$ outside V_1

for h inside V_1 then

$$\therefore h(x, y, z) = \frac{-1}{4\pi\mu} \int_{V_1} \left(\frac{1}{r_1} - \frac{1}{r_2} \right) \nabla^2 h \, dV_1 = \frac{1}{4\pi\mu} \int_{V_1} \left(\frac{1}{r_1} - \frac{1}{r_2} \right) \left[+2(1-\nu) \frac{\partial f_2(\xi, \eta, \zeta)}{\partial \xi} + \frac{\partial^2}{\partial \xi^2} (z f_2) \right] d\xi d\eta d\zeta$$

Now carry out the operation & let $V_1 \rightarrow 0$

$$h = \lim_{V_1 \rightarrow 0} \frac{1}{4\pi\mu} \int_{V_1} \left(\frac{1}{r_1} - \frac{1}{r_2} \right) \left[2(1-\nu) \frac{\partial f_2}{\partial \xi} + \frac{\partial^2}{\partial \xi^2} (z f_2) \right] d\xi d\eta d\zeta$$

$$= \frac{P_e}{4\pi\mu} \left[2(1-\nu) \frac{\partial}{\partial z} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) + c \frac{\partial^2}{\partial z^2} \left(\frac{1}{R_1} - \frac{1}{R_2} \right) \right]$$

$$\Rightarrow 2(1-\nu) \frac{\partial B_2}{\partial z} - \frac{\partial^2 \beta}{\partial z^2} = \frac{P_e}{4\pi\mu} \left[2(1-\nu) \left(\frac{1}{R_1} + \frac{1}{R_2} \right) + c \frac{\partial^2}{\partial z^2} \left(\frac{1}{R_1} - \frac{1}{R_2} \right) \right]$$

$$\therefore 2(1-\nu) B_2 - \frac{\partial \beta}{\partial z} = \frac{P_e}{4\pi\mu} \left[2(1-\nu) \left(\frac{1}{R_1} + \frac{1}{R_2} \right) + c \frac{\partial}{\partial z} \left(\frac{1}{R_1} - \frac{1}{R_2} \right) \right]$$

5/7/79

Continuation of the previous problem.

$$\text{Consider } \sigma_{zx} \Big|_{z=0} = 0 = (1-2\nu) \frac{\partial B_2}{\partial x} - \frac{\partial^2 \beta}{\partial x \partial z}$$

$$\text{let } h = (1-2\nu) \frac{\partial B_2}{\partial x} - \frac{\partial^2 \beta}{\partial x \partial z} \quad \text{for } z=0 \quad h=0 \quad \nabla^2 B_2 = -f_2/\mu \quad \nabla^2 \beta = z f_2$$

$$\nabla^2 h = -\frac{(1-2\nu)}{\mu} \frac{\partial f_2}{\partial x} - \frac{1}{\mu} \frac{\partial^2}{\partial x \partial z} (z f_2) \quad \text{Now using Green's theorem}$$

$$\text{thus } h(x, y, z) = \frac{-1}{4\pi\mu} \int_{V_1} \left[-G \left[(1-2\nu) \frac{\partial f_2(\xi, \eta, \zeta)}{\partial \xi} + \frac{\partial^2}{\partial \xi^2} (z f_2) \right] d\xi d\eta d\zeta \right]$$

$$\text{Now } \int_{V_1} G \frac{\partial f_2}{\partial \xi} d\xi d\eta d\zeta = \int_S G f_2 d\eta d\zeta - \int_{V_1} f_2 \frac{\partial G}{\partial \xi} d\xi d\eta d\zeta \quad G = \left(\frac{1}{r_1} - \frac{1}{r_2} \right)$$

$= 0$ on surface since $f_2 \neq 0$ on S

$$\frac{\partial G}{\partial \xi} = \frac{\partial}{\partial \xi} \left(\frac{1}{r_1} - \frac{1}{r_2} \right) = \frac{x-\xi}{r_1^3} - \frac{x-\xi}{r_2^3} \quad \text{since } r_{1,2} = \sqrt{(x-\xi)^2 + (y-\eta)^2 + (z-\zeta)^2}$$

$$\lim_{\xi \rightarrow 0} - \int f_2 \left[\frac{x-\xi}{r_1^3} - \frac{x-\xi}{r_2^3} \right] d\xi d\eta d\zeta \quad (\text{w/ } \xi \rightarrow 0 \quad r_1 \rightarrow R_1, \quad r_2 \rightarrow R_2) \quad \int f_2 d\xi d\eta d\zeta \rightarrow P_z$$

$$\Rightarrow -P_z \left[\frac{x}{R_1^3} - \frac{x}{R_2^3} \right] = P_z \frac{\partial}{\partial x} \left[\frac{1}{R_1} - \frac{1}{R_2} \right] = \int_V G \frac{\partial f_2}{\partial \xi} d\xi d\eta d\zeta$$

Integration by parts in 2 dim.

we must now find $\int \int G \frac{\partial^2 (\Sigma f_2)}{\partial z \partial y} dz dy ds = \int_s \int G \frac{\partial}{\partial y} (\Sigma f_2) dy ds - \int_v \int \frac{\partial G}{\partial z} \frac{\partial (\Sigma f_2)}{\partial y} dz dy ds$

integrate again $= - \int_s \int \frac{\partial G}{\partial z} (\Sigma f_2) dy ds + \int_v \int \Sigma f_2 \frac{\partial^2 G}{\partial z \partial y} dz dy ds$
 $f_2 = 0$ on surface

$$\frac{\partial^2 G}{\partial z \partial y} = \frac{\partial}{\partial z} \left(\frac{\partial G}{\partial y} \right) = \frac{\partial}{\partial z} \left[\frac{x-y}{r_1^3} - \frac{x-z}{r_2^3} \right] = \beta(x-y) \left[\frac{z-y}{r_1^5} + \frac{z-y}{r_2^5} \right]$$

take limit as $v_i \rightarrow 0$ of $\int_v \int \Sigma f_2 \frac{\partial^2 G}{\partial z \partial y} dz dy ds$ (w/ $\Sigma \rightarrow c$ $\int f_2 dV = P_z$
 $\int \int \rightarrow 0$ $r_1 \rightarrow R_1$, $r_2 \rightarrow R_2$)

$$\therefore \Rightarrow \beta x \left[\frac{z-c}{R_1^5} + \frac{z+c}{R_2^5} \right] P_z \cdot c = -c P_z \frac{\partial}{\partial z} \left[\frac{x}{R_1^3} + \frac{x}{R_2^3} \right] = c P_z \frac{\partial^2}{\partial z \partial x} \left[\frac{1}{R_1} + \frac{1}{R_2} \right]$$

$$\therefore h(x, y, z) = (1-2v) \frac{\partial B_z}{\partial x} - \frac{\partial \beta}{\partial x \partial z} = \frac{P_z}{4\pi\mu} \left\{ (1-2v) \frac{\partial}{\partial x} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) + c \frac{\partial^2}{\partial x \partial z} \left[\frac{1}{R_1} + \frac{1}{R_2} \right] \right\}$$

3rd BC gives something w/ $\frac{\partial}{\partial y}$ replacing $\frac{\partial}{\partial x}$ giving same result.

from above $\int w/r x \Rightarrow (1-2v) B_z - \frac{\partial \beta}{\partial z} = \frac{P_z}{4\pi\mu} \left\{ (1-2v) \left(\frac{1}{R_1} + \frac{1}{R_2} \right) + c \frac{\partial}{\partial z} \left[\frac{1}{R_1} + \frac{1}{R_2} \right] \right\}$

use this w/ $2(1-v) B_z - \frac{\partial \beta}{\partial z} = \frac{P_z}{4\pi\mu} \left[2(1-v) \left(\frac{1}{R_1} + \frac{1}{R_2} \right) + c \frac{\partial}{\partial z} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) \right]$ from before

to get $B_z = \frac{P_z}{4\pi\mu} \left[\frac{1}{R_1} + \frac{3-4v}{R_2} - 2c \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) \right]$

using one of the other eqns to get

$$\left| \frac{\partial \beta}{\partial z} = \frac{P_z}{4\pi\mu} \left[\frac{4(1-2v)(1-v)}{R_2} - c \frac{\partial}{\partial z} \left(\frac{1}{R_1} \right) - (3-4v)c \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) \right] \right|$$

integrate w/ rt \bar{z} to get

$$\left| \beta = \frac{P_z}{4\pi\mu} \left[-\frac{c}{R_1} - \frac{c(3-4v)}{R_2} + 4(1-2v)(1-v) \log(R_2 + z + c) \right] \right|$$

as $c \rightarrow \infty$ we should recover Kelvin Soln. (Body effects no longer important)
 $c \rightarrow 0$ " " " Boundary Soln / or Cerruti

Mindlin solved problem in different manner - assume a force at $(0, 0, c)$ Kelvin
 Problem + nuclei of strain in $-z$ region. \therefore we annihilate $\sigma_{zx}, \sigma_{zy}, \sigma_{zz}$
 at $z=0$

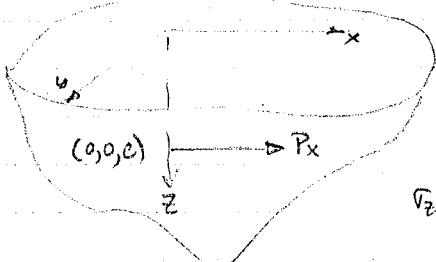
this is what Mindlin found

1. A singular force at $(0,0,+c)$
2. A " " " " $(0,0,-c)$
3. " double " " $(0,0,c)$
4. a center of compression at $(0,0,-c)$
5. A line of centers of compression extending from $z = -c$ to $z = \infty$
6. A doublet at $(0,0,-c)$

5/9/75

Exam: Friday - Notes & HW only allowed

Mindlin Problem - Part II - Force II to X axis



Not longer have axial symmetry \Rightarrow only $B_y \neq 0$
need B_x, B_z, β

we write tractions on surface

$$\tau_{zz} = \frac{\mu}{2(1-\nu)} \left[2(1-\nu) \frac{\partial B_z}{\partial z} + 2\nu \frac{\partial B_x}{\partial x} - x \frac{\partial^2 B_x}{\partial z^2} - \frac{\partial^2 \beta}{\partial z^2} \right] = 0$$

$$\tau_{zy} = \frac{\mu}{2(1-\nu)} \left[(1-2\nu) \left(\frac{\partial B_z}{\partial y} + \frac{\partial B_x}{\partial z} \right) - x \frac{\partial^2 B_x}{\partial x \partial z} - \frac{\partial^2 \beta}{\partial z \partial x} \right] = 0$$

$$(*) \quad \tau_{zy} = \frac{\mu}{2(1-\nu)} \left[(1-2\nu) \left(\frac{\partial B_z}{\partial y} \right) - x \frac{\partial^2 B_x}{\partial y \partial z} - \frac{\partial^2 \beta}{\partial y \partial z} \right] = 0$$

Consider P_x is a distributed force f_x over a small volume

remember $\nabla^2 B_x = f_x / \mu$ $\nabla^2 B_z = 0$ $\nabla^2 \beta = -f_x x / \mu$

$$P_x = \lim_{V \rightarrow 0} \int f_x dV$$

$$\text{Now take } \frac{\partial \tau_{zy}}{\partial y} - \frac{\partial \tau_{zy}}{\partial x} = 0 \Rightarrow \frac{\partial^2 B_x}{\partial y \partial z} = 0 \text{ on } z=0 \Rightarrow \frac{\partial B_x}{\partial z} = 0$$

$$\text{In general } \nabla^2 B_x = -f_x / \mu \Rightarrow \nabla^2 \frac{\partial B_x}{\partial z} = -\frac{1}{\mu} \frac{\partial f_x}{\partial z}$$

use green's theorem on $\frac{\partial B_x}{\partial z}$; $\nabla^2 \left(\frac{\partial B_x}{\partial z} \right) = -\frac{1}{\mu} \frac{\partial f_x}{\partial z}$ in V & $\frac{\partial B_x}{\partial z} = 0$ on S

$$\therefore \frac{\partial B_x}{\partial z}(x, y, z) = \frac{1}{4\pi\mu} \iint_V G \frac{\partial f_x}{\partial z} d\xi dy dz \quad G = \frac{1}{R_1} - \frac{1}{R_2} \quad \text{since we integrate by parts}$$

$$\text{now take limit } \rightarrow 0 \Rightarrow \frac{\partial B_x}{\partial z} = \frac{1}{4\pi\mu} P_x \frac{\partial}{\partial z} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) \Rightarrow B_x = \frac{P_x}{4\pi\mu} \left(\frac{1}{R_1} + \frac{1}{R_2} \right)$$

look at (*) since $\frac{\partial B_x}{\partial z} = 0$ on bdy $\Rightarrow \tau_{zy} = (1-2\nu) \frac{\partial B_z}{\partial y} - \frac{\partial^2 \beta}{\partial y \partial z} = 0$ or

$$(1-2\nu) B_z = \frac{\partial \beta}{\partial z} \quad \text{integrating w/ y:}$$

$$\text{in } z \quad \nabla^2 \left[(1-2v) B_2 - \frac{\partial \beta}{\partial z} \right] = -\frac{x}{\mu} \frac{\partial f_x}{\partial z} \quad \text{since } \nabla^2 B_2 \geq 0 \text{ as shown before}$$

now use greens fn. on $(1-2v) B_2 - \frac{\partial \beta}{\partial z} = h \text{ w/ } h=0 \text{ on } z=0 \text{ & } \nabla^2 h = -\frac{x}{\mu} \frac{\partial f_x}{\partial z}$

$$\therefore (1-2v) B_2 - \frac{\partial \beta}{\partial z} = \frac{1}{4\pi\mu} \int_V G \frac{\partial^2 f_x}{\partial z^2} d\zeta dy dz$$

in limit as $V \rightarrow 0 \quad \zeta \rightarrow 0 \quad G \rightarrow 0 \quad \int f_x d\zeta dy dz \rightarrow P_x \quad \text{do this by parts}$

$$(4) \quad \therefore (1-2v) B_2 - \frac{\partial \beta}{\partial z} = 0 \text{ everywhere}$$

need another relation between B_2, β use σ_{zz}

$$\text{let } X = 2(1-v) \frac{\partial B_2}{\partial z} + 2v \frac{\partial B_x}{\partial x} - x \frac{\partial^2 B_x}{\partial z^2} - \frac{\partial \beta}{\partial z^2} \quad h=0 \text{ on } z=0$$

$$\textcircled{14} \quad \text{rewrite } X = 2(1-v) \frac{\partial B_2}{\partial z} - \frac{\partial^2 \beta}{\partial z^2} - (1-2v) \frac{\partial B_x}{\partial x} - \frac{P_x c}{2\pi\mu} \frac{\partial^2}{\partial x \partial z} \left(\frac{1}{R_2} \right) \text{ using } B_x$$

$$\nabla^2 X = -\frac{x}{\mu} \frac{\partial^2 f_x}{\partial z^2} + \frac{(1-2v)}{\mu} \frac{\partial f_x}{\partial x}$$

$$\text{In } z \text{ using greens fn: } X = \frac{1}{4\pi\mu} \int_V G \zeta \frac{\partial^2 f_x}{\partial z^2} d\zeta dy dz = \frac{(1-2v)}{4\pi\mu} \int_V G \frac{\partial f_x}{\partial z} d\zeta dy dz \\ = 0 \text{ in limit after IBP} \quad \zeta \rightarrow 0 \text{ and } G=0 \text{ on Bdry} \quad \text{integrate by parts & take limit & } V \rightarrow 0$$

$$X = \frac{1-2v}{4\pi\mu} \int f_x \frac{\partial G}{\partial z} d\zeta dy dz : \text{take limit} = -\frac{(1-2v) P_x}{4\pi\mu} \frac{\partial}{\partial x} \left(\frac{1}{R_1} - \frac{1}{R_2} \right)$$

put this back into $\textcircled{14}$ and integrate wrt x to get

$$(++) \quad 2(1-v) B_2 - \frac{\partial \beta}{\partial z} = \frac{(1-2v) P_x x}{2\pi\mu R_2(R_2+z+c)} - \frac{P_x c x}{2\pi\mu R_2^3}$$

using (4) and (++) we can get

$$\boxed{B_2 = \frac{P_x \chi}{2\pi\mu R_2} \left[\frac{1-2v}{R_2+z+c} + \frac{c}{R_2^2} \right]}$$

$$\boxed{\beta = \frac{P_x (1-2v) x}{2\pi\mu (R_2+z+c)} \left[\frac{c}{R_2} - (1-2v) \right]}$$

Mindlin solved this using following symmetries (superposition of nuclei of state)

1) Single file at $(0, 0, c)$

2) " " at $(0, 0, -c)$ same dir

3) Doublet at $(0, 0, -c)$

4) Semi infinite line of doublets from $z=c$ to $z=-c$ w/ strength proportional to distance from $-c$

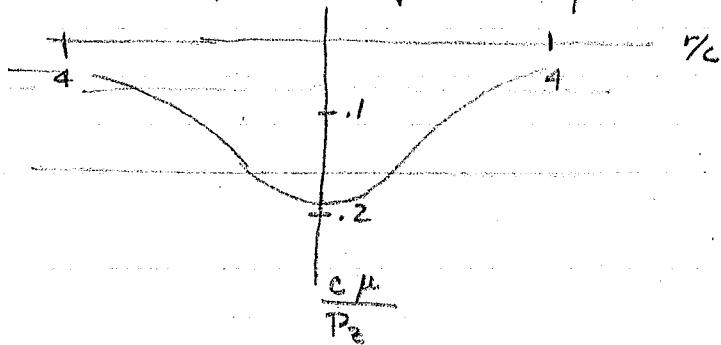
- 5) Double force w/moment at $(0, 0, -c)$
 6) Semi infinite line of double forces w/moment extending from $(-c, -\infty)$

In first problem
of mudline



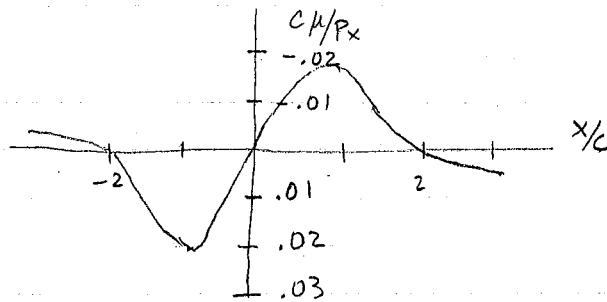
Suppose you want settlement.

Vertical displacement of Boundary in Part I



5/14/79

Part II Settlement - Vertical deflection of Boundary in Part II



Why wasn't Mindlin's problem solved by the following?

Kelvin Problem

→ calculate tractions
 ↓ apply opposite tractions
 by use of Kelvin-Boussinesq/Cerruti Problem.

Mindlin's background was limited in math - he tried it this way first but the integrals were hard so he abandoned it.

Stresses & disp. are found in his paper "Force at a Point in the Interior of a Semi-Infinite Solid" - R. D. Mindlin Physics, Vol 7, May 1936 Pg 195-202.

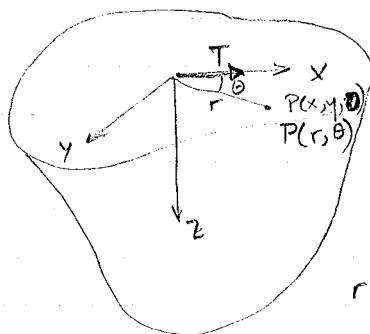
Mindlin problem is an intermediate case of the Kelvin (w/ BC far away) and the Boussinesq problem (w/ BC in fact plays a part).

Last few lectures - possible topics

Thermo elasticity

- Dynamic next fall class
- Large Deformation
- Complex potential } together
- Fracture Mechanics

Displacements of $z=0$ plane (x-y plane) for Cerruti Problem.



Consider tangential forces distributed over a circle in the surface of half-space as a prelude to contact problem: (using Cerruti Problem)

$$\text{Using Cerruti Problem } u_x(x, y, 0) = \frac{T}{2\pi\mu r} - \nu \frac{\partial}{\partial x} \left(\frac{Tx}{2\pi\mu r} \right)$$

$$r = \sqrt{x^2 + y^2}$$

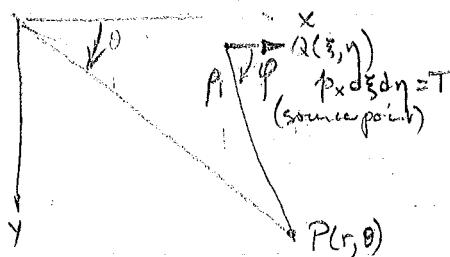
$$u_y(x, y, 0) = -\nu \frac{\partial}{\partial y} \left(\frac{Tx}{2\pi\mu r} \right)$$

thus

$$\begin{aligned} u_x(x, y, 0) &= \frac{T}{2\pi\mu r} \left[1 - \nu + \frac{\nu x^2}{r^2} \right] & \text{in polar coords.} &= \frac{T}{2\pi\mu r} [2 - \nu + \nu \cos 2\theta] \\ u_y(x, y, 0) &= \frac{\nu xy}{2\pi\mu r^3} & &= \frac{\nu}{2\pi\mu r} \frac{\sin 2\theta}{2} \end{aligned}$$

Now the above is only good when T is applied at $(0, 0, 0)$

if T is applied at (ξ, η) then



since p_x is an elementary load then

$$\text{we can write } du_x = \frac{p_x}{2\pi\mu r} [2 - \nu + \nu \cos 2\phi] d\xi dy$$

$$du_y = \frac{p_x \nu}{2\pi\mu r} \sin 2\phi d\xi dy$$

$$u_x = \left\{ \int_A \frac{p_x [2 - \nu + \nu \cos 2\phi]}{2\pi\mu r} d\xi dy \right\}$$

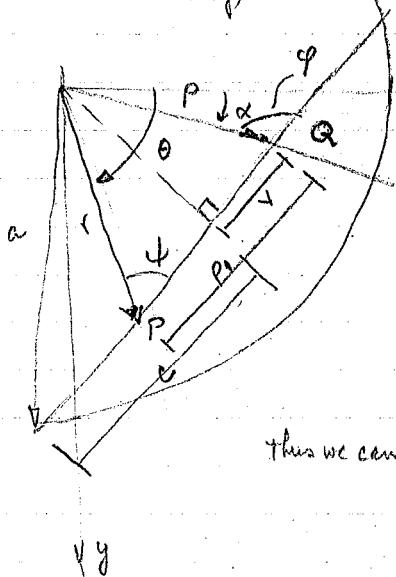
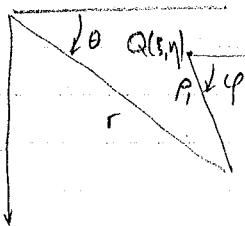
$$u_y = \left\{ \int_A \frac{p_x \nu \sin 2\phi}{2\pi\mu r} d\xi dy \right\}$$

where $p_x = p_x(\xi, \eta)$

Take Home final

5/16/79

Continuation of the last class



$$u_x = \int_A \frac{1}{4\pi\mu} p_x \frac{2 - v + v \cos 2\phi}{\rho_1} d\bar{x} dy$$

$$u_y = \int_A \frac{v}{4\pi\mu} p_x \frac{\sin 2\phi}{\rho_1} d\bar{x} dy$$

$$\phi = \theta + \psi$$

$$v = r \cos \psi - \rho_1$$

$$dv = d\rho_1$$

$$c^2 = a^2 - r^2 \sin^2 \psi$$

$$v^2 = \rho^2 - r^2 \sin^2 \psi$$

$$d\bar{x} dy = \rho_1 d\rho_1 d\psi \\ = \rho_1 dv d\psi$$

thus we can rewrite

$$u_x = \int_{-\pi/2}^{\pi/2} \int_{-c}^c p_x [2 - v + v \cos 2(\theta + \psi)] dv d\psi$$

$$u_y = \frac{v}{4\pi\mu} \int_{-\pi/2}^{\pi/2} \int_{-c}^c p_x [\sin 2\phi] dv d\psi$$

the above is for any p_x

$$\text{for } p_x = \frac{T}{2\pi a} \frac{1}{\sqrt{a^2 - \rho^2}} \text{ w/ } \int p_x dA = T$$

$$\therefore u_x = \frac{T}{8\pi^2 \mu a} \int_{-\pi/2}^{\pi/2} \int_{-c}^c \frac{2 - v + v \cos 2(\theta + \psi)}{\sqrt{c^2 - v^2}} d\psi dv$$

$$= \frac{T}{8\pi^2 \mu a} \int_{-\pi/2}^{\pi/2} [2 - v + v \cos 2(\theta + \psi)] \sin^{-1} \frac{v}{c} \int_{-c}^c d\psi$$

$$= \frac{T(2-v)}{8\mu a} + \frac{TV}{8\pi\mu a} \left[\frac{1}{2} \sin 2(\theta + \psi) \right]_{-\pi/2}^{\pi/2}$$

$$u_x = \frac{(2-v)T}{8\mu a}$$

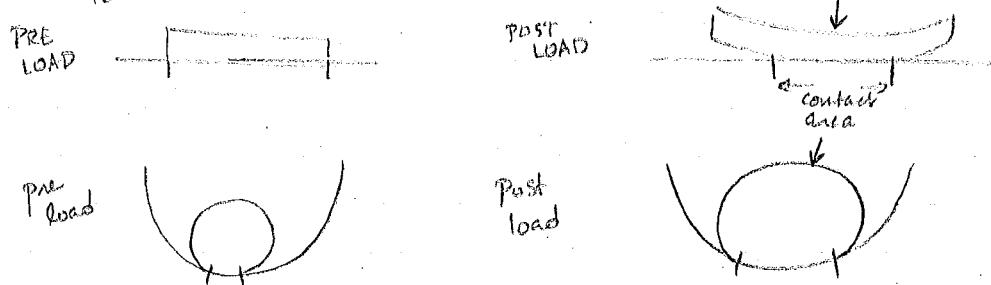
drift in direction of load is a constant i.e.
circle moves as a rigid body in x direction

$$u_y = \frac{V_2 T}{8 \pi \mu a} \int_{-\frac{\pi}{2}}^{\frac{\pi}{2}} \frac{\sin 2\phi}{\sqrt{c^2 - v^2}} dv d\phi = 0$$

thus the circle will displace in x direction as a rigid body only. No displacement in y direction. Similar to the Case 2 of Boussinesq problem.

This is important when we talk about CONTACT problems.

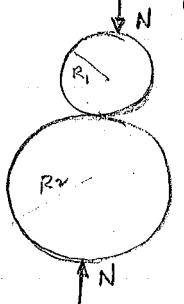
In contact problems what we need to know is the contact surface i.e.



The larger the load the smaller the contact area in beam problem

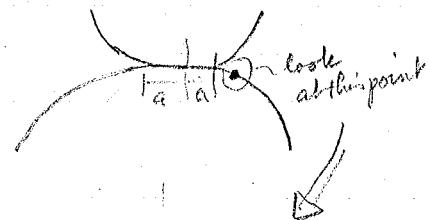
Contact Problems - Hertz about 100 years ago

Ball Bearings under applied loads



We assume

1. Contact area is flat
2. " " " circular
3. " circle radius $\ll R_1, R_2$ (look at local problem)
4. N is applied remotely & W does not affect contact problem.



$$\cos \delta = \sqrt{\cos^2 \delta = 1 - \sin^2 \delta} = \sqrt{1 - \left(\frac{r_1}{R_2}\right)^2}$$

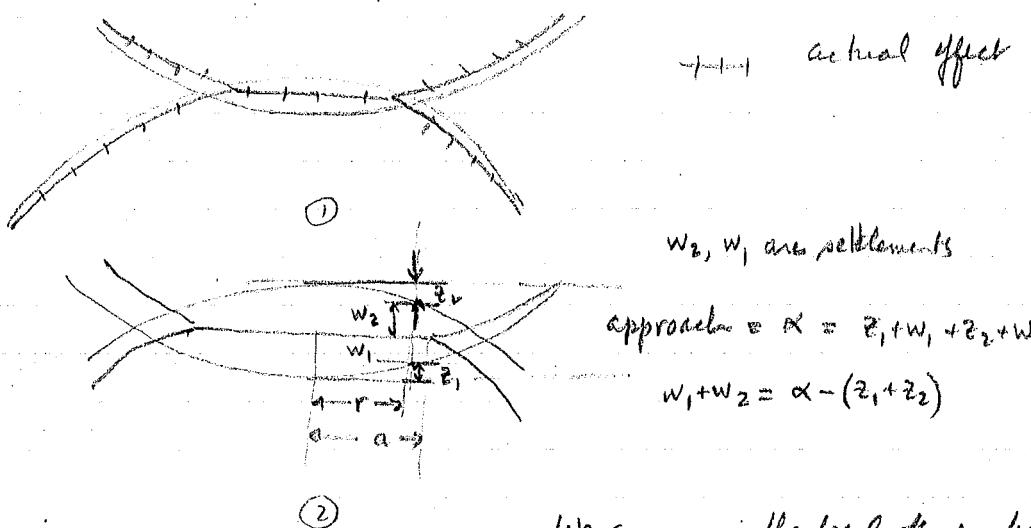
$$z_2 = R_2(1 - \cos \delta) = R_2 \left(1 - \sqrt{1 - \left(\frac{r_1}{R_2}\right)^2}\right)$$

R_2 if we assume r_1 and R_2 are small use binomial

$$z_2 = R_2 \left[1 - \left(1 - \frac{1}{2} \left(\frac{r_1}{R_2}\right)^2 + \dots\right)\right] = \frac{r_1^2}{2R_2}$$

define similarly z_1 on upper sphere $z_1 = \frac{r_1^2}{2R_1}$

Now sketch sphere if they penetrated & did not deform



w_2, w_1 are settlements

$$\text{approximate } \alpha = z_1 + w_1 + z_2 + w_2$$

$$w_1 + w_2 = \alpha - (z_1 + z_2)$$

②

We assume in the local that bearing ② looks like a half space to ① and no shear on the face. We can thus assume contact area of sphere is that of problem in a half space thus we can use settlements

5/19/79

Friday - 1 June : Take Home will be handed out and due the following Monday or Tuesday

$$w_1 + w_2 = \alpha - (z_1 + z_2) \quad \text{Recall } w = \frac{1-\nu}{2\pi\mu} \int_{-R_2}^{R_2} \int_{-c}^c p_2(v, \psi) dv d\psi$$

$$\therefore \left[\frac{1-\nu_1}{2\mu_1\pi} + \frac{1-\nu_2}{2\mu_2\pi} \right] \int_{-R_2}^{R_2} \int_{-c}^c p_2(v, \psi) dv d\psi = \alpha - \frac{r^2}{2} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) \quad \begin{matrix} \text{look at other} \\ \text{side for how } \nu_1, \nu_2 \\ \text{were obtained} \end{matrix}$$

look at case 3 of Boussinesq problem. : Recall for $p_2 = \frac{3N\sqrt{a^2 - r^2}}{2\pi a^3}$

$$\text{where } \iint p_2 dA = N \quad \text{then} \quad w_i = \frac{3(1-\nu_i)N}{16a^3\mu_i} (2a^2 - r^2) = \frac{3k_i N \pi}{8a^3} (2a^2 - r^2)$$

$$\text{if we let } \frac{1-\nu_i}{2\mu_i} = k_i \quad \text{thus } (k_1 + k_2) \left[\frac{3N\pi}{8a^3} (2a^2 - r^2) \right] = \alpha - \frac{r^2}{2} \left(\frac{1}{R_1} + \frac{1}{R_2} \right)$$

$(k_1 + k_2) \frac{3N\pi}{8a^3} = \alpha$ and $(k_1 + k_2) \frac{3N\pi}{8a^3} = \frac{1}{2} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) = \frac{1}{2} \left(\frac{R_1 + R_2}{R_1 R_2} \right)$
 k_1, k_2 are material constants, R_1, R_2 are physical const and N is applied load

Solve for α

$$\therefore \left(\alpha = \sqrt[3]{\frac{(k_1 + k_2) 3N\pi R_1 R_2}{4(R_1 + R_2)}} = K_1 N^{1/3} \right)$$

$$\left| \alpha = \sqrt[3]{\frac{9N^2\pi^2(k_1 + k_2)^2(R_1 + R_2)}{16R_1 R_2}} = \frac{K_1 N^{1/3}}{2} \right|$$

$$\left| \begin{array}{l} \text{if } \\ \sigma = \epsilon E \\ \epsilon = \frac{1}{E} \sigma \end{array} \right.$$

$$\sigma = E \epsilon \quad E \text{ is stiffness}$$

$$\epsilon = \frac{1}{E} \sigma \quad \frac{1}{E} \text{ is compliance}$$

we can define compliance $\frac{dx}{dN} = \frac{2}{3} k_2 N^{-\frac{1}{3}}$

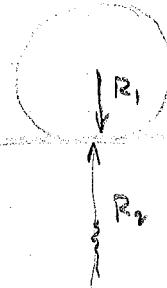
$$C = \frac{dx}{dN} = \frac{3}{6 R_1 R_2 N} \sqrt{\pi^2 (R_1 + R_2)^2 (R_1 + R_2)}$$

pressure distribution at the contact surface

to find stresses below surface we can use boundary problem superposed -
use boundary problem to get stresses below surface ~~but up to~~ up to what depth can we use it?

Problem

for a sphere on a flat surface take one of the radii to be a



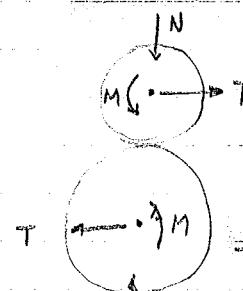
Problem: Two like spheres $R_1 = k_2 = k$, $R_1 \leq R_2$

$$C = \frac{3}{6 R^2 N} \sqrt{4 \pi^2 k^2 \cdot 2k} = \frac{3}{6} \sqrt{\frac{4 \pi^2 k^2}{R^2 N}} = \frac{\pi k}{a}$$

$$a = \sqrt[3]{\frac{3N\pi(2k) \cdot R^2}{8R}} = \sqrt[3]{\frac{3N\pi R k}{4}} = \frac{1-\nu}{2M}$$

An extension of the contact problem

Contact of 2 spheres under Normal & Tangential loads



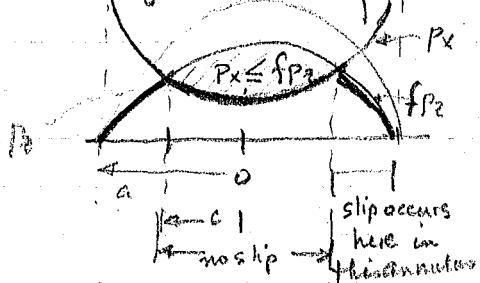
First N is applied giving rise to a contact circle
Now we apply T

look at The T problem only

$$\text{BC. } ① \text{ For } r < a \quad \tau_{zz}|_{r=a} = 0$$

② assume displ (produced by T) in y direction $u_y = 0$ (no warping of contact circle)
 ϵ_{xx} must be anti-symmetric due to T ; in the contact circle $\epsilon_{xx} = 0 \Rightarrow u_x = \text{const}$
 all this wrt $z=0$

Recall that for $u_y = 0$ and $u_x = \text{const}$ was produced on a half plane with
 tangential loadip on circle (after Gerini problem) $p_x = \frac{I}{2\pi a} (a^2 - r^2)^{1/2}$



is this a correct loadip? No since as $p_x \uparrow$
 the two bodies will slip. How do we find the area of slip? Recall law of statics friction

$$p_x \leq f p_z \quad p_z = \text{normal to surface of contact.}$$

where f is dependent on surface texture only

for $r \ll c$ $\sigma_{zz} = 0$, $u_y = 0$ $u_x = \text{const}$

$c < r < a$ $\sigma_{zz} = 0$, $\sigma_{zx} = f p_z$, $\sigma_{zy} = 0$ $<$ since slip has taken place restrictions
 $r > a$ $\sigma_{zz} = \sigma_{zy} = \sigma_{zx} = 0$ traction free b.c.

if $p_x = f p_z$ call this the (' problem)

$$p_x' = f p_z = \frac{3fN}{2\pi a^3} \sqrt{a^2 - r^2} \quad \text{let } fN = T'$$

For $c < r < a$

$$u_x' = \frac{3T'}{64c^3\mu} [2(2-\nu)(2a^2 - r^2) + \nu r^2 \cos 2\theta]$$

$$u_y' = \frac{3T' \nu r^2 \sin 2\theta}{64c^3\mu}$$

For $r \ll c$

+ to correct for $r \ll c$ add ("problem") such that $u_x' + u_x'' = \text{const}$
 $u_y' + u_y'' = 0$

$$\text{let } p_x'' = -\frac{3T''}{2\pi c^3} (c^2 - r^2)$$

$$u_x'' = -\frac{3T''}{64c^3\mu} [2(2-\nu)(2c^2 - r^2) + \nu r^2 \cos 2\theta]$$

$$u_y'' = -\frac{3T'' \nu r^2 \sin 2\theta}{64c^3\mu} \quad \text{take } T'' = T' \frac{c^3}{a^3} \text{ so that } u_y' + u_y'' = 0$$

$$u_x = u_x' + u_x'' = \frac{3T'}{64a^3\mu} [2(2-\nu)(2a^2 - 2c^2)] = \frac{3T'}{16a\mu} (2-\nu) \left(1 - \frac{c^2}{a^2}\right) = \text{const}$$

$$u_y = u_y' + u_y'' = 0$$

$$p_x = p_x' + p_x'' = \frac{3fN}{2\pi a^3} [(a^2 - r^2)^{1/2} - (c^2 - r^2)^{1/2}]$$

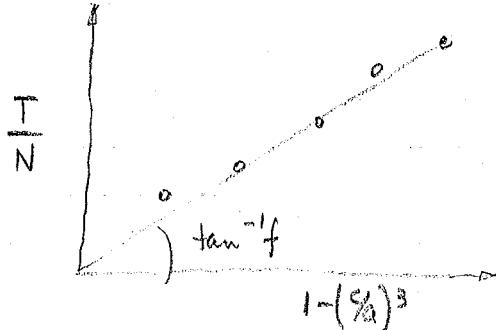
$$T = \int p_x dA = \frac{3fN}{\pi a^3} \iiint p_x r dr d\theta = \frac{3fN}{a^3} \left[\int_0^a r \sqrt{a^2 - r^2} dr - \int_0^c r \sqrt{c^2 - r^2} dr \right]$$

$$T = \frac{fN}{a^3} [a^3 - c^3] = fN(1 - \frac{c^3}{a^3})$$

solve for c

$$\left| c = a \sqrt[3]{1 - \frac{T}{fN}} \right|$$

$$\text{for } f \boxed{\left| f = \frac{T}{N(1 - (\frac{c}{a})^3)} \right|}$$



5/23/79

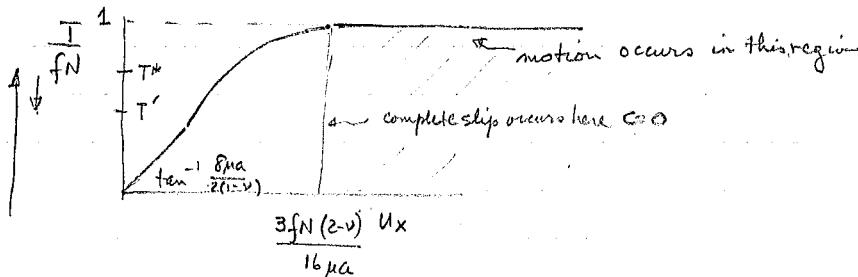
Now continuing

the horiz dip

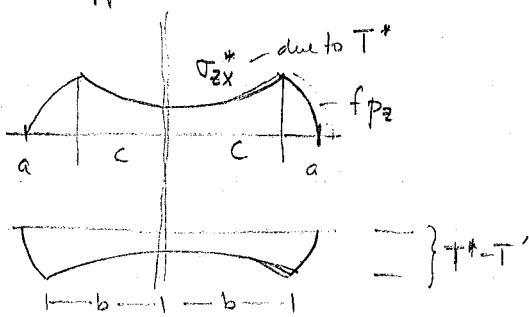
$$u_x = \frac{3fN(2-v)}{16\mu a} \left[1 - \frac{c^2}{a^2} \right] = \frac{3fN(2-v)}{16\mu a} \left[1 - \left\{ 1 - \frac{T}{fN} \right\}^{2/3} \right]$$

Calculate the compliance in shear

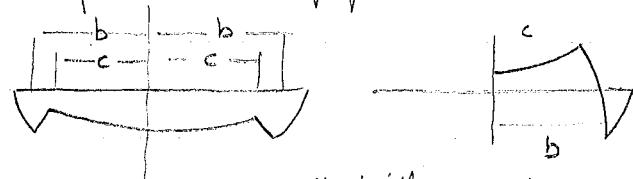
$$\frac{du_x}{dT} = \frac{2-v}{8\mu a} \left[1 - \frac{T}{fN} \right]^{-1/3}$$



Now suppose we increase T to T^* and then decrease from T^* to T' .



We now argue that by applying T to T^* and then decreasing to T' is a superposition

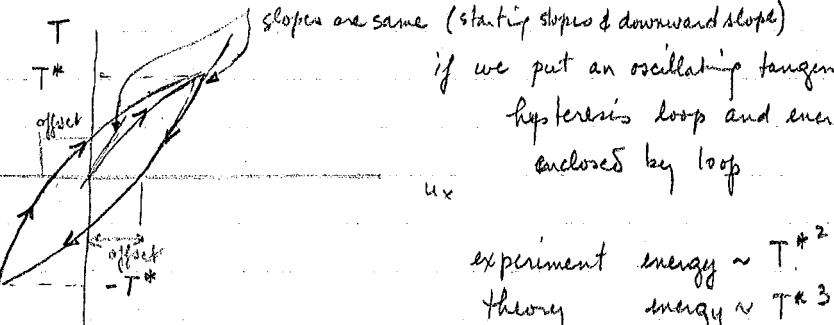


results in either one of these depending on the magnitudes of T^* and $T^* - T'$

by superposition we can find $b = a \left(1 - \frac{T^* - T'}{2fN}\right)^{1/3}$

$$u_x = \frac{3(2-\nu)fN}{16\mu a} \left[2 \left(1 - \frac{T^* - T'}{2fN}\right)^{2/3} - \left(1 - \frac{T^*}{fN}\right)^{2/3} - 1 \right]$$

displacement of adhered section of contact ~~area~~ circle in horiz direction.



Energy dissipated per cycle (for small T/fN)

$$E_d = \frac{(2-\nu) T^{*3}}{36\mu a fN} = \int T du_x$$

All these problems are dependent on order of application

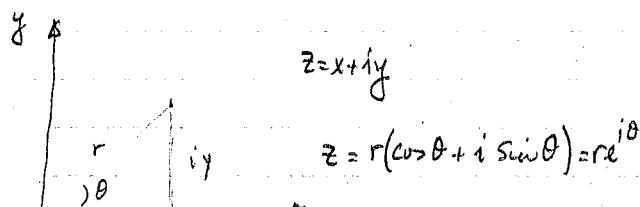
Mechanics of Contact Between Deformable Bodies DePater, Kalker 1975

5/25/79

Fracture Problems

Fathers of Fracture Problems : Kolosov Muskhelishvili

Prelude : plane problems given $f(z)$



$$\frac{\partial}{\partial x} f(z) = \frac{\partial f}{\partial z} \frac{\partial z}{\partial x} = \frac{\partial f}{\partial z}$$

$$\frac{\partial}{\partial y} f = \frac{\partial f}{\partial z} \frac{\partial z}{\partial y} = i \frac{\partial f}{\partial z} = i \frac{\partial f}{\partial x}$$

$$f(z) = \alpha(x, y) + i\beta(x, y) \Rightarrow$$

conjugate
harmonic fn. of x

$$\frac{\partial f}{\partial x} = \frac{\partial \alpha}{\partial x} + i \frac{\partial \beta}{\partial x}$$

$$\frac{\partial f}{\partial y} = \frac{\partial \alpha}{\partial y} + i \frac{\partial \beta}{\partial y}$$

$$z = x + iy$$

$$z = r(\cos \theta + i \sin \theta) = re^{i\theta}$$

$$\frac{\partial \alpha}{\partial x}$$

$$\frac{\partial \beta}{\partial x}$$

$$\frac{\partial \alpha}{\partial y}$$

$$\frac{\partial \beta}{\partial y}$$

$$\frac{\partial \alpha}{\partial x}$$

$$\frac{\partial \beta}{\partial x}$$

$$\frac{\partial \alpha}{\partial y}$$

$$\frac{\partial \beta}{\partial y}$$

\Rightarrow Cauchy Riemann Eqns: $\frac{\partial \alpha}{\partial x} = \frac{\partial \beta}{\partial y}$; $\frac{\partial \alpha}{\partial y} = -\frac{\partial \beta}{\partial x}$

satisfactions implies f is analytic and has all derivatives

and α, β satisfy $\nabla^2 \alpha = \nabla^2 \beta = 0$

Consider $\psi(x, y)$ -

what is $\nabla^2(\chi\psi) = \chi\nabla^2\psi + 2\frac{\partial\psi}{\partial x}$

if $\nabla^2\psi = 0$ then $\chi\psi$ is biharmonic

also $y\psi$ is biharmonic

$$\nabla^2\psi = (x^2+y^2)\psi \text{ is biharmonic}$$

Now we define the airy stress fn. ϕ define $P \equiv \nabla^2\phi = \sigma_{xx} + \sigma_{yy}$

$$\sigma_{xx} = \phi_{yy}, \quad \sigma_{yy} = \phi_{xx}, \quad \sigma_{xy} = -\phi_{xy}$$

to satisfy compat $\nabla^4\phi = 0 \Rightarrow \nabla^2P = 0 \quad P$ is harmonic

let Q be the conjugate harmonic fn of P Let $F(z) = P+iQ$

$$\text{consider } \frac{1}{4}\int F(z)dz \equiv \psi(z) = p+iq \quad \therefore \frac{d\psi}{dz} = \frac{1}{4}f(z) = \frac{\partial p}{\partial x} + i\frac{\partial q}{\partial x} = \frac{P+iQ}{4}$$

$$\Rightarrow \frac{\partial p}{\partial x} = \frac{1}{4}P, \quad \frac{\partial q}{\partial y} = \frac{Q}{4} \quad \text{from cauchy-reimann} \quad \Rightarrow \frac{\partial^2 p}{\partial x^2} = \frac{1}{4}P, \quad \frac{\partial^2 q}{\partial y^2} = \frac{Q}{4} = -$$

$\Rightarrow p_1 = \phi - xp - yq$ is biharmonic

$\therefore \phi = p_1 + xp + yq$ is biharmonic

using both p and q is not necessary $\Rightarrow \nabla^2(\phi - 2xp) = \nabla^2\phi - 4\frac{\partial p}{\partial x} = 0$
if $\nabla^2 p = 0$

$\therefore \phi - 2xp$ is biharmonic let $p_2 = \phi - 2xp \Rightarrow \phi = 2xp + p_2$ is biharmonic

$\phi = 2yq + p_3$ is also biharmonic

define $\chi(z) = p_1 + iq_1$, q_1 is conjugate harmonic to p_1

Consider $(x-iy)(p+iq) + p_1 + iq_1$

Real part $xp + yq + p_1$; thus any stress fn ϕ can be expressed as
 $\phi = \operatorname{Re}[\bar{z}\psi + \chi]$ ψ, χ are analytic fns. of a complex variable

$$\bar{z}\psi + \chi = r^2 \frac{\psi}{z} + \chi = r^2 p_4 + p_5$$

Stress in terms of stress function ϕ

$$\text{For plane stress } E \frac{\partial u_x}{\partial x} = \sigma_{xx} - \nu \sigma_{yy}$$

$$E \frac{\partial u_y}{\partial y} = \sigma_{yy} - \nu \sigma_{xx} \quad \tau_{zz} = 0$$

$$\mu \left(\frac{\partial u_y}{\partial x} + \frac{\partial u_x}{\partial y} \right) = \tau_{xy}$$

$$E \frac{\partial u_x}{\partial x} = \phi_{,yy} - \nu \phi_{,xx} = (P - \phi_{,xx}) - \nu \phi_{,yy} = P - (1+\nu) \phi_{,xx}$$

$$E \frac{\partial u_y}{\partial y} = -(1+\nu) \phi_{,yy} + P$$

$$\text{Let } P = \frac{4 \frac{\partial p}{\partial x}}{1+\nu} \quad \text{and} \quad \frac{4 \frac{\partial p}{\partial y}}{1+\nu}$$

$$\text{and integrate } 2\mu \frac{\partial u_x}{\partial x} = -\phi_{,xx} + \frac{4}{1+\nu} \frac{\partial p}{\partial x}$$

$$2\mu \frac{\partial u_y}{\partial y} = -\phi_{,yy} + \frac{4}{1+\nu} \frac{\partial p}{\partial y}$$

5/30/79

For any stress fn $\phi = \operatorname{Re} [\bar{z} \psi(z) + \chi(z)]$

$$2\mu \frac{\partial u}{\partial x} = -\phi_{,xx} + \frac{4}{1+\nu} \frac{\partial p}{\partial x} \rightarrow 2\mu u = -\frac{\partial \phi}{\partial x} + \frac{4}{1+\nu} p + f(y)$$

$$2\mu \frac{\partial v}{\partial y} = -\phi_{,yy} + \frac{4}{1+\nu} \frac{\partial p}{\partial y} \rightarrow 2\mu v = -\frac{\partial \phi}{\partial y} + \frac{4}{1+\nu} p + f(x)$$

$$\text{Now } \mu \left(\frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} \right) = \tau_{xy}$$

$$\Rightarrow -\phi_{,xy} + \frac{2}{1+\nu} \left(\frac{\partial p}{\partial x} + \frac{\partial p}{\partial y} \right) + \frac{1}{2} \frac{df}{dy} + \frac{1}{2} \frac{df_1}{dx} = \tau_{xy}$$

$$\Rightarrow \frac{df}{dy} + \frac{df_1}{dx} = 0 \quad \Rightarrow \frac{df}{dy} = \text{const} \quad (\text{rigid body rotation displacement})$$

$P \triangleq \nabla^2 \phi$ Q from C.R. relations can be obtained

Now $f(z) = \alpha + i\beta$ $\bar{f} = \alpha - i\beta$ then $f + \bar{f} = 2\alpha = 2\operatorname{Re} f$

$$2\phi = \bar{z} \psi(z) + \chi(z) + z \bar{\psi}(\bar{z}) + \bar{\chi}(\bar{z})$$

$$2 \frac{\partial \phi}{\partial x} = \bar{z} \psi(z) + \psi(z) + \chi'(z) + z \bar{\psi}'(\bar{z}) + \bar{\psi}(\bar{z}) + \bar{\chi}'(\bar{z}) \quad (4)$$

$$2 \frac{\partial \phi}{\partial y} = i \left[\bar{z} \psi'(z) - \psi(z) + \chi(z) - z \bar{\psi}'(\bar{z}) + \bar{\psi}(\bar{z}) - \bar{\chi}'(\bar{z}) \right] \quad (2)$$

$$\Rightarrow \text{take } \frac{1}{2} [2\phi + i2\frac{\partial \phi}{\partial y}] = \psi(z) + z \bar{\psi}'(\bar{z}) + \bar{\chi}'(\bar{z})$$

$$\text{Now take } \frac{\partial u}{\partial y} (u + iv) = - \left[\frac{\partial \phi}{\partial x} + i \frac{\partial \phi}{\partial y} \right] + \frac{4}{1+v} [p+iq]$$

$$\text{Now } \psi(z) = p+iq \Rightarrow \left| 2\mu(u+iv) = \frac{3-v}{1+v} \psi(z) - z\bar{\psi}'(z) - \bar{\chi}'(z) \right|$$

for plane stress

for plane strain replace ν by $\frac{v}{1-\nu}$

We now want stresses

$$\text{Now } \frac{\partial}{\partial x} \left[\frac{\partial \phi}{\partial x} + i \frac{\partial \phi}{\partial y} \right] = \psi(z) + z\bar{\psi}'(z) + \bar{\psi}'(z) + \bar{\chi}''(z) \quad (3)$$

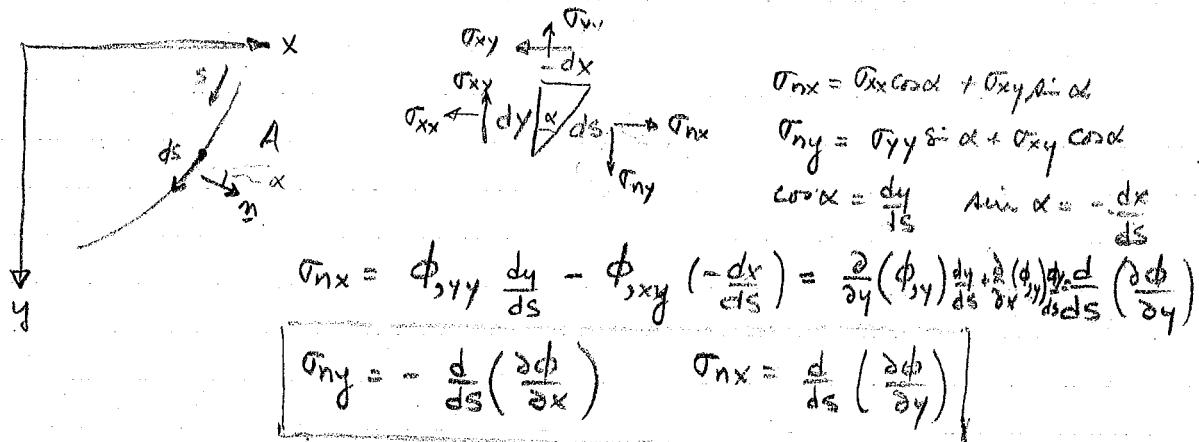
$$\text{and } i \frac{\partial}{\partial y} \left[\frac{\partial \phi}{\partial x} + i \frac{\partial \phi}{\partial y} \right] = -\psi(z) + z\bar{\psi}'(z) - \bar{\psi}'(z) + \bar{\chi}''(z) \quad (4)$$

$$\text{Now } \sigma_{xx} + \sigma_{yy} - \nabla^2 \phi = (3)-(4) = 2\bar{\psi}'(z) + 2\bar{\chi}''(z) = 4 \operatorname{Re} \psi'(z)$$

$$\text{Now } \sigma_{yy} - \sigma_{xx} - 2i\tau_{xy} = (3)+(4) = 2[z\bar{\psi}''(z) + \bar{\chi}''(z)]$$

this gives 3 eqns for $\sigma_{xx} + \sigma_{yy}$, $\sigma_{yy} - \sigma_{xx}$, $-2i\tau_{xy}$

B.C.



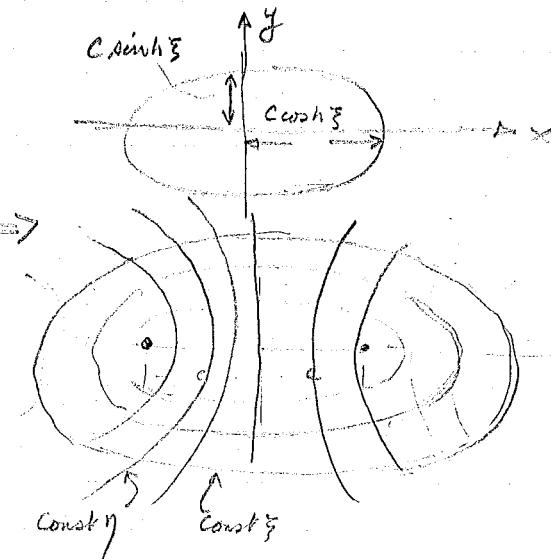
Elliptic Coordinates

$$x = C \cosh \xi \cos \eta$$

$$y = C \sinh \xi \sin \eta$$

$$\text{const } \xi \text{ curves } \frac{x^2}{C^2 \cosh^2 \xi} + \frac{y^2}{C^2 \sinh^2 \xi} = 1$$

$$\text{const } \eta \text{ curves } \frac{x^2}{C^2 \cos^2 \eta} - \frac{y^2}{C^2 \sin^2 \eta} = 1$$

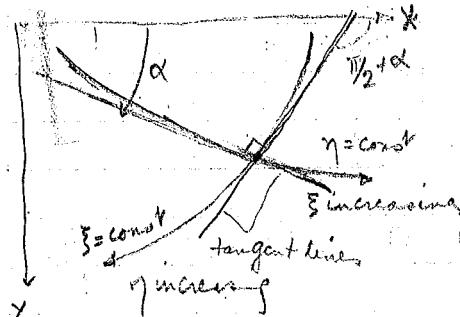


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$$\begin{aligned} x &= c \cosh \xi \cos \eta \\ y &= c \sinh \xi \sin \eta \end{aligned} \quad \left. \begin{array}{l} \text{elliptical coords} \\ \text{ } \end{array} \right\}$$

$$x+iy = c \cosh(\xi + i\eta) \Rightarrow \cosh(\alpha+i\beta) = \cosh \xi \cos \eta + i \sinh \xi \sin \eta$$

thus $\xi = c \cosh \beta$ where $\beta = \xi + i\eta$



$$\sigma_{\xi\xi} + \sigma_{\eta\eta} = \sigma_{xx} + \sigma_{yy}$$

$$\begin{aligned} \sigma_{\eta\eta} - \sigma_{\xi\xi} + 2i\sigma_{\xi\eta} &= -(\sigma_{xx} - \sigma_{yy}) \cos 2\alpha - 2\sigma_{xy} \sin 2\alpha \\ &\quad + 2i(\sigma_{xx} - \sigma_{yy}) \sin 2\alpha + 2i\sigma_{xy} \cos 2\alpha \\ &= -(\sigma_{xx} - \sigma_{yy}) e^{i2\alpha} + 2i\sigma_{xy} e^{i2\alpha} \\ &= e^{i2\alpha} [\sigma_{xx} - \sigma_{yy} + 2i\sigma_{xy}] \end{aligned}$$

$$\sigma_{\xi\xi} + \sigma_{\eta\eta} = 4 \operatorname{Re} \Psi(z) = 2 [\psi'(z) + \bar{\psi}'(\bar{z})] \quad \text{as shown in last lecture}$$

$$\sigma_{\eta\eta} - \sigma_{\xi\xi} + 2i\sigma_{\xi\eta} = 2e^{i2\alpha} [\bar{z}\psi''(z) + \chi''(z)] \quad \text{using results from previous lect.}$$

displ.

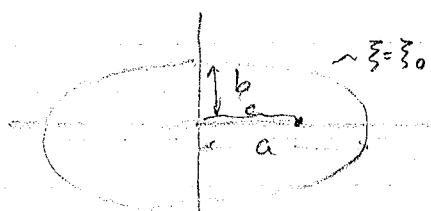
$$\begin{aligned} u_x &= u \cos \alpha + v \sin \alpha \\ u_y &= v \cos \alpha - u \sin \alpha \end{aligned} \quad \left. \begin{array}{l} \\ \end{array} \right\} \Rightarrow u_\xi + i u_\eta = e^{-i\alpha} [u + iv]$$

Note that $f'(5) = J \cos \alpha + i J \sin \alpha = Je^{i\alpha} \quad J, \alpha \text{ real}$
 $f'(\bar{5}) = Je^{-i\alpha} \quad \therefore f'(5)/f'(\bar{5}) = e^{2i\alpha}$

$$z = c \cosh \xi$$

$$\frac{dz}{d\xi} = c \sinh \xi$$

$$e^{2i\alpha} = \frac{\sinh \xi_0}{\sinh \bar{\xi}_0}$$



$$\begin{aligned} a &= c \cosh \xi_0 \\ b &= c \sinh \xi_0 \end{aligned}$$

look at problem #1 remote uniform tension S

$$\text{BC: at } \infty \quad \sigma_{xx} = \sigma_{yy} = S' \quad \text{as } \xi \rightarrow \infty \\ \Rightarrow 2\operatorname{Re} \psi(z) = S' \quad \text{also } \bar{z}\psi''(z) + \chi''(z)$$

$$\text{BC: at } \xi = \xi_0, \quad \sigma_{\xi\xi}, \sigma_{\xi\eta} = 0 \\ \text{must have periodicity in } \eta$$

To obtain χ, ψ we must use trial & error

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we note that $\sinh \eta \xi, \cosh \eta \xi, BC^2 \xi$ qualify for fns. ψ, χ

$$\text{let } \psi(z) = Ac \sinh \xi z \\ \psi'(z) = Ac \cosh \xi z \frac{dz}{dz} = Ac \cosh \xi z \left(\frac{dz}{d\xi}\right)^{-1}; \quad \text{since } z = \cosh \xi \quad \frac{dz}{d\xi} = c \sinh \xi \\ = A \coth \xi \quad \text{as } \xi \rightarrow \infty \quad \coth \xi \rightarrow 1 \\ \Rightarrow \text{if } 2A = S \text{ we satisfy 1st BC.} \quad \therefore 2\operatorname{Re} \psi'(z) = 2A = S \quad \text{as } \xi \rightarrow \infty$$

$$\psi''(z) = -\frac{A}{c} \frac{1}{\sinh^2 \xi z} \Rightarrow \bar{z}\psi''(z) = -\frac{A \cosh \xi z}{\sinh^3 \xi z}; \quad \text{let } \chi(z) = BC^2 \xi \Rightarrow \chi(z) = \frac{BC}{\sinh \xi z} \\ \Rightarrow \chi''(z) = -B \frac{\cosh \xi z}{\sinh^3 \xi z} \quad \text{then } \bar{z}\psi''(z) + \chi''(z) = 0 \Leftrightarrow \xi \rightarrow \infty \\ \text{since as } \xi \rightarrow \infty \quad \bar{z}\psi'' \rightarrow 0, \chi'' \rightarrow 0 \\ \text{thus we satisfy 2nd BC}$$

go back to the general expressions

$$(*) \quad \sigma_{\xi\xi} - i\sigma_{\xi\eta} = \psi'(z) + \psi'(\bar{z}) - e^{2iz} [\bar{z}\psi''(z) + \chi''(z)] \\ e^{2iz} = \sinh \xi z / \sinh \bar{\xi} z$$

Now substitute the obtained expressions and algebraate in (*)

$$(**) \quad \sigma_{\xi\xi} - i\sigma_{\xi\eta} = \frac{1}{\sinh^2 \xi z \sinh \bar{\xi} z} \left\{ A \left[\sinh \xi z \sinh (\xi + \bar{\xi}) + \cosh \bar{\xi} z \right] + B \cosh \xi z \right\}$$

we will now get A & B by using that $\sigma_{\xi\xi} = \sigma_{\xi\eta} = 0$ on $\xi = \xi_0$ $2\xi_0 = \xi + \bar{\xi} \Rightarrow \bar{\xi} = 2\xi_0 - \xi$
 plug into the (**) to get

$$\sigma_{\xi\xi} - i\sigma_{\xi\eta} = \frac{1}{\sinh^2 \xi z \sinh \bar{\xi} z} (A \cosh 2\xi_0 + B) \cosh \xi z$$

$$\text{but } A = S/2 \Rightarrow B = -\frac{S \cosh 2\xi_0}{2}$$

$$\Phi(z) = \frac{1}{2} Sc \sinh \xi$$

$$\chi(z) = -\frac{1}{2} Sc^2 (\cosh 2\xi_0) \xi$$

On the rim $\sigma_{33} = \sigma_{3\eta} = 0$

to find $\sigma_{\eta\eta}$ use first invariant: since $\sigma_{33} + \sigma_{\eta\eta} = \sigma_{\eta\eta}$ on rim
but $\sigma_{33} + \sigma_{\eta\eta} = 4 \operatorname{Re} \Phi(z) = 2Sc \operatorname{Re}(\coth \xi)$

$$\coth \xi \Rightarrow \frac{\sinh 2\xi - i \sin 2\eta}{\cosh 2\xi - \cos 2\eta} \Rightarrow \operatorname{Re} \coth \xi = \frac{\sinh 2\xi}{\cosh 2\xi - \cos 2\eta}$$

$$\Rightarrow \sigma_{\eta\eta} = \frac{2Sc \sinh 2\xi}{\cosh 2\xi - \cos 2\eta} \quad \text{when } \xi = \xi_0 \text{ then } \sigma_{\eta\eta} = \frac{2Sc \sinh 2\xi_0}{\cosh 2\xi_0 - \cos 2\eta}$$

$$(*) \quad \max_{\eta=0,\pi} \sigma_{\eta\eta} = \frac{2Sc \sinh 2\xi_0}{\cosh 2\xi_0 - 1} \quad \text{since: } c^2 = a^2 - b^2$$

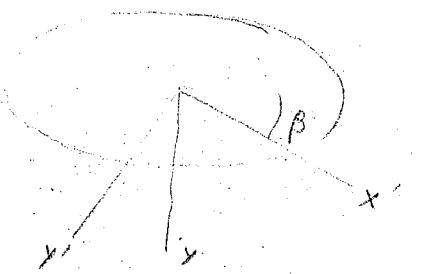
$$\text{and } \sinh 2\xi_0 = 2 \tanh \xi_0 \cosh \xi_0$$

$$\tanh \xi_0 = \frac{b}{c} \quad \sinh 2\xi_0 = \frac{2ab}{c^2}$$

$$\cosh \xi_0 = \frac{c}{b} \quad \cosh 2\xi_0 = \frac{c^2}{b^2} - 1$$

$$\text{plugging into (*) to get } \sigma_{\eta\eta} \Big|_{\max} = \frac{2Sc}{b}$$

Problem #2 - Elliptic cavity under uniaxial tension



$$\text{let } \Phi(z) = A \cosh \xi + B \cosh \xi$$

$$\begin{aligned} \Phi'(z) &= C \cosh \xi + D \cosh \xi \\ &\quad + E \sinh 2\xi \end{aligned}$$

$$\sigma_{xx'} + \sigma_{yy'} = \sigma_{xx} + \sigma_{yy}$$

$$\sigma_{yy'} - \sigma_{xx'} + 2i \sigma_{xy'} = e^{2i\beta} (\sigma_{xx} + \sigma_{yy} + 2i \sigma_{xy})$$

$$\Rightarrow 4 \operatorname{Re} \Phi'(z) = S \quad [2\Phi''(z) + \chi''(z)] = -Se^{-2i\beta} \quad \text{at } \infty$$

$$\text{at } \xi = \xi_0 \quad \sigma_{xx} = \sigma_{yy} = 0$$

again we use periodicity to obtain $A \rightarrow 0$
also by periodicity that $C + D$ are real

after algebraic

$$A = S e^{2\beta_0} \cos 2\beta \quad B = S(1 - e^{2\beta_0 + 2i\beta}) \quad C = -S(\cosh 2\beta_0 - \cos 2\beta)$$

$$D = -\frac{1}{2}S e^{2\beta_0} \cosh 2(\beta_0 + i\beta) \quad E = \frac{1}{2}S e^{2\beta_0} \sinh 2(\beta_0 + i\beta)$$

$$\text{we now use the first stress invariant to get } \sigma_{\eta\eta}|_{\beta=\beta_0} = S \frac{\sinh 2\beta_0 + \cos 2\beta - e^{2\beta_0} \cos 2\beta}{\cosh 2\beta_0 - \cos 2\beta}$$

$$\begin{aligned} & \text{if } \beta = \frac{\pi}{2} \\ & \sigma_{\eta\eta}|_{\beta=\beta_0} = S e^{2\beta_0} \left[\frac{\sinh 2\beta_0 (1 + e^{-2\beta_0})}{\cosh 2\beta_0 - \cos 2\beta} - 1 \right] \\ & \max \sigma_{\eta\eta}|_{\beta=\beta_0} = S (1 + 2\%) \end{aligned}$$

Continuation

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$$\begin{array}{c} \text{if } \beta=0 \\ \text{if } \beta=\frac{\pi}{4} \\ \text{if } \beta=\frac{3\pi}{4} \end{array}$$

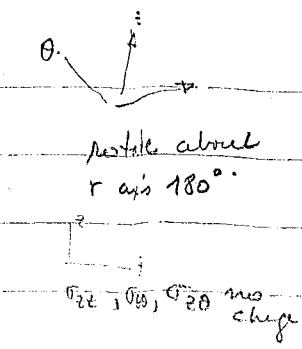
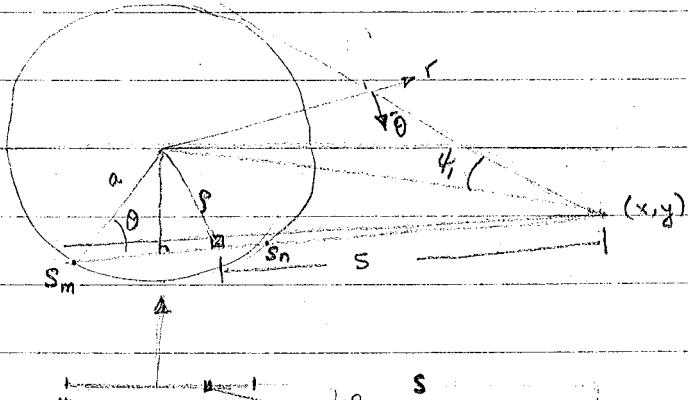
$$S = \left(1 + \frac{2b}{a}\right)$$

$$\text{for pure shear } S @ \beta = \frac{\pi}{4}$$

$$S @ \beta = \frac{3\pi}{4} \quad \sigma_{\eta\eta}|_{\beta=\beta_0} = 2S' e^{2\beta_0} \frac{\sin 2\beta}{\cosh 2\beta - \cos 2\beta}$$

$$\max \sigma_{\eta\eta} = \pm S \frac{(a+b)^2}{ab}$$

G.P. Cherepanov - Inverse Problems of the Plane Theory of Elasticity
 PMM - Appl. Math & Mech V38 No 6 1974 pg 915-931



$$s = r_0 \cos \psi.$$

$$d\xi d\eta = (sd\psi) ds.$$

$$\int \frac{pd\xi d\eta}{r_0} = p \iint \frac{sds d\psi}{\sqrt{s^2 + z^2}}$$

$$r = \sqrt{x^2 + y^2}.$$

$$2\pi \int_0^{2\pi} \left[\sqrt{s^2 + z^2} \right]_{s=s_m}^{s=s_n} d\psi.$$

$$s_m = r \cos \psi + \sqrt{a^2 - r^2 \sin^2 \psi} \quad \sqrt{s_m^2 + z^2}$$

$$s_n = r \cos \psi - \sqrt{a^2 - r^2 \sin^2 \psi}$$

Now if we remember that $r \sin \psi = a \sin \theta$

$$\text{then } d\psi = \frac{a \cos \theta d\theta}{r \cos \psi} = \frac{a \cos \theta d\theta}{r \sqrt{1 - (a/r)^2 \sin^2 \theta}} = \frac{a \cos \theta d\theta}{r \sqrt{1 - (q/r)^2 \sin^2 \theta}}$$

$$\text{or } 2\pi \int_0^{2\pi} \left(\sqrt{s_m^2 + z^2} - \sqrt{s_n^2 + z^2} \right) \cdot \frac{a \cos \theta d\theta}{r \sqrt{1 - (q/r)^2 \sin^2 \theta}}$$

$$\text{and } s_m = r \sqrt{1 - (q/r)^2 \sin^2 \theta} + a \cos \theta$$

$$s_n = r \sqrt{1 - (q/r)^2 \sin^2 \theta} - a \cos \theta$$

by symmetry of situation look along $r=x$ then.

