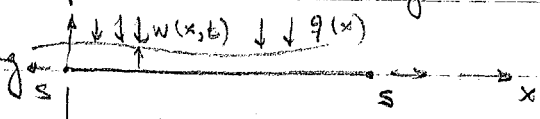


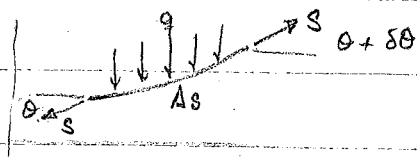
2-6-74

Wave Propagation - Stoker

Wave - unsteady & time dependent motion in a continuous media.

Perfect incompressible irrotational gravitating fluid.

We look at string  Perfectly Flexible $w(x,t)$ is small



$$\frac{\Delta s}{\Delta x} \sim \sqrt{1 + \left(\frac{\Delta w}{\Delta x}\right)^2} \sim 1$$

$$S \sin(\theta + \delta\theta) - S \sin \theta - q \Delta s = \rho \Delta s \frac{\partial^2 w}{\partial t^2}$$

$$\Delta s \approx \Delta x$$

$$S \theta + \delta\theta - S \theta - q \Delta x = \rho \Delta x \frac{\partial^2 w}{\partial t^2}$$

$$S \delta\theta - q \Delta x = \rho w_{tt}$$

$$\theta \sim \tan \theta = \frac{\partial w}{\partial x}$$

$$\therefore S w_{xx} + q = \rho w_{tt}$$

look at $\rho = \text{const}$ $q = 0$ with $S/\rho = c^2$

$$\therefore c^2 w_{xx} - w_{tt} = 0$$

$w = f(x-ct) + g(x+ct)$ is solution

wave: $t \rightarrow t + \Delta t$ $x \rightarrow c \Delta t + x \rightarrow$ motion is translation of curve to right with speed c .

I.V.P. $t=0$ $w(x,0) = F(x)$

Continuity is needed for $F(x)$

$w_t(x,0) = G(x)$

Continuity & piecewise cont deriv

d'Alembert Solution

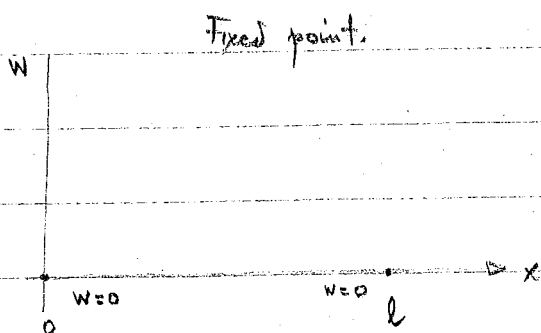
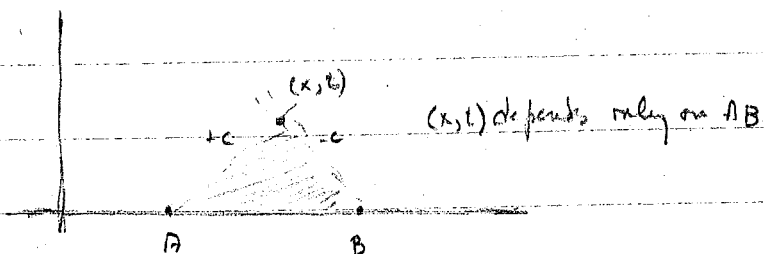
$$\left. \begin{aligned} F(x) &= f(x) + g(x) \\ G(x) &= -c(f' - g') \end{aligned} \right\} \begin{aligned} f(x) - g(x) &= -\frac{1}{c} \int^x G(s) ds \end{aligned}$$

$$2f = F - \frac{1}{c} \int^x G(s) ds$$

$$2g = F + \frac{1}{c} \int^x G(s) ds$$

$$w(x,t) = \frac{1}{2} [f(x-ct) + g(x+ct)] + \frac{1}{2c} \int_{x-ct}^{x+ct} G(s) ds$$

Problem



look for sol of form $w(x,t) = X(x)T(t)$

Standing waves & separation of variables.

$$c^2 X'' = -\frac{1}{T} T' = -v^2$$

$$c^2 \frac{X''}{X} = -\frac{1}{T} T' = -v^2 \Rightarrow T'' + v^2 T = 0 \quad \& \quad X'' + \frac{v^2}{c^2} X = 0$$

$$\left[\cos vt, \sin vt \right] \quad \left[\sin \frac{v}{c} x, \cos \frac{v}{c} x \right]$$

$w(x,t)$ is assumed

$T(0), T'(0)$ must be given.

These give rise to EV problems.

$$X'' + \frac{v^2}{c^2} X = 0 \quad X(0) = X(l) = 0 \Rightarrow X^* = 0 \text{ trivial sol.}$$

$$X = A \cos \frac{v}{c} x + B \sin \frac{v}{c} x \quad X(0) = 0 \Rightarrow X = B \sin \frac{v}{c} x$$

$$0 = A \sin \frac{v}{c} l \quad \& \quad \frac{v}{c} = n\pi l$$

$$\therefore w(x,t) = B_n \sin \frac{n\pi x}{l}$$

$$T = C_n \cos \frac{n\pi c t}{l} + D_n \sin \frac{n\pi c t}{l}$$

$$w = \sum_n \left\{ C_n \cos \frac{n\pi c t}{l} + D_n \sin \frac{n\pi c t}{l} \right\} \sin \frac{n\pi x}{l}$$

Impose I.C. $w(t=0) = -F(x)$

$$w_t(t=0) = G(x)$$

$$f(x) = a_0 + \sum_{n=1}^{\infty} (a_n \cos nx + b_n \sin nx)$$

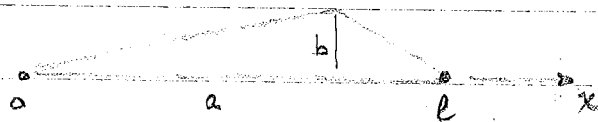
$$a_0 = \frac{1}{2\pi} \int_{-\pi}^{\pi} f(x) dx$$

$$a_n = \frac{1}{\pi} \int_{-\pi}^{\pi} f(x) \cos nx dx$$

$$b_n = \frac{1}{\pi} \int_{-\pi}^{\pi} f(x) \sin nx dx$$

$f(x)$ is piecewise continuous & piecewise differentiable

Vibration of Harp String



$$f(x) = \begin{cases} \frac{b}{a}x & 0 \leq x \leq a \\ \frac{b}{(a-l)}(x-l) & a \leq x \leq l \end{cases}$$

$$w_t = 0 \text{ for } t=0 \Rightarrow G(x) = 0 \therefore w(x,t) = \frac{1}{2} \{F(x+ct) + F(x-ct)\}$$

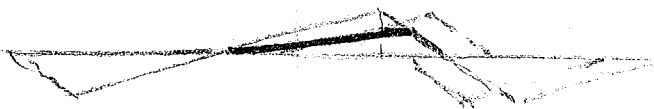
$$w|_{t=0} = f(x)$$

$$w(x,t) = \sum_{n=1}^{\infty} C_n \cos \frac{n\pi c t}{l} \sin \frac{n\pi x}{l}$$

$$w(x,0) = \sum_{n=1}^{\infty} C_n \sin \frac{n\pi x}{l} = f(x)$$

standing wave $w = \frac{2bl^2}{\pi^2 a(l-a)} \sum_{n=1}^{\infty} \frac{1}{n^2} \sin \frac{n\pi a}{l} \sin \frac{n\pi x}{l} \cos \frac{n\pi c t}{l}$

progressive wave sol. $= \frac{bl^2}{\pi^2 a(l-a)} \sum_{n=1}^{\infty} \frac{1}{n^2} \sin \frac{n\pi a}{l} \left[\sin \frac{n\pi}{l}(x-ct) + \sin \frac{n\pi}{l}(x+ct) \right]$



$$e^{imx} e^{i\omega t} = e^{i(mx \pm \omega t)}$$

$$\operatorname{Re}(e^{i(mx \pm \omega t)}) = \cos(mx \pm \omega t) = \cos m(x \pm \frac{\omega}{m}t)$$

if t fixed wavelength $\lambda = \frac{2\pi}{m}$

prop speed $= \pm \omega/m$

m is wave number

ω - circular freq

$$\text{freq} = \frac{\omega}{2\pi}$$

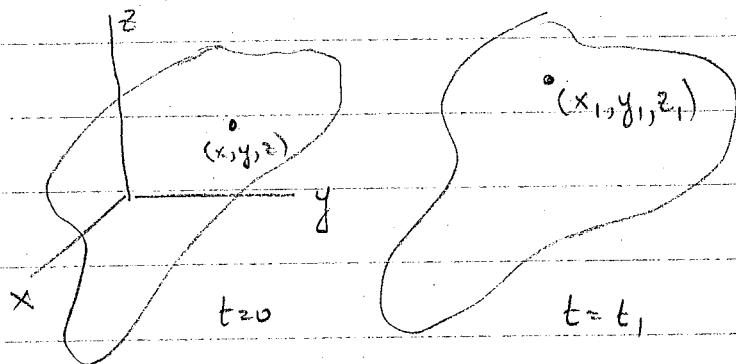
$\nabla \cdot \underline{u} = 0$ incompressible

$\iff$

$\rho = \text{const}$ for incompressible fluids

$\nabla \times \underline{u} = 0$ irrotational

Continuum Mechanics

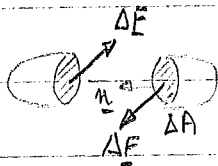
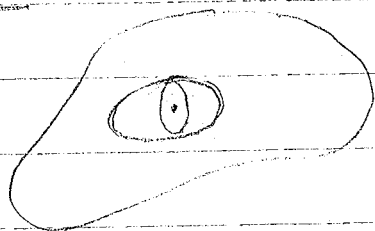


Assume topological mapping 1-1, continuous, differentiable to high order

Jacobian $\neq 0$

Assume curves remain curves in this mapping interior $\rightarrow$ interior
body $\rightarrow$ body

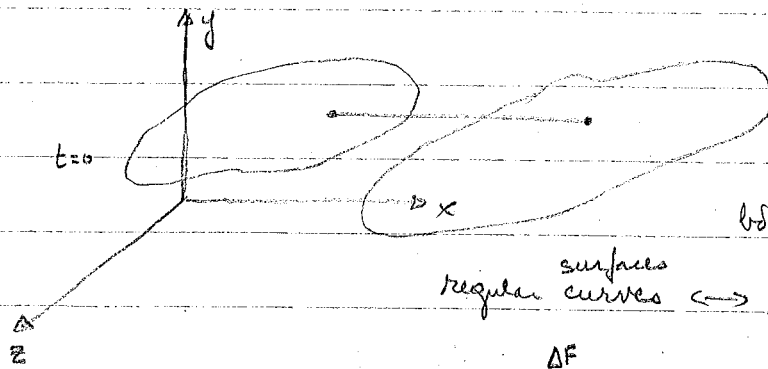
Stress



$$\lim_{\Delta A \rightarrow 0} \frac{\Delta F}{\Delta A} = \underline{\sigma}(x_1, y_1, z_1, t; \underline{n}) \quad \text{smooth, continuous}$$

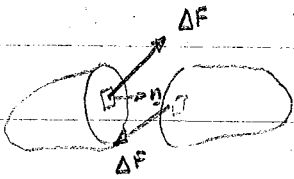
Perfect fluid cannot support shear $\underline{\sigma} = p \delta_{ij} \quad p(x_1, y_1, z_1, t)$

2-13-72

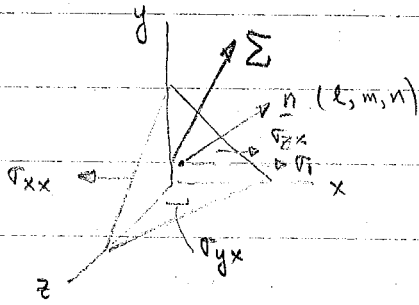


1-1 mapping inverse exists & differentiable
Jacobian $\partial/\partial > 0$

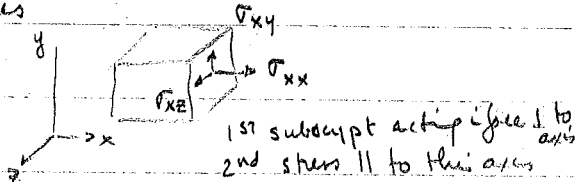
bdy pts $\leftrightarrow$ bdy pts interior pts $\leftrightarrow$ interior pts
surfaces $\leftrightarrow$ surfaces
regular curves $\leftrightarrow$ regular curves



$$\sigma_{av} = \frac{\Delta F}{\Delta A} ; \quad \sigma = (x, y, z, t, n)$$

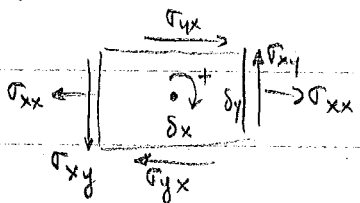


$\begin{pmatrix} l \\ m \\ n \end{pmatrix}$ direction cosines
notation :



