

$$y^3 + py^2 + qy + r = 0$$

can be reduced to $x^3 + ax + b = 0$ by letting $y = x - \frac{p}{3}$

$$a = \frac{1}{3}(3q - p^2) \quad b = \frac{1}{27}(2p^3 - 9pq + 27r)$$

Let

$$A = \sqrt[3]{-\frac{b}{2} + \sqrt{\frac{b^2}{4} + \frac{a^3}{27}}} \quad B = \sqrt[3]{-\frac{b}{2} - \sqrt{\frac{b^2}{4} + \frac{a^3}{27}}}$$

$$x = A + B, \quad -\frac{A+B}{2} + \frac{A-B}{2}\sqrt{-3}, \quad -\frac{A+B}{2} - \frac{A-B}{2}\sqrt{-3}$$

$\frac{b^2}{4} + \frac{a^3}{27} > 0$	1 real	2 imm
	≥ 0	3 real & 2 conj.
	< 0	3 real & unique

quartic

$x^4 + ax^3 + bx^2 + cx + d = 0$ can be rewritten

$$y^3 - by^2 + (ac - 4d)y - a^2d + 4bd - c^2 = 0$$

$$R = \sqrt{\frac{3a^2}{4} - b + y} \text{ where } y \uparrow$$

$$\text{if } R \neq 0 \quad D = \sqrt{\frac{3a^2}{4} - R^2 - 2b + \frac{4ab - 8c - a^3}{4R}}$$

$$E = \sqrt{\frac{3a^2}{4} - R^2 - 2b - \frac{4ab - 8c - a^3}{4R}}$$

$$R \neq 0 \quad D = \sqrt{\frac{3a^2}{4} - 2b + 2\sqrt{y^2 - 4d}}$$

$$E = \sqrt{\frac{3a^2}{4} - 2b - 2\sqrt{y^2 - 4d}}$$

then 4 roots

$$x = -\frac{a}{4} + \frac{R}{2} + \frac{D}{2}$$

$$x = -\frac{a}{4} - \frac{R}{2} + \frac{E}{2}$$

$$\frac{dv}{dt} + \frac{k}{m} v + g = 0$$

$$\frac{dv}{dt} + \frac{k}{m} v + g = 0 \quad v = \frac{m}{k}$$

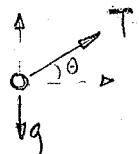
$$(\times \frac{m}{k}) \frac{dv}{dt} + v = 0 \quad v = -\frac{mg}{k}$$

$$e^{\int \frac{m}{k} dt} v = C_1 \quad v = C_1 e^{-\frac{mt}{k}}$$

$$v = C_1 e^{-\frac{mt}{k}}$$

$$m = m_0 - m t = m_0 + p t$$

①



$$\begin{aligned}\ddot{u} &= \bar{a} \cos \theta \\ \ddot{v} &= \bar{a} \sin \theta - g \\ \dot{x} &= u \\ \dot{y} &= v\end{aligned}$$

$$\ddot{a} = \frac{T}{m} = \frac{c\beta}{m}$$

$$\text{also } x=y=u=v=0 \quad t_0=0 \quad t_1=T$$

$$\bar{a} = -\frac{cdM}{Mdt} = +\frac{c}{M} \beta \quad \beta \text{ is already known}$$

no drag uniform field

$$\begin{array}{l} \uparrow v_1 \\ \rightarrow u_1 \end{array}$$

$$\begin{array}{l} y_1 \\ x_1 \end{array}$$

range

$$\tan \theta_1 = \frac{v_1}{u_1} = \frac{v_1}{u}$$

$$\sin \theta_1 = \frac{v_1}{\sqrt{u_1^2 + v_1^2}} = \frac{v_1}{v}$$

$$u_1 = v \cos \theta_1$$

$$v_1 = v \sin \theta_1 - gt$$

$$x_1 = (v \cos \theta_1) t_1$$

$$y_1 = (v \sin \theta_1) t_1 - \frac{1}{2} g t_1^2 = y_1$$

$$\frac{1}{2} g t_1^2 - v \sin \theta_1 t_1 - y_1 = 0$$

$$t_1 = \frac{v \sin \theta_1 \pm \sqrt{v^2 \sin^2 \theta_1 + 2gy_1}}{g}$$

$$t = \frac{v_1 + \sqrt{v_1^2 + 2gy_1}}{g}$$

$$x_r = v^2 \sin \theta_1 \cos \theta_1 \pm \sqrt{v^2 \sin^2 \theta_1 + 2gy_1} = \frac{u_1}{g} [v_1 + \sqrt{v_1^2 + 2gy_1}]$$

$$x_r = \frac{v_1 u_1}{g} + \frac{u_1}{g} \sqrt{v_1^2 + 2gy_1}$$

$$R = x_1 + \frac{u_1}{g} [v_1 + \sqrt{v_1^2 + 2gy_1}]$$

$$dx = du_1 + \frac{u_1}{g} dv_1, \quad dy = dv_1 - \frac{1}{2} g dt_1, \quad \frac{du_1}{dt_1} = \frac{u_1}{g}, \quad \frac{dv_1}{dt_1} = \frac{v_1}{g}$$

$$u_1, v_1, x_1, y_1, \theta$$

$$F = \lambda_u [\ddot{u} - a \cos \theta] + \lambda_v [\ddot{v} - a \sin \theta + g] + \lambda_x [\dot{x} - u] + \lambda_y [\dot{y} - v]$$

$$-\lambda_x - \lambda u = 0 \quad -\lambda y - \lambda v = 0 \quad -\lambda \dot{x} = 0 \quad -\lambda \dot{y} = 0$$

$$+ a (\lambda u \sin \theta - \lambda v \cos \theta) = 0 \quad \theta$$

$$\lambda x = c_1, \quad \lambda y = c_2, \quad \lambda u = -c_1 t + c_{11}, \quad \lambda v = -c_2 t + c_{22}$$

From transversality FI $\lambda_u \dot{u} + \lambda_v \dot{v} + \lambda_x \dot{x} + \lambda_y \dot{y} = 0$

transversality - dx $-Cdt + \lambda_u du + \lambda_v dv + \lambda_x dy + \lambda_y dy \Big|_0^T = 0$

$$C=0 \quad \lambda_{u_f} = 0, \quad \lambda_{v_f} = 0, \quad \lambda_{x_f} = 1, \quad \lambda_{y_f} = 0$$

$$\lambda_u = -c_1(t-T), \quad \lambda_v = -c_2(t-T)$$

$$\lambda_x = 1, \quad \lambda_y = 0$$

$$\text{transv} = -Cdt + \left[\lambda u + \frac{v_1 + r}{g} \right] du + \left[\lambda v + \frac{u_1(r+v_1)}{gr} \right] dv + \left[\lambda y + \frac{u_1}{r} \right] dy + \left[\lambda x + 1 \right] dx \Big|_{\lambda=0}$$

$$C=0 \quad \lambda u_f = -\frac{v_1+r}{g} \quad \lambda v_f = -\frac{u_1(r+v_1)}{gr} \quad \lambda y_f = -\frac{u_1}{r} \quad \lambda x_f = -1$$

$$r = \sqrt{v_1^2 + 2gy_1}$$

$$\lambda_1 = \lambda_x = -1 \text{ everywhere } \lambda_2 = \lambda_y = -\frac{u_1}{r} \text{ everywhere}$$

$$\tan \theta = \frac{\lambda v}{\lambda u} = \frac{u_1}{r} > \text{constant angle to horiz}$$

$$\lambda u_f = -C_1(T) + C_{11}$$

$$-\frac{(v_1+r)}{g} = +T + C_{11}, \quad C_{11} = -\left[T + \frac{(v_1+r)}{g}\right]$$

$$\lambda v_f = -C_2 T + C_{22} \quad -\frac{u_1(r+v_1)}{gr} = +\frac{u_1}{r} T + C_{22} \quad C_{22} = -\frac{u_1}{r} \left[T + \frac{(r+v_1)}{g} \right]$$

$$\therefore \lambda u = (t-T) - \frac{(v_1+r)}{g} \quad \lambda v = \frac{u_1}{r} \left[(t-T) - \frac{(r+v_1)}{g} \right] \quad \cos \theta = \frac{r}{\sqrt{r^2 + u_1^2 / r^2}}$$

$$\lambda x = -1 \quad \lambda y = -\frac{u_1}{r} \quad \sin \theta = \frac{u_1}{\sqrt{r^2 + u_1^2 / r^2}}$$

total flight time $T + \left(\frac{v_1+r}{g}\right)$

$$\text{Weierstrass Condition} \quad \lambda u \dot{u} + \lambda v \dot{v} + \lambda x \dot{x} + \lambda y \dot{y} \leq \lambda u \ddot{u} + \lambda v \ddot{v} + \lambda x \ddot{x} + \lambda y \ddot{y}$$

$$\text{where } \ddot{U} = \ddot{a} \cos \theta^*$$

$$\dot{V} = \ddot{a} \sin \theta^* - g$$

$$\ddot{x} = u = \dot{x} \quad \therefore \quad \lambda u \dot{u} + \lambda v \dot{v} \leq \lambda u \ddot{u} + \lambda v \ddot{v}$$

$$\ddot{y} = v = \dot{y} \quad \ddot{a} (\lambda u \cos \theta + \lambda v \sin \theta) \leq \ddot{a} (\lambda u \cos \theta^* + \lambda v \sin \theta^*)$$

$$-\frac{\ddot{a}}{g} (v_1 + r) \left[\cos \theta + \frac{u_1}{r} \sin \theta \right] \leq -\frac{\ddot{a}}{g} (v_1 + r) \left[\cos \theta^* + \frac{u_1}{r} \sin \theta^* \right]$$

$$\cos \theta + \frac{u_1}{r} \sin \theta \geq \cos \theta^* + \frac{u_1}{r} \sin \theta^* \quad \text{satisfied if } \theta^* \text{ if } \theta \text{ makes LHS max. } u_1 > 0$$

$\therefore a(t)$ is irrelevant in the weierstrass condition, θ is acute angle

$$\text{from PI} \quad \lambda u \dot{u} + \lambda v \dot{v} + \lambda x \dot{x} + \lambda y \dot{y} = C \quad \text{from conservation} \quad \lambda u_f = \lambda u_i$$

θ is not constant since it has a solution to $\tan \theta = \frac{u_1}{r}$

$$\lambda v_f = \lambda v_i$$

$$\lambda x_f = \lambda x_i$$

$$\lambda y_f = \lambda y_i$$

$$C_f = C_i = 0$$

$$\frac{c\beta}{m} \sin \theta = g t \\ c \sin \theta$$

$$\dot{m} = -\beta$$

$$\text{if } \dot{m} = \text{const}$$

$$\int du = -c \dot{m} \int \frac{\cos \theta}{m} dt = -c \dot{m} \cos \theta \int \frac{m dt}{m_1 + \dot{m} t} = c \cos \theta \ln(m_1 + \dot{m} t) \Big|_0^t$$

$$m = m_1 + \dot{m} t$$

$$u = -c \cos \theta \ln(m_1 + \dot{m} t) + c \cos \theta \ln(m_1)$$

$$u = c \cos \theta \ln\left(\frac{m_1}{m_1 + \dot{m} t}\right) = c \cos \theta \ln\left(\frac{m_1}{m_1 - \beta t}\right)$$

$$\rightarrow v = \int -\frac{c \dot{m}}{m} \sin \theta dt - g t = +c \sin \theta \ln\left(\frac{m_0}{m_0 - \beta t}\right) - g t$$

$$\rightarrow x = \int u dt = \int c \cos \theta \ln\left(\frac{m_1}{m_1 - \beta t}\right) dt = [c \cos \theta \ln(m_1)] t - c \cos \theta \left[\frac{t \ln(m_1 - \beta t) - t}{\beta} - \frac{m_1 \ln(m_1 - \beta t)}{m_1} \right]$$

$$\rightarrow y = \int v dt = \int c \sin \theta \ln\left(\frac{m_1}{m_1 - \beta t}\right) dt - \frac{g t^2}{2}$$

$$y = c \sin \theta \ln m_1 t - c \sin \theta \left[t \ln(m_1 - \beta t) - t - \frac{m_1 \ln(m_1 - \beta t)}{\beta} \right] - \frac{g t^2}{2}$$

$$v_1 = c \ln\left(\frac{m_0}{m_0 - \beta T}\right) \sin \theta - g T, \quad u_1 = c \cos \theta \ln\left(\frac{m_1}{m_1 - \beta T}\right), \quad \frac{m_1}{m_1 - \beta T} = B$$

$$y_1 = c \sin \theta \left[T \ln\left(\frac{m_1}{m_1 - \beta T}\right) + 1 \right] - \frac{m_1}{\beta} \ln\left(\frac{m_1}{m_1 - \beta T}\right) - \frac{g T^2}{2}$$

$$x_1 = c \cos \theta \left[T \ln\left(\frac{m_1}{m_1 - \beta T}\right) + 1 \right] - m_1 \ln\left(\frac{m_1}{m_1 - \beta T}\right)$$

$$r = \sqrt{v_1^2 + 2gy_1} = \sqrt{c^2 \ln^2(B) \sin^2 \theta + g^2 T^2 - 2gcT \ln B + 2g C \sin \theta \left[T \ln B + 1 \right] - \frac{m_1 \ln B}{\beta}}$$

$$u_1 = c \cos \theta \ln\left(\frac{m_0}{m_0 - \beta T}\right) = c \ln B \cos \theta$$

$$\tan \theta = \sqrt{\frac{c \cos \theta}{\sin^2 \theta + (D + EA) \sin \theta}}$$

$$A = (c \ln B)$$

$$D = 2gcT$$

$$E = -2g/m_1$$

$$B = \frac{m_0}{m_1}$$

$\tan \theta$ & Chaff

where θ is acute

$$C^2 \ln^2 B \sin^2 \theta + g^2 T^2 - 2gT C \ln B \sin \theta + 2gC \sin \theta \sqrt{T} \{ \ln \theta + 1 \} + \frac{m_1}{B}$$

$$- g^2 T^2$$

$$C^2 \ln^2 B \sin^2 \theta - 2gT C \ln B \sin \theta + 2gCT \sin \theta \ln B$$

$$+ 2gCT \sin \theta + 2g \frac{C m_1}{B} \ln B \sin \theta$$

$$\tan \theta = \frac{C \ln B \sin \theta}{C \ln B \cos \theta}$$

$$\sqrt{C^2 \ln^2 B \sin^2 \theta + 2gCT \sin \theta + 2g \frac{C m_1}{B} \ln B \sin \theta}$$

$$\sqrt{\sin^2 \theta + \frac{2gT}{C \ln B} \sin \theta + \frac{2g m_1}{B} \left[\frac{1}{C \ln B} \right] \sin \theta}$$

$$A = C \ln B$$

$$D = 2gCT$$

$$E = \frac{2g m_1}{B}$$

$$\cos \theta$$

$$\sqrt{\sin^2 \theta + \left(\frac{D+E A}{A^2} \right) \sin \theta}$$

$$\frac{\sin^2 \theta}{\cos^2 \theta} = \frac{\cos^2 \theta}{(\sin^2 \theta + \left(\frac{D+E A}{A^2} \right) \sin \theta)}$$