$$\iint P \delta(x-\xi) \delta(y-\eta) d\xi d\eta.$$

$$\sigma_z = -\frac{3q}{2\pi} \int \frac{z^3 \cdot sd\psi ds}{r_0^5} = -\frac{3qz^3}{2\pi} \int \frac{sds}{(\sqrt{s^2 + z^2})^5} d\psi$$

$$= -\frac{3qz^3}{2\pi} \int \frac{d\psi}{(s^2 + z^2)^{3/2}} \Big|_{s=s_m}^{s=s_n} = \frac{qz^3}{2\pi} \int d\psi \left[\frac{1}{(s_m^2 + z^2)^{3/2}} - \frac{1}{(s_n^2 + z^2)^{3/2}} \right]$$

$$\tau_{rz} = -\frac{3qz^2}{2\pi} \iint \frac{s \cdot sd\psi ds}{(s^2 + z^2)^{5/2}}$$

let $s=u$ $\frac{sds}{(\sqrt{s^2 + z^2})^{3/2}} = dv$
 $du=ds \quad \frac{1}{3} (\sqrt{s^2 + z^2})^{-3/2} = v$

$$\downarrow A \quad \int d\psi \left[-\frac{s}{3} (s^2 + z^2)^{-3/2} + \frac{1}{3} \int \frac{ds}{(s^2 + z^2)^{3/2}} \right] = A \int d\psi \left[-\frac{s}{3} (z^2 + s^2)^{-3/2} + \frac{1}{3} z^2 \frac{s}{(z^2 + s^2)^{3/2}} \right]_{s_m}^{s_n}$$

$$\tau_{rz} = -\frac{q}{2\pi} \int d\psi \left[-\frac{s^3}{(s^2 + z^2)^{3/2}} \right]_{s_m}^{s_n}$$

$$\sigma_r = \frac{q(1-\nu)}{2\pi} \int s d\psi ds \left\{ \frac{1}{s^2} - \frac{z}{s^2} \frac{1}{(s^2+z^2)^{1/2}} - \frac{3s^2}{(s^2+z^2)^{3/2}} \right\}$$

$$\int d\psi \left\{ \ln s + \frac{1}{2} \ln \left(\frac{z+(s^2+z^2)^{1/2}}{(s^2+z^2)^{1/2}-z} \right) - 3z \left[-\frac{1}{(s^2+z^2)^{1/2}} + \frac{z^2}{3(s^2+z^2)^{3/2}} \right] \right\}_{S_n}^{S_m}$$

$$\sigma_\theta = \frac{q(1-\nu)}{2\pi} \int s d\psi ds \left\{ -\frac{1}{s^2} + \frac{z}{s^2} \frac{1}{(s^2+z^2)^{1/2}} + \frac{z}{(s^2+z^2)^{3/2}} \right\}$$

$$\int d\psi \left\{ -\ln s - \frac{1}{2} \ln \left(\frac{(s^2+z^2)^{1/2}+z}{(s^2+z^2)^{1/2}-z} \right) - \frac{z}{(s^2+z^2)^{1/2}} \right\}_{S_n}^{S_m}$$

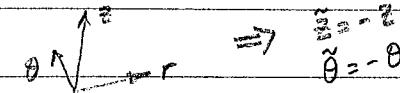
Now.

$$\sigma_r = \frac{\partial}{\partial z} (\nu \nabla^2 \phi - \phi_{rr}) \quad \nabla^2 = \frac{\partial^2}{\partial r^2} + \frac{1}{r} \frac{\partial}{\partial r} + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2} + \frac{\partial^2}{\partial z^2}$$

$$\sigma_\theta = \frac{\partial}{\partial z} (\nu \nabla^2 \phi - \frac{1}{r} \phi_{r\theta}) \quad \text{Now if we want}$$

$$\sigma_z = \frac{\partial}{\partial z} ((2-\nu) \nabla^2 \phi - \phi_{zz})$$

$$\tau_{rz} = \frac{\partial}{\partial r} ((1-\nu) \nabla^2 \phi - \phi_{rz})$$



$$\nabla^2(r, \theta, z) = \nabla^2(r, \tilde{\theta}, \tilde{z})$$

$$\tilde{\sigma}_r = -\sigma_r$$

$$\tilde{\sigma}_\theta = -\sigma_\theta$$

$$\tilde{\sigma}_z = -\sigma_z$$

$$\tilde{\tau}_{rz} = \tau_{rz}$$

$$\therefore \tilde{\sigma}_z = + \frac{qz^3}{2\pi} \int d\psi \left[\frac{1}{(s^2+z^2)^{3/2}} \right]_{S_n}^{S_m}$$

$$\tilde{\tau}_{rz} = - \frac{q}{2\pi} \int d\psi \left[\frac{s^3}{(s^2+z^2)^{3/2}} \right]_{S_n}^{S_m}$$

$$\tilde{\sigma}_r = - \frac{q}{2\pi} (1-\nu) \int d\psi \left\{ \ln s - \frac{1}{2} \ln \left[\frac{z+(s^2+z^2)^{1/2}}{(s^2+z^2)^{1/2}-z} \right] + 3z \left[-\frac{1}{(s^2+z^2)^{1/2}} + \frac{z^2}{3(s^2+z^2)^{3/2}} \right] \right\}_{S_n}^{S_m}$$

$$\sigma_\theta = - \frac{q}{2\pi} (1-\nu) \int d\psi \left\{ -\ln s + \frac{1}{2} \ln \left[\frac{(s^2+z^2)^{1/2}+z}{(s^2+z^2)^{1/2}-z} \right] \right\}_{S_n}^{S_m}$$

$$\lambda = \frac{2\mu\nu}{(1-2\nu)}$$

$$\sigma_{ij} = \lambda \delta_{ij} \epsilon_{kk} + 2\mu \epsilon_{ij}; \quad \epsilon_{ij} = \frac{(1+\nu)}{E} \sigma_{ij} - \frac{\nu}{E} \sigma_{kk} \delta_{ij}$$

IF g and $\frac{\partial g}{\partial n} = 0$ on S then $g(P) = -\frac{1}{4\pi} \int \frac{1}{R} \nabla^2 g dA$

$$\text{IF NOT } g(P) = \frac{1}{4\pi} \int \left(\frac{1}{R} \frac{\partial g}{\partial n} - g \frac{\partial^2}{\partial n^2} \right) dS - \int \frac{1}{R} \nabla^2 g dA$$

$$\text{IF } G = \nabla \phi + \nabla \times H \text{ w/ } \nabla \cdot H = 0 \text{ then } \phi = -\frac{1}{4\pi} \int \frac{\nabla \cdot G}{R} dA \quad H = -\frac{1}{4\pi} \int \frac{\nabla \times G}{R} dA$$

$$\text{IF } H = \nabla \phi + \nabla \times H \text{ then } G = H$$

$$\text{Boussinesq-Papkovich-Neuber fn: } u_i = B - \frac{1}{4(1-\nu)} \nabla (r \cdot B + \beta) \quad \text{with } \nabla^2 B = -f/\mu \quad \nabla^2 \beta = \frac{1}{\mu}$$

A. ∞ Body Problems $G_{km}(r) = \frac{1}{8\pi\mu} \left[\delta_{km} r_{pp} - \frac{\lambda + \mu}{\lambda + 2\mu} r_{km} \right]$

1. Kelvin problem - stress distib in an ∞ elastic body subject to concentrated force

$$B = \frac{P}{4\pi\mu r} \quad \beta = 0 \quad u_i = \text{found in HW #1} \quad \text{or also}$$

2. Double force (in tension) in z dir $A = \lim_{h \rightarrow 0} Ph$

$$B = \frac{Az e_z}{4\pi\mu r^3} \quad \beta = -\frac{A}{4\pi\mu r}$$

3. Center of Dilatation $\nabla \cdot u_i = 0$

$$B = \frac{A P}{4\pi\mu r^3} \quad \beta = -\frac{3A}{4\pi\mu r} \quad \text{or} \quad u_i = -\frac{1}{4(1-\nu)} \nabla \left(\frac{C}{r} \right) \quad C = \frac{(1-2\nu)A}{2\pi\mu}$$

$$\text{if } u_i = \frac{CR}{4(1-\nu)r^3}, \quad u_\theta = u_\phi = 0$$

4. Spherical Cavity under pressure p_i (assume center of dilatation)

$$\sigma_{RR} = -p_i (a/R)^3; \quad \sigma_{\theta\theta} = \sigma_{\phi\phi} = p_i (a/R)^3; \quad \sigma_{\theta\phi} = \sigma_{\theta R} = \sigma_{\phi R} = 0 \quad u_R = \frac{p_i a}{4\mu} (a/R)^2 \quad u_\theta = u_\phi = 0$$

5. Hollow sphere under internal & external pressure (assume center of dilatation + hydrostatic tension)

$$\sigma_{RR} = (p_i - p_e) \frac{a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 p_e - a^3 p_i}{a^3 - b^3} \quad \sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{a^3 b^3 (p_e - p_i)}{2(a^3 - b^3) R^3} + \frac{b^3 p_e - a^3 p_i}{a^3 - b^3}$$

More on pg 7:

Rigid inclusion in a sphere under external pressure - see midterm

6. Spherical cavity under uniaxial tension at ∞

see pg 8 and HW #2 and 3

$$4\pi h(x, y, z) = -2 \frac{\partial}{\partial z} \int_{r_0}^{\infty} \frac{h(\xi, \eta, 0)}{r_0} d\xi dy - \int_V \left(\frac{1}{r_1} - \frac{1}{r_2} \right) \nabla^2 h(\xi, \eta, z) d\xi dy dz$$

$$G = \frac{1}{r_1} - \frac{1}{r_2}, \quad \nabla^2 G = 0 \text{ in } V, \text{ Green's}$$

B. Half Space Problem

1. Distributed load in half space



$$\text{w/ } \sigma_{zx} = \sigma_{zy} = 0, \quad \sigma_{zz} = -p(z, \eta) \quad r_0 = \sqrt{(x-z)^2 + (y-\eta)^2 + z^2} \\ B_z = B_z B_z = \frac{1-v}{\pi \mu} \int \frac{p(\xi, \eta)}{r_0} d\xi dy \quad \beta = \int \frac{(1-2v)(1-v)}{\pi \mu} dz \int \frac{p(\xi, \eta)}{r_0} d\xi dy$$

2. Boussinesq - point force at origin (in z dir)

$$B_z = \frac{1-v}{\pi \mu} \frac{P_z}{R} \quad \beta = \frac{(1-2v)(1-v)}{\pi \mu} P_z \ln(R+z) \quad \text{Settlement - see midterm}$$

3. Distributed Normal loads

$$\text{Settlement } w = u_z(x, y, 0) = \frac{B_z}{2} \quad \text{where } B_z$$

load over
circle

$$a. \quad p(\xi, \eta) = p_0 \text{ over a circle of rad } a. \quad \text{VALID UNDER LOAD ONLY !!!}$$

$$w(x, y) = \frac{2(1-v)}{\pi \mu} p_0 a \int_0^{R/2} \sqrt{1 - (\frac{r}{a})^2} \sin^2 \psi d\psi \quad \text{a complete elliptic integral of the 2nd kind.}$$

$$b. \quad p = \frac{N}{2\pi a \sqrt{a^2 - r^2}} \quad \text{where } N = \int p(\xi, \eta) d\xi dy \quad w(x, y) = \frac{(1-v)N}{4\pi \mu a} = \text{const.}$$

for u_z outside load see HW #2.

$$c. \quad p = \frac{BN}{2\pi a^2} \sqrt{a^2 - r^2} \quad \text{where } N = \int p(\xi, \eta) d\xi dy \quad w(x, y) = \frac{3(1-v)N}{16\pi \mu a^3} (2a^2 - r^2)$$

4. tangential fractions over half space

$$B_y = 0, \quad B_x = \frac{1}{2\pi \mu} \int \frac{p_x(\xi, \eta)}{r_0} d\xi dy$$

$$2. \quad \sigma_{zz} = \sigma_{zy} = 0, \quad \sigma_{zx} = -p_x \quad \text{or. } \Phi \cdot \Psi$$

$$B_z = \int dz (1-2v) \frac{\partial B_x}{\partial x}, \quad r_0^2 = (x-\xi)^2 + (y-\eta)^2 + z^2$$

$$\beta = \int dz \left[(1-2v) B_z - x \frac{\partial B_x}{\partial z} + z \frac{\partial B_x}{\partial x} \right]$$

5. Cerruti Problem : concentrated force in tangential direction on half space (using 4)

$$B_x = \frac{P_x}{2\pi \mu R}, \quad B_z = \frac{(1-v) P_x x}{2\pi \mu R (R+z)}, \quad \beta = -\frac{(1-2v)^2 P_x x}{2\pi \mu (R+z)}$$

$u_x(x, y, 0)$ and $u_y(x, y, 0)$ are given on Pg 16

2. Mindlin problem : a concentrated force at any point in the interior of a half space

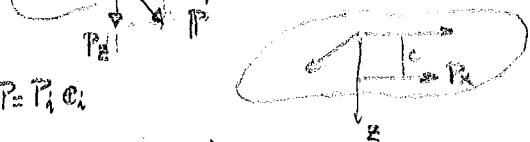
Special case



$$B_x, B_y = 0 \quad B_z = \frac{P_2}{4\pi\mu} \left[\frac{1}{R_1} + \frac{3-4\nu}{R_2} - 2C \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) \right]$$

$$\beta = \frac{P_2}{4\pi\mu} \left[-\frac{C}{R_1} - \frac{C(3-4\nu)}{R_2} + 4(1-2\nu)(1-\nu) \log(R_2+z+C) \right]$$

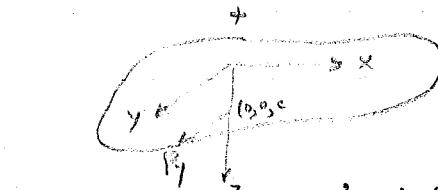
$$P = P_1 e_i$$



$$B_y = 0 \quad B_x = \frac{P_x}{4\pi\mu} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) \quad B_z = \frac{P_x}{2\pi\mu R_2} \left[\frac{1-2\nu}{R_2+z+C} - \frac{C}{R_2^2} \right]$$

$$\beta = \frac{P_x(1-2\nu)x}{2\pi\mu(R_2+z+C)} \left[\frac{C}{R_2} - (1-2\nu) \right]$$

$$(0, 0, C)$$

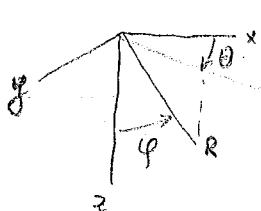


$$B_x = 0 \quad B_y = \frac{P_y}{4\pi\mu} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) \quad B_z = \frac{P_y}{2\pi\mu R_2} \left[\frac{1-2\nu}{R_2+z+C} - \frac{C}{R_2^2} \right]$$

$$\beta = \frac{P_y y (1-2\nu)}{2\pi\mu(R_2+z+C)} \left[\frac{C}{R_2} - (1-2\nu) \right]$$

$$R_2^2 = x^2 + y^2 + (z+C)^2 \quad R_1^2 = x^2 + y^2 + (z-C)^2$$

Mindlin problem : if $C \rightarrow \infty$ we approach the Kelvin problem ; as $C \rightarrow 0$ we should approach Cerruti/boussinesq



$$\frac{x}{r} = \sin\varphi \cos\theta \quad \frac{y}{r} = \sin\varphi \sin\theta \quad \frac{z}{r} = \cos\varphi$$

$$\begin{pmatrix} \Phi_r \\ \Phi_\theta \\ \Phi_\phi \end{pmatrix} = \begin{pmatrix} \cos\varphi & \sin\varphi \cos\theta & \cos\varphi \\ \sin\varphi & \sin\theta \cos\varphi & -\sin\theta \\ \cos\theta & \sin\theta \cos\varphi & -\sin\theta \\ -\sin\theta & \cos\theta & 0 \end{pmatrix} \begin{pmatrix} \Phi_1 \\ \Phi_2 \\ \Phi_3 \end{pmatrix}$$

$$\epsilon_{rr} = \frac{\partial u_r}{\partial r} ; \quad \epsilon_{\theta\theta} = \frac{1}{r \sin\varphi} \frac{\partial u_\theta}{\partial \theta} + \frac{u_r}{r} + u_\theta \frac{\cot\varphi}{r} ; \quad \epsilon_{\varphi\varphi} = \frac{1}{r} \frac{\partial u_\varphi}{\partial \varphi} + \frac{u_r}{r} ; \quad \epsilon_{r\theta} = \frac{1}{2} \left(\frac{1}{r \sin\varphi} \frac{\partial u_r}{\partial \theta} - \frac{u_\theta}{r} + \frac{\partial u_\varphi}{\partial r} \right)$$

$$\epsilon_{\theta\varphi} = \frac{1}{2} \left(\frac{1}{r} \frac{\partial u_\theta}{\partial \varphi} - \frac{u_\varphi}{r} \cos\varphi + \frac{1}{r \cos\varphi} \frac{\partial u_r}{\partial \theta} \right) ; \quad \epsilon_{\varphi r} = \frac{1}{2} \left(\frac{1}{r} \frac{\partial u_\varphi}{\partial r} - \frac{u_r}{r} + \frac{\partial u_\theta}{\partial \varphi} \right)$$

$$\nabla \Psi = \left[\Phi_r \frac{\partial}{\partial r} + \Phi_\theta \frac{1}{r \sin\varphi} \frac{\partial}{\partial \theta} + \Phi_\varphi \frac{1}{r} \frac{\partial}{\partial \varphi} \right] \Psi$$

$$\text{in cartesian } T_{EK} = \frac{\mu\nu}{(1-\nu)} B_{m,m} \delta_{EK} + \frac{\mu}{2(1-\nu)} \left\{ (1-2\nu) [B_{L,K,L} + B_{K,L}] - [x_i B_{i,EK} + \beta_{,EK}] \right\}$$

Contact Problems

See Timoshenko pg 409-421

- Preliminaries: for $p_x = \frac{T}{2\pi a} \frac{1}{\sqrt{a^2 - r^2}}$ on a circle (distributed tangential load on a circle on a half-space)

 $u_x = \text{const} = \frac{(2-\nu)T}{8\mu a}$ rigid body disp

 $u_y = 0$

- Assume
 1. Contact area flat & circular
 2. contact area radius \ll radius of either bodies
 3. N applied remotely & doesn't affect problem (St Venant Principle)
 4. For local problem Most of bodies are half-spaces & no shear applied
 5. We can use settlement formula



$$\alpha = (z_1 + z_2) + w_1 + w_2 ; \text{ if } k_i = \frac{(1-\nu_i)}{2\pi\mu_i} \quad w_i = \frac{3k_i N \pi}{8a^3} (2a^2 - r^2) ; \quad z_i = \frac{r_i^2}{2R_i}$$

$$a = \sqrt[3]{\frac{(k_1+k_2)3N\pi R_1 R_2}{4(R_1+R_2)}} = K_1 N^{1/3}; \quad \alpha = \sqrt[3]{\frac{9N^3\pi^2 (k_1+k_2)^2 (R_1+R_2)}{16R_1 R_2}} = K_2 N^{2/3}; \quad C_n = \frac{dx}{dN} = \sqrt[3]{\frac{\pi^2 (k_1+k_2)^2 (R_1+R_2)}{6R_1 R_2 N}} = \frac{2K_2}{3N^{1/3}}$$

contact radius approach compliance in Normal load

- Contact due to applic of tangential after normal loading using static friction law

Tangential
Problem only

1. Assume for $r < a$ for tangential problem only $\sigma_{22}|_{z=0} = 0$

$$2. u_y = 0 \quad u_x = \text{constant} \Rightarrow p_x' = \frac{T}{2\pi a} \frac{1}{\sqrt{a^2 - r^2}} \quad \text{and } p_x \leq f p_z$$

3. no slip radius is c ; slip occurs for $a < r < c$

$$p_x = \frac{3fN}{2\pi a^3} \left[(a^2 - r^2)^{1/2} - (c^2 - r^2)^{1/2} \right] \quad u_y = 0 \quad u_x = \frac{3fN}{16a\mu} (2-\nu) \left(1 - \frac{c^2}{a^2} \right) = \frac{3fN}{16a\mu} (2-\nu) \left[1 - \left(1 - \frac{T}{fN} \right)^{4/3} \right]$$

$$4. c = a^3 \sqrt[3]{1 - \frac{T}{fN}} \quad \text{or } f = \frac{T/N}{(1 - [c/a]^3)} \quad \text{see back of page 19}$$

no slip radius

$$C_s = \frac{du_x}{dT} = \frac{2-\nu}{8\mu a} \left[1 - \frac{T}{fN} \right]^{-1/3} \leftarrow \text{compliance in shear}$$

- If we increase T to T^* and then decrease to T'

$$\text{then a radius } b = a \left(1 - \frac{T^* - T'}{2fN} \right)^{1/3} \quad \text{and a } u_x = \frac{3(2-\nu)fN}{16\mu a} \left[2 \left(1 - \frac{T^* - T'}{2fN} \right)^{2/3} - \left(1 - \frac{T^*}{fN} \right)^{2/3} - 1 \right]$$

$$\text{Energy dissip} = E_d = \frac{(2-\nu)T^*{}^3}{36\mu a f N}$$

displacement of arched section of contact circle
in horiz direction
 $u_x = \frac{3(2-\nu)fN}{16\mu a} \left[2 \left(\frac{T^*}{fN} \right)^2 - \left(\frac{b}{a} \right)^2 - 1 \right] / 16\mu a$

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CHAPTER III

ELASTIC INCLUSIONS AND INHOMOGENEITIES

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§ 1. Introduction

This review is concerned with the two following problems in the infinitesimal theory of elasticity, and with their inter-relation and generalization.

(i) The transformation problem.

A region (the 'inclusion') in a homogeneous elastic medium undergoes a permanent change of form which, in the absence of the constraint imposed by its surroundings (the 'matrix'), would be a prescribed uniform strain. To find the elastic field in matrix and inclusion.

(ii) The inhomogeneity problem.

A region in an otherwise homogeneous elastic medium has elastic constants differing from those of the remainder. To find how an applied stress, uniform at large distances, is disturbed by the inhomogeneity.

We shall not consider two-dimensional problems, where complex variable methods can be used, and the number of special cases which may be solved is unlimited. The three-dimensional inhomogeneity problem has been discussed extensively, particularly the special case of a cavity, i.e. an 'inhomogeneity' with vanishing elastic constants. The inclusion problem has received less attention, but is encountered in the discussion of various phenomena in solid-state physics, for example martensitic transformations and the formation of precipitates. STERNBERG [1958] has given an excellent annotated bibliography of the three-dimensional inhomogeneity (and inclusion) problems which have been solved.

We shall be largely concerned with the special case where the inclusion or inhomogeneity takes the form of the general ellipsoid with three unequal axes. There are two reasons for this. First, it appears to be the most general case whose solution can be given in a manageable form. Secondly, in this particular case there is a close connexion between the transformation and inhomogeneity problems. It stems from the fact that, as we shall see, the stress is constant throughout

an ellipsoidal inclusion which has undergone a uniform transformation. As an illustration of this connexion suppose that it is required to solve the inhomogeneity problem for an ellipsoidal cavity. On the stress-field due to the inclusion superimpose everywhere a uniform stress equal and opposite to the uniform stress in the inclusion. The inclusion is then free of stress and may be removed. We are left with a stress-field which becomes uniform at infinity and which gives zero traction on the surface of the ellipsoid, as required. The general problem of an ellipsoidal inhomogeneity can be handled by an extension of this argument.

Among closed surfaces the ellipsoid alone has this convenient property. It shares it with other second-degree surfaces, and the analysis of §§ 3, 4 can in fact be applied with trivial modifications to hyperboloids and paraboloids. However, the properties of such infinite 'inclusions' are not of much interest, and we shall not consider them. The corresponding in homogeneity problem is essentially the problem of stress-concentration by hyperboloidal notches. The cases which are of practical use have been discussed by NEUBER [1958].

We begin (§ 2) by finding a solution to problem (i) for an inclusion of arbitrary shape. The argument used is somewhat intuitive, but it is verified that the solution which is obtained does in fact satisfy the conditions of the problem. In § 3 the special case of an ellipsoidal inclusion is worked out to a point where numerical calculation is possible. Section 4 begins with a discussion of the inhomogeneity problem for an inhomogeneous region of arbitrary shape (though not much progress can be made) and the solution for the ellipsoidal inclusion is obtained from the results of § 2 in the way already indicated. By taking advantage of the connexion between inclusions and Somigliana dislocations (§ 5) it is possible to solve the problem of an ellipsoidal inhomogeneity perturbing a non-uniform stress-field. In § 6 we present some selected physical applications of the theory.

We shall use the familiar suffix notation. Repeated suffixes are to be summed over the values 1, 2, 3 and suffixes following a comma will denote differentiation with respect to the Cartesian coordinates x_1, x_2, x_3 , e.g. $u_{ij,l} = \partial u_{ij} / \partial x_l$, $\psi_{ijk} = \delta^3\psi / \delta x_i \delta x_j \delta x_k$. Displacement u_{ij} and strain e_{ij} are related by

$$e_{ij} = \frac{1}{2}(u_{i,j} + u_{j,i}) . \quad (1.1)$$

Stress ρ_{ij} and strain are related by

$$\rho_{ij} = \lambda e_{kk} \delta_{ij} + 2\mu e_{ij} \quad (1.2)$$

in an isotropic medium with Lamé constants λ, μ . If a second pair of quantities ρ'_{ij}, e'_{ij} satisfy

$$\rho'_{ij} = \lambda e'_{kk} \delta_{ij} + 2\mu e'_{ij} \quad (1.3)$$

then

$\rho_{ij} e'_{ij} = \rho'_{ij} e_{ij} .$

A set of quantities bearing a common affix, e.g. u_k^c, e_k^c, ρ_k^c will be supposed to be related by (1.1) and (1.2) unless otherwise stated. It will sometimes be convenient to use the notation

$$f = f_{kk}, \quad f'_{ij} = f_{ij} - \frac{1}{3} f_{kk} \delta_{ij} \quad (1.4)$$

to denote the scalar and deviatoric parts of a second-order tensor. We shall often make use of the formula

$$\nabla^2(\rho\eta) = \rho \nabla^2\eta + q \nabla^2\rho + 2\rho_{,k} q_{,k} \quad (1.5)$$

for calculating the Laplacian of a product, and of Gauss's theorem in the form

$$\int_S A \dots dS_k = \int_V A \dots, k dV \quad (1.6)$$

where S is a closed surface enclosing the volume V . Here and elsewhere dS_k is an abbreviation for $n_k dS$, where dS is an element of surface and n_k is its normal.

§ 2. The General Transformed Inclusion

2.1. THE ELASTIC FIELD

In this section we give a formal solution of the following problem: A region bounded by a closed surface S in a homogeneous isotropic elastic medium undergoes a change of form which but for the constraint imposed by the surrounding material would be an arbitrary homogeneous strain e_{ij}^T . To find the resulting elastic field inside and outside S .

It will be convenient to refer to the material inside S as the 'inclusion' and to the material outside S as the 'matrix'. The strain will be called the 'transformation strain', or, following ROBINSON [1951], the 'stress-free strain'. We assume that adjacent points immediately inside and outside S suffer no relative displacement; in other words the inclusion is 'bonded' to the matrix before, during and after the transformation. The problem can also be formulated as follows. To find a state of self-stress in an infinite body, with the following property: on making a

cut over a prescribed closed surface S we are left with a stress-free cavity and a stress-free solid bounded by surfaces S_1 , S_2 such that S_2 is transformed into S_1 by the homogeneous strain $-e_y^T$.

TMOSHENKO and GOODIER [1951] have given a method for finding the elastic field in a material in which each volume element alters its unconstrained shape. Our basic inclusion problem is merely the special case in which the change of shape is identical for all the volume elements inside a certain surface S and is zero for all elements outside S . The method of calculation we shall use (ESHELBY [1957]) is essentially equivalent to theirs.

It is first necessary to decide how to define the displacement. Let us fix our attention on some marked point r in the material, with coordinates $x_i(r)$ and suppose that, as we watch, the transformation takes place gradually, by some physical mechanism unspecified. Every point of the medium moves, and when the transformation is over the marked point will have different coordinates, $x_i(r) + u_i^c(r)$ say. We take u_i^c as our displacement function. The state of zero displacement is the state of the material before the transformation has occurred.

We may contrast this definition of the displacement with another, perhaps equally natural, one. Suppose that after the transformation has occurred we make a cut over the surface S . Points on either side of the cut will move relatively as the stresses relax. For simplicity suppose that the two faces of the cut shrink away from each other everywhere, leaving a gap. During the relaxation every point of the matrix or inclusion suffers a certain displacement, $-u_i^D$ say. In place of u_i^c we might take u_i^D as our displacement function. In the matrix both u_i^c and u_i^D are measured from a state in which the matrix is free of stress. Consequently in the matrix $u_i^c = u_i^D$. In the inclusion u_i^c is measured from a state in which the inclusion is untransformed and unstressed, whereas u_i^D is measured from a state in which the inclusion has transformed but is free of stress because the constraint due to the matrix has been removed. Because of the gap which appears when matrix and inclusion are cut apart u_i^D is discontinuous across S , whereas u_i^c is continuous. It is generally more convenient to use u_i^c , but we shall refer to u_i^D again in § 5.

The displacement u_i^c will be calculated with the help of a sequence of imaginary cutting, straining and welding operations. This approach is somewhat alien to the usual methods of applied mathematics; the argument can, of course, be considered to be a purely heuristic one

which points to a result (eq. (2.8)) whose validity has to be tested. We shall suppose, to begin with, that the matrix extends to infinity. In the unstrained medium mark out the boundary S of the proposed transformed inclusion. Make a cut over S and remove the inclusion. Allow the transformation to occur. After it has suffered the uniform transformation e_y^T the inclusion can no longer be fitted without strain into the cavity from which it was taken. Apply surface tractions $-\hat{\rho}_y n_j$ to the surface of the inclusion, where

$$\hat{\rho}_y^T = \lambda e_{mm} \delta_{ij} + 2\mu e_y^T. \quad (2.1)$$

This produces a strain $-e_y^T$ in the inclusion and restores it to the form it had before transformation. Put the inclusion back in the cavity, still maintaining the surface tractions. Weld the material together across S . The surface tractions thus become an embedded layer of body force of amount

$$dF_i = -\hat{\rho}_y^T n_j dS_j \quad (2.2)$$

on each element dS of S . The matrix is now unstressed and there is a uniform stress $-\hat{\rho}_y^T$ in the inclusion. Further, every point in matrix or inclusion has the same position as it had before the transformation. That is, as regards displacement, the material is in the initial state from which we have agreed to measure u_i^c . This state only differs from the required final state by the presence of the layer of body force (2.2). To get rid of this unwanted layer we apply an equal and opposite layer of body force

$$dF_i = +\hat{\rho}_y^T n_j dS = \hat{\rho}_y^T dS_j \quad (2.3)$$

over S . The displacement induced by this operation is the displacement which we are trying to calculate.

A point force F_i at r' produces a displacement

$$u_i(r) = U_{ij} F_j$$

at r , where† (LOVE [1954])

$$U_{ij} = \frac{1}{4\pi\mu} \frac{\delta_{ij}}{|r - r'|} - \frac{1}{16\pi\mu(1 - \sigma)} \frac{\partial^2}{\partial x_i \partial x_j} |r - r'| \\ = \frac{1}{4\pi\mu} \left(\frac{1}{2} \nabla^2 - \frac{1}{4(1 - \sigma)} \frac{\partial^2}{\partial x_i \partial x_j} \right) |r - r'| \quad (2.4)$$

since

$$|r - r'|^{-1} = \frac{1}{2} \nabla^2 |r - r'|. \quad (2.5)$$

† σ is Poisson's ratio.

Hence

$$u_i^c(r) = \int_S dS_k \rho_{jk}^T U_{ij}(|r - r'|). \quad (2.6)$$

But by Gauss's theorem

$$\int_{\partial S} |r - r'| dS_k = \int \frac{\partial}{\partial x_k} |r - r'| dv' = -\frac{\partial}{\partial x_k} \int |r - r'| dv' \quad \text{and so}$$

$$u_i^c(r) = \frac{1}{16\pi\mu(1-\sigma)} \rho_{jk}^T \psi_{ijk} - \frac{1}{4\pi\mu} \rho_{ik}^T \varphi_{,k} \quad (2.7)$$

or

$$u_i^c = \frac{1}{8\pi(1-\sigma)} e_{jk}^T \psi_{ijk} - \frac{1}{2\pi} e_{ik}^T \varphi_{,k} - \frac{\sigma}{4\pi(1-\sigma)} e^T \varphi_{,t} \quad (2.8)$$

where

$$\varphi(r) = \int_V \frac{dv'}{|r - r'|} \quad \text{and} \quad \psi(r) = \int_V |r - r'| dv'. \quad (2.9)$$

φ is the ordinary (harmonic or Newtonian) potential of attracting matter of unit density filling the volume V bounded by S . ψ is the corresponding biharmonic potential. Geometrically, ψ/V is the mean distance of the point r from all the points inside S .

From (2.5)

$$\nabla^2 \psi = 2\varphi. \quad \varphi^* r \cdot \frac{2}{r} \quad (2.10)$$

The following results follow from the theory of attraction (POINCARÉ [1899], MACMILLAN [1958]).

$$\nabla^2 \varphi = 2\nabla^2 \varphi = \begin{cases} -8\pi & \text{Outside } S \\ 0 & \text{Inside } S \end{cases} \quad (2.11)$$

$\varphi, \varphi_{,t}$ are continuous across S

$\varphi_{,ij}(\text{out}) - \varphi_{,ij}(\text{in}) = 4\pi n_i n_j$. $\quad (2.12)$

$\varphi_{,ij}(\text{out}) - \varphi_{,ij}(\text{in}) = 4\pi n_i n_j$. $\quad (2.13)$

The last equation gives the difference in the second derivative at two adjacent points immediately inside and outside S at a point where the normal to S is n_i . We shall use a similar notation for other quantities which are discontinuous across S . Eq. (2.13) is a re-statement of the result that the discontinuity in attraction across a double layer is 4π times its moment. More generally, a function satisfying

$$\nabla^2 U = -4\pi \varrho$$

suffers a discontinuity in its second derivatives

$$U_{,kl}(\text{out}) - U_{,kl}(\text{in}) = -4\pi \{\varrho(\text{out}) - \varrho(\text{in})\} n_k n_l \quad (2.15)$$

on crossing a surface across which ϱ is discontinuous. But $\varphi_{,ij}$ satisfies (2.14) with $\varrho = -2\varphi_{,ij}/4\pi$ and so from (2.15) we obtain the relation

$$\varphi_{,ijkl}(\text{out}) - \varphi_{,ijkl}(\text{in}) = 8\pi n_i n_j n_k n_l. \quad (2.16)$$

By similar arguments one finds that

$$\psi_{,ij}, \varphi_{,ij}, \psi_{,ijkl} \text{ are continuous across } S. \quad (2.17)$$

The stress in the matrix is

$$\rho_{ij}^c = \lambda u_{mm} \delta_{ij} + \mu(u_{ii,j}^c + u_{jj,i}^c). \quad (2.18)$$

Since the inclusion was already subject to a uniform stress $-\rho_{ij}^T$ before the body force (2.3) was applied, the stress in the inclusion is

$$\rho_{ij}^I = \rho_{ij}^c - \rho_{ij}^T \quad (2.19)$$

with ρ_{ij}^c derived from $\psi_{,ij}$ as in (2.18).

We outline a method of verifying formally that the proposed solution (2.8), (2.18), (2.19) does in fact solve the inclusion problem. From (2.7) it follows that u_i^c satisfies the equilibrium equation

$$\mu \nabla^2 u_i + (1 + \mu) u_{im,m} = 0 \quad (2.20)$$

and from (2.12), (2.17) that it is continuous across S . If the stress is defined by (2.18) and (2.19) the relations (2.13) and (2.16) show that $\varphi_{,ijkl}$ is continuous across S . Let an additional displacement $-u_i^c(r')$ be imposed on all points r' or the inner boundary of the matrix. By the uniqueness theorem of the theory of elasticity (LOVE [1954]) it will produce an additional displacement $-u_i^c(r)$ throughout the matrix, and so leave it stress-free. Likewise an additional displacement $-u_i^c(r')$ imposed on all points r' of the surface of the inclusion induces a displacement $-u_i^c(r)$ throughout its interior and so by (2.19) leaves it in a state of uniform stress $-\rho_{ij}^T$. At this point the inner surface of the matrix and the outer surface of the inclusion still fit perfectly, because of the continuity of u_i^c . If the tractions $-\rho_{ij}^T$ on the inclusion are reduced to zero the inclusion suffers a uniform strain e_{ij}^T and we are left with an unstressed matrix and an unstressed inclusion between whose surfaces there is the required misfit.

Equation (2.7) can be written in the Boussinesq-Papkovich-Neuber form

$$u_i^c = B_i - \frac{1}{4\lambda(1-\sigma)} (x_m B_m + \beta_i) \quad (2.20)$$

with harmonic B_t and β :

$$4\pi\mu B_t = -\hat{p}_{ik}^T \varphi_{ik}, \quad 4\pi\mu\beta = \hat{p}_{ik}^T f_{ik} \quad (2.21)$$

where

$$f_{ij} = x_i \varphi_{ij} - \varphi_{ij}. \quad (2.22)$$

That f_{ij} (and hence β) is harmonic inside and outside the inclusion follows from (1.5) and (2.10). Further f_{ij} behaves like r^{-1} for large r , while its normal derivative

$$\frac{\partial}{\partial x_k} n_k = \varphi \mathbf{k} \cdot \bar{n} \star \delta f_{ij}/\delta n = \varphi_{ij} n_i + x_i \varphi_{jk} n_k - \varphi_{ij} n_k$$

suffers a discontinuity $4\pi x_i n_k n_{jk} = 4\pi x_i n_j$ on passing through S , by (2.13) and (2.16). Hence f_{ij} is the harmonic potential of a layer of attracting matter distributed over S with surface density $x_i n_j$. In this way the biharmonic potential ψ can be replaced by the harmonic potential β of a certain surface layer.

It is interesting to see how much information can be obtained when our knowledge of φ and ψ is incomplete. We know in any case that φ and ψ behave as V/r and Vr for larger r , and hence, from (2.7) that the field at large distance from the inclusion is given by

$$u_i^C(r) = \frac{V \hat{e}_{ik}^T g_{ij}}{8\pi(1-\sigma)r^2} \quad (2.23)$$

where

$$g_{ij} = (1-2\sigma)(\delta_{ij}l_k + \delta_{ik}l_j - \delta_{jk}l_i) + 3u_il_il_j$$

and l_i is a unit vector joining the origin to the point of observation r .

If only φ is known we can find the dilatation and rotation:

$$\varrho^C = -\frac{1-2\sigma}{8\pi\mu(1-\sigma)} \hat{p}_{ik}^T \varphi_{ik} \quad (2.24)$$

$$4\pi\omega_{ii}^C = 2\pi(u_{ii}^C - u_{i,i}^C) = \hat{e}_{ik}^T \varphi_{ik} - \hat{e}_{ik}^T \varphi_{ik}. \quad (2.25)$$

If e_{ij}^T happens to be a pure dilatation we can find the complete field in terms of φ :

$$u_i^C = -\frac{1+\sigma}{12\pi(1-\sigma)} e^T \varphi_{ik} \quad (e_{ij}^T = \frac{1}{3} e^T \delta_{ij}) \quad (2.26)$$

(GOODIER [1937]). In this special case the dilatation has the constant value $e^T(1+\sigma)/3(1-\sigma)$ inside the inclusion. In the matrix the dilatation is zero. Consequently the value of the bulk modulus κ^* of the matrix is irrelevant and (2.26) applies also to the case of an in-

clusion of elastic constants μ, κ in a matrix with constants μ, κ^* if we give to σ the value appropriate to the inclusion (CRUM, quoted by NABARRO [1940]; ROBINSON [1951]).

Again, it may happen that it is easier to calculate φ and ψ for points within the inclusion than for points outside it. (We shall see that this is so for an ellipsoidal inclusion.) Since the strains involve φ_{ij} and ψ_{ijkl} we can use (2.13) and (2.16) to find the strain e_{ij}^C (out) (and hence the stress) at a point immediately outside the inclusion from the values e_{ij}^C (in) at the adjacent point immediately inside the inclusion. Expressed in terms of the dilatational and deviatoric parts of e_{ij}^C the result is

$$\begin{aligned} e_{ij}^C(\text{out}) &= e_{ij}^C(\text{in}) - \frac{1}{3} \frac{1+\sigma}{1-\sigma} e^T - \frac{1-2\sigma}{1-\sigma} \hat{e}_{ij}^T n_i n_j \\ e_{ii}^C(\text{out}) &= 'e_{ii}^C(\text{in}) + \frac{1}{1-\sigma} \hat{e}_{jk}^T n_j n_k n_m - 'e_{ik}^T n_k n_l - 'e_{ik}^T n_i n_l \\ &\quad + \frac{1-2\sigma}{3(1-\sigma)} \hat{e}_{jk}^T n_j n_k \delta_{il} - \frac{1}{3} \frac{1+\sigma}{1-\sigma} e^T (n_i n_l - \frac{1}{3} \delta_{il}). \end{aligned} \quad (2.27)$$

The foregoing results all refer to a transformed inclusion in an infinite matrix. If the matrix has a finite boundary the displacements will be of the form

$$u_i^F = u_i^C + u_i^{\text{im}}$$

and the stress will be

$$\hat{\rho}_{ij}^F = \hat{\rho}_{ij}^C + \hat{\rho}_{ij}^{\text{im}}$$

in the matrix and

$$\hat{\rho}_{ij}^F - \hat{\rho}_{ij}^C$$

in the inclusion. The 'image field' u_i^{im} , $\hat{\rho}_{ij}^{\text{im}}$ is free of singularities in the medium and is determined by the requirement that the sum of the C-field and the image-field shall satisfy whatever boundary conditions are imposed on the outer surface of the matrix. If the outer boundary S_0 is stress-free we must have

$$\hat{\rho}_{ij}^{\text{im}} n_j = -\hat{\rho}_{ij}^C n_j \quad \text{on } S_0$$

that is, the image-field is the field produced by surface tractions $-\hat{\rho}_{ij}^C n_j$ acting on the outer surface. If the outer surface is held immovable we must have

$$u_i^{\text{im}} = -u_i^C \quad \text{on } S_0. \quad (2.28)$$

Thus when the C-quantities are known the determination of the image quantities reduces to a standard boundary-value problem. It is sometimes convenient to have a formal expression analogous to (2.6) exhibiting u_i^F as the displacement induced by the layer of body force (2.3). When the boundary condition is (2.28) (rigidly held boundary) we may evidently write

$$u_i^F(r) = \int_S dS_k \rho_{jk}^T U_{ij}(r, r') \quad (2.29)$$

where $F_j U_{ij}(r, r')$ is the displacement at r' due to a point-force $F = (F_1, F_2, F_3)$ acting at r' in a body at whose surface the displacement is required to be zero.

With a suitable alternative definition of $U_{ij}(r, r')$ (2.29) also applies to the case where the outer boundary is stress-free. However, we cannot simply say that $F_j U_{ij}$ is the displacement at r due to a point-force at r' in a body with a stress-free surface. No such solution of the elastic equations exists, since the integral of the surface traction over S_0 must be equal to $-F$. Instead we define $U_{ij}(r, r')$ as the displacement in the body with stress-free surfaces due to a point-force F at r' , a point-force $-F$ at the origin and a double force (LOVE [1954]) at the origin of moment $-F \times r'$. Since the resultant and moment of this set of forces are both zero the condition of zero traction over S_0 can be satisfied. It can easily be shown that when the elementary forces (2.3) are summed over S the corresponding auxiliary forces and moments at the origin add up to zero. Consequently, with this definition of U_{ij} the expression (2.29) actually gives the displacement due to the layer of body force (2.3) alone.

2.2. ENERGY RELATIONS

The elastic energy associated with the inclusion (i.e. the energy in the inclusion plus the energy in the matrix) can be calculated very simply by following the energy changes which occur during the imaginary operations leading to (2.6). Suppose first that the matrix is infinite. When the inclusion has been welded back into the matrix but is still held in its untransformed shape by the layer of body force (2.2) the energy in the matrix is zero and the energy in the inclusion is

$$\frac{1}{2} \int_V \rho_{ij}^T e_{ij}^T dv. \quad (2.30)$$

When the layer of body force is relaxed each element of S moves

$$E_{\text{inc}} = E_{\infty} \left\{ 1 - \frac{1}{4} (1 + \sigma) \frac{a^3}{h^3} \right\} \quad (2.34)$$

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$\omega = \frac{1}{2} \int_V \rho_{ij}^T (-e_{ij} + e_{ij}^C) dv + \int_S \rho_{ij}^T e_{ij}^C dv_{\text{ext}}$

$\rho_{ij}^T = \frac{1}{2} \int_S \rho_{ij}^T e_{ij}^T dv + \frac{1}{2} \sum_{i,j} \int_{S_0} \rho_{ij}^T u_i^C u_j^C + \frac{1}{2} \sum_{i,j} \int_{S_0} \rho_{ij}^C u_i^C u_j^C$

through a distance u_i^C as the force on it falls from dF_i to zero. The amount of energy extracted from the elastic solid during the relaxation is thus

$$-\frac{1}{2} \int_S u_i^C dF_i = \frac{1}{2} \int_S \rho_{ij}^T u_i^C dS_j = \frac{1}{2} \int_V \rho_{ij}^T e_{ij}^C dv. \quad (2.31)$$

The energy remaining in the medium is found by subtracting (2.31) from (2.30):

$$E_{\infty} = -\frac{1}{2} \int_V \rho_{ij}^T (e_{ij}^C - e_{ij}^T) dv = -\frac{1}{2} \int_V \rho_{ij}^T e_{ij}^T dv. \quad (2.32)$$

The suffix ∞ emphasizes that this is the energy for an inclusion in an infinite matrix. Evidently we only need to know the elastic field inside the inclusion. If the stress-free strain is a pure dilatation $e_{ij}^T = \frac{1}{3} \epsilon^T \delta_{ij}$ we have by (2.26)

$$E_{\infty} = \frac{2}{9} \mu V \frac{1 + \sigma}{1 - \sigma} (\epsilon^T)^2 \quad (2.33)$$

for an inclusion of any shape. According to the argument following (2.26) the expression is still correct if the bulk moduli of matrix and inclusion differ (NABARRO [1940], ROBINSON [1951]).

Suppose next that the inclusion is situated in a finite matrix bounded by the surface S_0 . No difference is made to (2.30) if we cut away the part of the unstrained matrix exterior to S_0 . The energy removed by relaxing the layer of body force is given by (2.31) with u_i^C replaced by $u_i^C + u_i^{\text{im}}$. The elastic energy due to the stress-field of the inclusion is thus

$$E_{\text{inc}} = E_{\infty} + E_{\text{im}}$$

where the ‘image term’

$$E_{\text{im}} = -\frac{1}{2} \int_V \rho_{ij}^T e_{ij}^{\text{im}} dv = -\frac{1}{2} \int_V \rho_{ij}^{\text{im}} e_{ij}^T dv$$

varies with the position of the inclusion. As an example we consider the case of a sphere of volume $V(1 + \epsilon^T)$ forced into a spherical cavity of volume V in a semi-infinite solid. In the infinite solid u_i^C is given by (2.26) with $\varphi = V/r$. E_{∞} is given by (2.33). To find E_{im} we have to calculate e_{ik}^{im} . It is the dilatation produced by surface tractions $-\rho_{ijk}^G n_j$ acting on the free surface of the semi-infinite solid, and can be found by well-known methods (LOVE [1954]). As the dilatation is harmonic we need only find its value at the centre of the inclusion, since the mean value of harmonic function of a sphere is equal to its value at the centre. The result is

$$E_{\text{inc}} = E_{\infty} \left\{ 1 - \frac{1}{4} (1 + \sigma) \frac{a^3}{h^3} \right\} \quad (2.34)$$

where a is the radius of the sphere and h is the distance of its centre from the free surface. If $\sigma = \frac{1}{3}$ the elastic energy is reduced to $\frac{2}{3}E_\infty$ if the inclusion just reaches the surface ($h = a$). It is worth recalling at this point that, in thermodynamic terms, 'elastic energy' represents internal energy under adiabatic conditions and Helmholtz free energy under isothermal conditions (SOKOLNIKOFF [1946]). A calculation such as the foregoing covers both cases; the distinction only appears when we decide to insert either the adiabatic or isothermal values of the elastic constants.

In some applications it is necessary to consider the changes of energy which occur when an inclusion is formed in a body which is already stressed by externally applied loads. For simplicity we shall suppose that the body is stressed by surface tractions which do not vary when the outer surface S_0 of the body is slightly deformed by the introduction of the inclusion; in engineering language this is the case of 'dead loading'. The stress due to the inclusion must satisfy $\hat{\rho}_i^F n_j = 0$ on S_0 .

Let the external loads produce stress and strain $\hat{\rho}_{ij}^A, e_{ij}^A$, not necessarily uniform. Before the inclusion has transformed the elastic energy in the material is

$$E_A = \frac{1}{2} \int S_0 \hat{\rho}_{ij}^A e_{ij}^A dv \quad (2.35)$$

with the integral extending over the whole volume of the material. Suppose next that the body is subject to the combined action of the stresses due to the load and the internal stresses due to the inclusion. We might expect that the elastic energy would be the sum of (2.35), (2.34) and a cross term, representing an 'interaction energy'. But in fact the cross term is zero. To see this, suppose that the transformation occurs first, in the absence of external loads. The elastic energy is E_{inc} . Now let the load be applied. Within the limits of the usual infinitesimal theory of elasticity the body responds to external forces just as it would if it were not self-stressed by the transformed inclusion. The work done on the body by the load is thus

$$\frac{1}{2} \int S_0 \hat{\rho}_{ij}^A u_i^A ds_j = \frac{1}{2} \int \hat{\rho}_{ij}^A e_{ij}^A dv = E_A$$

and so the total elastic energy is simply $E_{\text{inc}} + E_A$. The same conclusion can be reached analytically. If the volume integral of the energy density is converted into a surface integral over S and the outer surface of the matrix it will be found that the cross term between the A- and F-terms vanishes because $\hat{\rho}_{ij}^F n_j = 0$ on S_0 .

Despite the lack of a cross term in the elastic energy it is possible to define a physically meaningful interaction energy. (For a general discussion in the context of solid state theory cf. PEACH [1951], ESHELBY [1951, 1956].) To introduce the concept of interaction energy we begin by enquiring whether in the presence of the stresses arising from the external load it is energetically possible for the inclusion to form spontaneously. At first sight it appears as if the answer is no, since the elastic energy increases by the necessarily positive quantity E_{inc} when the inclusion is introduced. However, we have to consider not simply the elastic energy of the material, but rather the total energy, E_{tot} say, of the closed system made up of the body and the loading mechanism. When the transformation occurs in the presence of the external load the increase in the potential energy of the loading mechanism is equal to the work which the surface tractions $\hat{\rho}_{ij}^A n_j$ do on the body as the surface displacements change from u_i^A to $u_i^A + u_i^F$. Thus the increase in the energy of the whole system (inclusion plus matrix plus loading mechanism) is

$$\Delta E_{\text{tot}} = E_{\text{inc}} + E_{\text{int}} \quad \left. \begin{array}{l} \text{ak - he calls} \\ \text{the interaction } \hookrightarrow \\ \text{energy rise due} \\ \text{put loading mech.} \end{array} \right\} \text{in the (2.38) system} \quad (2.36)$$

where

$$E_{\text{int}} = - \int S_0 \hat{\rho}_{ij}^A u_i^F ds_j. \quad (2.37)$$

We may write

$$E_{\text{tot}} = E_0 + E_A + E_{\text{inc}} + E_{\text{int}} \quad \left. \begin{array}{l} \text{the interaction } \hookrightarrow \\ \text{energy rise due} \\ \text{put loading mech.} \end{array} \right\} \text{in the (2.38) system} \quad (2.38)$$

where E_0 is the potential energy of the loading mechanism in the absence of the inclusion. In (2.38) the first two terms depend only on the elastic field due to the load and the third only on the field due to the inclusion. The last term, which depends on both these fields, may be appropriately called the interaction energy. We can now answer the question posed in the preceding paragraph. If the applied stress is chosen so that $\Delta E_{\text{tot}} < 0$ (and this can always be done) it is energetically possible for the transformation to take place spontaneously. Generally ΔE_{tot} , whether it is positive or negative, may be called the energy of formation of the inclusion in the applied stress-field. This concept is familiar in thermodynamics. In fact, if the transformation leading to the formation of the inclusion occurs under adiabatic conditions ΔE_{tot} represents, in thermodynamic language, the increase in the enthalpy of the body, while if the transformation takes place under isothermal conditions ΔE_{tot} is the in-

crease in its Gibbs free energy. This follows at once from the definition of these quantities. The increase in enthalpy associated with some change in the state of a thermally isolated system is equal to the increase in its internal energy plus the work which it does on its environment during the course of the change. For a change at constant temperature the Gibbs free energy is similarly defined, replacing ‘internal energy’ by ‘Helmholtz free energy’. That ΔE_{tot} is the change in enthalpy or Gibbs free energy follows from the fact that, as we have seen, the ‘elastic energy’ has to be identified with the internal energy in adiabatic processes and with Helmholtz free energy in isothermal processes. Thus ΔE_{tot} is the enthalpy or Gibbs free energy of formation of the inclusion.

The expression (2.39) for the interaction energy can be put into a more useful form by the following artifice. We re-write (2.37) as

$$E_{\text{int}} = - \int_{S_0} (\phi_{ij}^A u_i^A - \phi_{ij}^C u_i^C) dS_j. \quad (2.39)$$

Because $\phi_{ij}^F u_i^A = 0$ on S_0 the added term is zero. Consider the divergence of the integrand. It can be split into two terms

$$(\phi_{ij}^{\text{im}} u_i^A - \phi_{ij}^A u_i^{\text{im}})_j = \phi_{ij}^{\text{im}} e_{ij}^A - \phi_{ij}^A e_{ij}^{\text{im}} \quad (2.40)$$

$$(\phi_{ij}^C u_i^A - \phi_{ij}^A u_i^C)_j = \phi_{ij}^C e_{ij}^A - \phi_{ij}^A e_{ij}^C. \quad (2.41)$$

By (1.3) the expression (2.40) is zero both in the matrix and the inclusion and so the image terms make no contribution. On the other hand, (2.41) vanishes in the matrix but not in the inclusion, since ϕ_{ij}^C is discontinuous across S . Hence in (2.39) we may replace $\phi_{ij}^F u_i^F$ by $\phi_{ij}^C u_i^C$ and carry out the integration over the boundary of the inclusion instead of over the outer surface of the body; that is,

$$E_{\text{int}} = \int_S (\phi_{ij}^C u_i^A - \phi_{ij}^A u_i^C) dS_j. \quad (2.42)$$

In (2.42) we may replace ϕ_{ij}^C by ϕ_{ij}^I since $\phi_{ijm}^C = \phi_{ijm}^I$ on S . The integral can then be converted to a volume integral over the inclusion:

$$E_{\text{int}} = \int_V (\phi_{ij}^I e_{ij}^A - \phi_{ij}^A e_{ij}^I) dV$$

By (1.3) the integrand is equal to $(\phi_{ij}^I - \phi_{ij}^C) e_{ij}^A$ and so by (2.19)

$$E_{\text{int}} = - \int_V \phi_{ij}^T e_{ij}^A dV = - \int_V \phi_{ij}^A e_{ij}^T dV. \quad (2.43)$$

Evidently to find the interaction energy we need only know the stress-free strain e_{ij}^T ; it is unnecessary to solve the elastic problems associated with the determination of u_i^C or u_i^{im} .

§ 3. The Ellipsoidal Inclusion

3.1. THE ELASTIC FIELD

When the inclusion is bounded by the ellipsoid

$$x^2/a^2 + y^2/b^2 + z^2/c^2 = 1$$

the elastic field may be found explicitly. The form of the harmonic potential φ is well-known (KELLOGG [1929]). There is an analogous expression for the biharmonic potential ψ (ESHELBY [1959b]), but in fact all the derivatives of ψ which enter (2.7) can be found in terms of derivatives of φ . Let us compare f_{12} (eq. (2.7)) with the function $g = a^2(x_1\varphi_2 - x_2\varphi_1)/(a^2 - b^2)$ which plays a role in the hydrodynamic theory of rotating ellipsoids. Each of these functions is harmonic inside and outside the ellipsoid, is continuous across its surface and falls to zero at infinity. Across the surface there is a discontinuity $4\pi x_1 n_2$ in the normal derivative of f_{12} (eq. (2.22)). The corresponding discontinuity for g is $4\pi a^2(x_1 n_2 - x_2 n_1)/(a^2 - b^2)$. This is simply $4\pi x_1 n_2$ in view of the relation $b^2 x_1 n_2 = a^2 x_2 n_1$ which follows from the expressions

$$\begin{aligned} n_1 &= x_1/a^2 h, \\ n_2 &= x_2/b^2 h, \quad (h^2 = x^2/a^4 + y^2/b^4 + z^2/c^4) \\ n_3 &= x_3/c^2 h, \end{aligned}$$

for the components of the normal to an ellipsoid. Hence f_{12} and g are identical, being harmonic potentials of the same surface distribution. We therefore have

$$f_{12} = x_1 \varphi_{12} - \varphi_{12} = \frac{a^2}{a^2 - b^2} (x_1 \varphi_{12} - x_2 \varphi_{11})$$

and similarly for the other f_{ij} with $i \neq j$. Thus

$$\varphi_{12} = \frac{a^2}{a^2 - b^2} x_2 \varphi_{11} + \frac{b^2}{b^2 - a^2} x_1 \varphi_{12}$$

$$\varphi_{23} = \frac{b^2}{b^2 - c^2} x_3 \varphi_{12} + \frac{c^2}{c^2 - b^2} x_2 \varphi_{13}$$

$$\varphi_{31} = \frac{c^2}{c^2 - a^2} x_1 \varphi_{13} + \frac{a^2}{a^2 - c^2} x_3 \varphi_{11}.$$

It is more difficult to derive expressions for f_{11} , f_{22} , f_{33} and hence for ψ_{11} , ψ_{22} , ψ_{33} . However, there is no need to obtain them, since all the third derivatives ψ_{ijj} appearing in (2.8) can be made to depend on φ , ψ_{12} , ψ_{23} , ψ_{31} . We may write, for example,

$$\psi_{112} = (\psi_{12})_{,1} \quad \psi_{111} = 2\varphi_{,1} - (\psi_{12})_{,2} - (\psi_{13})_{,3}.$$

The first relation is trivial; the second is obtained by differentiating $\nabla^2\varphi = 2\varphi$ with respect to x_1 .

Substitution in (2.8) gives

$$8\pi(1-\sigma)u_1^C = \frac{e_{22}^T - e_{11}^T}{a^2 - b^2} \frac{\partial}{\partial x_2} (a^2 x_2 \varphi_{,1} - b^2 x_1 \varphi_{,2}) +$$

$$\frac{e_{33}^T - e_{11}^T}{c^2 - a^2} \frac{\partial}{\partial x_3} (c^2 x_1 \varphi_{,3} - a^2 x_3 \varphi_{,1}) -$$

$$2\{(1-\sigma)e_{11}^T + \sigma(e_{22}^T + e_{33}^T)\}\varphi_{,1} -$$

$$4(1-\sigma)(e_{12}^T \varphi_{,2} + e_{13}^T \varphi_{,3}) + \frac{\partial}{\partial x_1} \beta \quad (3.1)$$

where

$$\beta = \frac{2e_{12}^T}{a^2 - b^2} (a^2 x_2 \varphi_{,1} - b^2 x_1 \varphi_{,2}) +$$

$$\frac{2e_{23}^T}{b^2 - c^2} (b^2 x_3 \varphi_{,2} - c^2 x_2 \varphi_{,3}) +$$

$$\frac{2e_{31}^T}{c^2 - a^2} (c^2 x_1 \varphi_{,3} - a^2 x_3 \varphi_{,1}).$$

u_2^C and u_3^C are found by cyclic permutation of (1, 2, 3). (a, b, c).

At an internal point

$$\varphi = \frac{1}{2}(a^2 - x^2)I_a + \frac{1}{2}(b^2 - y^2)I_b + \frac{1}{2}(c^2 - z^2)I_c \quad (3.2)$$

where I_a , I_b , I_c are constants depending only on the axial ratios of the ellipsoid. Consequently u_i^C is a linear function of the x_i and the strain and stress are uniform within the inclusion, as stated in the Introduction. The constant strains e_{ij}^C are linear functions of the e_{ij}^T and we may write

$$e_{ij}^C = S_{ijkl} e_{kl}^T. \quad (3.3)$$

The S_{ijkl} are symmetric in ij and in kl , but in general S_{ijkl} is different from S_{klij} . It is easy to verify that these coefficients vanish unless they are of the form S_{1111} , S_{1112} or S_{1122} ($i \neq j$; no summation). That is,

Sphere (Isotropy):

$$S_{1111} = \frac{2 - 5\alpha^2}{15(1-\alpha^2)}; \quad S_{1112} = -\frac{1 - 5\alpha^2}{15(1-\alpha^2)}; \quad S_{1122} = \frac{1 - 5\alpha^2}{15(1-\alpha^2)}$$

unless shears are not coupled, and shears are not coupled to extensions. From (3.1) we obtain

$$8\pi(1-\sigma)S_{1111} = \frac{a^2 I_a - b^2 I_b}{a^2 - b^2} + \frac{a^2 I_a - c^2 I_c}{a^2 - c^2} - \frac{1}{2}(1-\sigma)I_a \quad (3.4)$$

$$8\pi(1-\sigma)S_{1122} = -b^2 \frac{I_a - I_b}{a^2 - b^2} - (1-2\sigma)I_a$$

$$8\pi(1-\sigma)S_{1212} = -\frac{1}{2} \frac{a^2 + b^2}{a^2 - b^2} (I_a - I_b) + \frac{1}{2}(1-2\sigma)(I_a + I_b); \quad \text{check}$$

the remaining coefficients may be obtained by cyclic interchange. (For an alternative way of obtaining these coefficients see ESHELBY [1957].) The rotation inside the inclusion is also constant. We have at once from (2.25) and (3.2)

$$4\pi a \omega_{12}^C = (I_b - I_a) e_{22}^T, \quad (3.5)$$

$$4\pi a \omega_{23}^C = (I_c - I_b) e_{23}^T,$$

$$4\pi a \omega_{31}^C = (I_a - I_c) e_{31}^T.$$

In their role of demagnetising factors I_a , I_b , I_c have been plotted as functions of b/a and c/a by OSBORN [1945] with the notation $I_a = L$, $I_b = M$, $I_c = N$. They may also be found from tables of elliptic integrals $F(\theta, k)$, $E(\theta, k)$ using the relations

$$I_a = \frac{4\pi abc}{(a^2 - b^2)(a^2 - c^2)^{\frac{1}{2}}} (F - E), \quad (3.6)$$

$$I_b = 4\pi - I_a - I_c,$$

$$I_c = \frac{4\pi abc}{(b^2 - c^2)(a^2 - c^2)^{\frac{1}{2}}} \left\{ \frac{(b^2 K^2 - c^2)^{\frac{1}{2}}}{ac} - E \right\},$$

$$k^2 = \frac{a^2 - b^2}{a^2 - c^2}, \quad \theta = \sin^{-1} \left(1 - \frac{c^2}{a^2} \right)^{\frac{1}{2}}.$$

(Compare (3.7) below.)

For a point outside the ellipsoid the potential takes the form (KELLOGG [1929], MACMILLAN [1958])

$$\varphi = \frac{2\pi abc}{l^3} \left[\left(l^2 - \frac{x^2}{k^2} + \frac{y^2}{l^2} \right) F(\theta, k) + \left[\frac{x^2}{k^2} - \frac{y^2}{k^2 k'^2} + \frac{z^2}{l^2 k'^2} \right] E(\theta, k) \right. \\ \left. + \frac{l}{k'^2} \left[\frac{C}{AB} y^2 - \frac{B}{AC} z^2 \right] \right] \quad (3.7)$$

where

$$A = (a^2 + \lambda)^{\frac{1}{2}}, \quad B = (b^2 + \lambda)^{\frac{1}{2}}, \quad C = (c^2 + \lambda)^{\frac{1}{2}} \quad (3.8)$$

$$l = (a^2 - c^2)^{\frac{1}{2}}, \quad k^2 = 1 - k'^2 = \frac{a^2 - b^2}{a^2 - c^2}$$

$$a^2 > b^2 > c^2,$$

and F, E are elliptic integrals of modulus k and argument

$$\theta = \sin^{-1}(l/A).$$

λ is the greatest (and in fact the only positive) root of

$$x^2/A^2 + y^2/B^2 + z^2/C^2 = 1.$$

Equation (3.7) also gives the potential at an internal point if we put $\lambda = 0$; this gives (3.6).

To carry out the differentiations necessary to find the displacement or stress outside the inclusion one can make repeated use of

$$\partial F/\partial \lambda = -\frac{1}{2}V/ABC,$$

$$\partial \lambda/\partial x = 2x/Ah, \dots,$$

$$h^2 = x^2/A^4 + y^2/B^4 + z^2/C^4.$$

In forming the first (but not the higher) derivatives of φ , λ may be treated as a constant. The condition (3.9) is not really necessary. It ensures that $0 < k^2 < 1$ and $0 < \theta < \frac{1}{2}\pi$. If it is violated $F(\theta, k), E(\theta, k)$ can be made to depend on $F(\theta_1, k_1), E(\theta_1, k_1)$ with $0 < k_1^2 < 1, 0 < \theta_1 < \frac{1}{2}\pi$ with the help of known transformations. (See, for example, BYRD and FRIEDMAN [1954].) This is useful if it becomes convenient to ignore (3.9) at a late stage in a calculation.

The results we have obtained can only be applied to the sphere after a tedious passage to the limit. However, we may use the following expressions for the potentials of a sphere of radius a :

$$\begin{aligned} \varphi &= \frac{4}{3}\pi a^2 \left(\frac{3}{2} - \frac{1}{2} \frac{r^2}{a^2} \right), & r < a \\ &= \frac{4}{3}\pi a^2 \frac{a}{r}, & r > a \end{aligned} \quad (3.12)$$

$$\begin{aligned} \psi &= \frac{4}{3}\pi a^4 \left(\frac{3}{4} + \frac{1}{2} \frac{r^2}{a^2} - \frac{1}{20} \frac{r^4}{a^4} \right), & r < a \\ &= \frac{4}{3}\pi a^4 \left(\frac{1}{5} \frac{a}{r} + \frac{r}{a} \right), & r > a \end{aligned} \quad (3.13)$$

The expression for φ is well-known. ψ may be found by integrating

$$\frac{1}{r^2} \frac{d}{dr} \left(r^2 \frac{d\psi}{dr} \right) = 2\varphi$$

and calculating $\psi(0)$ by direct integration:

$$\psi(0) = 4\pi \int_0^a r \cdot r^2 dr = \pi a^4.$$

For the sphere (3.3) reduces to

$$e^C = \alpha e^T, \quad e_{ij}^C = \beta e_{ij}^T$$

with

$$\alpha = \frac{1}{3} \frac{1+\sigma}{1-\sigma}, \quad \beta = \frac{2}{15} \frac{4-5\sigma}{1-\sigma}. \quad (3.14)$$

KRÖNER [1958b] has given the S_{ijkl} (with the notation e_{ijkl}^{-1}) for prolate and oblate spheroids.

For some purposes it may be unnecessary to deal with the relatively complex field outside the inclusion. A knowledge of the e_{ij}^T alone is enough to give the field far from the inclusion (2.23) or the interaction energy with an applied field (2.43). When the numerical coefficients S_{ijkl} have been computed we can find the elastic field inside the inclusion, and also, with the help of (2.27), the field at points in the matrix immediately outside the inclusion.

The displacement (3.1) at points external to the ellipsoid may be regarded as the solution to one or other of the following boundary-value problems:

- (i) To find the elastic field falling to zero at infinity and having the displacement

$$u_{ij} = (e_{ij}^C + \omega_{ij}^C)x_j \quad (e_{ij}^C, \omega_{ij}^C \text{ constant})$$

over the surface of an ellipsoid.

- (ii) To find the elastic field falling to zero at infinity and giving surface tractions

$$T_i = \dot{\phi}_{ij}^I n_j \quad (\dot{\phi}_{ij}^I \text{ constant})$$

on the surface of an ellipsoid.

The solution (3.1) is designed to give the constrained elastic field directly when the stress-free strain e_{ij}^T is known. Thus in using it to solve (i) or (ii) the first step would have to be the solution of (3.3), or (2.19) and (3.3) to find the e_{ij}^T appropriate to the prescribed u_{ij}^C or $\dot{\phi}_{ij}^I$. Regarded as a solution of (i) the solution (3.1) is closely related to one already given by DANIELE [1911]. Again, if a constant stress-field $\dot{\phi}_{ij}^A = -\dot{\phi}_{ij}^I$ is superimposed on the solution of (ii) the ellipsoidal surface is free of stress and we are left with a solution representing an ellipsoidal cavity perturbing a uniform stress $\dot{\phi}_{ij}^A$. The solution has been given by SANDOWSKY and STERNBERG [1949] for the case where

$$\dot{\phi}_{12}^A = \dot{\phi}_{23}^A = \dot{\phi}_{31}^A = 0.$$

Daniele determined the displacements in an infinite medium outside

an ellipsoidal surface over which the displacement is required to be

$$(3.15) \quad u_i = \xi_{ij} \varphi_{ij}$$

with constant coefficients ξ_{ij} . He assumed that there was a solution of the form

$$u_i = \alpha_{ij} \varphi_{ij} + \alpha_{ijkl} x_j \varphi_{kl}, \quad (\alpha_{ij}, \alpha_{ijkl}, \lambda_{ij} \text{ constant}).$$

$\epsilon = \lambda_{ij} \varphi_{ij}$

The α_{ij} and α_{ijkl} were determined by substituting in the equilibrium equations, equating the coefficients of φ_{ij} and φ_{ijkl} to zero (subject to the conditions $\nabla^2 \varphi_{ij} = 0, \nabla^2 \varphi_{ijkl} = 0$) and applying the boundary condition (3.15). Actually, Daniele's solution is more general than (3.1) since we cannot prescribe e_{ij}^c and ω_{ij}^c independently. If the e_{ij}^c are given the e_{ij}^T follow from (3.3) and the ω_{ij}^c are then fixed by (3.5). This connection between the e_{ij}^c and ω_{ij}^c is due to the fact that there is no external couple acting on the inclusion. Thus, in physical terms, Daniele's solution gives the field about a rigid embedded inclusion which suffers a homogeneous deformation and is then rotated from its equilibrium position by an external couple. (The case where the ellipsoid is rotated but not deformed was discussed earlier by EDWARDES [1893].) When the requirement of zero couple is imposed Daniele's solution agrees with (3.1).

Sadowsky and Sternberg presented their solution in ellipsoidal coordinates. The problem they had in mind was that of an ellipsoidal cavity perturbing a stress field which at infinity is uniform and has its principal axes parallel to those of the ellipsoid; ROBINSON [1951] and NIESEL [1953] have pointed out that it can be applied to the inclusion problem. Robinson treats the case where e_{ij}^T is a pure dilatation; Niesel considers the more general case where $e_{11}^T, e_{22}^T, e_{33}^T$ are unequal and $e_{12}^T, e_{23}^T, e_{31}^T$ are zero.

Sadowsky and Sternberg's solution is expressed in the form (2.20)

$$\begin{aligned} B_1 &= \mathcal{C} A_1 X, & B_2 &= \mathcal{C} A_2 Y, & B_3 &= \mathcal{C} A_3 Z, \\ \beta &= \mathcal{C} A_4 F_1 + \mathcal{C} A_5 F_2 \end{aligned} \quad (3.16)$$

with

$$\begin{aligned} X &= -v k \varphi_{11}, & Y &= +v h k'^2 \varphi_{12}, & Z &= -v \frac{k'^2}{k} \varphi_{13} \\ F_1 &= -v k^3 f_{11} = v k^3 (2\varphi - x_i \varphi_{ii}) \\ F_2 &= \gamma_{bc} f_{11} + \gamma_{ca} f_{22} + \gamma_{ab} f_{33} \end{aligned} \quad (3.15)$$

with

$$\begin{aligned} \gamma_{qr} &= \frac{2(\beta - 1)}{3\pi abc} (d^2 - q^2)(d^2 - r^2) & (q, r = a, b, c) \\ v &= \frac{b^3}{4\pi abc}, & d^2 &= a^2 - 2 \frac{a^2 - b^2}{\beta(1 + k^2)} \\ \beta &= 2 + \frac{2(k'^2 + k^4)^{\frac{1}{2}}}{1 + k^2}. \end{aligned}$$

For completeness we have given the values of the numerical coefficients connecting φ and X, Y, Z, F_1, F_2 . We note that f_{11}, f_{22}, f_{33} (eq. (2.22)) only enter the solution through the two particular linear combinations F_1 and F_2 . This is a result of Sadowsky and Sternberg's requirement that F_1 and F_2 (and also X, Y, Z) shall take the form

$$f(\alpha_1) g(\alpha_2) h(\alpha_3) \quad (3.17)$$

when expressed in terms of their ellipsoidal coordinates $\alpha_1, \alpha_2, \alpha_3$. Thus the A_n cannot be found by direct comparison of (3.17) and (2.21). For example, if we choose A_4 and A_5 so that the coefficients of f_{11} and f_{22} agree with (2.7), the coefficient of f_{33} is fixed and so βf_{33} cannot be prescribed at will. This could be remedied by adding to β a further harmonic function, $\mathcal{C} A'_5 F'_2$, say, where F'_2 is derived from F_2 by changing the sign of the radical in the equation defining β . (Cf. SADOWSKY and STERNBERG [1949] eq. (41).) There is, however, no need to do this. In a representation such as (2.20) it is possible to modify B_i and β simultaneously and leave u_i^c unchanged. Sadowsky and Sternberg give a matrix relation (their eq. (50)) which enables the A_n to be determined in such a way that the resulting stress annuli the surface tractions on the ellipsoidal surface due to uniform stresses $\phi_{11} = \sigma_1, \phi_{22} = \sigma_2, \phi_{33} = \sigma_3$.