Claim: symmetric tensor

$$\begin{pmatrix} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{yx} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{zx} & \sigma_{zy} & \sigma_{zz} \end{pmatrix} = \sigma_{\text{tensor}}$$



$$\sigma_{yx} \delta x \delta z \frac{\delta y}{2} + \sigma_{yx} \delta x \delta z \frac{\delta y}{2} - \sigma_{xy} \delta z \delta y \frac{\delta x}{2} - \sigma_{xy} \delta z \delta y \frac{\delta x}{2} = 0$$

$$\Rightarrow I \propto O(\delta^5) \quad \therefore \sigma_{yx} = \sigma_{xy}$$

$$\sigma_1 \Delta A = l \Delta A \sigma_{xx} + m \Delta A \sigma_{yx} + n \Delta A \sigma_{zx} + \rho \int \delta x \delta y \delta z = \rho \int V a(x)$$

divide by ΔA & let tetrahedron shrink to origin since

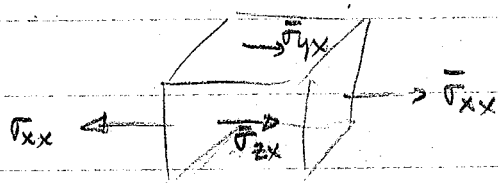
$$\sigma_1 = \sigma_{xx} l + \sigma_{yx} m + \sigma_{zx} n$$

$$\Sigma = (\sigma)^T \begin{pmatrix} l \\ m \\ n \end{pmatrix} \quad \exists \text{ principal axis } \Rightarrow \quad \sigma = \begin{pmatrix} \sigma_{xx} & 0 & 0 \\ 0 & \sigma_{yy} & 0 \\ 0 & 0 & \sigma_{zz} \end{pmatrix}$$

$\Rightarrow$ all shears are zero

in a perfect fluid no shear exists & all directions are principal axes hence σ is described by a single scalar called pressure

Equation of motion in $x \rightarrow$



$$\bar{F} = (x, y, z)$$

$\bar{F}$ denotes $\sigma_{xx} + \frac{\partial \sigma_{xx}}{\partial x} dx$

$$(\bar{\sigma}_{xx} - \sigma_{xx}) \delta y \delta z + (\bar{\sigma}_{xy} - \sigma_{xy}) \delta z \delta x + (\bar{\sigma}_{xz} - \sigma_{xz}) \delta y \delta x + \rho X \delta x \delta y \delta z = \rho \delta x \delta y \delta z a(x)$$

to first order

$$\frac{\partial \sigma_{xx}}{\partial x} \delta x \delta y \delta z + \frac{\partial \sigma_{xy}}{\partial y} \delta x \delta y \delta z + \frac{\partial \sigma_{xz}}{\partial z} \delta x \delta y \delta z + \rho X \delta x \delta y \delta z = \rho \delta x \delta y \delta z a(x)$$

$$\sigma_{xx,x} + \sigma_{xy,y} + \sigma_{xz,z} + \rho X = \rho a(x)$$

since we assume shears are zero & $\sigma_{xx} = -p$

$$-\frac{\partial p}{\partial x} + \rho X = \rho a(x)$$

$$-\frac{\partial p}{\partial y} + \rho Y = \rho a(y)$$

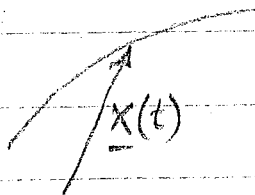
$$-\frac{\partial p}{\partial z} + \rho Z = \rho a(z)$$

deformed position & differentiation is done with respect to deformed position } Lagrange form.
 x, y, z are dependent variables since we really don't know the deformed position

$$-\frac{1}{\rho} \nabla p + \bar{F} = \underline{a}$$

In deformed position .

$F(x, y, z, t)$ what is $\frac{dF}{dt}$ along some curve
 $x = x(t) \quad y = y(t) \quad z = z(t)$



$$\frac{dF}{dt} = \frac{\partial F}{\partial t} + \frac{\partial F}{\partial x} \frac{dx}{dt} + \frac{\partial F}{\partial y} \frac{dy}{dt} + \frac{\partial F}{\partial z} \frac{dz}{dt}$$

$$\frac{d\mathbf{x}}{dt} = (\dot{x}, \dot{y}, \dot{z}) = (u, v, w) \quad \frac{dF}{dt} = \left(\frac{\partial}{\partial t} + \underline{q} \cdot \nabla \right) F$$

if $\mathbf{x} = (x(t), y(t), z(t))$

$$\underline{q} = \frac{d\mathbf{x}}{dt} = (\dot{x}(t), \dot{y}(t), \dot{z}(t)) = (u, v, w)$$

Euler fix & watch one pt in space

$$\underline{a} = \frac{d\underline{q}}{dt}$$

$$a_x = \frac{du}{dt} = u_t + u u_x + v u_y + w u_z$$

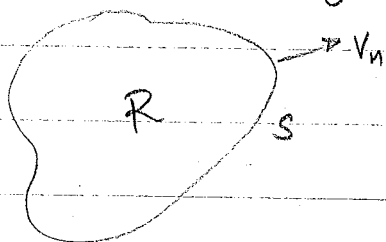
$$u_t + u u_x + v u_y + w u_z = X - \frac{1}{\rho} P_x$$

$$v_t + u v_x + v v_y + w v_z = Y - \frac{1}{\rho} P_y$$

$$w_t + u w_x + v w_y + w w_z = Z - \frac{1}{\rho} P_z$$

} Conservation of Momentum

Assume incompressible flow $\Rightarrow \Delta \text{volume} = 0$; $\rho = \text{const}$

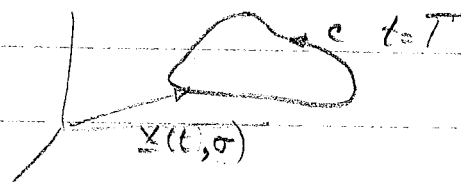
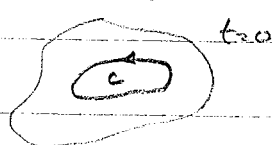


flow out of volume is $\iint v_n dS = \iiint \nabla \cdot \underline{v} d\tau$
 if $\underline{v} \cdot \underline{n} = 0 \Rightarrow \nabla \cdot \underline{v} = 0 \Rightarrow u_x + v_y + w_z = 0$

$$u_x + v_y + w_z = 0$$

Conservation of mass

Another simplifying assumption



Circulation $\Gamma(t) = \oint_C u dx + v dy + w dz = \oint_C \underline{q} \cdot d\underline{x} = \oint_C \underline{q} \cdot \underline{x}_\sigma d\sigma$
 Helmholtz law $\dot{\Gamma}(t) = 0$

For each time let s be arc length $\Rightarrow 0 \leq s \leq l$

let $\sigma = \frac{s}{l} \quad 0 \leq \sigma \leq 1$

then $\Gamma(t) = \int_0^1 \underline{q} \cdot \underline{x}_\sigma d\sigma$

$$\dot{\Gamma}(t) = \int_0^1 (\dot{\underline{q}} \cdot \underline{x}_\sigma + \underline{q} \cdot \dot{\underline{x}}_\sigma) d\sigma$$

$\dot{\underline{x}}_\sigma = \frac{\partial^2 \underline{x}}{\partial \sigma \partial t}$ by continuity $\dot{\underline{x}}_\sigma = \frac{\partial^2 \underline{x}}{\partial t \partial \sigma} = \underline{q}_\sigma$

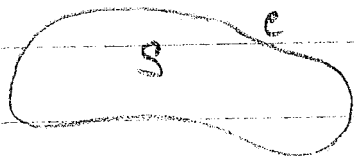
$$\dot{\Gamma}(t) = \int_0^1 (\dot{\underline{q}} \cdot \underline{x}_\sigma + \underline{q} \cdot \underline{q}_\sigma) d\sigma$$

$\underline{\dot{q}} = \underline{a}$ in Lagrangian $\underline{a} = -\frac{1}{\rho} \nabla p + \underline{\bar{F}}$ $\underline{\bar{F}} = (0, -g, 0)$
gravity

$$\dot{\Gamma}(t) = \int_0^1 \left(-\frac{1}{\rho} \underline{x}_\sigma \cdot \nabla p - g \underline{x}_\sigma \cdot \nabla y + \underline{q} \cdot \underline{q}_\sigma \right) d\sigma$$

$$\int_0^1 \left[-\frac{1}{\rho} p_{,\sigma} - g y_{,\sigma} + \frac{1}{2} (\underline{q} \cdot \underline{q})_{,\sigma} \right] d\sigma = 0$$

since $\sigma=0, \sigma=1$ is same pt. $\Rightarrow \Gamma(t) = \text{const}$



$$\oint_C \underline{q} \cdot d\underline{s} = \Gamma = \iint_S (\nabla \times \underline{q}) \cdot \underline{n} dA \quad \text{Stokes' Th.}$$

if Γ from rest $\Gamma=0$ if true for every closed curve

then $\nabla \times \underline{q} = 0$ irrotational $\Rightarrow \underline{q} = \nabla \Phi$ Φ a scalar fn.

$$\Phi(x, y, z, t) = \int_{x_1, y_1, z_1}^x u dx + v dy + w dz$$

$$\Phi_x = u \quad \Phi_y = v \quad \Phi_z = w$$

$\Rightarrow$ continuity $\Rightarrow \Delta \Phi = 0$

$$\begin{cases} u_t + u u_x + v u_y + w u_z = -\frac{1}{\rho} P_x \\ v_t + u v_x + v v_y + w v_z = -\frac{1}{\rho} P_y - g \end{cases}$$

Can write the above as $\nabla \Phi_t + \frac{1}{2} \nabla (\nabla \Phi)^2 = -\nabla(P/\rho) - \nabla(gy)$

Bernoulli law: $\Phi_t + \frac{1}{2} (\Phi_x^2 + \Phi_y^2 + \Phi_z^2) = -P/\rho - gy + \frac{1}{\rho} C(t)$

$\Rightarrow$ p can now be found we can assume $C(t) = 0$

On fixed body $\eta \cdot \nabla \Phi = 0$

Free surface

$$y = \eta(x, z, t)$$

will not be treated

$$S: y - \eta(x, z, t) = 0$$

$$\frac{d}{dt} (\eta(x, z, t) - y) = \Phi_x \eta_x - \Phi_y + \Phi_z \eta_z + \eta_t = 0$$

Kinematic free surface conditions.

If pressure prescribed on surface use Bernoulli law

$$\Phi_t + \frac{1}{2} (\nabla \Phi)^2 + g\eta = -P/\rho \quad \text{on } y = \eta(x, z, t)$$

Find a solution to $\Delta \Phi = 0$ with these two b.c. & two I.C.

I.C. initial position & velocity.