$$\sin^4 \theta + \sin^3 \theta \left[\frac{D+E A}{A^2} \right] \sin \theta = (1 - 2 \sin^2 \theta + \sin^4 \theta)$$

$$\left[\frac{D+E A}{A^2} \right] \sin^3 \theta + 2 \sin^2 \theta - 1 = 0$$

$$\sin^3 \theta + \frac{2 A^2}{D+E A} \sin^2 \theta - \frac{A^2}{D+E A} = 0$$

$$q=0 \quad r = -\frac{A^2}{D+E A}$$

$$p = \frac{2 A^2}{D+E A}$$

$$a = -\gamma_3 \left(\frac{4 A^4}{D^2 + 2 D E A + E^2 A^2} \right)$$

$$b = \frac{1}{27}$$

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$$D = kq$$

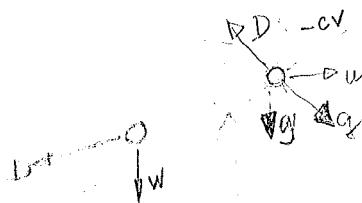


$$\ddot{u} = a \cos \theta = \frac{kq}{m} \cos \theta_0$$

$$\cos \theta_0 = \frac{u_0}{l}$$

$$\dot{\theta} = a \sin \theta = \frac{kq}{m} \sin \theta_0 - g$$

$$\sin \theta_0 = \frac{v_0}{l}$$



$$D = -kq \quad a_x = -\frac{D}{m_1} \cos \theta_0 = -\frac{kq}{m_1} \cos \theta_0$$

$$- \frac{a_0 \theta_0}{m_1} = - \frac{v_0}{l}$$

$$a_y = -\frac{D}{m_1} \sin \theta_0 - g = -\frac{kq}{m_1} \sin \theta_0 - g$$

$$- \frac{a_0 \theta_0}{m_1} = - \frac{v_0}{l}$$

$$\frac{du}{dt} = -\frac{kq}{m_1}$$

$$\frac{du}{u} = -\frac{k}{m_1} dt$$

$$\ln u = -\frac{kt}{m_1} + C \quad u = u_1 e^{-\frac{kt}{m_1}}$$

$$u_1 \left(\frac{k}{m_1} \right) e^{-\frac{kt}{m_1}} = -\frac{kq}{m_1} e^{-\frac{kt}{m_1}}$$

$$\frac{dvr}{dt} = -\frac{kq}{m_1} - g$$

$$\frac{dvr}{dt} = -kv - m_1 g$$

$$dt = \frac{m_1}{kv + m_1 g} dv$$

$$-kv - m_1 g$$

$$t = \frac{m_1}{kv + m_1 g} \int \frac{dvr}{kv + m_1 g} = \frac{m_1}{k} \ln \left[\frac{kv + m_1 g}{kv_1 + m_1 g} \right] + C$$

$$t=0 \quad v=v_1 \quad \frac{m_1}{k} \ln \left[\frac{kv_1 + m_1 g}{kv_1 + m_1 g} \right] = C$$

$$t = \frac{m_1}{k} \left[\ln \left(\frac{kv_1 + m_1 g}{kv_1 + m_1 g} \right) \right]$$

$$e^{\frac{kt}{m_1}} = \frac{kv_1 + m_1 g}{kv_1 + m_1 g}$$

$$e^{\frac{kt}{m_1}} \left(\frac{v_1 + m_1 g}{kv_1 + m_1 g} \right) = \frac{v_1 + m_1 g}{kv_1 + m_1 g}$$

$$e^{\frac{kt}{m_1}} = \frac{kv_1 + m_1 g}{kv_1 + m_1 g}$$

$$\left(\frac{v_1 + m_1 g}{kv_1 + m_1 g} \right) e^{\frac{kt}{m_1}} = \frac{m_1 g}{kv_1 + m_1 g} + v_1$$

$$\left| -\frac{m_1 g}{kv_1 + m_1 g} + \left(v_1 + \frac{m_1 g}{kv_1 + m_1 g} \right) e^{\frac{kt}{m_1}} = v \right| \checkmark$$

$$x_r = \int u_1 e^{-\frac{kt}{m_1}} dt = u_1 \left(-\frac{m_1}{k} \right) e^{-\frac{kt}{m_1}} + C \quad x_r = u_1 \frac{m_1}{k}$$

$$x_r = u_1 \left[1 - e^{-\frac{kt}{m_1}} \right] \quad \text{at } t=0$$

$$y = -\frac{m_1 g}{kv_1 + m_1 g} t - \frac{m_1}{k} \left(v_1 + \frac{m_1 g}{kv_1 + m_1 g} \right) e^{-\frac{kt}{m_1}} + \frac{m_1 \left(v_1 + m_1 g \right)}{kv_1 + m_1 g}$$

$$y_1 = -\frac{m_1}{kv_1 + m_1 g} (v_1 + m_1 g) + C$$

$$\frac{k}{m_1} = n_1$$

$$\frac{R}{m} = n_1$$

$$\text{let } t_1 = f(v_1, u_1)$$

$$dR = dv_1 + \frac{du_1}{K} m_1 \left[1 - e^{-\frac{kt_1}{m_1}} \right] + \frac{u_1 m_1}{K} \left[\left(+e^{-\frac{kt_1}{m_1}} \right) \left(+\frac{K}{m_1} dt_1 \right) \right]$$

$$0 = v_1 - \frac{m_1 g}{K} t_1 + \frac{m_1}{K} \left(v_1 + \frac{m_1 g}{K} \right) \left[1 - e^{-\frac{kt_1}{m_1}} \right]$$

$$0 = dy_1 - \frac{m_1 g}{K} dt_1 + \frac{m_1}{K} \left(dv_1 + \frac{m_1 g}{K} \right) \left[1 - e^{-\frac{kt_1}{m_1}} \right]$$

$$\left(\frac{m_1 v_1}{K} \right) + \left(\frac{m_1^2 g}{K^2} \right) \left[\quad \right]$$

$$\frac{m_1}{K} \left[\quad \right] dv_1 + \frac{m_1 v_1}{K} \left[d \quad \right] + \frac{m_1^2 g}{K^2} \left[d \quad \right]$$

$$0 = dy_1 - \frac{m_1 g}{K} dt_1 + \frac{m_1}{K} \left[1 - e^{-\frac{kt_1}{m_1}} \right] dv_1 + \frac{m_1}{K} \left[v_1 + \frac{m_1 g}{K} \right] \left[+e^{-\frac{kt_1}{m_1}} \left(+\frac{K}{m_1} dt_1 \right) \right]$$

$$\frac{m_1 g}{K} dt_1 = \left[v_1 + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}} dt_1 = dy_1 + \frac{m_1}{K} \left[1 - e^{-\frac{kt_1}{m_1}} \right] dv_1$$

$$dt_1 = \frac{dy_1 + \frac{m_1}{K} \left[1 - e^{-\frac{kt_1}{m_1}} \right] dv_1}{\frac{m_1 g}{K} - \left[v_1 + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}}}$$

$$dR = dv_1 + \frac{du_1}{K} m_1 \left[1 - e^{-\frac{kt_1}{m_1}} \right] + u_1 e^{-\frac{kt_1}{m_1}} \left[\frac{dy_1 + \frac{m_1}{K} \left[1 - e^{-\frac{kt_1}{m_1}} \right] dv_1}{\frac{m_1 g}{K} - \left[v_1 + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}}} \right]$$

$$\lambda_{x_f} = -1 \quad \lambda_{u_f} = -\frac{m_1}{K} \left[1 - e^{-\frac{kt_1}{m_1}} \right]$$

$$\lambda_{y_f} = \frac{u_1 e^{-\frac{kt_1}{m_1}}}{\left[v_1 + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}} - \frac{m_1 g}{K}}$$

$$\lambda_{v_f} = \frac{\frac{m_1 u_1}{K} e^{-\frac{kt_1}{m_1}} \left[1 - e^{-\frac{kt_1}{m_1}} \right]}{\left[v_1 + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}} - \frac{m_1 g}{K}}$$

$$\lambda_u = -\frac{m}{K+\beta} + C_{11} m^{-K/\beta}$$

$$\lambda_v = \frac{k u_1 e^{-\frac{kt_1}{m_1}}}{[kv_1 + m_1 g] e^{-\frac{kt_1}{m_1}} - m_1 g} \left(\frac{m_1}{K+\beta} \right) + C_{22} m^{-K/\beta}$$

$$\lambda_{uf} = -\frac{m_1}{K+\beta} + C_{11} m_1^{-K/\beta} = -\frac{m_1}{K} [1 - e^{-\frac{Kt_1}{m_1}}]$$

$$-\frac{m}{K+\beta} + \frac{C_{11}}{m^{-K/\beta}} = \left\{ \frac{m_1}{K+\beta} - \frac{m_1}{K} [1 - e^{-\frac{Kt_1}{m_1}}] \right\} \left(\frac{m_1}{m} \right)^{K/\beta} - \frac{m}{K+\beta}$$

$$\lambda_u = \left\{ \frac{Km_1}{(K+\beta)K} - \frac{m_1}{(K+\beta)K} [1 - e^{-\frac{Kt_1}{m_1}}] \right\} \left(\frac{m_1}{m} \right)^{K/\beta} - \frac{mK}{(K+\beta)(K)}$$

$$\lambda_{vf} = \frac{k u_1 e^{-\frac{kt_1}{m_1}}}{[kv_1 + m_1 g] e^{-\frac{kt_1}{m_1}} - m_1 g} \left(\frac{m_1}{K+\beta} \right) + C_{22} m_1^{-K/\beta} = \frac{m_1 u_1 e^{-\frac{kt_1}{m_1}} [1 - e^{-\frac{kt_1}{m_1}}]}{[kv_1 + m_1 g] e^{-\frac{kt_1}{m_1}} - m_1 g}$$

$$\frac{(K+\beta)m_1 u_1 e^{-\frac{kt_1}{m_1}} [1 - e^{-\frac{kt_1}{m_1}}]}{[kv_1 + m_1 g] e^{-\frac{kt_1}{m_1}} - m_1 g} \left(\frac{m_1}{m} \right)^{K/\beta} = \frac{C_{22}}{m^{K/\beta}}$$

$$-K^2 m_1 u_1 e^{-\frac{2kt_1}{m_1}} + (K+\beta) K m_1 u_1 e^{-\frac{2kt_1}{m_1}} - K^2 m_1 u_1 e^{-\frac{2kt_1}{m_1}} + K\beta m_1 u_1 e^{-\frac{2kt_1}{m_1}} [1 - e^{-\frac{kt_1}{m_1}}] \left(\frac{m_1}{m} \right)^{K/\beta} + \frac{km_1 u_1 e^{-\frac{kt_1}{m_1}}}{[kv_1 + m_1 g] e^{-\frac{kt_1}{m_1}} - m_1 g} \left(\frac{m_1}{m} \right)^{K/\beta} - \frac{km_1 u_1 e^{-\frac{kt_1}{m_1}}}{[kv_1 + m_1 g] e^{-\frac{kt_1}{m_1}} - m_1 g} (K+\beta)$$

$$\lambda_u = \frac{Km_1 [kv_1 + m_1 g] e^{-\frac{kt_1}{m_1}} - m_1 (K+\beta) [1 - e^{-\frac{kt_1}{m_1}}]}{[kv_1 + m_1 g] e^{-\frac{kt_1}{m_1}} - m_1 g} \left(\frac{m_1}{m} \right)^{K/\beta}$$

$$= Ku_1 e^{-\frac{kt_1}{m_1}} \left[\beta m_1 + (K+\beta)m_1 e^{-\frac{kt_1}{m_1}} \right] - \frac{mK}{(K+\beta)K}$$

$$\lambda_u = \frac{\beta m_1 + (K+\beta)m_1 e^{-\frac{kt_1}{m_1}}}{(K+\beta)K} \left(\frac{m_1}{m} \right)^{K/\beta} - \frac{mK}{(K+\beta)K}$$

$$u(x) = e^{K/\beta \ln m} \left(+ \frac{\beta c}{K} \cos \theta \ m^{-K/\beta} + S \right)$$

$$\theta = e^{K/\beta \ln m} \left(\frac{\beta c}{K} \cos \theta \ m_i^{-K/\beta} + S \right)$$

$$e^{-K/\beta \ln m_i}$$

$$m_i^{-K/\beta}$$

$$S = - \frac{\beta c}{K} \cos \theta \ m_i^{-K/\beta}$$

$$u(x) = e^{K/\beta \ln m} \left(\frac{\beta c}{K} \cos \theta [m^{-K/\beta} - m_i^{-K/\beta}] \right)$$

$$u = e^{K/\beta \ln m} \left[\frac{\beta c}{K} \cos \theta [m^{-K/\beta} - m_i^{-K/\beta}] \right]$$

$$u = m_i^{-K/\beta} e^{K/\beta \ln m} \left\{ \frac{\beta c}{K} \cos \theta \left[\left(\frac{m}{m_i} \right)^{-K/\beta} - 1 \right] \right\}$$

$$\boxed{u = e^{K/\beta \ln(m/m_i)} \left\{ \frac{\beta c}{K} \cos \theta \left[\left(\frac{m}{m_i} \right)^{-K/\beta} - 1 \right] \right\}}$$

m_0 = initial mass

$$\ddot{\theta} + \frac{K\dot{\theta}}{m} = a \sin \theta - g$$

$$u(x) = e^{+K \int \frac{dt}{m}} = e^{-K/\beta \ln m}$$

$$v = e^{K/\beta \ln m} \left[\int (a \sin \theta - g) e^{-K/\beta \ln(m_i - \beta t)} dt + c \right]$$

$$u = \left(\frac{m}{m_i} \right)^{K/\beta} \left\{ \frac{\beta c}{K} \cos \theta \left[\left(\frac{m}{m_i} \right)^{-K/\beta} - 1 \right] \right\}$$

$$\boxed{u = \frac{\beta c}{K} \cos \theta \left[1 - \left(\frac{m}{m_0} \right)^{K/\beta} \right]}$$

$$dm = \beta dt$$

$$v = e^{K/\beta \ln m} \left[c \beta \sin \theta \left\{ \frac{e^{-K/\beta \ln m}}{m} dt - g \left\{ e^{-K/\beta \ln(m_i - \beta t)} dt + c \right\} \right\} \right]$$

$$\left[-c \sin \theta \left\{ \frac{m}{m_i} m^{-(K/\beta + 1)} dm + g \frac{1}{\beta} \int m^{-(K/\beta)} dm \right\} \right]$$

$$m^{K/\beta} - c \sin \theta \left[\frac{m^{-K/\beta}}{-K/\beta} \right] + g \frac{1}{\beta} \frac{m^{-(K/\beta - 1)}}{-(K/\beta - 1)}$$

$$v = -c \beta \sin \theta \left[\right]$$

$$v = e^{\frac{c\beta}{K} \ln m} \left[\frac{c\beta}{K} \sin \theta \left(m^{-\frac{1}{K+\beta}} \right) - \frac{g}{K-\beta} (m)^{-\frac{1}{K-\beta}} + c \right]$$