According to NIESEL [1953] it is convenient to include the solution F'_2 in β if we wish to pass to the limiting case of a spheroid. To extend Sadowsky and Sternberg's solution to the case where the principal axes of the stress at infinity are not parallel to the axes of the cavity (or to apply it to the inclusion problem with non-vanishing $e_{11}^T, e_{22}^T, e_{33}^T$) it would be necessary to add to β terms $\mathcal{C} A_6(f_{12} + f_{21}), \mathcal{C} A_7(f_{23} + f_{32}), \mathcal{C} A_8(f_{31} + f_{13})$. These terms have the required form (3.17) when transcribed into ellipsoidal coordinates.

(We have introduced the constant $\mathcal{C} = -2(1 - \sigma)/\mu$ to make our notation agree with Sadowsky and Sternberg's.) The A_n are numerical coefficients and X, Y, Z, F_1, F_2 are harmonic functions chosen so that the boundary conditions can be satisfied. They are related to the potential φ (eq. (3.7)) as follows:

The use of ellipsoidal coordinates does not seem to offer much advantage. As we have seen, the problem may be set up and solved formally in Cartesian coordinates. To find the elastic field at a given point (x, y, z) outside the ellipsoid we have to solve the cubic (3.11) for λ . We appear to be spared this when ellipsoidal coordinates are used. But, in fact, we must have already prepared an ellipsoidal coordinate network in order to be able to locate points relative to the ellipsoid, and, further, a different network is required for each value of the axial ratio b/a .

LURIE [1952] has given a solution to Sadowsky and Sternberg's problem in the disconcertingly simple form

$$\begin{aligned} B_1 &= (M/a^2)\varphi_1, & B_2 &= (M/b^2 - N)\varphi_2, & B_3 &= (M/c^2 - N)\varphi_3 \\ \beta &= N(x\varphi_1 - 2\varphi) + P\varphi \end{aligned}$$

where M, N, P are disposable constants. (Our M, N, P differ from Lurie's by a common multiplicative factor.) Unfortunately the contributions to μ_i^c from the terms in N and P are both of the form const. $\varphi\alpha$. (This illustrates the fact mentioned above that different choices of B_i, β can give the same elastic field.) Consequently there are really only two disposable constants, and so only two of $\rho_{11}, \rho_{22}^T, \rho_{33}^T$ (or, in the cavity problem, $\sigma_1, \sigma_2, \sigma_3$) can be prescribed independently.

3.2. THE INHOMOGENEOUS INCLUSION

So far we have been concerned with a homogeneous inclusion, that is, one which has the same elastic constants as the matrix. We consider next the case where the elastic constants of matrix and inclusion are different. As in § 2 we imagine that the inclusion undergoes a transformation specified by a uniform stress-free strain whilst constrained by the matrix, and try to calculate the resulting elastic field. This must be distinguished from the problem considered in § 4 below. There we have, in effect, a *perfectly fitting* inhomogeneous insertion cemented into a cavity in the matrix, and consequently the material is everywhere stress-free in the absence of applied forces.

We have seen that (2.26) still applies when the bulk moduli of matrix and inclusion differ. This seems to be the only simple statement one can make for an inclusion of arbitrary shape. For the ellipsoid, on the other hand, we may solve the general problem very simply by taking advantage of the fact that the stress is constant inside a homogeneous ellipsoidal inclusion (ROBINSON [1951], NISER [1953], ESHELBY [1957]).

The use of ellipsoidal coordinates does not seem to offer much advantage. As we have seen, the problem may be set up and solved formally in Cartesian coordinates. To find the elastic field at a given point (x, y, z) outside the ellipsoid we have to solve the cubic (3.11) for λ . We appear to be spared this when ellipsoidal coordinates are used. But, in fact, we must have already prepared an ellipsoidal coordinate network in order to be able to locate points relative to the ellipsoid, and, further, a different network is required for each value of the axial ratio b/a .

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$$B_1 = (M/a^2)\varphi_1,$$

$$B_2 = (M/b^2 - N)\varphi_2,$$

$$B_3 = (M/c^2 - N)\varphi_3$$

$$\beta = N(x\varphi_1 - 2\varphi) + P\varphi$$

where M, N, P are disposable constants. (Our M, N, P differ from Lurie's by a common multiplicative factor.) Unfortunately the contributions to μ_i^c from the terms in N and P are both of the form const. $\varphi\alpha$. (This illustrates the fact mentioned above that different choices of B_i, β can give the same elastic field.) Consequently there are really only two disposable constants, and so only two of $\rho_{11}, \rho_{22}^T, \rho_{33}^T$ (or, in the cavity problem, $\sigma_1, \sigma_2, \sigma_3$) can be prescribed independently.

To find e_{ij}^T for the equivalent homogeneous inclusion we use (3.3) to express e_{ij}^c in terms of e_{ij}^T . For the non-diagonal components we have simply

$$e_{12}^T = \frac{\mu^*}{2(\mu^* - \mu)} S_{1212}^T + \mu e_{12}^T, \quad e_{23}^T = \dots$$

To find $e_{11}^T, e_{22}^T, e_{33}^T$ we have to solve the set of three simultaneous equations

$$(1^* - \lambda)e_{11}^c + \lambda e_{11}^T + 2(\mu^* - \mu) S_{1111}^T = \underbrace{\lambda^* e_{11}^T}_{e_{11}^G} + 2\mu e_{11}^T$$

$$+ 2\mu e_{21}^T$$

$$(ij = 11, 22, 33).$$

Only $e_{11}^T, e_{22}^T, e_{33}^T$ enter the kl -summation, and in the first term we have

$$e_{ii}^c = \frac{1 - 2\sigma}{4\pi(1 - \sigma)} (I_a e_{ii}^T + I_b e_{ii}^T + I_c e_{ii}^T) + \frac{\sigma}{1 - \sigma} e^T$$

by (2.24) and (3.2).

It is a simple matter to calculate the elastic energy E_∞^* associated with an inhomogeneous (ellipsoidal) inclusion in an infinite matrix.

For the equivalent homogeneous inclusion we have from (2.32)

$$E_\infty = -\frac{1}{2} V \rho_y^T e_{yy}^T$$

$$(V = \frac{4}{3}\pi abc).$$

For brevity let E denote the ellipsoidal inclusion we have been considering hitherto. It has the same elastic constants λ, μ as the matrix, and it has suffered a permanent change of shape characterized by the stress-free strain e_{ij}^T while embedded in the matrix. The strain e_{ij}^c relates its final form to its form before transformation. Take a second ellipsoid E^* which to begin with has the same form as E had before its transformation and which has elastic constants λ^*, μ^* . Let E^* undergo a stress-free strain e_{ij}^{T*} . To E^* apply surface tractions chosen so as to produce a uniform elastic strain $e_{ij}^c - e_{ij}^T$ in it. It then has precisely the same form as the embedded inclusion E . If this treatment should happen also to produce in E^* stresses identical with those in E we can replace E by E^* without upsetting the continuity of displacement and surface traction across the interface. The condition for this to be possible is (cf. eq. (2.19))

$$p_{ij}^T = \lambda(e^c - e^T)\delta_{ij} + 2\mu(e_{ij}^c - e_{ij}^T)$$

$$= \lambda^*(e^c - e^{T*})\delta_{ij} + 2\mu^*(e_{ij}^c - e_{ij}^{T*}).$$

When the values of λ^*, μ^* and e_{ij}^{T*} for the inhomogeneous inclusion are given we can solve (3.18) for the e_{ij}^T . The elastic field inside and outside the inhomogeneous inclusion is then identical with that of a homogeneous inclusion with the stress-free strain e_{ij}^T .

To find e_{ij}^T for the equivalent homogeneous inclusion we use (3.3) to express e_{ij}^c in terms of e_{ij}^T . For the non-diagonal components we have simply

$$e_{12}^T = \frac{\mu^*}{2(\mu^* - \mu)} S_{1212}^T + \mu e_{12}^T, \quad e_{23}^T = \dots$$

To find $e_{11}^T, e_{22}^T, e_{33}^T$ we have to solve the set of three simultaneous equations

$$(1^* - \lambda)e_{11}^c + \lambda e_{11}^T + 2(\mu^* - \mu) S_{1111}^T = \underbrace{\lambda^* e_{11}^T}_{e_{11}^G} + 2\mu e_{11}^T$$

$$+ 2\mu e_{21}^T$$

$$(ij = 11, 22, 33).$$

Only $e_{11}^T, e_{22}^T, e_{33}^T$ enter the kl -summation, and in the first term we have

$$e_{ii}^c = \frac{1 - 2\sigma}{4\pi(1 - \sigma)} (I_a e_{ii}^T + I_b e_{ii}^T + I_c e_{ii}^T) + \frac{\sigma}{1 - \sigma} e^T$$

by (2.24) and (3.2).

It is a simple matter to calculate the elastic energy E_∞^* associated with an inhomogeneous (ellipsoidal) inclusion in an infinite matrix.

For the equivalent homogeneous inclusion we have from (2.32)

$$E_\infty = -\frac{1}{2} V \rho_y^T e_{yy}^T$$

$$(V = \frac{4}{3}\pi abc).$$

The energy in the matrix is the same for E and E^* . The internal stress is ρ_{ij}^I for both, but the effective strain is $e_{ij}^C - e_{ij}^T$ for E and $e_{ij}^C - e_{ij}^T$ for E^* . Thus we have to add $\frac{1}{2}V\rho_{ij}^I(e_{ij}^C - e_{ij}^T) - \frac{1}{2}V\rho_{ij}^*(e_{ij}^C - e_{ij}^T)$ to (3.20), which gives the simple result (cf. ROBINSON [1951])

$$E_\infty^* = -\frac{1}{2}V\rho_{ij}^I e_{ij}^{T*}. \quad (3.21)$$

The results of this section may be adapted to the case of an ellipsoidal cavity containing a fluid under pressure. If the pressure P of the fluid is prescribed it is only necessary to put $\rho_{ij}^I = -P\delta_{ij}$ in (3.18) and solve for the e_{ij}^T or e_{ij}^{T*} . If instead we are given the excess volume v of fluid introduced (measured at zero pressure) we have to put $e^T = v$ and require ρ_{ij}^I to have the form $\frac{1}{2}\rho_{ij}^I\delta_{ij}$. The solution of (3.18) then gives ρ_{ij}^I , e_{ij}^T and e_{ij}^{T*} .

§ 4. The Ellipsoidal Inhomogeneity

4.1. THE ELASTIC FIELD

In this section we shall take up the second problem mentioned in the introduction. It may be formulated as follows:

An infinite solid has elastic constants λ^* , μ^* inside a region bounded by a closed surface S (the 'inhomogeneity') and elastic constants λ , μ in the region outside S (the 'matrix'). To find the elastic field everywhere when the strain is required to reduce to the constant value e_{ij}^A far from S .

Although the problem can only be solved in detail for an ellipsoidal inhomogeneity it is convenient to start from the case where the form of the inhomogeneity is arbitrary. The problem can be reduced to the determination of the elastic field produced by a certain layer of body-force distributed over S . To see this, suppose that the strain e_{ij}^A is impressed throughout the medium. The displacement is then

$$u_i^A = e_{ij}^A x_j.$$

plus an inessential rigid-body displacement. The equilibrium equations are satisfied inside and outside S . The traction on the inner boundary of the matrix is $(1e_{ij}^A + 2\rho_{ij}^A)\eta_j$, but the traction on the outer surface of the inhomogeneity is $(\lambda^*e_{ij}^A + 2\mu^*e_{ij}^A)\eta_j$. Consequently the required state of strain can only be maintained if there is a layer of body-force of surface density $\{(1^* - \lambda)e_{ij}^A + 2(\mu^* - \mu)e_{ij}^A\}\eta_j$ spread

over S . To find the actual elastic field we apply an equal and opposite layer of body force of surface density

$$T_i = \{(1 - \lambda^*)e^A \delta_{ij} + 2(\mu - \mu^*)e_{ij}^A\}\eta_j \quad (4.1)$$

and calculate the displacement u_i^C which it induces in the medium. The final displacement is then

$$u_i = e_{ij}^A x_j + u_i^C.$$

u_i^C is evidently given by the expression

$$u_i^C(r) = \{(1 - \lambda^*)e^A \delta_{kj} + 2(\mu - \mu^*)e_{kj}^A\} \int_S U_{ik}(r, r') dS_j, \quad (4.2)$$

where $U_{ik}(r, r')$ is the i -component of the displacement at r when a unit point-force is applied at r' parallel to the x_k -axis. Because the medium is inhomogeneous U_{ik} depends on r and r' separately and not simply on $|r - r'|$ as does the corresponding quantity in (2.4). It is not possible to determine U_{ik} for an arbitrary form of S , and hence a transformation of (9.2) corresponding to the step from (2.6) to (2.7) cannot be made. However, we shall find that the formulation in terms of a layer of force is useful in deriving certain energy relations.

When the inhomogeneity has the form of an ellipsoid the solution can be found from the solution for the ellipsoidal inclusion by making use of the fact that the stress in the inclusion is uniform. For the special case of a cavity ($\lambda^* = 0$, $\mu^* = 0$) the method has already been outlined in § 1. The general ellipsoidal inhomogeneity is handled similarly. On the elastic field of a homogeneous inclusion with stress-free strain e_{ij}^T superimpose a uniform strain e_{ij}^A . Let

$$\rho_{ij}^A = \lambda e^A \delta_{ij} + 2\mu e_{ij}^A$$

be the corresponding stress. The stress in the inclusion is now

$$\rho_{ij}^{inc} = \rho_{ij}^T + \rho_{ij}^A = \rho_{ij}^C - \rho_{ij}^T + \rho_{ij}^A,$$

and the strain in the inclusion is

$$e_{ij}^{inc} = e_{ij}^C + e_{ij}^A. \quad (\text{TOTAL STRAIN})$$

On account of the term $-\rho_{ij}^T$ in (2.19) (which appears because there is no stress associated with the stress-free transformation strain e_{ij}^T) ρ_{ij}^{inc} and e_{ij}^{inc} are not related by Hooke's law for material with elastic constants λ , μ . They are, however, related by Hooke's law for a material with constants λ^* , μ^* provided these satisfy

$$\rho_{ij}^{inc} = \lambda^* e^{inc} \delta_{ij} + 2(\mu^* - \mu)e_{ij}^{inc},$$

that is, if $(1 - \lambda^*)e^c + e^A = 2(\mu - \mu^*)(e_{ij}^c + e_{ij}^A) = (\lambda e^T \delta_{ij} + 2\mu e_{ij}^T) = \rho_{ij}^T$. (4.3)

An ellipsoid made of material with these constants can be used to replace the transformed inclusion with continuity of stress and displacement across the interface provided that in its unstressed state this ellipsoid coincides in shape and size with the untransformed inclusion. This replacement does not alter the stresses inside or outside the ellipsoid; they remain the same as the stresses due to a homogeneous transformed inclusion with stress-free strain e_{ij}^T , together with the original applied stress ρ_{ij} .

The argument has been presented as if e_{ij}^A and e_{ij}^T were given and λ^*, μ^* were to be found. In the actual inhomogeneity problem λ^*, μ^* and e_{ij}^A (or ρ_{ij}^A) are given and we have to determine e_{ij}^T . To do this we express ρ_{ij} and e_{ij}^c in terms of S_{ijkl} and e_{kl}^T and substitute in (4.3). This gives

$$(1^* - \lambda)S_{mmkl}e_{kl}^T \delta_{ij} + 2(\mu^* - \mu)S_{ijkl}e_{kl}^T + \lambda e^T \delta_{ij} + 2\mu e_{ij}^T \\ = (\lambda - \lambda^*)e^A \delta_{ij} + 2(\mu - \mu^*)e_{ij}^A. \quad (4.4)$$

As in the case of (3.18) the solution for the non-diagonal e_{ij}^T is immediate,

$$e_{12}^T = \frac{\mu - \mu^*}{2(\mu^* - \mu)S_{1212}} e_{12}^A, \quad e_{23}^T = \dots,$$

while for $e_{11}^T, e_{22}^T, e_{33}^T$ we have the three simultaneous equations

$$(1^* - \lambda)e^c + 2(\mu^* - \mu)S_{ijkl}e_{kl}^T + \lambda e^T + 2\mu e_{ij}^T = (\lambda - \lambda^*)e^A e_{ij}^A + 2(\mu - \mu^*)e_{ij}^A \quad (4.5)$$

($ij = 11, 22, 33$) with the value (3.19) for e^c .

For a sphere (4.4) reduces to

$$e^T = A e^A, \quad e_{ij}^T = B e_{ij}^A \quad (4.6)$$

where

$$A = A\{x, x^*\} = \frac{x^* - x}{(x - x^*)\alpha - \kappa}, \quad (4.7)$$

$$B = B\{\mu, \mu^*\} = \frac{\mu^* - \mu}{(\mu - \mu^*)\beta - \mu} \quad (4.8)$$

with the values (3.14) for α, β .

The e_{ij}^T found in this way is the stress-free strain of an 'equivalent homogeneous inclusion' from which the elastic field can be calculated. Note that e_{ij}^T goes to zero with e_{ij}^A . Consequently the solution does correspond to a perfectly fitting inhomogeneous ellipsoid in a body which is stress-free when not acted on by external forces.

The displacement inside and outside the ellipsoid is

$$u_i = u_i^A + u_i^C. \quad (4.8)$$

The term u_i^A represents the unperturbed displacement; it is equal to $e_{ij}^A x_j$ plus an arbitrary rigid-body displacement. The term u_i^C represents the perturbation due to the presence of the ellipsoid; it is calculated from (3.1) with the e_{ij}^T of the equivalent inclusion. The stress is

$$\rho_{ij} = \rho_{ij}^A + \rho_{ij}^C$$

outside the ellipsoid and

$$\rho_{ij} = \rho_{ij}^A + \rho_{ij}^I$$

inside it. The form

$$\rho_{ij} = \rho_{ij}^A + \lambda e^C \delta_{ij} + 2\mu e_{ij}^C$$

is valid inside and outside the ellipsoid; in calculating the interior field we use the 'wrong' elastic constants λ^*, μ in place of λ^*, μ^* and this compensates for the change from ρ_{ij}^C to ρ_{ij}^I .

It is perhaps worth mentioning that the results of this section and of § 3.2 for the inhomogeneous inclusion can be generalized to the case where the matrix is isotropic but the interior of the ellipsoid is anisotropic and has elastic constants c_{ijkl} , say. For the transformation problem it is only necessary to replace (3.18) by

$$\lambda(e^C - e^T)\delta_{ij} + 2\mu(e_{ij}^C - e_{ij}^T) = c_{ijkl}(e_{ij}^C - e_{ij}^T). \quad (4.9)$$

For the inhomogeneity problem (4.4) becomes

$$\lambda(S_{mmkl}e_{kl}^T - e^T + e^A)\delta_{ij} + 2\mu(S_{ijkl}e_{kl}^T - e_{ij}^T + e_{ij}^A) \\ = c_{ijkl}^*(S_{pqkl}e_{kl}^T + e_{pq}^A). \quad (4.10)$$

By solving (4.9) or (4.10) the e_{ij}^T of the equivalent inclusion can be found.

The case of an anisotropic ellipsoid in an isotropic medium may seem rather artificial. It finds an application, however, in the theory of aggregates of anisotropic crystals (cf. § 6).

The argument leading to the expression (2.6) for the displacement u_i^c due to a transformed inclusion of any shape applies equally to an anisotropic material. To obtain an explicit expression for u_i^c analogous to (2.7) we should have to know the form of the displacement due to a point force in an anisotropic material. Unfortunately an expression explicit enough for our purpose cannot be obtained (FREDHOLM [1900], LIFSHITZ and ROSENZWEIG [1947], KRÖNER [1953b]). It is, however, possible to carry the analysis far enough to show that the stress is uniform inside a transformed anisotropic ellipsoid in an anisotropic matrix (ESHELBY [1957]).

We may give a picturesque interpretation to (4.8) by saying that the applied field 'induces' an inclusion in the inhomogeneity, with a stress-free strain e_{ij}^T proportional to the applied stress (cf. KRÖNER [1958a]).

If the inhomogeneity is in a finite body we may write the perturbing field of the inclusion in the form

$$u_i^F = u_i^c + u_i^{im}, \quad p_{ij}^F = p_{ij}^c + p_{ij}^{im},$$

as in § 2.1, with the image terms chosen so that the boundary conditions on the outer surface S_0 of the body are satisfied. They are not, of course, quite identical with the image terms for the equivalent inclusion in a homogeneous medium.

4.2. ENERGY RELATIONS

It is sometimes necessary to compare the elastic behaviour of a body containing an inhomogeneity with the behaviour of a similar body which is homogeneous. In such cases it is convenient to imagine that we have a single body and that the inhomogeneity may be introduced or removed at will.

When an inhomogeneity is introduced into a body already stressed by some external mechanism there will be a change in its elastic energy. At the same time there may be a change in the potential energy of the loading mechanism. As in § 2.2 we can define a change in total energy (or in enthalpy or Gibbs free energy). We shall calculate this quantity for an ellipsoidal inhomogeneity in a body subjected to two types of loading, rigidly imposed surface displacements and constant surface tractions.

Consider first the case where a constant displacement is imposed on the outer boundary by a perfectly rigid external mechanism, producing uniform stress and strain ρ_{ij}^A, e_{ij}^A in the absence of the inhomogeneity.

The perturbing field when the inhomogeneity is introduced must thus satisfy the condition

$$u_i^F = u_i^c + u_i^{im} = 0 \quad \text{on } S_0. \quad (4.1)$$

The elastic energy E_A in the medium when the inclusion is absent is found by integrating the (constant) energy density over the whole volume of material. Let the inhomogeneity be introduced and, as in § 4.1, let the material be held in a state of uniform strain e_{ij}^A by a layer of body force $-T_i$ (eq. (4.9)). The elastic energy is found by replacing λ, μ by λ^*, μ^* in that part of the volume integral for E_A which refers to the interior of S . Thus at this stage the elastic energy is $E_A + W_1$ where

$$W_1 = \frac{1}{2}V[(\lambda^* - \lambda)e^A \delta_{ij} + 2(\mu^* - \mu)e_{ij}^A]e_{ij}^A.$$

If the layer of force $-T_i$ is relaxed to zero each element dS of S suffers a displacement u_i^F and an amount of energy

$$W_2 = \frac{1}{2} \int_S T_i u_i^F dS$$

is withdrawn from the medium. By Gauss's theorem and (4.11) this can be put in the form

$$W_2 = \frac{1}{2}V[(\lambda - \lambda^*)e^A \delta_{ij} + 2(\mu - \mu^*)e_{ij}^A]e_{ij}^C - E_{im}$$

where

$$E_{im} = \frac{1}{2}[(\lambda - \lambda^*)e^A \delta_{ij} + 2(\mu - \mu^*)e_{ij}^A] \int_V e_{ij}^{im} dv.$$

The increase in the elastic energy when the inhomogeneity is introduced is thus

$$\Delta E_{el} = W_1 - W_2.$$

This is also the increase in the total energy, ΔE_{tot} , since the rigid loading mechanism does no work. With the help of (4.3) we have

$$W_1 - W_2 = -\frac{1}{2}V\rho_{ij}^A e_{ij}^T + E_{im}.$$

We consider next the case where the body is loaded by constant surface tractions $\rho_{ij}^A n_j$. The perturbing field due to the introduction of the inhomogeneity must now satisfy

$$\rho_{ij}^F n_j = (\rho_{ij}^c + \rho_{ij}^{im})n_j = 0 \quad \text{on } S_0.$$

The elastic energy when the inhomogeneity is introduced and the layer of force $-T_i$ is present is again given by $E_A + W_1$, and W_2 still repre-

sents the energy removed on relaxing the layer of force. But in addition there will now be a movement of the outer boundary in which the surface tractions do an amount of work

$$W_3 = \int_{S_0} p_{ij}^A u_i^F dS_j$$

on the body. Thus

$$\Delta E_{el} = W_1 - W_2 + W_3.$$

If we are interested in the change in total energy we do not need to know the value of W_3 ; evidently W_3 also represents the decrease in the energy of the loading mechanism and so

$$\Delta E_{tot} = \Delta E_{el} - W_3 = W_1 - W_2.$$

It is in fact not difficult to establish the relation

$$W_3 = 2W_2 - W_1.$$

We make use of the same device as in (2.39) and write

$$W_3 = \int_{S_0} (p_{ij}^A u_i^F - p_{ij}^F u_i^A) dS_j, \quad (4.12)$$

change the surface of integration from S_0 to S and convert to a volume integral over the inhomogeneity. The relation (4.12) then follows if we note that in the inhomogeneity

$$p_{ij}^A = \lambda e^{Im} \delta_{ij} + 2\mu^* e_{ij}^A$$

by definition, but that, contrary to the general rule laid down in § 1

$$p_{ij}^{Im} = \lambda^* e^{Im} \delta_{ij} + 2\mu^* e_{ij}^{Im}$$

since p_{ij}^{Im} , e_{ij}^{Im} represent the actual image field in the inhomogeneity.

If we introduce the notation

$$E_{int} = -\frac{1}{2} V p_{ij}^A e_{ij}^T + E_{im}$$

these results may be summarised as follows:

(i) for a rigidly-held boundary

$$\Delta E_{tot} = E_{int}, \quad \Delta E_{el} = \Delta E_{tot}, \quad (4.14)$$

(ii) for a boundary subject to constant loads

$$\Delta E_{tot} = E_{int}, \quad \Delta E_{el} = -\Delta E_{tot}. \quad (4.15)$$

The difference in sign between (4.14) and (4.15) may be illustrated by considering the case where the inhomogeneity takes the form of a crack, that is, a narrow region in which the elastic constants are zero. Introduction of the crack into a body strained by rigidly-imposed

surface displacements will obviously reduce the elastic energy. On the other hand the presence of the crack makes the body more 'flexible'. Consequently given applied loads will deform it more, do more work on it and so increase its elastic energy.

The first term in (4.13) is precisely half the interaction energy (2.43) for the equivalent homogeneous inclusion. The image term E_{im} will have a different value in (i) and (ii) since e_{ij}^{Im} is derived from different boundary conditions. If the inhomogeneity is far from the boundaries it will be small compared with $-\frac{1}{2} V p_{ij}^A e_{ij}^T$, and we may say that provided the initial stress p_{ij}^A is the same in each case, the total energy changes in cases (i) and (ii) are the same, but that the changes in elastic energy are equal and opposite. It is possible to extend the analysis to the mixed case where a constant displacement is imposed on part of S_0 and a constant load on the remainder. As one might expect, ΔE_{tot} is still given by (4.14) (with the appropriate e_{ij}^{Im} inserted in E_{im}), but no general statement can be made about the relation between ΔE_{el} and ΔE_{tot} . This is no drawback since it is ΔE_{tot} which is the physically important quantity.

§ 5. Relation to the Theory of Dislocations

There is a close relation between the inclusion problem and the theory of dislocations, more particularly with the general type of dislocation introduced by SOMIGLIANA [1914, 1915]. It will be convenient, however, to begin with the more familiar type of dislocation which plays a part in the physical theory of plasticity (NABARRO [1952], SEEGER [1955]). For want of a better name we may refer to these as VOLTERRA [1907] dislocations, though in fact they only correspond to the first three of his six classes. For our purposes a Volterra dislocation may be defined as a state of self-stress in which the displacement has discontinuity b_i , the Burgers vector, across a surface bounded by an open or closed curve, the dislocation line. If the dislocation line forms a closed loop and lies in a plane, the dislocation is characterised by giving the form and orientation of the loop and the value of the Burgers vector.

The displacement at large distances from a Volterra dislocation loop of area A , Burgers vector b_i and normal n_i situated at the origin is

$$u_i = \frac{Ab_i n_i g_{ijk}}{3\pi(1 - \sigma_j)r^2} \quad (5.1)$$

with the notation of (2.23) (BURGERS [1939], NABARRO [1951]). This can be put in the alternative form

$$u_t = \frac{Ab_j n_k}{8\pi(1-\sigma)} \partial_{jk} r \quad (5.2)$$

where

$$\partial_{jk} = \frac{\partial^3}{\partial x_i \partial x_j \partial x_k} - \left\{ (\delta_{jk} \frac{\partial}{\partial x_i} + (1-\sigma)\delta_{ij} \frac{\partial}{\partial x_k} + (1-\sigma)\delta_{ik} \frac{\partial}{\partial x_j}) \nabla^2 \right\} \quad (5.3)$$

The limiting process

$$A \rightarrow 0, \quad b_j \rightarrow \infty, \quad Ab_j \rightarrow s_j \quad (5.4)$$

gives an elementary infinitesimal dislocation loop of strength s_j and normal n_k . The displacement at any distance from it is given by (5.1) or (5.2) if we identify Ab_j with s_j .

Comparison of (5.1) with (2.23) shows that the remote field of a (finite) dislocation loop is the same as the remote field of an inclusion of arbitrary shape whose volume and stress-free strain satisfy

$$\nabla e_{ij}^T = \frac{1}{2}(b_{ij} n_j + b_j n_i). \quad (5.5)$$

To an elementary dislocation loop there corresponds an elementary inclusion. It is natural to imagine it as a platelet coinciding with the loop. Its thickness and stress-free strain may be so chosen that in the limit (5.5) agrees with (5.4). A finite plane dislocation loop may be formed by spreading elementary loops over a plane surface. When each loop is replaced by an equivalent elementary inclusion we obtain an inclusion in the form of a thin disc, and the elastic field which it produces is the same as that of the dislocation. It is, in fact, possible to build up directly an expression for the elastic field of a finite plane dislocation loop starting from an inclusion in the form of a disc of small (and ultimately vanishing) thickness. The quantity B_t (2.7) is then the potential of a plane lamina, and the f_{ij} (2.22) are the potentials of certain double layers. The discontinuities in potential and attraction on crossing them can be found by elementary electrostatics, and the e_{ij}^T may then be so adjusted that there is a discontinuity b_{ij} in the displacement u_i^c between points on opposite faces of the disc. This procedure, however, does not completely determine the e_{ij}^T ; it is necessary to impose further conditions to ensure, for example, that there is not a line of dilatation running round the rim of the disc.

We turn now to the more fruitful connection between inclusions and

the general Somigliana dislocation. A Somigliana dislocation can be constructed as follows. Make a cut over a surface S (open or closed) and give the two faces of the cut an arbitrary small relative displacement, removing material where there would be interpenetration. Fill in any gaps and weld the material together again. Let $b_t(r)$ be the relative displacement at the point r on the cut. There are several ways of calculating the resulting state of self-stress. The simplest is to regard the Somigliana dislocation as equivalent to a distribution of elementary dislocations over S , the strength of the loop associated with the element of area $dS(r)$ being $b_t(r)dS$. By (5.2) the displacement is

$$u_t^D = \frac{1}{8\pi(1-\sigma)} \partial_{jk} I_{jk} \quad \text{looks like opposite of North & Lathe (5.6)}$$

where

$$I_{jk} = \int_S b_t(r') |r - r'| dS_k.$$

This can be shown to be equivalent to the expression given by SOMIGLIANA [1915].

To see the connection with the inclusion problem suppose that a cut is made over the surface S of the Somigliana dislocation. The faces spring apart leaving a gap $b_t(r)$ (in some places the 'gap' may in fact be an interpenetration of material). As we saw in § 2.1 the inclusion behaves in just this fashion. If we make a cut over the interface a gap appears. (To (5.7) we might possibly add a rigid body displacement; for the moment we ignore it.) Consequently the inclusion is equivalent to a Somigliana dislocation in which S is a closed surface coinciding with the interface and having the discontinuity (5.7). We should expect the displacement calculated from (5.7) and (5.6) to be not u_i^c but rather the quantity which, in anticipation, we called u_i^D in § 2.1. That is, it should coincide with u_i^c outside the inclusion and differ from u_i^c by the displacement $e_{ij}^T x_i$ inside the inclusion. This may be verified analytically. With (5.7) we have, using Gauss's theorem,

$$I_{jk} = e_{jk}^T \int_S x'_i |r - r'| dS_k = e_{jk}^T \int_V \frac{\partial}{\partial x'_k} \{x'_i |r - r'|\} dv$$

$$= e_{jk}^T \psi + e_{jk}^T X_{lk}$$

where ψ is the biharmonic potential (2.9) and

$$X_{lk} = \int_V x'_i \frac{\partial}{\partial x'_k} |r - r'| dv.$$

When inserted in (5.6) the term in ψ gives u_t^c in the form (2.8). It can be verified that

$$\partial_{ijk} X_{lk} = 2(1 - \sigma) \delta_{ij} \nabla^2 \int_V \frac{x_i}{|\mathbf{r} - \mathbf{r}'|} dv. \quad (5.8)$$

The integral is the harmonic potential of matter of density x_l filling S , and so its Laplacian is $-4\pi x_k$ inside S and zero outside. Consequently

$$\partial_{ijk} X_{lk} = -8\pi(1 - \sigma) \delta_{ij} x_l \quad \text{inside } S \quad (5.9)$$

and so

$$\begin{aligned} u_t^D &= u_t^c - e_{il}^T x_l && \text{inside } S \\ &= u_t^c && \text{outside } S \end{aligned} \quad (5.10)$$

as expected. If we had included a rigid-body rotation $\omega_{jk}^T x_l$ in (5.7) I_{jk} would have contained the additional terms $\omega_{jk}^T \psi + \omega_{jk}^T X_{lk}$. The first contributes nothing to u_t^D , since $\partial_{ijk} \psi$ is symmetric in jk . According to (5.8) the second term leaves u_t^D unchanged outside S . Inside S it gives a rigid-body rotation which in a structureless medium produces no observable effect.

It is clear that treatment in terms of a Somigliana dislocation offers no computational advantages when applied to an inclusion which has undergone a homogeneous stress-free strain. However, we may use the formula (5.6) to extend our results to the case where the inclusion suffers any small change of shape, not necessarily a homogeneous deformation.

Suppose that the inclusion undergoes a permanent change of form which, if the matrix were absent, would not be associated with stress. To specify this 'stress-free change of form' it is only necessary to give the displacement of each point of its surface,

$$u_t^T = u_t^T(\mathbf{r}), \quad \mathbf{r} \text{ on } S. \quad (5.11)$$

Alternatively, to maintain the analogy with § 2.1, we could introduce a variable stress-free strain $e_{ij}^T(\mathbf{r})$ defined throughout the interior of the inclusion and giving a displacement agreeing with (5.11) on S . (We recall (SOKOLOVSKOFF [1946]) that specification of a (compatible) strain throughout a region fixes the displacement in it to within a rigid body displacement.) Since e_{ij}^T would be largely arbitrary it seems better to work with u_t^T in the general case.

If we try to put the transformed inclusion into the matrix there will

be a misfit u_t^T at each point of S . When this misfit is removed by suitably straining the material and welding together corresponding points on either side of S we are left with a Somigliana dislocation for which $b_t = u_t^T$. Thus to find the elastic field when the inclusion is constrained by the matrix we have only to put $b_j(r') = u_j^T(r')$ in (5.6). The resulting u_t^P is the displacement measured from an initial state in which the matrix is unstressed and the inclusion is transformed and stress-free. Consequently the stress is given by

$$\rho_{ij}^D = \lambda u_{m,m}^D \delta_{ij} + \mu(u_{i,j}^D + u_{j,i}^D)$$

both inside and outside the inclusion.

This problem can also be solved by the method of § 2.1. Let

$$u_t^T = u_t^T(\mathbf{r}), \quad \mathbf{r} \text{ inside or on } S,$$

be any convenient continuous displacement which coincides with (5.11) on S , and put

$$\rho_{ij}^T = \lambda u_{m,m}^T \delta_{ij} + \mu(u_{i,j}^T + u_{j,i}^T).$$

Remove the transformed inclusion from the matrix and apply surface tractions $-\rho_{jk}^T n_j$ and a distribution of body force of amount $+\rho_{jk,k}^T$ per unit volume. This produces a displacement $-u_i^T$ and so restores the inclusion to its untransformed shape. Cement it back in the matrix and relax the unwanted forces. This gives the displacement (cf. (2.6))

$$u_t^C = \int_S dS_k \rho_{jk}^T(r') U_{ij} - \int_V dv \frac{\partial \rho_{jk}}{\partial x_k}(r') U_{ij}.$$

It can be shown by the same sort of analysis as led to (5.10) that $u_t^C = u_t^D$ in the matrix and that $u_t^C = u_t^P + u_t^T$ in the inclusion. Since u_t^T is by definition a stress-free displacement the stresses calculated from u_t^C and u_t^D are identical.

The solution for the elastic field due to an inclusion which has suffered a non-uniform transformation does not seem to have any obvious applications. However, for an ellipsoidal inclusion the solution has a property which enables it to be used to find how an ellipsoidal cavity or inhomogeneity perturbs a non-uniform stress-field of fairly general type. This property generalizes the result that a uniform transformation strain leads to a uniform strain in the constrained ellipsoid. It may be stated thus: if the stress-free transformation displacement is a polynomial in x_1, x_2, x_3 of degree N , then the dis-

placement is also a polynomial of degree N inside the constrained ellipsoid. We shall only outline its derivation and application. The details of the calculation can easily be filled in.

The analysis involves a number of polynomials in x_i or x'_i with constant coefficients. We shall denote them by script capitals and indicate only the argument and degree. Thus, for example, $\mathcal{P}(M, x)$ stands for a polynomial in x_1, x_2, x_3 whose highest term is of degree M . Similarly $\mathcal{G}_{jk}(M, x), i, j = 1, 2, 3$ denotes a set of twenty-seven polynomials in x'_1, x'_2, x'_3 , all of the same degree, and so forth. We shall also use R as an abbreviation for $|\mathbf{r} - \mathbf{r}'|$.

Let the ellipsoidal inclusion undergo the transformation specified by

$$u_i^T(\mathbf{r}) = \mathcal{T}_i(N, x) \quad (\mathbf{r} \text{ on } S).$$

If we put $b_i = u_i^T$ in (5.6) and use Gauss's theorem we find that I_{jk} has the form

$$I_{jk} = \int_V \mathcal{E}_{jk}(N-1, x') R \, dv + \int_V \mathcal{T}_i(N, x') \frac{\partial R}{\partial x'_k} \, dv. \quad (5.12)$$

In the first integrand introduce the factor

$$1 = \frac{(x_m - x'_m)(x_m - x'_m)}{|\mathbf{r} - \mathbf{r}'|^2}$$

and in the second write $\partial R / \partial x'_k$ as $(x'_k - x_k)/R$. After a little rearrangement (5.12) takes the form

$$\begin{aligned} I_{jk} = & x_m x_m \int_V \mathcal{F}_{jk}(N-1, x') \frac{dv}{R} + x_p \int_V \mathcal{G}_{pj}(N, x) \frac{dv}{R} \\ & + \int_V \mathcal{H}_{jk}(N+1, x) \frac{dv}{R}. \end{aligned} \quad (5.13)$$

The integrals are the harmonic potentials of solid ellipsoids whose densities are polynomial functions of the co-ordinates. FERRERS [1877] and DYSON [1891] have discussed the potentials of inhomogeneous ellipsoids of this type. Their results show that when the density is of the form $\mathcal{P}(M, x)$ the potential is of the form $\mathcal{Q}(M+2, x)$ inside the ellipsoid. The coefficients of the polynomial \mathcal{Q} can be calculated from the coefficients of \mathcal{P} and the three quantities I_a, I_b, I_c (eq. (3.2)). Outside the ellipsoid the potential is a similar polynomial in x_1, x_2, x_3 but with coefficients which are themselves functions of position. (Compare the relation between (3.2) and (3.7).) These variable coefficients can be expressed in terms of the coefficients of \mathcal{P} and the quantities

A, B, C, F, E of (3.8), (3.10). Consequently, by (5.13) and (5.6) the constrained displacement is a polynomial of the form

$$u_i^D(\mathbf{r}) = \mathcal{D}_i(N, x)$$

inside the ellipsoid. Thus the constrained and unconstrained displacements of the inclusion are similar polynomials with calculable relations between their coefficients.

The solution of the cavity problem follows the lines of § 4.1. Superimpose a displacement

$$u_i^A(\mathbf{r}) = -\mathcal{D}_i(N, x) \quad (5.14)$$

everywhere. The total displacement $u_i^A + u_i^D$ is zero inside the ellipsoid. Hence the inclusion is unstressed, and it can be removed without disturbing the matrix. The coefficients in \mathcal{T}_i can be chosen so as to make (5.14) any required polynomial of degree N . From these coefficients the coefficients of $\mathcal{F}, \mathcal{G}, \mathcal{H}$ (eq. (5.13)) can be found. The field $u_i^A + u_i^D$ in the matrix then follows from (5.13) and Ferrers' and Dyson's results. The extension to the ellipsoidal inhomogeneity follows as in § 4.1.

The calculations are not too unwieldy for small N . With $N = 2$ the displacement

$$u_i^A = \alpha_{ij}^A x_j + \beta_{ijk}^A x_j x_k$$

covers the case of an applied stress which is a combination of simple torsion, bending and flexure. Solutions, based on other methods, have been given for a number of such problems involving spheroids and spheres (NEUBER [1958], SEN [1933], DAS [1953]).

§ 6. Applications

The application of the results reviewed here to the actual calculation of stress in and about inclusions and inhomogeneities calls for no special comment. The engineering application of the theory of cavities to the calculation of stress concentrations has been thoroughly treated by NEUBER [1958]. In this section we indicate some of the applications to physical problems.

Expressions for the elastic energy and interaction energy of an inclusion find a use in the theory of martensitic transformations. (For a general review see KAUFMAN and COHEN [1958].) Suppose that a metal can exhibit two crystal structures, γ and α , the former stable

at high temperatures and the latter at low temperatures. Ideally we might expect that on cooling the whole of a single crystal of the metal would undergo a uniform stress-free strain in the sense of § 2, the e_{ij}^T being the deformation which carries the γ -lattice into the α -lattice. In a martensitic transformation, however, the low-temperature phase first appears in the form of inclusions of α embedded in the γ matrix. Thus, in considering the thermodynamics of the transformation, one must take into account the elastic energy of the misfitting inclusions of α and also, if there is an externally applied stress, their interaction energy with the latter.

The Gibbs free energy change associated with the formation of a martensitic inclusion may be written in the form

$$(6.1) \quad \Delta G = \Delta G_{\text{chem}} + \Delta G_{\text{surf}} + E_{\text{inc}} + E_{\text{int}}.$$

ΔG_{chem} is the 'chemical' contribution, the free energy change which would occur if the inclusion underwent its stress-free strain in the absence of the matrix. The theory of elastic inclusions can tell us nothing about it. ΔG_{surf} is a contribution due to the interface between matrix and inclusion. It may be estimated by using the theory of dislocations. E_{inc} is the elastic energy associated with the inclusion, and E_{int} is the interaction energy with any externally applied stress which may be present. According to the discussion in § 2.2 the terms E_{inc} and E_{int} taken together make up the elastic contribution to the change of Gibbs free energy.

It will usually be accurate enough to identify E_{int} with E_{∞} , the value in an infinite medium, and to ignore the difference between the elastic constants of matrix and inclusion. If the inclusion can be considered to be some form of ellipsoid E_{∞} can be calculated from (3.3) and (2.32) when the transformation strain e_{ij}^T is known. Suppose, for example, that the inclusion takes the form of a plate in the x_1x_2 -plane and that the transformation is made up of a shear parallel to this plane through an angle s , a uniform dilatation Δ and an extension ξ perpendicular to the plane of the plate. Then

$$e_{13}^T = e_{31}^T = \frac{1}{2}s, \quad e_{11}^T = e_{22}^T = \frac{1}{2}\Delta, \quad e_{33}^T = \xi + \frac{1}{2}\Delta \quad (6.2)$$

and the remaining components are zero. If the plate has the form of a thin oblate spheroid with $c \ll a = b$ we have

$$\begin{aligned} \frac{1-\sigma}{\mu V} E_{\infty} &= \frac{1}{8} \pi (2-\sigma) \frac{c}{a} s^2 + \frac{2}{9} (1+\sigma) \Delta^2 \\ &\quad + \frac{1}{4} \pi \frac{c}{a} \xi^2 + \frac{1}{3} \pi (1+\sigma) \frac{c}{a} \xi \Delta \end{aligned} \quad (6.3)$$

(CHRISTIAN [1958]). There is also a fairly simple expression for E_{∞} when the inclusion is a flat ellipsoid with $a > b \gg c$ (ESHELBY [1957]). The expression (6.3) has been used by CHRISTIAN [1958, 1959] and KAUFMAN [1959] to discuss the nucleation of martensite.

The interaction energy may be found from (2.43). In an actual experiment the applied stress will usually be a uniaxial tension; $\rho \hat{\delta}$ then takes the form $\tau n_i n_j$, where τ is the magnitude of the tension and n_i is its direction. If the transformation strain is given by (6.2) we have

$$E_{\text{int}} = -\tau C V \quad (6.4)$$

with

$$C = s \cos \beta \sin \beta + \xi \cos^2 \beta + \frac{1}{2} \Delta$$

where β is the angle between the x_3 -axis and the direction of the tension, and the latter is assumed to lie in the x_1x_3 -plane. A result equivalent to (6.4) was first obtained (with $\Delta = 0$) by PATEL and COHEN [1953] (cf. also MACHLIN and WEINIG [1953]). FISHER and TURNBULL [1953] have also discussed the effect of an applied stress on the free energy change associated with martensite formation. In effect, they take the interaction energy to be the cross-term in the elastic energy between the applied field and the field due to the inclusion. As we have seen (§ 2.2), this quantity should be zero. Since they only integrate the energy density over the immediate neighbourhood of the inclusion they obtain a finite result which, however, is not quite correct numerically and which, in addition, has the wrong sign. To correct this they have to make an arbitrary reversal of sign in the relation between applied stress and applied strain.

When a crystal having the high-temperature γ -structure is cooled, ΔG decreases. According to elementary thermodynamics a martensitic inclusion of the α -phase can form when a temperature is reached at which ΔG is zero. In fact a nucleation barrier has to be overcome and the transformation will occur when ΔG reaches some finite negative value, say ΔG_{nuc} . Let T_s be the temperature at which $\Delta G = \Delta G_{\text{nuc}}$ in the absence of external stress, i.e. with $E_{\text{int}} = 0$ in (6.1). When there is an applied stress the term E_{int} will alter the temperature at which $\Delta G_{\text{nuc}} = \Delta G$ to, say, $T_s + \Delta T_s$. If ΔG_{nuc} is independent of temperature and ΔG_{nuc} is the only strongly temperature-dependent term in (6.1) we must have

$$\frac{d\Delta G_{\text{nuc}}}{dT} \Delta T_s + E_{\text{int}} = 0,$$

and so from (6.4) we obtain the expression

$$\frac{dT_s}{d\bar{T}} = C / \bar{V} \frac{d\Delta G_{\text{chem}}}{d\bar{T}} \quad (6.5)$$

for the rate of change of transformation temperature with applied tensile stress. C can be found from crystallographic measurements, and the rate of change of chemical free energy with temperature may be obtained from thermodynamic data. Patel and Cohen found excellent agreement between theory and experiment for certain iron-nickel and iron-nickel-carbon alloys. They also showed how (6.5) should be modified when ΔG_{chem} varies with temperature.

For martensitic inclusions the transformation strain is essentially a pure shear. For particles of precipitate formed by diffusion it is essentially a dilatation. For an ellipsoidal precipitate the strain energy can be found from (3.21) with $e_{ij}^* = \frac{1}{3}\epsilon^{ij}\delta_{ij}$. ROBINSON [1951] has given a detailed treatment. In these calculations it is assumed that the inclusion is coherent with the matrix. NABARRO [1940] and KRÖNER [1954] have treated the problem of finding the minimum strain energy due to an inclusion which has broken away from the matrix.

The displacement need not be continuous across the interface, but the matrix and inclusion are supposed to be in contact everywhere. The volume misfit is prescribed, but the stress-free shape of the inclusion has to be determined so as to minimize the energy. According to KRÖNER [1953a, 1954] the energy is a minimum for the state in which the stress in the inclusion is purely hydrostatic. (Matrix and inclusion may be made of different anisotropic materials.)

The theory of cracks, i.e. empty cavities one of whose dimensions is evanescent, has played a part in Griffiths' treatment of rupture (see, for example, SNEDDON [1951]) and, more recently, in the theory of the brittle and ductile fracture of metals. The crack is supposed to have associated with it a surface energy γA , where A is its surface area and γ is a constant representing true surface energy or an effective surface energy associated with plastic deformation. According to the discussion in § 4.2 the total energy of the system made up of the body containing the crack and the external loading mechanism is

$$E_{\text{int}} + \gamma A + \text{const.}$$

By studying the way in which E_{int} and A vary when the form of the crack is altered slightly one can find whether it is energetically favourable for the crack to spread or contract.

We may find the necessary properties of an elliptical crack from the results of §§ 4.1, 4.2 by putting $\lambda^* = \mu^* = 0$ and letting the c -axis tend to zero. The equivalent stress-free strain e_{ij}^* approaches infinity for a fixed applied stress ϕ_{ij}^A , but the products $(\frac{\partial \lambda^*}{\partial x_i})_j$ (and hence also the potentials φ, ψ) remain finite.

There exist a number of calculations for the interaction energy of a circular or two-dimensional crack (INGLIS [1913], STARR [1928], SACK [1946], SEGEDIN [1951]). They may be deduced from the following results for the elliptical crack

$$\frac{x_1^2}{a^2} + \frac{x_2^2}{b^2} = 1.$$

- (i) The applied stress is a pure tension $\hat{\rho}_{33}^A$ normal to the plane of the crack (tensions $\phi_{ii}^A, \hat{\rho}_{22}^A$ evidently have no effect). Then

$$E_{\text{int}} = - \frac{2\pi(1-\sigma)ab^2}{3E(k)} \frac{(\hat{\rho}_{33}^A)^2}{\mu} \quad (6.6)$$

This is easily deduced from the results of GREEN and SNEDDON [1949].

- (ii) The applied stress is a pure shear $\hat{\rho}_{13}^A$ in the plane of the crack. Then (ESHELBY [1957])

$$E_{\text{int}} = - \frac{2\pi ab^2}{3\eta} \frac{(\hat{\rho}_{13}^A)^2}{\mu}$$

where

$$\eta = E(k) + \frac{\sigma}{1-\sigma} \frac{K(k) - E(k)}{k^2} \left(\frac{b}{a} \right)^2 \quad (6.7)$$

In (6.6) and (6.7) E, K are complete elliptic integrals of modulus $k = (1 - b^2/a^2)^{\frac{1}{2}}$. Eq. (6.7) is valid for both $a > b$ and $a < b$. In the latter case the elliptic integrals may be reduced to real form with the help of the relation

$$K(k_1) = (1 - k^2)^{\frac{1}{2}} K(k), \quad E(k) = E(k)/(1 - k^2)^{\frac{1}{2}},$$

where $k_1 = ik/(1 - k^2)^{\frac{1}{2}}$.

(Compare the remark following (3.6).) For the circular crack $a = b$, $E = \frac{1}{2}\pi, \eta = \pi(2 - \sigma)/4(1 - \sigma)$. Stroh [1958] has calculated the interaction energy for two-dimensional cracks in anisotropic materials. The concept of the interaction energy of an inhomogeneity with an applied stress simplifies the calculation of the bulk elastic constants of elastically inhomogeneous aggregates. Suppose, for example, that we wish to calculate the effective elastic constants of a material of bulk modulus κ and shear modulus μ containing a dispersion of

spherical inhomogeneities with elastic constants \varkappa^* , μ^* (ESHELBY [1957]). In the absence of the inhomogeneities unit volume of the material has elastic energy

$$E_0 = \frac{1}{2} \left(\frac{1}{9\kappa} \rho^A \rho^A + \frac{1}{2\mu} \rho_{ij}^A \rho_{ij}^A \right) \quad (6.8)$$

if it is subjected to the uniform stress

$$\rho_{ij}^A = \frac{1}{3} \rho^A \delta_{ij} + \rho_{ij}^A$$

(for the notation see § 1). If the spheres are introduced the elastic energy becomes

$$E = E_0 - \Sigma E_{\text{int}} \quad (6.9)$$

by (4.15), where Σ denotes summation over all the spheres. If we ignore the image term in (4.13) and also the interaction between the spheres (so that we limit ourselves to a dilute dispersion) (6.9) may be written as

$$E = \frac{1}{2} \left[\frac{1}{9\kappa} (1 + Av) \rho^A \rho^A + \frac{1}{2\mu} (1 + Bv) \rho_{ij}^A \rho_{ij}^A \right]$$

where v is the fraction of the volume of material occupied by the inhomogeneous spheres and A, B are given by (4.7). Comparing with (6.8) we see that the effective elastic constants are

$$\varkappa_{\text{eff}} = \varkappa/(1 + Av), \quad \mu_{\text{eff}} = \mu/(1 + Bv)$$

or, since the analysis is only valid for $v \ll 1$,

$$\varkappa_{\text{eff}} = \varkappa(1 - Av), \quad \mu_{\text{eff}} = \mu(1 - Bv). \quad (6.10)$$

These expressions have been obtained in another manner by HASHIN (REINER [1958]).

Much attention has been given to a related problem: namely, to determine the macroscopic elastic constants, \bar{c}_{ijkl}^0 , \bar{s}_{ijkl}^0 , say, of a polycrystalline aggregate whose actual constants c_{ijkl} , s_{ijkl} vary from grain to grain, not because the material is inhomogeneous, but because the crystal orientation varies. The estimates $\bar{c}_{ijkl}^0 = \bar{c}_{ijkl}$ and $\bar{s}_{ijkl}^0 = \bar{s}_{ijkl}$ are associated with the names of Voigt and Reuss. Here the bar denotes an average over all crystal orientations, weighted if necessary according to the relative frequency of each orientation. HILL [1952] has shown that

$$\bar{s}_{ijkl} \leq s_{ijkl}^0 \leq \bar{s}_{ijkl}$$

$$\bar{c}_{ijkl} \leq c_{ijkl}^0 \leq \bar{c}_{ijkl}$$

where \bar{c}_{ijkl} and \bar{s}_{ijkl} are, respectively, the tensors inverse to \bar{s}_{ijkl} and \bar{c}_{ijkl} . For an aggregate of cubic crystals the upper and lower estimates of the bulk modulus coincide and we have exactly

$$\varkappa = \frac{1}{3}(c_{11} + 2c_{12}). \quad (6.11)$$

Hill's relation seems to mark the limit of what can be proved precisely. To go further it is necessary to make assumptions which may be physically plausible, but cannot be proved conclusively. Most of the attempts in this direction are rather outside the scope of this article. HERSHER'S [1954] and KÖRNER'S [1958b] calculations, however, make use of the concept of an anisotropic inhomogeneity in an isotropic matrix. We shall not give their physical arguments, but instead present a simplified treatment which involves essentially the same mathematics.

We confine ourselves to an aggregate of cubic crystals which is macroscopically isotropic. Let the effective elastic constants be \varkappa , μ . Imagine that the material of each grain is replaced by isotropic material with constants \varkappa , μ , but that the boundaries between grains can still be distinguished. We thus obtain, trivially, an 'equivalent isotropic aggregate' with the same overall elastic constants as the polycrystalline aggregate. It seems clear that this equivalent aggregate must pass the following test: if any representative sample of its grains have their original anisotropic constants restored the bulk elastic constants are unaltered. In applying the test we may take the collection of re-transformed grains to be so far apart that the effect of the perturbing field of one grain on another may be neglected. Of course the isotropic aggregate must also pass the more severe test that its bulk elastic constants are unchanged when every grain is restored to its original anisotropic state. However, we may hope that if the weaker test gives definite values for \varkappa and μ , then these will be the same as those which would result from applying the severer test.

We shall suppose that the set of test grains have their crystal axes oriented at random and that they may be treated as spheres. This second assumption is hard to justify rigorously, but without it not much progress can be made. In the presence of a uniform applied strain e_{ij}^A the e_{ij} of the equivalent inclusion for one of the grains can be written in the form

$$e_{ij}^T = D_{ijkl} e_{kl}^A;$$

the D_{ijkl} depend on the orientation of the grain. When the anisotropic

test grains are introduced into the equivalent isotropic material its elastic constants are changed by an amount proportional to the sum of their interaction energies with the applied field, by the same argument as led to (6.9). Thus the test requires that

$$\sum (p_{ij}^A e_{kl}^A D_{ijkl}) = 0 \quad (6.12)$$

where Σ implies summation over all the re-transformed grains. If we choose fixed coordinate axes, $p_{ij}^A e_{kl}$ is the same for each term of the summation but D_{ijkl} varies from term to term. It is therefore more convenient to choose for each term axes parallel to the crystal axes of the grain in question. Then D_{ijkl} is the same for each term, but $p_{ij}^A e_{kl}^A$ is not. The calculation of the sum $\sum p_{ij}^A e_{kl}^A$ could be carried out as follows. Assign fixed values to the principal strains of e_{ij}^A , form $p_{ij}^A e_{kl}^A$, average the result over all orientations of the principal axes of the strain and multiply by the number of terms. The resulting expression will have the form of the most general isotropic tensor which has the symmetry of the suffixes in $p_{ij}^A e_{kl}^A = p_{kl}^A e_{ij}^A$. Consequently it must have the same form as the elastic constant tensor c_{ijkl} of an isotropic medium, that is

$$\sum p_{ij}^A e_{kl}^A = c_{11} \delta_{ij} \delta_{kl} + c_2 (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk}) \quad (6.13)$$

with arbitrary c_1 , c_2 . Inserting this in (6.12) we get $D_{ijkl} = 0$ and $D_{ijkl} = 0$, or in view of the symmetry (spherical grains, cubic crystal referred to its principal axes)

$$D_{1111} + 2D_{1122} = 0 \quad (6.13)$$

and

$$D_{1111} + 2D_{1212} = 0. \quad (6.14)$$

The values of the D_{ijkl} are easily calculated from (4.6) and (4.7) if we bear in mind that when extended along its axes a cubic crystal behaves like an isotropic material with $\alpha^* = \frac{1}{3}(c_{11} + 2c_{12})$, $\mu^* = \frac{1}{3}(c_{11} - c_{22})$ and that for a shear of the type e_{12} it behaves like an isotropic material with $\mu^* = c_{44}$. Eq. (6.14) gives $4\{\alpha, \frac{1}{3}(c_{11} + 2c_{12})\} = 0$ or $\alpha = \frac{1}{3}(c_{11} + 2c_{12})$ in agreement with the already known result (6.11). The expression for the shear modulus is more interesting. Eq. (6.14) gives

$$\frac{2}{3}B\{\mu, \frac{1}{3}(c_{11} - c_{12})\} - B[\mu, c_{44}] = 0.$$

On inserting the value of β (eq. (3.14)) and clearing of fractions we are left with a cubic equation:

$$16\mu^3 + 2(5c_{11} + 4c_{12})\mu^2 - 2c_{44}(7c_{11} - 4c_{12})\mu$$

$$- c_{44}(c_{11} - c_{12})(c_{11} + 2c_{12}) = 0 \quad (6.15)$$

for determining μ . It has only one real positive root.

HERSHEY [1954] obtained a quartic equation for μ which in fact is what one obtains on multiplying (6.15) by $8\mu + 9\alpha$. Evidently this introduces no new positive root. KRÖNER [1958b] obtained both the quartic and cubic and gave an argument to show that their respective positive roots are upper and lower bounds for μ , and that since these roots coincide the value so obtained is exact apart from statistical uncertainties due to the fact that the aggregate contains only a finite number of grains.

GOODIER [1936] has emphasized that there is a useful analogy between problems in the slow motion of viscous liquids and elastic problems for incompressible solids. In general special care has to be taken in extrapolating elastic solutions to the incompressible case $\sigma = \frac{1}{2}$, since we have, in effect, to make the simultaneous transition $\omega_{k,k} \rightarrow 0$, $\lambda \rightarrow \infty$. However, in the case of the solution (2.8) to the general inclusion problem there is no trouble. With $\sigma = \frac{1}{2}$ it satisfies

$$\mu \nabla^2 u^C = 0, \quad \operatorname{div} u^C = 0$$

outside the inclusion. These are just the Stokes equations governing the velocity u^C in an incompressible fluid of viscosity μ when inertial effects may be neglected and the hydrostatic pressure, p_0 , say, is independent of position. Consequently (2.8) with $\sigma = \frac{1}{2}$ and u_t^C interpreted as a velocity represents a certain state of viscous flow. The associated stress is

$$\rho u_t^C = \mu(u_{tt}^C + u_{jj}^C) - p_0 \delta_{tt}.$$

Eq. (2.8) describes the flow about a solid which is deforming in such a way that each point r of its surface S has a velocity $u_t^C(r)$. Since we cannot prescribe $u_t^C(r)$, but only the constants e_{ij}^A , the analogy is not of much use in the general case. Eq. (3.1), however, (with $\sigma = \frac{1}{2}$) describes the viscous flow around an ellipsoid which is undergoing a change of shape specified by the constant rate-of-strain tensor e_{ij}^C , and the e_{ij}^A appropriate to prescribed e_{ij}^C can be found by solving (3.3). The solution to the problem of an ellipsoidal inhomogeneity perturbing a uniform strain e_{ij}^A has a simple viscous interpretation when the ellipsoid is perfectly rigid and incompressible ($\lambda^* \rightarrow \infty$, $\mu^* \rightarrow \infty$). Eq. (4.3) gives $e_{ij}^C + e_{ij}^A = 0$ on S or equivalently

$$u_t^C + u_t^A = 0 \text{ on } S. \quad (6.16)$$

Translated into terms of velocity this is the condition that the liquid shall adhere to the surface of the ellipsoid, and so the field $\mathbf{u}_i^c = \mathbf{u}_i^A + \mathbf{u}_i^{\text{ext}}$ gives the velocity about a solid ellipsoid immersed in the uniform flow specified by the rate-of-strain tensor e_y^A . The elastic energy density translates into half the rate of dissipation of energy per unit volume. (In the one case we are concerned with the familiar half stress times strain, in the other with stress times rate of deformation.) Consequently $2E_{\text{int}}$ (eq. (4.13)) is equal to the additional rate of dissipation of energy, E_{diss} say, when a solid is introduced into a viscous liquid at whose boundary constant velocities are maintained.

If the image term may be neglected in (4.13) the interaction energy for an inhomogeneity is half the interaction energy for the equivalent transformed inclusion, and it follows that E_{diss} is given by the right hand side of (2.42), or, in view of (6.16), by

$$\begin{aligned} E_{\text{diss}} &= - \int_S (\phi_{ij}^c + \phi_{ij}^A) u_i^c dS_j \\ &= \int_S p_n \cdot \mathbf{u}^c dS \end{aligned}$$

where p_n is the actual surface traction on the solid (with the sign convention usual in hydrodynamics) and \mathbf{u}^c is the perturbation of velocity due to the presence of the solid. This is identical with a result of BRENNER's [1958]. The expression (6.10) for the effective shear modulus of a dispersion of spheres gives the Einstein viscosity formula $\mu_{\text{eff}} = (1 + \frac{4}{3}\nu)\mu$ when we put $\mu^* = \infty$, $\sigma = \frac{1}{2}$. The difference in sign between (4.14) and (4.15) has the following interpretation in terms of viscosity. The viscosity is always increased by the introduction of solid particles. Consequently a viscometer working at constant load produces a lower rate of deformation and so less energy dissipation, while a viscometer working at constant speed has to work harder to maintain the prescribed rate of strain.

The applications considered above have all been essentially macroscopic. Elastic inclusions and inhomogeneities have also been found useful as models of lattice defects in crystals. (For a general account see, for example, FRIEDEL [1956] or ESHELBY [1956].)

Suppose that one atom of a crystal lattice is replaced by a foreign atom. The foreign atom will generally have a different size from the host atoms. As a simple elastic model we may take a spherical hole in an isotropic continuum into which a misfitting elastic sphere is inserted (BULBY [1950]). Since the elastic constants of the foreign atom (in so far as one can speak of them for a single atom) will differ from

those of the host atoms the misfitting elastic sphere will in general have to be assigned elastic constants differing from those of the matrix. It is useful to consider two limiting cases. (i) A pure inclusion: the sphere and inclusion have the same elastic constants, and the misfit gives rise to a permanent elastic field. (ii) A pure inhomogeneity; there is no misfit, but the elastic constants of sphere and matrix differ. There is no permanent elastic field, but a field can be 'induced' by an applied field. The general case, intermediate between (i) and (ii), corresponds to the inhomogeneous inclusion of § 3.2.

It is not at all obvious that such a crude model will be of any use in discussing the behaviour of lattice defects. We shall discuss some of the consequences of taking the model seriously and then try to indicate the reasons for its success.

The elastic field of a pure inclusion is given by (2.26) with the value (3.12) for φ and e^T equal to the fractional volume misfit between hole and sphere. The stress falls off as r^{-3} and its hydrostatic component is zero. The field of a pure inhomogeneity in an applied field e_{ij}^A is found by calculating the equivalent e_{ij}^T from (4.6) and inserting them in (2.8) with the values (3.12), (3.13) for φ and ψ .

It may be shown that the expressions (2.43) and (4.13) for the interaction energies of inclusions and inhomogeneities remain valid when ϕ_{ij}^A , e_{ij}^A refer to the field produced by some source of internal stress, another inclusion or a dislocation for example. Thus the interaction energy between two pure inclusions is proportional to $\phi_{ij}^A e_{ij}^T$, where ϕ_{ij}^A is the stress produced by one inclusion and e_{ij}^T is the stress-free strain associated with the other. Since $e_{ij}^T = \frac{1}{3}e^T \delta_{ij}$, the interaction energy is proportional to ρ^A . But neither defect produces a hydrostatic pressure, and so the interaction energy is zero (BIRK [1931]). Again, let ϕ_{ij}^A refer to the field of a pure inclusion and let e_{ij}^T be the equivalent stress-free strain which it induces in a pure inhomogeneity at a distance r from it. Since the ϕ_{ij}^A are proportional to r^{-3} and the e_{ij}^T are proportional to the ϕ_{ij}^A it follows that the interaction energy between a pure inclusion and a pure inhomogeneity is proportional to the inverse sixth power of the distance between them. The interaction energy between two defects each of which is represented by a misfitting and inhomogeneous sphere is also proportional to r^{-6} . Its numerical value may be found by a detailed calculation (ESHELBY [1958]).

The stress-field at a distance r from a dislocation line is proportional to r^{-1} . It follows by the same kind of argument as before that the dislocation has an interaction energy proportional to r^{-1} with a pure

inclusion but an interaction energy proportional to r^{-2} with a pure inhomogeneity. This difference allows one to distinguish experimentally between the two types of defect. Since the interaction energy is a function of position there is an effective force

$$\mathbf{F} = -\operatorname{grad} E_{\text{int}}(\mathbf{r}) \quad (6.17)$$

on the defect. If the defect is capable of diffusing through the lattice, a drift velocity

$$\mathbf{v} = DF/kT$$

is superimposed on its random motion, where D is the diffusion coefficient, k is Boltzmann's constant and T is the absolute temperature. Thus, if dislocations are introduced into a crystal containing defects, the latter will be attracted to the dislocations. The resulting depletion of defects in the bulk of the material can be detected by measuring suitable physical properties (electrical resistance, internal friction). Calculation shows that for an r^{-1} -interaction the number of defects drawn into the dislocations is proportional to $t^{\frac{1}{2}}$, where t is the time since the interaction between defects and dislocations began. If, on the other hand, there is an r^{-2} interaction the expression $t^{\frac{1}{2}}$ is replaced by $t^{\frac{1}{3}}$ (FRIEDEL [1959]).

For a defect which can be represented by an inserted sphere which is both misfitting and inhomogeneous the interaction energy with a dislocation will evidently have the form $A r^{-1} + B r^{-2}$. Even if the coefficient B is relatively large the A -term will dominate at large distances and we should expect the $t^{\frac{1}{2}}$ law to be most nearly obeyed. The $t^{\frac{1}{2}}$ law has, in fact, been verified for carbon and nitrogen diffusing in iron. We might, perhaps, regard a vacant lattice site as a pure inhomogeneity ($A = 0$). However, such calculations as have been made (e.g. TEWORDDR [1958] for copper) indicate that there is an appreciable displacement of the atoms bordering the vacant site. On the elastic model this means that there is a stress-field associated with the vacancy even in the absence of an applied field, and so $A \neq 0$.

Rather surprisingly WINTENBERGER'S [1957] measurements on vacancies in aluminium follow the $t^{\frac{1}{2}}$ law, which indicates that, in aluminium at least, lattice vacancies behave as pure inhomogeneities.

We shall now try to indicate why the simple misfitting sphere model has been relatively successful. In the first place it is reasonable to suppose that sufficiently far from a lattice defect the crystal can be treated as an elastic continuum. The displacement representing the disturbance due to the defect can be expanded in ascending inverse

powers of r (the distance from the defect) each provided with a suitable angular factor. If we treat the material as isotropic the leading term of the expansion has precisely the form (2.23) with arbitrary symmetric e_y^T . Thus, if the material is isotropic, (2.23) gives the elastic field at large distances from the most general type of point defect. In many of the common metals the elastic field of the defect will have cubic symmetry. This physical condition, combined with the artificial limitation to isotropy, requires that e_y^T shall be of the form $\frac{1}{3}\epsilon r\delta e_y$. According to (2.26) the remote field is independent of the shape of the inclusion and we may take it to be a sphere. In this way we recover the misfitting sphere model. In some cases (in particular, interstitial carbon and nitrogen in iron) the assumption that e_y^T is a pure dilatation is inadequate, but the displacement may still be taken to have the form (2.23) with suitably chosen values for the e_y^T . The interaction energy (2.42) can be re-written in the form

$$E_{\text{int}} = \int_S (\rho_y^C u_i^A - \rho_y^A u_i^C) dS_j \quad (6.18)$$

where S' is any surface drawn in the material and enclosing the inclusion. (The expressions (2.42) and (6.18) are identical because the divergence of the integrand is zero between S and S' .) The expression for the interaction energy was derived on the assumption that the infinitesimal theory could be applied everywhere. This is obviously not true near an inclusion representing an atomic defect. However, it may be shown that for (6.18) to be correct it is only necessary that the infinitesimal theory be valid in the neighbourhood of the surface S' , a much less severe requirement. The following is an outline of the argument (cf. ESHELBY [1959a]). We may form an expression for the x_3 -component of the effective force (6.17) by subtracting from (6.18) the corresponding expression with ρ_y^C, u_i^C replaced by $\rho_y^A + \rho_y^C \epsilon, u_i^C + u_{i,1}^C \epsilon$, dividing by ϵ and letting ϵ approach zero. The result is

$$F_i = \int_{S'} (\rho_y^C u_i^A - \rho_y^A u_i^C) dS_j \quad (6.19)$$

It is possible (ESHELBY [1956]) to derive a general expression for F_i which is valid for an arbitrary non-linear stress-strain relation and for finite deformation. Like (6.19) this expression takes the form of an integral over a surface surrounding the defect on which the force is to be calculated. If it is permissible to apply the infinitesimal theory on this surface the general expression reduces to (6.19). Hence (6.19), (6.18) or the simple formula (2.43) can be used whenever it is possible to draw a

surface S' enclosing the defect and no other source of internal stress and far enough from it for the strains on S' to be reasonably small. Applied to the interaction of two point defects, for example, this means that (2.42) can be used if the linear theory is approximately obeyed halfway between them.