Assume $\Phi = \Phi_1 \epsilon + \dots$

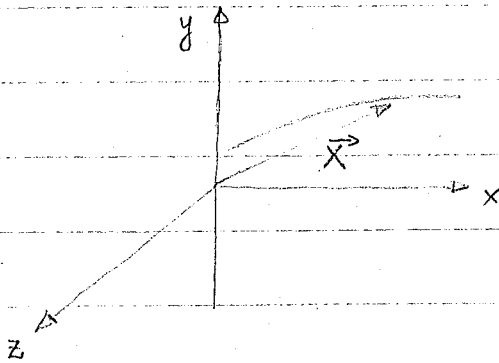
$$\eta = \epsilon \eta_1 + \dots$$

let $\Phi = \Phi(x, y, z, t) = \Phi(x, \epsilon \eta_1(x, z, t) + \dots, z, t)$ About some equil point:
 $= \Phi(x, 0, z, t) + \epsilon \Phi_y|_{y=0}$

for small first order case drop $\Phi_x \eta_x, \Phi_z \eta_z$

$$\Rightarrow \eta_t - \Phi_y|_{y=0} = 0 \quad \& \quad \Phi_t + g\eta = -P/\rho \quad \text{at } y=0$$

2-20-74



$\underline{q} = (u, v, w)$ velocity components

$p(x, y, z, t)$

$\underline{F} = (X, Y, Z) = (0, -g, 0)$

$\underline{X} = (x(t), y(t), z(t))$

$\frac{d\underline{X}}{dt} = (\dot{x}(t), \dot{y}(t), \dot{z}(t)) = \underline{q}$

$\frac{dx}{dt} = u(x, y, z, t)$

$\frac{dy}{dt} = v(x, y, z, t)$

$\frac{dz}{dt} = w(x, y, z, t)$

Conservation of Mass

$$\begin{cases} u_t + u u_x + v u_y + w u_z = \rho X - p_x \\ v_t + \dots = \rho Y - g - p_y \\ w_t + \dots = \rho Z - p_z \end{cases}$$

Conservation of mass $\nabla \cdot \underline{q} = u_x + v_y + w_z = 0$

$$\Gamma(t) = \oint \underline{v}_s \cdot d\underline{s}$$

$$\dot{\Gamma}(t) = 0$$

$\Phi(x, y, z, t)$

$\underline{q} = \nabla \Phi = (\Phi_x, \Phi_y, \Phi_z)$

$\nabla \times \underline{q} = 0$ irrotational

$$\Rightarrow \Phi_{xx} + \Phi_{yy} + \Phi_{zz} = 0$$

Bernoulli law $\Phi_t + \frac{1}{2} (\Phi_x^2 + \Phi_y^2 + \Phi_z^2) + p/\rho + gy = 0$

Energy, $E(t) = \rho \iiint_R \left[\frac{1}{2} (\Phi_x^2 + \Phi_y^2 + \Phi_z^2) + gy \right] dx dy dz$

$= - \iiint_R (p + \rho \Phi_t) dx dy dz$ no sources or sinks

$$\frac{dE}{dt} = \frac{d}{dt} \iiint_R f(x,y,z,t) dx dy dz = \iiint_R f_t dx dy dz + \iint_S f \underline{q} \cdot \underline{n} dS$$

$$\frac{dE}{dt} = \rho \iiint_R (\Phi_x \Phi_{x_t} + \Phi_y \Phi_{y_t} + \Phi_z \Phi_{z_t}) dx dy dz - \iint_S (p + \rho \Phi_t) \underline{q} \cdot \underline{n} dS$$

$$\iiint_R \nabla f \cdot \nabla g dx dy dz = - \iiint_R f \nabla^2 g dx dy dz + \iint_S f \frac{\partial g}{\partial n} dS$$

$$\frac{dE}{dt} = \rho \iiint_R \nabla \Phi \cdot \nabla \Phi_t dx dy dz - \iint_S (p + \rho \Phi_t) \underline{q} \cdot \underline{n} dS$$

$$= \rho \iint_S \Phi_t \frac{\partial \Phi}{\partial n} dS - \iiint_R \Phi_t \underbrace{\Delta \Phi}_{=0} dx dy dz - \iint_S (p + \rho \Phi_t) \underline{q} \cdot \underline{n} dS$$

$$\frac{dE}{dt} = \iint_S [\rho \Phi_t (\Phi_{,n} - v_n) - p v_n] dS \quad v_n = \underline{q} \cdot \underline{n}$$

a) Surface moving with fluid then $\dot{E} = - \iint p v_n dS$

b) " " " fixed in space & fluid moves along it $\dot{E} = 0$

c) " " " fixed in space but fluid moves through it. $\dot{E} = \rho \iint \Phi_t \Phi_{,n} dS$
 if $\Phi = \Phi(x-ct, y, z)$ $\xi = x-ct$ $\dot{E} = -\rho c \iint_S \Phi_{,\xi}^2 dS$

1. Solid $\Phi_n = 0$

2. free surface Kinematic condition if $y = \eta(x, z, t)$ $F = y - \eta(x, z, t) = 0$

particles in free surface stay in free surface. $\frac{dF}{dt} = \eta_t + u \eta_x - v + w \eta_z = 0$

$$\eta_x \Phi_x - \Phi_y + \eta_z \Phi_z + \eta_t = 0 \text{ on } y = \eta$$

dynamic free surface condition: pressure prescribed on free surface given by Bernoulli's law

$$\Phi_t + \frac{1}{2} (\Phi_x^2 + \Phi_y^2 + \Phi_z^2) + p/\rho + gy = 0 \text{ on } y = \eta$$

b.c. are not linear even though $\Delta\Phi=0$.

$$\text{if } p=0 \quad \eta=0 \text{ satisfies the cond } \Rightarrow \Phi_y=0 \text{ or } V_{\text{vertical}}=0 \\ \Rightarrow \Phi = Ux \rightarrow \text{Bernoulli's law} \Rightarrow \frac{1}{2}U^2$$

we assume small perturbation, in some variable ε , about exact soln.

$$\Phi(x, y, z, t, \varepsilon), \quad \eta(x, z, t, \varepsilon)$$

$$\Phi = \varepsilon \Phi_1 + \varepsilon^2 \Phi_2 + \dots$$

$$\Phi_i = \Phi_i(x, y, z, t, 0)$$

$$\eta = \varepsilon \eta_1 + \varepsilon^2 \eta_2 + \dots$$

$$\eta_i = \eta_i(x, z, t, 0)$$

$$p/p = p_0 + \varepsilon p_1 + \dots$$

$$\Delta\Phi = \varepsilon \Delta\Phi_1 + \dots$$

$$\text{since } \Delta\Phi=0 \Rightarrow \Delta\Phi_i=0$$

$$\eta_x \Phi_x - \Phi_y + \eta_z \Phi_z + \eta_t = 0 \Rightarrow$$

$$(\varepsilon \eta_{1x} + \varepsilon^2 \eta_{2x} + \dots)(\varepsilon \Phi_{1x} + \varepsilon^2 \Phi_{2x} + \dots) - (\varepsilon \Phi_{1y} + \varepsilon^2 \Phi_{2y} + \dots) \\ + (\varepsilon \eta_{1z} + \varepsilon^2 \eta_{2z} + \dots)(\varepsilon \Phi_{1z} + \varepsilon^2 \Phi_{2z} + \dots) + (\varepsilon \eta_{1t} + \dots) = 0$$

look at ε terms

$$-\Phi_{1y} + \eta_{1t} = 0$$

If we assume $y=y_0$ plane & η measured from it

$$\Rightarrow \eta(x, y_0, z, t) \Rightarrow \text{from Bernoulli's}$$

$$\frac{p_0}{\rho} + g y_0 + \eta(x, y_0, z, t) + \dots \quad \text{or } p_0 = -\rho g y_0 \text{ hydrostatic pressure}$$

$$\left. \begin{aligned} p/p + \Phi_{1t} + g \eta_1 &= 0 \\ -\Phi_{1y} + \eta_{1t} &= 0 \\ \Delta\Phi_1 &= 0 \end{aligned} \right\}$$

2-D $\Delta\Phi=0$ $0 > y > -\infty$ $|x| < \infty$

$p=0$ on bdy

$$\left. \begin{aligned} \Phi_t + g\eta &= 0 \\ -\Phi_y + \eta_t &= 0 \end{aligned} \right\} y=0$$

Bddness at ∞

$\Phi(x, y, t) = e^{i\sigma t} \varphi(x, y)$

$\Phi \equiv 0, \eta \equiv 0$ are soln.

diff 1st by t
$$\left\{ \begin{aligned} \Phi_{tt} + g\eta_t &= 0 \\ -g\Phi_y + g\eta_t &= 0 \end{aligned} \right\}$$

$$\Phi_{tt} + g\Phi_y = 0$$

$$(-\nabla^2 \varphi + g\varphi_y) e^{i\sigma t} = 0 \quad \Delta\Phi = \Delta\varphi = 0$$

let $\frac{\sigma^2}{g} = m \quad \varphi_y - m\varphi = 0 \quad \text{for } y=0$

φ & φ_y are bdd at infinity.

introduce $\psi(x, y) = \varphi_y - m\varphi$ since φ is harmonic $\Rightarrow \Delta\psi = 0$ with $\psi=0$ at $y=0$

ψ is bdd at ∞ use reflection principle since ψ is harmonic & bdd i

whole plane $\psi = \text{const} = 0$ by bdy cond $\therefore \varphi_y - m\varphi = 0$ everywhere

$\varphi = c(x) e^{my} \quad \Delta\varphi = 0 \Rightarrow c_{xx} + m^2 c = 0 \quad \text{or } c = e^{\pm i x m}$

$c(x) = a \cos mx + b \sin mx$

$$\Phi(x, y, t) = \begin{cases} \cos mx & \cos \sigma t \\ \sin mx & \sin \sigma t \end{cases}$$

Standing waves

$$\Phi(x, y, t) = A \cos mx \cos \sigma t e^{my} \quad \lambda = \frac{2\pi}{m}$$

Progressing waves.
$$\Phi(x, y, t) = \cos(m(x \pm \sigma t)) e^{my}$$

$m(x \pm \frac{\sigma}{m} t)$

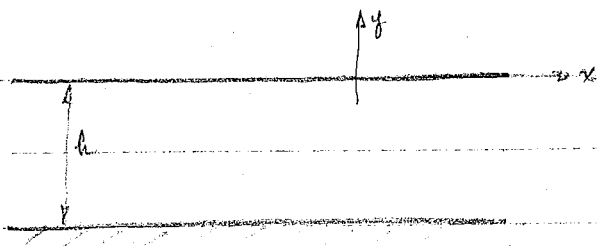
$\frac{\text{phase speed}}{\text{propagation speed}} = c = \frac{\sigma}{m} = \frac{\sigma \lambda}{2\pi}$

$\sigma^2 = g m = \frac{2\pi g}{\lambda}$; c: speed phase

$T = \frac{2\pi}{\sigma}$

$c = \sqrt{\frac{g\lambda}{2\pi}}$

dispersion



$$\Delta \phi = 0$$

$$\phi_y - m\phi = 0 \quad y=0$$

$\phi=0$ is a soln.

$$\phi_y = 0 \quad y=h$$

$$\phi = \cosh m(y+h) \begin{cases} \cos mx \\ \sin mx \end{cases}$$

$$\phi_y = m \sinh m(y+h) \begin{cases} \cos mx \\ \sin mx \end{cases} \quad \text{at } y=-h = 0$$

$$\phi_y - m\phi = m \sinh mh - m \cosh mh = 0$$

$$\Rightarrow \tanh mh = 1 \Rightarrow$$

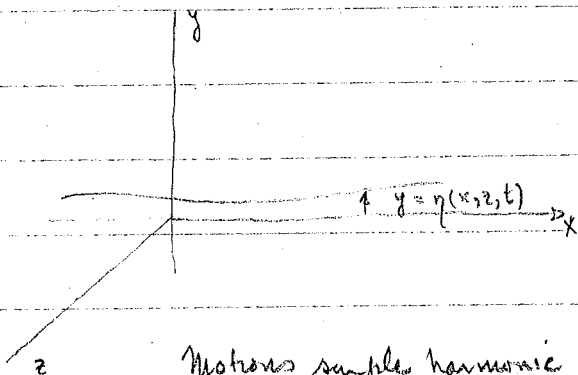
$$\sigma^2 = gm \tanh mh = g \frac{2\pi}{\lambda} \left[\frac{2\pi h}{\lambda} - \frac{1}{3} \left(\frac{2\pi h}{\lambda} \right)^3 + \dots \right]$$

$$\text{let } \frac{h}{\lambda} \rightarrow 0$$

$$\sigma^2 \rightarrow \left(\frac{2\pi}{\lambda} \right)^2 gh = m^2 gh$$

$$c = \frac{\sigma}{m} \rightarrow \sqrt{gh}$$

2-27-74



Flow governed by $\Phi(x, y, z, t)$ $(u, v, w) = (\Phi_x, \Phi_y, \Phi_z)$

flow perfect, irrotational $\nabla^2 \Phi = 0$

$$\text{B.C. } \left. \begin{aligned} -\Phi_y + \eta_t &= 0 \\ \Phi_t + g\eta + p_p &= 0 \end{aligned} \right\} y=0$$

Motions simple harmonic in time studied

Consider $\Phi = e^{i\sigma t} \varphi(x, y)$ ansatz

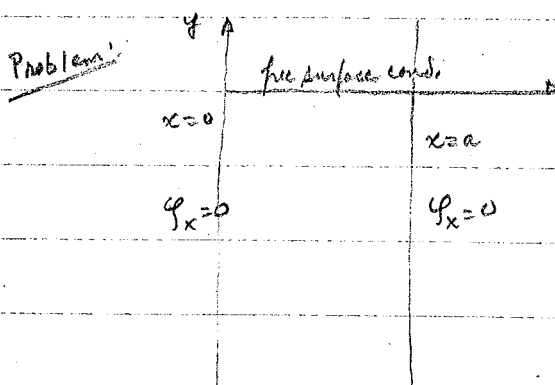
$$\text{solutions } \varphi(x, y) = e^{my} \begin{cases} \cos mx \\ \sin mx \end{cases}$$

$$\rightarrow \Phi = A e^{my} \cos(mx \pm \sigma t + \alpha)$$

$$+ \sin \sigma t \quad - \sin \sigma t$$

$$\bar{\Phi} = \Phi(x \pm ct, y)$$

Problem:



$x=0$
 $\varphi_x=0$

$x=a$
 $\varphi_x=0$

$\Phi = e^{i\omega t} \varphi(x, y)$ point spectrum
 $\nabla^2 \varphi = 0$: $\varphi = g(y) \cos \frac{n\pi x}{a}$ + free body cond.