$$0 = e^{\frac{c\beta}{K} \ln m_1} \left[\frac{c\beta}{K} \sin \theta \left(m_1^{-\frac{1}{K+\beta}} \right) - \frac{g}{K-\beta} (m_1)^{-\frac{1}{K-\beta}} + c \right]$$

$$c = \frac{c\beta}{K} \sin \theta (m_1^{-\frac{1}{K+\beta}}) + \frac{g}{K-\beta} (m_1^{-\frac{1}{K-\beta}})$$

$$v = e^{\frac{c\beta}{K} \ln m} \left[\frac{c\beta}{K} \sin \theta \left[m^{-\frac{1}{K+\beta}} - m_1^{-\frac{1}{K+\beta}} \right] - \frac{g}{K-\beta} (m/m_1^{\frac{1}{K+\beta}} - m_1(m_1)^{-\frac{1}{K-\beta}}) \right]$$

$$v = e^{c \ln m - \frac{g}{K-\beta} \left(\frac{c\beta}{K} \sin \theta \left[\left(\frac{m}{m_1} \right)^{-\frac{1}{K+\beta}} - 1 \right] \right)} - \frac{g}{K-\beta} \left[m \left(\frac{m}{m_1} \right)^{-\frac{1}{K-\beta}} - m_1 \right]$$

$$\left(\frac{m}{m_1} \right)^{\frac{1}{K+\beta}} \left[\frac{c\beta}{K} \sin \theta \left(1 - \left(\frac{m}{m_1} \right)^{\frac{1}{K+\beta}} \right) - \frac{g}{K-\beta} \left[m - m_1 \left(\frac{m}{m_1} \right)^{\frac{1}{K-\beta}} \right] \right]$$

m_0 initial mass

$$v = \left\{ \frac{c\beta}{K} \sin \theta \left[1 - \left(\frac{m}{m_0} \right)^{\frac{1}{K+\beta}} \right] - \frac{g}{K-\beta} \left[m - m_0 \left(\frac{m}{m_0} \right)^{\frac{1}{K-\beta}} \right] \right\}$$

$$x = \int u dt = -\frac{1}{\beta} \int u dm \quad dm = -\beta dt$$

$$= -\frac{1}{\beta} \int \left[\frac{c\beta}{K} \cos \theta \left(1 - \frac{m^{\frac{1}{K+\beta}}}{m_1^{\frac{1}{K+\beta}}} \right) \right] dm$$

$$x = -\frac{c \cos \theta}{K} \int \left[dm - \frac{m^{\frac{1}{K+\beta}} dm}{m_1^{\frac{1}{K+\beta}}} \right] \quad m^{\frac{1}{K+\beta}+1}$$

$$x = -\frac{c \cos \theta}{K} \left[m - \frac{\beta}{(K+\beta)m_1^{\frac{1}{K+\beta}}} m^{(K+\beta)+1} + C \right]$$

$$0 = m_1 - \frac{1}{K+\beta+1} m_1^{\frac{1}{K+\beta}} m_1 + C$$

$$0 = m_1 \left[1 - \frac{\beta}{K+\beta} \right] + C$$

$$C = -m_1 \left[\frac{\beta}{K+\beta} \right]$$

$$m_1 \left[\frac{\beta}{K+\beta} \right] + C = 0$$

$$x = -\frac{c \cos \theta}{K} m \left[\frac{m}{m_1} - \left(\frac{m}{m_1} \right)^{\frac{1}{K+\beta}} \frac{\beta}{K+\beta} \frac{m}{m_1} - \frac{1}{K+\beta} \left[\frac{K}{K+\beta} \right] \right]$$

$$-c \cos \theta m \left[m - \left(\frac{m}{m_1} \right)^{\frac{1}{K+\beta}+1} \frac{\beta}{K+\beta} - \frac{K}{K+\beta} \right]$$

$$x = \frac{c \cos \theta m_0}{K} \left[\frac{\kappa}{\kappa + \beta} + \left(\frac{m}{m_0} \right)^{\frac{K+\beta}{\beta}} \frac{\beta}{(\kappa + \beta)} - \frac{m}{m_0} \right]$$

$$y = \int v dt = \frac{v}{\beta} \int v dm$$

$$= -\frac{c}{K} \sin \theta \int [1 - \left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}}] dm + \frac{g}{\beta(K-\beta)} \int [m - m \left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}}] dm$$

$$y = -\frac{c}{K} \sin \theta \left[m - \frac{\beta m^{\frac{K+\beta}{\beta}}}{(K+\beta)m_1^{\frac{K+\beta}{\beta}}} \right] + \frac{g}{\beta(K-\beta)} \left[\frac{m^2}{2} - \frac{m_1 \beta}{K+\beta} m_1^{\frac{K+\beta}{\beta}} \right] + C$$

$$0 = -\frac{c}{K} \sin \theta \left[m_1 - \frac{m_1 \beta}{K+\beta} \right] + \frac{g}{\beta(K-\beta)} \left[\frac{m_1^2}{2} - \frac{m_1 \beta}{K+\beta} m_1 \right] + C$$

$$C = +\frac{c}{K} \sin \theta \left[m_1 - \frac{m_1 \beta}{K+\beta} \right] - \frac{g}{\beta(K-\beta)} \left[\frac{m_1^2}{2} - \frac{\beta}{K+\beta} m_1^2 \right]$$

$$y = -\frac{c}{K} \sin \theta \left[m - \frac{\beta}{K+\beta} \left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}} m - m_1 + \frac{m_1 \beta}{K+\beta} \right] + \frac{g}{\beta(K-\beta)} \left[\frac{m^2}{2} - \frac{\beta m_1}{K+\beta} \left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}} m - \frac{v}{2} \right. \\ \left. + \frac{\beta}{K+\beta} m_1^2 \right]$$

$$y = \frac{c}{K} \sin \theta \left[\left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}} \frac{m \beta}{K+\beta} - \frac{m_1 \beta}{K+\beta} + m_1 - m \right] + \frac{g}{\beta(K-\beta)} \left[\frac{m^2}{2} - \frac{\beta m_1}{K+\beta} \left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}} m - \frac{m_1^2}{2} \right. \\ \left. + \frac{\beta}{K+\beta} m_1^2 \right]$$

$$y = \frac{c}{K} m_1 \sin \theta \left[\left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}} \left(\frac{\beta}{K+\beta} \right) - \frac{\beta}{K+\beta} + 1 - \frac{m}{m_1} \right]$$

$$\frac{c}{K} m_1 \sin \theta \left[\left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}} \left(\frac{\beta}{K+\beta} \right) + \frac{K}{K+\beta} - \frac{m}{m_1} \right] + \frac{g m_1^2}{\beta(K-\beta)} \left[\frac{1}{2} \left(\frac{m}{m_1} \right)^2 - \left(\frac{\beta}{K+\beta} \right) \left(\frac{m}{m_1} \right)^{\frac{K+\beta}{\beta}} - \frac{1}{2} \right]$$

$$y = \frac{c}{K} m_0 \sin \theta \left[\left(\frac{m}{m_0} \right)^{\frac{K+\beta}{\beta}} \frac{\beta}{K+\beta} + \frac{K}{K+\beta} - \frac{m}{m_0} \right] + \frac{g m_0^2}{\beta(K-\beta)} \left[\frac{1}{2} \left(\frac{m}{m_0} \right)^2 - \frac{\beta}{K+\beta} \left(\frac{m}{m_0} \right)^{\frac{K+\beta}{\beta}} - \frac{K-\beta}{2(K+\beta)} \right]$$

$$-\frac{1}{2} + \frac{2\beta}{2(K+\beta)} \rightarrow K+\beta \quad \frac{2\beta + 2\beta - 2\beta}{2(K+\beta)} = \frac{2\beta}{2(K+\beta)} = \frac{(K-\beta)}{2(\beta+K)}$$

$$-K+\beta \rightarrow -K+\beta \quad -\frac{2(\beta+K)}{2(K+\beta)} = \frac{K+\beta}{2} - \frac{2\beta}{2(K+\beta)} = \frac{K-\beta}{2(\beta+K)}$$

$$v_1^2 = \left(\frac{c\beta}{K}\right)^2 \sin^2 \theta \left[1 - \left(\frac{m_1}{m_0}\right)^{\frac{K+\beta}{\beta}} \right]^2 + \left[\frac{g^2 m_0^2}{(K-\beta)^2} \left[\left(\frac{m_1}{m_0}\right) - \left(\frac{m_1}{m_0}\right)^{\frac{K+\beta}{\beta}} \right] \right]^2$$

$$= \frac{2c\beta g \sin \theta m_0}{K(K-\beta)} \left[1 - \left(\frac{m_1}{m_0}\right)^{\frac{K+\beta}{\beta}} \right] \left[\left(\frac{m_1}{m_0}\right) - \left(\frac{m_1}{m_0}\right)^{\frac{K+\beta}{\beta}} \right]$$

$$2gy_1 = \frac{2gc}{K} \sin \theta m_0 \left[\frac{K}{K+\beta} \right] + \frac{2g^2 m_0^2}{\beta(K-\beta)} \left[\frac{K}{K+\beta} \right]$$

$$\frac{2gc K \sin \theta m_0}{K(K-\beta)} \left[\frac{K}{K+\beta} \right] - \frac{2gc \beta \sin \theta m_0}{K(K-\beta)} \left[\frac{K}{K+\beta} \right]$$

$$v_1^2 = \left(\frac{c\beta}{K}\right)^2 \sin^2 \theta - 2\left(\frac{c\beta}{K} \sin \theta\right) \left(\frac{m_1}{m_0}\right)^{\frac{K+\beta}{\beta}} + \left(\frac{c\beta \sin \theta}{K} \frac{m_1}{m_0}\right)^2 \\ + \frac{g^2 m_1^2}{(K-\beta)^2} - \frac{2g^2 m_0^2}{(K-\beta)^2} \left(\frac{m_1}{m_0}\right)^{\frac{K+\beta}{\beta}} + \frac{g^2 m_0^2}{(K-\beta)^2} \left(\frac{m_1}{m_0}\right)^{\frac{2K+\beta}{\beta}}$$

$$2gy_1 = \frac{2gc \sin \theta m_0}{K+\beta} + \frac{2gc \sin \theta m_0 \beta}{K(K+\beta)} \left(\frac{m_1}{m_0}\right)^{\frac{K+\beta}{\beta}} - \frac{2gc \sin \theta m_1}{K} \\ + \frac{g^2 m_1^2}{\beta(K-\beta)} - \frac{2g^2 m_0^2}{(K-\beta)(K+\beta)} \left(\frac{m_1}{m_0}\right)^{\frac{K+\beta}{\beta}} - \frac{g^2 m_0^2}{\beta(K+\beta)}$$

$$\sqrt{v_1^2 + 2gy_1} \quad v_1 = A \sin \theta + B$$

$$y_1 = D \sin \theta + E$$

$$r = \sqrt{v_1^2 + 2gy_1}$$

$$r = \sqrt{(A^2 \sin^2 \theta + 2BA \sin \theta + B^2 + 2gD \sin \theta + 2gE)} \\ = \frac{r}{K(A \sin \theta + B) + K\sqrt{A^2 \sin^2 \theta + B^2 + 2gD \sin \theta + 2gE}}$$

$$u_1 = A \cos \theta$$

$$A' = A^2$$

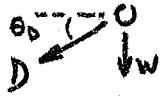
$$B' = 2BA + 2gD$$

$$C' = 2gE$$

$$\tan \theta = \frac{A \cos \theta}{\sqrt{A'^2 \sin^2 \theta + B'^2 \sin^2 \theta + C'}} e^{-m_1 g t}$$

$$\sin^2 \theta (A'^2 \sin^2 \theta + B'^2 \sin^2 \theta + C') = A \cos^2 \theta e^{-m_1 g t}$$

$$\text{where } D' = -\frac{A K}{m_1 g}, \quad E' = -\frac{B K}{m_1 g}, \quad F' = \frac{A' K^2}{m_1^2 g^2}, \quad G' = \frac{B' K^2}{m_1^2 g^2}, \quad H' = \frac{C' K^2}{m_1^2 g^2}$$



$$\dot{u} = -\frac{D}{m_r} \cos \theta_D = -\frac{ku}{m_r}$$

$$\dot{v} = -\frac{D}{m_r} \sin \theta_D = -\frac{kv}{m_r} - g$$

This leads to

$$u = u_i e^{-\frac{kt}{m_r}} = u_i e^{-n_1 t} \quad \text{where } n_1 = k/m_r$$

$$v = -\frac{m_r g}{k} + (v_i + \frac{m_r g}{k}) e^{-n_1 t}$$

$$x = \frac{u_i}{n_1} \left[1 - e^{-n_1 t} \right]$$

$$y = y_i - \frac{m_r g}{k} t + \frac{m_r}{k} \left(v_i + \frac{m_r g}{k} \right) \left[1 - e^{-\frac{kt}{m_r}} \right]$$

Assume drag is small as in the case of launching at high altitudes

$\therefore t_1$ is the same as t_1 for the ballistic trajectory for drag-free flight

$$t_1 = \frac{v_i}{g} + \frac{r}{g} = \frac{v_i + r}{g} \quad \text{where } r = \sqrt{v_i^2 + 2gy_i}$$

$$\therefore x_{\text{BALLISTIC}} = \frac{u_i}{n_1} \left[1 - e^{-n_1 \left[\frac{v_i + r}{g} \right]} \right]$$

$$F = \lambda u \left(\dot{u} - a \cos \theta + \frac{k}{m} u \right) + \lambda v \left(\dot{v} - a \sin \theta + \frac{kv}{m} + g \right) + \lambda x \left(\dot{x} - u \right) \\ + \lambda y \left(\dot{y} - v \right)$$

$$E-L \text{ in } u \quad n \lambda u - \lambda x - \lambda \dot{u} = 0 \quad E-L \text{ in } x \quad -\lambda \dot{x} = 0$$

$$E-L \text{ in } v \quad n \lambda v - \lambda y - \lambda \dot{v} = 0 \quad E-L \text{ in } y \quad -\lambda \dot{y} = 0$$

$$E-L \text{ in } \theta \quad a(\lambda u \sin \theta - \lambda v \cos \theta) = 0$$

$$\therefore C_1 = \lambda x \quad C_2 = \lambda y \quad \tan \theta = \frac{\lambda v}{\lambda u} \quad \lambda u = \frac{C_1 + C_2 e^{-nt}}{n} + C_3 \left[\frac{m}{k+r} \right] e^{-nt} + C_4$$

$$\lambda v = \frac{C_2}{n} + C_3 \left[\frac{m}{k+r} \right] e^{-nt} + C_4 \left[\frac{m}{k+r} \right] e^{-nt} + C_5$$

$$dR = dx + du, \left[\frac{1}{n} - \frac{e^{-n_1 \left[\frac{v_i + r}{g} \right]}}{n_1} \right] - \frac{u_i}{n_1} e^{-n_1 \left[\frac{v_i + r}{g} \right]} \left\{ -\frac{n_1}{g} + \frac{n_1 v_i}{gr} \right\} dv,$$

$$+ u_i e^{-n_1 \left[\frac{v_i + r}{g} \right]} \left\{ -n_1 g \right\} dy.$$

$$dR = dx_1 + du_1 \left[\frac{1 - e^{-n_1(\frac{v_1+r}{g})}}{n_1} \right] + \left[\frac{u_1 + u_1 v_1}{gr} \right] e^{-n_1(\frac{v_1+r}{g})} dv_1$$

$$+ \frac{u_1}{r} e^{-n_1(\frac{v_1+r}{g})} dy_1$$

Transversality

$$-C dt + du \left(\lambda u + \frac{1 - e^{-n_1(\frac{v_1+r}{g})}}{n_1} \right) + dv \left(\lambda v + \left\{ \frac{u_1}{g} + \frac{u_1 v_1}{gr} \right\} e^{-n_1(\frac{v_1+r}{g})} \right)$$

$$dx (\lambda x + 1) + dy \left(\lambda y + \frac{u_1}{r} e^{-n_1(\frac{v_1+r}{g})} \right) = 0$$

$$\lambda_{uf} = - \left[\frac{1}{n_1} - \frac{e^{-n_1(\frac{v_1+r}{g})}}{n_1} \right] \quad \lambda_{vf} = - \left[\frac{u_1(v_1+r)}{gr} e^{-n_1(\frac{v_1+r}{g})} \right]$$

$$\lambda_{xf} = -1 \quad \lambda_{yf} = - \frac{u_1}{r} e^{-n_1(\frac{v_1+r}{g})}$$

$$\text{Since } \lambda_x = C_1 = -1 \quad \lambda_y = C_2 = - \frac{u_1}{r} e^{-n_1(\frac{v_1+r}{g})}$$