The treatment of lattice defects by continuum methods sometimes gives useful results even in cases too extreme for the above considerations to apply. FRIEDEL [1954] has shown, for example, that in some cases it is possible to get a reasonable estimate for the energy of solution of atoms of a metal X in a metal Y by associating with each dissolved X-atom a strain energy (3.21), taking for the volumes of hole and inclusion the atomic volumes of X and Y and for (λ, μ) , (λ^*, μ^*) the ordinary elastic constants of the metals Y and X. An amusing example is provided by JACOBS' [1954] calculation of the effect of hydrostatic pressure on the frequency of the absorption band of an F-centre in an alkali halide crystal. An F-centre is an electron trapped at a negative-ion vacancy. An empirical rule of MOLLWO's [1933] states that in passing from one alkali halide crystal with the sodium chloride structure to another the product νa^2 remains constant; ν is the frequency of the maximum of the F-absorption band and a is the lattice parameter. It is reasonable to suppose that the same relation will govern the behaviour of a given alkali halide when its lattice parameter is changed by compression. We should then have the relation $-(d\nu/\nu)/(da/a) = 2$. In place of 2, Jacobs' experiments gave values between 4.4 and 3.4. This discrepancy can be reconciled if we admit that the characteristics of an F-centre are determined by the positions of the atoms bordering the vacancy. Then Mollwo's rule must be interpreted as $\nu R^2 = \text{const.}$, where R is, say, the distance of a neighbouring atom from the centre of the vacancy, so that

$$-\frac{d\nu/\nu}{dR/R} = 2. \quad (6.20)$$

On passing from one type of crystal to another $(dR/R)/(da/a)$ is unity, since R/a depends only on the lattice geometry. However, when a given crystal is compressed, da/a and dR/R are not identical. In fact, if we idealize the vacancy as a spherical hole in an isotropic continuum we have

$$\frac{dR/R}{da/a} = \frac{e_A + e_C}{e_A} \quad (6.21)$$

in the notation of (4.6), with $e^* = 0$ in A. For the alkali halides we

may put $\sigma = \frac{1}{4}$; the ratio (6.21) is then 2.25, and from (6.20) and (6.21) we have

$$-\frac{d\nu/\nu}{da/a} = 4.5$$

in much better agreement with experiment. The above is admittedly rather a travesty of Jacobs' argument, but in his more rigorous calculation he also found it necessary to introduce the magnification factor (6.21).

References

- BULEY, B. A., 1950, Proc. Phys. Soc. A **63** 191.
- BITTER, F., 1931, Phys. Rev. **37** 1526.
- BRENNER, H., 1958, The Physics of Fluids **1** 388.
- BURGERS, J. M., 1939, Proc. Acad. Sci. Amsterdam **42** 293.
- BYRD, P. F. and M. D. FRIEDMAN, 1954, Handbook of Elliptic Integrals (Springer-Verlag, Berlin 1954).
- CHRISTIAN, J. W., 1958, Acta Met. **6** 377.
- CHRISTIAN, J. W., 1959, Acta Met. **7** 218.
- DANIELE, E., 1911, Nuovo Cimento **[6]** 1 211.
- DAS, S. C., 1953, Bull. Calcutta Math. Soc. **45** 55.
- DYSON, F. W., 1891, Q. J. Pure Appl. Math. **25** 259.
- EDWARDES, D., 1893, Q. J. Pure Appl. Math. **26** 70.
- ESHELBY, J. D., 1951, Phil. Trans. Roy. Soc. A **244** 87.
- ESHELBY, J. D., 1956, in Solid State Physics (ed. Seitz and Turnbull) **3** 79.
- ESHELBY, J. D., 1957, Proc. Roy. Soc. A **241** 376.
- ESHELBY, J. D., 1958, Ann. Physik **[7]** 1 116.
- ESHELBY, J. D., 1959a, in Internal Stresses and Fatigue in Metals (ed. Rassweiler and Grube; Elsevier, Amsterdam, 1959) 41.
- ESHELBY, J. D., 1959b, Proc. Roy. Soc. A **252** 561.
- FERRERS, N. M., 1877, Q. J. Pure Appl. Math. **14** 1.
- FISHER, J. C. and D. TURNBULL, 1953, Acta Met. **1** 310.
- FREDHOLM, I., 1900, Acta Math. **23** 1.
- FRIEDEL, J., 1954, Advances in Physics **3** 446.
- FRIEDEL, J., 1956, Les Dislocations (Gauthier-Villars, Paris, 1956).
- FRIEDEL, J., 1959, in Internal Stresses and Fatigue in Metals (ed. Rassweiler and Grube; Elsevier, Amsterdam, 1959) 220.
- GOODIER, J. N., 1936, Phil. Mag. **22** 678.
- GOODIER, J. N., 1937, Phil. Mag. **23** 1017.
- GREEN, A. E. and I. N. SNEDDON, 1949, Proc. Camb. Phil. Soc. **46** 159.
- HERSHEY, A. V., 1954, J. Appl. Mech. **21** 236.
- HILL, R., 1952, Proc. Phys. Soc. A **65** 349.
- INGLIS, C. E., 1913, Trans. Inst. Naval Arch. **55** 219.
- JACOBS, I. S., 1954, Phys. Rev. **93** 93.
- KAUFMAN, L., 1959, Acta Met. **7** 216.
- KAUFMAN, L. and M. COHEN, 1958, Progress in Metal Physics **7** 165.
- KELLOGG, O. D., 1929, Potential Theory (Springer-Verlag, Berlin, 1929).
- KRÖNER, E., 1953a, Diplomarbeit (Stuttgart, 1953).

- KRÖNER, E., 1953b, Z. Phys. 136 404.
— KRÖNER, E., 1954, Acta Met. 2 302.
KRÖNER, E., 1958a, Kontinuumstheorie der Versetzungen und Eigenspannungen (Springer-Verlag, Berlin, 1958).
KRÖNER, E., 1958b, Z. Phys. 151 504.
LIFSHITZ, I. M. and N. ROSENZWEIG, 1947, J. Eksp. Teor. Fiz. 17 783.
LOVE, A. E. H., 1954, Mathematical Theory of Elasticity (Cambridge University Press, 1954).
LURIE, A. I., 1952, Doklady Akad. Nauk SSSR 87 709.
MACHLIN, E. S. and S. WEINIG, 1953, Acta Met. 1 480.
MACMILLAN, W. D., 1958, The Theory of the Potential (Dover Publications, New York, 1958) 175.
MOLLWO, E., 1933, Z. Phys. 85 56.
NABARRO, F. R. N., 1940, Proc. Roy. Soc. A 175 519.
NABARRO, F. R. N., 1951, Phil. Mag. 42 1224.
NABARRO, F. R. N., 1952, Advances in Physics 1 269.
NEUBER, H., 1958, Kerbspannungslehre (Springer-Verlag, Berlin, 1958).
NIESEL, W., 1953, Inauguraldissertation, Karlsruhe.
OSBORN, J. A., 1945, Phys. Rev. 67 351.
PATEL, J. R. and M. COHEN, 1953, Acta Met. 1 531.
PEACH, M. O., 1951, J. Appl. Phys. 22 1359.
POINCARÉ, H., 1899, Théorie du Potentiel Newtonien (Carré et Naud, Paris, 1899) 118.
REINER, M., 1958, Encyclopedia of Physics VI (Springer-Verlag, Berlin, 1958) 528.
ROBINSON, K., 1951, J. Appl. Phys. 22 1045.
SACK, R. A., 1946, Proc. Phys. Soc. 58 729.
SADOWSKY, M. A. and E. STERNBERG, 1949, J. Appl. Mech. 16 149.
SEEGER, A., 1955, Encyclopedia of Physics VII (1) 383.
SEGEDIN, C. M., 1951, Proc. Camb. Phil. Soc. 47 396.
SEN, B., 1933, Bull. Calcutta Math. Soc. 25 107.
SNEDDON, I. N., 1951, Fourier Transforms (McGraw-Hill Book Company, New York, 1951).
SOKOLNIKOFF, I. S., 1946, Mathematical Theory of Elasticity (McGraw-Hill Book Company, New York, 1946).
SOMIGLIANA, C., 1914, R. C. Accad. Lincei [5] 23 (1) 463.
SOMIGLIANA, C., 1915, R. C. Accad. Lincei [5] 24 (1) 655.
STARR, A. T., 1928, Proc. Camb. Phil. Soc. 24 489.
STERNBERG, E., 1958, Appl. Mech. Rev. 11 1.
STROH, A. N., 1958, Phil. Mag. 30 623.
TEWORDT, L., 1958, Phys. Rev. 109 61.
TIMOSHENKO, S. and J. N. GOODIER, 1951, Theory of Elasticity (McGraw-Hill Book Company, New York, 1951) 425.
VOLTERRA, V., 1907, Ann. Ecole Norm. Super. [3] 24 400.
WINTENBERGER, M., 1957, C. R. Acad. Sci. Paris 244 2800.

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ME 238C - THEORY OF ELASTICITY

TENTATIVE SCHEDULE FOR THE FIRST 9 LECTURES

1. THE 3-D ELASTIC GREEN'S FUNCTION

2. ELLIPSOIDAL INHOMOGENEITIES AND INCLUSIONS

REFERENCE - J.D. ESHELBY, ELASTIC INCLUSIONS AND INHOMOGENEITIES

IN PROGRESS IN SOLID MECHANICS, v. II., edited by

I.N. SNEDDON AND R.HILL, NORTH-HOLLAND CO., AMSTERDAM.

CHAPTER 3, p. 89-140 (1961).

THE 3-D ELASTIC GREEN'S FUNCTIONS

FROM LAST QUARTER, WE KNOW THAT THE 2-D GREEN'S

FUNCTION $G_{im}(x, y; x', y')$ IS DEFINED TO BE THE i^{th} COMPONENT

OF THE ELASTIC DISPLACEMENT AT (x', y') IN AN INFINITE

MEDIUM DUE TO A UNIT LINE FORCE APPLIED IN THE m^{th}

DIRECTION AT (x', y') . IN 3-D WE DEFINE THE INFINITE

MEDIUM GREEN'S FUNCTION

$$G_{im}(x, y, z; x', y', z')$$

AS THE i^{th} COMPONENT OF THE ELASTIC DISPLACEMENT AT

(x, y, z) DUE TO A UNIT POINT FORCE APPLIED AT (x', y', z')

IN THE m^{th} (x_m) DIRECTION. $i, m = 1, 2, 3$. FOR BREVITY

WE WRITE

$$G_{im}(x, x')$$

NOW THE G_{im} SATISFY THE EQUATIONS OF EQUILIBRIUM WRITTEN IN TERMS OF DISPLACEMENTS, NAMELY

$$C_{ijkm} \frac{\partial^2 G_{km}}{\partial x_j \partial x_i} + \delta_{jm} \delta(\underline{x} - \underline{x}') = 0$$

DIVERGENCE OF $\underline{0}$ POINT FORCE (BODY FORCE).

WE EXPECT THAT $G_{km}(\underline{x}, \underline{x}') \rightarrow 0$ AS $|\underline{x} - \underline{x}'| \rightarrow \infty$ AND SIMILARLY FOR ITS DERIVATIVES.

SOLUTION FOR G_{km} USING FOURIER TRANSFORMS

GENERAL REMARKS: THE FOURIER TRANSFORM $g(k)$ OF THE FUNCTION $G(x)$ IN ONE-DIMENSION IS DEFINED BY

$$g(k) = \int_{-\infty}^{\infty} dx e^{ikx} G(x). \quad (1)$$

IT CAN BE SHOWN THAT THE FOLLOWING INVERSION RELATION IS VALID, NAMELY,

$$G(x) = \frac{1}{2\pi} \int_{-\infty}^{\infty} dk e^{-ikx} g(k). \quad (2)$$

THE GENERALIZATION OF EQUATIONS (1) AND (2) TO 3-D IS

$$g(k_1, k_2, k_3) = \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} dx dy dz e^{i(k_1 x + k_2 y + k_3 z)} G(x, y, z) \quad (3)$$

$$G(x, y, z) = \left(\frac{1}{2\pi}\right)^3 \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} dk_1 dk_2 dk_3 e^{-i(k_1 x + k_2 y + k_3 z)} g(k_1, k_2, k_3). \quad (4)$$

CALLING $x = x_1, y = x_2, z = x_3$ AND NOTING

$$\underline{k} = k_i \underline{e}_i = k_1 \underline{e}_1 + k_2 \underline{e}_2 + k_3 \underline{e}_3 \quad (5)$$

$$\underline{x} = x_i \underline{e}_i = x_1 \underline{e}_1 + x_2 \underline{e}_2 + x_3 \underline{e}_3$$

SO THAT

$$\underline{k} \cdot \underline{x} = k_1 x_1 + k_2 x_2 + k_3 x_3 \quad (6)$$

WHERE \underline{e}_i IS A UNIT VECTOR ALONG THE x_i -AXIS,

WE CAN WRITE

$$g(\underline{k}) = \iiint_{-\infty}^{\infty} d^3 \underline{x} e^{i \underline{k} \cdot \underline{x}} G(\underline{x}) \quad (7)$$

$$G(\underline{x}) = \frac{1}{8\pi^3} \iiint_{-\infty}^{\infty} d^3 \underline{k} e^{-i \underline{k} \cdot \underline{x}} g(\underline{k}) \quad (8)$$

\underline{k} IS CALLED THE WAVE VECTOR.

COROLLARY: SUPPOSE $G(x, y, z) = \delta(x) \delta(y) \delta(z) = \delta(\underline{x})$. FROM (3)

$$1 = \iint_{-\infty}^{\infty} \iint_{-\infty}^{\infty} \iint_{-\infty}^{\infty} dx dy dz e^{i \underline{k} \cdot \underline{x}} \delta(\underline{x}) \quad (9)$$

SO THAT THE FOURIER TRANSFORM OF A DELTA FN. IS ONE.

THEN FROM (4)

$$\delta(x) = \left(\frac{1}{2\pi}\right)^3 \iint_{-\infty}^{\infty} \iint_{-\infty}^{\infty} \iint_{-\infty}^{\infty} d^3 \underline{k} e^{-i \underline{k} \cdot \underline{x}} \quad (10)$$

NOW LET US USE THESE RESULTS TO SOLVE FOR THE
ELASTIC GREEN'S FUNCTIONS.

RETURN TO

$$C_{ijk} e \frac{\partial^2}{\partial x_i \partial x_j} G_{km}(\underline{x}, \underline{x}') + \delta_{jm} \delta(\underline{x} - \underline{x}') = 0. \quad (1)$$

WE CAN ALWAYS SHIFT OUR ORIGIN SO THAT $\underline{x}' = 0$ (i.e.,

PT. FORCE IS APPLIED AT ORIGIN). THUS, WE MUST SOLVE

$$C_{ijk} e \frac{\partial^2}{\partial x_i \partial x_j} G_{km}(\underline{x}) + \delta_{jm} \delta(\underline{x}) = 0. \quad (2)$$

CALL

$g_{km}(\underline{k})$ = FOURIER TRANSFORM OF $G_{km}(\underline{x})$, i.e.,

$$g_{km}(\underline{k}) = \iiint_{-\infty}^{\infty} d^3x e^{i\underline{k} \cdot \underline{x}} G_{km}(\underline{x}) \quad (3)$$

SO THAT

$$G_{km}(\underline{x}) = \frac{1}{8\pi^3} \iiint_{-\infty}^{\infty} d^3k e^{-i\underline{k} \cdot \underline{x}} g_{km}(\underline{k}). \quad (4)$$

USING (4) AND (1), WE CAN REWRITE (2) AS

$$C_{ijk} e \frac{\partial^2}{\partial x_i \partial x_j} \left\{ \frac{1}{8\pi^3} \iiint_{-\infty}^{\infty} d^3k e^{-i\underline{k} \cdot \underline{x}} g_{km}(\underline{k}) \right\}$$

$$+ \delta_{jm} \frac{1}{8\pi^3} \iiint_{-\infty}^{\infty} d^3k e^{-i\underline{k} \cdot \underline{x}} = 0 \quad (5)$$

SINCE $\frac{\partial^2}{\partial x_i \partial x_j} e^{-i\underline{k} \cdot \underline{x}} = (-ik_i)(-ik_j) e^{-i\underline{k} \cdot \underline{x}}$

$$= -k_i k_j e^{-i\underline{k} \cdot \underline{x}}, \quad (6)$$

$$\iiint_{-\infty}^{\infty} d^3k e^{-i\underline{k} \cdot \underline{x}} \left\{ -C_{ijk} k_i k_j g_{km}(\underline{k}) + \delta_{jm} \right\} = 0 \quad (7)$$

EQUATION (17) IS EASILY SATISFIED IF $\underline{g}_{km}(\underline{x})$ SATISFIES

$$(C_{ijk}\ell_i\ell_k)\underline{g}_{km}(\underline{x}) = \delta_{jm}. \quad (18)$$

NOW THE UNIT VECTOR $\underline{\xi}$ ALONG \underline{k} IS DEFINED BY

$$\underline{\xi} = \frac{\underline{k}}{|\underline{k}|} \quad (19)$$

SO THAT

$$\ell_i = |\underline{k}| \xi_i \quad (20)$$

HENCE

$$|\underline{k}|^2 C_{ijk} \xi_i \xi_k \underline{g}_{km}(\underline{x}) = \delta_{jm}. \quad (21)$$

DEFINE THE SYMMETRIC MATRIX $[M]$ BY

$$M_{jk} = C_{ijk} \xi_i \xi_k. \quad (22)$$

THEN

$$\underline{g}_{km}(\underline{x}) = \frac{M_{km}^{-1}(\underline{\xi})}{|\underline{k}|^2} \quad (23)$$

NOTE:

$$M_{km} \in \text{HENCE } M_{km}^{-1} \text{ (THE INVERSE OF } [M])$$

DEPEND ONLY ON $\underline{\xi}$ (THE DIRECTION OF \underline{k}) BUT NOT ON $|\underline{k}|!!$

IT TURNS OUT THAT EQUATION (23) IS ALL WE NEED TO KNOW

TO SOLVE ELLIPSOIDAL INCLUSION PROBLEMS. NEVERTHELESS WE

WILL GO AHEAD AND SOLVE FOR $G_m(x)$ ANYWAY.

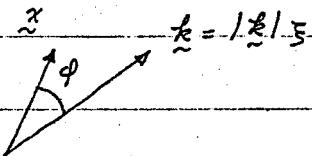
NOW, USING (14),

$$G_{km}(\underline{x}) = \frac{1}{8\pi^3} \iiint_{-\infty}^{\infty} d\underline{k}_1 d\underline{k}_2 d\underline{k}_3 e^{-i\underline{k} \cdot \underline{x}} \frac{M_{km}^{-1}(\underline{s})}{|\underline{k}|^2}. \quad (24)$$

THE INVERSION INTEGRAL IS MOST EASILY DONE IN SPHERICAL POLAR COORDINATES:

$$d\underline{k}_1 d\underline{k}_2 d\underline{k}_3 = |\underline{k}|^2 \sin\varphi d\theta d\varphi d\varphi \quad (25)$$

WHERE φ IS THE ANGLE BETWEEN \underline{k} AND \underline{x} .



$$\text{i.e., } \underline{k} \cdot \underline{x} = |\underline{k}| |\underline{x}| \cos\varphi$$

THEN

$$G_{km}(\underline{x}) = \frac{1}{8\pi^3} \int_0^{2\pi} d\theta \int_0^\pi \sin\varphi d\varphi \int_0^\infty d|\underline{k}| M_{km}^{-1}(\underline{s}) e^{-i|\underline{k}| |\underline{x}| \cos\varphi}$$

OR

$$\text{LETTING } u = |\underline{x}| / |\underline{k}| ; \quad du = |\underline{x}| d|\underline{k}|$$

$$G_{km}(\underline{x}) = \frac{1}{8\pi^3 |\underline{x}|} \int_0^{2\pi} d\theta \int_0^\pi \sin\varphi d\varphi M_{km}^{-1}(\underline{s}) \int_0^\infty du e^{-iu \cos\varphi}. \quad (27)$$

$$\text{Now } e^{-iu \cos\varphi} = \cos(u \cos\varphi) - i \sin(u \cos\varphi). \quad (28)$$

AND CLEARLY $G_{km}(\underline{x})$ IS REAL, SO WE NEED ONLY

WORRY ABOUT THE REAL PART OF THE INTEGRAL IN (27).

THUS $G_{km}(x) = \frac{1}{8\pi^3 |x|} \int_0^{2\pi} d\theta \int_0^\pi \sin\varphi d\varphi M_{km}^{-1}(\xi) \int_0^\infty du \cos(u \cos\varphi)$. (29)

NOW,

$$\boxed{\int_0^\infty du \cos(u \cos\varphi) = \pi \delta(\cos\varphi)} \quad (30)$$

PROOF: USING THE 1-D FORM OF (10),

$$\begin{aligned} \delta(x) &= \frac{1}{2\pi} \int_{-\infty}^{\infty} dk e^{-ikx} \\ &= \frac{1}{2\pi} \int_{-\infty}^{\infty} dk \{ \cos kx - i \sin kx \} \end{aligned}$$

BUT

$$\int_{-\infty}^{\infty} dk \sin kx = 0 \quad \text{SINCE } \sin kx \text{ IS ODD IN } k$$

$$\int_{-\infty}^{\infty} dk \cos kx = 2 \int_0^{\infty} dk \cos kx$$

$$\therefore \delta(x) = \frac{1}{2\pi} \cdot 2 \int_0^{\infty} dk \cos kx$$

OR $\int_0^{\infty} dk \cos kx = \pi \delta(x)$.

LETTING

$k \rightarrow u$, $x \rightarrow \cos\varphi$, WE GET EQUATION (30)

HENCE

$$G_{km}(x) = \frac{1}{8\pi^2 |x|} \int_0^{2\pi} d\theta \int_0^\pi d\varphi \sin\varphi \delta(\cos\varphi) M_{km}^{-1}(\xi) \quad (31)$$

NOW $\delta(\cos\varphi) = 0$ UNLESS $\varphi = \frac{\pi}{2} \rightarrow 0 \leq \varphi \leq \pi$

AND

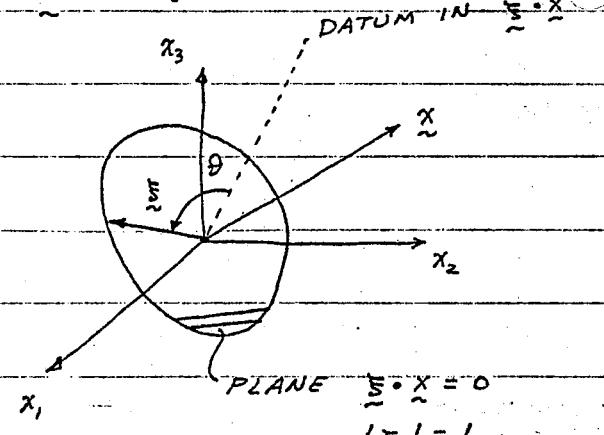
$$\int_0^\pi d\varphi f(\varphi) \delta(\cos\varphi) = f(\varphi = \frac{\pi}{2}).$$

$$G_{km}(x) = \frac{1}{8\pi^2/|x|} \int_0^{2\pi} d\theta [M_{km}^{-1}]_{\varphi=\frac{\pi}{2}}$$

$$G_{km}(x) = \frac{1}{8\pi^2/|x|} \int_0^{2\pi} d\theta \left\{ M_{km}^{-1}(\xi) \right\} \quad (32)$$

ALL ξ SUCH THAT
 $\xi \cdot x = 0$

* $\varphi = \frac{\pi}{2} \rightarrow \xi \cdot x = 0$ AND $\xi \cdot \tilde{x} = 0$.



EQUATION (32) IS VALID FOR

ARBITRARY ANISOTROPY! LET

US SPECIALIZE TO ELASTIC

ISOTROPY. WE CAN PERFORM

THE INTEGRAL OVER θ EXACTLY.

ISOTROPIC SOLIDS

$$C_{ijkl} = \lambda \delta_{ij} \delta_{kl} + \mu (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk}) \quad (33)$$

$$M_{jk} = C_{ijkl} \xi_i \xi_l = \lambda \xi_j \xi_k + \mu (\xi_j \xi_k + \xi_l \xi_l \delta_{jk}) \quad (34)$$

BUT

$$\underline{\xi}_j \cdot \underline{\xi}_k = \underline{\xi} \cdot \underline{\xi} = 1 \quad \text{SINCE } \underline{\xi} \text{ IS A UNIT VECTOR.} \quad (35)$$

THUS

$$M_{jk} = \mu \left\{ \delta_{jk} + \frac{\lambda + \mu}{\mu} \underline{\xi}_j \underline{\xi}_k \right\}. \quad (36)$$

GUESS

$$M_{km}^{-1} = \frac{1}{\mu} \left\{ \delta_{km} + A \underline{\xi}_k \underline{\xi}_m \right\} \quad (37)$$

AND SEE IF WE CAN FIND A VALUE OF A SUCH

THAT

$$M_{jk} M_{km}^{-1} = \delta_{jm}. \quad (38)$$

NOW

$$\begin{aligned} M_{jk} M_{km}^{-1} &= \left\{ \delta_{jk} + \frac{\lambda + \mu}{\mu} \underline{\xi}_j \underline{\xi}_k \right\} \left\{ \delta_{km} + A \underline{\xi}_k \underline{\xi}_m \right\} \\ &= \delta_{jk} \delta_{km} + A \underline{\xi}_j \underline{\xi}_m + \frac{\lambda + \mu}{\mu} \underline{\xi}_j \underline{\xi}_m + \\ &\quad \frac{\lambda + \mu}{\mu} A \underline{\xi}_j \underline{\xi}_m (\underline{\xi}_k \underline{\xi}_k) \\ &= \delta_{jm} + \underline{\xi}_j \underline{\xi}_m \left\{ A + \frac{\lambda + \mu}{\mu} + A \frac{\lambda + \mu}{\mu} \right\} \end{aligned}$$

EQUATION (38) IS SATISFIED IF

$$A \left\{ 1 + \frac{\lambda + \mu}{\mu} \right\} = - \frac{\lambda + \mu}{\mu}$$

OR

$$A = - \frac{\lambda + \mu}{\lambda + 2\mu} \quad (39)$$

THUS FOR ISOTROPY

$$M_{km}^{-1} = \frac{1}{\mu} \left\{ \delta_{km} - \frac{\lambda + \mu}{\lambda + 2\mu} \underline{\xi}_k \underline{\xi}_m \right\} \quad (40)$$

*10

IN THE PLANE $\underline{x} \cdot \underline{x} = 0$ WE CAN
WRITE

$$\underline{\xi} = \underline{\alpha} \cos \theta + \underline{\beta} \sin \theta$$

ANY

WHERE $\underline{\alpha}$ AND $\underline{\beta}$ ARE A TWO FIXED ORTHOGONAL UNIT
VECTORS. \curvearrowleft IN THE PLANE $\underline{x} \cdot \underline{x} = 0$ \curvearrowright THUS

$$\underline{\xi}_k = \alpha_k \cos \theta + \beta_k \sin \theta$$

$$(\underline{\xi}_k \underline{\xi}_m) = \alpha_k \alpha_m \cos^2 \theta + \beta_k \beta_m \sin^2 \theta + (\alpha_k \beta_m + \alpha_m \beta_k) \sin \theta \cos \theta$$

NOW

$$\int_0^{2\pi} d\theta = 2\pi ; \quad \int_0^{2\pi} \cos^2 \theta d\theta = \int_0^{2\pi} \sin^2 \theta d\theta = \pi$$

$$\int_0^{2\pi} \sin \theta \cos \theta d\theta = 0$$

THUS, FORMULA (32) REDUCES TO

$$G_{km}(\underline{x}) = \frac{1}{8\pi |\underline{x}|} \frac{1}{\mu} \left\{ 2\delta_{km} - \frac{\lambda + \mu}{\lambda + 2\mu} [\alpha_k \alpha_m + \beta_k \beta_m] \right\}. \quad (41)$$

$$\text{NOW } \alpha_k \alpha_m + \beta_k \beta_m + \tau_k \tau_m = \delta_{km} \quad (42)$$

WHERE

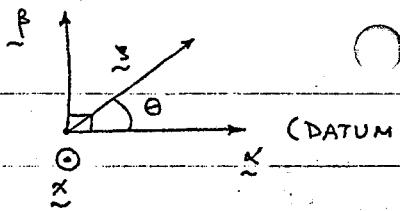
$\underline{\tau}$ IS A UNIT VECTOR ALONG \underline{x} , i.e.,

$$\underline{\tau}_k = \frac{\underline{x}_k}{|\underline{x}|} .$$

PROOF OF (42): PICK 1, 2, 3 AXES ALONG $\underline{\alpha}, \underline{\beta}$, AND

$$\underline{\tau} (\alpha_1 = 1, \alpha_2 = 0, \alpha_3 = 0; \beta_1 = 0, \beta_2 = 1, \beta_3 = 0; \tau_1 = \tau_2 = 0 \& \tau_3).$$

FORMULA WORKS AND, BEING A RELATION AMONG TENSORS,
IS TRUE IN ANY COORDINATE SYSTEM.



HENCE

$$G_{km}(\underline{x}) = \frac{1}{8\pi\mu|\underline{x}|} \left\{ \left(2 - \frac{\lambda+\mu}{\lambda+2\mu} \right) \delta_{km} + \frac{\lambda+\mu}{\lambda+2\mu} \frac{x_k x_m}{|\underline{x}|^2} \right\}$$

$$= \frac{1}{8\pi\mu|\underline{x}|} \left\{ \frac{\lambda+3\mu}{\lambda+2\mu} \delta_{km} + \frac{\lambda+\mu}{\lambda+2\mu} \frac{x_k x_m}{|\underline{x}|^2} \right\} \quad (43)$$

NOW $\frac{\lambda+3\mu}{\lambda+2\mu} = \frac{3-4\nu}{2(1-\nu)}$ (44)

$$\frac{\lambda+\mu}{\lambda+2\mu} = \frac{1}{2(1-\nu)} \quad (45)$$

$\therefore G_{km}(\underline{x}) = \frac{1}{16\pi\mu(1-\nu)|\underline{x}|} \left\{ (3-4\nu) \delta_{km} + \frac{x_k x_m}{|\underline{x}|^2} \right\} \quad (46)$

HENCE $G_{km}(\underline{x}) = G_{mk}(\underline{x})$ AND

$$G_{km}(\underline{x}, \underline{x}') = \frac{1}{16\pi\mu(1-\nu)|\underline{x}-\underline{x}'|} \left\{ (3-4\nu) \delta_{km} + \frac{(x_k - x'_k)(x_m - x'_m)}{|\underline{x}-\underline{x}'|^2} \right\}$$

$$* |\underline{x}-\underline{x}'| = \sqrt{(x_1 - x'_1)^2 + (x_2 - x'_2)^2 + (x_3 - x'_3)^2}$$

EXAMPLES OF THE USE OF $G_{km}(\underline{x}, \underline{x}')$

1. CONSTRUCTION OF THE 2-D LINE FORCE GREEN'S FUNCTIONS

(WE FOUND THEM LAST QUARTER)

CALL $g_{km}(x, y; x', y')$ THE DISPLACEMENT IN THE k^{\perp}

DIRECTION AT (x, y) DUE TO A LINE OF FORCE

OF UNIT STRENGTH/UNIT LENGTH APPLIED IN THE m^{\perp}

DIRECTION AT (x', y') .

CLEARLY

$$g_{km}(x, y; x', y') = \lim_{R \rightarrow \infty} \int_{-R}^R dz' G_{km}(x, y, z; x', y', z').$$

WE NEED

$$I_1 = \lim_{R \rightarrow \infty} \int_{-R}^R \frac{dz'}{\{(x-x')^2 + (y-y')^2 + (z-z')^2\}^{3/2}}$$

AND

$$I_{km} = \lim_{R \rightarrow \infty} \int_{-R}^R \frac{(x_k - x'_k)(x_m - x'_m) dz'}{\{(x-x')^2 + (y-y')^2 + (z-z')^2\}^{3/2}}$$

NOW, CALLING $\rho^2 = (x-x')^2 + (y-y')^2$

$$I_1 = \lim_{R \rightarrow \infty} \int_{-R}^R \frac{dz'}{\sqrt{\rho^2 + (z-z')^2}}$$

OR

$$\text{LETTING } z' = u + z$$

$$I_1 = \lim_{R \rightarrow \infty} \int_{-R-z}^{R-z} \frac{du}{(\rho^2 + u^2)^{1/2}}$$

$$= \lim_{R \rightarrow \infty} \left\{ \int_0^{R-z} \frac{du}{\sqrt{\rho^2 + u^2}} + \int_0^{R+z} \frac{du}{\sqrt{\rho^2 + u^2}} \right\}$$

$$= \lim_{R \rightarrow \infty} \left\{ \ln \frac{R-z + \sqrt{\rho^2 + (R-z)^2}}{\rho} + \ln \frac{R+z + \sqrt{(R+z)^2 + \rho^2}}{\rho} \right\}$$

$$= \lim_{R \rightarrow \infty} \left\{ \ln \frac{2R}{\rho} + \ln \frac{2R}{\rho} \right\}$$

$$= \lim_{R \rightarrow \infty} \{ \ln 2R \} - 2 \ln \rho$$

↑

AN INFINITE CONSTANT \Rightarrow RIGID TRANSLATION WHICH WE CAN DISCARD

FOR $K, m = 1$ OR 2 $(x_k - x'_k)(x_m - x'_m)$ IS CONSTANT

AND CAN BE PULLED OUT OF THE INTEGRAL FOR I_{km} .

$[K, m = 1$ OR $2]$:

$$\begin{aligned} I_{km} &= (x_k - x'_k)(x_m - x'_m) \lim_{R \rightarrow \infty} \int_{-R}^R \frac{dz'}{\{p^2 + (z-z')^2\}^{3/2}} \\ &= (x_k - x'_k)(x_m - x'_m) \lim_{R \rightarrow \infty} \left\{ \int_0^{R-z} \frac{du}{(p^2 + u^2)^{3/2}} + \int_0^{R+z} \frac{du}{(p^2 + u^2)^{3/2}} \right\} \\ &= (x_k - x'_k)(x_m - x'_m) \lim_{R \rightarrow \infty} \left\{ \frac{R-z}{p^2 \sqrt{p^2 + (R-z)^2}} + \frac{R+z}{p^2 \sqrt{p^2 + (R+z)^2}} \right\} \\ &= \frac{2(x_k - x'_k)(x_m - x'_m)}{p^2} \end{aligned}$$

THUS, FOR $K, m = 1$ OR 2 (DISCARDING THE INFINITE CONSTANT)

$$g_{km}(x, y; x', y') = -\frac{3-4w}{8\pi\mu(1-w)} \ln p + \frac{1}{8\pi\mu(1-w)} \frac{(x_k - x'_k)(x_m - x'_m)}{p^2}$$

THIS IS PRECISELY THE EQUATION FOR THE 2-D GREEN'S FUNCTIONS WE DERIVED LAST QUARTER.

NOW SUPPOSE $K = 1$ OR 2 AND $m = 3$ OR $K = 3$ AND $m = 1$ OR 2 .

THEN g_{13}, g_{23}, g_{31} , AND g_{32} WILL BE PROPORTIONAL

TO

$$\lim_{R \rightarrow \infty} \int_{-R}^R \frac{(z-z') dz'}{\{p^2 + (z-z')^2\}^{3/2}} = J$$

OR

$$-J = \lim_{R \rightarrow \infty} \left\{ \int_0^{R-z} \frac{udu}{(p^2 + u^2)^{3/2}} - \int_0^{R+z} \frac{udu}{(p^2 + u^2)^{3/2}} \right\}$$

$$-J = \lim_{R \rightarrow \infty} \left\{ -\frac{1}{\sqrt{\rho^2 + (R-z)^2}} + \frac{1}{\rho} + \frac{1}{\sqrt{\rho^2 + (R+z)^2}} - \frac{1}{\rho} \right\} = 0 \quad \textcircled{1}$$

THUS $g_{13} = g_{23} = g_{31} = g_{32} = 0$

THE ONLY REMAINING TERM IS $g_{33}(x, y; x', y')$. APART FROM THE TERM PROPORTIONAL TO δ_{km} (WHICH WE HAVE ALREADY EVALUATED) THE REMAINING TERM IN g_{33} IS PROPORTIONAL TO

$$Q = \lim_{R \rightarrow \infty} \int_{-R}^R \frac{(z-z')^2 dz'}{\{\rho^2 + (z-z')^2\}^{3/2}} = \lim_{R \rightarrow \infty} \left\{ \int_0^{R-z} \frac{u^2 du}{(\rho^2 + u^2)^{3/2}} + \int_0^{R+z} \frac{u^2 du}{(\rho^2 + u^2)^{3/2}} \right\} \quad \textcircled{1}$$

$$Q = \lim_{R \rightarrow \infty} \left\{ -\frac{R-z}{\sqrt{\rho^2 + (R-z)^2}} + \ln \left\{ R-z + \sqrt{\rho^2 + (R-z)^2} \right\} - \ln \rho \right. \\ \left. - \frac{R+z}{\sqrt{\rho^2 + (R+z)^2}} + \ln \left\{ R+z + \sqrt{\rho^2 + (R+z)^2} \right\} - \ln \rho \right\}$$

OR

$$Q = -2 - 2 \ln \rho + \lim_{R \rightarrow \infty} 2 \ln 2R \\ \uparrow \qquad \qquad \qquad \uparrow \\ \text{CONSTANT-} \qquad \qquad \qquad \text{INFINITE CONSTANT - RIGID TRANSLATION} \\ \text{RIGID TRANSLATION}$$

HENCE, APART FROM A RIGID TRANSLATION (WHICH IS INFINITE)

$$g_{33}(x, y; x', y') = \frac{1}{16\pi\mu(1-\nu)} \left\{ (3-4\nu)(-2 \ln \rho) - 2 \ln \rho \right\} \\ = \frac{-2[4(1-\nu)]}{16\pi\mu(1-\nu)} \ln \rho \quad \textcircled{1}$$

$$g_{33}(x,y; x',y') = -\frac{1}{2\pi'\mu} \ln\rho \quad (\text{JUST LIKE PROBLEM 2}$$

ON FINAL EXAM)

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Question	Score
1	20
2	10
3	20
4	
5	
6	
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8	
9	
10	
Total	60/60

Name of Student CESAR LORY

Date of Examination 1 JUNE 79

Subject ME 238 C ELASTICITY

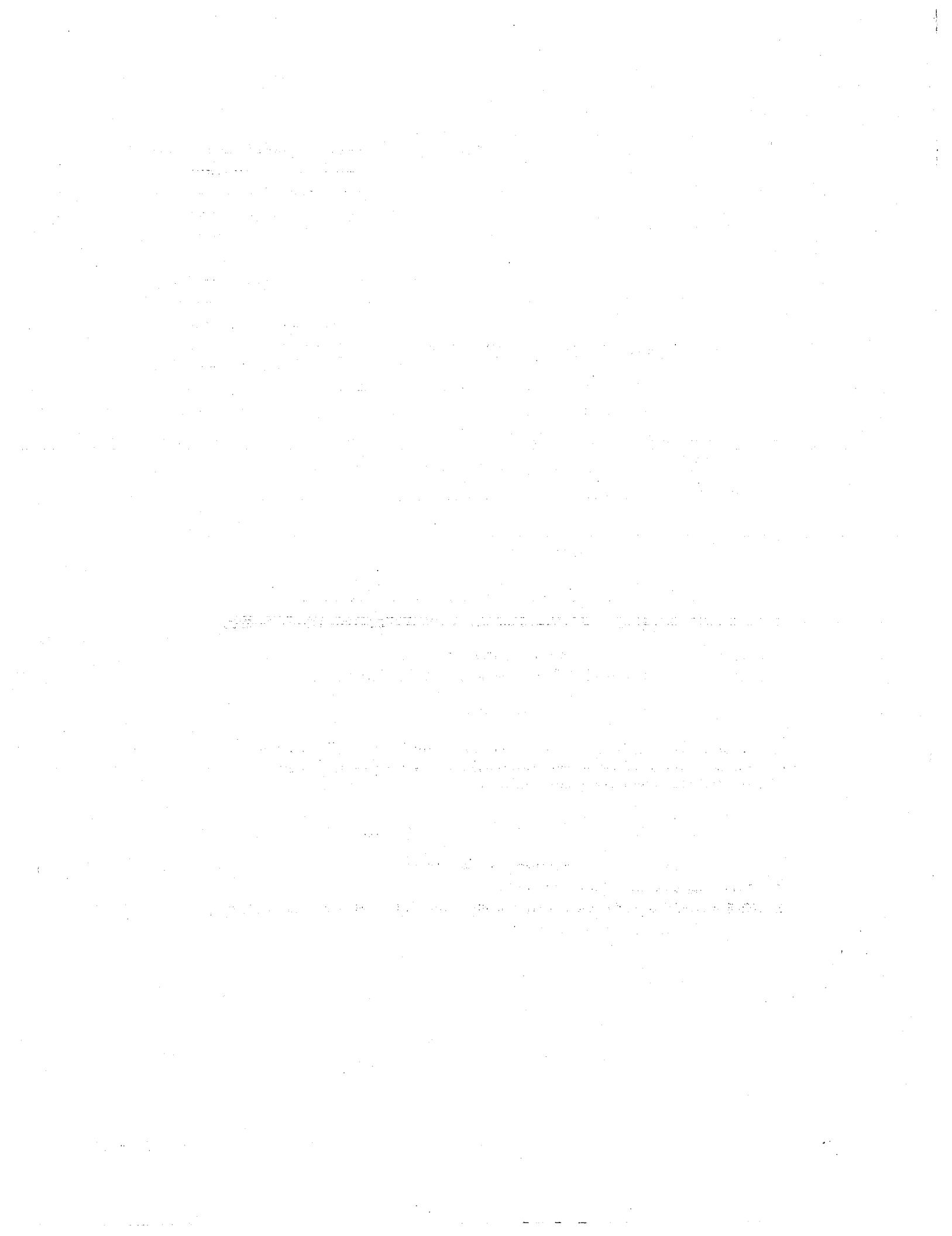
HONORABLE CONDUCT
in academic work is the spirit of conduct in this University.

In recognition of and in the spirit of the Honor Code, I certify that I will neither receive nor give unpermitted aid on this examination and that I will report, to the best of my ability, all Honor Code violations observed by me.

(signed) Cesar Lory
Name _____

SUGGESTIONS FOR CONDUCT

1. Occupy alternate seats where possible.
2. When in doubt as to the meaning of a question, consult the instructor, who will be found in his or her office.



Due Wednesday Tuesday 12 Noon

DIVISION OF APPLIED MECHANICS
DEPARTMENT OF MECHANICAL ENGINEERING

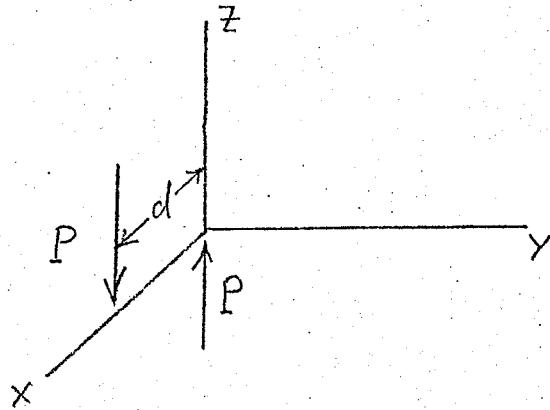
STANFORD UNIVERSITY

ME 238C Theory of Elasticity

Spring 1979

Final Exam

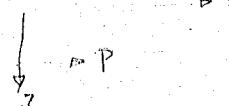
1. Find the displacement field of an infinite body subjected to a couple of magnitude M at the origin, as shown.



2. Consider a semi-infinite elastic solid bounded by the plane $z = 0$, the positive z-axis being directed into the body. A force P is applied at $(0, 0, +c)$ and acts in the positive x-direction. Show that the displacement component u_x in the x-direction at the point $(0, 0, 0)$ is given by

$$u_x(0, 0, 0) = \frac{P(3 - 2v)}{8\pi\mu c}$$

where μ is the shear modulus and v is Poisson's ratio.



3. A rigid sphere of radius R is impressed with a normal force N into an elastic half-space of shear modulus μ and Poisson's ratio v . Determine the radius of contact, the maximum pressure at the interface and the settlement.

1. For the Kelvin Problem we had found that a force of magnitude P at the origin produced a displacement field $u = \nabla B = \frac{1}{4(1-\nu)} \nabla (r \cdot B + \beta)$ where $B = \frac{P\hat{r}}{4\pi\mu r}$ and $\beta = 0$ with: \hat{r} is the vector describing the force and $r = \|r\|$ is the distance from the origin to the field point. For a force located at the point $(d, 0, 0)$ we again find a $u' = \nabla B' = \frac{1}{4(1-\nu)} \nabla (r' \cdot B' + \beta')$ where $B' = \frac{P'\hat{r}'}{4\pi\mu r'}$, and $\beta' = 0$ with: \hat{r}' is the vector describing the force and $r' = \|r'\|$ is the distance from the point of application of the force to the field point

$$u = \frac{Pe_z}{4\pi\mu r} - \frac{1}{4(1-\nu)} \nabla \left(\frac{zP}{4\pi\mu r} \right) \quad r^2 = x^2 + y^2 + z^2$$

$$u' = -\frac{Pe_z}{4\pi\mu r'} + \frac{1}{4(1-\nu)} \nabla \left(\frac{zP}{4\pi\mu r'} \right) \quad r'^2 = (x-d)^2 + y^2 + z^2$$

$$r' = \sqrt{(x-d)^2 + y^2 + z^2}$$

The displacement field due to these two forces are

$$u = u + u' = \frac{Pe_z}{4\pi\mu} \left[\frac{1}{r} - \frac{1}{r'} \right] = \frac{P}{16\pi\mu(1-\nu)} \nabla \left[z \left(\frac{1}{r} - \frac{1}{r'} \right) \right]$$

if we let $d \rightarrow 0$ but keep $Pd = M$ fixed then using the taylor expansion we can find that $\frac{1}{r} - \frac{1}{r'} = d \frac{\partial}{\partial x} \left(\frac{1}{r} \right) \Big|_{d=0} - \frac{d^2}{2!} \frac{\partial^2}{\partial x^2} \left(\frac{1}{r} \right) \Big|_{d=0} + \dots$

$$\text{then the limit } u = \lim_{\substack{d \rightarrow 0 \\ Pd = M}} \frac{M e_z}{4\pi\mu} \left\{ \frac{\partial}{\partial x} \left(\frac{1}{r} \right) \right\} = \frac{M}{16\pi\mu(1-\nu)} \nabla \left[z \frac{\partial}{\partial x} \left(\frac{1}{r} \right) \right]$$

$$\text{or } u = -\frac{M e_z}{4\pi\mu} \frac{x}{r^3} + \frac{M}{16\pi\mu(1-\nu)} \nabla \left[\frac{zx}{r^3} \right] \quad \text{using } \frac{\partial}{\partial x} \frac{1}{r} = -\frac{x}{r^3}$$

$$\text{now } \frac{\partial}{\partial x} \left(\frac{zx}{r^3} \right) = \frac{z}{r^3} + zx \frac{\partial}{\partial x} \left(\frac{1}{r^3} \right) = \frac{z}{r^3} - \frac{3zx^2}{r^5}$$

$$\frac{\partial}{\partial y} \left(\frac{zx}{r^3} \right) = zx \frac{\partial}{\partial y} \left(\frac{1}{r^3} \right) = -\frac{3zy}{r^5}$$

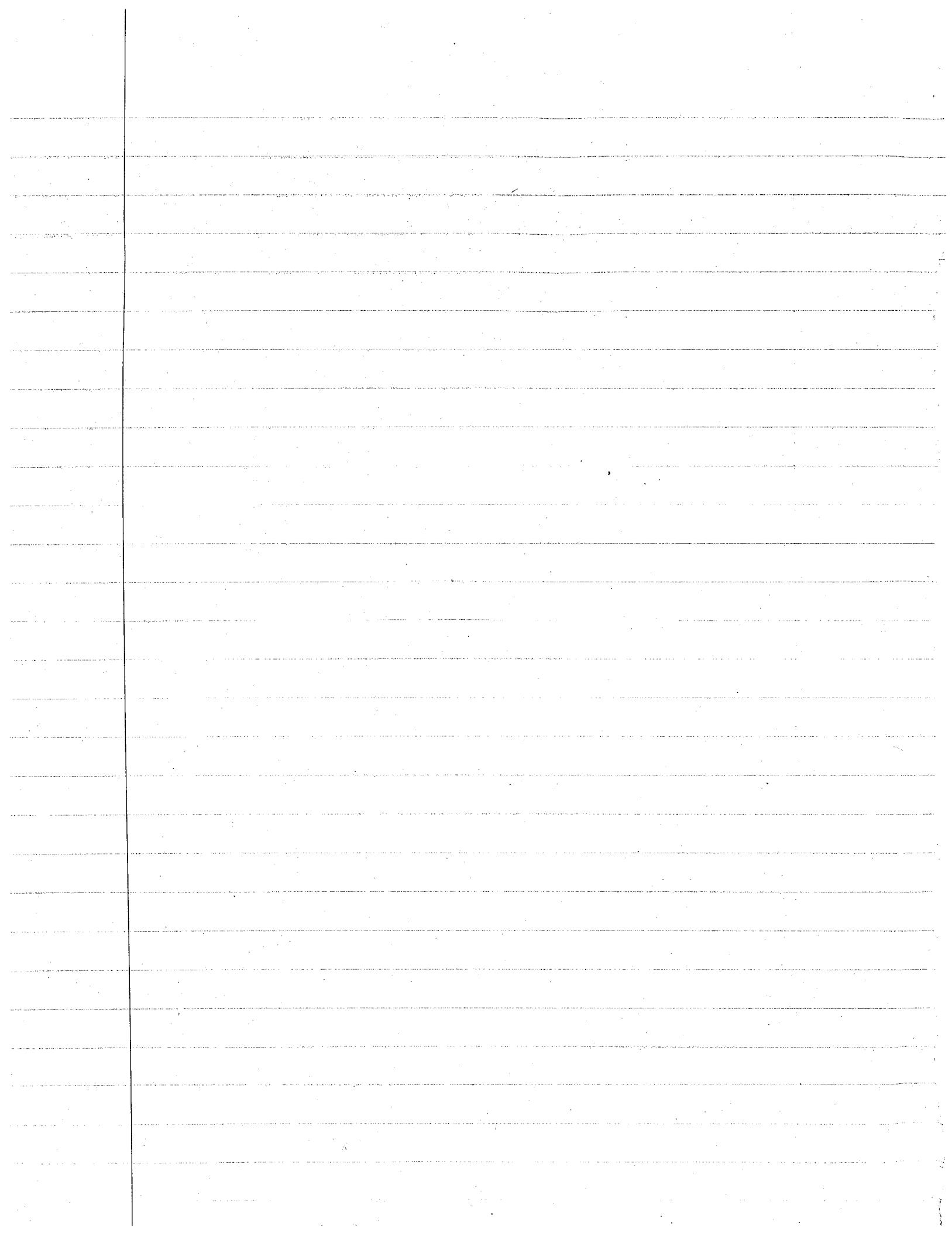
$$\frac{\partial}{\partial z} \left(\frac{zx}{r^3} \right) = \frac{x}{r^3} + zx \frac{\partial}{\partial z} \left(\frac{1}{r^3} \right) = \frac{x}{r^3} - \frac{3z^2x}{r^5}$$

thus

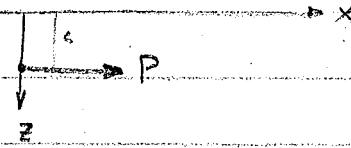
$$u_x = \frac{M}{16\pi\mu(1-\nu)} \frac{z}{r^3} \left[1 - \frac{3x^2}{r^2} \right]; \quad u_y = \frac{M}{16\pi\mu(1-\nu)} \left(-\frac{3zy}{r^5} \right)$$

$$u_z = \frac{-Mx}{4\pi\mu r^3} + \frac{M}{16\pi\mu(1-\nu)} \left[\frac{x}{r^3} - \frac{3z^2x}{r^5} \right] = \frac{Mx}{16\pi\mu(1-\nu)r^3} \left[\frac{(4\nu-3)}{r^3} - \frac{3z^2}{r^2} \right]$$

20/20



2.



$$\text{Show that } u_x(0,0,0) = \frac{P(1-2\nu)}{8\pi\mu c}$$

The problem above was given as Mindlin's Problem Part II with the following

$$u = \bar{B} - \frac{1}{4(1-\nu)} \nabla \cdot (\mathbf{r} \cdot \nabla \bar{B} + \beta) \quad \text{where } \bar{B} = B_z \mathbf{e}_z \text{ and } B_y = 0; B_x = \frac{P}{4\pi\mu} \left(\frac{1}{R_1} + \frac{1}{R_2} \right)$$

$$B_z = \frac{Px}{2\pi\mu R_2} \left[\frac{1-2\nu}{R_2+z+c} - \frac{c}{R_2^2} \right]; \quad \beta = \frac{P(1-2\nu)x}{2\pi\mu(R_2+z+c)} \left[\frac{c}{R_2} - (1-2\nu) \right]$$

$$\text{and } R_1^2 = x^2 + y^2 + (z-c)^2 \quad R_2^2 = x^2 + y^2 + (z+c)^2 \quad \text{derived and defined}$$

$$\text{To get } u_x(x,y,z) \text{ we need } u_x = B_x - \frac{1}{4(1-\nu)} \frac{\partial}{\partial x} [x B_x + z B_z + \beta]$$

$$u_x = B_x - \frac{1}{4(1-\nu)} \left\{ x \frac{\partial B_x}{\partial x} + B_x + z \frac{\partial B_z}{\partial x} + \frac{\partial \beta}{\partial x} \right\}$$

$$\text{Now we need } \frac{\partial B_x}{\partial x} = \frac{P}{4\pi\mu} \frac{\partial}{\partial x} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) = \frac{P}{4\pi\mu} \left[-\frac{x}{R_1^3} - \frac{x}{R_2^3} \right]$$

$$\frac{\partial B_z}{\partial x} = \frac{P}{2\pi\mu R_2} \left[\frac{1-2\nu}{R_2+z+c} - \frac{c}{R_2^2} \right] = \frac{Px^2}{2\pi\mu R_2^3} \left[\frac{1-2\nu}{R_2+z+c} - \frac{c}{R_2^2} \right] + \frac{Px}{2\pi\mu R_2} \left[\frac{1-2\nu}{(R_2+z+c)^2} \frac{x}{R_2} \right.$$

$$\left. + \frac{c}{R_2^4} x \right]$$

$$\frac{\partial \beta}{\partial x} = \frac{P(1-2\nu)}{2\pi\mu(R_2+z+c)} \left[\frac{c}{R_2} - (1-2\nu) \right] = \frac{P(1-2\nu)x^3}{2\pi\mu(R_2+z+c)^2 R_2} \left[\frac{c}{R_2} - (1-2\nu) \right]$$

$$+ \frac{P(1-2\nu)x}{2\pi\mu(R_2+z+c)} \left[-\frac{c}{R_2^3} x \right]$$

$$\text{Now at } x=0, y=0, z=0 \quad R_1^2 = c^2 = R_2^2 \Rightarrow \text{then } \left. \frac{\partial B_x}{\partial x} \right|_{x=y=0} = 0;$$

$$\left. \frac{\partial B_z}{\partial x} \right|_{x=y=0} = \frac{P}{2\pi\mu c} \left[\frac{1-2\nu}{2c} - \frac{1}{c} \right] = \frac{-P(1+2\nu)}{4\pi\mu c^2}; \quad \left. \frac{\partial \beta}{\partial x} \right|_{x=y=0} = \frac{P(1-2\nu)}{4\pi\mu c} [2\nu]$$

$$B_x = \frac{P}{4\pi\mu} \left(\frac{c}{c} \right); \Rightarrow u_x = \frac{P}{2\pi\mu c} - \frac{1}{4(1-\nu)} \left[0 \cdot 0 + \frac{P}{2\pi\mu c} + 0 \cdot \frac{P(1-2\nu)}{4\pi\mu c^2} + \frac{P(1-2\nu)\nu}{2\pi\mu c} \right]$$

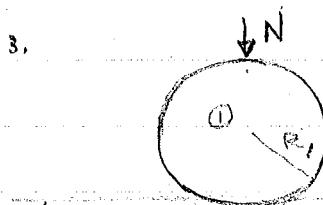
$$u_x = \frac{P}{2\pi\mu c} = \frac{1}{4(1-v)} \left[\frac{P}{2\pi\mu c} \{ 1 + (1-2v)v \} \right] = \frac{P}{2\pi\mu c} \left\{ 1 - \frac{1}{4(1-v)} [(1-v)(1+2v)] \right\}$$

$$= \frac{P}{2\pi\mu c} \left\{ 1 - \frac{1+2v}{4} \right\} = \frac{P}{2\pi\mu c} \cdot \frac{3-2v}{4} = \frac{(3-2v)P}{8\pi\mu c} \quad \checkmark$$

QED

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/ 20

20/10

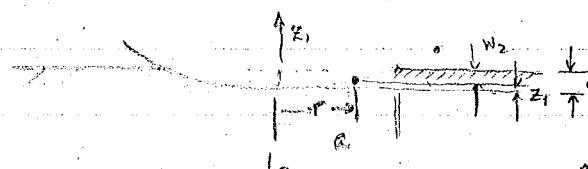


A rigid sphere implies no settlement here in the sphere i.e. $w_1 = 0$

$$w_2 \neq 0$$

$$\mu > 1 \quad R_2 = 0 \quad \Rightarrow \quad z_2 = 0$$

$$R_2 = \infty$$



If we assume that the contact area is circular with radius $a \ll R_1$ or R_2 and N is applied remotely and doesn't affect contact problem then

$$(1) \quad \boxed{w_2 = \alpha - z_1 = \alpha - \frac{r^2}{2R_1}} \quad \text{as shown in class. Now since } w_2 \text{ is the settlement then}$$

$$(2) \quad \boxed{w_2 = \frac{1-\nu}{2\pi\mu} \int_{-R_2}^{\infty} \int_{-a}^{a} p_2(v, \psi) dv d\psi}, \quad \text{In writing this we are assuming that the contact area is flat and to a first approximation this is correct. Now since } w_2 \text{ is proportional to } r^2$$

we can use the 3rd case of normal circular loads studied in class i.e

$$(3) \quad \boxed{p_2 = \frac{3N(a^2 - r^2)^{1/2}}{8\pi a^3} \Rightarrow \boxed{w_2 = \frac{3(1-\nu)N}{16a^3\mu} (2a^2 - r^2)} \quad (4)}$$

$$\text{thus equating (4) and (1) we find that } \alpha = \frac{3(1-\nu)N \cdot 2a^2}{16a^3\mu} = \frac{3(1-\nu)N}{8a\mu}$$

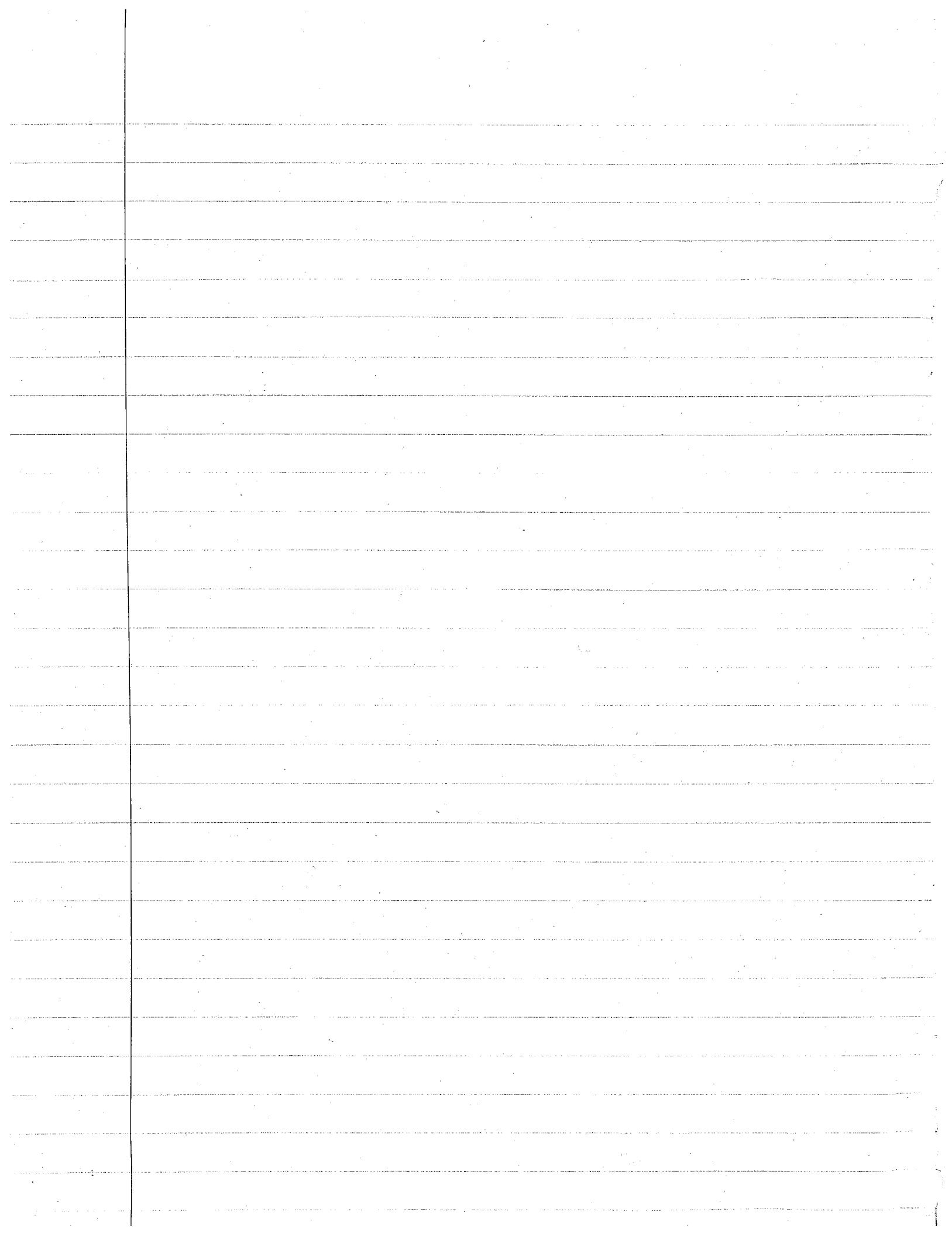
$$\text{and } \frac{1}{2R_1} = \frac{3(1-\nu)N}{16a^3\mu} \Rightarrow \boxed{a = \left(\frac{3(1-\nu)NR_1}{8\mu}\right)^{1/3}} \quad (5)$$

to find the maximum pressure we go to equation (3) and find that when

$$r=0 \quad \boxed{p_2 = p_{2\max} = \frac{3N}{2\pi a^2}} \quad \text{or if } \exists \text{ a passage } = \frac{N}{\pi a^2} \text{ then } p_{2\max} = \frac{3}{2} p_{2\min}$$

SUMMARY

- thus knowing R_1, μ, ν, N we can find the radius of the contact circle $a = \left[\frac{3(1-\nu)NR_1}{8\mu}\right]^{1/3}$
- then knowing μ, ν, N and a we can find the settlement $w_2 = \frac{3(1-\nu)N}{16a^3\mu} (2a^2 - r^2)$
- then knowing N and a we find that at $r=0$ $p_2 = p_{2\max} = \frac{3N}{2\pi a^2}$ at the interface



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Question	Score
1	18/20
2	12/20
3	14/20
4	
5	
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7	
8	
Total	44/60

read the
question

$$\frac{44}{60} \times 100 = 73$$

Name of student Cesar Levy

Date of examination 11 May 1979

Course ME 238 C

THE STANFORD UNIVERSITY HONOR CODE

- A. The Honor Code is an undertaking of the students, individually and collectively:
- (1) that they will not give or receive aid in examinations; that they will not give or receive unpermitted aid in class work, in the preparation of reports, or in any other work that is to be used by the instructor as the basis of grading;
 - (2) that they will do their share and take an active part in seeing to it that others as well as themselves uphold the spirit and letter of the Honor Code.
- B. The faculty on its part manifests its confidence in the honor of its students by refraining from proctoring examinations and from taking unusual and unreasonable precautions to prevent the forms of dishonesty mentioned above. The faculty will also avoid, as far as practicable, academic procedures that create temptations to violate the Honor Code.
- C. While the faculty alone has the right and obligation to set academic requirements, the students and faculty will work together to establish optimal conditions for honorable academic work.

I acknowledge and accept the Honor Code.

(Signed) Cesar Levy

*Interpretations and applications of the Honor Code
appear on the back cover of this examination book.*

what I didn't do!!

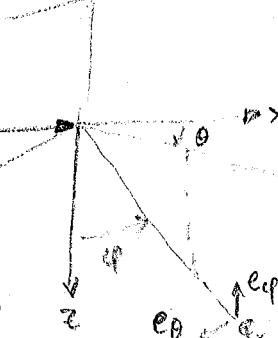
tractions on $z=0$ normal to $z=0$ is σ_{φ}
 $\Rightarrow \sigma_{\varphi} \cdot \mathbf{e} = \sigma_{\varphi r} \mathbf{e}_r + \sigma_{\varphi \theta} \mathbf{e}_{\theta} + \sigma_{\varphi \varphi} \mathbf{e}_{\varphi}$

but $\sigma_{\varphi r} = \sigma_{\varphi \theta} = \sigma_{\varphi \varphi} = 0$ everywhere
thus on plane of applied load traction
is zero, & since load is tangential this
is Bernoulli problem !!

$$x = \sin \varphi \cos \theta$$

$$\rightarrow T_{RR} = - \frac{3 P_x \sin \varphi \cos \theta}{4 \pi r^2}$$

$$T_{RR} = 0 \text{ when } \varphi = 0 \text{ or } \theta = 90^\circ$$



Cesar Levy

DIVISION OF APPLIED MECHANICS
DEPARTMENT OF MECHANICAL ENGINEERING
STANFORD UNIVERSITY

ME 238C Theory of Elasticity

Spring 1979

Midterm Examination

1. Consider the solution of the Kelvin problem and show that if Poisson's ratio is $1/2$, then this solution also provides the solution to the Boussinesq and Cerruti problems.
(Hint: evaluate components of stress in spherical coordinates)
2. In the Boussingesq problem, calculate and sketch the settlement, as well as the displacement of points along the line of action of the force in the direction of the force. ✓
3. An elastic sphere of radius b contains a fixed spherical inclusion of radius a and is subjected to a uniform external pressure p . Find the traction on the surface of the inclusion.

From HW #1 we found $\sigma_{pr} = \frac{P \sin \varphi}{8\pi r^2} \frac{(1-2\nu)}{1-\nu}$ $\sigma_{\theta\varphi} = \sigma_{r\theta} = 0$

$$\sigma_{rr} = \frac{P \cos \varphi}{4\pi r^2} \left(\frac{\nu-2}{1-\nu} \right) \quad \sigma_{\theta\theta} = \sigma_{\varphi\varphi} = \frac{P \cos \varphi}{8\pi r^2} \left(\frac{1-2\nu}{1-\nu} \right)$$

for Kelvin problem.

Now with $\nu = 1/2$ $\sigma_{pr} = \sigma_{\theta\varphi} = \sigma_{r\theta} = \sigma_{\theta\theta} = \sigma_{\varphi\varphi} = 0$

$$\sigma_{rr} = \frac{P \cos \varphi}{4\pi r^2} [-3] = -\frac{3P \cos \varphi}{4\pi r^2}$$

$\sigma_{rr} = 0$ when $\varphi = 90^\circ$

3. Since $B_z = \frac{1-2\nu}{\pi R} P_z$ and $\beta = \frac{(1-2\nu)(1-\nu)}{\pi \mu} P_z L(R+z)$

$$\sigma_{rr} = e_{rr} \sigma_{rr} = \nu^2 \theta \sin^2 \varphi \sigma_{rr} = -3P$$

$\sigma_{rr} = -\frac{3Pz}{4\pi r^3}$ * note that since problem in class was taken along z direction we find $P = P_z$.

where P is in any direction since $\nabla P = 0$ on $z=0$

we have

Since for boundary $B_x, B_y = 0$ $B_z = \frac{1-\nu}{\pi \mu} P_z$ $\beta = \frac{(1-2\nu)(1-\nu)}{\pi \mu} P_z L(R+z)$

$$\text{for } \nu = 1/2 \quad B_z = \frac{P_z}{2\pi \mu R}, \quad \beta = 0$$

Kelvin problem to rigid body rotation if $P_x, P_y = 0$ $B_z = \frac{P_z}{4\pi \mu r}$ $\beta = 0$

when $z \geq 0$ except at $r=0$ $\sigma_{rr}=0$ but all other stresses are

zero on plane $z=0$ this is same as Boussinesq problem

ie point loading in z dir on a half space

why is this equivalent to Cerruti?

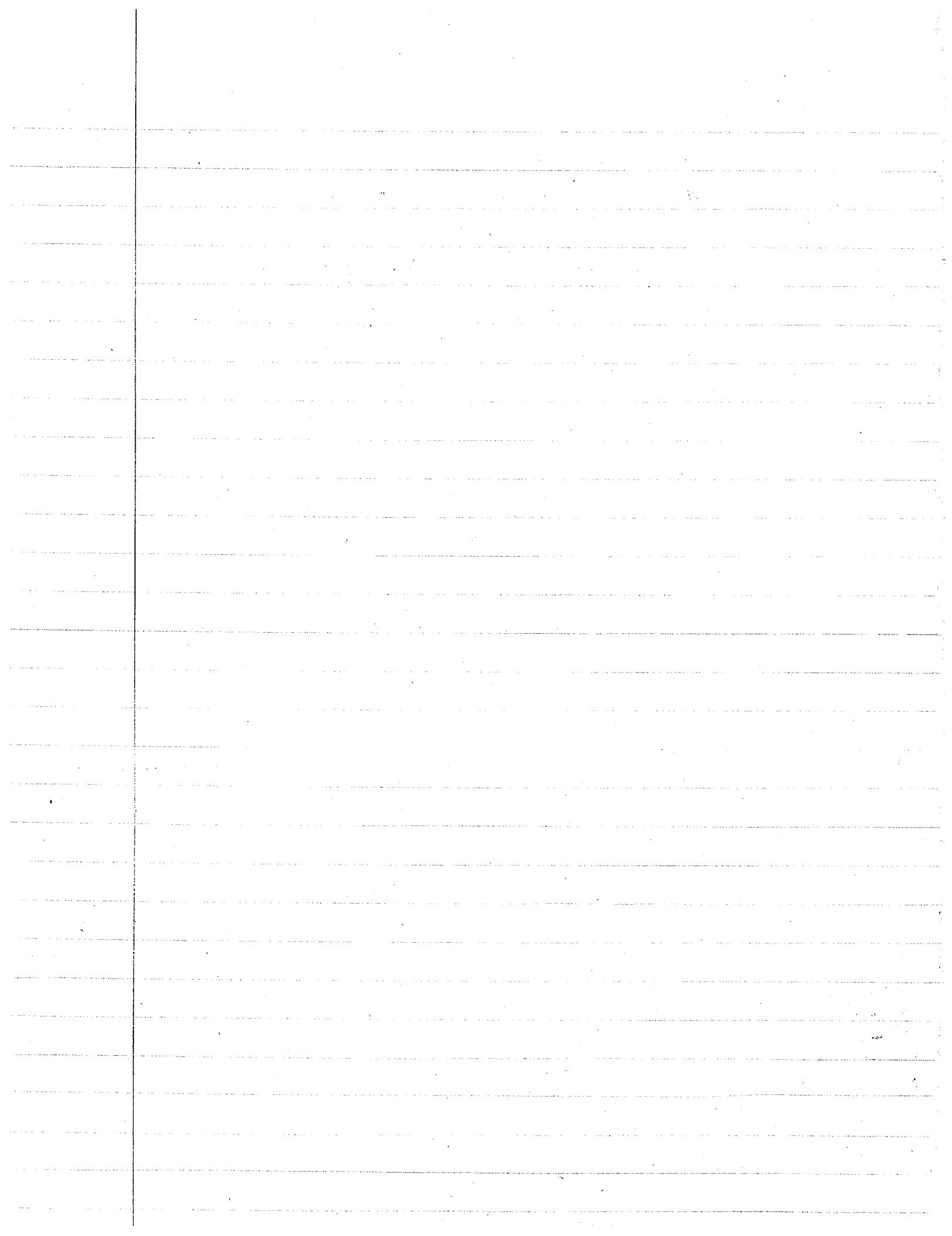
on what plane does it give zero traction?