Now consider 2-D case uniform but finite height h -



$\varphi = \cosh m(y+h) \begin{cases} \cos mx \\ \sin mx \end{cases}$

$\varphi_y = 0$ at $-h$

for free surface cond $\sigma^2 = g m \tanh m h$ $m = \frac{2\pi}{\lambda}$

Soln $\varphi = \cosh m(y+h) \cos(mx - \sigma t + \alpha)$

$c = \frac{\sigma}{m}$
 $\sigma^2 = \frac{2\pi g}{\lambda} \left[\frac{2\pi h}{\lambda} - \frac{1}{3} \left(\frac{2\pi h}{\lambda} \right)^3 - \dots \right]$

$\sigma^2 = \left(\frac{2\pi}{\lambda} \right)^2 g h = m^2 g h$ as $\frac{h}{\lambda} \rightarrow 0$ $c \rightarrow \sqrt{g h}$

$\delta x, \delta y$

$\frac{d}{dt} \delta x = \Phi_x = -m A \cos \sigma t \cosh m(y+h) \sin mx$

$\frac{d}{dt} \delta y = \Phi_y = m A \cos \sigma t \tanh m(y+h) \cos mx$

$\delta x = -\frac{m A}{\sigma} \sin \sigma t \cosh m(y+h) \sin mx$

$\delta y = \frac{m A}{\sigma} \sin \sigma t \sinh m(y+h) \cos mx$

particle path are ellipses

$\Phi = A \cosh m(y+h) \cos(mx \pm \sigma t + \alpha)$

$\frac{\delta x^2}{a^2} + \frac{\delta y^2}{b^2} = 1$

Assume $\Phi = e^{i\omega t} \varphi(x, y, z) = e^{i\omega t} \varphi(r, y)$ not dependent on θ

$\nabla^2 \varphi(r, y) = \frac{1}{r} \partial_r(r \partial_r \varphi) + \partial_y^2 \varphi = 0$ $-\infty < y < 0$ $0 \leq r < \infty$

$$\varphi_y - m\varphi = 0 \text{ at } y=0$$

$$\text{Look for } \varphi = e^{my} g(r)$$

$$\Rightarrow \frac{1}{r} \partial_r (r \partial_r g) + m^2 g = 0$$

$$\text{let } \rho = \log r \Rightarrow \nabla^2 \varphi = e^{-2\rho} \frac{\partial^2 \varphi}{\partial \rho^2} + \frac{\partial^2 \varphi}{\partial y^2} = 0$$

$$\text{let } \psi = \varphi_y - m\varphi \quad \psi \text{ satisfies } \nabla^2 \psi = 0$$

$$g = J_0(mr), Y_0(mr)$$

entire fn. log singularity.

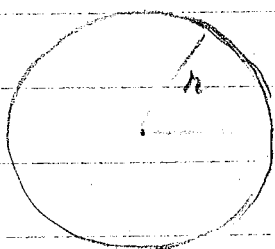
Asymptotic large values of r

$$J_0(mr) \sim \sqrt{\frac{2}{\pi mr}} \cos\left(mr + \frac{\pi}{4}\right)$$

$$Y_0(mr) \sim \sqrt{\frac{2}{\pi mr}} \sin\left(mr + \frac{\pi}{4}\right)$$

$$\text{take } H_0 = J_0 + i Y_0$$

$$\begin{aligned} \Phi(r, y, t) &= A e^{my} e^{-i\omega t} \sqrt{\frac{2}{\pi mr}} e^{i(mr - \pi/4)} \\ &= A e^{my} \frac{2}{\sqrt{\pi mr}} e^{i(mr - \omega t)} e^{-i\pi/4} \end{aligned}$$

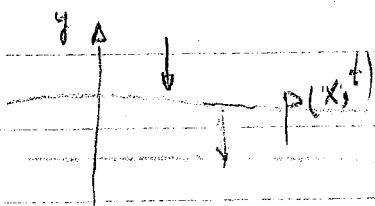


energy flux out of cylinder $\int$ for 1 period

$$F = \int_0^T \left\{ \int_0^{2\pi} \Phi_t \Phi_r r d\theta \right\} dt$$

$\Phi_t \Phi_r \cdot r$ are indep of r to first order approx

& hence $F \neq F(r) \neq 0$ Φ_t, Φ_r have opp signs & hence $F < 0$



$\Rightarrow x$

$$\Phi(x, y, t) = 0$$

$$\partial_y \eta = -\Phi_t - p/\rho \quad y=0$$

$$\eta_t = \Phi_y \quad y=0$$

$$p(x,t) = \bar{p}(x) \sin \sigma t$$

$$\Phi = \varphi(x,y) \cos \sigma t + \chi(x,y) \sin \sigma t$$

$$g \eta_t = -\Phi_{tt} - \frac{p_t}{\rho}$$

$$g \eta_t = \Phi_{yy} g$$

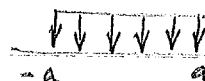
$$g \Phi_{yy} + \Phi_{tt} + \frac{p_t}{\rho} = 0$$

$$y=0 \begin{cases} \varphi_y - m \varphi = -\frac{\sigma}{\rho g} \bar{p}(x) \\ \chi_y - m \chi = 0 \end{cases}$$

χ satisfies previous cond. since harmonic

$$\therefore \chi(x,y) = A e^{my} \begin{cases} \cos mx \\ \sin mx \end{cases}$$

we will assume $\bar{p}(x) = \text{const. } P \quad |x| < a$



0

$$|x| > a$$

$$\varphi_y - m \varphi = \begin{cases} c & |x| < a \\ 0 & |x| > a \end{cases} \quad m = \frac{\sigma^2}{g} \quad c = -\frac{P\sigma}{\rho g}$$

Boundary at $\infty \quad \varphi, \varphi_x, \varphi_y \rightarrow 0$.

at $\pm a$ p is discontinuous we assume η is continuous $\Rightarrow \Phi_t$ must

be discontinuous; we assume $\varphi_x, \varphi_y \sim \frac{1}{|x|^\varepsilon} \quad \varepsilon > 0$

let $x_1 = mx \quad y_1 = my \quad a_1 = ma \quad c_1 = \frac{c}{m}$

$$\Rightarrow \nabla^2 \varphi = 0 \quad \varphi_{y_1} - \varphi = \begin{cases} c_1 & |x_1| \leq a_1 \\ 0 & |x_1| > a_1 \end{cases}$$

drop subscript $\nabla^2 \varphi = 0 \quad \varphi_y - \varphi = \begin{cases} c & |x| \leq a \\ 0 & |x| > a \end{cases}$

$$\varphi(x, y) \quad \& \quad \text{an anal } \psi(x, y) \Rightarrow \varphi + i\psi = f(z)$$

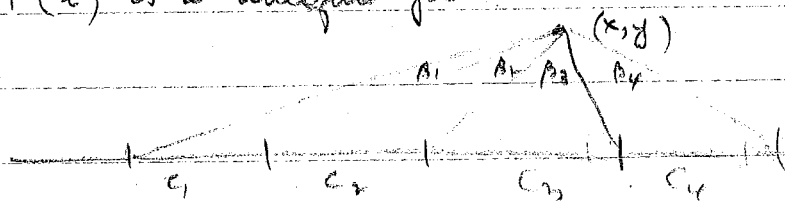
$$\frac{df}{dz} = \varphi_x + i\psi_x = \varphi_x - i\psi_y \quad \text{velocity: } (\varphi_x, \varphi_y) \quad \varphi_x + i\varphi_y = \text{complex vel.}$$

$$\begin{aligned} (\varphi_y - \psi) &= \left(\frac{\partial}{\partial y} - 1\right) \varphi = \operatorname{Re} \left(\frac{\partial}{\partial y} - 1\right) (\varphi + i\psi) \\ &= \operatorname{Re} \left(\frac{\partial}{\partial y} - 1\right) f(z) = \operatorname{Re} \left(i \frac{df}{dz} - f\right) \end{aligned}$$

$$\varphi_y - \psi = \begin{cases} c \\ 0 \end{cases} \rightarrow \operatorname{Re} (i f' - f) = \begin{cases} c \\ 0 \end{cases} \text{ on real axis}$$

$$\text{let } F'(z) = i f' - f. \quad \text{for uniqueness let } G_1 = G_2 = G$$

$\operatorname{Re} G$ is real on real axis continue it by Schwarz Reflection
at $\pm a$ we define the f_n is a nbh about the pts $\Rightarrow$ since sing
is less than pole it is a removable singularity $\Rightarrow$ that
means G is harmonic & is bdd in entire plane hence $G = \text{const}$
(real const) since zeros on real axis $\Rightarrow G = 0 \therefore G_1 = G_2$
hence $F(z)$ is a unique fn.



$$= \arctan \frac{y}{x-a}$$

$$\varphi = \frac{1}{\pi} (c_1 \beta_1 + c_2 \beta_2 + c_3 \beta_3 + c_4 \beta_4)$$

$$F(z) = \frac{ic}{4} \log \frac{z-a}{z+a} = \frac{ic}{\pi} (\log R_1 + i\theta_1 + \log R_2 - i\theta_2)$$

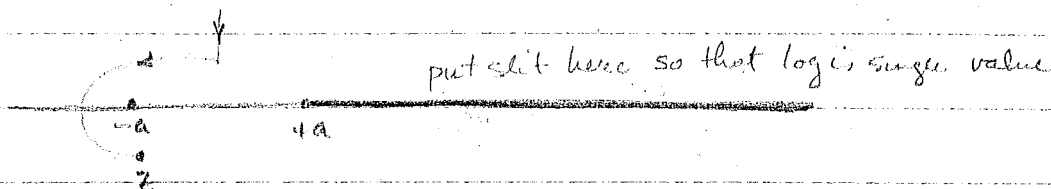
$$= -\frac{c}{\pi} (\theta_1 - \theta_2) + 2\arg.$$

$$if' - f = \frac{ic}{\pi} \log \frac{z-a}{z+a}$$

sol to $|x| > a$

$$if' - f = 0 \quad f = e^{-iz} = e^y e^{-ix} \quad \text{since } y < 0 \text{ O.K.}$$

$$f(z) = \frac{c}{\pi} e^{-iz} \int_{z_0}^z e^{it} \log \frac{t-a}{t+a} dt + Ae^{-iz}$$

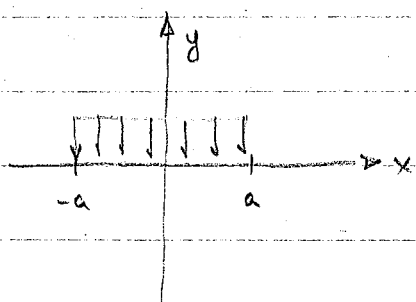


Integrate by parts

$$f(z) = \frac{ci}{\pi} \left[-\log \frac{z-a}{z+a} + e^{-iz} \int_{i\infty}^z e^{it} \left(\frac{1}{t-a} - \frac{1}{t+a} \right) dt \right]$$

$$\log \frac{1 - a/z}{1 + a/z} = \log \left(1 - \frac{za}{z} + \dots \right)$$

3-6-74



$$\Phi(x, y, t) = \varphi(x, y) e^{i\omega t}$$

$$p(x, t) = \begin{cases} P \sin \omega t & |x| \leq a \\ 0 & |x| > a \end{cases} \quad P = \text{const}$$

Introduce $f(z) = \varphi(x, y) + i\psi(x, y) \quad z = x + iy$

B.c. $\varphi_y - m\varphi = -\frac{\sigma}{\rho g} P, \quad y=0 \quad |x| \leq a \quad \left\| \begin{array}{l} \psi_y - m\psi = 0 \\ \psi = 0 \end{array} \right. \quad |x| > a \quad m = \frac{\sigma^2}{g}$