~~$$\therefore \lambda u = -\frac{1}{n_1} + C_{11} e^{nt}; \text{ at } t=T \quad \lambda_{uf} = -\frac{1}{n_1} + C_{11} e^{n_1 T} = -\frac{1}{n_1} + \frac{e^{-n_1(\frac{v_1+r}{g})}}{n_1}$$~~

~~$$C_{11} = \frac{1}{n_1} e^{-n_1(\frac{v_1+r+gT}{g})}$$~~

~~$$\lambda u = \frac{1}{n_1} + \frac{1}{n_1} e^{nt - n_1(\frac{v_1+r+gT}{g})}$$~~

~~$$\therefore \lambda v = -\frac{u_1}{nr} e^{-n_1(\frac{v_1+r}{g})} + C_{22} e^{nt}; \text{ at } t=T \quad \lambda_{vf} = -\frac{u_1}{n_1 r} e^{-n_1(\frac{v_1+r}{g})} + C_{22} e^{n_1 T}$$~~

~~$$\text{also } \lambda_{vf} = - \left[\frac{u_1(v_1+r)}{gr} e^{-n_1(\frac{v_1+r}{g})} \right] \quad \therefore \lambda v = \frac{u_1}{gr} e^{-n_1(\frac{v_1+r}{g})} \left[\frac{g - n_1(v_1+r)}{n_1} e^{nt - n_1 T} - \frac{g}{n} \right]$$~~

$$\tan \theta = \frac{\frac{u_i e^{-\frac{n_i(v_i+r)}{g}}}{gr} \left[\frac{g - n_i(v_i+r)}{n_i} e^{nt - n_i T} - \frac{g}{n} \right]}{-\frac{1}{n} + \frac{e^{nt - n_i(v_i+r+gT)}}{n_i}}$$

note that if $m+g \gg K\beta$,

$$\lambda_{x_f} = -1 \quad \lambda_{u_f} = -\frac{1}{n_i} \left[1 - e^{-\frac{n_i(v_i+r)}{g}} \right]$$

$$\lambda_{v_f} = -\frac{u_i(r+v_i)}{gr} e^{-\frac{n_i(v_i+r)}{g}} \quad \lambda_{y_f} = -\frac{u_i}{r} e^{-\frac{n_i(v_i+r)}{g}}$$

$$\lambda_x : \lambda_x = -1 \quad \lambda_u = \frac{-m}{K+\beta} + C_{11} m^{-K/\beta}$$

$$\lambda_y = -\frac{u_i}{r} e^{-\frac{n_i(v_i+r)}{g}} \quad \lambda_v = -\frac{u_i}{r(K+\beta)} e^{-\frac{n_i(v_i+r)}{g}} m + C_{22} m^{-K/\beta}$$

$$\lambda_{v_f} = \frac{-u_i}{r(K+\beta)} e^{-\frac{n_i(v_i+r)}{g}} m_1 + C_{22} m_1^{-K/\beta} = -\frac{u_i(r+v_i)}{gr} e^{-\frac{n_i(v_i+r)}{g}}$$

$$+ \frac{u_i}{r} e^{-\frac{n_i(v_i+r)}{g}} \left[\frac{m_1}{K+\beta} - \frac{(r+v_i)}{g} \right] = C_{22}$$

$$\lambda_v = \frac{-u_i}{r(K+\beta)} e^{-\frac{n_i(v_i+r)}{g}} m + \left[\frac{u_i}{r} e^{-\frac{n_i(v_i+r)}{g}} \left\{ \frac{m_1}{K+\beta} - \frac{(r+v_i)}{g} \right\} \left(\frac{m_1}{m} \right)^{K/\beta} \right]$$

$$\lambda_v = \frac{u_i}{r} e^{-\frac{n_i(v_i+r)}{g}} \left[\frac{-m}{K+\beta} + \left\{ \frac{m_1}{K+\beta} - \frac{(r+v_i)}{g} \right\} \left(\frac{m_1}{m} \right)^{K/\beta} \right]$$

$$\lambda_u = -\frac{m}{K+\beta} + C_{11} m^{-K/\beta}$$

$$\frac{m_1 g - (K+\beta)(r+v_i)}{K(K+\beta)g}$$

$$\lambda_{u_f} = -\frac{m_1}{K+\beta} + C_{11} m_1^{-K/\beta} = -\frac{m_1}{K} \left[1 - e^{-\frac{n_i(v_i+r)}{g}} \right]$$

$$-\frac{m_1 \cancel{K}}{K(K+\beta)} + C_{11} m_1^{-K/\beta} = -\frac{m_1(K+\beta)}{K(K+\beta)} + \frac{m_1}{K}$$

$$y(x) = \frac{1}{\mu(x)} \left[\int_0^x \mu g \, ds + c \right]$$

$$u + \frac{k u}{m} = a \cos \theta$$

$$\mu(x) = e^{\int_0^x p \, dt}$$

$$p = \frac{K}{m}$$

$$k \int_0^x \frac{dt}{m} = -\frac{K}{\beta} \ln[m - \beta t]$$

$$\mu = e^{k \int_0^x \frac{dt}{m}} = e^{-\frac{K}{\beta} \ln[m - \beta t]}$$

$$y(x) = \frac{1}{e^{-\frac{K}{\beta} \ln(m - \beta t)}} \left[\int_0^x e^{-\frac{K}{\beta} \ln(m - \beta t)} a \cos \theta \, ds + c \right]$$

$$u = \frac{1}{e^{-\frac{K}{\beta} \ln(m - \beta t)}} \left[c \beta \cos \theta \int_0^x \frac{e^{-\frac{K}{\beta} \ln(m - \beta t)}}{m} \, dt + c \right]$$

$$\frac{e^{-\frac{K}{\beta} \ln(m - \beta t)}}{m-1} \frac{du}{dt} = \frac{1}{\beta} \ln(m - \beta t)$$

$$e^{-\frac{K}{\beta} \ln(m - \beta t)} \left[-\frac{1}{\beta} \ln(m - \beta t) \right] + \frac{K}{\beta} \int e^{-\frac{K}{\beta} \ln(m - \beta t)} \, dt$$

$$-\frac{1}{\beta} \int e^{-\frac{K}{\beta} \ln m} \ln m \, dm \quad dm = -\beta dt$$

$$-c \beta \cos \theta \int e^{-\frac{K}{\beta} \ln m} \, dm = -c \cos \theta \int e^{-\frac{K}{\beta} \ln m} \, dm$$

$$u(x) = e^{+\frac{K}{\beta} \ln m} \left[-c \cos \theta \int e^{-\frac{K}{\beta} \ln m} \, dm + c \right]$$

$$e^{-\frac{K}{\beta} \ln m} = \ln m^{-\frac{K}{\beta}}$$

$$e^{\ln m} = m^{-\frac{K}{\beta}}$$

$$\ln e^{\ln m} = \ln m = C K \quad m = K$$

$$\ln m^{-\frac{K}{\beta}} = -\frac{K}{\beta} \ln m$$

$$\int m^{-\frac{K}{\beta}-1} \, dm$$

$$m^{-(\frac{K}{\beta}+1)} \, dm$$

$$-\frac{\beta}{K} m^{-\frac{K}{\beta}}$$

$$\int m^{-(\frac{K}{\beta}+1)} \, dm = m^{-(\frac{K}{\beta}+1)+1} = m^{-\frac{K}{\beta}}$$

$$\frac{1}{m_1} \frac{d^2\theta}{dt^2}$$

$$= \frac{1}{m_1} \left[\frac{1}{k+D} + \frac{1}{m_0} \right] \frac{d\theta}{dt} - \frac{1}{m_1} \frac{d\theta}{dt}$$

$$= \frac{k}{m_0} \left[\left(\frac{m_0}{m_1} \right)^{\frac{1}{k+D}} + \frac{k}{k+D} \cdot \frac{1}{m_1} \right] \frac{d\theta}{dt} = D \cdot e^{-\frac{1}{k+D} \cdot \frac{1}{m_1} t} = D \cdot e^{-\frac{1}{k+D} \cdot \frac{1}{m_1} t}$$

$$A = \left[\sqrt{\frac{m_0}{m_1}} \left(\frac{m_0}{m_1} \right)^{\frac{1}{k+D}} - 1 \right] \frac{1}{m_1} = A$$

$$= \frac{k^2 g}{m_1^2} \ln \left[1 + \frac{m_1 A}{k D} \right] + \frac{B}{k D} + m_1 g \frac{1}{k D} = E$$

$$= \frac{k^2 g}{m_1^2} \ln \left[1 + \frac{m_1 A}{k D} \right] + m_1 B \frac{k}{k D} + m_1 g \frac{1}{k D} = E$$

$$e = \frac{m_1 A}{k D}$$

$$m_1 \left[1 + \frac{m_1 A}{k D} \right] = - \frac{k}{m_1} t + I_1 = - \frac{k}{m_1} \ln \left[1 + \frac{m_1 A}{k D} \right]$$

$$Q_1 = - m_1 A e = D \quad -e = \frac{K D}{m_1} = e^{\frac{K D}{m_1} t} = 1$$

$$m_1 q \dot{\theta}_1 = m_1 \left[A \sin \theta + B + m_1 g \right] e = D \sin \theta + E$$

$$T_1 = \frac{1}{2} \left[\frac{1}{m_1} + m_1 g \right] e = \frac{1}{2} m_1 q \dot{\theta}_1^2 = \frac{1}{2} m_1 q^2 \theta_1^2$$

$$V_1 = A \sin \theta + E \quad \dot{\theta}_1 = D \sin \theta + E$$

$$d\sin \theta = \frac{d\theta}{dt} \cdot \frac{e}{m_1} \quad \text{Calculation}$$

$$m_1 g \left(1 - e^{-\frac{K D}{m_1} t} \right) = V_1 e^{-\frac{K D}{m_1} t}$$

$$\tan \theta = \frac{A \cos \theta e^{-\frac{kt_1}{m_1}}}{\frac{m_1 g}{K} (1 - e^{-\frac{kt_1}{m_1}}) - (A \sin \theta + B) e^{-\frac{kt_1}{m_1}}}$$

$$e^{-\frac{kt_1}{m_1}} = \frac{e^0}{e^{\frac{kt_1}{m_1}}} = e^{-\frac{kt_1}{m_1}}$$

$$\frac{m_1 g t_1}{K} - \frac{m_1}{K} \left[A \sin \theta + B + \frac{m_1 g}{K} \right] (1 - e^{-\frac{kt_1}{m_1}}) = D \sin \theta + E$$

$$\tan \theta \left[\frac{m_1 g}{K} (1 - e^{-\frac{kt_1}{m_1}}) - (A \sin \theta + B) e^{-\frac{kt_1}{m_1}} \right] = A \cos \theta e^{-\frac{kt_1}{m_1}}$$

$$\frac{+ \left[D \sin \theta + E - \frac{m_1 g t_1}{K} \right] \frac{K}{m_1} + \frac{m_1 g}{K}}{(1 - e^{-\frac{kt_1}{m_1}})} = -(A \sin \theta + B)$$

$$\tan \theta \left[\frac{m_1 g}{K} (1 - e^{-\frac{kt_1}{m_1}}) + \frac{m_1 g}{K} e^{-\frac{kt_1}{m_1}} + \left[D \sin \theta + E - \frac{m_1 g t_1}{K} \right] \frac{K}{m_1} e^{-\frac{kt_1}{m_1}} \right] = \frac{A \cos^2 \theta}{\cos \theta} e^{-\frac{kt_1}{m_1}}$$

$$\frac{m_1 g \tan \theta}{K} = \frac{m_1 g \tan \theta}{K} e^{-\frac{kt_1}{m_1}} - A \frac{\sin^2 \theta}{\cos \theta} e^{-\frac{kt_1}{m_1}} - B \tan \theta e^{-\frac{kt_1}{m_1}} = \frac{A \cos^2 \theta}{\cos \theta} e^{-\frac{kt_1}{m_1}}$$

$$\frac{m_1 g \sin \theta}{K} = \frac{m_1 g \sin \theta}{K} e^{-\frac{kt_1}{m_1}} - B \sin \theta e^{-\frac{kt_1}{m_1}} = \frac{A \sin \theta}{\cos \theta} e^{-\frac{kt_1}{m_1}}$$

$$\sin \theta \frac{m_1 g}{K} = [A \cos \theta + \frac{m_1 g}{K} \sin \theta + B \sin \theta] e^{-\frac{kt_1}{m_1}}$$

$$\frac{m_1 g t_1}{K} - \frac{m_1}{K} \left[A \sin \theta + B + \frac{m_1 g}{K} \right] + \frac{m_1}{K} \left[A \cos \theta + B + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}} = D \sin \theta + E$$

$$\theta = \theta_0 - \frac{f(\theta)}{f'(\theta)}$$

$$\theta = g(t)$$

$$f'(\theta) = \frac{df}{dt} \frac{d\theta}{dt}$$

$$\theta = \theta_0 - \frac{f(\theta_0)}{f'(\theta_0)}$$

$$\frac{m_1 g t_1}{K} - \frac{m_1}{K} \left[B + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}} = D \sin \theta + E + \frac{m_1 A}{K} \sin \theta e^{-\frac{kt_1}{m_1}}$$

$$f'(g(t)) = \frac{df}{d\theta} \frac{d\theta}{dt}$$

$$\left\{ \begin{array}{l} \frac{m_1 g t_1}{K} - \frac{m_1}{K} \left[B + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}} - E = \sin \theta \\ (D + \frac{m_1 A}{K} e^{-\frac{kt_1}{m_1}}) \end{array} \right.$$

$$(D + \frac{m_1 A}{K} e) \left[\frac{m_1 g}{K} e - \left[B + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}} \right] = \left[\frac{m_1 g t_1}{K} - \frac{m_1}{K} \left[B + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}} \right] A e^{\frac{kt_1}{m_1}}$$

$$f(\theta(t_1)) \Rightarrow \sin \theta = \frac{\frac{m_1 g t_1}{K} - \frac{m_1}{K} \left[B + \frac{m_1 g}{K} \right] e^{-\frac{kt_1}{m_1}}}{D + \frac{m_1 A}{K} e}$$

$$1 - e = e^{-\frac{kt_1}{m_1}}, \quad \frac{d}{dt} (1 - e) = -\frac{k e}{m_1},$$

$$e = 1 - e^{-\frac{kt_1}{m_1}} = 1 - e^{-\frac{kt_1}{m_1}}$$

$$\frac{de}{dt} = \frac{k}{m_1} e^{-\frac{kt_1}{m_1}}$$

$$f(\theta(t_1)) \Rightarrow \tan \theta = \frac{A \cos \theta (1 - e)}{\frac{m_1 g}{K} e - (A \sin \theta + B)(1 - e)}$$

$$[A \cos \theta (1 - e)] / [A \cos \theta + B] = [A \sin \theta + B]$$

$$\frac{A}{A \sin \theta} \frac{\cos \theta}{1 - e} = \frac{\sin \theta}{A \sin \theta} \frac{\cos \theta}{1 - e} = \frac{\cos \theta}{\cos^2 \theta}$$

$$\frac{\cos \theta}{\cos^2 \theta}$$

$$\frac{m_1 g}{K} - \frac{m_1 g}{K} (1 - e) = [A \cos \theta + B](1 - e)$$

$$\frac{m_1 g}{K} [1 - 1 + e] = \frac{m_1 g}{K} e = [A \cos \theta + B](1 - e)$$

$$\theta(t_1) = \theta_0(t_1) - f(\theta_0(t_1)) \Rightarrow \theta_0(t_1) = \theta_0(t_1) - \frac{f(\theta_0(t_1))}{f'(\theta_0(t_1))} = \theta_0(t_1) - \frac{df}{d\theta_0} \frac{d\theta_0}{dt_1}$$