-2

if we took Kelvin problem in x direct we would have found

$$\sigma_{rr} = -\frac{3Px}{4\pi r^3} \quad \& \text{all other } \sigma_{ij} = 0 \quad \text{where } P = P_x$$

this would have been equivalent to Cerruti problem if tangential load on a half space



$$2. \text{ If } \frac{P_z}{B_z} = W = u_z(x, y, 0)$$

$$\text{then } W = B_z P_z + \frac{1}{4(1-\nu)} \nabla (1r \cdot \nabla B + \beta)$$

$$= \frac{1-\nu}{4\pi\mu} \frac{P_z \cdot P_z}{R} - \frac{1}{4(1-\nu)} \nabla \left(\frac{1-\nu}{\pi\mu} \frac{P_z \cdot z}{R} + \frac{(1-2\nu)(1-\nu)}{\pi\mu} P_z \ln(R+2) \right)$$

$$= \frac{1-\nu}{\pi\mu} \frac{P_z \cdot P_z}{R^2} - \cancel{\frac{1}{4(1-\nu)}} \frac{P_z}{4\pi\mu} \nabla \left(\frac{z}{R} + (1-2\nu) \ln(R+2) \right)$$

$$\nabla \left(\frac{z}{R} + (1-2\nu) \ln(R+2) \right) = z \nabla \left(\frac{1}{R} \right) + \frac{P_z}{R^2} + (1-2\nu) \nabla \ln(R+2)$$

$$= -z \frac{R}{R^3} + \frac{P_z}{R^2} + (1-2\nu) \left[\frac{1}{R+2} \right] \text{ etc}$$

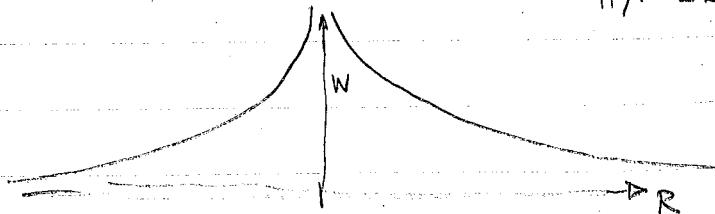
$$\text{for } u_z(x, y, z) = \frac{1-\nu}{\pi\mu} \frac{P_z}{R} + \left(-\frac{z^2}{R^3} + \frac{1}{R} + (1-2\nu) \frac{1}{R+2} \cdot \left(\frac{z}{R} + 1 \right) \frac{1}{R} \right) \frac{P_z}{4\pi\mu}$$

$$= \frac{1-\nu}{\pi\mu} \frac{P_z}{R} - \frac{P_z}{4\pi\mu} \left(\frac{(1-2\nu)}{R} + \frac{1}{R} - \frac{z^2}{R^3} \right)$$

$$= \left(\frac{2(1-\nu)}{R} - \frac{z^2}{R^3} \right)$$

$$(1-\nu) \frac{P_z}{\pi\mu} \left[\frac{1}{2R} \right] - \frac{P_z}{4\pi\mu} \left(-\frac{z^2}{R^3} \right)$$

$$@ z=0 \quad u_z(x, y, 0) = W = \frac{(1-\nu)}{\pi\mu} \frac{P_z}{2R} \quad R = \sqrt{x^2 + y^2}$$



along line of action
-g

also find $u_z(0, 0, z)$

$$u_z(0, 0, z) = \frac{(1-\nu)P_z}{\pi\mu} \left[\frac{1}{2z} \right] - \frac{P_z}{4\pi\mu} \left[-\frac{1}{2z} \right]$$

what's
I did?
do!!

$u_z(0, 0, z)$

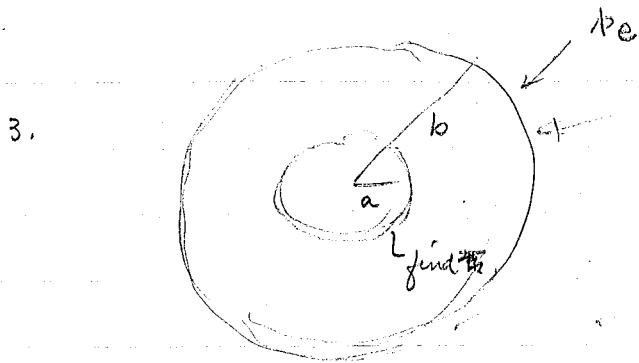
$$= \frac{P_z}{\pi\mu} \left\{ \frac{2(1-\nu)}{2 \cdot 2z} + \frac{1}{4z} \right\}$$

$$= \frac{P_z}{\pi\mu} \left\{ \frac{3-2\nu}{4z} \right\}$$

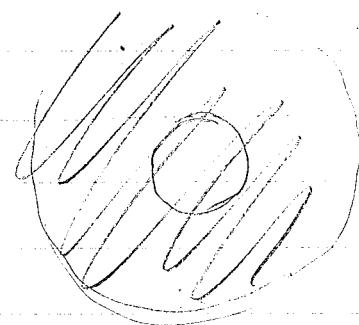
$$\begin{aligned}
 +p_i &= \frac{(p_i - p)(a^3 b^3)}{(a^3 - b^3)ab} + \frac{b^3 p - a^3 p_i}{a^3 b^3} \\
 &= \cancel{(p_i - p) b^3} \quad p_i \left(\frac{b^3 - a^3}{a^3 - b^3} \right) + \cancel{p_i \left(\frac{b^3 + a^3}{a^3 - b^3} \right)} = 0 \\
 &\quad + \cancel{p_3} - \cancel{p_3} \cancel{+ p_3} \\
 + 2p_i &= -p \quad \boxed{p_i = \cancel{-p}} = 0
 \end{aligned}$$

I can't find
the sign
somewhere

14/20

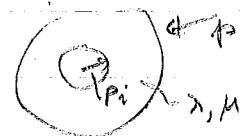


Since it's an inclusion. $u_1 \text{ sphere}_a = u_1 \text{ sphere}_b$ at interface = 0. rigid inclusion
fractions are continuous also



Let us use ~~as~~ concentric sphere

Soln.



+ spherical cavity
 P_{pi}
 $\lambda^*, \mu^* = \infty$

3. u_1 are ~~continuous~~
across bndry.
fractions are contin.

Spherical cavity

$$u_R = \frac{P_i a^3}{4\pi R^2} \quad u_\theta = u_\phi = 0$$

$$\sigma_{RR} = \frac{P_i a^3}{R^3} \quad \sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{P_i a^3}{2R^3} \quad \sigma_{\theta\phi} = \sigma_{\theta R} = \sigma_{\phi R} = 0$$

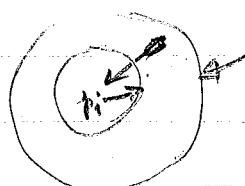
$$\tau_{rr} = \tau_{rr}$$

$$\tau_{\theta\theta} = \tau_{\theta\theta}$$

$$\tau_{\phi\phi} = \tau_{\phi\phi}$$

$$\tau_r$$

Sphere



$$\sigma_{RR} = \frac{(P_i - P) a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 P - a^3 P_i}{a^3 - b^3} = P_i$$

$$\sigma_{RR} = \frac{(P_i - P) a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 P - a^3 P_i}{a^3 - b^3} \quad u_r = ?$$

$$\sigma_{\theta\phi} = \sigma_{\theta\theta} = \frac{a^3 b^3 (P_i - P)}{2(a^3 - b^3) R^3} + \frac{b^3 P - a^3 P_i}{a^3 - b^3} \quad \text{equate } u_r \text{ incl to } u_r / \text{bph.}$$

$$\tau_r = -\tau_{rr}$$

\therefore I expect symmetry \Rightarrow

$$\tau_\theta = -\tau_{R\theta}$$

$$+ P_i = (P_i - P)(a^3 b^3) + \frac{b^3 P - a^3 P_i}{a^3 - b^3}$$

$$\tau_\phi = -\tau_{R\phi}$$

$$(a^3 - b^3) R^3 \quad a^3 - b^3$$

$$- S(1+\nu)(ab^3) - a^4 S(2-4\nu)$$

$$pab^3(1+\nu) + pab^3(2-4\nu) = 3(1-\nu)pab^3$$

$$u_r \text{ of cavity} \Big|_{r=a} = u_r \text{ of hollow sphere} \Big|_{r=a}$$

and Π_r are continuous across boundary in this case $\Pi_p = \Pi_{rr}$ only

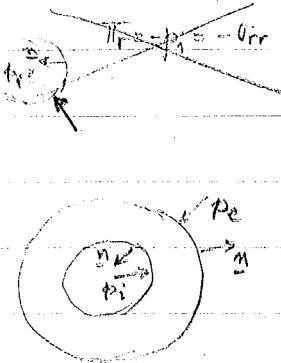
$$\Pi_0 = 0 \quad \Pi_p = 0$$

for a spherical cavity under p_i we showed

$$u_r = -\frac{\rho_i a^3}{8\mu R^2} = -\frac{S a^3}{R^3} \quad \text{and} \quad \sigma_{rr} = +\frac{\rho_i a^3}{8\mu R^2} = \frac{S a^3}{R^3}$$

for a hollow sphere with $p_i = S$ and $\sigma_{rr} \Big|_{r=a} = \sigma_{rr} \Big|_{r=b}$

$$\sigma_{rr} = (S - p) \frac{a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 p - a^3 S}{a^3 - b^3}$$



$$\text{Now } \sigma_{rr} \text{ on } r=a \quad \sigma_{rr} = (S - p) \frac{b^3}{a^3 - b^3} + \frac{b^3 p - a^3 S}{a^3 - b^3} = -S \quad \text{w}$$

Now we will find the displacement of $r=a$ in the hollow sphere

$$\epsilon_{rr} = \frac{\partial u_r}{\partial r} = \frac{\sigma_{rr}}{E} = \frac{1}{E} (\sigma_{00} + \sigma_{pp})$$

$$\text{and } \sigma_{00} = \sigma_{pp} = \frac{a^3 b^3 (p_e - S)}{2(a^3 - b^3) R^3} + \frac{b^3 p_e - a^3 S}{a^3 - b^3}$$

$$\therefore \sigma_{00} + \sigma_{pp} = \frac{a^3 b^3 (p_e - S)}{(a^3 - b^3) R^3} + 2 \left(\frac{b^3 p_e - a^3 S}{a^3 - b^3} \right)$$

$$\therefore \epsilon_{rr} = \frac{(S - p)}{E} \frac{a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 p - a^3 S}{E (a^3 - b^3)} = \frac{1}{E} \frac{a^3 b^3 (p_e - S)}{(a^3 - b^3) R^3} = \frac{2}{E} \left(\frac{b^3 p_e - a^3 S}{a^3 - b^3} \right)$$

$$\frac{\partial u_r}{\partial r} = \frac{S - p}{E} \frac{a^3 b^3}{(a^3 - b^3) R^3} \left[1 + \frac{1}{2} \nu \right] + \frac{b^3 p - a^3 S}{E (a^3 - b^3)} \left[1 - 2\nu \right]$$

$$\therefore u_r = \frac{S - p}{E} \frac{a^3 b^3}{(a^3 - b^3) R^2} \left[\frac{1+\nu}{2} \right] + \frac{b^3 p - a^3 S}{E (a^3 - b^3)} [1 - 2\nu] R + C$$

$$u_r \Big|_{r=a} = \frac{S - p}{E} \left[\frac{ab^3}{a^3 - b^3} \right] \left(\frac{1+\nu}{2} \right) + \frac{ab^3 p - a^4 S}{E (a^3 - b^3)} (1 - 2\nu)$$

$$= -\frac{S - p}{2\mu} \left(\frac{ab^3}{a^3 - b^3} \right) \frac{1+\nu}{1+\nu} \frac{ab^3 p - a^4 S}{a^3 - b^3} \frac{2 - 4\nu}{4\mu (1+\nu)} = \frac{3(1-\nu) p a b^3 - S [(1+\nu) a b^3 + a^4 (2-4\nu)]}{4\mu (1+\nu) (a^3 - b^3)}$$

Since problem is symmetric all around
C is a part
 $\epsilon_{rr} = 0$
since this rigid body translation

for rigid inclusion $u_r = 0$

$r=a$

$$u_r = \frac{3pab^3(1-\nu) - S [ab^3(1+\nu) + a^4(2-4\nu)]}{4\mu(1+\nu)(a^3-b^3)} = 0$$

$$\therefore \frac{3pab^3(1-\nu)}{4\mu(1+\nu)(a^3-b^3)} = \frac{S [ab^3(1+\nu) + a^4(2-4\nu)]}{4\mu(1+\nu)(a^3-b^3)}$$

$$\therefore S = \frac{3pab^3(1-\nu)}{ab^3(1+\nu) + 2a^4(1-2\nu)}$$

1. Bessmey Problem: Normal Point based on a half-space

Cerruti Problem: Show point load on a half-space

Show Kelvin gives solution to Bessmey & Cerruti problems.

$$\delta^A \psi = 0$$

~~3) Bessy~~ This will be true if the Kelvin problem gives no traction.

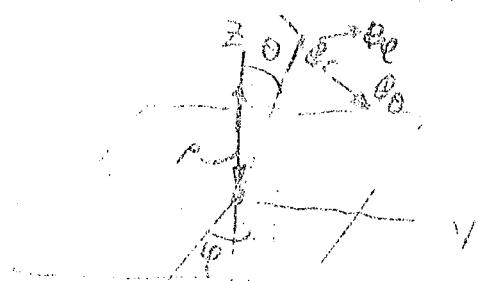
Effect on a plane in which the load lies and in a plane normal to

the load. Also for bessmey

In spherical coordinates, referring to parts of Hestad and setting

$$\psi = \theta \sin \phi \cos \theta = \theta \cos \phi = \theta \psi = 0$$

$$\delta \psi = -\frac{3R \cos \theta}{4\pi R^2}$$



The traction on a plane whose normal

is $\theta \psi$ is $\theta \psi \delta \psi \sin \theta \cos \theta$.

Taking $\theta = \frac{\pi}{2} - \frac{\theta}{2}$, the plane contains the load

and therefore we have proven the Cerruti problem is solved.

Taking $\theta = \frac{\pi}{2} - \frac{\theta}{2}$, this plane is normal to the load
and we've proven the Bessmey problem is solved.

It is easily verified that the integrated traction conditions are also satisfied in both cases.

2

ME 238C Matheron Spring 1979

Boussinesq problem of settlement = $\delta z/z=0$

Given in notes: $B_z = \frac{1-\nu}{\pi(1-\nu)} R^2 P$, $B_x = B_y = 0$

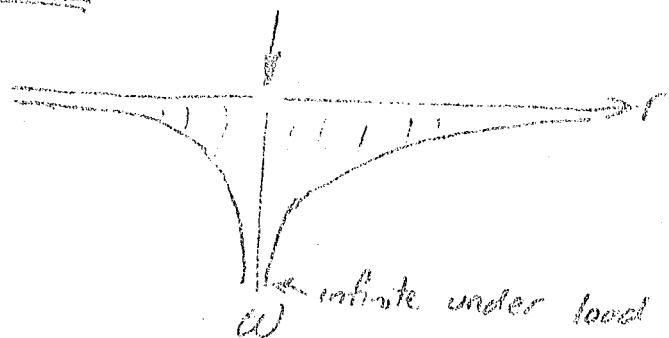
$$B = \frac{(1-2\nu)(1-\nu)}{\pi(1-\nu)} P h_0 (R+z)$$

$$w = B - \frac{P}{4\pi(1-\nu)} [v - B + \beta] = B_z \delta z - \frac{1}{4\pi(1-\nu)} v (2B_z + \beta)$$

$$\begin{aligned} B_z &= B_z - \frac{1}{4\pi(1-\nu)} \frac{\partial}{\partial z} (2B_z + \beta) = B_z - \frac{1}{4\pi(1-\nu)} (B_z + \frac{2-2\nu}{\nu} B_z + \frac{\beta}{\nu}) \\ &= \frac{P}{4\pi h_0} \left[\frac{3-4\nu}{R} + \frac{z^2}{R^2} - \frac{1-2\nu}{R} \right] = \frac{P}{4\pi h_0} \left[\frac{2(1-\nu)}{R} + \frac{z^2}{R^2} \right] \end{aligned}$$

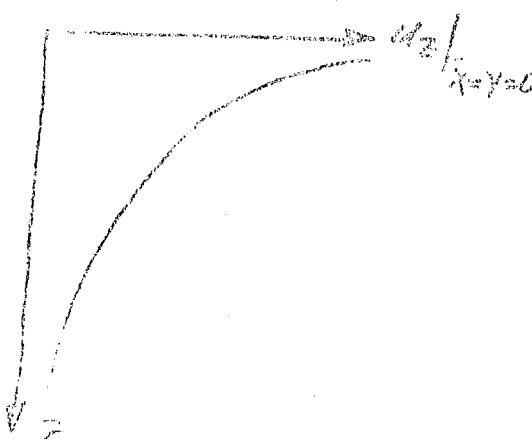
settlement $w = \delta z/z=0$

$$= \frac{P}{4\pi h_0} \frac{(1-\nu)}{R} \quad \text{where } r = \sqrt{x^2+y^2}$$

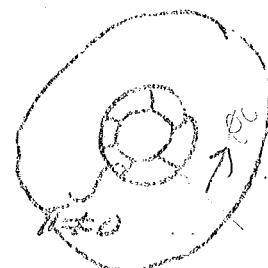
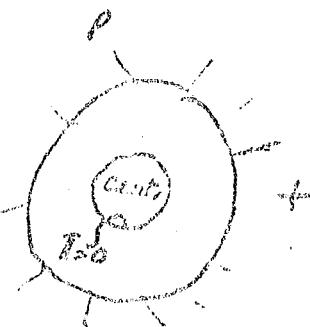
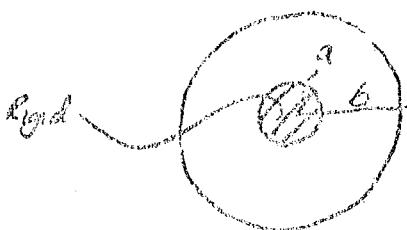


Q) on line of force (Z-axis) $x=y=0$, $R^2=z$

$$\Rightarrow \delta z = \frac{P}{4\pi h_0} \left(\frac{3-2\nu}{z} \right)$$



3) Superposition:



(A)

(B)

$$\text{Interface condition: } \frac{\partial u_r + u_B}{r-a} = 0$$

A.) determine displacement at $r=a$ if no inclusion were present

To note, it is shown solution to spherical cavity under internal and external pressure is given by superposition of center of dilation of strength A and hydrostatic tension B , where A, B were found to be

$$A = \frac{(4-2V) \alpha^3 b^3 (P_e - P_i)}{8(1-V)}$$

$$B = \frac{63 \pi e^{-\alpha^3 b^3}}{a^2 b^3}$$

$$\text{displacement: Center of dilation, } u_r = \frac{A}{4\pi(1-V)^2} \quad u_\theta = u_\phi = 0$$

$$\text{Hydrostatic tension, } E\epsilon_r = \frac{E}{2}(A - 4\alpha^3 b^3) = \frac{E}{2}(1-2V) = \frac{3(1-2V)}{2(1-V)^2}$$

$$= E_0 \epsilon_r = E_0 \epsilon_{pp}$$

$$\text{Integrate strain-displacement relation } \Rightarrow u_r = \frac{3(1-2V)}{2(1-V)^2} R - B_0 a^2 \theta = 0$$

$$\text{Superposing, } u_r = \frac{R}{4(1-V)^2} + \frac{3(1-2V)}{2(1-V)^2} R$$

Subst. for A & B their values when $B=B_0$, $P_i=0$

$$\Rightarrow u_r = \frac{a^3 b^3 P}{24(1-V)^2} + \frac{(1-2V) b^3 P}{24(1-V) \alpha^3 b^3} R$$

$$u_r/r = a^2 = \frac{a^6 b^3 P}{24(1-V)^2 \alpha^2} + (1-2V) \frac{a^3 b^3 P}{24(1-V) \alpha^3 b^3} = \frac{3P a^6 (1-2V)}{24(1-V) \alpha^3 b^3}$$

ME 238C Problem - Spring 1979

B) determine traction at sea needed to push the interface back to where it belongs

for $P_0=0$, $\sigma_r = -T_r$, this is the spherical cavity in infinite space

$$\frac{4\sigma_r}{T_r} = a^{-2} = \frac{ab^3 T_r}{4\pi(1+2\nu)} + \frac{T_r a^4 (1-2\nu)}{(a^2-b^2) 2\pi(1+\nu)}$$

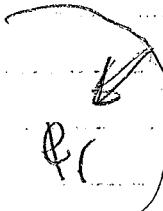
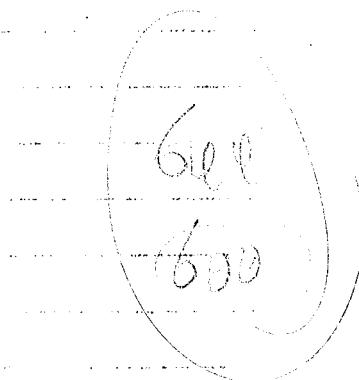
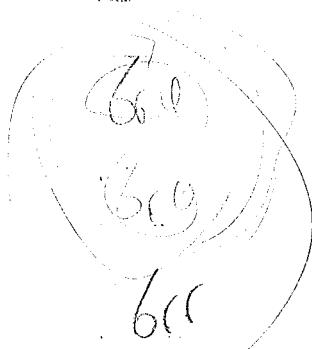
$$= T_r \left[\frac{ab^3(1+\nu) + 2a^4(1-2\nu)}{4\pi(1+\nu)(a^2-b^2)} \right]$$

$$(2) \quad \Pi_{r2} = \Pi_{r1} = -\sigma_r$$

$$\frac{3\rho ab^3(1-\nu)}{4\pi(1+\nu)(a^2-b^2)}$$

$$\left. \begin{array}{l} T_r = \frac{3\rho ab^3(1-\nu)}{a^2(1+\nu) + 2a^4(1-2\nu)} \\ \text{and } T_\theta = T_\theta = 0 \end{array} \right\}$$

$$\frac{3\rho b^3(1-\nu)}{b^3(1+\nu) + 2(1-\nu)a^3}$$



Π^θ

$$T_\theta = \phi_\theta \cdot \Pi_r = \phi_\theta \cdot \delta \cdot \phi_r$$

DIVISION OF APPLIED MECHANICS
DEPARTMENT OF MECHANICAL ENGINEERING
STANFORD UNIVERSITY

Spring 1979

ME 238C Theory of Elasticity

Problem Set No. 1

1. Evaluate the Cartesian components of stress for the Kelvin problem.
2. Evaluate the components of traction on a spherical surface of arbitrary radius with its origin at the point at which the concentrated force is acting in the Kelvin problem.
3. Show that the system of tractions evaluated in (2) above is statically equivalent to a single force.
4. Evaluate the normal strain along the line of action of the force in the Kelvin problem. *What does this mean?*
5. Evaluate the components of displacement in spherical coordinates for the Kelvin problem.
6. Evaluate the spherical components of stress for the Kelvin problem and show in particular that the components $\sigma_{\theta\phi}$ and $\sigma_{\phi r}$ vanish.



$$\rightarrow M = Pe, P$$



$$v = \frac{\lambda}{2(\lambda+\mu)} \quad 1-v = \frac{2(\lambda+\mu)-\lambda}{2(\lambda+\mu)} = \frac{\lambda+2\mu}{2(\lambda+\mu)} \quad \frac{1}{4(1-v)} = \frac{2(\lambda+\mu)}{2\lambda(\lambda+2\mu)}$$

$$1 - \frac{1}{4(1-v)} = \frac{2(\lambda+2\mu) - (\lambda+\mu)}{2(\lambda+2\mu)} = \frac{\lambda+3\mu}{2(\lambda+2\mu)}$$

$$\therefore \frac{P_i}{8\pi\mu r} \frac{\lambda+3\mu}{(\lambda+2\mu)} \quad \text{if } P_1=0=P_2 \quad P_3=P_3$$

$$\therefore u_i = \frac{\lambda+3\mu}{8\pi\mu(\lambda+2\mu)} \frac{P_3}{r} \delta_{i3} + \frac{\lambda+\mu}{8\pi\mu(\lambda+2\mu)} \frac{x_3 x_i}{r^3} P_3 \quad \text{let } \frac{(\lambda+\mu)P}{8\pi\mu(\lambda+2\mu)}$$

$$u_i = C \left[\frac{(\lambda+3\mu)}{(\lambda+\mu)} \frac{\delta_{i3}}{r} + \frac{x_3 x_i}{r^3} \right] \quad \frac{\partial}{\partial x_k} \left(\frac{1}{r} \right) = -\frac{1}{r^3} x_k \quad \frac{\partial r}{\partial x_k} = \frac{x_k}{r}$$

$$u_{i,k} = C \left[\frac{\lambda+3\mu}{(\lambda+\mu)} \delta_{i3} \left(-\frac{x_k}{r^3} \right) + \frac{\delta_{3k} x_i}{r^3} + \frac{x_3 \delta_{ik}}{r^3} - \frac{3x_3 x_i x_k}{r^5} \right]$$

$$u_{k,i} = C \left[\frac{\lambda+3\mu}{\lambda+\mu} \delta_{k3} \left(-\frac{x_i}{r^3} \right) + \frac{\delta_{3i} x_k}{r^3} + x_3 \frac{\delta_{ki}}{r^3} - \frac{3x_3 x_i x_k}{r^5} \right]$$

$$\text{add } = C \left\{ \frac{\lambda+3\mu}{\lambda+\mu} \left[-\frac{x_k \delta_{i3}}{r^3} - \frac{x_i \delta_{k3}}{r^3} \right] + \left[\frac{\delta_{3k} x_i}{r^3} + \frac{\delta_{3i} x_k}{r^3} + \frac{2x_3 \delta_{ik}}{r^3} - \frac{6x_3 x_i x_k}{r^5} \right] \right\}$$

$$u_{i,i} = C \left[\frac{\lambda+3\mu}{\lambda+\mu} \delta_{i3} \left(-\frac{x_i}{r^3} \right) + \frac{\delta_{3i} x_i}{r^3} + \left(\frac{\delta_{ii} x_3}{r^3} - \frac{3x_3 x_i x_i}{r^5} \right) \right]$$

$$= C \left[\frac{\lambda+3\mu}{\lambda+\mu} \left(-\frac{\delta_{i3} x_i}{r^3} \right) + \frac{\delta_{3i} x_i}{r^3} + \frac{3x_3}{r^3} - \frac{3x_3 r^2}{r^5} \right]$$

$$= C \left[\frac{\delta_{3i} x_i}{r^3} \left\{ 1 - \frac{\lambda+3\mu}{\lambda+\mu} \right\} \right] = C \left[\frac{-2\mu}{\lambda+\mu} \right] \frac{\delta_{3i} x_i}{r^3}$$

$$\therefore \sigma_{ij} = C \lambda \left(-\frac{2\mu}{\lambda+\mu} \right) \left[\frac{\delta_{3k} x_k}{r^3} \right] \delta_{ij} + \mu C \left\{ \frac{\lambda+3\mu}{\lambda+\mu} \left[\frac{-x_j \delta_{i3} + x_i \delta_{j3}}{r^3} \right] + \left[\frac{\delta_{3k} x_i}{r^3} + \frac{\delta_{3i} x_k}{r^3} + \frac{2x_3 \delta_{ij}}{r^3} - \frac{6x_3 x_i x_j}{r^5} \right] \right\}$$

$$= C \left\{ \frac{x_3}{r^3} \left[\frac{-2\lambda\mu + 2\mu}{\lambda+\mu} \right] \delta_{ij} - \frac{x_j \delta_{i3}}{r^3} \left[\frac{\mu\lambda + 3\mu^2}{\lambda+\mu} + \mu \right] - \frac{x_i \delta_{j3}}{r^3} \left[\underbrace{\frac{\mu\lambda + 3\mu^2}{\lambda+\mu} - \mu}_{\frac{2\mu^2}{\lambda+\mu}} \right] - \mu \frac{6x_3 x_i x_j}{r^5} \right\}$$

$$\begin{aligned}
 \int T_3 r^2 \sin\varphi d\varphi d\theta &= -\frac{2C\mu^2}{\lambda+\mu} \iint_0^{2\pi} \sin\varphi d\varphi d\theta = 6C\mu \int_0^{2\pi} \int_0^\pi \cos^2\varphi \sin\varphi d\varphi d\theta \\
 &+ \frac{4\pi C\mu^2}{\lambda+\mu} \left[\cos\varphi \right]_0^{\pi} = 6C\mu \cdot 2\pi \left[\frac{\cos^3\varphi}{3} \right]_0^{\pi} \\
 &- \frac{8\pi C\mu^2}{\lambda+\mu} = -4C\mu \cdot 2\pi \\
 &- 8C\mu\pi \\
 &- 8\pi C\mu \left[\frac{\mu}{\lambda+\mu} + 1 \right] \\
 -8\pi C\mu \cdot \frac{\lambda+2\mu}{\lambda+\mu} &= -\frac{8\pi C\mu (\lambda+2\mu)}{\lambda+\mu} \cdot \frac{(\lambda+\mu)P}{8\pi C\mu (\lambda+2\mu)} = P.
 \end{aligned}$$

$$\begin{aligned}
 5. \quad U_i = U_i e_i &= U_r e_r + U_\theta e_\theta + U_\varphi e_\varphi \\
 &= U_x e_x + U_y e_y + U_z e_z \\
 \text{then } U_r &= U \cdot e_r \quad U_\theta = U \cdot e_\theta \quad U_\varphi = U \cdot e_\varphi \\
 U_r &= \frac{Ce\cos\varphi}{r} \quad U_\theta = 0 \quad U_\varphi = 0
 \end{aligned}$$

$$\begin{aligned}
 e_r &= \begin{pmatrix} \cos\theta \sin\varphi & \sin\theta \sin\varphi & \cos\varphi \\ \cos\theta \cos\varphi & \sin\theta \cos\varphi & -\sin\varphi \\ -\sin\theta & \cos\theta & 0 \end{pmatrix} \begin{pmatrix} e_1 \\ e_2 \\ e_3 \end{pmatrix} \\
 e_\varphi &= \begin{pmatrix} -\sin\theta \sin\varphi \\ \cos\theta \sin\varphi \\ \cos\theta \end{pmatrix} \\
 e_\theta &= \begin{pmatrix} \cos\theta \sin\varphi \\ \cos\theta \cos\varphi \\ -\sin\theta \end{pmatrix}
 \end{aligned}$$

$$6. \quad \text{Using sohleinko (fig 184)} \quad \overline{e_{rr}} = \frac{\partial u_r}{\partial r} \quad \overline{e_{\theta\theta}} = \overline{e_{\varphi\varphi}} = \frac{u_r}{r} \quad \overline{e_{r\varphi}} = \overline{e_{r\theta}} = \overline{e_{\theta\varphi}} = 0$$

$$\overline{e_{rr}} = \frac{\lambda-2\mu}{r^2} C \cos\varphi \quad \overline{e_{\theta\theta}} = \overline{e_{\varphi\varphi}} = \frac{\lambda+2\mu}{r^2} C \cos\varphi \quad \overline{e_{r\theta}} = \overline{e_{r\varphi}} = \overline{e_{\theta\varphi}} = 0$$

$$\begin{aligned}
 4. \quad \text{hermann said find } \frac{\partial u_3}{\partial z} & \quad e_3 = u_{3,3} = C \left[-\frac{\lambda+3\mu}{\lambda+\mu} \frac{x_3}{r^3} + \frac{x_3}{r^3} + \frac{x_3}{r^3} - \frac{3x_3^3}{r^5} \right] \\
 & \quad \frac{2x_3}{r^3} \\
 & \quad C \left[\frac{x_3}{r^3} \left(\frac{2\lambda+2\mu-\lambda-3\mu}{\lambda+\mu} \right) - \frac{3x_3^3}{r^5} \right] \\
 & \quad \frac{Cx_3}{r^3} \left[\frac{\lambda-\mu}{\lambda+\mu} - \frac{3(x_3)^2}{r^2} \right]
 \end{aligned}$$

$$\therefore \sigma_{ij} = \frac{2C}{r^3}\mu \left\{ (x_3\delta_{ij} - x_j\delta_{i3} - x_i\delta_{j3}) \frac{\mu}{\lambda + \mu} - \frac{3x_3x_ix_j}{r^2} \right\}$$

$$\sigma_{11} = \frac{2C}{r^3}\mu \left\{ x_3 \frac{\mu}{\lambda + \mu} - 3x_3 \left(\frac{x_1}{r} \right)^2 \right\}; \quad \sigma_{31} = \sigma_{13} = \frac{2C}{r^3}\mu \left\{ (-x_1) \frac{\mu}{\lambda + \mu} - \frac{3x_3x_1x_3}{r^2} \right\}$$

$$\sigma_{21} = \sigma_{12} = \frac{2C}{r^3}\mu \left(-\frac{3x_3x_1x_2}{r^2} \right); \quad \sigma_{22} = \frac{2C}{r^3}\mu \left\{ x_3 \frac{\mu}{\lambda + \mu} - 3x_3 \left(\frac{x_2}{r} \right)^2 \right\}; \quad \sigma_{23} = \sigma_{32} = \frac{2C}{r^3}\mu \left\{ -\frac{x_2\mu}{\lambda + \mu} - \frac{3x_3^2x_2}{r^2} \right\}$$

$$\sigma_{33} = \frac{2C}{r^3}\mu \left\{ (x_3 - 2x_3) \frac{\mu}{\lambda + \mu} - 3x_3 \left(\frac{x_3}{r^2} \right)^2 \right\} = -\frac{2C}{r^3}\mu x_3 \left\{ 3 \left(\frac{x_3}{r} \right)^2 + \frac{\mu}{\lambda + \mu} \right\}$$

2. $T_i = \sigma_{ij}n_j$ now ~~$\sum n_j = 0$~~ $= \frac{x_j \phi_j}{r} = n_j \phi_j$

$$T_1 = \sigma_{11}n_1 + \sigma_{12}n_2 + \sigma_{13}n_3 = -\frac{2C}{r^3}\mu x_3 \left[3 \left(\frac{x_1}{r} \right)^2 + \frac{\mu}{\lambda + \mu} \right] \frac{x_1}{r} - \frac{2C}{r^3}\mu \left(3x_3x_1x_2 \right) \frac{x_2}{r} - \frac{2Cx_1\mu}{r^3} \left\{ 3 \left(\frac{x_3}{r} \right)^2 + \frac{\mu}{\lambda + \mu} \right\} x_3$$

$$= -\frac{2C}{r^4}\mu \left\{ 3 \left(\frac{x_1}{r} \right)^2 x_3 x_1 - \frac{x_3 x_1 \mu}{\lambda + \mu} + 3x_3 x_1 \left(\frac{x_2}{r} \right)^2 + 3 \left(\frac{x_3}{r} \right)^2 x_1 x_3 + \frac{\mu}{\lambda + \mu} x_1 x_3 \right\}$$

$$-\frac{2C}{r^4}\mu \left\{ 3x_3 x_1 \right\}$$

$$T_i = \sigma_{ij}n_j = \frac{2C}{r^4}\mu \left\{ (x_3 x_j \delta_{ij} - x_j x_j \delta_{i3} - x_i x_j \delta_{j3}) \frac{\mu}{\lambda + \mu} - \frac{3x_3 x_i x_j}{r^2} \right\}$$

$$\text{if } i, j \neq 3 \quad -\frac{6Cx_3x_i\mu}{r^4} = T_i$$

$$T_3 = \frac{2C}{r^4}\mu \left\{ \cancel{(x_3^2 - r^2 - x_3^2)} - \frac{x_3^2}{\lambda + \mu} - \frac{3x_3^2}{r^2} \right\} = -\frac{2C}{r^4}\mu \left\{ \frac{x_3^2\mu}{\lambda + \mu} + 3x_3^2 \right\}$$

$$= -\frac{2C\mu^2}{\lambda + \mu r^2} - \frac{6C\mu x_3^2}{r^4} \quad w/r = a$$

$$3. \int_S T_i \, dS = \iint_0^{2\pi} T_i r \sin\varphi \, d\varphi \, d\theta = -\frac{6Cx_3x_i\mu}{r^4a^2} \int_0^{2\pi} \sin\varphi \, d\varphi \, d\theta$$

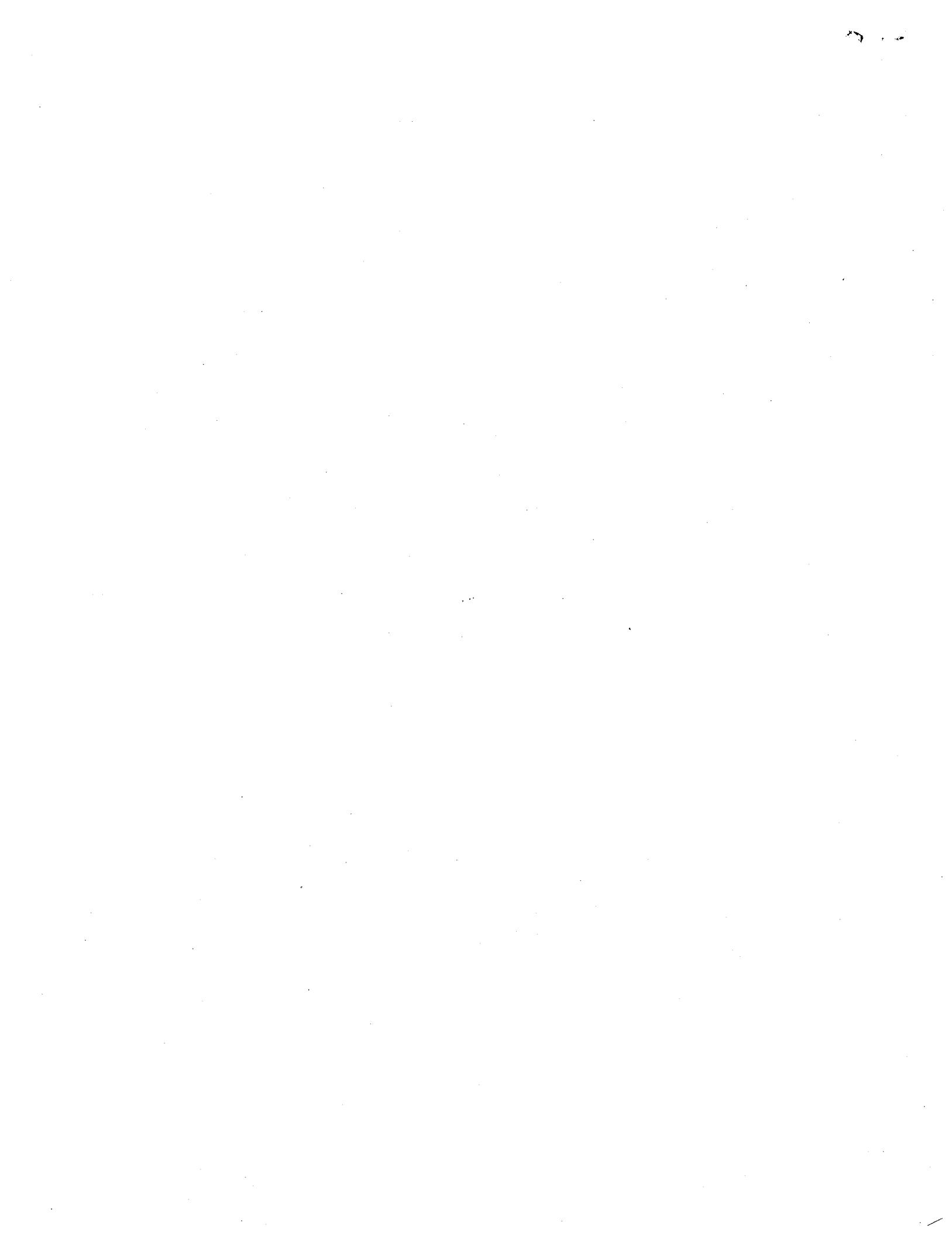
$$= -6C \left(\frac{x_3}{a} \right) \cdot \left(\frac{x_i}{a} \right) \mu$$

$$= -6C \cancel{\mu} \cos\varphi \cdot \cancel{a \cos\theta \sin\varphi} \left\{ \sin\varphi \, d\varphi \, d\theta \right\}$$

$$= -6C\mu \cos\varphi \left(1 - \frac{\cos 2\varphi}{2} \right) \left\{ \sin\theta \right\} d\varphi \, d\theta = 0$$

$$\int_0^{2\pi} \cancel{\mu \cos\theta \, d\theta} = 0 \quad \text{for } i=1,2$$

Go to back of 1



Kelvin Problem in cartesian

Problem Sub #1

$$1. \text{ if } \nu = \frac{\lambda}{2(\lambda+\mu)} \text{ then } \frac{1}{4(1-\nu)} = \frac{\lambda+\mu}{2(\lambda+2\mu)} \quad \text{and} \quad 1 - \frac{1}{4(1-\nu)} = \frac{\lambda+3\mu}{2(\lambda+\mu)}$$

thus

5/60

$$u_i = \frac{\lambda+3\mu}{8\pi\mu(\lambda+2\mu)} B_i = \frac{\lambda+\mu}{8\pi\mu(\lambda+2\mu)} x_j B_{j,i} \quad \text{using } B_i = \frac{P_i}{4\pi\mu r^3}$$

then with $P_1 = P_2 = 0, P_3 = P$

$$u_i = \frac{\lambda+3\mu}{8\pi\mu(\lambda+2\mu)} \frac{P \delta_{i3}}{r} + \frac{\lambda+\mu}{8\pi\mu(\lambda+2\mu)} \frac{x_2 x_i}{r^3} P = C \left[\frac{\lambda+3\mu}{\lambda+\mu} \frac{\delta_{i3}}{r} + \frac{x_2 x_i}{r^3} \right] \text{ where } C = \frac{(\lambda+\mu)P}{8\pi\mu(\lambda+2\mu)}$$

$$\text{Now } u_{ij} = C \left[- \frac{\lambda+3\mu}{\lambda+\mu} \frac{x_j}{r^3} \delta_{i3} + \frac{\delta_{3j} x_i}{r^3} + \frac{\delta_{ij} x_3}{r^3} - \frac{3x_3 x_i x_j}{r^5} \right]$$

also

$$u_{ii} = C \left[- \frac{\lambda+3\mu}{\lambda+\mu} \frac{x_i \delta_{i3}}{r^3} + \frac{\delta_{3i} x_i}{r^3} + \frac{(\delta_{ii} x_3 - 3x_3 x_i x_i)}{r^5} \right] = C \left(\frac{-2\mu}{\lambda+\mu} \right) \frac{\delta_{ii} x_i}{r^3} = \frac{-2\mu C x_3}{(\lambda+\mu) r^3}$$

$$\text{thus } \sigma_{ij} = \lambda u_{k,k} \delta_{ij} + \mu(u_{ij,j} + u_{ji,i}) = \frac{2C\mu}{r^3} \left\{ (x_3 \delta_{ij} - x_j \delta_{i3} - x_i \delta_{j3}) \frac{\mu}{\lambda+\mu} - \frac{3x_3 x_i x_j}{r^2} \right\} \text{ after combining terms and simplifying.}$$

$$\text{Thus } \sigma_{11} = -\frac{2C\mu x_3}{r^3} \left\{ 3 \left(\frac{x_1}{r} \right)^2 - \frac{\mu}{\lambda+\mu} \right\}; \quad \sigma_{12} = -\frac{6C\mu}{r^2} \left(x_1 x_2 x_3 \right); \quad \sigma_{13} = -\frac{2C\mu x_1}{r^3} \left\{ 3 \left(\frac{x_2}{r} \right)^2 + \frac{\mu}{\lambda+\mu} \right\}$$

$$\sigma_{22} = -\frac{2C\mu x_3}{r^3} \left\{ 3 \left(\frac{x_2}{r} \right)^2 - \frac{\mu}{\lambda+\mu} \right\}; \quad \sigma_{23} = -\frac{2C\mu x_2}{r^3} \left\{ 3 \left(\frac{x_3}{r} \right)^2 + \frac{\mu}{\lambda+\mu} \right\}; \quad \sigma_{33} = -\frac{2C\mu x_3}{r^3} \left\{ 3 \left(\frac{x_3}{r} \right)^2 + \frac{\mu}{\lambda+\mu} \right\}$$

10

$$2. \text{ Now } T_i = \sigma_{ij} n_j \quad \text{and the normal to the sphere is } \hat{r} = x_j \hat{e}_j / \sqrt{(x_i x_i)} \quad \therefore n_j = x_j / r = \frac{x_j}{r}$$

$$\therefore T_i = \sigma_{ij} n_j = \frac{2C\mu}{r^4} \left\{ (x_3 x_j \delta_{ij} - x_j x_j) \delta_{i3} - x_i x_j \delta_{j3} \right\} \frac{\mu}{\lambda+\mu} - \frac{3x_3 x_i \left(\frac{x_j x_j}{r^2} \right)}{r^2} \quad \checkmark$$

10

$$\text{thus } T_i = -\frac{6x_3 x_i \mu}{r^4} \quad \text{for } i=1,2 \quad T_3 = \frac{2C\mu^2}{(\lambda+\mu)r^2} = \frac{6C\mu x_3^2}{r^4}$$

$$3. \int_S T_i dS = \iint_0^{2\pi} \int_0^\pi T_i r^2 \sin\varphi d\varphi d\theta = \int_0^{2\pi} \int_0^\pi -\frac{6C\mu x_i \mu r^2}{r^4} \sin\varphi d\varphi d\theta \quad \text{for constant } r \text{ and } i=1,2 \\ 0 \leq \theta \leq 2\pi; 0 \leq \varphi \leq \pi$$

$$\text{now } \frac{x_3}{r} = \cos\varphi \quad \frac{x_i}{r} = (\cos\theta) \sin\varphi \quad i=1,2; \quad \text{thus } \int_S T_i dS = \int_0^{2\pi} \int_0^\pi -6C\mu \cos\theta \sin^2\varphi d\varphi (\frac{\cos\theta}{\sin\theta}) d\theta$$

$$\text{but } \int_0^{2\pi} \left(\frac{\cos\theta}{\sin\theta} \right) d\theta = 0 \quad \therefore \int_S T_i dS = 0 \quad i=1,2 \quad \checkmark$$

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$$\int_S T_3 r^2 \sin\varphi d\varphi d\theta = -\frac{2C\mu^2}{\lambda+\mu} \int_0^{2\pi} \int_0^\pi \sin\varphi d\varphi d\theta = 6C\mu \int_0^{2\pi} \int_0^\pi \cos^2\varphi \sin\varphi d\varphi d\theta$$

where $\frac{x_3}{r} = \cos\varphi$; $\int_S T_3 ds = -\frac{8\pi C\mu^2}{\lambda+\mu} = 8C\mu\pi = -8C\mu \left(\frac{\lambda+2\mu}{\lambda+\mu}\right) = -P$ when we substitute back for C .

Since the volume of the sphere contains the concentrates for P , for equil. to exist the surface must support the self equilibrating force $-P$.

Shear tractions produce no moment -4

4. According to Prof. Hermann we need to evaluate $\frac{\partial u_2}{\partial z} = \epsilon_2 = \epsilon_2 \cdot \delta \cdot \epsilon_2$

but we know

$$u_{i,j} = C \left[-\frac{\lambda+3\mu}{\lambda+\mu} \delta_{ij} \frac{x_j}{r^3} + \frac{\delta_{ij}}{r^3} \frac{x_i}{r^3} + \delta_{ij} \frac{x_3}{r^3} - \frac{3x_3 x_i x_j}{r^5} \right]$$

for $i=j=3$

$$u_{3,3} = C \left[-\frac{\lambda+3\mu}{\lambda+\mu} \frac{x_3}{r^3} + \frac{x_3}{r^3} + \frac{x_3}{r^3} - \frac{3x_3^3}{r^5} \right] = C \frac{x_3}{r^3} \left[\frac{\lambda-\mu}{\lambda+\mu} - 3 \left(\frac{x_3}{r} \right)^2 \right]$$

$$\text{where } C = \frac{(\lambda+\mu)P}{8\pi\mu(\lambda+2\mu)}$$

simplify = ok

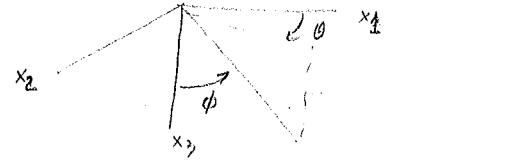
note $\delta = x_3$

$$\text{along } x_3 = r = x_3 \quad \therefore \quad u_{3,3} = \frac{C}{r^2} \left[\frac{\lambda-\mu}{\lambda+\mu} - 3 \right]$$

$$\begin{aligned} &= \frac{C}{r^2} \left[\frac{\lambda-\mu-3\lambda-3\mu}{\lambda+\mu} \right] = -\frac{C}{r^2} \left[\frac{2(\lambda+2\mu)}{\lambda+\mu} \right] \\ &= -\frac{P}{4\pi\mu r^2} \end{aligned}$$

10

see 4th page to show fractions are 0.



5. Now $u_1 = u_1 \mathbf{e}_1 + u_2 \mathbf{e}_2 + u_3 \mathbf{e}_3 = u_r \mathbf{e}_r + u_\theta \mathbf{e}_\theta + u_\varphi \mathbf{e}_\varphi$ where

$$u_r = \mathbf{e}_1 \cos \theta \sin \varphi + \mathbf{e}_2 \sin \theta \sin \varphi + \mathbf{e}_3 \cos \varphi; \quad \mathbf{e}_\varphi = \mathbf{e}_1 \cos \theta \cos \varphi + \mathbf{e}_2 \sin \theta \cos \varphi - \mathbf{e}_3 \sin \varphi$$

$$\mathbf{e}_\theta = -\mathbf{e}_1 \sin \theta + \mathbf{e}_2 \cos \theta$$

Now $u_r = u_1 \cdot \mathbf{e}_r = u_1 \cos \theta \sin \varphi + u_2 \sin \theta \sin \varphi + u_3 \cos \varphi$

$$u_\theta = u_1 \cdot \mathbf{e}_\theta = -u_1 \sin \theta + u_2 \cos \theta$$

$$u_\varphi = u_1 \cdot \mathbf{e}_\varphi = u_1 \cos \theta \cos \varphi + u_2 \sin \theta \cos \varphi - u_3 \sin \varphi$$

$$u_1 = \frac{C}{r} \cos \theta \cos \varphi \sin \varphi \quad u_2 = \frac{C}{r} \cos \theta \sin \theta \sin \varphi \quad u_3 = \frac{C}{r} \cos^2 \theta \quad C = \frac{(\lambda + \mu) P}{8\pi \mu (\lambda + 2\mu)}$$

$$u_r = \frac{C}{r} [\cos \theta \cos^2 \theta \sin^2 \varphi + \cos \theta \sin^2 \theta \sin^2 \varphi + \cos^3 \theta] = \frac{C}{r} [\cos \theta] \quad -3$$

$$u_\theta = \frac{C}{r} [-\cos \theta \cos \theta \sin \theta + \cos \theta \sin \theta \cos \theta \sin \theta] = 0 \quad X$$

$$u_\varphi = \frac{C}{r} [\cos^2 \theta \cos^2 \theta \sin \varphi + \cos^3 \theta \sin^2 \theta \sin \varphi - \sin \theta \cos^2 \theta] = 0 \quad \checkmark$$

(7)

6. Now since we know what the components are then for the non zero displacement components

$$\epsilon_{rr} = \frac{\partial u_r}{\partial r} = -\frac{C \cos \theta}{r^2} \quad \epsilon_{\varphi\varphi} = \frac{C}{r^2} [\cos \theta] = \frac{u_r}{r} \quad \epsilon_{\theta\theta} = \frac{C}{r^2} \cos \theta = \frac{u_r}{r}$$

$$\epsilon_{r\theta} = 0 \quad \epsilon_{r\varphi} = 0 \quad \epsilon_{\theta\varphi} = 0$$

using the formulae found on pg 184 of Sokolnikov w/ $\theta = \text{tang } \varphi$; $\alpha = \text{tang } \theta$

Since $\tau_{ij} = \lambda \epsilon_{KK} \delta_{ij} + 2\mu \epsilon_{ij}$ with $\epsilon_{KK} = \frac{C}{r^2} \cos \theta = \epsilon_{rr} + \epsilon_{\theta\theta} + \epsilon_{\varphi\varphi}$
 $= \lambda \frac{C}{r^2} \cos \theta \delta_{ij} + 2\mu \epsilon_{ij}$

$$\tau_{rr} = \frac{\lambda - 2\mu}{r^2} C \cos \theta \quad \tau_{\theta\theta} = \tau_{\varphi\varphi} = \frac{\lambda + 2\mu}{r^2} C \cos \theta \quad \tau_{r\theta} = 2\mu \epsilon_{r\theta} = 0 \quad X \quad -2$$

$$\tau_{r\varphi} = 2\mu \epsilon_{r\varphi} = 0 \quad \tau_{\theta\varphi} = 2\mu \epsilon_{\theta\varphi} = 0. \quad X$$

(8)

See next page



Kelvin in spherical

#5. Butler $u = \mathbf{B} = \frac{1}{4(1-\nu)} \nabla(r \cdot \mathbf{B})$ where $\mathbf{B} = \frac{\mathbf{P} \mathbf{e}_3}{4\pi\mu r} = \frac{\mathbf{P}(\cos\varphi \mathbf{e}_r - \sin\varphi \mathbf{e}_\theta)}{4\pi\mu r}$

$$\Phi_3 = A \mathbf{e}_r + B \mathbf{e}_\theta + C \mathbf{e}_\phi \quad A = \mathbf{e}_3 \cdot \mathbf{e}_r \quad B = \mathbf{e}_3 \cdot \mathbf{e}_\theta \quad C = \mathbf{e}_3 \cdot \mathbf{e}_\phi$$

$$r \cdot \mathbf{B} = r \mathbf{e}_r (\cos\varphi \mathbf{e}_r) = \frac{r \cos\varphi}{4\pi\mu r} = \text{const.} \quad u_r = \frac{\mathbf{P}}{4\pi\mu r} \cos\varphi \mathbf{e}_r = \frac{\mathbf{P} \sin\varphi}{4\pi\mu r} \mathbf{e}_\theta - \frac{1}{4(1-\nu)} \nabla \left(\frac{\mathbf{P}\varphi}{4\pi\mu r} \right)$$

$$\nabla = \frac{\partial}{\partial r} \mathbf{e}_r + \frac{1}{r \sin\varphi} \frac{\partial}{\partial \theta} \mathbf{e}_\theta + \frac{1}{r \sin\varphi} \frac{\partial}{\partial \phi} \mathbf{e}_\phi \quad \therefore \nabla \frac{\cos\varphi}{4\pi\mu r} = -\frac{1}{4\pi\mu r} \sin\varphi \mathbf{e}_\theta$$

$$\therefore u_r = \frac{\mathbf{P}}{4\pi\mu r} \cos\varphi \mathbf{e}_r = \frac{\mathbf{P} \cos\varphi}{4\pi\mu r} \mathbf{e}_\theta + \frac{1}{4(1-\nu)} \frac{\sin\varphi}{4\pi\mu r} \mathbf{e}_\phi$$

$$\boxed{u_r = \frac{\mathbf{P} \cos\varphi}{4\pi\mu r} \quad u_\theta = 0 \quad u_\phi = \frac{\mathbf{P} \sin\varphi}{4\pi\mu r} \left[-1 + \frac{1}{4(1-\nu)} \right] = \frac{4\nu-3}{4(1-\nu)} \frac{\mathbf{P} \sin\varphi}{4\pi\mu r}}$$

$$\text{Let } \frac{\mathbf{P}}{4\pi\mu} = C$$

#6 $\epsilon_{rr} = \frac{\partial u_r}{\partial r} = -\frac{\cos\varphi}{r^2}$ $\epsilon_{\theta\theta} = \frac{1}{r} \sin\varphi \frac{\partial u_\theta}{\partial \theta} + \frac{u_r}{r} + u_\phi \frac{\cos\varphi}{r}$
 $= \frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} + \frac{4\nu-3}{4(1-\nu)} \frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} = \frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} \frac{1}{4(1-\nu)}$

 $\epsilon_{\phi\phi} = \frac{1}{r} \frac{\partial u_\phi}{\partial \phi} + \frac{u_r}{r} = \frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} \left[1 + \frac{4\nu-3}{4(1-\nu)} \right] = \frac{\mathbf{P} \cos\varphi}{4(1-\nu) \cdot 4\pi\mu r^2}$

$$\boxed{\epsilon_{r\theta} = \frac{1}{2} \left(\frac{1}{r \sin\varphi} \frac{\partial u_r}{\partial \theta} - \frac{u_\theta}{r} + \frac{\partial u_\theta}{\partial r} \right) = 0 \quad \epsilon_{\theta\phi} = \frac{1}{2} \left(\frac{1}{r} \frac{\partial u_\theta}{\partial \phi} - \frac{u_\phi}{r} + \frac{1}{r \cos\varphi} \frac{\partial u_\phi}{\partial \theta} \right) = 0}$$

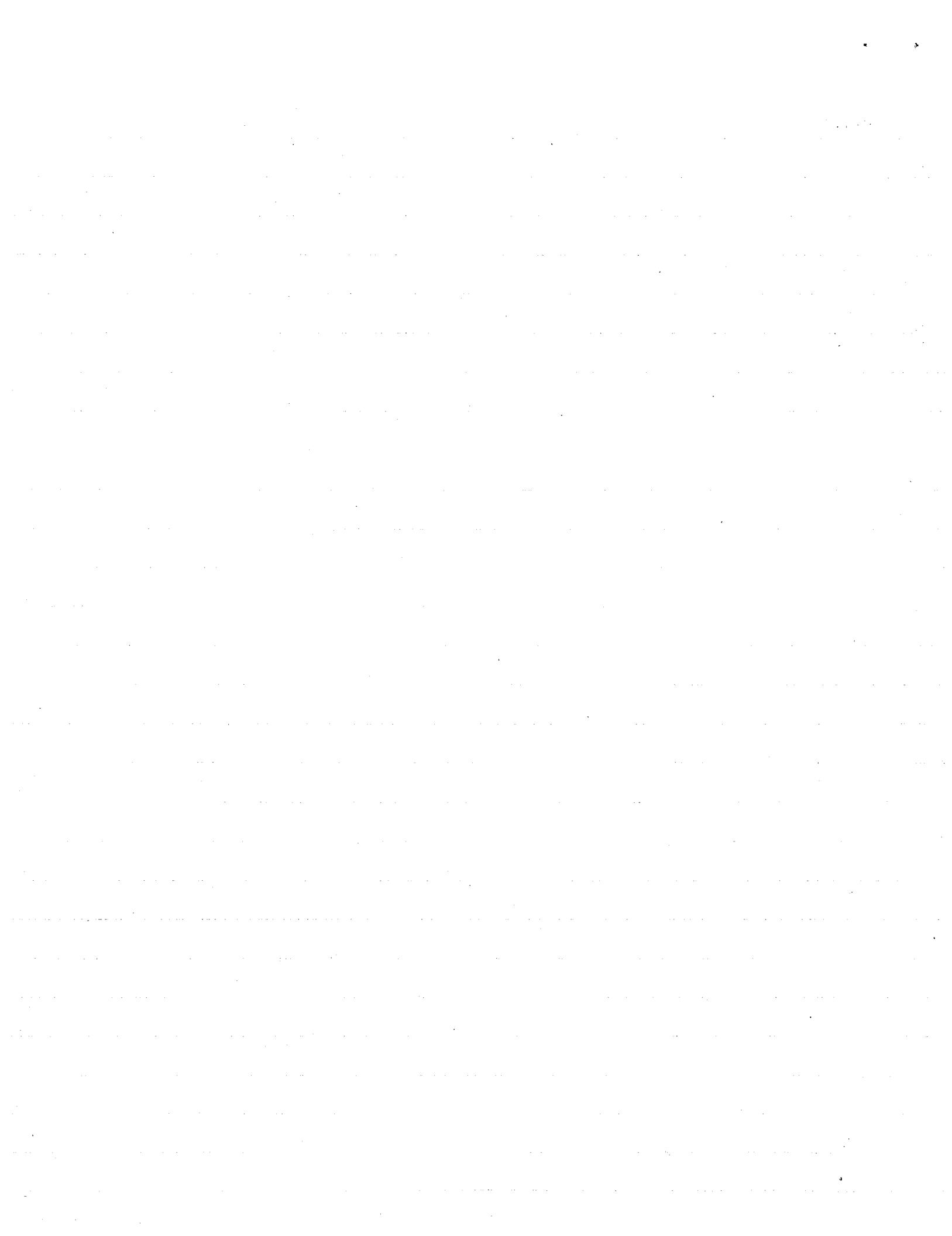
$$\epsilon_{\phi r} = \frac{1}{2} \left(\frac{1}{r} \frac{\partial u_r}{\partial \phi} - \frac{u_\phi}{r} + \frac{\partial u_\phi}{\partial r} \right) = \frac{1}{2} \left(-\frac{\mathbf{P} \sin\varphi}{4\pi\mu r^2} - 2 \frac{4\nu-3}{4(1-\nu)} \frac{\mathbf{P} \sin\varphi}{4\pi\mu r^2} \right)$$
 $= -\frac{1}{2} \frac{\mathbf{P} \sin\varphi}{4\pi\mu r^2} \left[\frac{4-4\nu+8\nu-6}{4(1-\nu)} \right]$
 $= -\frac{1}{2} \frac{\mathbf{P} \sin\varphi}{4\pi\mu r^2} \frac{(2\nu-1)2}{4(1-\nu)} = \frac{\mathbf{P} \sin\varphi (1-2\nu)}{16\pi\mu r^2 (1-\nu)}$

$$\boxed{\epsilon_{rr} = -\frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} \quad \epsilon_{\theta\theta} = \epsilon_{\phi\phi} = \frac{\mathbf{P} \cos\varphi (1-2\nu)}{16\pi\mu r^2 (1-\nu)} \quad \epsilon_{r\theta} = \epsilon_{\theta\phi} = 0 \quad \epsilon_{\phi r} = \frac{\mathbf{P} \sin\varphi (1-2\nu)}{16\pi\mu r^2 (1-\nu)}}$$

$$\sigma_{ij} = \lambda \delta_{ij} \epsilon_{KK} + \epsilon_{ij} 2\mu \quad \boxed{\sigma_{\theta\phi} = 2\mu \epsilon_{\phi r} = \frac{\mathbf{P} \sin\varphi (1-2\nu)}{8\pi\mu r^2 (1-\nu)} \quad \sigma_{\theta\phi} = \sigma_{r\theta} = 0}$$

$$\epsilon_{KK} = \frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} \left[-1 + \frac{1}{4(1-\nu)} \right] = \frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} \left[\frac{-2+4\nu}{4(1-\nu)} \right] = -\frac{\mathbf{P} \cos\varphi}{8\pi\mu r^2} \frac{(1-2\nu)}{1-\nu}$$

$$\sigma_{rr} = -\frac{\lambda \mathbf{P} \cos\varphi (1-2\nu)}{8\pi\mu r^2 (1-\nu)} + 2\mu \frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} = \frac{\mathbf{P} \cos\varphi}{4\pi\mu r^2} \left(\frac{2\nu-2\lambda}{1-\nu} \right)$$



$$T_{\theta\theta} = -\frac{\lambda P_{\cos\varphi}(1-2\nu)}{8\pi\mu r^2(1-\nu)} + \frac{P_{\cos\varphi}}{16\pi\mu(1-\nu)r^2} \cdot \frac{2\mu}{8\pi r^2} = \frac{P_{\cos\varphi}}{8\pi r^2} \left(\frac{1-2\nu}{1-\nu} \right) = T_{\varphi\varphi}$$

* 3. For Tongues to exist let us take moments about the origin T_{rr} produces no moments
 $T_{\theta\varphi}, T_{r\theta} = 0$

$$\begin{aligned} \mathbf{T} &= T_i \mathbf{e}_i \quad \sum M_i \mathbf{e}_i = \int \mathbf{r} \times \mathbf{T}_i dA = \int (x_1 T_2 - x_2 T_1) \mathbf{e}_3 dA + \int (x_2 T_3 - x_3 T_2) \mathbf{e}_1 dA \\ \mathbf{r} &= x_j \mathbf{e}_j \quad \mathbf{T} = T_i \mathbf{e}_i \quad \mathbf{r} \times \mathbf{T} = x_j T_i \epsilon_{ijk} \mathbf{e}_k \\ dA &= r^2 \sin\varphi d\varphi d\theta \quad x_1 = r \cos\theta \sin\varphi \quad x_2 = r \sin\theta \sin\varphi \quad x_3 = r \cos\varphi \end{aligned}$$

$$T_1 = -\frac{6\mu \cos\varphi \cos\theta \sin\varphi}{r^2} \quad T_2 = -\frac{6\mu \cos\varphi \sin\theta \sin\varphi}{r^2} \quad T_3 = -\frac{2C\mu^2}{(\lambda+\mu)r^2} - \frac{6C\mu x_3^2}{r^4}$$

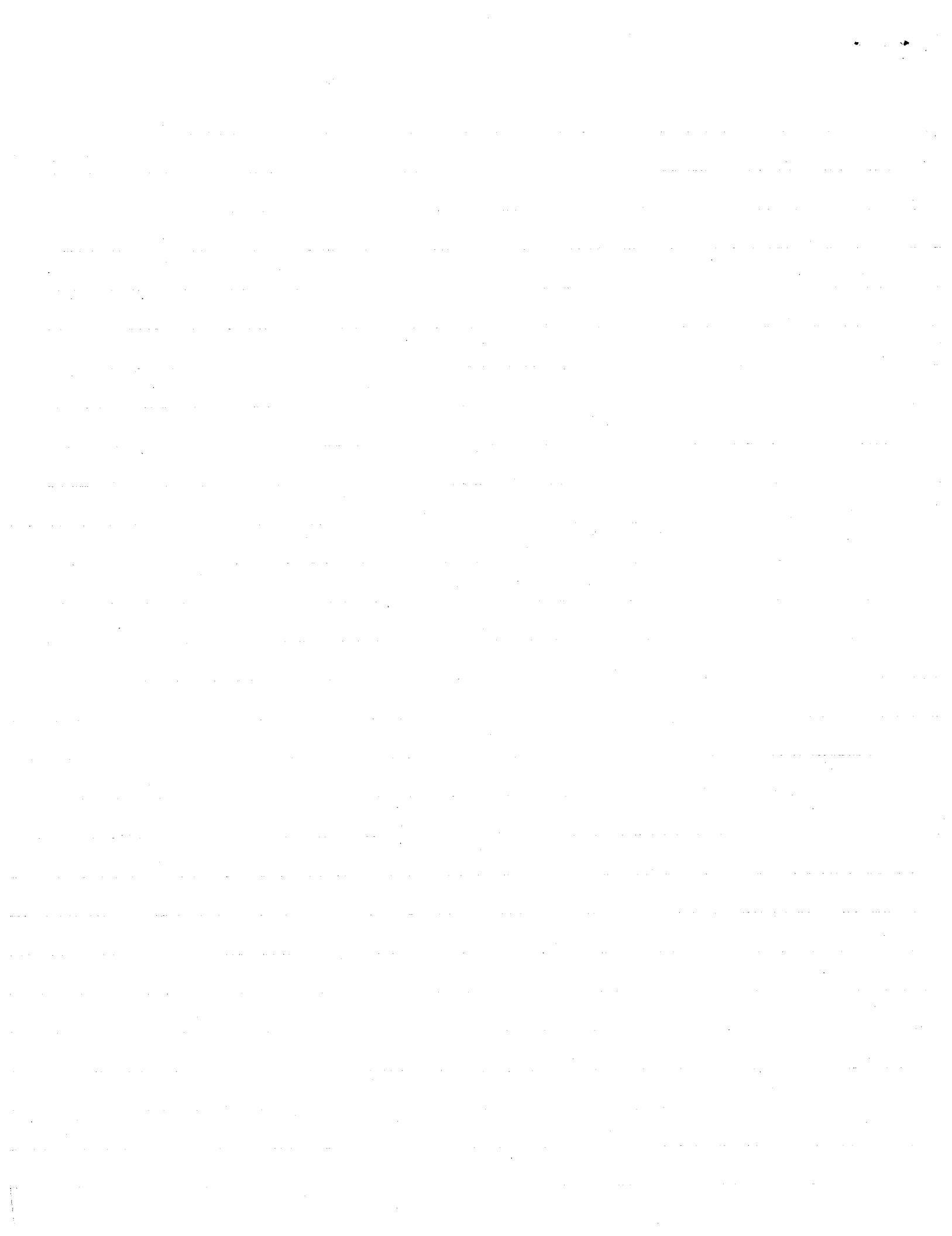
$$M_3 = \int (x_1 T_2 - x_2 T_1) dA = -6 \int \frac{r^2 \sin\varphi d\varphi d\theta}{r^2} \left[r \cos\theta \sin^2\varphi \cos\varphi \sin\theta - r \sin\theta \sin^2\varphi \cos\varphi \cos\theta \right] = 0$$

$$M_2 = \int r^2 \sin\varphi d\theta d\varphi \left(r \sin\theta \sin\varphi \cdot \left[-\frac{2C\mu^2}{(\lambda+\mu)r^2} - \frac{6C\mu \cos^2\varphi}{r^2} \right] + 6\mu \frac{\cos\varphi}{r^2} \left[\cos\varphi \sin\theta \sin\varphi \right] \right)$$

$$\int_0^{2\pi} \sin\theta d\theta = 0 \Rightarrow M_2 = 0$$

$$M_1 = \int r^2 \sin\varphi d\theta d\varphi \left(r \cos\varphi \cdot \left(-\frac{6\mu \cos\varphi \cos\theta \sin\varphi}{r^2} - r \cos\theta \sin\varphi \right) \left[f(\varphi) \text{ only} \right] \right)$$

$$\int_0^{2\pi} \cos\theta d\theta = 0 \Rightarrow M_1 = 0$$



DIVISION OF APPLIED MECHANICS
DEPARTMENT OF MECHANICAL ENGINEERING
STANFORD UNIVERSITY

ME 238C Theory of Elasticity

Spring 1979

Problem Set No. 2

1. Calculate the displacement associated with a doublet (double center of compression-dilatation).
2. Consider an extended elastic solid containing a spherical cavity and subjected to remote uniaxial tension. Determine the principal stresses on the surface of the cavity and show, by superposition, that if isotropic (all-around) remote tension were applied, the stress concentration factor would be 1.5.
3. Determine the settlement outside a rigid circular stamp impressed into an elastic half-space.

$$u = TB = \frac{1}{4(1-\nu)} \nabla (r \cdot B + A)$$

$$r = r e_r$$

$$\frac{\partial}{\partial z} = \frac{\partial}{\partial R} \frac{\partial R}{\partial z} + \frac{\partial}{\partial \phi} \frac{\partial \phi}{\partial z} + \frac{\partial}{\partial \theta} \frac{\partial \theta}{\partial z}$$

$$= \frac{1}{\cos \phi} \frac{\partial}{\partial R} - \frac{1}{R \sin \phi} \frac{\partial}{\partial \phi}$$

$$z = R \cos \phi$$

$$1 = \frac{\partial R \cos \phi}{\partial z}$$

$$1 = R \sin \phi \frac{\partial \phi}{\partial z}$$

$$\frac{\partial}{\partial z} = e_z \cdot \nabla$$

$$dz = -R \sin \phi d\phi + \cos \phi dR$$

$$\frac{\partial z}{\partial \phi} \quad \frac{\partial z}{\partial R}$$

$$= e_r \cdot e_z \frac{\partial}{\partial r} + e_\theta \cdot e_z \frac{\partial}{\partial \theta} \left[\frac{1}{R \sin \phi} \right] r \sin \phi + e_\phi \cdot e_z \frac{\partial}{\partial \phi}$$

$$= \cos \phi \frac{\partial}{\partial r} + 0 + -\frac{\sin \phi}{r} \frac{\partial}{\partial \phi}$$

$$e_z \cdot \nabla = e_z \cdot e_r \frac{\partial}{\partial r}$$

$$\cos \phi \frac{\partial}{\partial r} = \frac{1}{r \sin \phi} \cdot 0 = -\frac{\sin \phi}{r} \frac{\partial}{\partial \phi}$$

$$\nabla = u_r \frac{\partial}{\partial r} + u_\phi \frac{\partial}{\partial \phi} + \frac{u_\theta}{r \sin \phi} \frac{\partial}{\partial \theta}$$

$$\nabla = e_x \frac{\partial}{\partial x} + e_y \frac{\partial}{\partial y} + e_z \frac{\partial}{\partial z} = e_r \frac{\partial}{\partial r} + e_\phi \frac{1}{r} \frac{\partial}{\partial \phi} + e_\theta \frac{1}{r \sin \phi} \frac{\partial}{\partial \theta}$$

$$e_z \cdot \nabla = \frac{\partial}{\partial \phi} = \cos \phi \frac{\partial}{\partial r} + \frac{\sin \phi}{r} \frac{\partial}{\partial \phi}$$

$$\frac{\partial}{\partial z} \left(\frac{1}{r} \right) = \cos \phi \left[-\frac{1}{r^2} \right] = -\frac{\cos \phi}{r^2}$$

$$TB = B_3 e_z$$

$$-\frac{1}{r^2} \cancel{\frac{\partial}{\partial z}}$$

$$B = A \frac{\partial}{\partial z} \left(\frac{1}{R} \right) e_z = -A \frac{\cos \phi}{r^2} [\cos \phi e_r - \sin \phi e_\phi]$$