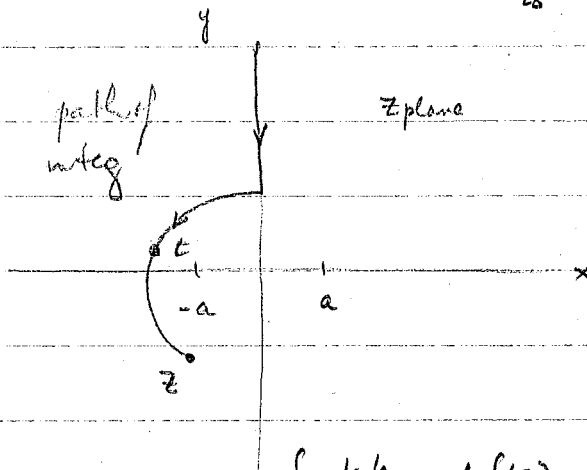
$$\text{Re} \left(i \frac{df}{dz} - f \right) = \begin{cases} c, & |x| \leq a \\ 0 & |x| > a \end{cases}$$

$$x_1 = mx, \quad y_1 = my \quad a_1 = ma \quad c_1 = \frac{c}{m}$$

$$\text{Let } F(z) = i \frac{df}{dz} - f$$

$$F(z) = i f'(z) - f = \frac{ic}{\pi} \log \frac{z-a}{z+a}$$

$$f(z) = Ce^{-iz} + \frac{c}{\pi} e^{-iz} \int_{z_0}^z e^{it} \log \frac{t-a}{t+a} dt$$



first term of $f(z)$ gives $e^y \{ A \cos x + B \sin x \}$

look at last integral

$$e^{-iz} \int_{z_0}^z e^{it} \log \frac{t-a}{t+a} dt = -\frac{ci}{\pi} e^{-iz} \int_{z_0}^z e^{it} \left(\frac{1}{t-a} - \frac{1}{t+a} \right) dt + \frac{c}{\pi} \log \frac{(z-a)}{(z+a)}$$

last part $\rightarrow 0$ as $z \rightarrow \infty$

$$I(z) = e^{-iz} \int_{i\infty}^z \frac{e^{it}}{t \pm a} dt = \frac{-i}{z \pm a} - i \int_{i\infty}^z \frac{e^{i(t-z)}}{(t \pm a)^2} dt$$

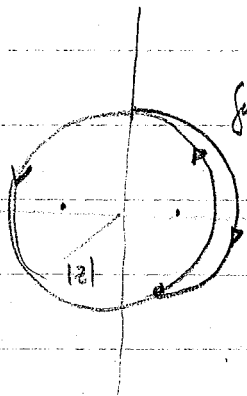
$$\frac{1}{t^2(1 \pm \frac{a}{t})^2} = \frac{1}{t^2} \left[1 \mp \frac{2a}{t} + \dots \right] \quad t-z \text{ is positive}$$

Re part is negative since $y-y_0$

$\therefore e^{i(t-z)}$ is decreasing

$$\therefore I(z) \leq \frac{1}{|z \pm a|} + \int_{\pi|z|}^{\infty} \frac{|dt|}{|t|^2} \sim \frac{1}{|z|}$$

for $\text{Re } z > 0$



for $\exists t \Rightarrow \text{Re } z - t > 0$ we close off the circle

in half plane

and retrace it $\Rightarrow$ from closed circle we get

residue from $-a, a$ and from we get the

same thing as before.

hence we get from residue

$$2\pi i e^{-iz} (e^{ia} - e^{-ia})$$

$$= e^{-iz} \sin a$$

$$f(z) \sim \begin{cases} \frac{1}{z} & \text{Re } z < 0 \\ -4c i \sin a e^{-iz} + O\left(\frac{1}{z}\right) & \text{Re } z > 0 \end{cases}$$

$$\varphi(x, y) \sim \text{Re } f(z) \sim \begin{cases} \frac{1}{x}, & x < 0 \\ -4c i \sin a e^{y \sin x} + O\left(\frac{1}{x}\right) & x > 0 \end{cases}$$

$$\Phi(x, y, t) = (\varphi(x, y) + A e^{y \sin x} + B e^{y \cos x}) e^{\cos t} + O\left(\frac{1}{x}\right) \quad x < 0$$

we had assumed $\Phi(x, y, t) = y \sin t + x \cos t$

$$\Phi(x, y, t) = (C e^{y \sin x} + D e^{y \cos x}) \sin t$$

we want at $-\infty$:

$$\Phi \sim e^y [H \sin(x+ot) + K \cos(x+ot)] + O\left(\frac{1}{x}\right)$$

at $+\infty$:

$$\Phi = e^y [L \sin(x-ot) + M \cos(x-ot)] + y \cos ot$$

4 eq & 4 unknowns $\Rightarrow L = -H \quad L = H - 4c \sin a$

$$M = K \quad M = -K \Rightarrow 0 = M = K$$

$$L = -2c \sin a \quad H = 2c \sin a$$

for right hand side alone because of sign about y axis

$$\Phi = \frac{2P_0}{\rho g m} \sin ma e^{my} \sin(mx - \omega t) + O\left(\frac{1}{n}\right) \quad x \geq 0$$

at ∞ $m = \frac{2\pi}{\lambda}$; if $\sin ma = 0$ $ma = n\pi$ then $2a = \text{interval of pressure application}$ $\therefore 2a = n\lambda_{\infty}$

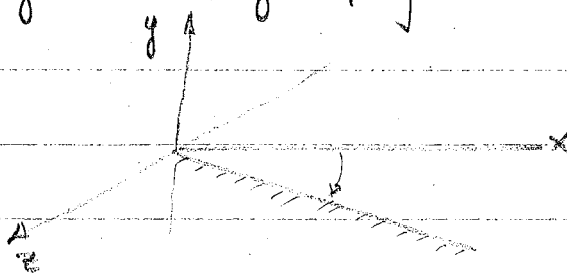
oscillatory pressure does no net work on the fluid since flux is $\int \Phi \Phi_t dx dt$
 $= \int O\left(\frac{1}{n}\right) O\left(\frac{1}{n}\right) dx \sim O\left(\frac{1}{n}\right) \rightarrow 0$ as $n \rightarrow \infty$
 $\therefore$ no flux at ∞ for 1 period

let $2Pa = 1$ let $P \uparrow$ at ∞ $\Rightarrow$ we get delta function $\delta(x) = 1$ limit

Exercise show that if $p(x, t) = \delta(x) \sin \omega t$

$$f(z) = \frac{C}{\pi} e^{-iz} \int_0^z \frac{e^{it}}{t} dt$$

Progressing waves along sloping beach



Breaking of waves absorbs energy of incoming waves.

$$P(x, t) = p(x) \sin \omega t \quad t > 0$$

$$\Phi(x, y, 0) = 0$$

since Φ harmonic $\Rightarrow \eta$ is harmonic

$$-\Phi_y + \eta_t = 0$$

$\Rightarrow p$ is harmonic

$$\Phi_t + g\eta = -p/\rho$$

since we assume $p = \text{const}$

} $y = 0$

$$g \Phi_y + \Phi_{tt} = -i\omega p(x) e^{i\omega t} \quad t > 0$$

$$\Delta \Phi = 0$$

$$\Phi = \Phi_t = 0 \text{ at } t=0$$

$$\Phi_t + g\eta = 0 \quad \text{at } y=0 \Rightarrow \eta=0 \text{ which is O.K.}$$

Fourier Integral Th. $f(x)$: piecewise cont / ^{1st} deriv with finite discont.

$$-\infty < x < \infty \quad f(x) = \frac{1}{\pi} \int_0^\infty d\alpha \int_{-\infty}^\infty f(\eta) \cos \alpha(\eta-x) d\eta$$

$$\& \int_{-\infty}^\infty |f(x)| dx < \infty$$

$$f(x) = \frac{1}{\pi} \lim_{\xi \rightarrow \infty} \int_{-\infty}^\infty f(\eta) d\eta \int_0^\xi \cos s(\eta-x) ds$$

$$\int_{-\xi}^\xi \cos s(\eta-x) ds = 2 \int_0^\xi$$

$$\int_{-\xi}^\xi \sin s(\eta-x) ds = 0$$

$$\int_0^\xi \cos s(\eta-x) ds = \frac{1}{2} \int_{-\xi}^\xi e^{is(x-\eta)} ds$$

$$\therefore f(x) = \frac{1}{2\pi} \int_{-\infty}^\infty e^{isx} ds \int_{-\infty}^\infty f(\eta) e^{-i\eta s} d\eta$$

$$f(s) = \frac{1}{\sqrt{2\pi}} \int_{-\infty}^\infty f(\eta) e^{-i\eta s} d\eta$$

$$f(x) = \frac{1}{\sqrt{2\pi}} \int_{-\infty}^\infty f(s) e^{isx} ds$$

$$\overline{\frac{d^n f}{dx^n}} = \frac{1}{\sqrt{2\pi}} \left. \frac{d^{n-1} f}{dx^{n-1}} e^{-isx} \right|_{-\infty}^{\infty} + is \int_{-\infty}^{\infty} f^{(n-1)} e^{-isx} dx$$

we assume derivs are cont. we assume $f^{(n-1)} \rightarrow 0$ at $x \rightarrow \pm\infty$

then

$$\overline{\frac{d^n f}{dx^n}} = (-is)^n \bar{f}$$

$$\bar{\Phi}_{xx} + \bar{\Phi}_{yy} = 0$$

$$\left[(-is)^2 + \frac{d^2}{dy^2} \right] \bar{\Phi} = 0$$

$$\bar{\Phi} = \bar{a}_{\pm}(s,t) e^{\pm sy} \quad \text{from bddness of } \bar{\Phi}$$

$$\text{for } y \rightarrow -\infty \Rightarrow \bar{\Phi} = \bar{a}_+(s,t) e^{+sy}$$

3-13-74

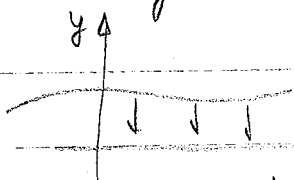
Fourier transform

$$\bar{f}(s) = \frac{1}{\sqrt{2\pi}} \int_{-\infty}^{\infty} f(\eta) e^{-i\eta s} d\eta \quad \int_{-\infty}^{\infty} |f(\eta)| d\eta < \infty$$

$$f(x) = \frac{1}{\sqrt{2\pi}} \int_{-\infty}^{\infty} \bar{f}(s) e^{+ixs} ds$$

Courant-Hilbert Vol 2

Unsteady Motion



$$p(x,t) = p(x) e^{i\omega t} \quad \text{at } y=0$$

$$\text{veloc potential } \bar{\Phi}(x,y,t) : \Delta \bar{\Phi} = 0 \quad y < 0$$

$$\text{with free surface cond. } g \bar{\Phi}_y + \bar{\Phi}_{tt} = -\frac{i\omega}{\rho} p(x) e^{i\omega t} \quad t > 0, y=0$$

$$\text{apply fourier transf on } \Delta \bar{\Phi} = 0 \Rightarrow -s^2 \bar{\Phi} + \bar{\Phi}_{yy} = 0$$

$$\bar{\Phi} = b(s,t) e^{-sy} + a(s,t) e^{sy} \quad \text{bddness cond} \Rightarrow b(s,t) = 0$$

$$\bar{\Phi} = a(s,t) e^{sy}$$

put into b.c.