↳ f'
↳ f

$$\frac{d\theta}{dt_1} = \frac{\left((D + \frac{m_1 A}{K} e) \left[\frac{m_1 g}{K} e - B(1 - e) \right] - \left[\frac{m_1 g t_1}{K} - \frac{m_1}{K} \left[B + \frac{m_1 g}{K} \right] e - E \right] A(1 - e) \right)}{\left(D + \frac{m_1 A}{K} e \right)^2 \cos \theta}$$

$$f'(t_1) \cdot \sec^2 \theta \frac{d\theta}{dt_1} = \left\{ \begin{array}{l} \left[\frac{m_1 g}{K} e - (A \sin \theta + B)(1 - e) \right] \left[-A(1 - e) \sin \theta \frac{d\theta}{dt_1} - \frac{k}{m_1} A \cos \theta (1 - e) \right] \\ - [A \cos \theta (1 - e)] \left[-g(1 - e) + \frac{k}{m_1} (A \sin \theta + B)(1 - e) - (1 - e) A \cos \theta \frac{d\theta}{dt_1} \right] \end{array} \right\}$$

$$\left[\frac{m_1 g}{K} e - (A \sin \theta + B)(1 - e) \right]^2$$

$$t_{new} = t_0 + \frac{f(t_0)}{f'(t_0)}$$

$$t_{\text{new}} = t_0 - \frac{f(t_0)}{f'(t_0)}$$

$$\epsilon = 1 - e^{-\frac{kt_0}{m}}$$

$$f(t_0) = \tan \theta - \frac{A \cos \theta (1-\epsilon)}{\frac{m_1 g \epsilon}{K} - (A \sin \theta + B)(1-\epsilon)}$$

starting equation

$$\text{input } t \rightarrow \text{use } \text{DELTA} = \frac{m_1 g t_1}{K} - \frac{m_1}{K} [B + \frac{m_1 g}{K}] \epsilon - E \quad \text{to get } \theta \left\{ \begin{array}{l} y \\ \theta \end{array} \right\}$$

$$D = \frac{m_1 A}{K} \epsilon$$

$$\text{get } \frac{d\theta}{dt_1} = \left\{ \frac{(D + \frac{m_1 A}{K} \epsilon) [\frac{m_1 g}{K} \epsilon - B(1-\epsilon)] - [\frac{m_1 g t_1}{K} - \frac{m_1}{K} [B + \frac{m_1 g}{K}] \epsilon - E] A(1-\epsilon)}{(D + \frac{m_1 A}{K} \epsilon)^2 \cos \theta} \right\}$$

input into

$$f'(t_0) = \sec^2 \theta \frac{d\theta}{dt} + \left(\frac{\text{GAMMAA} + \text{GAMMAB}}{\left[\frac{m_1 g}{K} \epsilon - (A \sin \theta + B)(1-\epsilon) \right]^2} \right)$$

$$\text{GAMMAA} = \left[\frac{m_1 g}{K} \epsilon - (A \sin \theta + B)(1-\epsilon) \right] \left[\sin \theta \frac{d\theta}{dt} + \frac{k_1}{m_1} \cos \theta \right] A(1-\epsilon)$$

$$\text{GAMMAB} = [A \cos \theta (1-\epsilon)] \left[g + \frac{k_1}{m_1} (A \sin \theta + B) - A \cos \theta \frac{d\theta}{dt} \right] (1-\epsilon)$$

$$\theta = A \sin(\text{DELTA})$$

Space Program" Forum).

The number of pieces of paper processed by Leonard Rosenberg's office-services team loses meaning as it adds digits. A suggestion of the magnitude: Over 10,000,000 sheets of meeting papers alone were preprinted!

Managing the West Coast Headquarters in support of all national activities, and having responsibility for the Los Angeles Section and the Pacific Aerospace Library (PAL), was Joseph P. Ryan, Assistant Executive Secretary, Western Operations, assisted by Nan MacCandless, PAL Director, and by Jeff Moreau.

Fiscal 1967 confirmed the gradually changing service posture of the Pacific Aerospace Library. Once largely supported by half a dozen aerospace corporations, PAL's importance to these large aerospace companies diminished as their own technical-information centers strengthened through centralization of library facilities, mechanized operations, and the availability of services and documents from the federal government.

In 1967 the acting PAL Advisory Committee was augmented by six members from the six industries most critically involved in PAL's history. An operating and management plan was drafted to direct PAL's operations, cushion it through a period of transition, and assure its viable role in future information activities.

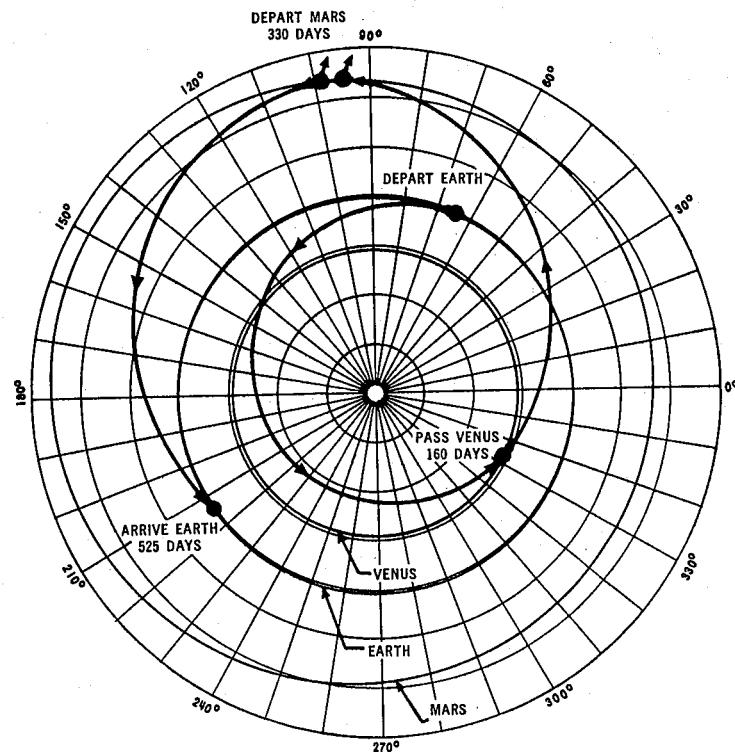
The Advisory Committee's study showed a continuing need for PAL by dozens of small aerospace subcontractors in the area, as well as confirming its limited need by the larger companies. A coupon service started late in 1966 generated more than 35 new clients for PAL's services in 1967.

Our other far-flung post, the three-man London office, was managed by Sanford M. Harris largely in support of IAA.

Keeping track of the ebb and flow of finances to cover these activities—almost \$4,000,000 in FY 1967—were AIAA Controller Joseph J. Maitan and Assistant Controller Vincent Gallo and seven others in the Accounting Department.

The cost of operating the AIAA staff is considerable—payroll alone amounting to more than \$1,620,000, or more than \$40 per member for every one of our 33,000 members and 7000 student members. For the skeptical member who wonders if it's all worth it, and is nonplussed by the fact that the payroll/member figure is twice his dues payment, it seems relevant to bring up the fact that the \$20 dues actually represents only about one-fifth of the \$100/member that AIAA put into services to the profession in FY 1967. The other \$80 comes largely from staff-administered services to aerospace firms, to NASA, DOD, and other government agencies, and to individuals, universities and other organizations in almost every country of the world.

James J. Harford
Executive Secretary



Why not Venus en route to Mars?

Is there a penalty we must pay in total trip time to get a look at Venus en route to Mars?

Or can fuel savings be realized by this mission mode?

Among the advanced interplanetary missions being investigated at Bellcomm is a Venus swingby mission to Mars. Many more interplanetary as well as lunar and earth orbital missions will be studied as part of our assignment for NASA's Office of Manned Space Flight.

And many more specialists are needed in all technical disciplines bearing on analysis

of planetary missions—flight mechanics, guidance and navigation, communications, bioastronautics, propulsion and power systems. Bellcomm also needs aeronautical and mechanical engineers broadly experienced in vehicle systems or mission planning.

If you are interested and qualified, Bellcomm offers rewarding opportunities. Send your résumé in confidence to Mr. N. W. Smusyn, Personnel Director, Bellcomm, Inc., Room 1604-N, 1100 17th St., N.W., Washington, D.C. 20036. Bellcomm is an equal opportunity employer.



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the space region in which the planet, rather than the Sun can be regarded as the center body and in which the concept of the hyperbola flight path holds with good approx.

$$r_{\text{act}} = R_{\text{pl}} \left(\frac{K_{\text{pl}}}{K_0} \right)^{1/5}$$

Heliocentric departure velocity,
transfer to outer planet.

$$V_p^* = \sqrt{\frac{2n}{n+1}} \quad n = R_A/R_p \quad V_p = \left[\frac{K}{R_p} \left(\frac{2n}{n+1} \right) \right]^{1/2}$$

transfer to inner planet.

$$V_A^* = \sqrt{\frac{2}{n+1}} = \sqrt{\frac{K_0}{R_A}} \quad n = R_A/R_p \quad K = GM$$

$$V_a = \left[\frac{K}{R_a} \left(\frac{2}{n+1} \right) \right]^{1/2}$$

Departure

$$V_p - 1 + \Delta V_1 = V_{\infty}$$

where

$$\Delta V_1 = |V_p - V_c|$$

Arrival

$$V_A - 1 - \Delta V_1 = V_{\infty}$$

where

$$\Delta V_1 = |V_A - V_c|$$

So that

Planetocentric escape velocity = V_{launch}

$$V_{\text{launch}} = \sqrt{(\Delta V_1)^2 + V_{\infty, \text{esc}}^2}$$

Or from orbit

$$\Delta V =$$

Arrive $\Delta V_2 = V_{\infty, 2} = V_p - \sqrt{K_0/a_{\text{pe}}}$ inner planet

$$\Delta V_2 = V_{\infty, 2} = \sqrt{K_0/a_{\text{pe}}} - V_a \text{ outer planet}$$

Capture energy requirement.

Circular Motion

Circle is a conic section

eccentricity = 0.0 & a = semi-major axis

$\omega = \sqrt{GM/a^3}$

period = $2\pi\sqrt{a^3/GM}$

$\propto \sqrt{a^3} \propto \sqrt{T^2}$

velocity = $\sqrt{GM/a} = \sqrt{a\omega}$

angular velocity = $\sqrt{GM/a^3}$

period = $2\pi\sqrt{a^3/GM}$

velocity = $\sqrt{GM/a} = \sqrt{a\omega}$

angular velocity = $\sqrt{GM/a^3}$

period = $2\pi\sqrt{a^3/GM}$

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velocity = $\sqrt{GM/a} = \sqrt{a\omega}$

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period = $2\pi\sqrt{a^3/GM}$

velocity = $\sqrt{GM/a} = \sqrt{a\omega}$

angular velocity = $\sqrt{GM/a^3}$

period = $2\pi\sqrt{a^3/GM}$

velocity = $\sqrt{GM/a} = \sqrt{a\omega}$

angular velocity = $\sqrt{GM/a^3}$

period = $2\pi\sqrt{a^3/GM}$

velocity = $\sqrt{GM/a} = \sqrt{a\omega}$

$V = \omega r$ - angular speed is ω

Acc. is independent of mass

$V \uparrow \rightarrow T \uparrow \rightarrow K.E. \uparrow$

W to E \rightarrow Prograde Orbit

E to W \rightarrow Retrograde Orbit

Inertia \propto mass $\propto a^2$ from rev. law

Orbital vel. \propto both body masses

Time zones = 24 \rightarrow TZ = 15° long.

Prime vertical is L to horizon circle

Ecliptical plane - earth rev. about sun plane [ecliptic axis minor]

Equatorial " - earth equator projected.

flat sun second = $a = 6.8183$ in equatorial plane

= 1.0973 in ecliptical plane

= 0.9968 in equatorial plane projected

flat earth sun second = $a = 0.9113$ in flat sun second

= 0.62137 in equatorial plane

= 0.8777 in ecliptical plane projected

flat sun year, addition = 3.2808 in flat sun second

= 2.2369 in equatorial plane

= 3.600 in ecliptical plane projected

flat sun year, addition = 0.874784 in equatorial plane second

= 0.6093 in ecliptical plane second

$g \sim I_2(\text{rotational})$, ellipticity & distribution of mass

THE DOPPLER EFFECT

Suppose that a source of waves is moving directly toward the observer from a fixed position with a velocity v , and let t be the time between the emission of two successive waves. Then, during this time, the source has moved $v \cdot t$ more by a distance $v \cdot t$. If c is the velocity of wave propagation (velocity of light) the second wave takes a time $t + v/v - t$ to reach the observer. Hence, the time between successive waves reaches up the observer and increases by a factor of $\frac{c+v}{c}$. The time between successive waves is inversely related to the frequency of the waves; hence, the frequency of the waves, remaining the fixed observer from receiving a wave, will be decreased by a factor of $\frac{c+v}{c}$ if the emitted frequency. In this case of an observer moving toward the observer, the frequency would be increased by the same factor.