~~$$TB = A \frac{\partial}{\partial z} \left(\frac{1}{R} \right) \Rightarrow r \cdot B = r e_r [-A \cos \phi \cos \phi e_r + 0] = -A \cos^2 \phi$$~~

$$B = \frac{D}{R} + C \frac{\partial}{\partial z} \left[\frac{\partial}{\partial z} \left(\frac{1}{R} \right) \right] = \frac{D}{R} + C \frac{\partial}{\partial z} \left[-\frac{\cos \phi}{r^2} \right] = \frac{D}{R} + C \left\{ \frac{2 \cos^2 \phi}{r^3} - \frac{\sin^2 \phi}{r^3} \right\}$$

$$\frac{\partial}{\partial z} \left(-\frac{\cos \phi}{r^2} \right) = \cos \phi \frac{\partial}{\partial r} \left(-\frac{\cos \phi}{r^2} \right) - \frac{\sin \phi}{r} \frac{\partial}{\partial \phi} \left(-\frac{\cos \phi}{r^2} \right)$$

$$= -\cos^2 \phi \left(-\frac{2}{r^3} \right) + \frac{\sin \phi}{r^3} (-\sin \phi)$$

$$\therefore u = +\frac{A}{r^2} \left[\sin \frac{2\phi}{2} e_\phi - \cos^2 \phi e_r \right] - \frac{1}{4(1-\nu)} \nabla \left\{ -\frac{A \cos^2 \phi + D}{r} + \frac{C}{r^3} \left\{ \cos 2\phi + \cos^3 \phi \right\} \right\}$$

~~$$\nabla = e_r \frac{\partial}{\partial r} + e_\phi \frac{1}{r} \frac{\partial}{\partial \phi} + e_\theta \frac{1}{r \sin \phi} \frac{\partial}{\partial \theta}$$~~

$$\left[\frac{A \cos^2 \phi - D}{r^2} + \frac{3C}{r^4} (\cos 2\phi + \cos^3 \phi) \right] e_r$$

$$+ \left[+ \frac{A \cdot 2 \cos \phi \sin \phi}{r^2} + \frac{C}{r^4} \left\{ -3 \sin 2\phi + 2 \cos \phi \sin \phi \right\} \right] e_\phi$$

$$u = e_\phi \left[\frac{A}{r^2} \frac{\sin 2\phi}{2} - \frac{1}{4(1-\nu)} \left\{ A \frac{\sin 2\phi}{r^2} - 3C \frac{\sin 2\phi}{r^4} \right\} \right] + e_r \left[-\frac{A \cos^2 \phi}{r^2} + \frac{A \cos^2 \phi - D}{r^2} - \frac{3C}{r^4} (\cos 2\phi + \cos^3 \phi) \right]$$

~~$$u_r = -\frac{D}{r^2} - \frac{3C}{r^4} (\cos 2\phi + \cos^3 \phi)$$~~

$$u_\phi = \frac{A(1-2\nu)}{4r^2(1-\nu)} \sin 2\phi + \frac{3C}{4(1-\nu)} \frac{\sin 2\phi}{r^4}$$

$$\epsilon_{\varphi\varphi} = \frac{\sin 2\varphi}{4(1-\nu)} \left[\frac{A(1-2\nu)}{r^2} + \frac{3C}{r^4} \right] \quad u_r = \frac{\cos^2\varphi}{4(1-\nu)} \left\{ -\frac{A(5-4\nu)}{r^2} + \frac{3C}{r^4} \right\} + \frac{D}{4(1-\nu)r^2} + \frac{3C}{4(1-\nu)} \frac{\cos 2\varphi}{r^4}$$

$$\epsilon_{rr} = \frac{\partial u_r}{\partial r} = \frac{\cos^2\varphi}{4(1-\nu)} \left[\frac{2A(5-4\nu)}{r^3} - \frac{24C}{r^5} \right] - \frac{2D}{4(1-\nu)r^3} + \frac{12C}{4(1-\nu)} \frac{\cos 2\varphi}{r^5}$$

$$\epsilon_{\theta\theta} = \frac{\cos^2\varphi}{4(1-\nu)} \left\{ -\frac{A(5-4\nu)}{r^3} + \frac{3C}{r^5} \right\} + \frac{D}{4(1-\nu)r^3} + \frac{3C}{4(1-\nu)} \frac{\cos 2\varphi}{r^5} + \frac{\cos^2\varphi}{4(1-\nu)} \left[\frac{2A(1-2\nu)}{r^3} + \frac{6C}{r^5} \right] = \frac{u_r + u_\varphi \cos 2\varphi}{r}$$

$$= \frac{\cos^2\varphi}{4(1-\nu)} \left[-\frac{3A}{r^3} + \frac{9C}{r^5} \right] + \frac{D}{4(1-\nu)r^3} + \frac{3C}{4(1-\nu)} \frac{\cos 2\varphi}{r^5} = \frac{\cos^2\varphi}{4(1-\nu)} \left[-\frac{3A}{r^3} + \frac{12C}{r^5} \right] - \frac{3C}{4(1-\nu)} \frac{\sin 2\varphi}{r^5} + \frac{D}{4(1-\nu)r^3}$$

$$\epsilon_{\varphi\varphi} = \frac{1}{r} \frac{\partial u_\varphi}{\partial \varphi} + \frac{u_r}{r} = \frac{8 \cos 2\varphi}{9(1-\nu)} \left[\frac{2A(1-2\nu)}{r^3} + \frac{6C}{r^5} \right] + \frac{\cos^2\varphi}{4(1-\nu)} \left\{ -\frac{A(5-4\nu)}{r^3} + \frac{3C}{r^5} \right\} + \frac{D}{4(1-\nu)r^3} + \frac{3C}{4(1-\nu)} \frac{\cos 2\varphi}{r^5}$$

$$= \frac{\cos 2\varphi}{4(1-\nu)} \left[\frac{2A(1-2\nu)}{r^3} + \frac{9C}{r^5} \right] + \frac{\cos^2\varphi}{4(1-\nu)} \left\{ -\frac{A(5-4\nu)}{r^3} + \frac{3C}{r^5} \right\} + \frac{D}{4(1-\nu)r^3}$$

$$\epsilon_{r\theta} = 0 \quad \epsilon_{\theta\varphi} = 0 \quad \frac{\cos 2\varphi}{4(1-\nu)} \left[-\frac{3A}{r^3} + \frac{12C}{r^5} \right] \Rightarrow \frac{\sin 2\varphi}{4(1-\nu)} \left[\frac{2A(1-2\nu)}{r^3} + \frac{9C}{r^5} \right] + \frac{D}{4(1-\nu)r^3}$$

$$\epsilon_{rr} = \frac{1}{2} \left[\frac{-\sin 2\varphi}{4(1-\nu)} \left\{ -\frac{A(5-4\nu)}{r^3} + \frac{3C}{r^5} \right\} - \frac{6C}{4(1-\nu)} \frac{\sin 2\varphi}{r^5} - \frac{\sin 2\varphi}{4(1-\nu)} \left\{ \frac{A(1-2\nu)}{r^3} + \frac{3C}{r^5} \right\} + \frac{\sin 2\varphi}{4(1-\nu)} \left\{ -2A(1-2\nu) - \frac{12C}{r^3} \right\} \right]$$

$$= \frac{\sin 2\varphi}{8(1-\nu)} \left\{ \frac{A(5-4\nu)}{r^3} - \frac{A(1-2\nu)}{r^3} - \frac{2A(1-2\nu)}{r^3} - \frac{3C}{r^5} - \frac{6C}{r^5} - \frac{3C}{r^5} - \frac{12C}{r^5} \right\}$$

$$\frac{\sin 2\varphi}{8(1-\nu)} \left\{ \frac{A(2+2\nu)}{r^3} - \frac{24C}{r^5} \right\} = \frac{\sin 2\varphi}{4(1-\nu)} \left\{ \frac{A(1+\nu)}{r^3} - \frac{12C}{r^5} \right\}$$

$\sigma_{\varphi r} = 2\mu \epsilon_{\varphi r} = \mu \frac{\sin \varphi \cos 2\varphi}{(1-\nu)} \left\{ \frac{A(1+\nu)}{r^3} - \frac{12C}{r^5} \right\}$	\checkmark	$\sigma_{r\theta} = \sigma_{\theta\varphi} = 0 \quad \checkmark$
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$$\epsilon_{rr} + \epsilon_{\theta\theta} + \epsilon_{\varphi\varphi} = \frac{\cos^2\varphi}{4(1-\nu)} \left[\frac{10A - 7A\nu - 3A - 5A + 8A\nu}{r^3} + \frac{-12C + 9C + 3C}{r^5} \right] + \frac{D}{4(1-\nu)r^3} \left[1 - 2 + 1 \right]$$

$$\frac{\cos 2\varphi}{4(1-\nu)r^3} \left[\frac{-12C + 3C + 9C}{r^5} + \frac{2A(1-2\nu)}{r^3} \right] = \frac{\cos^2\varphi}{2(1-\nu)} \frac{A}{r^3} + \frac{\cos 2\varphi}{2(1-\nu)} \frac{A(1-2\nu)}{r^3}$$

$$= \frac{\cos^2\varphi A}{2(1-\nu)r^3} + \frac{\cos^2\varphi A(1-2\nu)}{2(1-\nu)r^3} - \frac{\sin^2\varphi A(1-2\nu)}{2(1-\nu)r^3}$$

$$= \frac{\cos^2\varphi A}{r^3} - \frac{\sin^2\varphi A(1-2\nu)}{2(1-\nu)r^3}$$

$$\sigma_{rr} = \lambda \epsilon_{kk} + 2\mu \epsilon_{rr} = 2\mu \left[\frac{\nu}{(1-2\nu)} \epsilon_{kk} + \epsilon_{rr} \right]$$

$$2\mu \left\{ \frac{\cos^2\varphi}{(1-2\nu)} \left[\frac{A}{2(1-\nu)} + \frac{\cos 2\varphi}{2(1-\nu)} \frac{A(1-2\nu)}{r^3} \right] + \frac{\cos^2\varphi}{2A(1-\nu)} \left[\frac{2A(5-4\nu)}{r^3} - \frac{6C}{r^5} \right] - \frac{2D}{4(1-\nu)r^3} - \frac{12C}{4(1-\nu)r^3} \frac{6C}{2A(1-\nu)} \frac{\cos 2\varphi}{r^5} \right\}$$

$$2\mu \left\{ \frac{\cos^2\varphi}{2(1-\nu)} \left[\frac{\nu A}{(1-2\nu)} + A(5-4\nu) \right] + \frac{\cos 2\varphi}{2(1-\nu)} \right\}$$

$$\cos^2\phi - \sin^2\phi$$

$$\epsilon_{rr} + \epsilon_{\theta\theta} + \epsilon_{\phi\phi} = \frac{\cos^2\phi}{4(1-v)} + \frac{2A}{r^3} + \frac{\cos^2\phi \cdot 2A(1-2v)}{4(1-v)r^3} = \frac{\cos^2\phi}{4(1-v)} \left[\frac{2A}{r^3} + \frac{2A(1-2v)}{r^3} \right] = \sin^2\phi \cdot \frac{2A(1-2v)}{4(1-v)r^3}$$

$$\sigma_{ij} = \lambda \epsilon_{KK} \delta_{ij} + 2\mu \epsilon_{ij}$$

$$= \frac{2\mu v}{1-v} \quad 2v(\lambda+\mu) = \lambda \quad \lambda [F-2v] = \frac{2v\mu}{1-2v}$$

$$2\mu \left[\frac{v}{1-2v} \epsilon_{KK} \delta_{ij} + \epsilon_{ij} \right]$$

$$\frac{v}{1-2v} \epsilon_{rr} + \epsilon_{rr} + \frac{v}{1-2v} (\epsilon_{\theta\theta} + \epsilon_{\phi\phi})$$

$$\frac{1-v}{1-2v} \epsilon_{rr}$$

$$\sigma_{rr} = \frac{\cos^2\phi}{4(1-2v)} \left[\frac{2A(5-4v)}{r^3} - \frac{12C}{r^5} \right] - \frac{2D}{4(1-2v)r^3} - \frac{12C}{4(1-2v)} \frac{\cos^2\phi}{r^5} + \frac{v}{1-2v} \left[\frac{\cos^2\phi}{4(1-v)} \left(-A \frac{2A}{r^3} + \frac{18C}{r^5} \right) \right]$$

$$+ \frac{v}{1-2v} \left\{ \frac{\cos^2\phi}{4(1-v)} \left[\frac{-4A(2-v)}{r^3} + \frac{12C}{r^5} \right] + \frac{2D}{4(1-v)r^3} + \frac{\cos^2\phi}{4(1-v)} \left[\frac{12C}{r^5} + \frac{2A(1-2v)}{r^3} \right] \right\}$$

$$\frac{\cos^2\phi}{4(1-2v)} \left\{ \frac{2A(5-4v)}{r^3} - \frac{4A(2-v)v}{(1-v)r^3} \right\} + 2A(1-2v) \frac{v}{1-2v} \frac{\cos^2\phi}{4(1-v)} \left[\frac{2A(1-2v)}{r^3} \right]$$

$$2\cos^2\phi - 1$$

$$\frac{\cos^2\phi \cdot 2A}{4(1-2v)r^3} \left\{ \frac{(5-4v)(1-v)}{5-9v+4v^2-4v+2v^2} - 2v(2-v) \right\} = \frac{2A \cos^2\phi}{4r^3} \frac{5-3v}{1-v} + \frac{\cos^2\phi \cdot 2A}{4r^3(1-v)} 2v$$

$$\frac{5-13v+6v^2}{(5-3v)(1-2v)} (5-v) A$$

$$\sigma_{\phi\phi} = 2\mu \left[\frac{v}{1-2v} \epsilon_{\phi\phi} + \frac{v}{1-2v} (\epsilon_{\theta\theta} + \epsilon_{rr}) \right]$$

$$= 2\mu \left[\frac{2\cos^2\phi - 1}{4(1-2v)} \left\{ \frac{2A(1-2v)}{r^3} + \frac{9C}{r^5} \right\} + \frac{\cos^2\phi}{4(1-2v)} \left\{ -A \frac{(5-4v)}{r^3} + \frac{3C}{r^5} \right\} + \frac{D}{4(1-2v)r^3} \right]$$

$$+ \frac{v}{1-2v} \left\{ \frac{\cos^2\phi}{4(1-v)} \left[\frac{(7-8v)A}{r^3} - \frac{3C}{r^5} \right] - \frac{D}{4(1-v)r^3} + \left(\frac{2\cos^2\phi - 1}{4(1-v)} \right) \frac{9C}{r^5} \right\}$$

$$\frac{\cos^2\phi}{4(1-2v)} \left\{ \frac{4A(1-2v)}{r^3} + \frac{(7-8v)vA}{(1-v)r^3} - \frac{5A+4Av}{r^3} \right. \\ \left. - A(1+4v)(1-v) + A(7v-8v^2) \right)$$

$$A[-1+3v+4v^2+7v-8v^2] = A[-1+4v-4v^2] = Am (1-4v+4v^2) = -A(1-2v)(1-2v)$$

$$\rightarrow 2\mu \left[\frac{\cos^2\phi}{4(1-2v)} \cdot \frac{-A(1-2v)}{(1-v)r^3} - \frac{2A}{4r^3} + \frac{\cos^2\phi}{4(1-2v)} \cdot \frac{21(1-2v)c}{(1-v)r^5} - \frac{1}{4(1-2v)} \cdot \frac{9c(1-2v)}{(1-v)r^5} + \frac{D(1-2v)}{4(1-2v)r^3(1-v)} \right]$$

$$\frac{\cos^2\phi}{4(1-2v)} \left\{ \frac{8C}{r^5} + \frac{3C}{r^5} \right. \\ \left. - \frac{21c}{r^5} - \frac{21vC}{r^5} \right\} = 1 \cdot \frac{9C}{r^5} + \frac{v}{1-v} \frac{9C}{r^5} + \frac{D}{4(1-2v)r^3} \left[\frac{-v}{1-v} + 1 \right]$$

$$\left| \begin{array}{l} \text{at } \phi = 90^\circ \\ \text{at } \phi = 0^\circ \end{array} \right| \sigma_{\phi\phi}(\phi=90^\circ) + \left| \begin{array}{l} \text{at } \phi = 0^\circ \\ \text{at } \phi = 90^\circ \end{array} \right| \sigma_{\phi\phi}(\phi=0^\circ) + \left| \begin{array}{l} \text{at } \phi = 90^\circ \\ \text{at } \phi = 0^\circ \end{array} \right| \sigma_{\theta\theta}(\phi=90^\circ)$$

$$\sigma_{\phi\phi} = 2\mu \left[\frac{\cos^2 \phi}{4(1-v)} \left(-\frac{A}{r^3} \right) (1-2v) + \frac{\cos^2 \phi}{4(1-v)} \cdot \frac{2C}{r^5} - \frac{2A}{4r^3} - \frac{9C}{4(1-v)r^5} + \frac{D}{4(1-v)r^3} \right]$$

take out $4(1-v)r^3$

$$\sigma_{\phi\phi} = \frac{\mu}{2(1-v)r^3} \left\{ \cos^2 \phi \left[(2v-1)A + \frac{2C}{r^2} \right] - 2(1-v)A - \frac{9C}{r^2} + D \right\} + T \sin^2 \phi$$

$$\sigma_{\phi\phi} = \sigma_{\phi\phi} + T \sin^2 \phi \quad @ \quad r=a \quad \phi=90^\circ$$

$$\sigma_{\phi\phi} = \frac{\mu}{2(1-v)a^2} \left\{ +2(1-v) \left[\frac{5(1-v)ac^5T}{\mu(7-5v)} \right] + \frac{9}{a^2} \frac{(1-v)}{\mu(7-5v)} a^5T - \frac{(1-v)(6-5v)a^5T}{\mu(7-5v)} \right\} + T$$

$$= \frac{T}{2(7-5v)} \left\{ +10 + 10v + 9 - 6 + 5v \right\} \frac{-7 + 15v}{2(7-5v)} T + \frac{2(7 + 5v)}{2(7-5v)} T$$

$$13 - 5v + 14 - 10v = 7 + 5v$$

$$\frac{27 - 15v}{2(7-5v)} = \frac{3(9-5v)}{7-5v}$$

$$\phi=90^\circ$$

$$\phi=0^\circ$$

$$\frac{3(9-5v)}{2(7-5v)} - \frac{-3-15v}{2(7-5v)} = 3 + 15v$$

$$-5vA - 5D + \frac{30C}{a^2} = 0 \quad (5-v)A - \frac{18C}{a^2} = -\frac{T(1-v)a^3}{\mu}$$

$$(1+v)A - \frac{12C}{a^2} = -\frac{T(1-v)a^3}{\mu}$$

$$6A - \frac{30C}{a^2} = 0 \quad A = +\frac{30C}{4a^2}$$

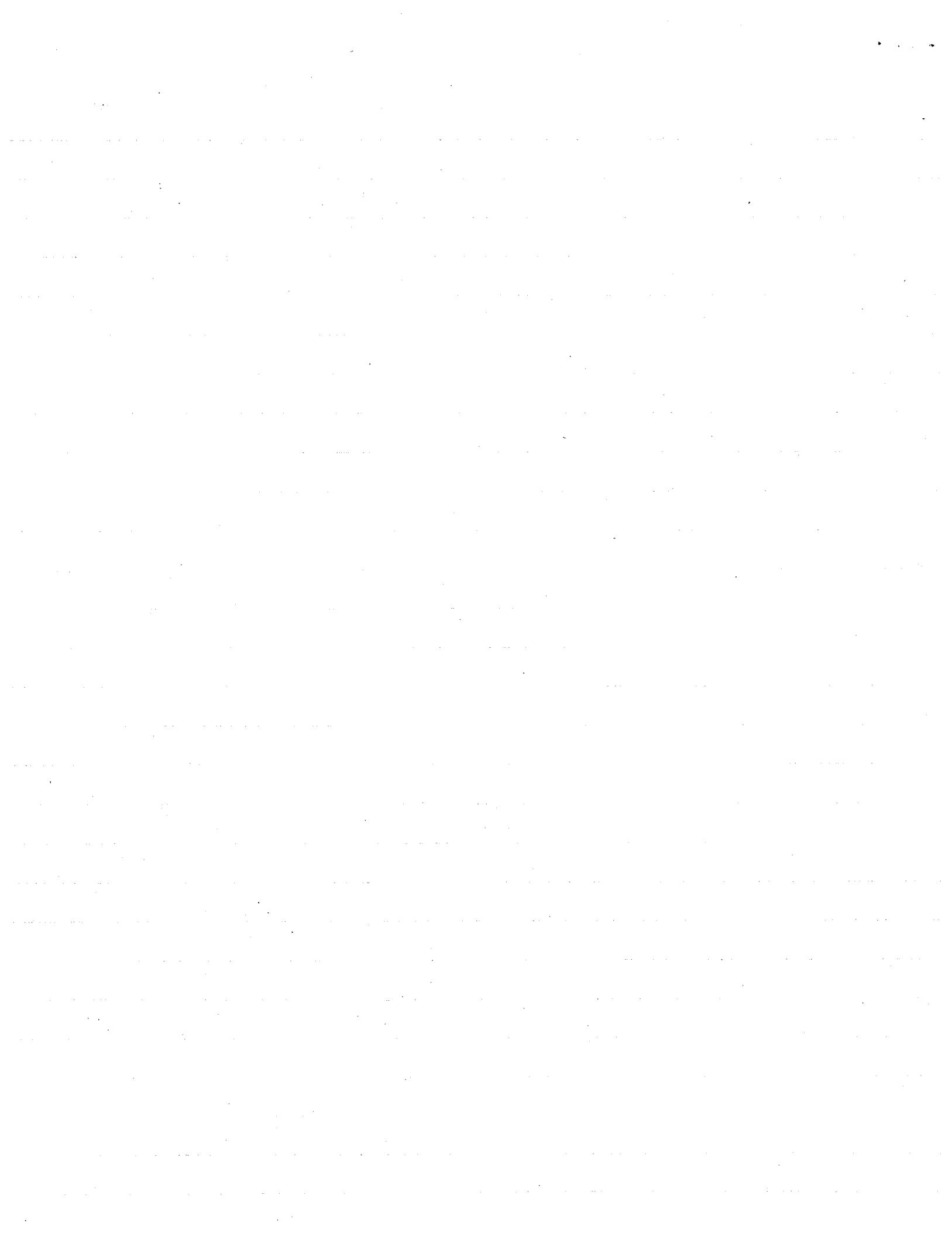
$$(5-v)\frac{5C}{a^2} - \frac{18C}{a^2}$$

$$\frac{7C}{a} - \frac{(7-5v)C}{a^2} = -\frac{T(1-v)a^5}{\mu(7-5v)} = C$$

$$-\frac{v5C}{a^2} + \frac{6C}{a^2} = D$$

$$\frac{C}{a^2}(6-5v)$$

$$\sigma_{\phi\phi}$$



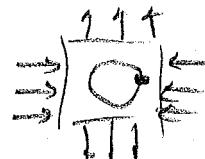
$$\frac{\mu v}{1-v} A\left(\frac{1}{R}\right)_{,xx} = \frac{\mu}{2(1-v)} \left[\left\{ \frac{2}{3} A\left(\frac{1}{R}\right)_{,xx} + D\left(\frac{1}{R}\right)_{,xx} + C\left(\frac{1}{R}\right)_{,xx} \right\}_{xx} \sin^2 \theta + \left\{ \frac{2}{3} A\left(\frac{1}{R}\right)_{,yy} + D\left(\frac{1}{R}\right)_{,yy} + C\left(\frac{1}{R}\right)_{,yy} \right\}_{yy} \right]$$

$$\cos^2 \theta - 2 \sin \theta \cos \theta \left\{ \frac{2}{3} A\left(\frac{1}{R}\right)_{,xy} + D\left(\frac{1}{R}\right)_{,xy} + C\left(\frac{1}{R}\right)_{,xy} \right\}$$

$$\sigma_{00} = - \left(\frac{3-15v}{14-10v} \right) T = - \frac{3}{2} \left(\frac{1-5v}{7-5v} \right) T.$$

$$\sigma_{pp} = \frac{3(9-5v)}{2(7-5v)} T. \quad (\text{at } \varphi = 90^\circ) \quad \sigma_{pp} \text{ (at } \varphi = 0) = \frac{-3-15v}{2(7-5v)} T = \frac{-3(1+5v)}{2(7-5v)}$$

if T is reversed $\sigma_{pp} = \frac{3(9-5v)}{2(7-5v)} T + \frac{3(1+5v)}{2(7-5v)} T = \frac{15T}{7-5v}$



J Goodier Pg 39 APM-55-7 Trans ASME 1932 Vol 55.

We need to add $\sigma_{pp} \text{ at } \varphi = 90^\circ + \sigma_{pp} \text{ at } \varphi = 0 + \sigma_{00} = T \left[\frac{27-15v - 3-15v - 3+15v}{2(7-5v)} \right]$

$$= T \left[\frac{21-15v}{2(7-5v)} \right] = \frac{3T}{2} \left(\frac{7-5v}{7-5v} \right) = \frac{3T}{2} \text{ N/mm}$$

$$\frac{A}{a^3} = - \frac{T}{8\mu} \frac{13-10v}{7-5v}; \quad \frac{B}{a^5} = \frac{T}{8\mu} \frac{1}{7-5v}; \quad \frac{C}{a^3} = \frac{T}{8\mu} \frac{5(1-2v)}{7-5v}$$

and $\sigma_{RR} = 2\mu \left\{ \frac{2A}{r^3} - \frac{2}{1-2v} \frac{C}{r^3} + 12 \frac{B}{r^5} + \left[- \frac{2(5-v)}{1-2v} \frac{C}{r^3} + \frac{36B}{r^5} \right] \cos 2\varphi \right\} + T \cos^2 \varphi$

$$\sigma_{pp} = 2\mu \left\{ -\frac{A}{r^3} - \frac{2v}{(1-2v)} \frac{C}{r^3} - \frac{3B}{r^5} + \left[\frac{C}{r^3} - 21 \frac{B}{r^5} \right] \cos 2\varphi \right\} + T \sin^2 \varphi$$

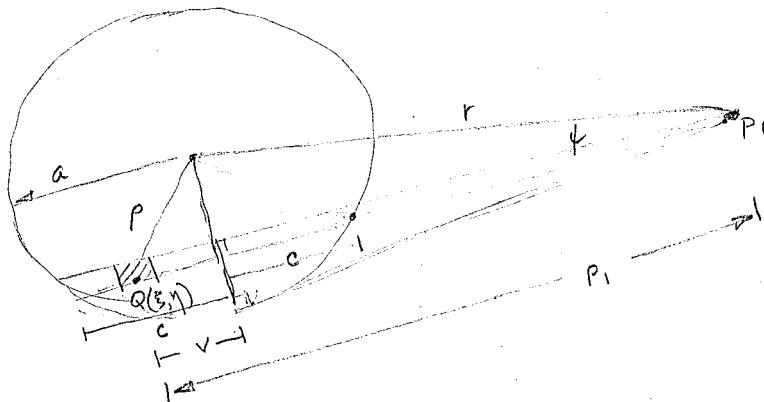
$$\sigma_{00} = 2\mu \left\{ -A/r^3 - \frac{2(1-v)}{1-2v} \frac{C}{r^3} - 9B/r^5 + \left[3C/r^3 - 15B/r^5 \right] \cos 2\varphi \right\} + 0$$

$$\sigma_{pq} = 2\mu \left\{ -\frac{2(1+v)}{1-2v} \frac{C}{r^3} + \frac{24B}{r^5} \right\} \sin 2\varphi - \frac{T}{2} \sin 2\varphi$$

$$\begin{aligned} &= \frac{2\mu T}{A \frac{2}{8\mu} \frac{(1-2v)}{(7-5v)}} \left\{ \frac{(1+v)}{(1-2v)} \cdot \frac{5(1-2v)}{r^3} \cdot \frac{a^3}{r^3} + 12B \cdot 5(1-2v) \frac{a^5}{r^5} \right\} \cdot \frac{\sin 2\varphi}{2} \\ &= \frac{5T}{2(7-5v)r^5} \left\{ -5(1+v)r^2 + 12(1-2v)a^2 \right\} \sin 2\varphi \end{aligned}$$



$$W = \frac{1-v}{2\pi\mu} \int \frac{N/2\pi a}{\sqrt{a^2 - \rho^2}} \cdot \frac{d\xi dy}{\sqrt{(x-\xi)^2 + (y-\eta)^2}}$$



$$v = \rho_1 - r \cos \psi$$

$$a^2 - r^2 \sin^2 \psi = c^2$$

$$\rho^2 = v^2 + r^2 \sin^2 \psi$$

$$\rho_1 = \sqrt{(x-\xi)^2 + (y-\eta)^2}$$

$$\sqrt{a^2 - \rho^2} = \sqrt{c^2 - v^2}$$

$$\rho_1^2 = r^2 \cos^2 \psi + v^2$$

$$W = \frac{(1-v)N}{4\pi^2\mu a} \int_{\psi} \frac{d\xi dy}{\sqrt{(x-\xi)^2 + (y-\eta)^2} (\sqrt{a^2 - \rho^2})} = \frac{(1-v)N}{4\pi^2\mu a} \int \frac{R_1 d\psi d\rho_1}{\rho_1 (\sqrt{a^2 - \rho^2})}$$

$$\frac{2(1-v)N}{4\pi^2\mu a} \int_{-c}^c \int_{-c}^c \frac{dv d\psi}{\sqrt{c^2 - v^2}} \quad \psi_1 = \arcsin \frac{v}{r}, \quad i.e. c=0$$

$$\frac{2(1-v)N}{4\pi^2\mu a} \int \sin^{-1} \frac{v}{r} \Big|_{-c}^c \int_0^{\arcsin \frac{v}{r}} d\psi$$

$$\frac{(1-v)N}{3\pi\mu a} \arcsin \frac{a}{r}$$

$$\begin{aligned} & \frac{2\mu}{4(1-2v)} \left[\frac{1}{4(1-2v)} \left\{ \cos^2 \varphi \left(\frac{-3A}{r^3} + \frac{12C}{r^5} \right) - \frac{3C \sin^2 \varphi}{r^5} + \frac{D}{r^3} \right\} + \frac{v}{4(1-2v)} \left\{ \cos^2 \varphi \left[\frac{-3A}{r^3} + \frac{12C}{r^5} \right] - \frac{\sin^2 \varphi}{r^3} \left[\frac{2A(1-2v)}{r^3} + \frac{9C}{r^5} \right] + \frac{D}{r^5} \right\} \right] \\ & - 3A = 10Av - 8Av^2 - A(3+10v+8v^2) \\ & + \frac{\cos^2 \varphi}{4(1-2v)} \left[\frac{2A(5-4v)}{r^3} - \frac{24C}{r^5} \right] + \frac{12C}{r^3} \sin^2 \varphi - \frac{2D}{r^5} \\ & \frac{2\mu}{4(1-2v)} \left[\frac{\cos^2 \varphi}{r^3} \left[\frac{-3A}{1-v} + \frac{2A(5-4v)}{1-v} \right] + \frac{\cos^2 \varphi}{r^5} \left[\frac{12C}{1-v} + \frac{12Cv}{1-v} \right] - \frac{24Cv}{1-v} \right] + \frac{\sin^2 \varphi}{r^5} \left[-3C - \frac{9Cv}{1-v} + \frac{12Cv}{1-v} \right] - \frac{3C+3Cv+3Cv}{1-v} = \frac{3C(1-2v)}{1-v} \\ & + \frac{D}{r^3} \left[1 + \frac{v}{1-v} - \frac{4v}{1-v} \right] \end{aligned}$$

$$\therefore \frac{2\mu}{2A(1-2v)^2} \left\{ \frac{\cos^2 \varphi}{r^3} \left[-A(1-2v)(3-4v) \right] + \frac{\cos^2 \varphi}{r^5} \left[\frac{12C(1-2v)}{1-v} \right] + \frac{\sin^2 \varphi}{r^5} \left[-3C(1-2v) \right] - \frac{\sin^2 \varphi}{r^3} \left[\frac{2A(1-2v)}{1-v} \right] + \frac{D}{r^3} \left(\frac{1-2v}{1-v} \right) \right\}$$

$$\frac{\mu}{2(1-v)} \left[\left\{ \frac{(4v-3)A}{r^3} + \frac{12C}{r^5} \right\} \cos^2 \varphi + \sin^2 \varphi \left[\frac{-3C}{r^5} - \frac{2Av}{r^3} \right] + \frac{D}{r^3} \right]$$

$$\frac{\mu}{2(1-v)R^3} \left[-\frac{3C}{R^2} - 2Av + D \right]$$

$$\frac{\cancel{\lambda}}{2(\cancel{\lambda}\cancel{\lambda})} \left[\frac{+3(\cancel{\lambda}\cancel{\lambda})T}{\cancel{\lambda}(7-5v)} + 2v \cdot \frac{5(\cancel{\lambda}\cancel{\lambda})T}{\cancel{\lambda}(7-5v)} - \frac{(\cancel{\lambda}\cancel{\lambda})(6-5v)T}{\cancel{\lambda}(7-5v)} \right] = \frac{T}{2(7-5v)} \left[\begin{matrix} 3+10v-6+5v \\ -3+15v \end{matrix} \right]$$

~~cancel~~

$$= 3T \quad \frac{1-5v}{2(7-5v)}$$

Problem Set #2

1. For a center of dilatation located at $(0,0,0)$ $u_i = -\frac{c}{4(1-\nu)} \nabla \left(\frac{1}{r}\right)$ where $r = (x_i x_i)^{1/2}$

For a center of compression located at $(h,0,0)$ we find that for the same strength

$$u_{i2} = -\frac{c}{4(1-\nu)} \nabla \left(\frac{-1}{r'}\right) \text{ where } r = [(x_i - h\delta_{ii})(x_i - h\delta_{ii})]^{1/2}$$

Thus in an infinite body affected by a center of dilatation & a center of compression a distance h away gives (for a linearly elastic body where superposition holds)

$$u_i = u_{i1} + u_{i2} = -\frac{c}{4(1-\nu)} \nabla \left(\frac{1}{r} - \frac{1}{r'}\right);$$

for small h we can write that $\frac{1}{r'} = \frac{1}{r} - h \frac{\partial}{\partial x} \left(\frac{1}{r'}\right) \Big|_{h=0} + \frac{h^2}{2!} \frac{\partial^2}{\partial x^2} \left(\frac{1}{r'}\right) \Big|_{h=0} + \dots$

$$\text{thus } \frac{1}{r} - \frac{1}{r'} = h \frac{\partial}{\partial x} \left(\frac{1}{r'}\right) \Big|_{h=0} - \frac{h^2}{2!} \frac{\partial^2}{\partial x^2} \left(\frac{1}{r'}\right) \Big|_{h=0} + \dots$$

$$\therefore u_i = -\frac{c}{4(1-\nu)} \nabla \left[h \frac{\partial}{\partial x} \left(\frac{1}{r'}\right) \Big|_{h=0} - \frac{h^2}{2!} \frac{\partial^2}{\partial x^2} \left(\frac{1}{r'}\right) \Big|_{h=0} + \dots \right]; \text{ if } \lim_{h \rightarrow 0} Ch \stackrel{\Delta}{=} D$$

$$\text{thus } u_i = -\frac{D}{4(1-\nu)} \nabla \left(\frac{\partial}{\partial x} \left[\frac{1}{r}\right]\right) = -\frac{D}{4(1-\nu)} \frac{\partial}{\partial x} \nabla \left(\frac{1}{r}\right) = \frac{D}{4(1-\nu)} \nabla \left(\frac{x}{r}\right)$$

$$\text{and } u_i = \frac{D}{4(1-\nu)r^3} \left[\delta_{ii} - \frac{3x_i x_i}{r^2} \right] \quad \boxed{20/20}$$

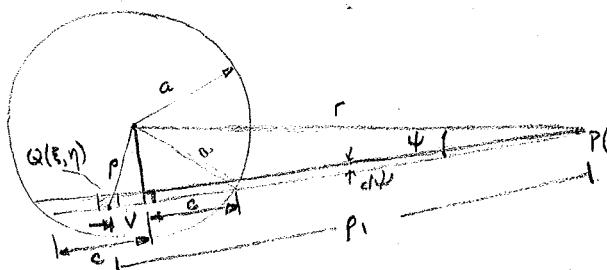
another method: if $u_{i2} = u_{i1}(h,0,0)$ then $u_{i2} = u_{i1} - h \frac{\partial u_{i1}}{\partial x} \Big|_{h=0} + \dots$ where u_{i1} is disp of center of dilatation. For center of compression $u_{i2} = -[u_{i1} - h \frac{\partial u_{i1}}{\partial x} \Big|_{h=0} + \dots]$

For center of comp + center of dilatation $u_{i1} + u_{i2} = h \frac{\partial u_{i1}}{\partial x} \Big|_{h=0} + O(h^2) = -\frac{Ch}{4(1-\nu)} \frac{\partial}{\partial x} \left(\nabla \frac{1}{r}\right) + O(h^2)$

using $\lim_{h \rightarrow 0} Ch \stackrel{\Delta}{=} D$ then $u_i = \lim_{h \rightarrow 0} [u_{i1} + u_{i2}] = -\frac{D}{4(1-\nu)} \frac{\partial}{\partial x} \left(\nabla \frac{1}{r}\right)$ as before.

$$3. w = \frac{1-\nu}{2\pi\mu} \int \frac{N/2\pi a r}{\sqrt{a^2 - \rho^2}} \frac{d\xi dy}{\sqrt{(x-\xi)^2 + (y-\eta)^2}} \quad \text{where } N = \int \rho(\xi, \eta) d\xi d\eta$$

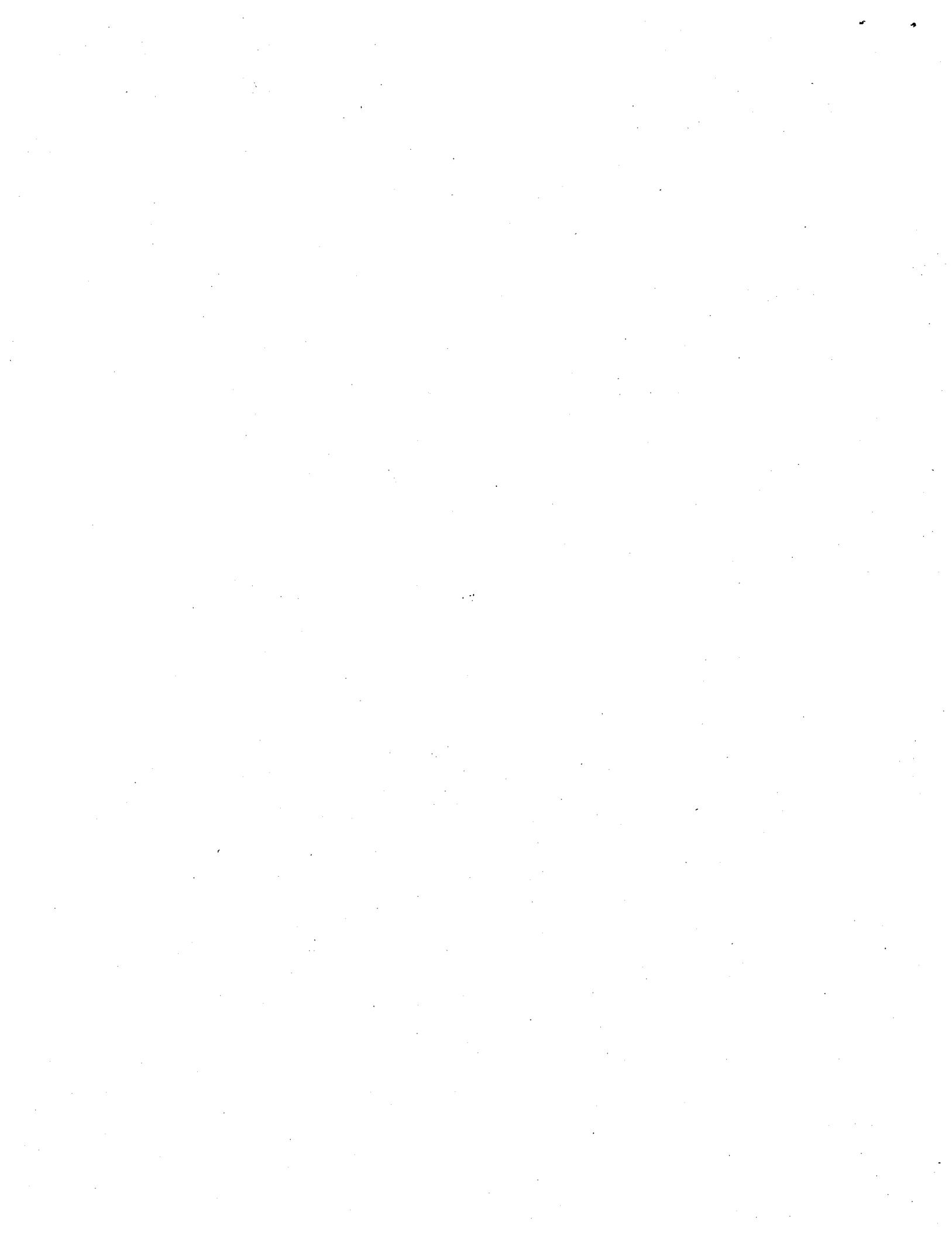
For points outside



$$v = \rho_1 r \cos \psi$$

$$\left. \begin{aligned} a^2 - r^2 \sin^2 \psi &= c^2 \\ \rho^2 &= v^2 + r^2 \sin^2 \psi \end{aligned} \right\} \Rightarrow \sqrt{a^2 - \rho^2} = \sqrt{c^2 - v^2}$$

$$\rho_1 = \sqrt{(x-\xi)^2 + (y-\eta)^2}$$



Now if $P(x, y)$ is a fixed point and we draw the rays as shown so that $Q(\xi, \eta)$ is any point in the circle, then the area $d\xi d\eta = p_1 d\psi dp_1$,

$$\therefore w = \frac{1-v}{4\pi^2\mu a} N \int \frac{p_1 d\psi dp_1}{\sqrt{a^2 - p_1^2}} = \frac{(1-v)N}{4\pi^2\mu a} \int \frac{d\psi dp_1}{\sqrt{a^2 - p_1^2}} \quad \text{where } \psi \text{ goes from } -\psi_1 \text{ to } +\psi_1 \text{ and}$$

$\psi_1 = \arcsin \frac{y}{r}$ along constant ψ lines. $dv = dp_1$ and v goes from $-c$ to $+c$. Thus

$$w = \frac{(1-v)N}{4\pi^2\mu a} \int_{-\psi_1}^{\psi_1} \int_{-c}^c \frac{dv d\psi}{\sqrt{c^2 - v^2}} = \frac{(1-v)N}{4\pi^2\mu a} \int_{-\psi_1}^{\psi_1} d\psi \int_{-c}^c \frac{dv}{\sqrt{c^2 - v^2}} = \frac{(1-v)N}{4\pi^2\mu a} \cdot 2\psi_1 \cdot \arcsin \frac{c}{v} \Big|_{-c}^c$$

$$= \frac{(1-v)N}{4\pi^2\mu a} \cdot 2 \arcsin \frac{y}{r} \cdot \pi = \frac{(1-v)N}{2\pi\mu a} \arcsin \frac{y}{r}$$

$$\therefore w(x, y, z=0) = \frac{(1-v)N}{2\pi\mu a} \arcsin \frac{y}{r} \quad r = \sqrt{x^2 + y^2} \quad 29/20$$

2. Starting with $u = B - \frac{1}{4(1-v)} \nabla(r \cdot B + \beta)$ with $r = R e_R$ & $\nabla = e_R \frac{\partial}{\partial R} + \frac{e_\theta}{R} \frac{\partial}{\partial \theta} + \frac{e_\phi}{R \sin \theta} \frac{\partial}{\partial \phi}$

we find that we need $\frac{\partial}{\partial z}$ in spherical coordinates

$$\nabla = e_x \frac{\partial}{\partial x} + e_y \frac{\partial}{\partial y} + e_z \frac{\partial}{\partial z} \Rightarrow e_z \cdot \nabla = \frac{\partial}{\partial z} = e_z \cdot e_R \frac{\partial}{\partial R} + e_z \cdot e_\theta \frac{1}{R} \frac{\partial}{\partial \theta} + e_z \cdot e_\phi \frac{1}{R \sin \theta} \frac{\partial}{\partial \phi}$$

from a previous homework we had found $e_z \cdot e_R = \cos \phi \quad e_z \cdot e_\theta = -\sin \phi \quad e_z \cdot e_\phi = 0$

$$\therefore \frac{\partial}{\partial z} = \cos \phi \frac{\partial}{\partial R} - \frac{\sin \phi}{R} \frac{\partial}{\partial \theta} \quad \therefore \frac{\partial}{\partial z} \left(\frac{1}{R} \right) = \cos \phi \left(-\frac{1}{R^2} \right) = -\frac{\cos \phi}{R^2}$$

To do this problem we need to find $T_{\theta\theta}$. But we are given

$$B_3' = A \frac{\partial}{\partial z} \left(\frac{1}{R} \right) \quad \text{and} \quad \beta'' = \frac{D}{R} + \frac{\partial^2}{\partial z^2} \left(\frac{C}{R} \right) \quad \text{where} \quad A = -\frac{5(1-v)a^3 T}{\mu(7-5v)} \quad D = -\frac{(1-v)(6-5v)a^3 T}{\mu(7-5v)}$$

$$\text{and} \quad C = -\frac{(1-v)a^5 T}{\mu(7-5v)} ; \quad B_1 \text{ and } B_2 \text{ are also given as zero}$$

$$B'' = A \frac{\partial}{\partial z} \left(\frac{1}{R} \right) e_z = B_3 e_z = -A \frac{\cos \phi}{R^2} [\cos \phi e_R - \sin \phi e_\phi]$$

$$r \cdot B'' = R e_R \left[-\frac{A \cos \phi}{R^2} \cos \phi e_R + \text{term which gives zero} \right] = -\frac{A \cos^2 \phi}{R}$$

$$\beta'' = \frac{D}{R} + C \frac{\partial}{\partial z} \left[\frac{1}{R} \right] = \frac{D}{R} + C \frac{\partial}{\partial z} \left(-\frac{\cos \phi}{R^2} \right) = \frac{D}{R} + C \left\{ \frac{2 \cos^2 \phi}{R^3} - \frac{\sin^2 \phi}{R^3} \right\}$$

plugging these into u and using ∇ in spherical we obtain u and its components are

$$u_R'' = \frac{\cos^2 \phi}{4(1-v)} \left[-\frac{A(5-4v)}{R^2} + \frac{3C}{R^4} \right] + \frac{D}{4(1-v)R^2} + \frac{3C}{4(1-v)} \frac{\cos^2 \phi}{R^4} ; \quad u_\theta'' = \frac{\sin 2\phi}{4(1-v)} \left[\frac{A(1-2v)}{R^2} + \frac{3C}{R^4} \right]$$

$$u_\phi'' = 0$$



Plugging into

$$\epsilon_{RR}'' = \frac{\partial u_R}{\partial R} = \frac{\cos^2 \varphi}{4(1-v)} \left[\frac{2A(5-4v)}{R^3} - \frac{12C}{R^5} \right] - \frac{2D}{4(1-v)R^3} - \frac{12C}{4(1-v)} \frac{\cos 2\varphi}{R^5} \quad \cos 2\varphi = 2\cos^2 \varphi - 1$$

$$\epsilon_{\theta\theta}'' = \frac{u_R}{R} + u_\theta \frac{\cos 2\varphi}{R^2} = \frac{\cos^2 \varphi}{4(1-v)} \left[-\frac{3A}{R^3} + \frac{9C}{R^5} \right] + \frac{D}{4(1-v)R^3} + \frac{3C}{4(1-v)} \frac{\cos 2\varphi}{R^5}$$

$$\epsilon_{\varphi\varphi}'' = \frac{1}{R} \frac{\partial u_\varphi}{\partial \varphi} + \frac{u_R}{R} = \frac{\cos 2\varphi}{4(1-v)} \left[\frac{2A(1-2v)}{R^3} + \frac{9C}{R^5} \right] + \frac{\cos^2 \varphi}{4(1-v)} \left[-\frac{A(5-4v)}{R^3} + \frac{3C}{R^5} \right] + \frac{D}{4(1-v)R^3}$$

$$\epsilon_{R\theta}'' = \epsilon_{\theta\varphi}'' = 0$$

$$\epsilon_{\varphi R}'' = \frac{\sin 2\varphi}{4(1-v)} \left[\frac{A(1+v)}{R^3} - \frac{12C}{R^5} \right]$$

from these we get $\sigma_{R\theta\theta}'' = 0$ $\sigma_{\theta\varphi\varphi}'' = 0$ $\sigma_{\varphi R R}'' = 2\mu \epsilon_{\varphi R}'' = \mu \frac{\sin 2\varphi \cos \varphi}{(1-v)} \left[\frac{A(1+v)}{R^3} - \frac{12C}{R^5} \right]$

$$\sigma_{RR}'' = \lambda \epsilon_{RR}'' \delta_{RR} + 2\mu \epsilon_{RR}'' = (\lambda + 2\mu) \epsilon_{RR}'' + \lambda (\epsilon_{\theta\theta}'' + \epsilon_{\varphi\varphi}'') = \frac{\mu \cos^2 \varphi}{(1-v)R^3} \left[(5-v)A - \frac{18C}{R^2} \right] + \frac{\mu}{(1-v)R^3} \left[-2A - D + \frac{6C}{R^2} \right]$$

$$\sigma_{\theta\theta}'' = 2\mu \left[\frac{1-v}{1-2v} \epsilon_{\theta\theta}'' + \frac{v}{1-2v} (\epsilon_{\varphi\varphi}'' + \epsilon_{R\theta\theta}'') \right] = \frac{\mu}{2(1-v)} \left[\left\{ \frac{(4v-3)A}{R^3} + \frac{12C}{R^5} \right\} \cos^2 \varphi + \lambda \sin^2 \varphi \left\{ \frac{-3C}{R^5} - \frac{2Av}{R^3} \right\} + \frac{D}{R^3} \right]$$

$$\sigma_{\varphi\varphi}'' = \frac{\mu}{2(1-v)R^3} \left[\cos^2 \varphi \left[(2v-1)A + \frac{21C}{R^2} \right] - 2(1-v)A - \frac{9C}{R^2} + D \right] \left\{ \begin{array}{l} \frac{3(2v-1)A}{R^3} + \frac{15C}{R^5} \\ \cos^2 \varphi \\ \frac{-3C}{R^5} - \frac{2Av}{R^3} \end{array} \right\}$$

$$\sigma_{R\theta\theta\text{tot}} = \sigma_{RR}'' + T \cos^2 \varphi = 0 \text{ or } r = a + t \varphi \text{ (show)} \quad \sigma_{R\varphi\varphi\text{tot}} = \sigma_{\varphi\varphi}'' - T \sin^2 \varphi \quad \sigma_{\varphi\varphi\text{tot}} = \sigma_{\varphi\varphi}'' + T \sin^2 \varphi \quad \sigma_{\theta\theta\text{tot}} = \sigma_{\theta\theta}'' = ? \text{ on } r = a$$

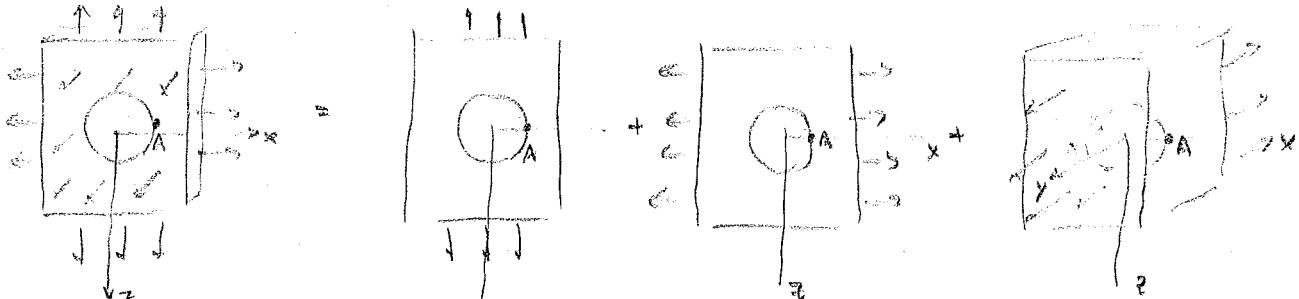
$$\sigma_{R\theta\theta\text{tot}} = 0 \quad \sigma_{\theta\varphi\varphi\text{tot}} = 0$$

We find that $\sigma_{\varphi\varphi\text{max}}$ occurs on $\varphi = 90^\circ$ plane since both A, C are negative and $2v-1 < 0$ for $0 < v < \frac{1}{2}$ and $\frac{\partial^2 \sigma_{\varphi\varphi}}{\partial \varphi^2} = -\cos 2\varphi$ negative no. \Rightarrow for max $\Rightarrow \cos 2\varphi > 0$ but $\frac{\partial \sigma_{\varphi\varphi}}{\partial \varphi} = 0 \Rightarrow \varphi = 90^\circ, 0^\circ \Rightarrow$ max at $\varphi = 90^\circ$

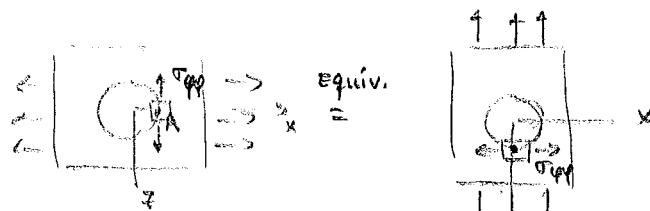
We also find that $\sigma_{\theta\theta\text{max}}$ occurs at $\varphi = 90^\circ$

We also find that $\sigma_{\varphi\varphi} = 0$ at $\varphi = 90^\circ$ $\therefore \sigma_{RR}, \sigma_{\varphi\varphi}, \sigma_{\theta\theta}$ are indeed principal stresses at $\varphi = 90^\circ$
They are principal everywhere on $r = a$

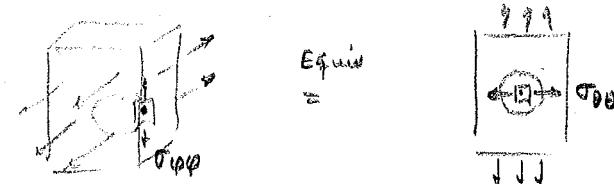
Now let us look at point A



The second picture



The third picture



$$\therefore \sigma_{\phi\phi} \Big|_{at A} = \sigma_{\phi\phi} \Big|_{\varphi=90^\circ} + \sigma_{\phi\phi} \Big|_{\varphi=0^\circ} + \sigma_{\theta\theta} \Big|_{\varphi=90^\circ}$$

but in class we calculated $\sigma_{\phi\phi} \Big|_{\varphi=90^\circ} = \frac{3(9-5v)}{2(7-5v)} T$; also $\sigma_{\phi\phi} \Big|_{\varphi=0^\circ} = -\frac{3(1+5v)}{2(7-5v)} T$ and $\sigma_{\theta\theta} \Big|_{\varphi=90^\circ} = -\frac{3}{2} \frac{(1-5v)}{7-5v} T$

when we add these 3 together we get $\frac{3T}{2} \checkmark$

$\frac{26}{30}$



DIVISION OF APPLIED MECHANICS
DEPARTMENT OF MECHANICAL ENGINEERING
STANFORD UNIVERSITY

ME 238C Theory of Elasticity

Spring 1979

Problem Set No. 3

1. Show that the displacement equations of equilibrium are satisfied in terms of a vector \mathbf{g} if one takes

$$U_1 = \nabla^2 \mathbf{g} - \frac{1}{2(1-\nu)} \nabla(\nabla \cdot \mathbf{g})$$

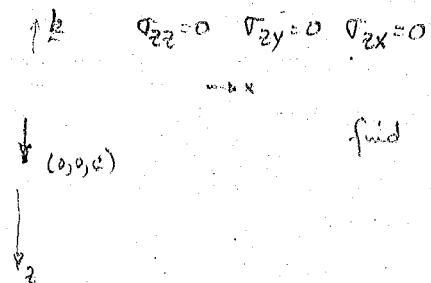
and provided \mathbf{g} (Galerkin vector) is biharmonic. Express the Boussinesq-Papkovich functions in terms of the Galerkin vector.

2. Verify the expressions for the components of stress σ_{rr} , $\sigma_{r\phi}$ derived in class due to displacement potentials

don't multiply

$$B_z = A \frac{\partial}{\partial z} \left(\frac{1}{r} \right); \quad \beta = -\frac{D}{r} + \frac{\partial^2}{\partial z^2} \left(\frac{c}{r} \right)$$

3. Consider a half-space with a traction-free bounding plane at $z = 0$ and subjected to a concentrated force in the z direction acting at $(0, 0, c)$. Calculate the displacement in the z direction.





$$u = V^2 G - \frac{1}{2(1-\nu)} \nabla (\nabla \cdot G)$$

$$\text{if } u = V\phi + V \times H \text{ then } \phi = -\frac{1}{4\pi} \int_V \frac{\nabla \cdot u}{R} dV \quad H = \frac{1}{4\pi} \int_V \frac{\nabla \times u}{R} dV$$

$$u = \nabla \cdot B - \frac{V}{2(1-\nu)} [r \cdot B + \beta]$$

$$\Rightarrow B = V^2 G \text{ and } r \cdot B + \beta = V \cdot G \therefore r \cdot \frac{V^2 G}{2} + \beta = V \cdot G$$

$$\beta = 2V \cdot G - r \cdot V^2 G \quad V^2(a \cdot b) = 0$$

$$V^2 \beta = 2V \cdot G - 2V \cdot V^2 G = V^2 (r \cdot V^2 G)$$

what to show that u satisfies $\nabla^2 u + \frac{1}{1-2\nu} \nabla (\nabla \cdot u) = 0$

$$\text{now } \nabla \cdot u = \nabla \cdot (V^2 G) = \frac{1}{2(1-\nu)} \nabla \cdot [\nabla (V \cdot G)]$$

$$= \frac{\partial e_i \cdot \frac{\partial^2 G_j}{\partial x_i \partial x_j}}{\partial x_k} e_k = \frac{\partial e_i \cdot \frac{\partial e_k}{\partial x_k} \frac{\partial G_j}{\partial x_j}}{\partial x_i} = V^2 (V \cdot G) = \frac{1-2\nu}{2(1-\nu)} V \cdot G$$

$$\frac{1}{1-2\nu} \nabla (\nabla \cdot u) = \frac{V^2}{2(1-\nu)} \nabla (V \cdot G)$$

$$\frac{\nabla (V \cdot u)}{1-2\nu} + \nabla^2 u = V^4 G - \frac{V^2}{2(1-\nu)} \nabla (V \cdot G) + \frac{V^2}{2(1-\nu)} V (V \cdot G) = 0 \quad \text{since } G \text{ satisfies } V^4 G = 0$$

and $f=0$

$$\nabla^2 \beta = 2 \nabla \cdot \nabla^2 G - \nabla^2 (r \cdot \nabla^2 G)$$

$$2 \nabla \cdot \nabla^2 G - 2 \nabla^2 (V \cdot G) = \frac{\partial}{\partial x_k} \left[\frac{\partial e_k}{\partial x_l} \frac{\partial}{\partial x_l} \left[\frac{\partial^2 G_i}{\partial x_j \partial x_j} \right] \right] = \frac{\partial^2 e_k}{\partial x_k \partial x_l} \frac{\partial}{\partial x_l} \left[x_i \frac{\partial^2 G_i}{\partial x_j \partial x_j} \right]$$

$$+ r \cdot \nabla^4 G = \frac{\partial}{\partial x_k} \left[\frac{\partial^2 e_k}{\partial x_l \partial x_l} \left\{ \delta_{il} \frac{\partial^2 G_i}{\partial x_j \partial x_j} e_{jl} + x_i \frac{\partial^3 G_i}{\partial x_l \partial x_j \partial x_j} e_{jl} \right\} \right]$$

$$+ \frac{\partial}{\partial x_k} \left[\frac{\partial^2 e_k}{\partial x_l \partial x_l} \left\{ \frac{\partial^2 G_i}{\partial x_j \partial x_j} e_{il} + x_i \frac{\partial^3 G_i}{\partial x_l \partial x_j \partial x_j} e_{il} \right\} \right]$$

$$+ \frac{\partial}{\partial x_k} \left[\frac{\partial^2 e_k}{\partial x_l \partial x_l} \delta_{il} + \delta_{ik} \frac{\partial^3 G_i}{\partial x_l \partial x_j \partial x_j} \right]$$

$$+ x_i \frac{\partial^3 G_i}{\partial x_k \partial x_l \partial x_j}$$

$$\nabla^2 (V \cdot G) + \nabla^2 (V \cdot G) + x_i \nabla^4 G;$$

$$u_1 = \nabla^2 G - \frac{1}{2(1-\nu)} \nabla (\nabla \cdot G) \quad \text{must satisfy } \sigma_{ij,j} = 0 \quad \text{or} \quad \nabla^2 u_1 + \frac{1}{1-2\nu} \nabla (\nabla \cdot u_1) = 0$$

$$\begin{aligned}\nabla(\nabla \cdot G) &= \nabla \cdot \nabla G + (G \cdot \nabla) \nabla + \nabla \times (\nabla \times G) + G \times (\nabla \times \nabla) \\ &= \nabla^2 G + (G \cdot \nabla) \nabla + \nabla(G \cdot \nabla) - \nabla^2 G + G \times (\nabla \times \nabla)\end{aligned}$$

$$\text{using } \nabla \times (\nabla \times G) = \nabla(\nabla \cdot G) - \nabla^2 G$$

$$u_1 =$$

$$(\nabla \cdot \nabla) G = \nabla(\nabla \cdot G) - \nabla \times \nabla \times G = \nabla(\nabla \cdot G) - [\nabla(\nabla \cdot G) - (\nabla \cdot \nabla) G]$$

$$u_1 = \nabla(\nabla \cdot G) - \nabla \times \nabla \times G - \frac{1}{2(1-\nu)} \nabla(\nabla \cdot G)$$

$$\Rightarrow \frac{\cancel{1}(1-\nu)}{\cancel{2}(1-\nu)} = \frac{1}{2} \nabla(\nabla \cdot G) - \cancel{\nabla \times \nabla \times G}$$

$$u_1 = \frac{1-2\nu}{2(1-\nu)} \nabla(\nabla \cdot G) - \nabla \times \nabla \times G$$

$$\nabla \cdot u_1 = \frac{1-2\nu}{2}$$

$$\nabla \cdot u_1 = \nabla^2(\nabla \cdot G) - \frac{1}{2(1-\nu)} \nabla^2(\nabla \cdot G) = \nabla^2(\nabla \cdot G) - \left[\frac{2-2\nu-1}{2(1-\nu)} \right] = \frac{1-2\nu}{2(1-\nu)} \nabla^2(\nabla \cdot G)$$

$$\nabla(\nabla \cdot u_1) = \frac{1-2\nu}{2(1-\nu)} \nabla^2[\nabla(\nabla \cdot G)] \quad \frac{1}{1-2\nu} \nabla(\nabla \cdot u_1) = \frac{1}{2(1-\nu)} \nabla^2[\nabla(\nabla \cdot G)]$$

$$\cancel{\nabla^2 u_1 = \nabla^2 G}$$

$$\nabla \cdot \{ \nabla(\nabla \cdot G) \} =$$

$$u_1 = \nabla^2 G - \frac{1}{2(1-\nu)} \nabla(\nabla \cdot G)$$

$$\nabla \cdot u_1 = \nabla^2(\nabla \cdot G) + (G \cdot \nabla) \nabla^2 - \frac{1}{2(1-\nu)} \nabla^2(\nabla \cdot G)$$

$$\Rightarrow \frac{1-2\nu}{2(1-\nu)} \nabla^2(\nabla \cdot G) + (G \cdot \nabla) \nabla^2$$

$$\frac{1}{1-2\nu} \nabla(\nabla \cdot u_1) = \frac{\nabla^2 \nabla(\nabla \cdot G)}{2(1-\nu)} - \frac{\nabla^2 \nabla(G \cdot \nabla)}{1-2\nu}$$

$$w \cdot v$$

$$\begin{aligned}
& \left(\frac{1}{R_2} \right) = - \frac{1}{R_2^2} \frac{\partial R_2}{\partial z} \quad \frac{\partial R_2}{\partial z} = \frac{1}{2} R_2^{-1} \cdot 2(z+c) = \frac{z+c}{R_2} \\
& \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) = - \frac{(z+c)}{R_2^3} - \frac{(z+c)^2 \left(\frac{3}{R_2^5} \right)}{R_2^5} = - \frac{(x^2+y^2+(z+c)^2) - 3(z+c)^2}{R_2^5} = - \frac{x^2+y^2-2(z+c)^2}{R_2^5} \\
& \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) = \frac{2}{R_2} \left(- \frac{z+c}{R_2^3} \right) = - \frac{1}{R_2^3} - \frac{(z+c)(\frac{3}{R_2^5})}{R_2^5} = - \frac{1}{R_2^3} - \frac{2}{(z+c)^3} \\
& \frac{1}{z+c} \quad \frac{\partial}{\partial z} \left(- \frac{1}{(z+c)^2} \right) = \frac{2}{(z+c)^3} \\
& u_2 = \frac{1}{4(1-v)} \frac{P_2}{4\pi\mu} \left\{ \frac{9-24v+16v^2}{R_2} - \frac{5-12v+8v^2}{R_2} + 2Cz \frac{(x^2+y^2-2(z+c)^2)}{R_2^5} + \frac{(z-c)^2}{R_1^3} + \frac{3-4v}{R_1} \right. \\
& \left. + 2C \frac{(3-4v)(z+c)}{R_2^3} + \frac{(3-4v)(z-c)(z+c)}{R_2^3} \right. \\
& \left. - \frac{(3-4v)(z+c)^2}{R_2^3} \right\} \quad \cancel{[z=c]} \\
& \frac{\partial B_2}{\partial z} = \frac{P_2}{4\pi\mu} \left[- \frac{1}{(z-c)^2} + \frac{3-4v}{(z+c)^2} + \frac{4c}{(z+c)^3} \right] \quad + 2C \frac{1}{(z+c)^2} - \frac{4c}{(z+c)^3} \\
& B_2 = \frac{P_2}{4\pi\mu} \left[\frac{1}{z-c} + \frac{3-4v}{z+c} + \frac{2c}{(z+c)^2} \right] \\
& \frac{\partial B}{\partial z} = \frac{P_2}{4\pi\mu} \left[\frac{4(1-2v)(1-v)}{z+c} + \frac{c}{(z-c)^2} + \frac{(3-4v)c}{(z+c)^2} \right] \\
& \frac{(3-4v)}{z-c} + \frac{(3-4v)}{z+c} + \frac{2C(3-4v)}{(z+c)^2} + \frac{2}{(z-c)^2} + \frac{(3-4v)z}{(z+c)^2} + \frac{4zc}{(z+c)^3} - 4(1-\frac{3v+2v^2}{z+c}) - \frac{c}{(z-c)^2} - \frac{(3-4v)c}{(z+c)^2} \\
& \frac{5-12v+8v^2}{z+c} + \frac{(z+c)(3-4v)}{(z+c)^2} = \frac{8(1-v)^2}{(z+c)} + \frac{2c}{(z-c)} + \frac{3-4v}{z-c} \\
& \frac{8(1-v)^2}{z+c} + \frac{4(1-v)}{z-c} + \frac{4zc}{(z+c)^3} \\
& u_2 = \frac{P_2}{16\pi\mu(1-v)} \left[\frac{5-12v+8v^2}{R_2} + \frac{(3-4v)(z+c)^2}{R_2^3} - 2Cz \frac{[R_2^2-3(z+c)^2]}{R_2^5} + \frac{(z-c)^2}{R_1^3} + \frac{3-4v}{R_1} \right]
\end{aligned}$$



Cesar Levy
Spring 79

Problem Set #3

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Given $u_1 = \nabla^2 G - \frac{1}{2(1-\nu)} \nabla(\nabla \cdot G)$ and $\nabla^4 G = 0$ prove that it satisfies the displacement equilibrium Eqns and express the Boussinesq-papkovich functions in terms of the Galerkin vector G .