$$A_{tt} + g s A = -\frac{i\omega}{\rho} \bar{p}(s) e^{i\omega t} \quad \text{for } y \geq 0$$

$$\begin{aligned}\bar{p} &= \frac{1}{\sqrt{2\pi}} \int_{-a}^a c e^{-isx} ds = \frac{ci}{\sqrt{2\pi}s} [e^{-isa} - e^{isa}] \\ &= \frac{ci}{\sqrt{2\pi}s} [-2i \sin sa] = \frac{2aci}{\sqrt{2\pi}} \left[\frac{-2i \sin sa}{2as} \right]\end{aligned}$$

$$\text{let } a \rightarrow 0 \quad \& \quad 2ac = 1$$

$$\therefore \lim_{a \rightarrow 0} \bar{p} = \frac{1}{\sqrt{2\pi}}$$

$$\therefore A_{tt} + gSA = -\frac{i\omega}{\rho} \frac{1}{\sqrt{2\pi}} e^{i\omega t}; \quad \text{we have also that fluid at rest at } t=0 \quad \Phi = \Phi_t = 0 \Rightarrow A, A_t = 0 \text{ at } t=0$$

$$A = \kappa(t) \cos \sqrt{gs} t + \lambda(t) \sin \sqrt{gs} t \quad \text{but } \Phi = 0 \text{ at } t=0 \Rightarrow$$

$$A(s,t) = -\frac{1}{\sqrt{2\pi}} \frac{i\omega}{\rho} \int_0^t \frac{e^{i\omega(t-\tau)}}{\sqrt{gs}} \sin \sqrt{gs} \tau \, d\tau$$

$$\Phi(x,y,t) = -\frac{i\omega}{\rho\pi} \int_0^\infty e^{sy} \cos sx \int_0^t \frac{e^{i\omega(t-\tau)}}{\sqrt{gs}} \sin \sqrt{gs} \tau \, d\tau \, ds$$

$$\text{let } \sin \sqrt{gs} \tau = \frac{e^{i\sqrt{gs}\tau} - e^{-i\sqrt{gs}\tau}}{2i}$$

$$= -\frac{i\omega e^{i\omega t}}{\pi\rho} \int_0^\infty e^{sy} \cos sx \cdot \left\{ -\frac{1}{2\sqrt{gs}} \frac{e^{i(\sqrt{gs}-\omega)t}}{\sqrt{gs}-\omega} \right.$$

$$\left. -\frac{1}{2\sqrt{gs}} \frac{e^{-i(\sqrt{gs}+\omega)t}}{\sqrt{gs}+\omega} + \frac{1}{gs-\omega^2} \right\} ds$$

Re $i(\sqrt{gs} \pm \omega) < 0$ (make branch cut along $-s$ axis) and $\sqrt{gs} < \frac{\pi}{2}$
we get no contrib from the exponentials at $t \rightarrow \infty$

$$\therefore \underset{t \text{ large}}{\Phi(x,y,t)} \sim \frac{i\omega}{\pi\rho} e^{i\omega t} \int \frac{e^{sy} \cos sx}{gs - \omega^2} ds \quad \text{Lamb}$$

went $x \rightarrow \infty$ let $\cos sx = \frac{e^{isx} + e^{-isx}}{2}$

$\Phi \sim \frac{\omega}{\rho} e^{i\omega t} \int_{-l}^l \left(\frac{e^{sy} e^{isx}}{2(qs - \omega^2)} + \frac{e^{sy} e^{-isx}}{2(qs - \omega^2)} \right) ds$

& add & subtract an arc 

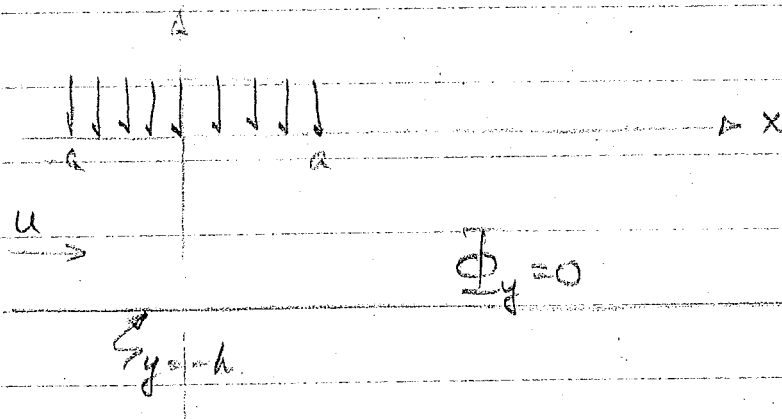
$\sim \frac{\omega}{\rho} e^{i\omega t} \left[\frac{1}{2\pi i} \oint \frac{e^{sy} e^{isx}}{qs - \omega^2} + \frac{1}{2\pi i} \oint \frac{e^{sy} e^{-isx}}{qs - \omega^2} \right] ds$

$\sim \frac{\omega i}{\rho} e^{\frac{\omega^2 y}{g}} e^{-i(\frac{\omega^2}{g} x - \omega t)}$

= 0 since analytic in upper half plane

$\therefore \Phi \sim \frac{m i}{\sigma \rho} e^{m y} e^{-i(mx - \sigma t)}$ let $\sigma = \omega$

if $\frac{\omega^2}{g} = m$



$\Phi(x, y, t) = Ux + \varphi(x, y, t)$

free surface $\left\{ \begin{aligned} \frac{\partial \varphi}{\partial t} + g\eta + \varphi_t + U\varphi_x + \frac{1}{2} U^2 &= 0 \end{aligned} \right.$ about $U^2 \rightarrow \rho$

kinematic $y=0 \left\{ \begin{aligned} \eta_t + U\eta_x - \varphi_y &= 0 \end{aligned} \right.$

found that the sol depends on $U^2/g h$

from before $U^2 = \frac{g\lambda}{2\pi} \tanh \frac{2\pi h}{\lambda}$ $\frac{U^2}{gh} = \frac{1}{2\pi} \frac{\lambda}{h} \tanh \frac{2\pi h}{\lambda}$

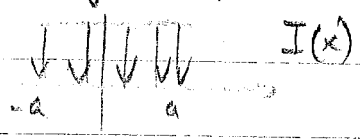
as $\lambda \rightarrow \infty$ $\therefore \frac{U^2}{gh} \rightarrow 0$ $U^2/g h \rightarrow 1$

$\frac{u^2}{gh} < 1$ need radiation condition

$\frac{u^2}{gh} > 1$ no radiation condition

for $\frac{u^2}{gh} = 1$ look at time dependent case by solving free surface for η
 & put into denominator, use fourier transform now let $t \rightarrow \infty$ & the
 $x \rightarrow \infty \Rightarrow \Phi \sim t^{2/3}$ velocities behave like $t^{1/3}$

take infinite depth discuss motion of pressure impulse over some region



$$\Phi(x, 0, 0) = -\frac{1}{\rho} I(x)$$

obtained from $\Phi_t + g\eta = -\frac{1}{\rho} p$; $y=0$
 $\int_0^T p dt = -\rho \Phi(x, 0, z, T) - g\rho \int_0^T \eta dt$

as $z \rightarrow 0$ η remains bdd
 $I = -\rho \Phi + 0 \quad \Rightarrow \quad \Phi(x, 0, 0) = -\frac{1}{\rho} I(x)$

assume $\Phi_t(x, 0, 0) = 0$? we will go to diff problem

let $\Phi_t = -g\eta$ and let $\Phi = -g \int \eta dt = \delta$

$$\Phi = \frac{\sqrt{g}}{\pi} \int_0^\infty e^{sy} \cos sx \sin \sqrt{gs} t \frac{ds}{\sqrt{s}}$$

$$\eta = -\frac{1}{g} \Phi_t \Big|_{y=0} = \frac{1}{\pi} \lim_{y \rightarrow 0} \int_0^\infty e^{sy} \cos sx \cos \sqrt{gs} t ds$$

$$I(k) = \int_a^b \varphi(\xi) e^{ik\varphi(\xi)} d\xi$$

want to know what happens when $k \gg 1$ assume ψ not oscillating
 ψ & ψ' real

Kelvin Douglas


 $\varphi'(\xi) \neq 0$ in interval
 distinct. does not like $1/k$

if $\exists$ an $\alpha \Rightarrow \varphi'(\alpha) = 0, \varphi''(\alpha) = 0$ disturbance goes like $\frac{1}{\gamma_k}$

Principal of stationary phase.

Poincaré proved rigorously for ψ, φ analytic

integrate by parts

integrate by parts if $\varphi'(\xi) \neq 0$ in interval

$$I(k)_c = \int_c^d \frac{\psi(\xi)}{iK\varphi'(\xi)} \frac{d}{d\xi} \left[e^{iK\varphi(\xi)} \right] d\xi$$

$$= \frac{\psi(\xi)}{iK\varphi'(\xi)} e^{iK\varphi(\xi)} \Big|_c^d - \frac{1}{iK} \int_c^d e^{iK\varphi} \frac{d}{d\xi} \left(\frac{\psi}{\varphi'} \right) d\xi$$

us $\alpha + x$ $\varphi(x) = \varphi(\alpha) + \frac{1}{2} x^2 \varphi''(\alpha^*)$ if $\varphi'(\alpha) = 0$

$$I = \psi(x) e^{ik\varphi(x)} \int_{x-x}^{x+x} e^{\frac{ik}{2} \varphi''(x) x^2} dx$$

let $\tau^2 = \pm \frac{k}{2} \varphi''(x)$

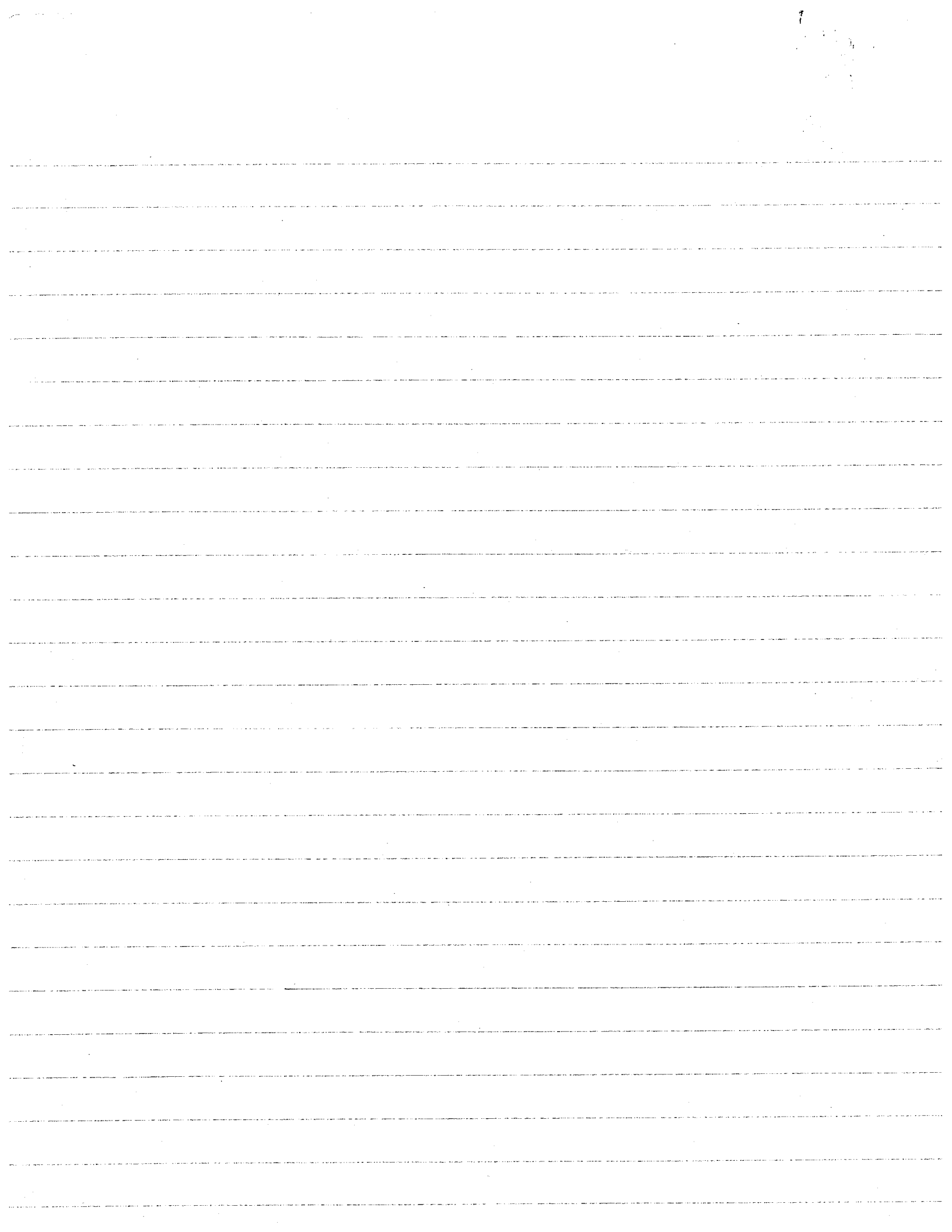
$$I = \psi(x) e^{ik\varphi(x)} \int_{x-x}^{x+x} e^{i(\pm x^2 \mp^2)} dx$$

don't change integral by going $\int_{-\infty}^{\infty}$

$$I = \psi(x) e^{ik\varphi(x)} \frac{\sqrt{\pi}}{2} e^{\pm i\pi/4} \approx \sqrt{2\pi} \psi(x) e^{ik\varphi(x) \pm i\pi/4}$$

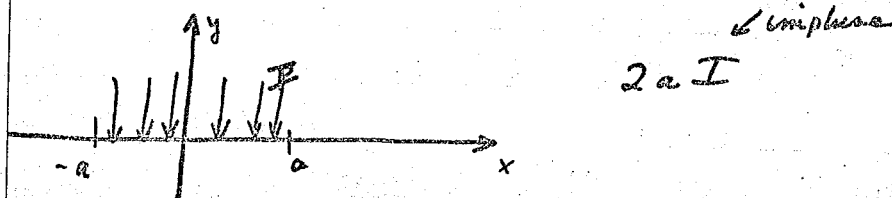
$$\pm \exists, \sqrt{|K(\varphi''(\alpha))|} \text{ positive}$$

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then take $\sum$ over all such pts where $\varphi'(x) \neq 0$
these are asymptotic expansions.