If to the frequency of the incident wave f_0 we add v/c the source, the frequency changes noted by the observer would be (the Doppler frequency shift) is equal to

$f_0 + \frac{v}{c}$

where

Source

f_0
 $f_0 + \frac{v}{c}$

$$I_t = F t_b \quad F_i = \dot{m}_i v_i = \frac{\dot{m}_i}{g} \bar{v}_i \quad I_{sp_i} = \frac{F_i t_{bi}}{\dot{m}_i}$$

$$\bar{v}_i = I_{sp_i} g \quad F_i = \dot{m}_i I_{sp_i}$$

$$v_f - v_0 = \sum_i \bar{v}_i \ln \frac{M_{0i}}{M_{bi}} \quad M_{0i} = M_{bi} + \dot{m}_i t_{bi}$$

$$\text{at } t=0 \quad x=y=u=v=0; \quad m=m_0$$

$$\text{at } t=0 \quad y=0; \quad m=m_1$$

$$T = -cm = c\beta$$

$$x=u$$

$$y=v$$

$$u = \frac{c\beta}{m} \cos\theta$$

$$v = \frac{c\beta}{m} \sin\theta - g$$

$$\dot{m} = -\beta$$

$$0 \leq \beta \leq \beta_{max}$$

$$F = \lambda_x(x-u) + \lambda_y(y-v) + \lambda_u(u - \frac{c\beta}{m} \cos\theta)$$

$$+ \lambda_v(v - \frac{c\beta}{m} \sin\theta + g) + \lambda_m(\dot{m} + \beta)$$

$$+ \lambda_\gamma(\beta[\beta_{max} - \beta] - \gamma^2) = 0$$

~~x, y, u, v, β , m, θ , γ~~
 ~~$\lambda_x, \lambda_y, \lambda_u, \lambda_v, \lambda_m, \lambda_\gamma$~~

$$\beta(\beta_{max} - \beta) - \gamma^2 = 0$$

$$x, y, u, v, \beta, m, \theta, \gamma$$

$$\frac{\partial F}{\partial x} - \frac{d}{dt} \left(\frac{\partial F}{\partial \dot{x}} \right) = -\lambda \ddot{x} = 0 \quad \lambda x = c_1 \quad \text{by rank.}$$

$$\frac{\partial F}{\partial y} - \frac{d}{dt} \left(\frac{\partial F}{\partial \dot{y}} \right) = -\lambda \ddot{y} = 0 \quad \lambda y = c_2$$

$$\frac{\partial F}{\partial u} - \frac{d}{dt} \left(\frac{\partial F}{\partial \dot{u}} \right) = -\lambda_x - \lambda \ddot{u} = 0 \quad c_1 + \lambda \ddot{u} = 0 \quad c_1 t + \lambda u = c_{11}$$

$$\frac{\partial F}{\partial v} - \frac{d}{dt} \left(\frac{\partial F}{\partial \dot{v}} \right) = -\lambda_y - \lambda \ddot{v} = 0 \quad c_2 + \lambda \ddot{v} = 0 \quad c_2 t + \lambda v = c_{22}$$

$$\frac{\partial F}{\partial \beta} - \frac{d}{dt} \left(\frac{\partial F}{\partial \dot{\beta}} \right) = -\frac{c}{m} \cos\theta \cdot \lambda_u - \frac{c}{m} \sin\theta \lambda_v + [\beta_{max} - 2\beta] \lambda_\gamma + \lambda_m = 0$$

$$\frac{\partial F}{\partial m} - \frac{d}{dt} \left(\frac{\partial F}{\partial \dot{m}} \right) = \lambda_u \frac{c\beta}{m} \cos\theta + \lambda_v \frac{c\beta}{m} \sin\theta - \lambda \dot{m} = 0$$

$$\frac{\partial F}{\partial \theta} - \frac{d}{dt} \left(\frac{\partial F}{\partial \dot{\theta}} \right) = \lambda_u \frac{c\beta}{m} \sin\theta + \lambda_v \frac{c\beta}{m} \cos\theta = 0$$

$$\frac{\partial F}{\partial \gamma} - \frac{d}{dt} \left(\frac{\partial F}{\partial \dot{\gamma}} \right) = -2\gamma \lambda_\gamma = 0 \quad \gamma = 0 \quad \lambda \gamma = 0$$

$$\begin{cases} \frac{\lambda v}{\lambda u} = \tan\theta \\ \frac{\lambda u}{\lambda v} = \frac{c_{22} - c_2 t}{c_1 - c_1 t} \end{cases}$$

$$L = \lambda_x u + \lambda_y v + \lambda_u \left[\frac{c\beta}{m} \cos\theta \right] + \lambda_v \left[\frac{c\beta}{m} \sin\theta - g \right] - \lambda_m \beta + \dots$$

$$\frac{\partial L}{\partial \theta} = -\lambda_u \frac{c\beta}{m} \sin\theta + \lambda_v \frac{c\beta}{m} \cos\theta = 0 \quad \tan\theta = \frac{\lambda_v}{\lambda_u} \quad \begin{matrix} \lambda_u \sin\theta = \lambda_v \cos\theta \\ \lambda_u \cos\theta = \lambda_v \sin\theta \end{matrix}$$

$$\frac{\partial^2 L}{\partial \theta^2} \geq 0 \quad -\lambda_u \frac{c\beta}{m} \cos\theta - \lambda_v \frac{c\beta}{m} \sin\theta \geq 0 \quad \text{only if } -\text{Axis of Agent}$$

$$+ \frac{c\beta}{m} (\lambda_u^2 + \lambda_v^2)$$

$$-Cdt \left| \int_{t_0}^{t_f} (\lambda_x dx + \lambda_y dy + \lambda_v dv + \lambda_w dw + \lambda_m dm) - d\gamma_f \right.$$

$$\gamma_f = 0 \quad \lambda_{x_f} + 1 = 0 \quad \lambda_x = +1 \quad dy_f = 0 \quad dy_i = 0$$

$$\lambda_{v_f} = 0 \quad \lambda_{w_f} = 0 \quad dm_f = 0 \quad dm_i = 0$$

$$\tan \theta = -(u/v) \quad \lambda_x \dot{x} + \lambda_y \dot{y} + \lambda_v \dot{v} + \lambda_w \dot{w} + \lambda_m \dot{m} = 0$$

$$\text{now } \lambda_{u_f} = 0 = C_{11} - C_1 T = C_{11} + T \quad C_{11} = -T$$

$$\therefore \lambda_u = -T - t = - (T + t)$$

$$\cos \theta = \frac{-1}{\sqrt{1+C_2^2}}$$

$$\lambda_{v_f} = C_{22} - C_2 T = 0 \quad C_{22} = C_2 T$$

$$\sin \theta = \frac{C_2}{\sqrt{1+C_2^2}}$$

$$\lambda_v = C_2(T-t)$$

$$\lambda_v / \lambda_w = \tan \theta = +C_2, \quad \lambda_w = -C_2 \cos \theta$$

$$\text{if } \gamma = 0 \quad \beta = \beta_{\max} \text{ or } \beta = 0$$

$$m = -C_2$$

$$\lambda_u \frac{c \cos \theta}{m} + C_2 \lambda_w \frac{c \beta \sin \theta}{m} \leq 0$$

$$\lambda_u \frac{c \beta \cos \theta}{m} - C_2^2 \frac{\lambda_w c \beta \sin \theta}{m} \leq 0$$

$$\frac{\partial L}{\partial \beta} = \lambda_u \frac{c}{m} \cos \theta + \lambda_v \frac{c}{m} \sin \theta - \lambda_m = K = 0 \quad \frac{c \beta (T-t)}{m} [\cos \theta + C_2^2 \cos \theta] \leq 0$$

$$\frac{c}{m} [\sqrt{\lambda_u^2 + \lambda_v^2}] = \lambda_m \quad \frac{c (T-t)}{m} \sqrt{1+C_2^2} = \lambda_m$$

$$L = fK$$

only if $\lambda \neq$

$$\frac{\partial L}{\partial \theta} = \frac{\partial L}{\partial \theta} d\theta + \frac{\partial L}{\partial \beta} d\beta$$

$$\frac{\partial L}{\partial \theta} < 0$$

$$\frac{\partial L}{\partial \beta} = 0$$

~~$$\lambda_u \frac{c}{m} \sin \theta + \lambda_v \frac{c}{m} \cos \theta$$~~

$$\lambda v = -C_2 \lambda u$$

~~$$\lambda_u \frac{c}{m} [\sin \theta + C_2 \cos \theta]$$~~

$$\lambda v = \sin \theta = -C_2$$

~~$$+ (T-t) \frac{c}{m} [0] = 0$$~~

$$\frac{c \beta}{m^2} \left[\lambda_u \frac{c \cos \theta}{m} + \lambda_v \frac{c \sin \theta}{m} \right] = \lambda \dot{m} = 0$$

$f_{xx} f_{yy} > f_{xy}^2$ no max exists

$$\frac{c \beta}{m^2} (T-t) \sqrt{1+C_2^2} - \lambda \dot{m} = 0$$

$$\lambda_m = \int_0^T \frac{c \beta}{m^2} (T-t) \sqrt{1+C_2^2} dt$$

$$-\left[\lambda_u \frac{c}{m} \cos \theta + \lambda_v \frac{c}{m} \sin \theta \right] = -\lambda \dot{m} \left[\frac{m}{\beta} \right]$$

$$\sqrt{1+C_2^2} c \int \frac{m}{m^2} (T-t) dt$$

$$-\lambda \dot{m} \frac{m}{\beta} + [\beta_{\max} - 2\beta] \lambda \gamma + \lambda m = 0$$

f

$$\frac{c}{r_1 + c_1^2} = \frac{m_0}{m_1} \left[\frac{2 + c_1^2}{c_1^2} \right]$$

and let $\frac{m_0}{m_1}$

$$x_f = \frac{c_1 \cos \theta}{m_1}$$

$$u_x = \frac{c_1 \cos \theta}{m_1} \frac{1 - \frac{m_0}{m_1}}{1 - \frac{c_1 \cos \theta}{m_1}}$$

$$\dot{x} = u$$

Euler lag

$$\dot{x} = 0$$

$$\dot{y} = v$$

$$\dot{u} = \frac{c\beta}{m} \cos \theta$$

$$\dot{v} = \frac{c\beta}{m} \sin \theta - g$$

$$\dot{m} = -\beta$$

$$\beta(\beta_{\max} - \beta) - \delta^2 = 0$$

$$\dot{x} = 0$$

$$\dot{y} = 0$$

$$\lambda x + \lambda w = 0$$

$$\lambda y + \lambda v = 0$$

$$K_B = \frac{c}{m} [\lambda u \cos \theta + \lambda v \sin \theta] - \lambda m; K_B = \lambda v [\beta_{\max} - 2\beta] = 0$$

$$\frac{c\beta}{m^2} [\lambda u \cos \theta + \lambda v \sin \theta] - \lambda m = 0$$

$$\frac{c\beta}{m} [\lambda u \sin \theta - \lambda v \cos \theta] = 0$$

$$-2\gamma \lambda \beta = 0$$

This leads to $\lambda x = c_1$, $\lambda y = c_2$, $\lambda u = c_{11} - c_{1t}$, $\lambda v = c_{22} - c_{2t}$

$$\gamma = 0 \text{ or } \lambda \gamma = 0 \quad \tan \theta = \frac{\lambda v}{\lambda u} = \frac{c_{22} - c_{2t}}{c_{11} - c_{1t}} = \frac{c_2(t-t)}{c_1(t-t)}$$

Also at t=0 $x=y=u=v=0$, $m=m_0$

$$@ \text{ end} \quad y=0 \quad m=m_1$$

First integral $C = \lambda x \dot{x} + \lambda y \dot{y} + \lambda u \dot{u} + \lambda v \dot{v} + \lambda m \dot{m}$

$$\text{trans. } -cdt \int_{t_0}^{t_f} + \lambda x dx + \lambda y dy + \lambda u du + \lambda v dv + \lambda m dm + dxf = 0$$

$$\text{or } C_f = 0 = C \quad \lambda x = +1 \quad \lambda u_f = 0 \quad \lambda v_f = 0 \quad d_{yf} = dy_i = 0$$

OK

$$\text{Weierstrass } \lambda u \frac{c\beta \cos \theta}{m} + \lambda v \left(\frac{c\beta}{m} \sin \theta - g \right) + \lambda x u + \lambda y v - \lambda m \beta \leq$$

$$\lambda u \frac{c\beta^* \cos \theta^*}{m} + \lambda v \left(\frac{c\beta^*}{m} \sin \theta^* - g \right) + \lambda x u + \lambda y v - \lambda m \beta^*$$

$$\lambda u \frac{c\beta}{m} \cos \theta + \lambda v \frac{c\beta}{m} \sin \theta - \lambda m \beta \leq \lambda u \frac{c\beta^*}{m} \cos \theta^* + \lambda v \left(\frac{c\beta^*}{m} \sin \theta^* \right) - \lambda m \beta^*$$

$$\text{Over a null thrust arc } \beta = 0 \quad K = \lambda m - \frac{c}{m} [\lambda u \cos \theta + \lambda v \sin \theta] \quad K \leq 0$$

$$\kappa \beta \geq K^* \beta^* \quad \text{only if } \lambda m \leq \frac{c}{m} \lambda_i l_i^* = \frac{c}{m} [\lambda u^2 + \lambda v^2]^{\frac{1}{2}}$$

$$\text{Over an informed } 0 < \beta \leq \beta_{\max} \quad \text{only poss if } \lambda m \geq \frac{c}{m} [\lambda u^2 + \lambda v^2]^{\frac{1}{2}}$$

$$\text{Over a max } \beta = \beta_{\max} \quad \text{only if } \lambda m \geq \frac{c}{m} [\lambda u^2 + \lambda v^2]^{\frac{1}{2}} \quad K \geq 0$$

Since λ 's are already defined

$$K \geq 0$$

at final first integral $\lambda x = -1$ $\lambda v_f = 0$ $\lambda u_f = 0$

$$0 = \lambda y \dot{y} - \lambda m \beta \quad \text{if we turn then}$$

$$\lambda m = \frac{c}{m} [\lambda u^2 + \lambda v^2]^{1/2} = 0 \Rightarrow \lambda y = 0 \text{ or } \dot{y} = 0$$

but if $\neq 0$ $\lambda y = 0 \Rightarrow c_2 = 0 \Rightarrow$ no constraints would exist on y

this is a contradiction to $y = 0$ at end of flight from transversality

if max thrust $\lambda m \geq 0$ $\dot{y} = \frac{\lambda m \beta}{\lambda y} \Rightarrow \lambda y > 0$

~~$\dot{y} \geq 0$ but this is not possible since~~

~~this is not possible since integration for λm shows $\lambda m = c \left[\frac{(T-t)}{m} + \int_0^T \frac{dt}{m} \right]^{1/2}$~~

~~and $\int_0^T \frac{dt}{m}$~~ $\dot{y} = \frac{\lambda m \beta}{\lambda y} \quad \text{if } \lambda y > 0, \tan \theta < 0 \quad \theta \text{ in 2nd q}$

$\dot{y} > 0, y > 0$ this is not possible.