$\nabla \cdot u_1 = \nabla^2(\nabla \cdot G) - \frac{1}{2(1-\nu)} \nabla^2(\nabla \cdot G) = \frac{1-2\nu}{2(1-\nu)} \nabla^2(\nabla \cdot G)$

and $\frac{1}{1-2\nu} \nabla(\nabla \cdot u_1) = \frac{\nabla^2}{2(1-\nu)} \nabla(\nabla \cdot G)$ since ∇^2 is a linear operator and commutes with ∇

Now $\nabla^2 u_1 = \nabla^4 G - \frac{\nabla^2}{2(1-\nu)} \nabla(\nabla \cdot G)$; therefore

$\nabla^2 u_1 + \frac{1}{1-2\nu} \nabla(\nabla \cdot u_1) = \nabla^4 G$

but since G satisfies $\nabla^4 G = 0$ then $\nabla^2 u_1 + \frac{1}{1-2\nu} \nabla(\nabla \cdot u_1) = 0$. But this is precisely the equilibrium equations (with $f=0$) written in terms of displacements. ✓

since $u_1 = \nabla^2 G - \frac{1}{2(1-\nu)} \nabla(\nabla \cdot G) = B - \frac{1}{4(1-\nu)} \nabla(r \cdot B + \beta)$

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then $\boxed{\nabla^2 G = B}$ and $\nabla \cdot G = \frac{r \cdot B + \beta}{2}$ thus $\boxed{2\nabla \cdot G - r \cdot \nabla^2 G = \beta}$

2. Beginning with $u_1 = B - \frac{1}{4(1-\nu)} \nabla(r \cdot B + \beta)$ and defining $r = r\hat{e}_r$ and

$\nabla = \hat{e}_r \frac{\partial}{\partial r} + \frac{\hat{e}_\theta}{r} \frac{\partial}{\partial \theta} + \frac{\hat{e}_\phi}{r \sin \theta} \frac{\partial}{\partial \phi}$ then $\frac{\partial}{\partial z} = \cos \theta \frac{\partial}{\partial r} - \frac{\sin \theta}{r} \frac{\partial}{\partial \phi}$

$B_z'' = A \frac{\partial}{\partial z} \left(\frac{1}{r} \right) = -A \frac{\cos \theta}{r^2}$ then $B = B_z \hat{e}_z = -A \frac{\cos \theta}{r^2} [\cos \theta \hat{e}_r - \sin \theta \hat{e}_\phi]$

$B_z = \frac{D}{r} + C \frac{\partial}{\partial z} \left[\frac{\partial}{\partial z} \left(\frac{1}{r} \right) \right] = \frac{D}{r} + C \left\{ \frac{2 \cos^2 \theta}{r^3} - \frac{\sin^2 \theta}{r^3} \right\}$

we find that

Plugging all these into u_1 we find that

$u_r'' = \frac{\cos^2 \theta}{4(1-\nu)} \left\{ -A \frac{(5-4\nu)}{r^2} + \frac{6C}{r^4} \right\} + \frac{D}{4(1-\nu)r^2} + \frac{3C}{4(1-\nu)} \frac{\sin^2 \theta}{r^4}$

$u_\theta'' = \frac{\cos^2 \theta}{4(1-\nu)} \left[A \frac{(1-2\nu)}{r^2} + \frac{3C}{r^4} \right]$

$u_\phi'' = 0$

$$\text{and } \epsilon_{rr}'' = \frac{\partial u_r''}{\partial r} = \frac{\cos^2 \theta}{4(1-\nu)} \left[2A \frac{(5-4\nu)}{r^3} - \frac{24C}{r^5} \right] - \frac{2D}{4(1-\nu)r^3} + \frac{12C \sin^2 \theta}{4(1-\nu)r^5}$$

$$\sin \frac{\partial u_\theta''}{\partial \theta} = 0$$

$$\epsilon_{\theta\theta}'' = \frac{u_r''}{r} + u_\theta'' \frac{\cot \theta}{r} = \frac{\cos^2 \theta}{4(1-\nu)} \left[-3A \frac{1}{r^3} + \frac{12C}{r^5} \right] - \frac{3C}{4(1-\nu)} \frac{\sin^2 \theta}{r^5} + \frac{D}{4(1-\nu)r^3}$$

$$\epsilon_{\phi\phi}'' = \frac{1}{r} \frac{\partial u_\theta''}{\partial \theta} + \frac{u_\phi''}{r} = \frac{\cos^2 \theta}{4(1-\nu)} \left[-3A \frac{1}{r^3} + \frac{12C}{r^5} \right] - \frac{\sin^2 \theta}{4(1-\nu)} \left[2A \frac{(1-2\nu)}{r^3} + \frac{9C}{r^5} \right] + \frac{D}{4(1-\nu)r^3}$$

$$\epsilon_{\theta\phi}'' = \frac{1}{2} \left\{ \frac{1}{r \sin \theta} \frac{\partial u_\theta''}{\partial \theta} + \frac{1}{r} \frac{\partial u_\phi''}{\partial \theta} - \frac{\cot \theta u_\phi''}{r} \right\} = 0$$

$$\frac{\partial u_r''}{\partial r} + \frac{\partial u_\theta''}{\partial r} - \frac{u_\phi''}{r} = 0$$



$$\text{and } \epsilon_{\phi r}'' = \frac{\sin \phi \cos \phi}{2(1-v)} \left[\frac{A(1+v)}{r^3} - \frac{12C}{r^5} \right] = \frac{1}{2} \left\{ \frac{\partial u_r}{\partial \phi} + \frac{\partial u_\phi}{\partial r} - \frac{u_\phi}{r} \right\}$$

20/20

$$\text{Now } \sigma_{\phi r}'' = 2\mu \epsilon_{\phi r}'' = \frac{\mu \sin \phi \cos \phi}{1-v} \left[\frac{A(1+v)}{r^3} - \frac{12C}{r^5} \right]$$

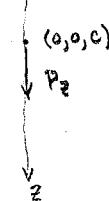
$$\text{and } \sigma_{rr}'' = (\lambda + 2\mu) \epsilon_{rr}'' + 2\mu (\epsilon_{\theta\theta}'' + \epsilon_{\phi\phi}'') = 2\mu \left[\frac{v}{1-2v} \epsilon_{rr}'' + \epsilon_{\theta\theta}'' + \epsilon_{\phi\phi}'' \right] = \frac{\mu}{(1-v)r^3} \left\{ \cos^2 \phi \left[(5-v)A - \frac{18C}{r^2} \right] + [vA + D - \frac{6C}{r^2}] \right\}$$

$$\text{now at } r=a \quad \sigma_{r\phi}'' = T \sin \phi \cos \phi \quad \text{and } \sigma_{rr}'' = -T \cos^2 \phi \Rightarrow A = \frac{6(1-v)a^3 T}{\mu(7-5v)} \quad C = +\frac{a^2 A}{5} \quad D = \left(\frac{6-5v}{a^2} \right) C$$

$$\text{Now } \sigma_{rr, \text{tot}} = \sigma_{rr}'' + T \cos^2 \phi \quad \text{and } \sigma_{r\phi, \text{tot}} = \sigma_{r\phi}'' - T \sin \phi \cos \phi$$

This problem is part one of Mindlin's Problem for which

$$\sigma_{zz}=0 \quad \sigma_{zx}=0 \quad \sigma_{zy}=0$$



it was found in class

$$B_z = \frac{P_2}{4\pi\mu} \left[\frac{1}{R_1} + \frac{3-4v}{R_2} - 2C \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) \right]$$

$$\text{and } \beta = \frac{P_2}{4\pi\mu} \left[-\frac{C}{R_1} - C \frac{(3-4v)}{R_2} + 4(1-2v)(1-v) \log(R_2 + z + C) \right]$$

$$\text{with } B_x = B_y = 0$$

$$R_1^2 = x^2 + y^2 + (z-c)^2$$

$$R_2^2 = x^2 + y^2 + (z+c)^2$$

$$\text{Thus we need to find } u_z \quad \text{or} \quad u_z = B_z - \frac{1}{4(1-v)} \frac{\partial}{\partial z} \left[v \cdot \beta B_z + \beta \right] = B_z - \frac{1}{4(1-v)} \left[B_z + z \frac{\partial B_z}{\partial z} + \frac{\partial \beta}{\partial z} \right]$$

$$\text{thus } u_z = \frac{1}{4(1-v)} \left[(3-4v)B_z + z \frac{\partial B_z}{\partial z} - \frac{\partial \beta}{\partial z} \right]$$

$$\text{also } \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) = -\frac{z+c}{R_2^3}; \quad \frac{\partial}{\partial z} \left(\frac{1}{R_1} \right) = -\frac{z-c}{R_1^3}; \quad \frac{\partial^2}{\partial z^2} \left(\frac{1}{R_2} \right) = -\frac{2(z+c)}{R_2^5}$$

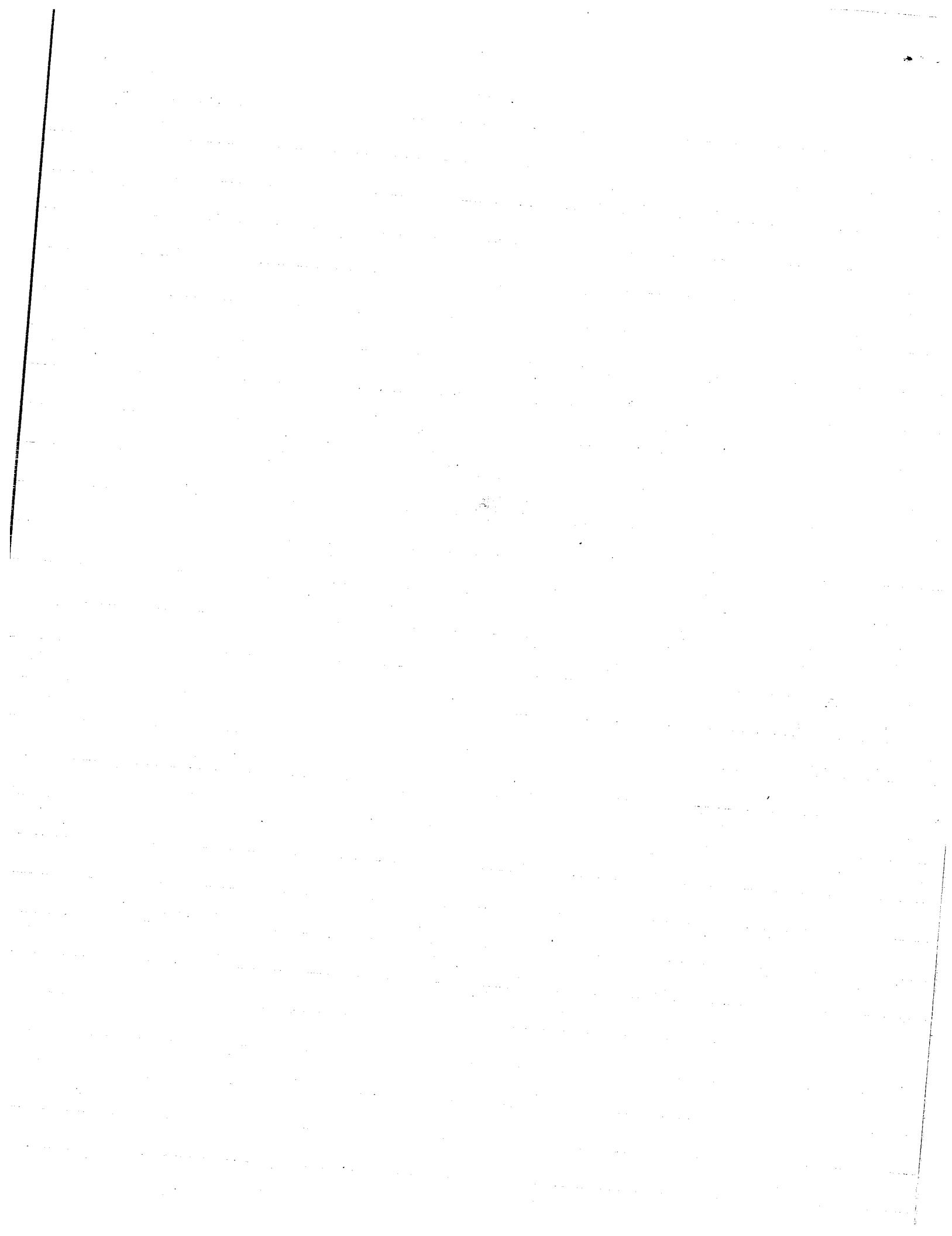
$$\text{In class we had shown } \frac{\partial \beta}{\partial z} = \frac{P_2}{4\pi\mu} \left[\frac{4(1-2v)(1-v)}{R_2} - C \frac{\partial}{\partial z} \left(\frac{1}{R_1} \right) - (3-4v)C \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) \right]$$

$$\text{also } z \frac{\partial B_z}{\partial z} = \frac{P_2}{4\pi\mu} \left[-2Cz \frac{\partial^2}{\partial z^2} \left(\frac{1}{R_2} \right) + z \frac{\partial}{\partial z} \left(\frac{1}{R_1} \right) + (3-4v)z \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) \right]$$

$$(3-4v)B_z = \frac{P_2}{4\pi\mu} \left[\frac{(3-4v)^2}{R_2} + \frac{3-4v}{R_1} - 2C(3-4v) \frac{\partial}{\partial z} \left(\frac{1}{R_2} \right) \right]$$

$$\text{thus } u_z = \frac{1}{4(1-v)} \frac{P_2}{4\pi\mu} \left[\frac{5-12v+8v^2}{R_2} + \frac{(3-4v)(z+c)^2}{R_2^3} - \frac{2Cz \left[R_2^2 - 3(z+c)^2 \right]}{R_2^5} + \frac{(z-c)^2 + \frac{3-4v}{R_1}}{R_1^3} \right]$$

$$\text{along the } z \text{ axis} \quad R_2 = z+c \quad R_1 = z-c \quad \text{and } u_z = \frac{P_2}{4\mu\pi(1-v)} \left[\frac{2(1-v)^2}{z+c} + \frac{(1-v)}{z-c} + \frac{2C}{(z+c)^2} \right]$$



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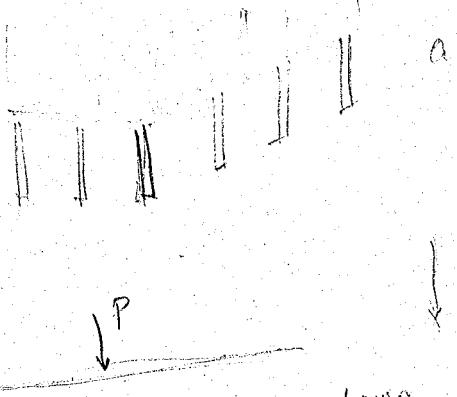
Spring 1980

ME 238C Theory of Elasticity

Midterm Examination

1. Consider a concentrated force acting normally to the surface of a half-space (Boussinesq problem). Calculate and sketch
- the settlement of the bounding plane, 5 for B 5 for β 5 for $u_1 = u_1(B)$
 - the displacement of points along the line of action of the face in the direction of the force. 10 for u_2 5 for graph. 5 for graph
- 15 for final answer
2. An elastic hollow sphere of inner radius a and outer radius b is placed in a rigid spherical housing of radius b and subjected to internal pressure p_i . Determine the change in radius a and the traction exerted by the housing.
- $u_R \& u_\theta, u_\phi$ 25 pts 15 to one
- $T_R \& T_\theta, T_\phi$ 25 pts 15 to T_R

13. Take exam. Akimoto - Net here



A helicopter is to be built on legs, as shown. The legs are designed to take the loading equally. You are asked to find the settlement and the depth under each leg. How would you solve this problem and what simplifying assumption would you make?

A needle rests on a record, and produces a settlement of $\frac{1}{2}k$ at a distance of $3l$ from the needle point (k being a constant having units of Force-length⁴/Force) as shown.

$$w = \frac{1}{2}k \cdot (1-\nu)^2$$

$$\frac{1}{2}k \cdot (1-\nu)^2 = \frac{\pi}{4} \cdot \mu \cdot 3l$$

$$\frac{1}{2}k \cdot (1-\nu)^2 = \frac{\pi}{4} \cdot \mu \cdot 3l$$

$$\begin{aligned}
 & \frac{13k}{108l^3} = \frac{k}{9l^3} = \frac{1-\nu}{24\mu + 48} \frac{P}{P} \\
 & \frac{13k}{8\pi\mu k} = \frac{(1-\nu)}{\frac{12\mu k}{8\pi\mu k}} = \frac{1-\nu}{\frac{12\mu k}{8\pi\mu k}} \frac{P}{P} \\
 & P = 2.872k \frac{P}{P} \\
 & 7.123
 \end{aligned}$$

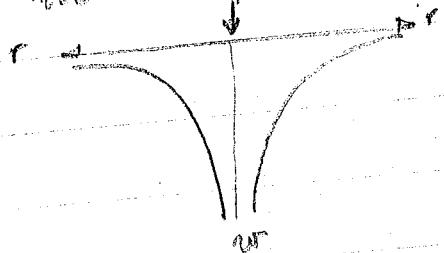
1. From Boundary Value Problem

$$B_x = \frac{1-v}{\pi \mu R} P \quad B_x = B_y = 0 \quad \beta = (1-2v)(1-v) \frac{P \ln(R/r)}{\pi \mu},$$

Using $u = B - \frac{v}{4(1-v)} [u \cdot B + \beta]$ obtain

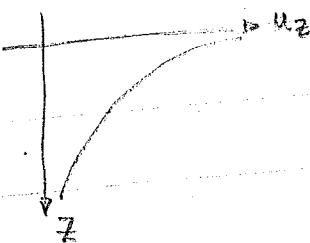
$$u_z = \frac{P}{4\pi\mu} \left[\frac{2(1-v)}{R} + \frac{r^2}{R^3} \right] = \frac{P}{4\pi\mu R} \left[2(1-v) + \left(\frac{r}{R}\right)^2 \right]$$

Settlement $w = u_z \Big|_{z=0} = \frac{(1-v)P}{2\pi\mu R}$ where $r = (x^2+y^2)^{1/2}$



Along line of Force $x=y=0 \quad R \neq z$

$$\therefore u_z = \frac{P}{4\pi\mu} \left[\frac{2(1-v)}{z} + \frac{1}{z} \right] = \frac{P(3-2v)}{4\pi\mu z}$$

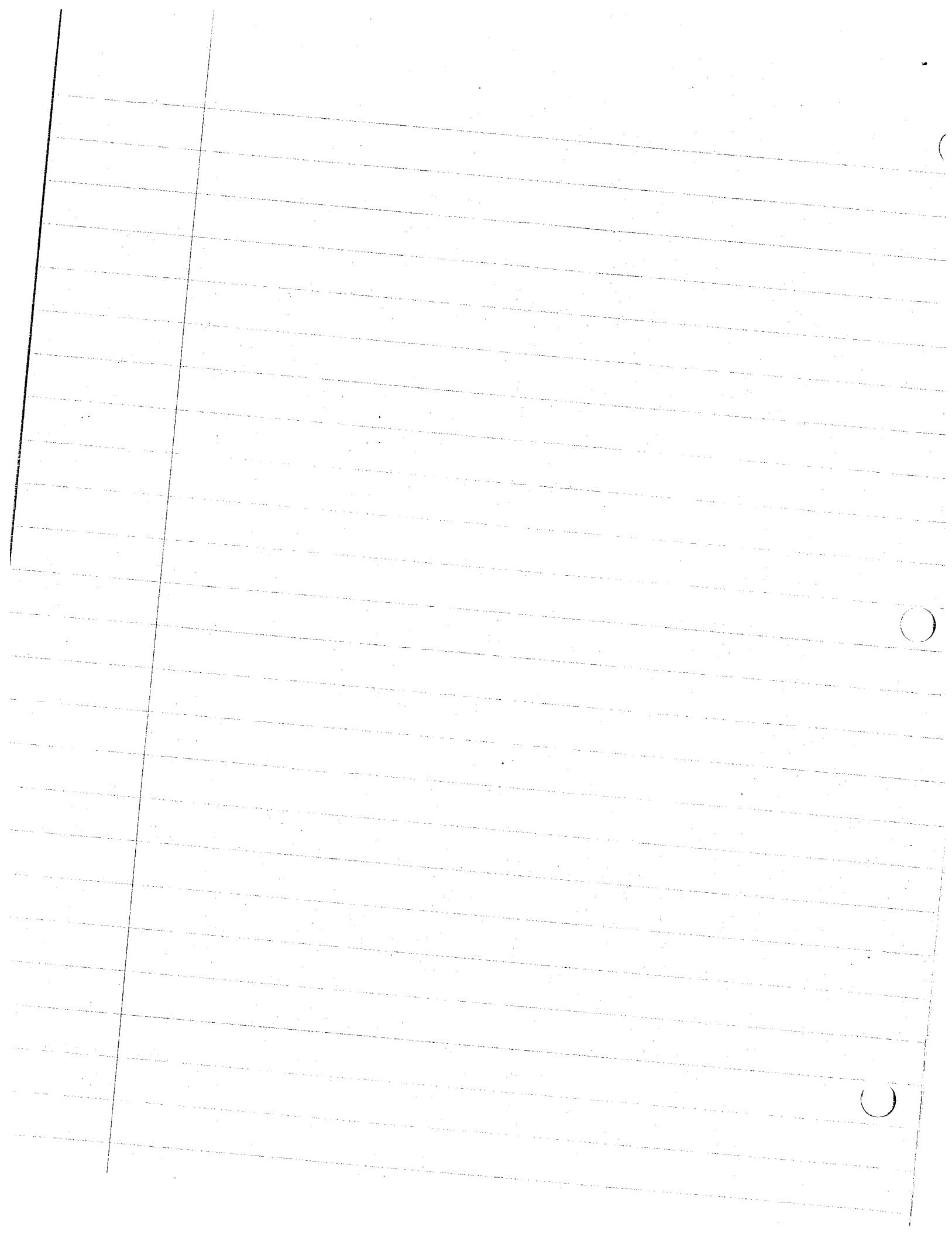


2. From symmetry of problem we expect only σ_{rr} and no σ_{rz} & σ_{rp} at $r=b$. But our b.c. are

(i) $r=a \quad \sigma_{rr}=\sigma_{\theta\theta}=0$

(ii) $r=b \quad u_r=u_\theta=u_\phi=0$

using the results of the hollow sphere under internal & external pressure we note that $u_r=u_\theta=u_\phi=0$ every where. Thus we satisfy the $u_r=u_\theta=u_\phi=0$ conditions at $r=b$.



Now all that is left is to pick $\sigma_{rr} = -p_e$ so that $u_r = 0$ at $r=b$



$$\text{thus } \sigma_{RR} = \frac{\chi}{R^3} + Y$$

$$\sigma_{\theta\theta} = \sigma_{\phi\phi} = -\frac{\chi}{2R^3} + Y \quad \text{where } \chi = (\rho_i - \rho_e) \frac{a^3 b^3}{a^3 - b^3} \quad Y = \frac{b^3 \rho_e - a^3 \rho_i}{a^3 - b^3}$$

$$\text{Now form } \epsilon_{rr} = \frac{du_r}{dr} = \frac{1+v}{E} \sigma_{rr} = \frac{1}{E} (\sigma_{RR} + \sigma_{\theta\theta} + \sigma_{\phi\phi}) \\ = \frac{1+v}{E} \left(\frac{\chi}{R^3} + Y \right) = \frac{1}{E} (3Y),$$

$$\text{Integrate to get } u_r = \frac{1-2v}{E} Y R + \left(\frac{1+v}{E} \right) \left(\frac{\chi}{2R^2} \right) + f(\theta, \varphi)$$

by symmetry $f(\theta, \varphi) = 0$; thus

$$u_r = \frac{1-2v}{E} Y R - \left(\frac{1+v}{E} \right) \frac{\chi}{2R^2}$$

since $u_r|_{r=b} = 0$ and $Y = Y(p_e)$, $\chi = \chi(p_e)$ solve

$$\frac{1-2v}{E} Y b - \left(\frac{1+v}{E} \right) \frac{\chi}{b^2} = 0 \quad \text{for } p_e$$

thus

$$\left. \begin{aligned} p_e &= \frac{3(1-v)}{(1+v)a^3 + 2(1-2v)b^3} \rho_i a^3 \end{aligned} \right\}$$

Now since ϵ_{rr} and σ_{rr} depend on u_r , u_θ or u_ϕ then $\sigma_{\theta\theta} = \sigma_{\phi\phi} = 0$ everywhere hence the fraction at $r=b$ is dependent on p_e only.

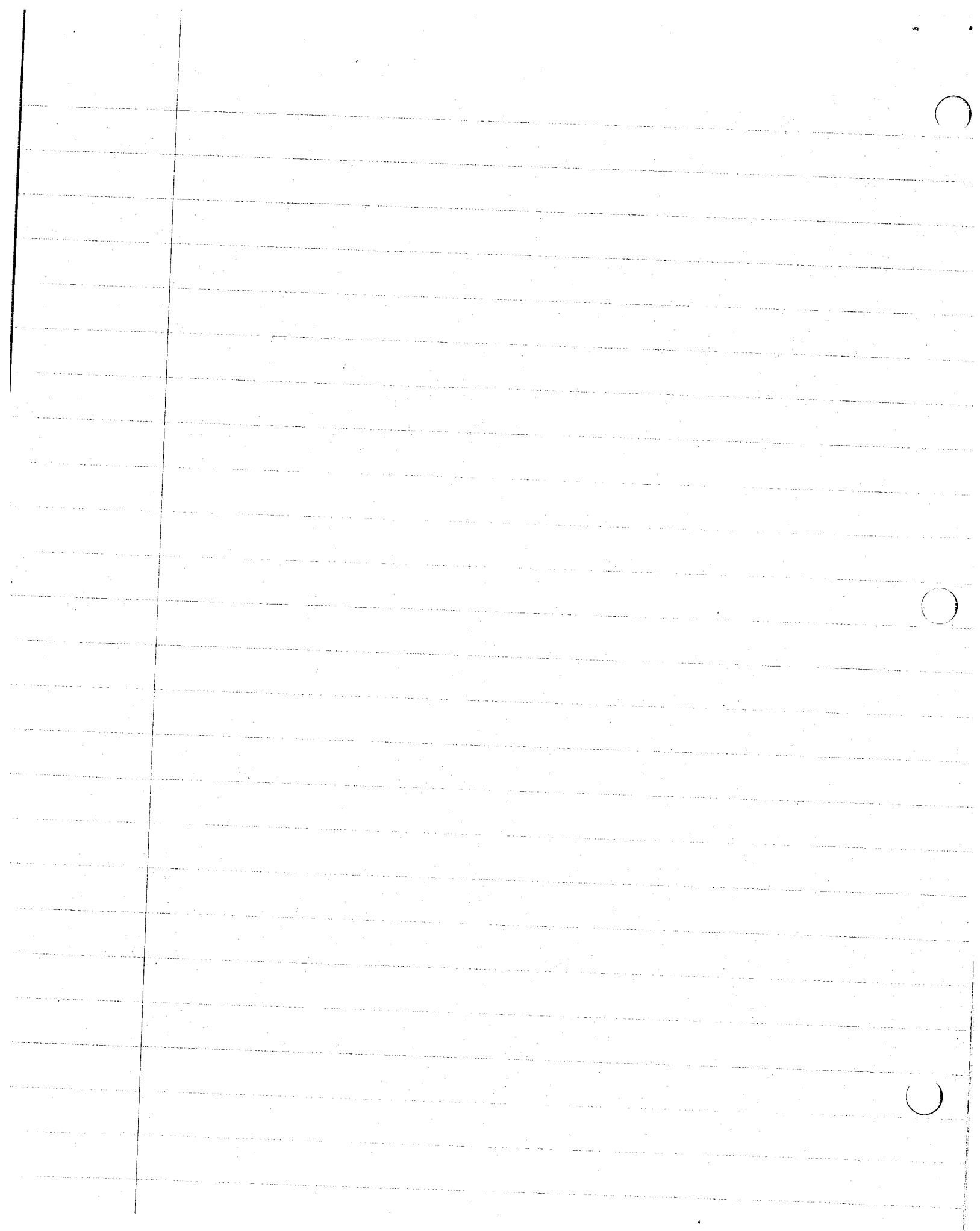
b. The change in radius a just means evaluation of $u_r|_{R=a}$

$$\text{from } u_r = \frac{1-2v}{E} Y R - \left(\frac{1+v}{E} \right) \frac{\chi}{2R^2}$$

let $R=a$ and use p_e in the expressions for Y and χ

this leads to a value of u_r

$$\left. \begin{aligned} u_r|_{R=a} &= \frac{2a(b^3 - a^3)(1-2v)\rho_i}{\mu [(1+v)a^3 + 2(1-2v)b^3]} \end{aligned} \right\}$$

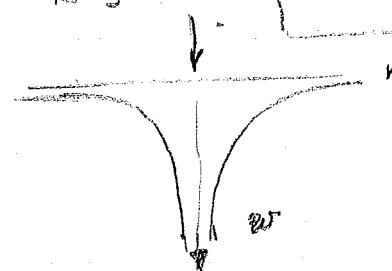


1. a) Boundary
 $B_x = \frac{1-\nu}{\pi \mu R} P$ $B_y = B_z = 0$ $\beta = \frac{(1-2\nu)(1-\nu)}{\pi \mu} P \ln(R+2)$

$$w = B_3 - \frac{V}{4(1-\nu)} [r \cdot B_3 + \beta] = B_3 \alpha_r - \frac{1}{4(1-\nu)} V [r B_3 + \beta],$$

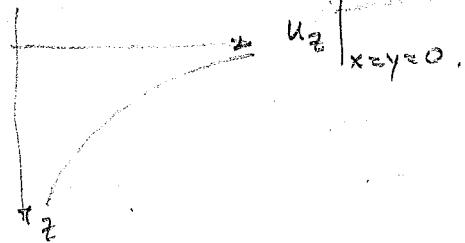
$$u_2 = \frac{P}{4\pi\mu} \left[\frac{2(1-\nu)}{R} + \frac{z^2}{R^2} \right] \quad \left| w = u_2 \right|_{z=0} = \frac{(1-\nu)P}{2\pi\mu r}$$

where $r = \sqrt{x^2+y^2}$



b. along the axis of force $x=y=0$ $R \neq z$

$$u_2 = \frac{P}{4\pi\mu} \left[\frac{2(1-\nu)}{z} + \frac{1}{z^2} \right] = \frac{P(3-2\nu)}{4\pi\mu z} = u_2 \Big|_{x=y=0},$$

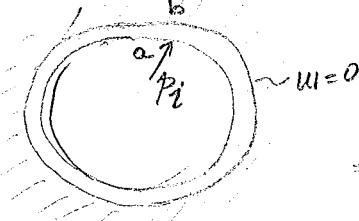


Not included c. when would the displacement of b be equal to the settlement

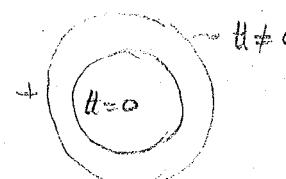
$$\frac{P(3-2\nu)}{4\pi\mu z} = \frac{P(1-\nu)}{2\pi\mu R} \quad \therefore z = \frac{3-2\nu}{1-2\nu} R$$

$$\text{or } r = \frac{(1-\nu)z}{3-2\nu}$$

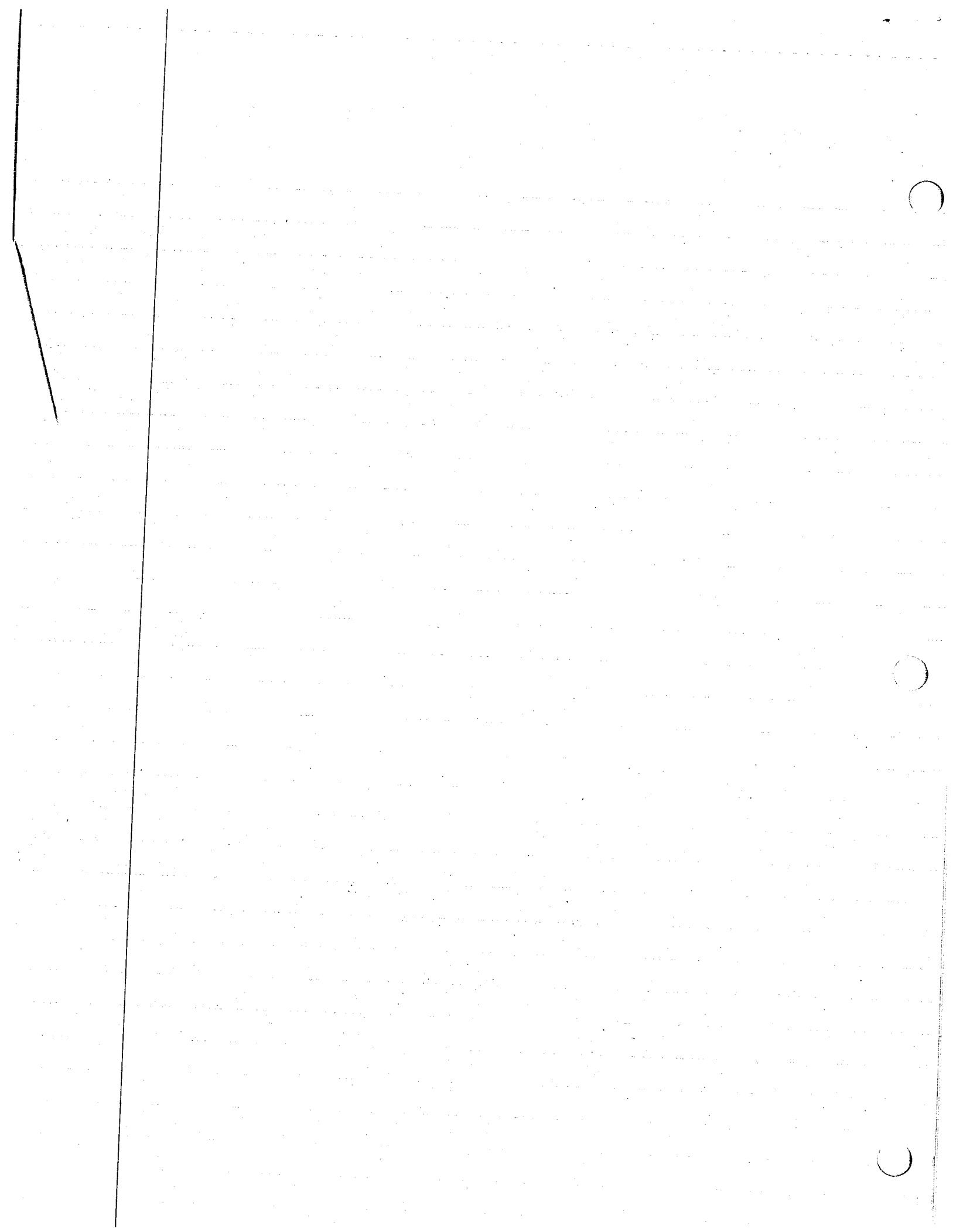
2.



i.e. find u_r (the change in radius a)
and find the fractions on surface



A with load, the A.R. being $w=0$



by symmetry of problem expect only a T_r and no T_θ, T_ϕ & $u_\theta, u_\phi = 0$
 $u_r \neq 0$.

for a sphere under internal & external pressures

$$\sigma_{rr} = (p_i - p_e) \frac{a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 p_e - a^3 p_i}{a^3 - b^3} = \frac{x}{R^3} + Y$$

$$\sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{a^3 b^3 (p_e - p_i)}{2(a^3 - b^3) R^3} + \frac{b^3 p_e - a^3 p_i}{a^3 - b^3} = -\frac{x/2}{R^3} + Y$$

$$\sigma_{rr} + \sigma_{\theta\theta} + \sigma_{\phi\phi} = 3 \left(\frac{b^3 p_e - a^3 p_i}{a^3 - b^3} \right) = I_0 = 3Y$$

$$\text{then } \epsilon_{rr} = \frac{\partial u_r}{\partial r} = \frac{1+Y}{E} \sigma_{rr} = \frac{Y}{E} I_0$$

$$= \frac{1+Y}{E} \left(\frac{x}{R^3} + Y \right) = \frac{Y}{E} \frac{3Y}{2}$$

$$\therefore \frac{1+Y}{E} Y R = \frac{1+Y}{E} \frac{x/2}{R^2} = \frac{Y}{E} \frac{3Y}{2} R = u_r$$

$$\frac{1+2Y}{E} Y R = \frac{1+Y}{E} \frac{x/2}{R^2} = u_r$$

$$\epsilon_{\theta\theta} = \epsilon_{\phi\phi} = \frac{u_r}{R} = \frac{1+Y}{E} \sigma_{\theta\theta} = \frac{Y}{E} I_0 = \frac{1+Y}{E} \sigma_{\theta\theta} = \frac{Y}{E} 3Y$$

$$= \frac{1+Y}{E} \left(-\frac{x/2}{R^3} + Y \right) = \frac{Y}{E} 3Y$$

$$= \frac{1+2Y}{E} Y = \frac{1+Y}{E} \frac{x/2}{R^3}$$

$$\therefore u_\theta = u_\phi = 0 \text{ and } u_r \Big|_{R=b} = 0 \quad \therefore \frac{1-2Y}{E} Y b - \frac{1+Y}{E} \frac{x/2}{b^2} = 0$$

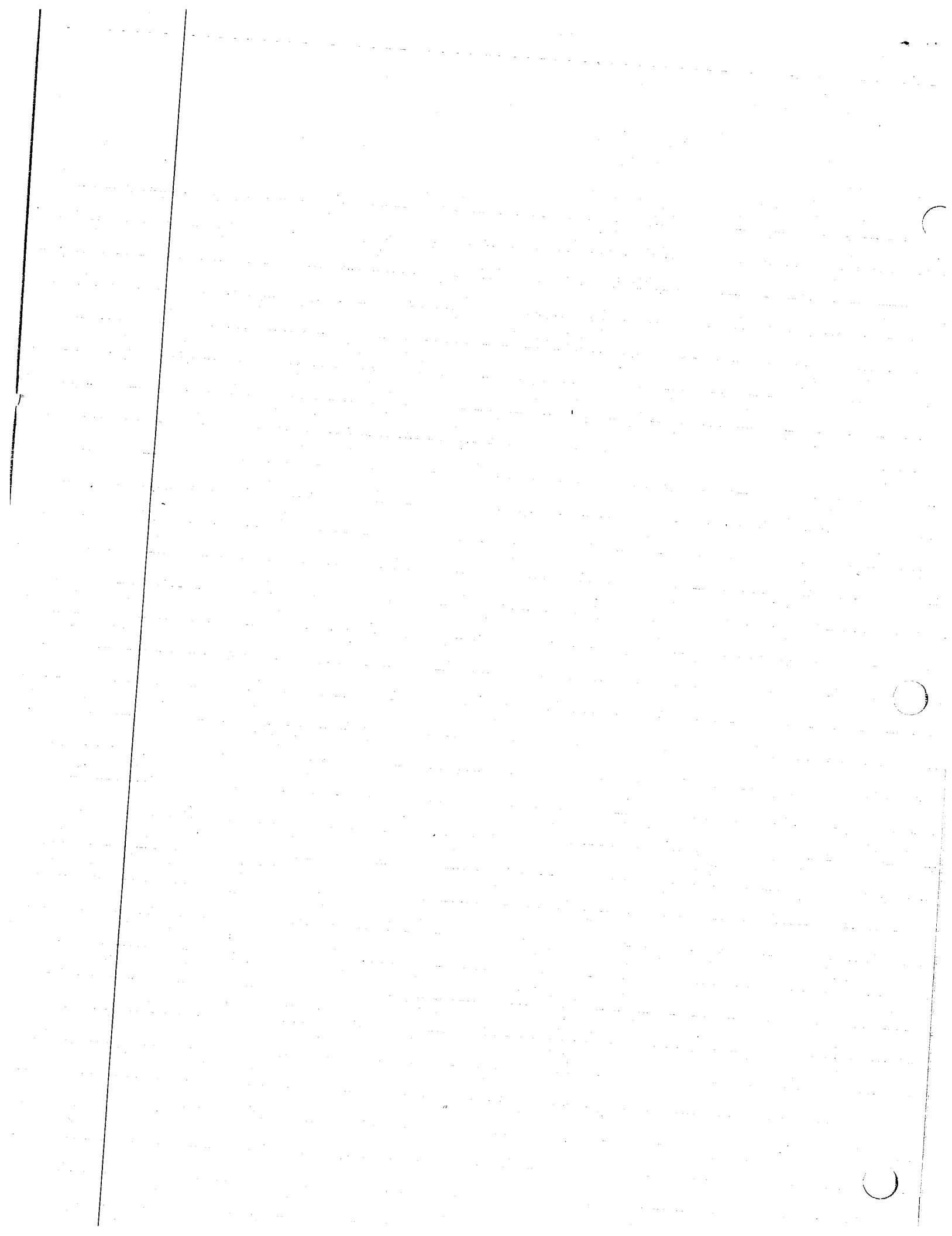
\therefore pick $\bar{Y}(p_e)$ and $\bar{X}(p_e)$ $\Rightarrow u_r \Big|_{R=b} = 0$

$$\therefore \frac{(1-2Y)}{E} \frac{Y b}{b} = \frac{1+Y}{2E b^2} \bar{X}$$

$$\frac{(1-2Y)}{E} b \left[\frac{b^3 p_e - a^3 p_i}{a^3 - b^3} \right] = \frac{1+Y}{2E b^2} \left[\frac{a^3 b^3 (p_i - p_e)}{(a^3 - b^3)} \right]$$

$$\begin{aligned} p_e \left[\frac{2(1-2Y)}{E} b^3 - 2(1+Y) \frac{a^3 p_i}{E} \right] &= (1+Y) a^3 \frac{b^3 p_i}{E} - (1+Y) a^3 \frac{b^3 p_e}{E} \\ p_e \left\{ \frac{(2-4Y)}{E} b^3 + (1+Y) a^3 \right\} &= \left\{ (1+Y) a^3 + 2(1-2Y) a^3 p_i \right\} \frac{b^3}{E} \\ p_e = T_r &= 3(1-Y) p_i a^3 \end{aligned}$$

$$\left. \begin{aligned} &\left((1+Y) a^3 + 2(1-2Y) b^3 \right) \\ &12Y \frac{(p_i - p_e) a^3 b^3}{a^3 - b^3} \end{aligned} \right\}$$



$$u_R = \frac{1-2v}{E} \bar{\gamma} a - \frac{1+v}{2E} \frac{\bar{\gamma}}{a^2}$$

$$= 2 \frac{(1-2v)}{2Ea^2} \bar{\gamma} a^3 - \frac{1+v}{2Ea^2} \frac{\bar{\gamma}}{a^2} = \frac{1}{2Ea^2} \left\{ 2(1-2v) \frac{b^3 p_e - a^3 p_i}{a^3 - b^3} a^3 - \frac{(1+v)(p_i - p_e) a^3 b^3}{a^3 - b^3} \right\}$$

$$= \frac{1}{2Ea^2(a^3 - b^3)} \left\{ a^3 p_e \left[\frac{2(1-2v)[b^3] + (1+v)b^3}{b^3(2-4v+1+v)} \right] - a^3 p_i \left[2(1+2v)a^3 + (1+v)b^3 \right] \right\}$$

$$= \frac{1}{2E(a^3 - b^3)} \left\{ \frac{3(1-v)a^3 b^3 p_e}{3(1-v)a^3 b^3} - a \left[2(1-2v)a^3 + (1+v)b^3 \right] p_i \right\}$$

$$p_e = \frac{3(1-v)p_i a^3}{(1+v)a^3 + 2(1-2v)b^3} \Rightarrow$$

$$u_R = \frac{1}{2E(a^3 - b^3)} \left\{ \frac{9(1-v)^2 a^4 b^3 p_i}{(1+v)a^3 + 2(1-2v)b^3} - a \left[2(1-2v)a^3 + (1+v)b^3 \right] p_i \right\}$$

$$= \frac{a p_i}{2E(a^3 - b^3)} \left\{ \frac{9(1-v)^2 a^3 b^2}{a^3 b^3 (9-18v+9v^2)} - \left[4(1-2v)^2 a^3 b^3 + 2(1-2v)(1+v)[b^6 + a^6] + (1+v)^2 a^3 b^3 \right] \right.$$

$$\left. \frac{4-4v-8v^2}{4(1-4v+4v^2)+1+2v+v^2} \right\} = \frac{(5-14v+17v^2)a^3 b^3 + 2(1-2v)(1+v)(b^6 + a^6)}{a^3 b^3 (9-18v+9v^2)}$$

$$\left\{ \frac{a^3 b^3 4(1-2v)(1+v)}{a^3 b^3 4(1-2v)(1+v)} - 2(1-2v)(1+v)(b^6 + a^6) \right\} \left\{ \frac{b^6 - 2a^3 b^3}{b^6 + a^6} \right\} = \frac{2(b^3 - a^3) a p_i (1-2v)}{\mu [(1+v)a^3 + 2(1-2v)b^3]}$$

$$\frac{a p_i}{2E(a^3 - b^3)} \left\{ \frac{-2(1-2v)(1+v)[(a^3 - b^3)^2]}{(1+v)a^3 + 2(1-2v)b^3} \right\} = \frac{2(b^3 - a^3) a p_i (1-2v)}{\mu [(1+v)a^3 + 2(1-2v)b^3]}$$

Thus $\boxed{u_R = \frac{(b^3 - a^3)a + 2(1-2v)}{\mu [(1+v)a^3 + 2(1-2v)b^3]} p_i}$

$$u_D = U_f e^{-C}$$

$$\sigma_{eff} = \sigma_{D0} = \frac{a^3 (p_e + p_i) + 2b^3 / e^{2a^3 p_i}}{2(a^3 - b^3) + (a^3 + b^3)}$$

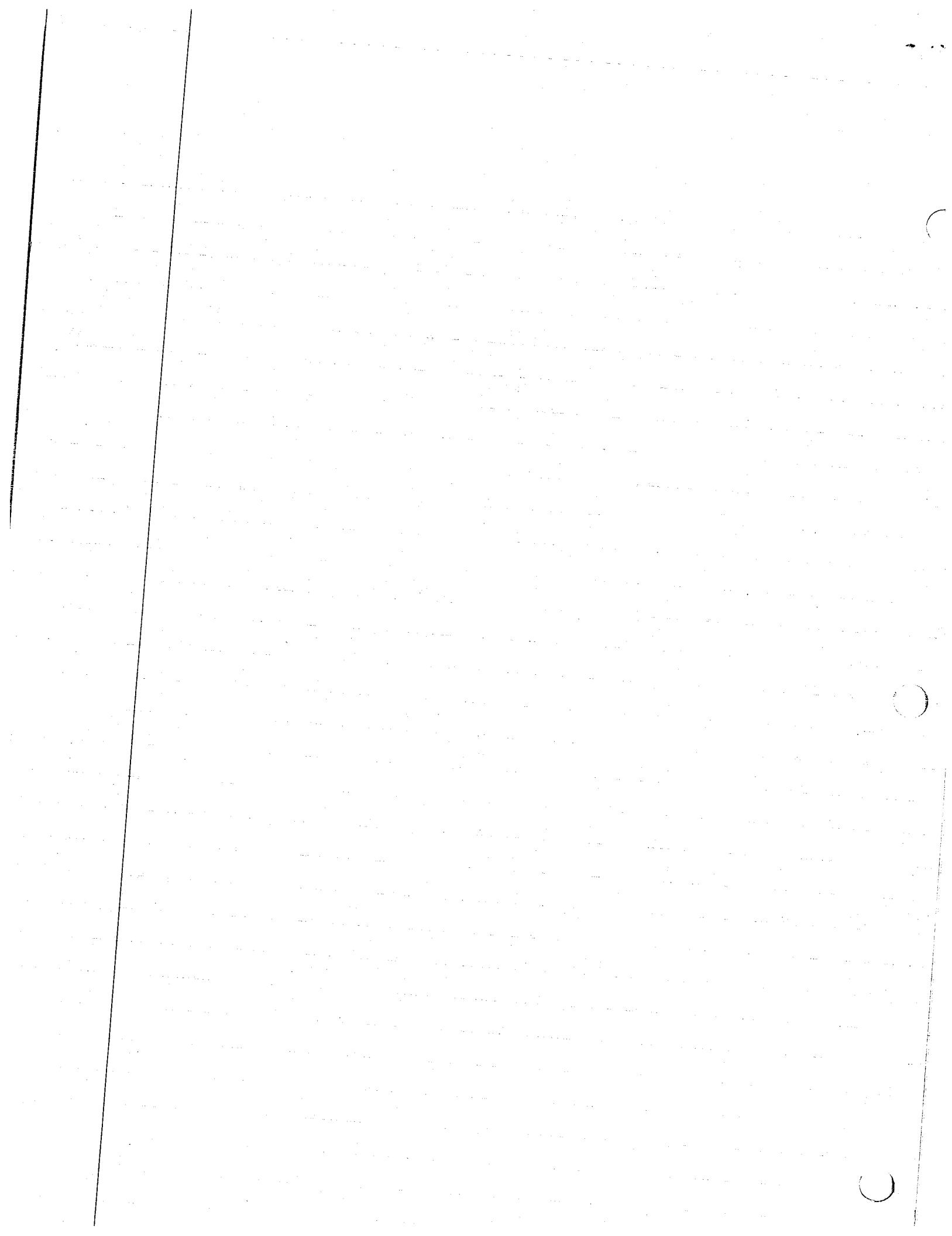
$$= \frac{(a^3 + 2b^3)p_e + 2a^3 p_i}{2(a^3 - b^3)}$$

$$= [a^3 + 2b^3] \left\{ 3(1-v) p_i / a^3 \right\}$$

$$3a^3 p_i$$

$$= \frac{(1+v)a^2 + 1 - (1-2v)b^2}{(a^3 + b^3) 3(1-v)} - 3(1+v)a^2 - 6(1-2v)b^2$$

$$+ \frac{a^3 p_i}{6v(a^3 - b^3)}$$



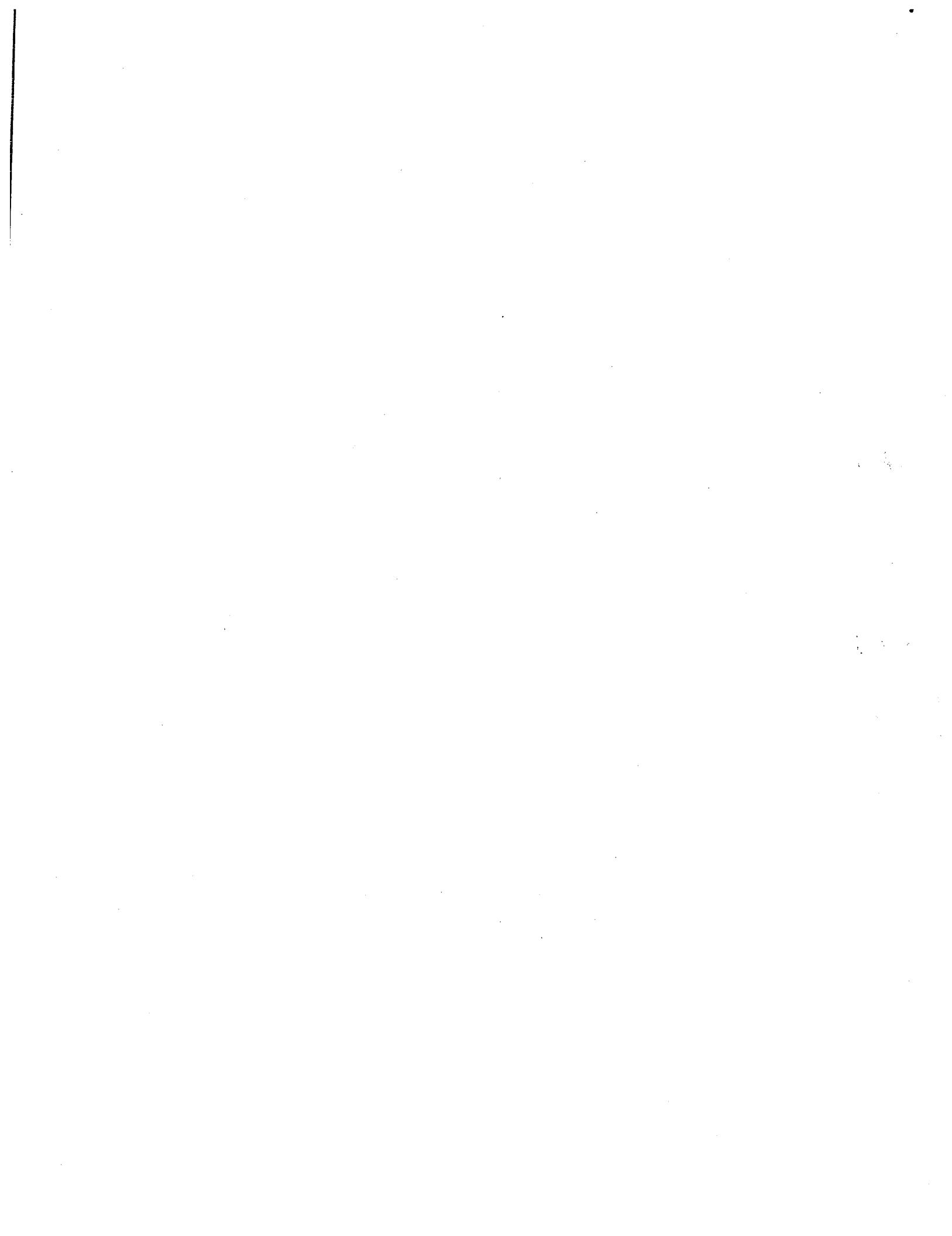
DIVISION OF APPLIED MECHANICS
DEPARTMENT OF MECHANICAL ENGINEERING
STANFORD UNIVERSITY

ME 238C Theory of Elasticity

Spring 1980

Problem Set No. 1

1. Derive the expression for the six components of stress in terms of the four Boussinesq-Papkovich functions for an isotropic elastic solid with reference to a Cartesian system of coordinates. Use the shear modulus and Poisson's ratio as elastic constants.
2. Evaluate the Cartesian components of stress for the Kelvin problem.
3. Evaluate the components of traction on a spherical surface of arbitrary radius with its origin at the point at which the concentrated force is acting in the Kelvin problem.
4. Show that the system of tractions evaluated in (2) above is statically equivalent to a single force.
5. Evaluate the normal strain along the line of action of the force in the Kelvin problem.
6. Evaluate the components of displacement in spherical coordinates for the Kelvin problem.
7. Evaluate the spherical components of stress for the Kelvin problem and show in particular that the components $\sigma_{\theta\phi}$ and $\sigma_{\phi r}$ vanish.
8. Consider an incompressible elastic medium (Poisson's ratio is 1/2) and reduce the results in (7) above to this special case.



-1 for non expansion

$$\textcircled{12} \quad \sigma_{ij} = 2\mu \left[\frac{\nu}{1-2\nu} \delta_{ij} \left\{ \frac{1-2\nu}{2(1-\nu)} B_{KK} - \frac{1}{4(1-\nu)} [x_K B_{L,KK} + \beta_{,KK}] \right\} + \left\{ \frac{1-2\nu}{4(1-\nu)} [B_{ij,j} + B_{jj,i}] - \frac{1}{4(1-\nu)} [x_m B_{m,ij} + \beta_{,ij}] \right\} \right]$$

$$\textcircled{12} \quad 2. \quad \sigma_{ii} = \frac{-P}{8\pi(1-\nu)r^2} \left(\frac{x_3}{r} \right) \left[\begin{array}{l} 2\nu-1 + 3 \left(\frac{x_3}{r} \right)^2 \\ -(2\nu-1) \end{array} \right] \quad \text{for } \sigma_{33}$$

$$\sigma_{12} = \frac{-3P}{8\pi(1-\nu)r^2} \left(\frac{x_1 x_2}{r^3} \right) \quad \sigma_{13} = \frac{-P}{8\pi(1-\nu)r^2} \left(\frac{x_1}{r} \right) \left[1-2\nu + 3 \left(\frac{x_3}{r} \right)^2 \right]$$

$$\textcircled{9} \quad 3. \quad T_1 = \frac{-3P}{8\pi(1-\nu)} \frac{x_1 x_3}{r^4} \quad T_2 = \frac{-3P}{8\pi(1-\nu)} \frac{x_2 x_3}{r^4} \quad T_3 = \frac{-P}{8\pi(1-\nu)r^2} \left[(1-2\nu) + 3 \left(\frac{x_3}{r} \right)^2 \right]$$

$$\textcircled{12} \quad 4. \quad F_1 = F_2 = 0 \quad F_3 = -P \quad M_1 = M_2 = M_3 = 0$$

$$\textcircled{6} \quad 5. \quad \epsilon_{33} = \frac{P}{4\pi\mu r^2} \quad -1 \text{ for non eval. along } x_3$$

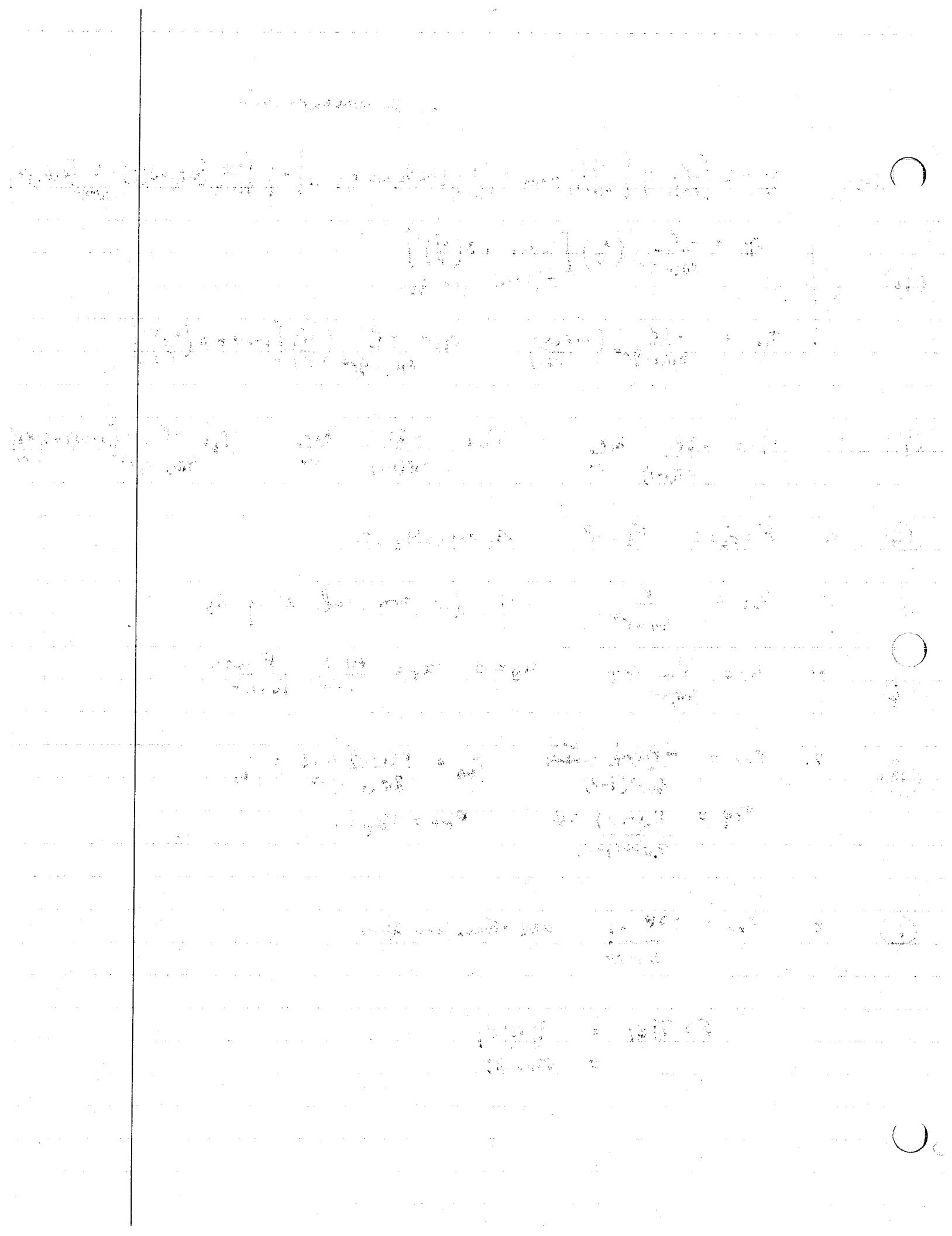
$$\textcircled{6} \quad 6. \quad u_r = \frac{P}{4\pi\mu r} \cos\phi \quad u_\theta = 0 \quad u_\phi = \frac{4\nu-3}{1-\nu} \frac{P \sin\phi}{16\pi\mu r}$$

$$\textcircled{12} \quad 7. \quad \sigma_{rr} = \frac{-P \cos\phi}{8\pi r^2 (1-\nu)} \quad \sigma_{\theta\theta} = \frac{P (1-2\nu)}{8\pi (1-\nu) r^2} \cos\phi = \sigma_{\phi\phi}$$

$$\sigma_{r\phi} = \frac{P (1-2\nu)}{8\pi r^2 (1-\nu)} \sin\phi \quad \sigma_{r\theta} = \sigma_{\theta\phi} = 0$$

$$\textcircled{6} \quad 8. \quad \sigma_{rr} = \frac{-3P \cos\phi}{4\pi r^2} \quad \text{all others are zero}$$

$$\bar{T} = T_i e_i = \sigma_{ij} n_j e_i \\ = \nabla \cdot n e_i$$



$$u_{ij} = B_{ij} - \frac{1}{4(1-v)} (B_i + x_k B_{k,i} + \beta_{,ij}),_j = B_{ij} - \frac{1}{4(1-v)} (B_{ij} + B_{ji} + x_k B_{k,ij} + \beta_{,ij})$$

$$= \frac{3-4v}{4(1-v)} B_{ij} - \frac{1}{4(1-v)} (B_{ji} + x_k B_{k,ij} + \beta_{,ij})$$

$$u_{ji} = \frac{3-4v}{4(1-v)} B_{ji} - \frac{1}{4(1-v)} (B_{ij} + x_k B_{k,ji} + \beta_{,ji})$$

$$u_{ij} + u_{ji} = \frac{3-4v}{4(1-v)} \left[\frac{2-4v}{4(1-v)} (B_{ji} + B_{ij}) - \frac{1}{4(1-v)} (2x_k B_{k,ij} + 2\beta_{,ij}) \right]$$

$$\epsilon_{ij} = \frac{1-2v}{4(1-v)} (B_{ji} + B_{ij}) - \frac{1}{4(1-v)} (x_k B_{k,ij} + \beta_{,ij})$$

$$\epsilon_{kk} = \frac{1-2v}{2(1-v)} B_{kk} - \frac{1}{2(1-v)} (x_k B_{k,ii} + \beta_{,ii})$$

$$\tau_{ij} = \frac{2Gv}{1-2v} \delta_{ij} \left[\frac{1-2v}{2(1-v)} B_{kk} - \frac{1}{2(1-v)} (x_k B_{k,kk} + \beta_{,kk}) \right] + 2G \left[\frac{1-2v}{4(1-v)} (B_{ji} + B_{ij}) \right.$$

$$\left. - \frac{1}{2(1-v)} (x_k B_{k,ij} + \beta_{,ij}) \right]$$

$$= 2G \delta_{ij} \left[\frac{v}{2(1-v)} B_{kk} \delta_{ki} x_k B_{k,kk} \delta_{kj} \beta_{,ij} \right] = \beta_{,kk} \delta_{ik} \delta_{jk}$$

$$= 2G \left[\frac{v}{2(1-v)} \nabla \cdot B - \frac{v}{2(1-v)(1-2v)} [r \cdot \nabla^2 B + \nabla^2 \beta] \right] + 2G \left[\frac{1-2v}{4(1-v)} [\nabla B + B \nabla] \right]$$

$$- \frac{1}{2(1-v)} \left[\frac{\cancel{v(1-2v)}}{\cancel{2(1-v)}} + \beta \right] \frac{\cancel{2}}{\cancel{2}} \cdot (r \cdot \nabla^2 B) \cdot \cancel{I} + \cancel{I} \cdot \nabla^2 \beta \cdot I$$

$$\frac{v}{1-v} G B_{kk} \delta_{ij} + G \frac{(1-2v)}{2(1-v)} [B_{ji} + B_{ij}] = \frac{Gv}{(1-2v)(1-v)} \delta_{ij} x_k B_{k,kk} - \frac{G}{1-v} x_k B_{k,ij}$$

$$- \frac{Gv}{(1-v)(1-2v)} \delta_{ij} \beta_{,kk} \rightarrow \frac{G}{2(1-v)} \beta_{,ij}$$

$$4-4v-1 = \frac{3-4v}{4(1-v)} = \frac{\lambda+3\mu}{2(\lambda+\mu)}$$

~~$$\frac{3-4v}{4(1-v)}$$~~
$$\frac{3-4v}{4(1-v)} - v = \frac{3\mu}{2(\lambda+\mu)}$$

$$3-4v-4v+4v^2$$

$$\frac{(1-2v)(1-v)}{1-3v+2v^2} = \frac{3-8v+4v^2}{2(1-v)} \cancel{\frac{2}{3}}$$

$$1 - \frac{1}{2(1-\nu)} = \frac{1-2\nu}{2(1-\nu)} \cdot \frac{\chi_{\mu\nu}}{1-2\nu}$$

$$\mu \frac{1-2\nu}{1-\nu} B_{1,1}$$

problem #2

$$w = \frac{(1-v)N}{2\pi\mu a} \sin^{-1} \frac{a}{r} = \frac{2(1-v)N}{4\pi^2 \mu a} \int_0^{\infty} \int_{-c}^c \frac{dv d\phi}{\sqrt{c^2 - v^2}} \quad \phi_i = \arcsin \frac{v}{r}$$

problem #1

$$\sigma_{RR} = \frac{2A \cos^2 \varphi}{4r^3} \frac{8-Bv}{1-v} + \frac{\cos^3 \varphi \cdot 2A}{4r^3(1-v)} \cancel{2x}$$

~~if $\theta = 0$~~

~~for $\theta = 0$~~

$$\sigma_{RR} = \frac{\mu \cos^2 \varphi}{(1-v)R^3} \left[(5-v)A - \frac{18C}{R^2} \right] + \frac{\mu}{(1-v)R^3} \left[-vA - D + \frac{6C}{R^2} \right] + T \cos^2 \varphi$$

$$\sigma_{\theta\theta} = \frac{\mu}{2(1-v)R^3} \left\{ \left(\frac{(4v+3)A}{R^2} + \frac{12C}{R^2} \right) \left(\cos^2 \varphi + \sin^2 \varphi \right) - \frac{3C}{R^2} - \frac{2Av}{R^2} \right\} + \frac{D}{R^2}$$

$$\sigma_{\theta\theta} = \left[\left(6v-3 \right) A + \frac{15C}{R^2} \right] \cos^2 \varphi + D - \frac{3C}{R^2} - 2Av$$

$$\sigma_{\varphi\varphi} = \frac{\mu}{2(1-v)R^3} \left[\cos^2 \varphi \left[\left(2v-1 \right) A + \frac{2C}{R^2} \right] - 2(1-v)A - \frac{9C}{R^2} + D \right] + T \sin^2 \varphi$$

$$A = - \frac{5(1-v)a^3 T}{\mu(7-5v)}$$

$$C = - \frac{(1-v)a^5 T}{\mu(7-5v)}$$

$$D = - \frac{(1-v)(6-5v)a^3}{\mu(7-5v)}$$

$$I_{\theta} = \frac{\mu \cos^2 \varphi}{(1-v)R^3} \left[\frac{2(5-v)A}{2} - \frac{18C}{2R^2} + \frac{4v-3A}{2} + \frac{12C}{2R^2} + \frac{(2v-1)A}{2} + \frac{21C}{2R^2} \right]$$

$$+ \frac{\mu}{(1-v)R^3} \left[\frac{-2vA-2D}{2} + \frac{12C}{2R^2} + \frac{D}{2} - \frac{2(1-v)A}{2} - \frac{9C}{2R^2} + \frac{D}{2} \right] + \frac{\mu \sin^2 \varphi}{2(1-v)R^3} \left[\frac{-3C}{R^2} - A \right]$$

$$= \frac{\mu \cos^2 \varphi}{(1-v)R^3} \left[\left(3+2v \right) A - \frac{3C}{R^2} \right] + \frac{\mu}{(1-v)R^3} \left[-A + \frac{3C}{R^2} \right] + \frac{\mu \sin^2 \varphi}{2(1-v)R^3} \left[\frac{-3C}{R^2} - A \right] - v + v \cos^2 \varphi$$

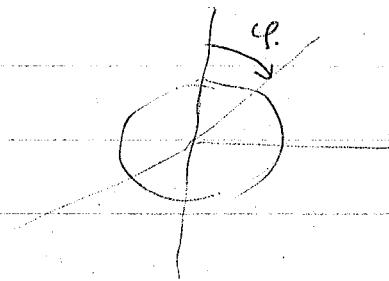
$$= \frac{\mu A}{(1-v)R^3} \left[\left(3+2v \right) \cos^2 \varphi - A - \frac{3C}{R^2} - v \sin^2 \varphi \right] + T = 0$$

$$\frac{\mu A}{(1-v)R^3} = - \frac{5a^3 T}{(7-5v)R^3}$$

$$- \frac{TR^3}{(7-5v)} \frac{(1-v)}{\mu A} = f(\theta)$$

$$R^3 = -\frac{f(\theta) \mu A}{T(1-\nu)}$$

$$\boxed{R = \frac{\sqrt[3]{-f(\theta) \mu A}}{T(1-\nu)}}$$



$$(3+2\nu) [1 - \sin^2 \varphi] - 1 - \nu \sin^2 \varphi \approx f(\theta)$$

$$3+2\nu - 1 - \sin^2 \varphi [3+2\nu + \nu]$$

$$2(1+\nu) - \sin^2 \varphi \cdot 3(1+\nu)$$

$$(1+\nu) [2 - 3 \sin^2 \varphi] \approx f(\theta)$$

$$+(1+\nu) (2 - 3 \sin^2 \varphi) \cdot \frac{5a^3}{7-5\nu} = R^3 \frac{\mu A}{(1-\nu)\pi} = \cancel{\frac{(-5a^3)}{7-5\nu}}$$

$$\boxed{R = \sqrt[3]{\left(\frac{1+\nu}{7-5\nu}\right) \cdot 5a^3 (2 - 3 \sin^2 \varphi)}}$$

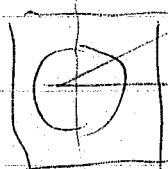
or

$$2 \sqrt{-\frac{1}{3} \left[\frac{R^3}{5a^3} \left(\frac{7-5\nu}{1+\nu} \right) \right] + \frac{2}{3}} = \sin \varphi$$

2-D

$$\varphi = \frac{\pi}{2} - \theta$$

$$\sigma_{RR} = \frac{\sigma}{2} \left(1 - \frac{a^2}{r^2} \right) - \frac{\sigma}{2} \left(1 - \frac{4a^2}{r^2} + \frac{3a^4}{r^4} \right) \cos 2\theta$$



$$\sigma_{\theta\theta} = \frac{\sigma}{2} \left(1 + \frac{a^2}{r^2} \right) + \frac{\sigma}{2} \left(1 + \frac{3a^4}{r^4} \right) \cos 2\theta$$

for plane stress $\sigma_{zz} = 0$ if plane strain $\sigma_{zz} = \nu (\sigma_{RR} + \sigma_{\theta\theta})$

$$\cos \varphi = \cos(\frac{\pi}{2} - \theta)$$

$$\sigma_{RR} + \sigma_{\theta\theta} = \sigma + 2\sigma \frac{a^2}{r^2} \cos 2\theta$$

$$= \sin \theta$$

$$\text{for plane stress } I_0 = 0 \Rightarrow 1 + \frac{2a^2}{r^2} \cos 2\theta = 0 \text{ or}$$

$$\cos \theta = \cos(\frac{\pi}{2} - \varphi)$$

$$= \sin \varphi$$

$$-\frac{r^2}{4a^2} + \frac{1}{2} = \sin^2 \varphi$$

$$r = a \sqrt{2 - 4 \sin^2 \varphi}$$

$$\sin \varphi = \sqrt{\frac{1}{2} - \frac{r^2}{4a^2}} \quad r^2 = 2a^2 - 4a^2 \cos^2 \theta$$

$$\frac{1 + \cos 2\theta}{2} = \cos^2 \theta$$

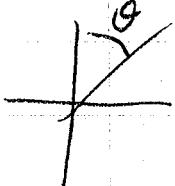
$$\text{for plane strain } \bar{\epsilon} = \sigma_{RR} + \sigma_{\theta\theta} + \sigma_{zz} = (1+\nu)(\sigma_{RR} + \sigma_{\theta\theta}) = 0$$

gives same results indep of ν

$$r = a \sqrt{2 - 4 \sin^2 \varphi} \quad \sin \varphi = \sqrt{\frac{1}{2} - \frac{1}{4} \left(\frac{r}{a}\right)^2}$$

$$\text{for 3D} \quad \sin \varphi = \sqrt{\frac{2}{3} - \frac{1}{15} \left(\frac{7-5\nu}{1+\nu}\right) \left(\frac{R}{a}\right)^3} \quad R = a \sqrt[3]{\left(\frac{5+5\nu}{7-5\nu}\right) \left(2 - 3 \sin^2 \varphi\right)}$$

$$\begin{aligned} (3 \cos^2 \phi - 1) &= \frac{7-5\nu}{5(1+\nu)} \left(\frac{R}{a}\right)^3 \\ 2 - 3 \sin^2 \phi &= +\frac{2}{3} - \frac{7-5\nu}{15(1+\nu)} \left(\frac{R}{a}\right)^3 \quad \cos \theta = \sqrt{\frac{1}{3}(1 + \frac{7-5\nu}{5(1+\nu)} \left(\frac{R}{a}\right)^3)} \\ &= +\frac{2}{3} - \frac{7-5\nu}{15(1+\nu)} \left(\frac{R}{a}\right)^3 \quad \frac{R}{a} = \left[\frac{5(3 \cos^2 \phi - 1)(1+\nu)}{7-5\nu}\right]^{\frac{1}{3}} \end{aligned}$$



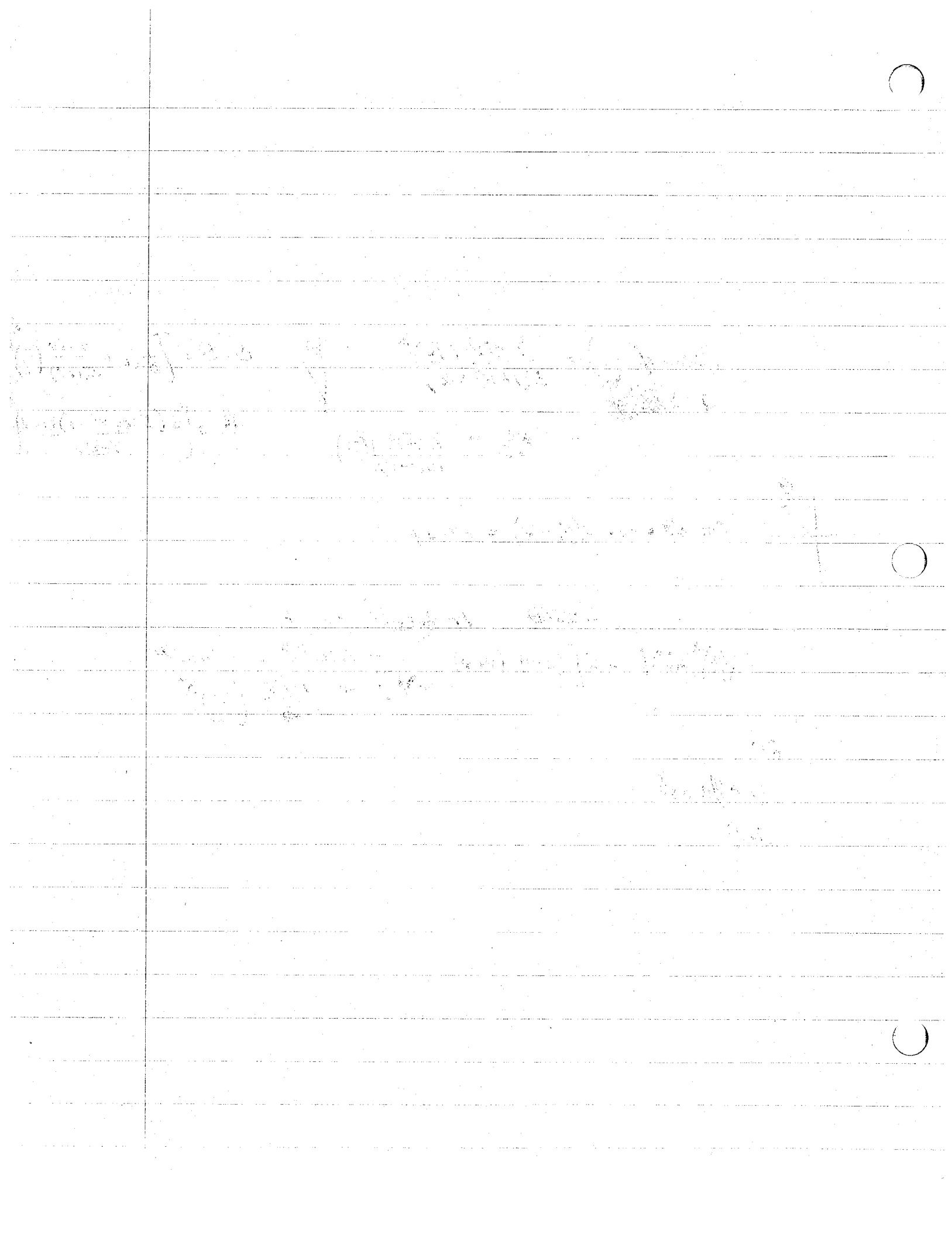
$$\cos 2\theta = \cos 2(\frac{\pi}{2} - \varphi) = -\cos 2\varphi \checkmark$$

$$\begin{aligned} 3 \left(\frac{a}{R}\right)^2 \sin^2 \phi + 5 \left[(2-\nu) - (\nu+4) \frac{1 - 2 \cos^2 \theta}{\cos^2 \phi} \right] + \frac{7-5\nu}{3} \left(\frac{R}{a}\right)^5 &= \frac{1 + \cos 2\theta \cos 2\varphi}{2} \\ &\quad - \cos 2\theta \end{aligned}$$