Unsteady Motions

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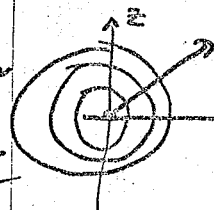


$$\Phi(x, y, t) = -\frac{2Pa}{\pi \rho} \int_0^\infty \frac{\sin sa}{sa} e^{sy} \cos sx \cos(\sqrt{g}s t) dt$$

$$\eta(x, t) = -\frac{1}{g} \frac{\partial \Phi}{\partial t} \Big|_{t=0} = -\frac{2Pa}{\pi \rho \sqrt{g}} \lim_{y \rightarrow 0} \int_0^\infty \frac{\sin sa}{sa} e^{sy} \cos(\sin \sqrt{g}s t) \sqrt{g} ds$$

$$\eta(x, t) = \frac{1}{\pi} \lim_{y \rightarrow 0} \int_0^\infty e^{sy} \cos sx \cos(\sqrt{g}s t) ds$$

(case of initial elevation)



$$\eta(r, t) = \frac{-1}{2\pi \sqrt{g}} \lim_{y \rightarrow 0} \int_0^\infty e^{sy} J_0(sr) \cos(\sqrt{g}s t) s^{3/2} ds$$

H. transf.

exact: $\eta(x, t) = \frac{1}{\pi x} \left\{ \frac{gt^2}{2x} - \frac{1}{1 \cdot 3 \cdot 5} \left(\frac{gt^2}{2x} \right)^3 + \dots \right\}$

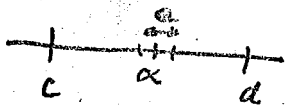
for $\frac{gt^2}{2x}$ small conv. , entire set.

$\therefore$ as soon as disturbance starts, its felt throughout fluid

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$$I(k) = \int_a^b \psi(s) e^{ik \phi(s)} ds, \quad k, \phi \text{ real set}$$

for $\psi'(s) \neq 0$ on $[c, d]$



Then $I(k) \sim \frac{1}{k}$

$$I(k) \approx \frac{\sqrt{2\pi} \psi(\alpha)}{\sqrt{\pm \psi'(\alpha)}} e^{i(k\psi(\alpha) \pm \pi/4)} + O\left(\frac{1}{k}\right)$$

$$\psi'(\alpha) = \psi''(\alpha) = 0, \quad \psi'''(\alpha) \neq 0$$

$$\psi(x) \approx \psi(\alpha) + \frac{1}{2} \psi''(\alpha) x^2$$

$$\psi(k) = \psi(\alpha) + \frac{1}{3!} \psi'''(\alpha) x^3$$

$$I(k) \approx \frac{1}{k^{3/2}} + O\left(\frac{1}{k^{5/2}}\right)$$

Asymptotic approx. for well pot.

$$\Phi(x, y, t) = -\frac{\sqrt{g}}{\pi g} \int_0^\infty e^{\frac{\sigma^2}{g} y} \cos \sigma x \sin(\sqrt{g} \sigma t) \frac{d\sigma}{\sqrt{g}}$$

$$\int_0^\infty \psi(z) e^{i k \phi(z)} dz$$

let $\sigma = \sqrt{g} s$, $2\sigma d\sigma = g ds$, $m = \frac{\sigma^2}{g}$

p161

$$\Phi(x, y, t) = -\frac{1}{\rho \pi} \int_0^\infty e^{\frac{\sigma^2}{g} y} \sin\left(\frac{\sigma^2 x}{g} + \sigma t\right) d\sigma$$

$$- \int_0^\infty e^{\frac{\sigma^2}{g} y} \sin\left(\frac{\sigma^2 x}{g} - \sigma t\right) d\sigma$$

let $h = \frac{2x^2}{gx}$, $s = \frac{-2x}{g} \sigma$

$$\Phi = \frac{-2t}{2\pi \rho x} \left\{ \int_0^\infty e^{m y} e^{i k (s^2 + 2s)} ds - \int_0^\infty e^{m y} e^{i k (s^2 - 2s)} ds \right\}$$

$$\varphi(s) = 5^2 \pm 25$$

$$\varphi'(s) = 23 \pm 2$$

$s=1$ (only pt. of stationary phase)

$$\varphi''(s) = 2$$

$$\eta(x,t) \approx \frac{g^{1/2} t}{\sqrt{\pi} x^{3/2}} \cos\left(\frac{gt^2}{4x} - \pi/4\right)$$

$$\eta(r,t) = -\frac{1}{2\pi g \sqrt{g}} \int_0^\infty e^{sy} J_0(sr) (\sin \sqrt{g} s t) \cdot s^{3/2} ds$$

$$J_0(sr) = \frac{2}{\pi} \int_0^{\pi/2} \cos(sr \cos \theta) d\theta$$

$$= \operatorname{Re} \frac{2}{\pi} \int_0^{\pi/2} e^{i sr \cos \theta} d\theta$$

$$\eta(r,t) \approx -\frac{gt^3}{2^{3/2} \pi g^{1/4}} \sin \frac{gt^2}{4x}$$

$$\frac{gt^2}{4x}$$

p166

phase

$$k = \frac{gt^2}{4x}$$

propagation of
phases

$$\frac{gt^2}{4x} = k = \text{const.}$$

$$x = \frac{gt^2}{4k}$$

phase moving with an acceleration.

$$\frac{dx}{dt} = \frac{gt}{2k}$$

$$\varphi = \frac{gt^2}{4x}$$

x_0, t_0

$$\varphi = \frac{gt_0^2}{4x_0} \left(1 + 2 \frac{\Delta t}{t_0} + \left(\frac{\Delta t}{t_0} \right)^2 \right)$$

$\Delta t/t_0$ small

p169

$$\Delta\phi = \phi - \phi_0 \approx \frac{2\pi t}{t_0} \left(\frac{g t_0^2}{4x_0} \right)$$

T period

$$\Delta\phi = 2\pi, \quad T \approx \frac{4\pi x_0}{g t_0}$$

valid if

$\frac{T}{t_0}$ is small

$$T/t_0 = \frac{4\pi x_0}{g t_0^2} \text{ has to be } \text{small in the region of validity}$$

$$\lambda \approx \frac{8\pi x_0^2}{g t_0^2} \text{ local wave length}$$

$$\frac{\lambda_0}{x_0} \text{ is small}$$

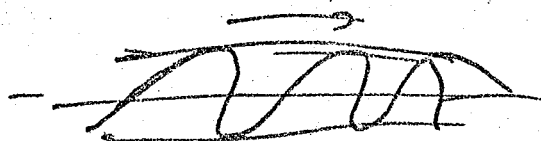
$$\frac{g t^2}{4x} \approx k$$

$$\frac{dx}{dt} = \frac{g t}{2xk} = \frac{2x}{t}$$

$$\approx \sqrt{\frac{g t}{2\pi}}$$

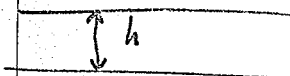
what's the propagation speed of λ ?
how much must you move along to observe the same wave length

$$x \approx \frac{1}{2} \sqrt{\frac{g}{2\pi}} t$$



crest moves half as fast as the wave

group vel. is $\frac{1}{2} \sqrt{\frac{g\lambda}{2\pi}}$



$$c^2 = \frac{g\lambda}{2\pi} \tanh \frac{2\pi h}{\lambda} \quad (1)$$

$$\Phi_1 = A \sin(m_1 x - \sigma_1 t)$$

$$\Phi_2 = A \sin[(m_1 + \delta m)x - (\sigma_1 + \delta \sigma)t]$$

$$\Phi = \Phi_1 + \Phi_2 = 2A \cos \frac{1}{2}(\delta m x - \delta \sigma t) \sin \left[(m_1 + \frac{\delta m}{2})x - (\sigma_1 + \frac{\delta \sigma}{2})t \right]$$



$$= B \sin(m' x - \sigma' t)$$

Sommerfeld $\int_{m_0 - \epsilon}^{m_0 + \epsilon} A(m) e^{i(m_0 x - \sigma_0 t)} dm$

$$\sigma = \sigma(m)$$

$$\Phi = c e^{i(m_0 x - \sigma_0 t)}$$

$$\frac{\sigma - \sigma_0}{m - m_0}$$

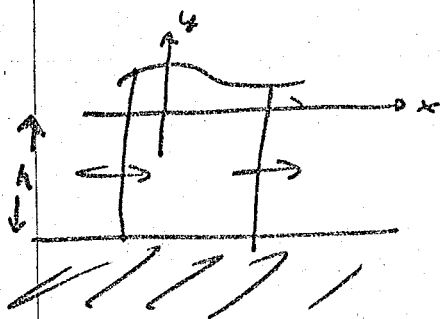
modulated $C(x, t) = \int_{m_0 - \epsilon}^{m_0 + \epsilon} A(m) e^{i(m - m_0)x - (\sigma - \sigma_0)t} dm$

capillary waves (short wave length, large curvature)



velocity of energy (flow of energy)





average
flux

$$F_{av} = \frac{A^2 g \sigma m h}{4\pi} \left(1 + \frac{\sin 2mh}{2mh} \right)$$

$$\sigma^2 = gm \tanh mh$$

$$m = \frac{2\pi}{\lambda}$$

$$F_{av} = U \frac{A^2 g \sigma^2}{2\pi} \cosh^2 mh$$

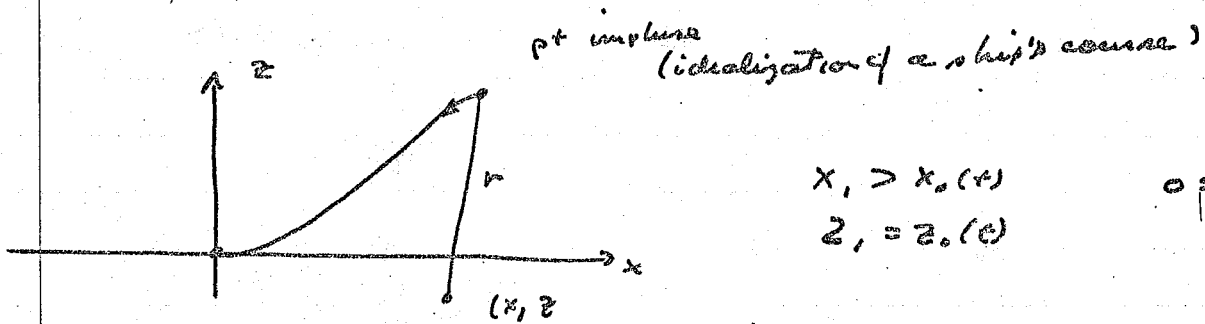
$$U = \frac{1}{2} c \left(1 + \frac{2mh}{\sinh 2mh} \right)$$

v. energy

$$\frac{E_{ave}}{unit} = \frac{4^2 \rho \sigma^2}{2\pi} \cosh^2 mh, \quad c \text{ given by (1)}$$

group velocity

p221



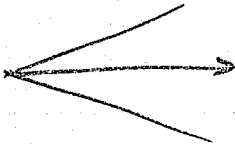
$$x_1 > x_0(t)$$

$$z_1 = z_0(t)$$

$$0 \leq t \leq T$$

$$r^2 = (x - x_1)^2 + (z - z_1)^2 \quad \left| \sum \right| \xrightarrow{\text{lin}} \int_0^T u(r, t) dt$$

Simplified case, ship moves along x-axis



$$\text{let } \sigma^2 = g m$$

$$m = \frac{g \tau^2}{4 \pi^2} \tau^2$$

p222

$$\int_0^\pi \int_0^\pi \int_0^{\pi/2} \dots \int e^{i K \left(\frac{25 - 5^2 \cos \Delta}{2} \right)} + e^{i K(t)}$$

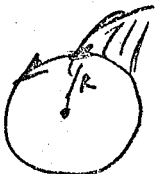
$$l = \frac{1}{K}$$

$$\vec{c} = (\vec{x}_1, \vec{z}_1)$$

$$K = \frac{g R}{4 c^2}$$

Wave pattern comes out very accurately

Bromberg



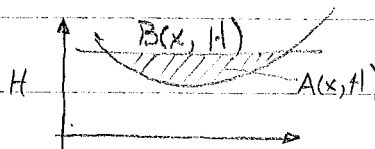
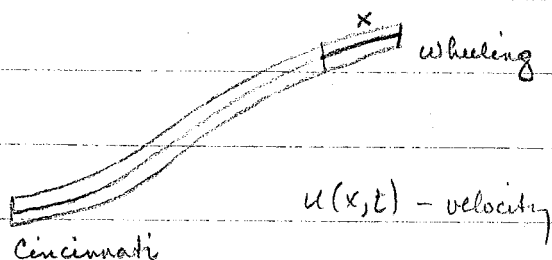
destroyer

R = ?