∴ final arc must be null thrust arc. Since rocket velocity is initially velocity must be positive

$$0 = -\dot{x}_f + \lambda y \dot{y}_f - \lambda m \beta \quad \therefore \lambda y \dot{y} = \dot{x} \quad c_2 = (\dot{x}/\dot{y})$$

$$\tan \theta = \frac{\dot{y}}{\dot{x}} = -\frac{\dot{x}}{\dot{y}} \quad \dot{y} < 0$$

~~90° if $\dot{x}/\dot{y} \geq 0$ this is impossible~~

Max thrust arc $0 = +\dot{x} - \lambda y \dot{y}, = -\lambda m \beta_{max} \quad \dot{x} - \lambda y \dot{y} < 0$

if $\lambda \beta = 0$ $K_p = 0$ $K_\beta = 0$ but $K_p = -\frac{c}{m} (\lambda x \cos \theta + \lambda y \sin \theta)$

for $K_p = 0$ then $\lambda x^2 + \lambda y^2 = 0$ this is not possible $\therefore \lambda \neq 0$

$\therefore \beta = 0$ for β_{max}, β

from first integral $-\dot{x}_f + \lambda y \dot{y}_f - \lambda m \beta = 0 \quad \text{if } \beta = \beta_{max}, \lambda m \geq \frac{c}{m} [\lambda u^2 + \lambda v^2]$

$$|\lambda y \dot{y}_f - \dot{x}_f| = +\lambda m \beta_{max} \leq 0 \quad \lambda y = \frac{\dot{x}_f - \lambda m \beta_{max}}{\dot{y}_f}$$

Assume $\lambda y > 0$

$$u = \int \frac{c \beta}{m} \cos \theta dt = \frac{c}{m} \int_0^T \frac{du}{m} = \frac{c}{m} \int_0^T \frac{1}{1 + C_2^2} \left[\ln \frac{m_1}{m_0} \right] dt = \frac{c}{m} \frac{1}{1 + C_2^2} \ln \frac{m_1}{m_0} \quad \dot{x}_f < 0$$

$$\dot{y}_f = \dot{x}_f - \lambda m \beta_{max}$$

$$\lambda y = \dot{x}_f - \lambda m \beta_{max}$$

$$\beta_{\max} \Delta m_f = \frac{c}{\sqrt{1+c_2^2}} \ln \frac{m_0}{m_1} \sqrt{1+c_2^2}$$

$$y_f = \frac{c}{\sqrt{1+c_2^2}} \ln \frac{m_1}{m_0} + c \ln \frac{m_1}{m_0} \sqrt{1+c_2^2}$$

$$= \frac{c}{\sqrt{1+c_2^2}} \ln \frac{m_1}{m_0} \left[1 + \frac{c^2 + c_2^2}{c_2^2} \right]$$

if $c_2 > 0$ $y_f < 0$ but $c_2 > 0$ is the case $y_f = \frac{c}{\sqrt{1+c_2^2}} \ln \frac{m_1}{m_0} \left[\frac{c^2 + c_2^2}{c_2^2} \right]$

\Rightarrow physically impossible to max. range [$m \gg T$]

if $c_2 < 0$ $y_f > 0$ again physically impossible if $y > 0$ for impact to occur.

i.e. last phase reached by coasting.

At junction pt m is discontin. i.e. $R=0$ $c_u = c_v$ $x_u = x_v$

i.e. integration of eq for Δm leads to

$$m = m_0 - \beta_{\max} t \quad 0 \leq t \leq t_1 \quad t_1 \text{ is burnout}$$

$$m = m_1 \quad t_1 \leq t \leq T$$

$$\Delta m = -c \sqrt{1+c_2^2} \left[\frac{T-t}{m} + \frac{1}{\beta_{\max}} \ln \frac{m}{m_1} \right] \quad 0 \leq t \leq t_1$$

$$\Delta m = -c \sqrt{1+c_2^2} \left[\frac{T-t_1}{m_1} \right] \quad t_1 \leq t \leq T$$

$$\Delta x = -1 \quad \text{everywhere} \quad t_1 = \frac{m_0 - m_1}{\beta_{\max}}$$

$$\Delta y = c_2$$

$$\Delta u = - (T-t) \quad 0 \leq t \leq T$$

$$\Delta v = c_2 (T-t)$$

$$\gamma = 0$$

$$\tan \theta = -c_2$$

$$\beta = \beta_{\max} \quad 0 \leq t \leq t_1$$

$$\beta = 0 \quad t_1 \leq t \leq T$$

$$\Delta s = -\frac{c \sqrt{1+c_2^2}}{\beta_{\max}} \ln \frac{m}{m_1} \quad 0 \leq t \leq t_1$$

$$\Delta s = c \sqrt{1+c_2^2} \left[\frac{T-t_1}{m_1} \right] \quad t_1 \leq t \leq T$$

$$\ddot{v} = \frac{c\beta}{m} \sin \theta - g$$

$$v = c \sin \theta \ln \frac{m_0}{m} - gt \quad 0 \leq t \leq t_1$$

$$v = c \sin \theta \ln \frac{m_0}{m_1} - gt \quad t_1 \leq t \leq T$$

$$\ddot{u} = \frac{c\beta}{m} \cos \theta$$

$$u = c \cos \theta \ln \frac{m_0}{m} \quad 0 \leq t \leq t_1$$

$$u = c \cos \theta \ln \frac{m_0}{m_1} \quad t_1 \leq t \leq T$$

$$\dot{y} = v$$

$$y = \frac{c \sin \theta}{\beta_{\max}} \left\{ m \ln \frac{m}{m_0} + \beta t \right\} - gt^2 \quad 0 \leq t \leq t_1$$

$$= \frac{c \sin \theta}{\beta_{\max}} \left\{ M \ln \frac{m_1}{m_0} + \beta_{\max} t_1 \right\} - gt^2 \quad t_1 \leq t \leq T$$

$$M = m_0 + \beta_{\max} t_1$$

$$\dot{x} = u$$

$$x = \frac{c \cos \theta}{\beta_{\max}} \left\{ m \ln \frac{m}{m_0} + \beta t \right\} \quad 0 \leq t \leq t_1$$

$$= \frac{c \cos \theta}{\beta_{\max}} \left\{ M \ln \frac{m_1}{m_0} + \beta t_1 \right\} \quad t_1 \leq t \leq T$$

$$\text{if BRAVO} = c \sin \theta + \ln \frac{m_1}{m_0} \quad \text{CHARLY} = - \frac{c \sin \theta}{\beta_{\max}} \left[m_0 \ln \frac{m_1}{m_0} + \beta_{\max} t_1 \right]$$

$$\text{ALPHA} = \alpha/2$$

$$\text{then } T = \frac{-\text{BRAVO} + \sqrt{(\text{BRAVO})^2 - 4 * \text{ALPHA} * \text{CHARLY}}}{2 * \text{ALPHA}}$$

$$\text{from 1st Integral } C_2 = \frac{\dot{x}_f}{\dot{y}_f} = \frac{c \cos \theta \ln \frac{m_0}{m_1}}{c \sin \theta \ln \frac{m_0}{m_1} - gt} = \frac{c \cos \theta \ln \frac{m_0}{m_1}}{-\gamma (\text{BRAVO})^2 - 4 * \text{ALPHA} * \text{CHARLY}}$$

$$\cos \theta = \frac{C_2}{\sqrt{1+C_2^2}}$$

$$\sin \theta = \frac{1}{\sqrt{1+C_2^2}}$$

$$\sin \theta = \frac{c_1}{\sqrt{1+c_2^2}}$$

$$\sin \theta^* = \frac{c_2}{\sqrt{1+c_2^2}}$$

$$-gt + c \sin \theta^* \ln \frac{m_1}{m_0} = \sqrt{c^2 \sin^2 \theta^* \ln \frac{m_1}{m_0} + \frac{2gc \sin \theta^*}{\beta_{max}} \left(m_0 \ln \frac{m_1}{m_0} + \beta_{max} t_1 \right)}$$

$$C_2 = 0$$

$$C_2 \dot{y}_f - 1 \dot{x}_f = 0 \quad C_2 = \frac{\dot{x}_f}{\dot{y}_f} = \frac{c \cos \theta^* \ln \frac{m_1}{m_0}}{c \sin \theta^* \ln \frac{m_1}{m_0} - gt}$$

※

$$\tan \theta = C_2 = - \frac{\dot{y}_f}{\dot{x}_f}$$

$$-C_2 =$$

$$y = \frac{c \sin \theta}{\beta_{max}} \left[m_0 \ln \frac{m_1}{m_0} - \beta_{max} t \ln \frac{m_1}{m_0} + \beta_{max} t_1 \right] - gt^2/2$$

$$\frac{c \sin \theta}{\beta_{max}} \left[m_0 \ln \frac{m_1}{m_0} - t \ln \frac{m_1}{m_0} + t_1 \right] - gt^2/2$$

$$gt^2/2 + c \sin \theta \ln \frac{m_1}{m_0} t - c \sin \theta \left[\frac{m_0}{\beta_{max}} \ln \frac{m_1}{m_0} + t_1 \right] = 0$$

$$T = c \sin \theta \ln \frac{m_1}{m_0} + \sqrt{c^2 \left(\ln \frac{m_1}{m_0} \right)^2 \sin^2 \theta + \frac{2gc \sin \theta}{\beta_{max}} \left[m_0 \ln \frac{m_1}{m_0} + \beta_{max} t_1 \right]}$$

g

$$C_2 = \frac{\dot{x}_f}{\dot{y}_f} = \frac{c \cos \theta \ln \frac{m_1}{m_0}}{c \sin \theta \ln \frac{m_1}{m_0} - c \sin \theta \ln \frac{m_1}{m_0} - \sqrt{c^2 \left(\ln \frac{m_1}{m_0} \right)^2 \sin^2 \theta + \frac{2gc \sin \theta}{\beta_{max}} \left[m_0 \ln \frac{m_1}{m_0} + \beta_{max} t_1 \right]}}$$

$$\tan^2 \theta = \frac{c^2 \cos^2 \theta \left(\ln \frac{m_1}{m_0} \right)^2}{c^2 \left(\ln \frac{m_1}{m_0} \right)^2 \sin^2 \theta + \frac{2gc \sin \theta}{\beta_{max}} \left[m_0 \ln \frac{m_1}{m_0} + \beta_{max} t_1 \right]}$$

$$\sin^2 \theta \left\{ c \ln B \right\}^2 + \frac{2gc \sin^3 \theta}{\beta_{max}} \left\{ -m_0 \ln \frac{m_1}{m_0} + t_1 \right\} = \left\{ c \ln B \right\}^2 \cos^4 \theta$$

$$\text{call } B = \frac{m_0}{m_1}, \quad D = 2gct_1$$

$$A = c \ln B$$

$$\therefore A^2 \sin^4 \theta + \sin^3 \theta \{ AE + D \} = A^2 (1 - 2 \sin^2 \theta + \sin^4 \theta)$$

$$E = -\frac{2gma}{\beta_{max}}$$

$$\sin^3 \theta + \frac{2A^2}{D+E A} \sin^2 \theta - \frac{A^2}{D+E A} = 0 \quad \checkmark \checkmark$$

DIMENSION U(100), V(150), X(150), Y(150), MA(100), ULAM(100), VLAM(100)

100 READ(S,I) : , C, MZERO, CAPT, BETA, G, EPSI

REAL MA, MZERO

1 FORMAT

$$B = MZERO / (MZERO - BETA * CAPT)$$

$$D = 2 * G * C * CAPT$$

$$E = 2 * G * MZERO / BETA * (-1,00) \quad \text{change}$$

$$\text{THETA} = 30$$

$$\text{OMEGA} = \text{THETA} * (4 * \text{ATAN}(1.00)) / 180.$$

$$10 FTHETA = (\sin(\text{OMEGA})) ** 3 + ((2 * A ** 2) / (D + E * A)) * (\sin(\text{OMEGA})) ** 2 - ((A ** 2) / (D + E * A))$$

$$DF = (1.5 * \sin(\text{OMEGA}) + (2 * A ** 2) / (D + E * A)) * \sin(2 * \text{OMEGA})$$

$$\text{OMA} = \text{OMEGA} - FTHETA / DF$$

IF (ABS(DF), LE, EPSI) STOP

IF (ABS(OMA - OMEGA) - EPSI) 20, 20, 15

$$15 \text{ OMEGA} = \text{OMA}$$

GO TO 10

①

$$20 \text{ THETA} = \text{OMA} * 180. / (4 * \text{ATAN}(1.00)) \quad \begin{matrix} T=0 \\ I=1 \end{matrix}$$

30

$$AA = C * \text{ALOG}(MZERO / (MZERO - BETA * T))$$

$$BB = C * T * (\text{ALOG}(MZERO / (MZERO - BETA * T)) + 1)$$

$$CC = (C * MZERO / BETA) * \text{ALOG}(MZERO / (MZERO - BETA * T))$$

$$U(I) = AA * \cos(\text{OMA}) \quad \begin{matrix} MA(I) = MZERO - BETA * T \\ \text{MA} \end{matrix}$$

$$V(I) = AA * \sin(\text{OMA}) - G * T$$

$$Y(I) = ((BB - CC) * \sin(\text{OMA}) - G * T * T / 2) \quad \text{change}$$

$$X(I) = (BB - CC) * \cos(\text{OMA})$$

$$ULAM(I) = (T - CAPT) = (V1 + R) / G$$

$$VLAM(I) = (V1 / R) * ULAM(I)$$

IF (T, EQ, CAPT) GO TO 40

$$T = T + 10$$

(1)

$$AA = C * ALOG(MZERO / (MZERO - BETA + CAPT))$$

$$BB = C * CAPT * (ALOG(MZERO / (MZERO - BETA + CAPT)) + 1)$$

$$CC = (C * MZERO / BETA) * ALOG(MZERO / (MZERO - BETA + CAPT))$$

$$V1 = AA * \cos(OHA)$$

$$V1 = AA * \sin(OHA) - G + CAPT$$

$$Y1 = (BB + CC) * \sin(OHA) - G + CAPT + Z/2 \rightarrow \text{change?}$$

$$X1 = (BB + CC) * \cos(OHA)$$

$$R = \sqrt{V1^2 + Z^2 + G^2 + Y1^2}$$

$$FLTT = CAPT + (V1 + R) / G$$

$$N = \text{IFIX}(FLTT / 10 + 1)$$

40 GO TO 30

40 M = I

50 T = T + 10

I = I + 1

60

$$V(I) = V1 - G * (T - CAPT)$$

$$X(I) = X1 + V1 * (T - CAPT)$$

$$Y(I) = Y1 + V1 * (T - CAPT) - 0.5 * G * (T - CAPT) * Z$$

IF (I, LE, (N-2)) GO TO 50

IF (I, EQ, N) GO TO 70

T = FLTT

I = I + 1

GO TO 60

70 WRITE(6,75) :, MZERO, BETA, C, CAPT, FLTT, XLAM, YLAM, THETA

75 FORMAT(4(SX, E15.7)/4(SX, E15.7))

DO 80 I = 1, M

80 WRITE(6,85) I, VLAM(I), VLAM(I), V(I), V(I), X(I), Y(I), MA(I)

85 FORMAT(SX, IS, 7(2X, E14.7))

M = M + 1

DO 90 I = M, N

90 WRITE(6,95) I, V(I), X(I), Y(I)

95 FORMAT(SX, IS, 57X, 3(SX, E14.7))

GOTO 100

B

DIMENSION U(100), V(100), X(100), Y(100), MA(100), VLAM(100), VLAM(100)