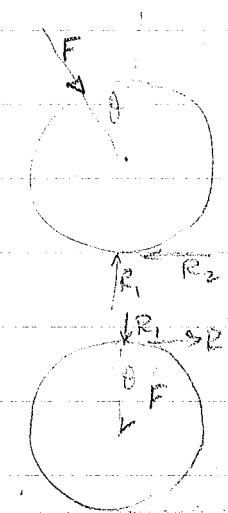
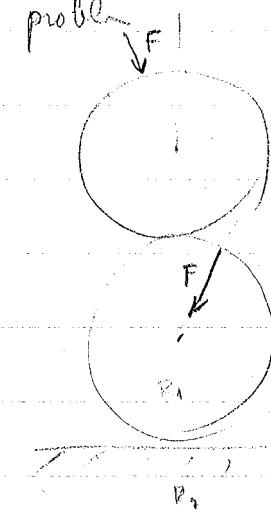
23

$$(1+\nu) \cos^2 \phi$$

$$(2-\nu)$$



HW prob



$$R_2 = F R_2 \theta \approx F \theta = T$$

$$R_1 = F c_0 \theta \approx F = N$$

$$N = 2 F c_0 \theta \approx 2 F$$

$$T = 0$$

$$\frac{1}{P_1 + P_2} = \frac{R_1 R_2}{R_1^2 R_2}$$

$$\text{For Ball/Surface } R_1 = R \quad R_2 = \infty \quad a = \sqrt[3]{\frac{(k_1 + k_2) \cdot 6F\pi R}{4}} \approx \sqrt[3]{\frac{(k_1 + k_2) \cdot 3F}{2}}$$

$$\frac{R_2 + R_1}{R_1 R_2} = \frac{1}{R_1} + \frac{1}{R_2}$$

$$\alpha = \sqrt[3]{\frac{36 F^2 \pi^2 (k_1 + k_2)^2}{16 R}} = \sqrt[3]{\frac{9 F^2 \pi^2 (k_1^2 + k_2^2)}{4 R}} \quad a = \sqrt[3]{\frac{(k_1 + k_2) \cdot 3F\pi R}{2}}$$

$$\text{for b/b contact } \frac{2}{R} = \frac{R_2 + R_1}{R_1 R_2} \text{ for a hard surface } k_2 = 0 \quad \therefore a = \sqrt[3]{\frac{3F\pi k_1 R}{2}} \quad \alpha = \sqrt[3]{\frac{9 F^2 \pi^2 k_1^2}{16 R^2}}$$

$$= \sqrt[3]{\frac{3F(1-v)R}{8\mu}} \quad = \sqrt[3]{\frac{9F(1-v)^2}{16 R^2}}$$

For ball/ball contact.

$$a = \sqrt[3]{\frac{2k_1 \cdot 3F\pi R^2}{4 \cdot 2R}} = \sqrt[3]{\frac{3k_1 F\pi R}{4}} = \sqrt[3]{\frac{3(1-v)F\pi R}{8\mu}}$$

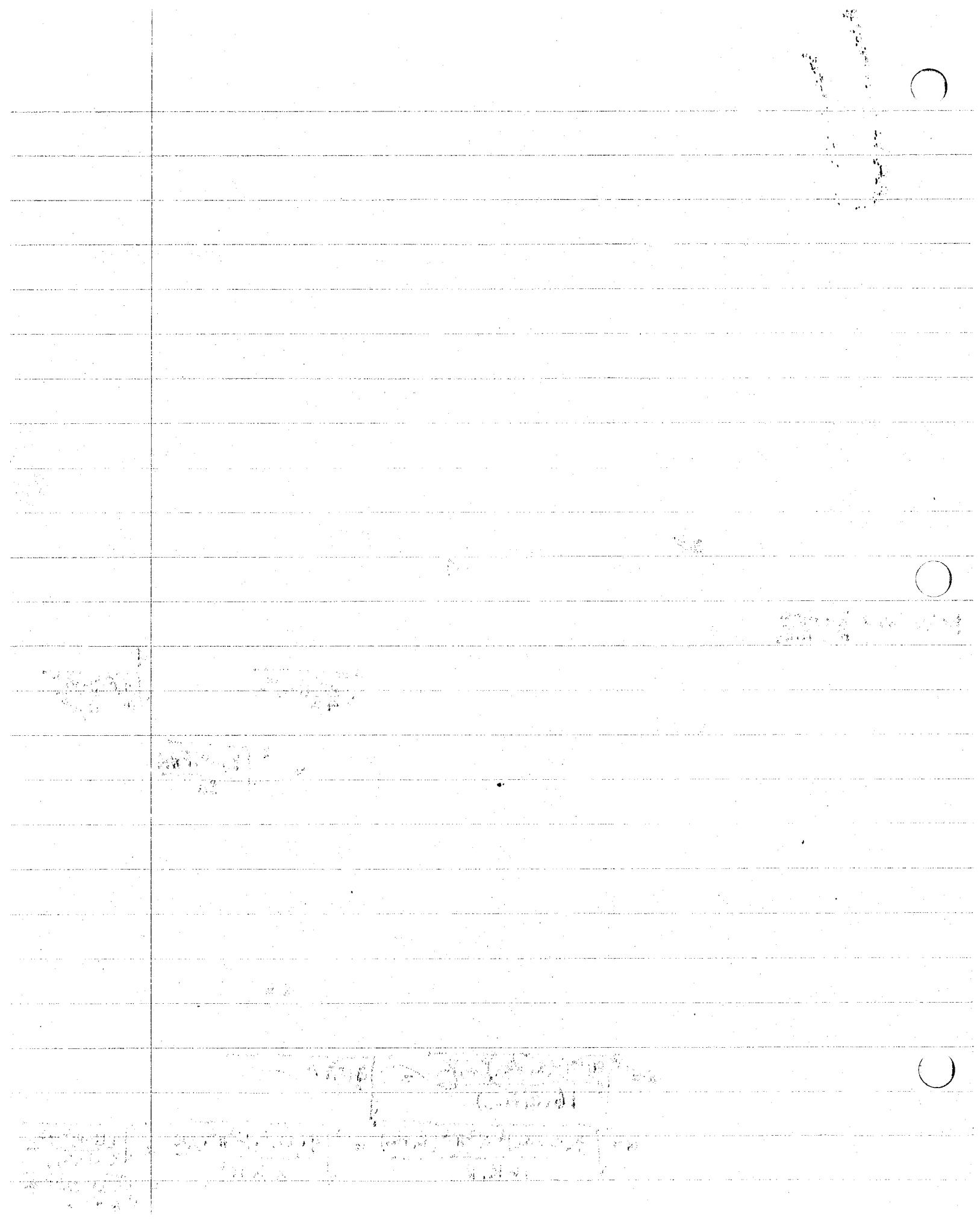
$$c = a \sqrt[3]{1 - \frac{T}{fN}} = a \sqrt[3]{1 - \frac{\theta}{f}} \approx a \left(1 - \frac{\theta}{3f}\right)$$

$$u_x = \frac{3fN}{16\mu} (2-v) \left(1 - \frac{\theta^2}{a^2}\right) \approx \frac{3fF}{16\mu} (2-v) \cdot \frac{2\theta}{3f} = \frac{2F\theta(2-v)}{16\mu}$$

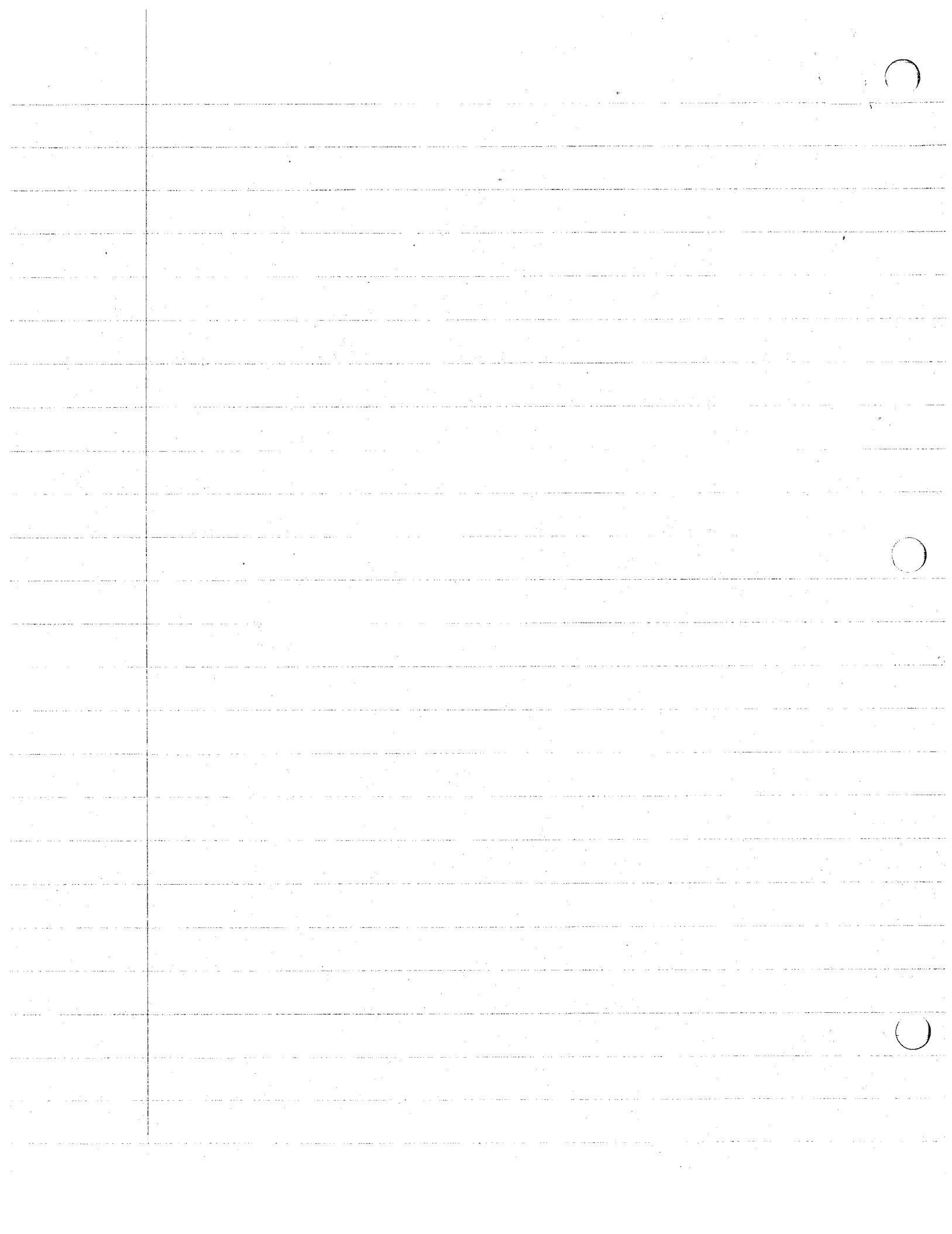
$$\text{or } \frac{2F\theta(2-v)}{16\mu \sqrt[3]{\frac{3k_1 F\pi R^2}{4}}} = u_x \quad \alpha =$$

$$\alpha = \sqrt[3]{\frac{9N^2\pi^2(k_1 + k_2)^2 R_1 R_2}{16(R_1 + R_2)}} = \sqrt[3]{\frac{9F^2\pi^2}{2}}$$

$$\alpha = \sqrt[3]{\frac{9(k_1 + k_2)^2 N^2 \pi^2 (R_1 + R_2)}{16 R_1 R_2}} = \sqrt[3]{\frac{9 \cdot 4k^2 \cdot F^2 \pi^2 \cdot 2R}{2 \cdot 16 R^2}} = \sqrt[3]{\frac{9(1-v)^2 F^2 \pi^2}{8 \mu^2 R}} \\ = \sqrt[3]{\frac{9(1-v)^2 F^2 \pi^2}{8 \mu^2 R}}$$



1. Consider a concentrated force acting tangentially to the surface of a half space. Calculate
 - a. The settlement of the sounding plane, and
 - b. If θ is the angle measured on the surface of the half space, measured from the positive x -axis, Find those values of θ and those physically possible values of ν for which the settlement vanishes (Remember $0 \leq \nu \leq \frac{1}{2}$).
2. An elastic sphere of radius b contains a fixed spherical inclusion of radius a and is subjected to uniform external pressure p . Find the traction on the surface of the inclusion.



For 1

$$w = B_3 - \frac{P}{4(1-v)} [r \cdot B_3 + \beta]$$

$$\text{from class } B_3 = \frac{P}{2\pi\mu R} ex + \frac{(1-v)Px}{2\pi\mu R(R+v)} \theta_2$$

for settlement need u_2

$$u_2 = B_2 - \frac{\partial}{\partial z} \frac{1}{4(1-v)} [r \cdot B_3 + \beta]$$

$$= B_2 - \frac{1}{4(1-v)} \left[\left(\frac{\partial}{\partial z} r \right) \cdot B_3 + r \cdot \frac{\partial B_3}{\partial z} + \frac{\partial \beta}{\partial z} \right]$$

$$= \left[\theta_2 \cdot B_3 + x \cdot \frac{\partial B_3}{\partial z} + \frac{\partial B_3}{\partial z} + \frac{\partial \beta}{\partial z} \right]$$

in class you derived $\frac{\partial B_3}{\partial z}$, $\frac{\partial B_2}{\partial z}$, $\frac{\partial \beta}{\partial z}$

plugging in and let $z=0$

$$w = u_2(x, y, 0) = \frac{(1-v)Px}{2\pi\mu r(r)} - \frac{1}{4(1-v)} \left[\frac{(1-v)Px}{2\pi\mu r^2} + x \cdot \frac{\partial B_3}{\partial z} + \frac{\partial B_2}{\partial z} + \frac{(1-v)^2 Px}{2\pi\mu r^2} \right]$$

$$= \frac{Px}{2\pi\mu r^2} \left[\frac{4(1-v)(1-v) - (1-v)(1-2v)}{4(1-v)} \right]. \quad \frac{1}{4(1-v)} = \frac{Px(1-2v)(2(1-v))}{2\pi\mu r^2 4(1-v)}$$

$w=0$ when $x=0$ $\Rightarrow \cos\theta = \frac{y}{r}$, $3y_2$ ie $x=0$

Also $\nabla w = 0$ when $1-2v=0$ i.e. $v=\frac{1}{2}$ ie when half space is rigid. Therefore $\theta = \frac{\pi}{2}$ or $\theta = \frac{3\pi}{2}$ is physically possible

i.e. $w=0$ when $\theta = \frac{\pi}{2}, \frac{3\pi}{2}$ or $\frac{\pi}{4}, \frac{3\pi}{4}$



For 2. B.C. are at $r=b$ $\sigma_{RR}=-p$, $\sigma_{R\phi}=\sigma_{R\theta}=0$

at $r=a$ $u_R=0$, $u_\theta=0$, $u_\phi=0$

use notes from hollow sphere. From symmetry this means that the only stress on inclusion walls are σ_{RR} , $\sigma_{\theta\theta}$, $\tau_{rr\theta}$; assume $\sigma_{RR}=-p$

then from notes

$$\sigma_{RR} = (\rho_i - \rho) \frac{a^3 b^3}{(a^3 - b^3) R^3} + b^3 p - a^3 p_i = \frac{\Delta}{R^3} + Y$$

$$\sigma_{xx} = \sigma_{yy} = \frac{a^3 b^3 (\rho - \rho_i)}{2(a^3 - b^3) R^3} + \frac{b^3 p - a^3 p_i}{a^3 - b^3} = \frac{-\Delta}{2R^3} + Y$$

this solution automatically satisfies $u_x = u_y = 0$ everywhere

$$\text{thus } \frac{\partial u_R}{\partial R} = \epsilon_{RR} = \frac{1+2v}{E} \sigma_{RR} = \frac{1}{E} (\sigma_{RR} + \sigma_{xx} + \sigma_{yy})$$

$$u_R = \frac{1-2v}{E} Y R = \frac{1+v}{E} \frac{x}{2R^2} + \text{r.h.s. motion}$$

$$u_R|_{x=0} = 0 \Rightarrow \frac{1-2v}{E} Y a = \frac{1+v}{2Ea^2} x$$

Then solving for $\rho_i = \frac{3(1-v) \rho b^3}{(1+v) b^3 + 2(1-2v) a^3}$ this is traction



$$= \sigma_{RR} = T_r = \rho_i$$

For 1

$$w = IB - \frac{P}{4(1-v)} [ir \cdot IB + \beta]$$

$$\text{from class } B = \frac{P}{2\pi\mu R} ex + \frac{(1-v)Px}{2\pi\mu R(R+v)} \theta_2$$

for settlement need u_z

$$u_z = B_2 - \frac{\partial}{\partial z} \frac{1}{4(1-v)} [ir \cdot IB + \beta]$$

$$= B_2 - \frac{1}{4(1-v)} \left[\left(\frac{\partial}{\partial z} ir \right) \cdot IB + ir \cdot \frac{\partial B}{\partial z} + \frac{\partial \beta}{\partial z} \right] \\ = \left[\frac{\theta_2 \cdot IB}{B_2} + x \cdot \frac{\partial B_x}{\partial z} + z \frac{\partial B_2}{\partial z} + \frac{\partial \beta}{\partial z} \right]$$

in class you derived $\frac{\partial B_x}{\partial z}, \frac{\partial B_2}{\partial z}, \frac{\partial \beta}{\partial z}$

plugging in and let $z=0$

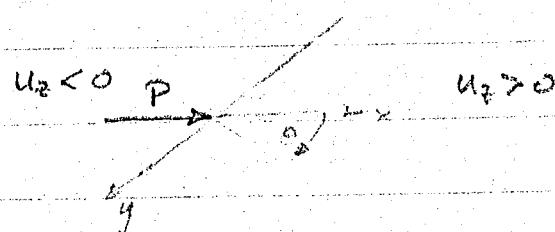
$$w = u_z(x, y, 0) = \frac{(1-v)Px}{2\pi\mu r(r)} - \frac{1}{4(1-v)} \left[\frac{(1-v)Px}{2\pi\mu r^2} + x \cdot \frac{\partial B_x}{\partial z} + z \frac{\partial B_2}{\partial z} + \frac{(1-v)^2 Px}{2\pi\mu r^2} \right]$$

$$= \frac{Px}{2\pi\mu r^2} \left[\frac{4(1-v)(1-v) - (1-v)^2}{(1-v) + (1-v)} \right]. \frac{1}{4(1-v)} = \frac{Px(1-v)}{2\pi\mu r^2 \cdot 4(1-v)}$$

$\therefore w=0$ when $x=0 \Rightarrow \cos\theta = \frac{\pi}{2}, \frac{3\pi}{2}$ ie $x=0$

Also $w=0$ when $1-2v=0 \Rightarrow v=\frac{1}{2}$ ie when half space is rigid. Therefore $v=\frac{1}{2}$ is physically possible

$$\therefore w=0 \quad \theta = \frac{\pi}{2}, \frac{3\pi}{2} \quad v = \frac{1}{2}$$



For 2. B.C. are at $r=b$ $\sigma_{RR}=-P$ $\sigma_{R\phi}=\sigma_{R\theta}=0$

at $r=a$ $u_R=0, u_\theta=0, u_\phi=0$

Use notes from hollow sphere. From symmetry this means that the only stress on inclusion walls are $\sigma_{RR}, \sigma_{\theta\theta}, \tau_{\theta\phi}$; assume $\sigma_{RR}=-p_i$

then from notes

$$\sigma_{RR} = (p_i - p) \frac{a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 p - a^3 p_i}{a^3 - b^3} = \frac{8}{R^3} + Y$$

$$\sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{a^3 b^3 (p_i - p)}{2(a^3 - b^3) R^3} + \frac{b^3 p - a^3 p_i}{a^3 - b^3} = \frac{-8}{2R^3} + Y$$

This solution automatically satisfies $u_3 = u_4 = 0$ everywhere

$$\text{thus } \frac{\partial u_2}{\partial R} = \sigma_{RR} = \frac{1+Y}{E} \sigma_{RR} - \frac{Y}{E} (\sigma_{rr} + \sigma_{\theta\theta} + \sigma_{\phi\phi})$$

$$u_R = \frac{1-2Y}{E} Y R - \frac{1+Y}{E} \frac{X}{2R^2} + \text{a. b. motion}$$

$$\frac{u_R}{R_{\text{ext}}} = 0 \Rightarrow \frac{1-2Y}{E} Y a = \frac{1+Y}{2Ea^2} X$$

Then solving for $p_i = \frac{3(1-Y)p b^3}{(1+Y)b^3 + 2(1-2Y)a^3}$ | This is traction

$$- \sigma_{RR} = T_r = p_i$$



For 1

$$w = 1B - \frac{P}{4(1-v)} [1r \cdot 1B + \beta]$$

$$\text{from class } B = \frac{P}{2\pi\mu R} ex + (1-v) \frac{Px}{2\pi\mu R(P+2)} \quad \varphi_2$$

for settlement need u_z

$$u_z = B_z - \frac{\partial}{\partial z} \frac{1}{4(1-v)} [1r \cdot 1B + \beta]$$

$$= B_z - \frac{1}{4(1-v)} \left[\left(\frac{\partial}{\partial z} 1r \right) \cdot 1B + 1r \cdot \frac{\partial B}{\partial z} + \frac{\partial \beta}{\partial z} \right]$$

$$= \left[\varphi_2 \cdot 1B + x \cdot \frac{\partial B_x}{\partial z} + z \frac{\partial B_z}{\partial z} + \frac{\partial \beta}{\partial z} \right]$$

in class you derived $\frac{\partial B_x}{\partial z}, \frac{\partial B_z}{\partial z}, \frac{\partial \beta}{\partial z}$

plugging in and let $z=0$

$$w = u_z(x, y, 0) = \frac{(1-v) Px}{2\pi\mu r(r)} - \frac{1}{4(1-v)} \left[\frac{(1-v) Px}{2\pi\mu r^3} + x \cdot \frac{\partial B_x}{\partial z} + z \frac{\partial B_z}{\partial z} + \frac{(1-v)^2 Px}{2\pi\mu r^2} \right]$$

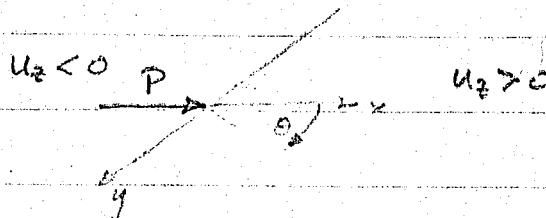
$$= \frac{Px}{2\pi\mu r^2} \left[\frac{4(1-v)(1-v) - (1-v)^2}{(1-v) - (1-v) - (1-v)^2} \right]. \quad \frac{1}{4(1-v)} = \frac{Px(1-v)}{2\pi\mu r^2 \cdot 4(1-v)}$$

$\therefore w=0$ when $x=0 \Rightarrow \cos\theta = \frac{1}{2}, \frac{3\pi}{2}$, ie $x=0$

Also $w=0$ when $1-2v=0 \therefore v=\frac{1}{2}$ ie when half-space is rigid.

Therefore $v=\frac{1}{2}$ is physically possible

$$\therefore w=0 \quad \theta = \frac{\pi}{2}, \frac{3\pi}{2} \quad v=\frac{1}{2}$$



For 2. B.C. are at $r=b$ $\sigma_{RR}=-P$, $\sigma_{R\phi}=\sigma_{R\theta}=0$

at $r=a$ $u_r=0$, $u_\theta=0$, $u_\phi=0$

Use notes from hollow sphere. From symmetry this means that the only stress on inclusion walls are σ_{RR} , $\sigma_{\theta\theta}$, $\sigma_{\phi\phi}$; assume $\sigma_{RR}=-p_i$

Then from notes

$$\sigma_{RR} = (\rho_i - p) \frac{a^3 b^3}{(a^3 - b^3) R^3} + \frac{b^3 p - a^3 \rho_i}{a^3 - b^3} = \frac{x}{R^3} + y$$

$$\sigma_{\theta\theta} = \sigma_{\phi\phi} = \frac{a^3 b^3 (p - \rho_i)}{2(a^3 - b^3) R^3} + \frac{b^3 p - a^3 \rho_i}{a^3 - b^3} = \frac{-x}{2R^3} + y$$

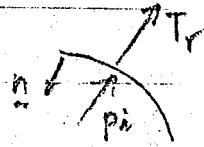
This solution automatically satisfies $u_\theta = u_\phi = 0$ everywhere

thus $\frac{\partial u_R}{\partial R} = \sigma_{RR} = \frac{1+\nu}{E} \sigma_{RR} - \frac{\nu}{E} (\sigma_{RR} + \sigma_{\theta\theta} + \sigma_{\phi\phi})$

$$u_R = \frac{1-2\nu}{E} YR - \frac{1+\nu}{E} \frac{x}{2R^2} + \text{a. b. motion}$$

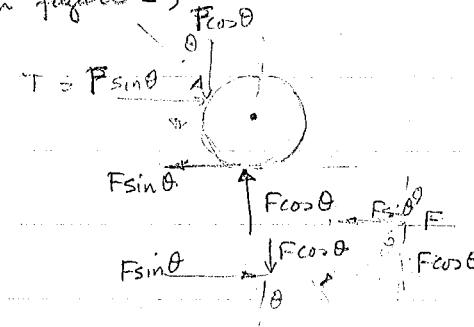
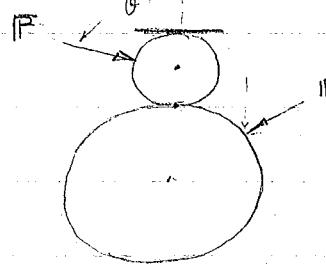
$$u_R|_{R=0} = 0 \Rightarrow \frac{1-2\nu}{E} Ya = \frac{1+\nu}{2Ea^2} X$$

then solving for $\rho_i = \frac{3(1-\nu)p b^3}{(1+\nu)b^3 + 2(1-2\nu)a^3}$



$$-\sigma_{RR} = Tr = \rho_i$$

not slide
Free body diagram
at time t :
the forces on the ball bearing are moving so slowly that the problem may be considered
and at time t , they appear as shown in figure 2, Find:



What approximations

$$R_2 \approx R_1$$

$$\therefore R_2 = 0 \quad R_1 = 2F \cos \theta$$

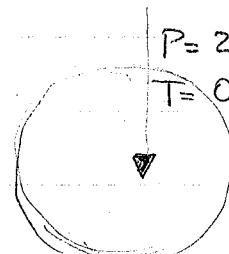


$$P = F \cos \theta$$

$$T = F \theta$$

$$P = 2F \cos \theta$$

what is approach of ball 1 relative to ball 2.

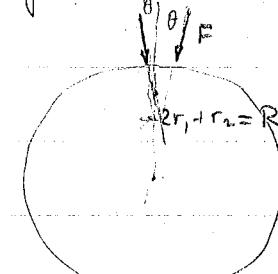


what is approach of ball 2 relative to surface.

" " " ball 1 " " surface.

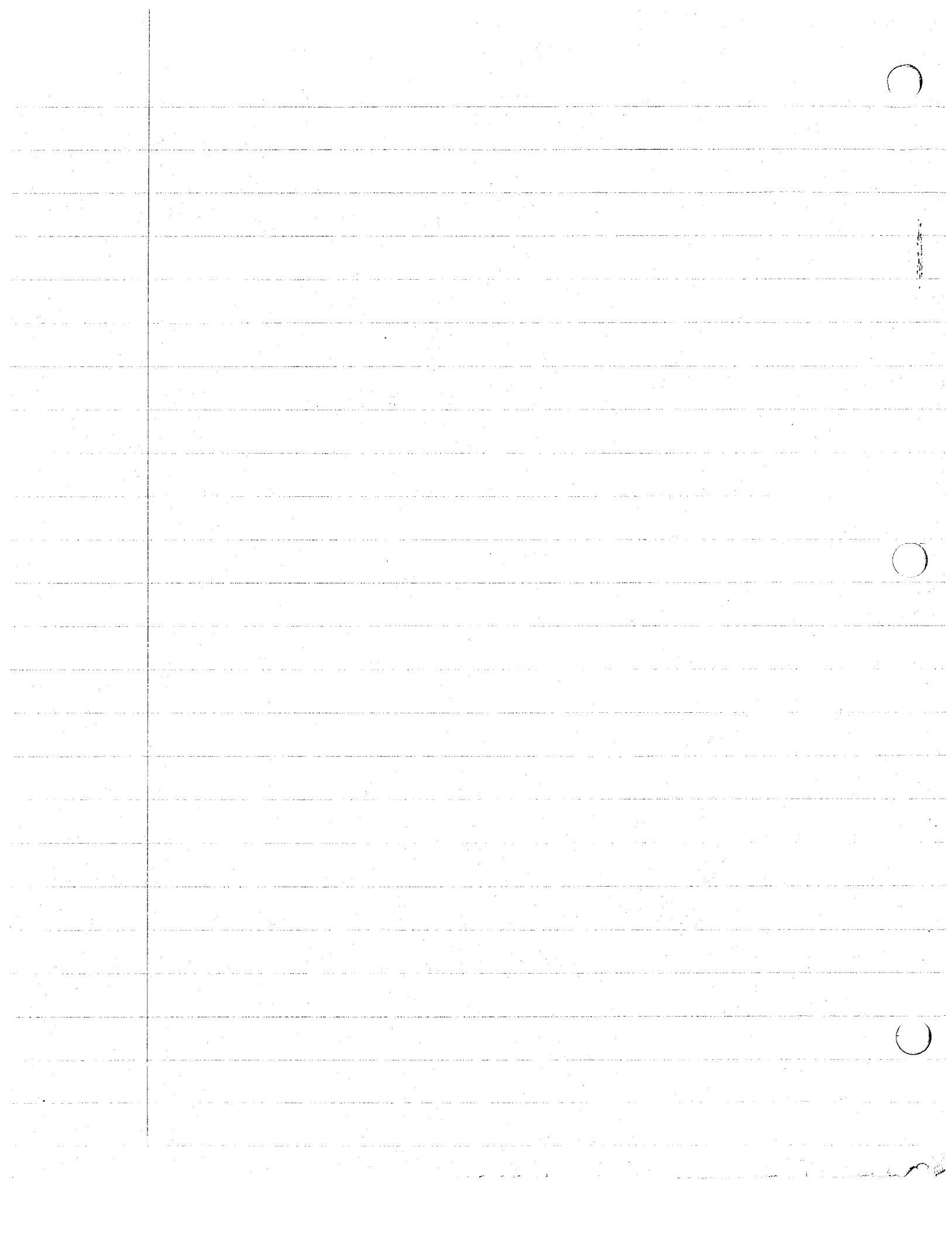


If ball 1 & 2 are replaced by 1 ball with same force system,



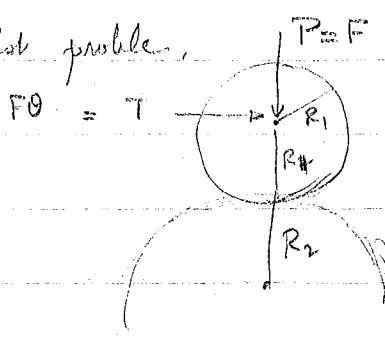
which sysyem would cause largest contact surface on the half space

$2r_1 + r_2 = R$



for small θ

for first problem,



$$F\theta = T$$

use these results to find

$$\alpha = (k_1 + k_2) \frac{3P\pi}{4a} \quad k_i = \frac{1 - v_i^2}{\pi E_i}$$

where

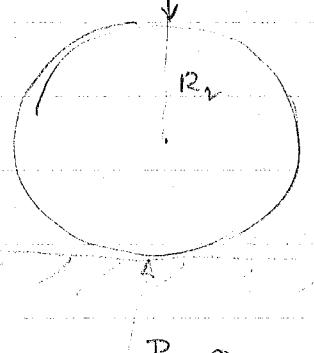
$$a = \sqrt[3]{(k_1^b + k_2^s) 3P\pi R_1 R_2}$$

$$\text{also } \phi_2 = \frac{3P}{2\pi a^2}$$

what is no slip requires if coeff of friction is f_1

$$c_1 = a_1 \sqrt[3]{1 - \frac{\theta}{f_1}}$$

for second problem,



$$u_x = \frac{3f_1 F(2-v)}{16} \left[1 - \left\{ 1 - \frac{\theta}{f_1} \right\}^{2/3} \right]$$

$$\alpha = (k_1 + k_2) \frac{6P\pi}{4a}$$

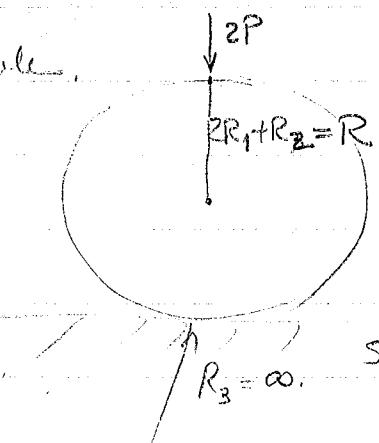
where

$$a = \sqrt[3]{(k_1^b + k_2^s) 6P\pi R_2}$$

since there is no T $c_2 = a_2$

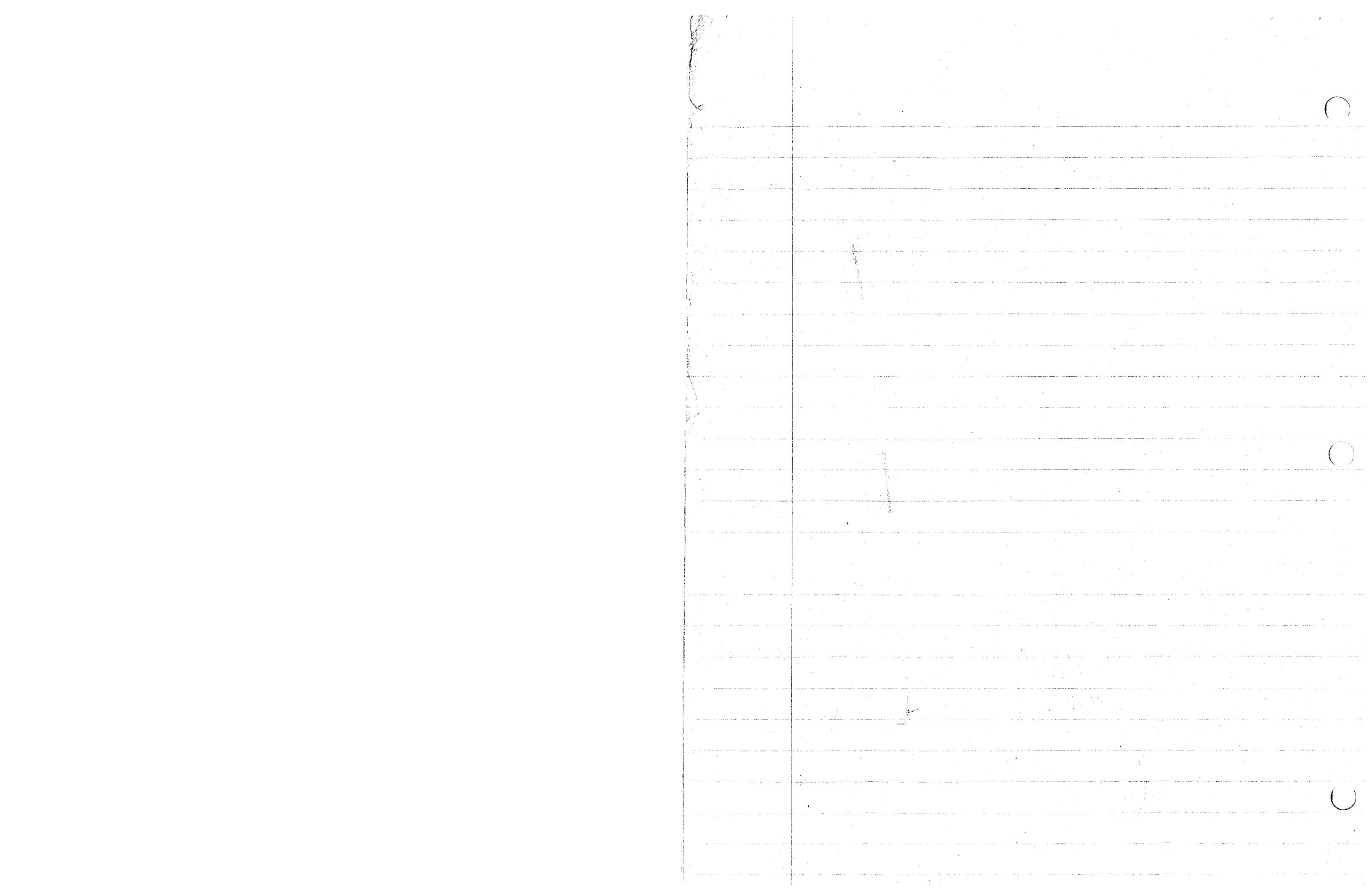
$$R_3 = \infty$$

for 3rd problem,



$$\alpha = (k_1^b + k_2^s) \frac{6P}{4a}$$

$$\text{where } a = \sqrt[3]{(k_1^b + k_2^s) 6P\pi (2R_1 + R_2)}$$



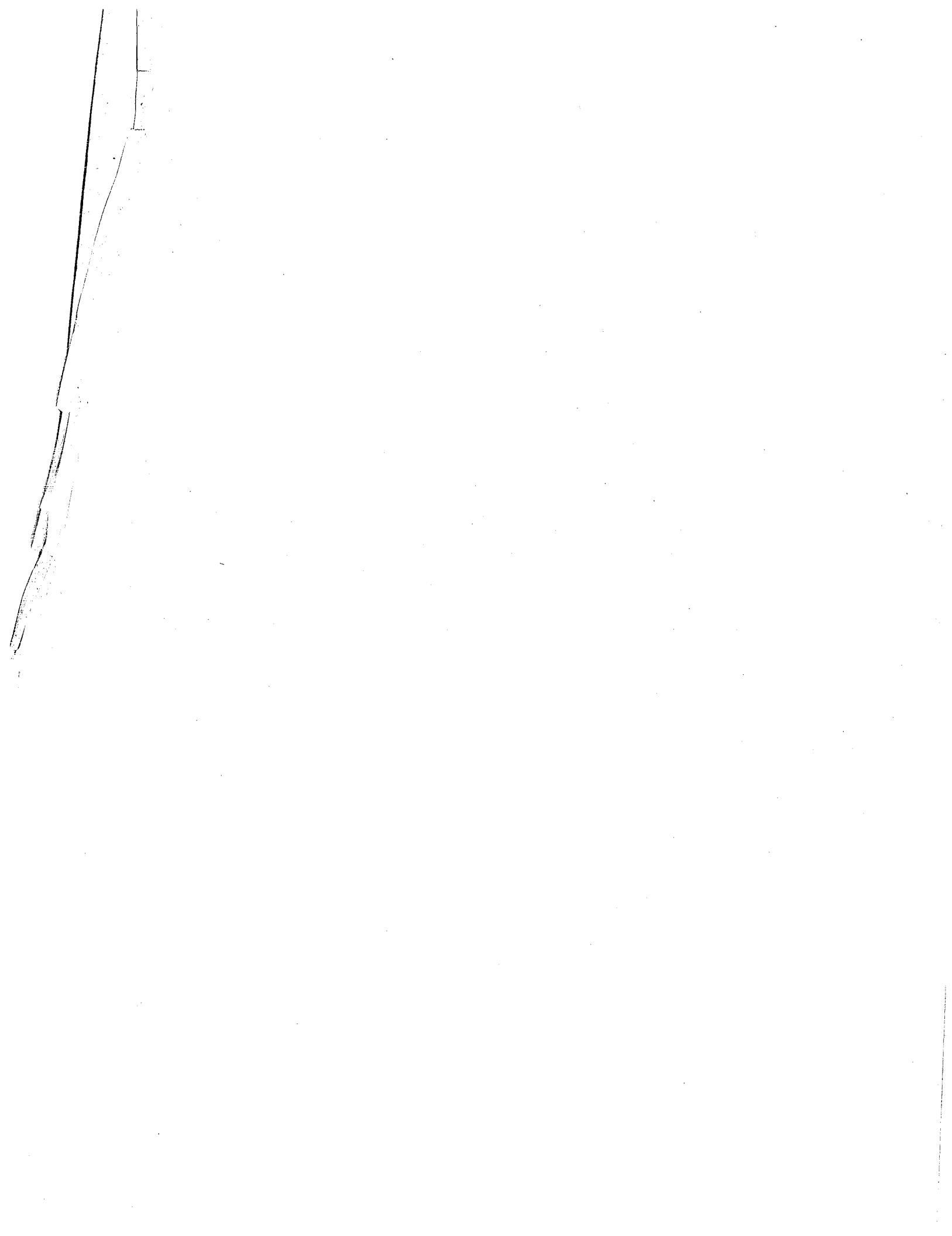
DIVISION OF APPLIED MECHANICS
DEPARTMENT OF MECHANICAL ENGINEERING
STANFORD UNIVERSITY

ME 238C Theory of Elasticity

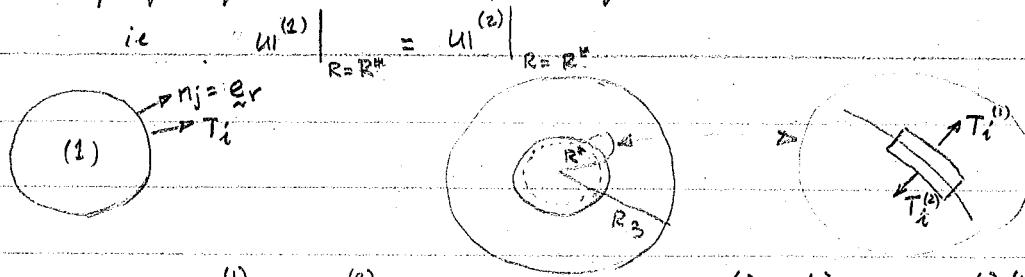
Spring 1980

Final Exam

1. A sphere of radius R_1 is subjected to uniform hydrostatic pressure, such that its radius is reduced to a value R_2 . Then it is placed into a hollow sphere, (which, of course, had to be cut open and then glued together again) of inner radius R_2 and outer radius R_3 . The given material properties of the two spheres are different. Determine the stress distribution in the system consisting of the two spheres. Compare your results to the results of the corresponding two-dimensional problem, i.e. shrink-fitted cylinders.
2. Determine the displacement distribution (which would lead to determination of stresses) produced by pinching the arm of a friend. Consider the arm in first approximation as a half-space.
3. Determine the area of contact and the approach in the hip joint subjected to a normal load. The joint is essentially a ball (radius R_1) and socket (radius R_2) arrangement. Be explicit as to your assumptions.



1. Due to symmetry of problem the only nonzero traction will be T_R
 our boundary conditions will be the continuity of traction and
 continuity of displacements at the interface



Similarly by equilb $\sigma_{ij}n_j = \sigma_{ij}^{(2)}n_j^{(2)}$; this comes from $T_i^{(1)} + T_i^{(2)} = 0 \Rightarrow \sigma_{ij}n_j^{(1)} = T_i^{(1)} = -T_i^{(2)}$
 but $-T_i^{(2)} = -\sigma_{ij}^{(2)}n_j^{(2)}$; however $-n_j^{(2)} = n_j^{(1)}$ $\therefore \sigma_{ij}n_j^{(1)} = \sigma_{ij}n_j^{(2)}$

If we define everything from the deformed state

Assume that the traction when the system reaches equilibrium

is p Then by results in class for sphere (2)

$$\sigma_{RR}^{(2)} = p \frac{R^* R_3^3}{(R^* R_3^3 - R_3^3) R_3^3} + p \frac{R^*}{(R^* R_3^3 - R_3^3)} = p R^* \frac{R_3^3 [R_3^3 - R^3]}{(R^* R_3^3 - R_3^3) R_3^3}$$

$$\sigma_{\phi\phi}^{(2)} = \sigma_{\theta\theta}^{(2)} = -p \frac{R^* R_3^3}{2(R^* R_3^3 - R_3^3) R_3^3} = -p \frac{R^*}{(R^* R_3^3)} = -p R^* \frac{R_3^3 [R_3^3 + 2R^3]}{2R^3 [R^3 - R_3^3]}$$

all the other stresses are zero. Now $I_T^{(2)} = \sigma_{RR}^{(2)} + \sigma_{\theta\theta}^{(2)} + \sigma_{\phi\phi}^{(2)} = \frac{-3pR_2^3}{R_2^3 - R_3^3}$

For the inner sphere (1)

$$\sigma_{RR}^{(1)} = -p \quad \sigma_{\phi\phi}^{(1)} = \sigma_{\theta\theta}^{(1)} = -p \quad I_T^{(1)} = -3p$$

$$\text{thus } \epsilon_{RR}^{(2)} = \frac{\partial u_R^{(2)}}{\partial R} = \frac{1 + \nu^{(2)}}{E^{(2)}} \sigma_{RR}^{(2)} = \frac{\nu^{(2)}}{E^{(2)}} I_T^{(2)} = \frac{\sigma_{RR}^{(2)}}{E^{(2)}} - \frac{\nu^{(2)}}{E^{(2)}} (\sigma_{\phi\phi}^{(2)} + \sigma_{\theta\theta}^{(2)}) \\ = \frac{1 + \nu^{(2)}}{E^{(2)}} p \frac{R^* R_3^3}{(R^* R_3^3 - R_3^3) R_3^3} + \frac{2\nu - 1}{E^{(2)}} \frac{p R^*}{R^* R_3^3 - R_3^3}$$

$$\text{thus } u_R^{(2)} = -\left(\frac{1 + \nu^{(2)}}{2E^{(2)}}\right) \frac{p R^* R_3^3}{(R^* R_3^3 - R_3^3) R_3^3} + \frac{2\nu - 1}{E^{(2)}} \frac{p R^*}{R^* R_3^3 - R_3^3} R + f(\theta, \phi) \text{ by symmetry}$$

$$\text{Similarly } \epsilon_{RR}^{(1)} = \frac{\partial u_R^{(1)}}{\partial R} = \frac{1 + \nu^{(1)}}{E^{(1)}} \sigma_{RR}^{(1)} - \frac{\nu^{(1)}}{E^{(1)}} I_T^{(1)} = -\frac{1 + \nu^{(1)}}{E^{(1)}} p + \frac{3\nu^{(1)}}{E^{(1)}} p = \frac{2\nu - 1}{E^{(1)}} p$$

$$\text{and } u_R^{(1)} = \frac{2\nu - 1}{E^{(1)}} p R + f(\theta, \phi) \text{ by symmetry}$$

$$\text{thus at } R = R^* \quad u_R^{(1)} = u_R^{(2)} \quad \therefore \frac{2\nu - 1}{E^{(1)}} p R^* = -\left(\frac{1 + \nu^{(1)}}{2E^{(1)}}\right) \frac{p R^* R_3^3}{(R^* R_3^3 - R_3^3) R_3^3} + \frac{2\nu - 1}{E^{(2)}} \frac{p R^* R_3^3}{(R^* R_3^3 - R_3^3) R_3^3}$$

$$\text{Let } \frac{2V-1}{E} = A \quad \frac{1+V}{2E} = B$$

$$A^{(1)} [R^{*3} - R_3^3] + B^{(2)} R_3^3 - A^{(2)} R^{*3} = 0$$

$$R^{*3} [A^{(1)} - A^{(2)}] + B^{(2)} R_3^3 - A^{(2)} R_3^3 = 0$$

$$R^* = \left[\frac{A^{(2)} - B^{(2)}}{A^{(1)} - A^{(2)}} \right]^{1/3} R_3 \quad \therefore \quad \left(\frac{R^*}{R_3} \right)^3 = \frac{A^{(2)} - B^{(2)}}{A^{(1)} - A^{(2)}}$$

$$\text{Now } u^{(2)} = -\left(\frac{1+V^{(2)}}{2E^{(2)}} \right) \frac{PR^{*3} R_3^3}{(R^{*3} - R_3^3) R^2} + \frac{2V^{(2)} - 1}{E^{(2)}} \frac{PR^{*3} R}{(R^{*3} - R_3^3)}$$

$$\begin{aligned} \text{at } R = R^* \quad u^{(2)} &= R^* - R_2 = -\left(\frac{1+V^{(2)}}{2E^{(2)}} \right) \frac{PR^* R_3^3}{R^{*3} - R_3^3} + \frac{(2V^{(2)} - 1)}{E^{(2)}} \frac{PR^{*4}}{R^{*3} - R_3^3} \\ &= -B^{(2)} PR^* + \frac{A^{(2)} PR^{*4}}{A^{(1)} - 2A^{(2)} + B^{(2)}} \\ &= -PR^* \left\{ \frac{1 - \left(\frac{A^{(2)} - B^{(2)}}{A^{(1)} - A^{(2)}} \right)}{A^{(1)} - 2A^{(2)} + B^{(2)}} - R_3^3 \left[1 - \left(\frac{A^{(2)} - B^{(2)}}{A^{(1)} - A^{(2)}} \right) \right] \right\} \\ &= -PR^* \left\{ \frac{-B^{(2)} (A^{(1)} - A^{(2)})}{A^{(1)} - 2A^{(2)} + B^{(2)}} + \frac{A^{(2)} \left[\frac{A^{(1)} - B^{(2)}}{A^{(1)} - A^{(2)}} \right]}{A^{(1)} - 2A^{(2)} + B^{(2)}} \right\} \end{aligned}$$

$$-PR^* \frac{[A^{(1)}]^2 - B^{(2)} A^{(1)}}{A^{(1)} - 2A^{(2)} + B^{(2)}} = R^* - R_2 \quad \therefore P = \frac{(1 - R_2/R^*)[A^{(1)} - 2A^{(2)} + B^{(2)}]}{[A^{(1)}]^2 - B^{(2)} A^{(1)}}$$

$$V_{RR}^{(2)} = \frac{P(R^* R_3)^3}{(R^{*3} - R_3^3) R^3} = \frac{PR^{*3}}{(R^{*3} - R_3^3)}$$

$$\sigma_{\phi\phi}^{(2)} = \sigma_{\theta\theta}^{(2)} = -P \frac{(R^* R_3)^3}{2(R^{*3} - R_3^3) R^3} = \frac{PR^{*3}}{R^{*3} - R_3^3}$$

$$\sigma_{R\phi}^{(2)} = \sigma_{R\theta}^{(2)} = \sigma_{\phi\theta}^{(2)} = 0$$

and

$$\sigma_{RR}^{(1)} = \sigma_{\phi\phi}^{(1)} = \sigma_{\theta\theta}^{(1)} = -P \quad \sigma_{R\phi}^{(1)} = \sigma_{R\theta}^{(1)} = \sigma_{\phi\theta}^{(1)} = 0$$

$$u_{R_{\text{idle}}} = -P_{\text{idle}} (1 - 2V)(1 + V)$$

Another way of doing it is to measure everything from the undeformed state

$$\sigma_{ij}\eta_j = \sigma_{ij}\eta'_j \quad \text{at } R=R_2 \quad \Rightarrow \quad \sigma_{RR}^{(1)} = \sigma_{RR}^{(2)}$$

$$u_R^{(1)} - u_R^{(2)} = R_2 - R_1 \quad \text{at } R_2 = R$$

$$\sigma_{RR}^{(2)} = \frac{p(R_2 R_3)^3}{(R_2^3 - R_3^3) R^3} = \frac{p(R_2)^3}{(R_2^3 - R_3^3)} \quad \sigma_{\theta\theta}^{(2)} = \sigma_{\phi\phi}^{(2)} = -\frac{p(R_2 R_3)^3}{2(R_2^3 - R_3^3) R^3} = \frac{-p R_2^3}{R_2^3 - R_3^3}$$

all others are zero

$$\text{and } \sigma_{RR}^{(1)} = \sigma_{\theta\theta}^{(1)} = \sigma_{\phi\phi}^{(1)} = -p \quad \text{all others are zero.}$$

$$\text{Now } u_R^{(2)} = -B^{(2)} \frac{p(R_2 R_3)^3}{(R_2^3 - R_3^3) R^2} + A^{(2)} \frac{p R_2^3 R}{(R_2^3 - R_3^3)} ; \quad \text{at } R=R_2 \quad u_R^{(2)} = -\frac{p A^{(2)} R_2 R_3^3}{R_2^3 - R_3^3} + \frac{B^{(2)} p R_2^4}{R_2^3 - R_3^3}$$

$$\text{and } u_R^{(1)} = A^{(1)} p R \quad ; \quad \text{at } R=R_2 \quad A^{(1)} p R_2 = u_R^{(1)}$$

$$\therefore u_E^{(1)} - u_R^{(2)} = R_2 - R_1 = p R_2 \left\{ \frac{A^{(1)} (R_2^3 - R_3^3)}{R_2^3 - R_3^3} + B^{(2)} \frac{R_2^3}{R_2^3 - R_3^3} + A^{(2)} \frac{R_2^3}{R_2^3 - R_3^3} \right\}$$

$$1 - \frac{R_1}{R_2} = p \left[\frac{R_2^3 \{ A^{(1)} - B^{(2)} \} + R_3^3 \{ A^{(2)} - A^{(1)} \}}{R_2^3 - R_3^3} \right]$$

$$\therefore p = \left(1 - \frac{R_1}{R_2} \right) \frac{(R_2^3 - R_3^3)}{R_2^3 \{ A^{(1)} - B^{(2)} \} + R_3^3 \{ A^{(2)} - A^{(1)} \}}$$

$$\text{for 2-D case} \quad \sigma_{RR} = \frac{R_2^2 R_3^2 (-p_i)}{r^4 \frac{R_2^3 - R_3^3}{R_3^3 - R_2^3}} + \frac{p_i R_2^2}{R_3^3 - R_2^3} = -\frac{A p_i}{r^2} + \frac{B p_i}{r^2} \quad \text{for outer}$$

$$\sigma_{\theta\theta} = \frac{R_2^2 R_3^2 p_i}{r^4 (R_3^3 - R_2^3)} + \frac{p_i R_2^2}{R_3^3 - R_2^3} \quad \sigma_{zz} = \nu (\sigma_{RR} + \sigma_{\theta\theta})$$

$$\text{for inner} \quad \sigma_{RR} = -p_e \quad \sigma_{\theta\theta} = -p_e \quad \sigma_{zz} = \nu (-2p_e)$$

$$\begin{aligned} E_{RR} &= \frac{\sigma_{RR}}{E} = \frac{1}{E} ((1+\nu) \sigma_{\theta\theta} + \nu \sigma_{zz}) = \frac{1-\nu^2}{E} \sigma_{\theta\theta} = \frac{\nu(1+\nu)}{E} \sigma_{\theta\theta} \\ &= \frac{1-\nu^2}{E} \left(-\frac{A p_i}{r^2} + \frac{B p_i}{r^2} \right) = \frac{\nu(1+\nu)}{E} \left(A p_i + B p_i \right) \\ &= \frac{A p_i}{E r^2} \left[-\left(1 - \frac{R_1}{R_2} \right) - \frac{R_2^3 - R_3^3}{R_3^3 - R_2^3} \right] + \frac{B p_i}{E r^2} \left(1 - \frac{R_1}{R_2} - \frac{R_2^3 - R_3^3}{R_3^3 - R_2^3} \right) \\ \frac{\partial \sigma_{RR}}{\partial r} &= -\frac{((1+\nu) A p_i + B p_i)(1-2\nu)(1+\nu)}{E r^3} \end{aligned}$$

$$u_R = +\frac{(1+\nu) A p_i}{E} + \frac{B p_i}{E} (1-2\nu)(1+\nu) r ; \quad \text{inner } E_{RR} = \frac{p_e}{E} = \frac{p_e}{E} (p_e - 2\nu p_e) = \frac{p_e}{E} (1 - \nu - 2\nu^2)$$

$$\frac{1}{R(R+2)} + \frac{x}{(R+2)} - \frac{x}{R^3} + \frac{x}{R^2(R+2)^2}$$

$$\frac{1}{R(R+2)} + \frac{-x^2(R+2)}{R^3(R+2)^2} = \frac{x^2}{R(R+2)^2}$$

$$\frac{\partial}{\partial x} \frac{1}{R^2} = \frac{-2}{R^3} \cdot \frac{2}{R^2} = \frac{-3x}{R^5}$$

$$\frac{\partial}{\partial x} \frac{x^2}{(R^2+Rz)^2} = \frac{x^2}{(R^4 R z)^2} + \frac{-2x^2 z}{(R^2+Rz)^3} \cdot \left(\frac{2z}{R} + \frac{2}{R^2} + \frac{R}{R} \right)$$

$$= x^2(R^2+Rz) - 4x^2 z^2 = 2x^2 z^2 - 2x^2 z R$$

$$x^2 R z^2 - x^2 z R = 2x^2 z^2 - 4x^2 z^2$$

$$\frac{\partial}{\partial x} \left(\frac{x}{R(R+2)} \right) = \frac{1}{R(R+2)} + \frac{x}{R^2} \cdot \left(\frac{2z}{R} + \frac{2}{R^2} \right)$$

$$\frac{\partial}{\partial x} \left(\frac{x}{R} \right) = \frac{1}{R} + \frac{x^2}{R^2} \cdot \frac{1}{R^2(R+2)^2} = \frac{1}{R^2(R+2)} \cdot \frac{x^2}{R^2(R+2)^2}$$

$$\frac{\partial}{\partial x} \left[\frac{Rz}{R(R+2)} \right] = \frac{2}{R(R+2)} + \frac{xz}{R(R+2)} \cdot \left(\frac{2z}{R} + \frac{2}{R^2} \right)$$

$$\frac{\partial}{\partial x} \frac{x}{R(R+2)} = \frac{1}{R+2} - \frac{x^2}{R(R+2)^2}$$

$$\frac{\partial}{\partial x} \left[\frac{1-2v}{2\pi\mu} A \left[\frac{-3x^2}{R^4} + \frac{1}{R^2} \right] \right] = \frac{1}{4(1-v)\pi\mu} \left[(1-2v) \left[\frac{-3x^2}{R^4} + \frac{1}{R^2} \right] - (1-2v)^2 \left(\frac{6x^2}{R^4} - \frac{1}{R^2} \right) \right]$$

$$= \frac{1}{2\pi\mu} \left[\frac{(1-2v)^2}{R^2} \left(\frac{-3x^2}{R^2} + \frac{1}{R^2} \right) + vR^2(R+2)^2 \left(\frac{6x^2}{R^4} - \frac{1}{R^2} \right) \right]$$

$$= \frac{1}{(R+2)^2} \left(\frac{6x^2}{R^2} - \frac{1}{R^2} \right)$$

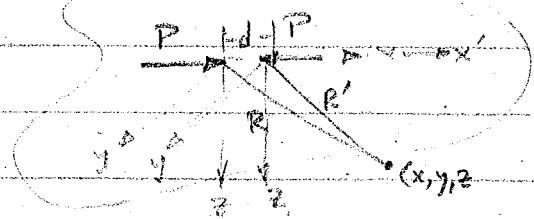
$$\frac{1-2v}{2\pi\mu} \left[\frac{-x^2 + 1}{R^4} + \frac{1}{4(1-v)} \left[\frac{x^2 + 1}{R^4} - \frac{(1-2v)^2}{R^4} \right] \right]$$

$$= 4(1-v) + 1$$

$$\frac{-3+4v}{4(1-v) R^4} x^2$$

for concentrated forces

$$U_{11} = \frac{1}{2} B_3 + \frac{1}{4(1-v)} \nabla \left((R \cdot B_3 + \beta) \right)$$



$$B_{11} = \frac{P \epsilon_x}{2\pi\mu R} + \frac{(1-2v) Px \epsilon_z}{2\pi\mu R(R+2)}$$

$$\beta_{11} = -\frac{(1-2v)^2 Px}{2\pi\mu(R+2)}$$

$$U_{12} = U_{11}(x', y, z; P) \text{ where } x' = x-d \text{ and } x \neq x'+d$$

$$U^{(2)} = U^{(1)} = \frac{\partial U^{(1)}}{\partial x}, \quad U^{(2)}_{(1)} = -U^{(1)} + d \frac{\partial U^{(1)}}{\partial x}$$

$$U_{11} = U_{11(1)} + U_{11(2)} = \frac{P \epsilon_x}{2\pi\mu} \left[\frac{1}{R} - \frac{1}{R'} \right] + \frac{1-2v}{2\pi\mu} P \epsilon_z \left[\frac{x}{R(R+2)} - \frac{(x-d)}{R'(R'+2)} \right] - \frac{1}{4(1-v)} \nabla \left\{ \frac{P}{2\pi\mu} \left[\frac{x}{R} - \frac{x}{R'} \right] \right.$$

$$\left. + \frac{1-2v}{2\pi\mu} P \left[\frac{xz}{R(R+2)} - \frac{(x-d)z}{R'(R'+2)} \right] - \frac{(1-2v)^2 P}{2\pi\mu} \left[\frac{x}{R+2} - \frac{(x-d)}{R'+2} \right] \right\}$$

$$\text{Now } \lim_{d \rightarrow 0} \frac{1}{R} - \frac{1}{R'} = -\frac{x}{R^3} \Rightarrow \frac{d}{RR'} \frac{R'-R}{d} = \frac{d}{RR'} \frac{\sqrt{R^2+d^2}-2dx-R}{d} = \frac{d}{RR} \left[\frac{R(1+d^2/2dx)}{2R^2} - \frac{d}{2R} \left[\frac{d}{2R} - \frac{d}{R} \right] \right] \Rightarrow$$

$$\lim_{d \rightarrow 0} \frac{x(R^2+R'^2) - x(R^2+Rz) + d(R^2+Rz)}{RR'(R+2)(R'+2)} = \frac{xd}{R^2(R+2)^2} \left\{ \frac{(x-d)^2-x^2}{d} + 2 \frac{(R'-R)}{d} \right\} + \frac{d}{R(R+2)}$$

$$\lim_{d \rightarrow 0} \left(\frac{x}{R(R+2)} - \frac{x-d}{R'(R'+2)} \right) = \frac{xd}{R^2(R+2)^2} \left\{ -2x - \frac{zx}{R} \right\} + \frac{d}{R(R+2)}$$

$$\lim_{d \rightarrow 0} \left[\frac{x}{R+2} - \frac{(x-d)}{R'+2} \right] = \lim_{d \rightarrow 0} \frac{XR' - XR + d(R+2)}{(R+2)^2} = \frac{dx}{(R+2)^2} \frac{(R'-R)}{d} = \frac{dx}{(R+2)^2} \left[-\frac{x}{R} \right] + \frac{d}{R+2}$$

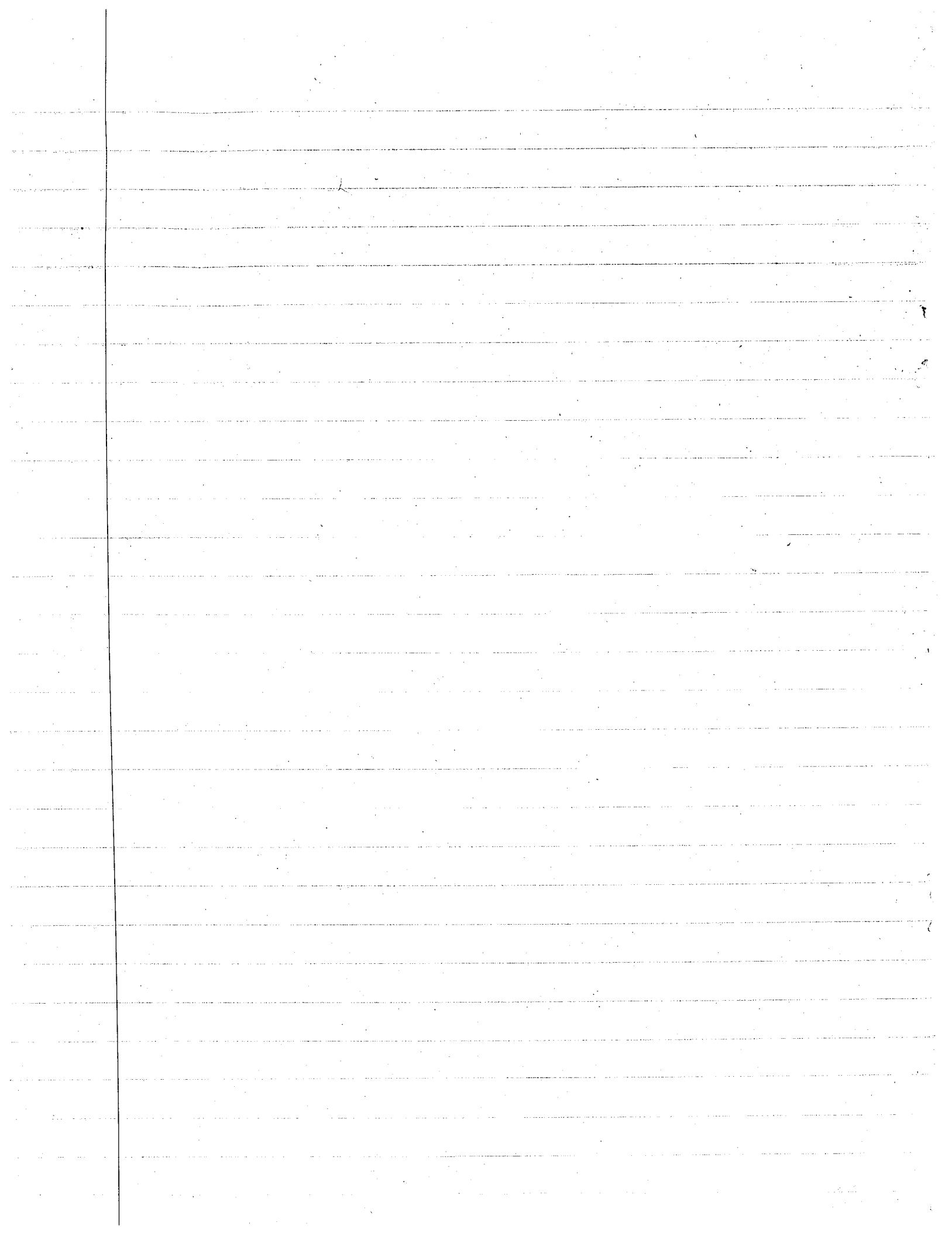
$$\therefore U_{11} = \frac{A \epsilon_x}{2\pi\mu} \left[-\frac{x}{R^3} \right] + \frac{1-2v}{2\pi\mu} A \epsilon_z \left[-\frac{2x^2}{R^2(R+2)^2} - \frac{2x^2}{(R^2(R+2))^2} \right] - \frac{1}{4(1-v)} \nabla \left\{ \frac{A}{2\pi\mu} \left(\frac{-x^2}{R^2} + \frac{1}{R} \right) \right.$$

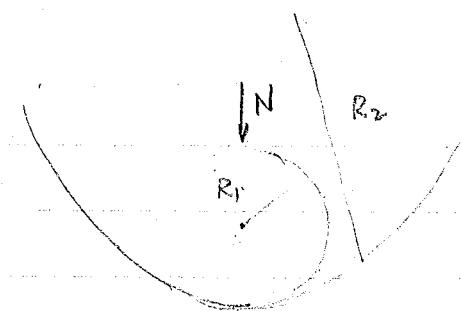
$$\left. + \frac{1-2v}{2\pi\mu} A \left[\frac{-2x^2}{R^2(R+2)^2} - \frac{zx^2}{R^3(R+2)^2} \right] - \frac{(1-2v)^2 A}{2\pi\mu} \left[\frac{-x^2}{R(R+2)^2} + \frac{1}{(R+2)} \right] \right\}$$

$$U_x = -\frac{A}{2\pi\mu R^3} - \frac{A}{4(1-v)2\pi\mu} \frac{\partial}{\partial x} \left[\left(\frac{-x^2}{R^3} + \frac{1}{R} \right) (1-2v) \left[\frac{-2x^2}{R^2(R+2)^2} - \frac{2x^2}{R^3(R+2)^2} \right] \right] - (1-2v)^2 \left(\frac{-x^2}{R(R+2)^2} \right)$$

$$U_y = -\frac{A}{4(1-v)2\pi\mu} \frac{\partial}{\partial y} \left[\left(\frac{-x^2}{R^3} + \frac{1}{R} \right) (1-2v) \left[\frac{-2x^2}{R^2(R+2)^2} - \frac{2x^2}{R^3(R+2)^2} \right] \right] - (1-2v)^2 \left(\frac{-x^2}{R(R+2)^2} \right)$$

$$U_z = \frac{1-2v}{2\pi\mu} A \left[\frac{-2x^2}{R^2(R+2)^2} - \frac{zx^2}{R^3(R+2)^2} \right] - \frac{A}{4(1-v)2\pi\mu} \frac{\partial}{\partial z} \left[\left(\frac{-x^2}{R^3} + \frac{1}{R} \right) (1-2v) \left[\frac{-2x^2}{R^2(R+2)^2} - \frac{zx^2}{R^3(R+2)^2} \right] \right] - (1-2v)^2 \left(\frac{-x^2}{R(R+2)^2} \right)$$





$$a = \frac{3}{4} \sqrt{\frac{(k_1+k_2)3N\pi}{R_1 R_2}} \sqrt{\frac{(k_1+k_2)\mu_1 \mu_2 R_1 R_2}{4(R_2 - R_1)}} \quad \text{Eq 3}$$

$$\therefore \text{area of contact } \pi a^2 = \pi \left[\frac{(k_1+k_2)3N\pi R_1 R_2}{4(R_2 - R_1)} \right]^{1/2} \quad \text{Eq 3}$$

R ball = R_1 , R socket = $-R_2$

$$\begin{aligned} \text{if socket is rigid } k_2 = 0 \quad \pi a^2 &= \pi \left[\frac{1-\nu_1}{2\mu_1} \cdot \frac{3N R_2 R_1}{4(R_2 - R_1)} \right]^{1/2} \\ &= \pi \left[\frac{1-\nu_1}{\mu_1} \cdot \frac{3N R_2 R_1}{8(R_2 + R_1)} \right]^{1/2} \end{aligned}$$

$$\alpha = \frac{3}{16} \sqrt{\frac{9N^2 \pi^2 (k_1+k_2)^2}{R_1 R_2}} = \sqrt{\frac{9N^2 \pi^2 (k_1+k_2)^2 (R_2 - R_1)}{16 R_1 R_2}} \quad \text{for rigid socket } k_2 = 0$$

$$\alpha = \sqrt{\frac{9N^2 \pi^2 k_1^2 (R_2 - R_1)}{16 R_1 R_2}} = \sqrt{\frac{9N^2 (1-\nu)^2 (R_2 - R_1)}{64 \mu_1^2 R_1 R_2}}$$

$$\text{if socket pliable, } a = \sqrt{\frac{k_1 \cdot 3N R_2 R_1}{2(R_2 R_1)}} = \sqrt{\frac{(1-\nu)}{4\mu_1} \cdot \frac{3N R_2 R_1}{R_2 - R_1}}$$

$$\alpha = \sqrt{\frac{9N^2 \pi^2 k_1 (R_2 - R_1)}{4 R_1 R_2}} = \sqrt{\frac{9N^2 (1-\nu)^2 (R_2 - R_1)}{16 \mu_1^2 R_1 R_2}}$$

$$\text{cont area} = \pi \sqrt{\frac{(1-\nu)^2 \cdot 9N^2 R_2 R_1}{16 \mu_1^2 (R_2 + R_1)^2}}^{1/2}$$