REAL KA, K, MZERO

100 READ (5,1) LS, C, WZERO, CAPT, WBETA, G, EPSI

1 FORMAT

$$.88 * MZERO / BETA = CAPT$$

$$MZERO = WZERO / G$$

$$BETA = WBETA / G$$

$$NONE = MZERO - BETA + CAPT$$

$$A = (BETA + C / K) * \left(1.0 - \frac{MONE}{(MZERO - BETA + CAPT) / MZERO} \right)^{** (K / BETA)}$$

$$B = G / (BETA - K) * \left(\frac{MONE}{(MZERO - BETA + CAPT) / MZERO} - MZERO + \frac{MONE}{(MZERO - BETA + CAPT) / MZERO} \right)^{** (K / BETA)}$$

$$D = (C * MZERO / K) * \left(\frac{K / (K + BETA)}{1.0} + \beta / (K + \beta) * \left(\frac{MONE}{(MZERO - BETA + CAPT) / MZERO} \right)^{** (K / BETA)} \right. \\ \left. - \frac{MONE}{(MZERO - BETA + CAPT) / MZERO} \right)$$

$$E = G * MZERO ** 2 / (BETA + (K - BETA)) * \left(0.5 * \left(\frac{MONE}{MZERO} \right)^{** 2} - BETA / (K + BETA) * \left(\frac{MONE}{MZERO} \right) \right. \\ \left. * * (1 + K / BETA) - (K - BETA) / (2.0 * (BETA + K)) \right)$$

$$T = 10.00$$

$$10 \quad ETA = 1.0 - EXP(-K * T / MONE)$$

$$DELTA = (MONE * G + T - MONE / K + (B + MONE * G / K) * ETA - E) / (D + MONE * A * ETA / K)$$

$$OMA = ASIN(DELTA)$$

$$PTO = TAN(OMA) = A * \cos(OMA) * (1 - ETA) / (MONE * G + ETA / K - (A * \sin(OMA) \\ + B) * (1 - ETA))$$

$$DTDHT = ((D + MONE * A * ETA / K) * (MONE * G * ETA / K - B * (1 - ETA)) - (MONE * G * T / K) \\ - MONE / K * (B + MONE * G / K) * ETA - E) * A * (1 - ETA) / ((D + MONE * A * ETA / K) \\ * * 2 * \cos(OMA))$$

$$GAMMAA = \left(MONE * G * ETA / K - (A * \sin(OMA) + B) * (1 - ETA) \right) * \left(\sin(\frac{OMA}{\pi}) * DTHDT \right. \\ \left. + K / MONE * \cos(\frac{OMA}{\pi}) \right) * A * (1 - ETA)$$

$$GAMMAB = (A * \cos(\frac{OMA}{\pi}) * (1 - \epsilon)) * (G + K / MONE * (A * \sin(\frac{OMA}{\pi}) + B) - \\ A * \cos(\frac{OMA}{\pi}) * DTHDT) * (1 - \epsilon)$$

$$DFTO = SEC(\frac{OMA}{\pi}) * * 2 * DTHDT + (GAMMAA + GAMMAB) / (MONE * G * ETA / K -$$

$$(A * \sin(\frac{\text{OMA}}{180}) + B) * (1 - e^{\beta A})$$

$$T_{\text{NEW}} = T - FTO / DFTO$$

IF (ABS(DFTO), LE, EPS1) STOP

IF (ABS(T - T_{\text{NEW}}) - EPS1) 20, 20, 15

15 T = T_{\text{NEW}}

GO TO 10

$$20 \quad \text{THETA} = \text{OMA} + 180. / (4. + \text{ATAN}(1.00))$$

~~$$U_1 = A * \cos(\text{OMA})$$~~

$$FLTT = CAPT + T$$

~~$$V_1 = A * \sin(\text{OMA}) + B$$~~

$$N = \text{IFIX}(FLTT / 10. + 1)$$

~~$$X_1 = D * \cos(\text{OMA})$$~~

$$T = 0$$

~~$$Y_1 =$$~~

$$\begin{aligned} I &= 1 \\ V_1 &= A * \cos(\text{OMA}) \\ Y_1 &= A * \sin(\text{OMA}) + B \end{aligned}$$

$$30 \quad MONG = MZERO - BETAT * T$$

$$\text{recopy } A, B, D, E$$

$$U(I) = A * \cos(\text{OMA})$$

$$V(I) = A * \sin(\text{OMA}) + B$$

$$X(I) = D * \cos(\text{OMA})$$

$$Y(I) = D * \sin(\text{OMA}) + E$$

$$VLAM(I) =$$

$$MA(I) = MZERO - BETAT * V_{\text{LAM}}(I)$$

IF (T, EQ, CAPT) GO TO 40

$$T = T + 10.$$

$$I = I + 1$$

GO TO 30

$$\begin{aligned} X_u &= \left\{ \frac{Km_1 - m_1 K - m_1 \beta + (K + \beta)m_1 e^{-\frac{Kt_1}{m_1}}}{(K + \beta)K} \right\} \left(\frac{m_1}{m} \right)^{\frac{K}{\beta}} - \frac{mK}{K(K + \beta)} \\ &\quad - \frac{1}{(K + \beta)K} \left\{ [m_1 \beta - (K + \beta)m_1 e^{-\frac{Kt_1}{m_1}}] \left(\frac{m_1}{m} \right)^{\frac{K}{\beta}} + mK \right\} \end{aligned}$$

~~At~~ $\frac{du}{dt} = \text{constant}$ $\frac{du}{dt} = \frac{R u}{m}$

$$P = m u t + C \quad \text{or} \quad P = \frac{R u}{m} t + C$$

Suppose that at burnout (i.e. at $t=T$), $x=x_1, y=y_1, u=u_1, v=v_1$. Present perfect may measure t from burnout. We consider this place. Then $m = \frac{m_1}{e^{-\frac{Rt}{m_1}} - 1} \quad (\text{const.})$

~~At~~ $\frac{dv}{dt} =$

$$\frac{dv + m_1 g}{P} = \left(\frac{Rv + m_1 g}{k} \right) e^{-\frac{Rt}{m_1}}$$

$$\text{also } x_1 + \frac{m_1 u_1}{R}$$

$$\frac{dy}{dt} = u_1 e^{-\frac{Rt}{m_1}}$$

$$x = x_1 + u_1 \frac{m_1}{R} \left[1 - e^{-\frac{Rt}{m_1}} \right]$$

(Refer S. 96,
t measured from T)

$$\text{th. } \left(1 - e^{-\frac{Rt_1}{m_1}} \right) \frac{m_1}{k} \left(v_1 + \frac{m_1 g}{k} \right) = \frac{m_1 g t_1 - y_1}{P}$$

$$1 - e^{-\frac{Rt_1}{m_1}} = \frac{\left(\frac{m_1 g t_1 - y_1}{P} \right) \frac{k}{m_1}}{\frac{v_1 + m_1 g}{k}}$$

There is no B to divide out this
affx. 1.

If $\frac{k}{m_1}$ is negligible: $\frac{v_1 + m_1 g}{k}$

then we have $\frac{k}{m_1}$ negligible $\frac{g t_1 - y_1}{P} \frac{k_1}{m_1}$

so we have $\frac{k}{m_1}$ negligible $\frac{g t_1 - y_1}{P} \frac{k_1}{m_1}$

It may be best to suffice by small drag, so let's can off.

Perhaps even assume that t_1 is unaffected by drag.

Looking at S. 9 & p. 146, eq. 6.227, this implies

$$\frac{mg}{R} \gg \frac{k}{m_1} v_1 \quad \text{or } mg \gg k v_1$$

i.e. drag < rot.
Momentum lost per v_1
not very large.

Still, can agree that this is O.K. for suff. small R .

After all, we are looking for effect of drag. So it's been all that's left
but do this. Then t_1 is as in orig. p. 1 of problem (no drag)

$$x_R = x_1 + \frac{v_1 m_1}{R} \left\{ 1 - \left(1 - \frac{kt_1}{m_1} + \dots \right) \right\} = \text{as before.}$$

Up to a first approx., x_R -offset is as before.

$$\frac{1}{n} = \frac{m}{R} \quad mg \gg kv_1$$

$$e^{kv_1 m/R}$$

$$\frac{m_1 g - k(v_1 + r)}{m_1} e^{nt-niT} - \frac{mg}{R}$$

$$m_1 g \quad g e^{nt-niT}$$

(1)

From Bell (Aero. Quart. Aug. 1865)
or Lardner (1865)

Project missile launched with v_0 initial vel at $t=0$; burnout occurs at a known instant $t=T$. Assume const. acceleration, f caused by the motor thrust is const. Find thrust dist. & maximize total range of rocket.

Sol. $\dot{q}_1 = \ddot{u} - f \cos \theta = 0$; $\dot{q}_2 = \dot{v} - f \sin \theta + g = 0$
 $\dot{q}_3 = \dot{x} - u = 0$; $\dot{q}_4 = \dot{y} - v = 0$.

(where $f = -c \frac{d.m}{m.t}$)

End cond: $x_0 = y_0 = u_0 = v_0 = t_0 = 0$; $t_1 = T$.

To maximize $x = x_1 + \frac{u_1}{g} [\sqrt{v_1^2 + v_1^2 + 2gy_1}]$.

Write $F = \lambda_u (\dot{u} - f \cos \theta) + \lambda_v (\dot{v} - f \sin \theta + g) + \lambda_x (\dot{x} - u)$
 $+ \lambda_y (\dot{y} - v)$. Variables are u, v, x, y, θ .

Euler-Lag. eqns for $F \Rightarrow$,

$$\dot{\lambda}_u = -\lambda_x; \quad \dot{\lambda}_v = -\lambda_y; \quad \dot{\lambda}_x = 0; \quad \dot{\lambda}_y = 0;$$

$$\lambda_u f \sin \theta - \lambda_v f \cos \theta = 0.$$

Transversality cond $\Rightarrow (\sin \theta |_{t_0} = dt_1 |_{t_0} = 0)$,

$$[\lambda_u du + \lambda_v dv + \lambda_x dx + \lambda_y dy]_0^T + dx \left[\frac{du}{g} \right]_0^T + \left(\frac{du}{g} \right) \left[\sqrt{v^2 + v^2 + 2gy_1} \right]_0^T + \frac{u_1}{g} [dv + \frac{1}{r} (v_1 dv + g dy_1)]_0^T = 0.$$

This $\Rightarrow (\lambda_u)_1 = -\frac{(v_1 + r)}{g}; \quad (\lambda_v)_1 = -\frac{u_1(v_1 + r)}{gr}$ (ans'd)

$$(\dot{x}_x)_1 = -1, (\lambda_y)_1 = -\frac{u_1}{r}, \text{ where } r \equiv \sqrt{v_1^2 + 2gy_1} \quad (2)$$

Sol. 2) Euler-Lag. Eqn: $\lambda_u = at + b, \lambda_v = a't + b'$,

$$\lambda_x = -a, \lambda_y = -a', \tan \theta = \frac{a't + b'}{at + b}.$$

From the transversality cond: $a = 1, a' = \frac{u_1}{r}, b = -k,$

$$b' = -\frac{u_1 k}{r}, \text{ where } k = T + \frac{(v_1 + r)}{g}. \text{ whence}$$

find $\tan \theta = \frac{u_1}{r} \therefore \theta = \text{const. throughout.}$

Bill does second variation to imply this indeed gives max. range.
Leverrier does that Weierstrass cond $\Rightarrow \theta$ acute & pos. ($\because \theta$ must by extrema).

Values of a and b are not specified in advance. To obtain explicit solution: $u_1 = f T \cos \theta; v_1 = (f \sin \theta - g)T$

$$r = (f \sin \theta - g)t = \dot{y} \therefore y_1 = \frac{(f \sin \theta - g)T^2}{2}$$

$$r = T \sqrt{(f \sin \theta - g)^2 + g(f \sin \theta - g)}$$

$$\therefore \tan \theta = \frac{u_1}{T \sqrt{}}; \tan^2 \theta = \frac{f \cos^2 \theta}{(f \sin \theta - g) \sin \theta}$$

Set $f \sin \theta - g$ ~~(contd)~~ must require $\tan \theta = \frac{u_1}{r}$, thereby find θ .
 (contd)

get $f(1 - \sin^2 \theta)^2 = \sin^3 \theta (f \sin \theta - g)$, or (3)
 $f(1 - 2 \sin^2 \theta + \sin^4 \theta) = f \sin^4 \theta - g \sin^3 \theta$, or
 $f(-2 \sin^2 \theta) + g \sin^3 \theta = 0$, or
 $\sin^3 \theta - 2\left(\frac{f}{g}\right) \sin^2 \theta + \left(\frac{f}{g}\right) = 0.$

$\left|\frac{f}{g}\right| \gg 1$, then approximately, $\sin^2 \theta = \frac{1}{2}$, $\theta \approx 45^\circ$.

$$x = \frac{1}{2} f t^2 \cos \theta, \quad y = \frac{1}{2} (f \sin \theta - g) t^2.$$

The opt. trajectory is a str. line at angle ϕ to horizontal.

$$\text{where } \tan \phi = \tan \theta - \frac{g}{f} \sec \theta.$$

(If f and g are const. the max. range will be $f T^2 \left(\frac{f}{g} \cot \theta - \frac{1}{2} \sec^2 \theta \right)$
 (using $r = \frac{u_1}{\tan \theta}$).

I Inclusion intro.: Basically, an elementary theory.
At least 3 types of procedures for analysis of
optimization problems: (a) Calc. of Var. (b) Adjoint Eqs.
(c) Pontryagin Max. Princ. You will emphasize (a) & (b).
Might give some idea of what has been done by others:
e.g. Larden, Bell, Leitmann, Miele, Fossler, Bliss,
Hoffman, Thomson, Pontryagin.

II Calc of Var.: Problem of Bolza, Mayer & Legendre, with
emphasis on Mayer Prob.

Euler-Lag. Eqs., transversality condition,
Corner condition, Legendre-Clebsch, Weierstrass
conds. Try to get clear idea of what these do (e.g., nec.
suff., etc.) & how they can be derived.

III A Problem with unbounded controls.

(a) "Bell's" problem. ~~seems without drag~~.
(b) Problem (a) with drag. Assume $D = \sqrt{v}$ if
helps. Might then still be able to use Cartesian coords.
Noted that if in this problem (as we are assuming) $f = \text{const.}$, then
 $\frac{dm}{m} = -\frac{f}{c} dt$, where $m = m_0 e^{-\frac{f}{c} t}$.

(May see Syznev Griffiths for Dr. v; pp. 141-142).

~~(It is feasible, but in (a), assume a different~~

III A Problem with bounded controls

(a) Milne one-dim problem p. 128

(b) Range problem. Milne p. 141

IV

Method I adapt Eqs.

Show this especially for
~~I illustrate with~~ the max. range problem, i.e.

max. $x(T)$ p. 4(1) Faulkner (Is this same as
Milne range problem?). Would be desirable, if same
to show that get same answers for the optimal control.