Speed = ?

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Shock Waves



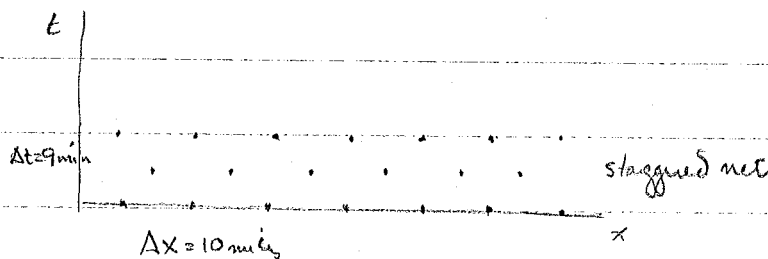
$$BH_t + \underbrace{(Au)_x}_{\text{cross sectional area of river}} = \underbrace{q(x,t)}_{\text{source term}}$$

Dynamical Condition $u_t + uu_x + gH_x = - \underbrace{F(x,H)}_{\text{force of gravity}} \underbrace{u|u|}_{\text{resistance term}}$

propagation speed $c = \sqrt{g \frac{A}{B}}$

characteristics $\frac{dx}{dt} = u \pm c$

Numerical Scheme



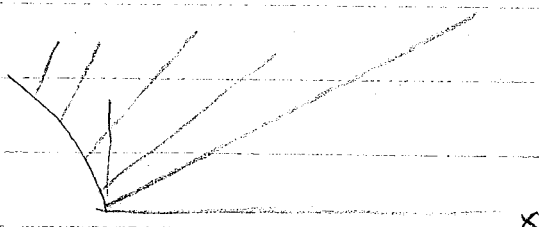
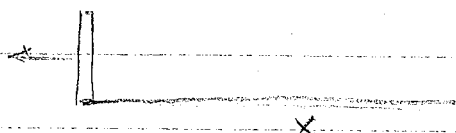
Coefficients are values taken at M

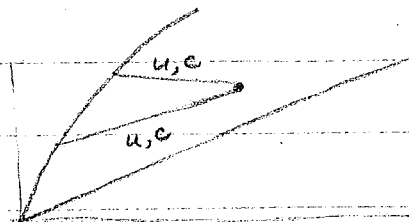
$$U_{mid} = \frac{U_L + U_R}{2}$$

$$U_t = \frac{U_p - U_w}{\Delta t}$$

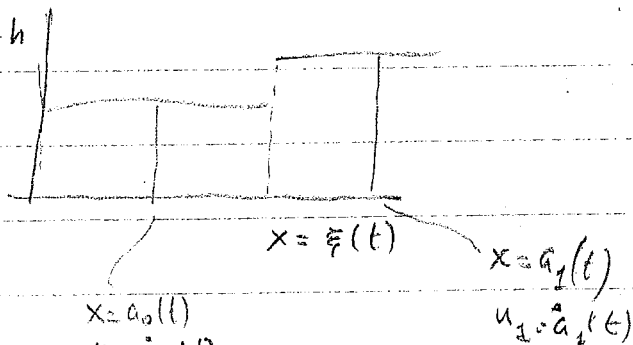
B, A, I contain unknown H.

Consider tank with movable end





discontinuity in $\eta+h$



Mass conservation $\frac{d}{dt} \int_{a_0(t)}^{a_1(t)} \rho(\eta+h) dx = 0$

Mom $\frac{d}{dt} \int_{a_0(t)}^{a_1(t)} \rho(\eta+h) u dx = \int_{-h}^{\eta_0} p_0 dy - \int_{-h}^{\eta_1} p_1 dy$

$$= \frac{1}{2} g \rho (\eta_0 + h)^2 - \frac{1}{2} g \rho (\eta_1 + h)^2$$

all integrals have form $\int_{a_0(t)}^{a_1(t)}$

$$I \int_{a_0(t)}^{a_1(t)} \psi(x, t) dx$$

$$\frac{dI}{dt} = \frac{d}{dt} \int_{a_0(t)}^{\xi(t)} \psi dx + \frac{d}{dt} \int_{\xi(t)}^{a_1(t)} \psi dx$$

use Leibniz Rule $\frac{d}{dt} \int_{a_0}^{\xi} = \int_{a_0}^{\xi} \frac{\partial}{\partial t} + \psi(\xi, t) \dot{\xi}(t) - \psi(a_0, t) \dot{a}_0$

Li $\frac{dI}{dt} = \psi_1(u_1, -\xi) - \psi_0(u_0, -\xi)$
 $a_1 \rightarrow a_0$
 $= \psi_1 v_1 - \psi_0 v_0$

$$\rho(\eta, h) v_1 - \rho(\eta_0 + h) v_0 = 0$$

$$\rho(\eta, h) u_1 v_1 - \rho(\eta_0 + h) u_0 v_0 = \frac{1}{2} \rho g (\eta_0 + h)^2 + \frac{1}{2} \rho g (\eta_1 + h)$$

$$\bar{\rho}_1 v_1 - \bar{\rho}_0 v_0 = 0$$

$$\bar{\rho}_1 u_1 v_1 - \bar{\rho}_0 u_0 v_0 = \bar{p}_0 - \bar{p}_1; \bar{p} = \frac{g}{2\rho} \bar{p}^2$$

$$\text{define } \bar{\rho}_1 v_1 = \bar{\rho}_0 v_0 = m$$

$$u_1 - u_0 = v_1 - v_0 \quad \text{since } v_1 - u_1 = -\dot{\eta}_0$$

$$m(v_1 - v_0) = \bar{p}_0 - \bar{p}_1$$

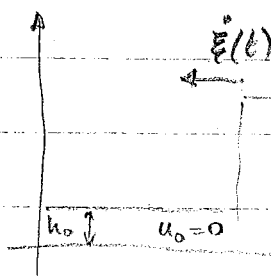
$$\frac{dE}{dt} = \left\{ \int_{\eta_0}^{\eta_1} \left[\rho(\eta + h) \frac{u^2}{2} + \frac{\rho g}{2} (\eta + h)^2 \right] dx + \int_{-h}^{\eta_1} \bar{\rho}_1 u_1 dy - \int_{-h}^{\eta_0} \bar{\rho}_0 u_0 dy \right\}$$

$$\frac{dE}{dt} = \frac{1}{2} \bar{\rho}_1 u_1^2 v_1 - \frac{1}{2} \bar{\rho}_0 u_0^2 v_0 + \bar{p}_1 v_1 - \bar{p}_0 v_0 + \bar{p}_1 u_1 - \bar{p}_0 u_0$$

$$\dot{E} [\bar{\rho}_1 u_1 v_1 - \bar{\rho}_0 u_0 v_0] - \dot{E} (\bar{p}_0 - \bar{p}_1) = 0$$

$$\frac{dE}{dt} = \frac{mg}{\bar{p}} \frac{(\bar{p}_0 - \bar{p}_1)^3}{4\bar{p}_1 \bar{p}_0}$$

$$\frac{dE}{dt} < 0 \quad \text{energy is lost}$$



$$m \neq 0 \quad m > 0$$

$$v_1, v_0 > 0$$

$$\frac{dE}{dt} < 0 \Rightarrow \bar{p}_0 < \bar{p}_1 \Rightarrow v_1 > v_0$$

$$v_1 v_0 = \frac{\bar{p}_0 - \bar{p}_1}{\bar{p}_0 - \bar{p}_1}$$

$$u_0 = 0$$

$$v_0 = -\dot{\xi}$$

$$v_1 = u_1 - \dot{\xi}$$

$$-\dot{\xi}(u_1 - \dot{\xi}) = \left(\frac{g\bar{p}_0^2}{2\rho} - \frac{g\bar{p}_1^2}{2\rho} \right) / (\bar{p}_0 - \bar{p}_1)$$

$$= \frac{g}{2\rho} (\bar{p}_0 + \bar{p}_1)$$

$$\begin{cases} -\dot{\xi}(u_1 - \dot{\xi}) = \frac{g}{2\rho} (\bar{p}_0 + \bar{p}_1) \\ \bar{p}_1 (u_1 - \dot{\xi}) = -\bar{p}_0 \dot{\xi} \end{cases} \rightarrow \dot{\xi}^2 = \frac{g\bar{p}_1}{2\rho} \left(1 + \frac{\bar{p}_1}{\bar{p}_0} \right)$$

$$\text{or } -\dot{\xi}(u_1 - \dot{\xi}) = \frac{g\bar{p}_1}{2\rho} \left(1 - \frac{u_1 - \dot{\xi}}{\dot{\xi}} \right)$$

unknown $\dot{\xi}, u_1, \bar{p}_0, \bar{p}_1$

$$\text{Case 1: } \bar{p}_1, \bar{p}_0 \text{ given} \Rightarrow \dot{\xi}^2 = f(\bar{p}_1, \bar{p}_0) \text{ gives } \dot{\xi}$$

$$\text{put into } -\dot{\xi}(u_1 - \dot{\xi}) = f(\bar{p}_0, \bar{p}_1) \Rightarrow u_1$$

$$\dot{\xi} = -\sqrt{g \frac{h_1}{h_0} \left(\frac{h_1 + h_0}{2} \right)} \quad h_1 = \eta_1 + h \quad h_0 = \eta_0 + h$$

if $h_1 > h_0$

$$\text{then } |\dot{\xi}| = \sqrt{g \frac{h_1}{h_0} \left(\frac{h_1 + h_0}{2} \right)} > \sqrt{gh_0}$$

$$v_1 = u_1 - \dot{\xi} \quad u_1 = \dot{\xi} \left(1 - \frac{\bar{p}_0}{\bar{p}_1} \right) \Rightarrow v_1 = -\frac{h_0}{h_1} \dot{\xi}$$

$$v_1 = \sqrt{g \frac{h_0}{h_1} \left(\frac{h_1 + h_0}{2} \right)} \quad |v_1| < \sqrt{gh_1}$$

5/8/74

2 wks Arthur Peters
from today "Weierhoff Technique"
C. (Integral equations)
tech. for linear prob.

Term paper (just summarizing, if you want or can expand)
suggested topics

1. Cox, A., Decaying Shocks, Comm. P.A. Math., Vol. 1, Section 7-1, 7-2, 1948



rarefaction waves go forward.
what happens?

2. John, F. Motion of Floating Bodies I, $\textcircled{H}$ CPAM, 1950

recommended
This one

$$u_{xx} + u_{yy} - c u_{tt} = 0$$

$$u = e^{i\omega t} w(x, y)$$

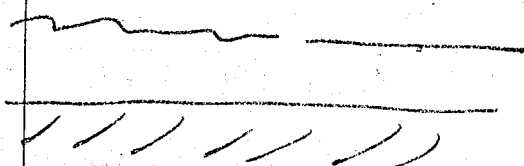
$$w_{xx} + w_{yy} + c^2 w^2 w = 0$$

how does this behave at ∞

reduced wave eq
(elliptic)

3. Heiss, A.E., Waves over channel of finite depth, Am. Journal Math., vol. 70, 1948 1-1, 1-2

deals with Weierhoff Technique



wave comes in from ∞
into dock

4. Two dimensional study supersonic (supersonic) flow.

Anneli Lax CAM 1948 Vol 1 247-257

RO Friedrichs " 1948 Vol 1 211-245

Supercritical - Steady Flow

(in gas dynamics version very interesting)

$$C = \sqrt{g h}$$

simple waves exist in same sense as in 1-dim
hyperbolic problem

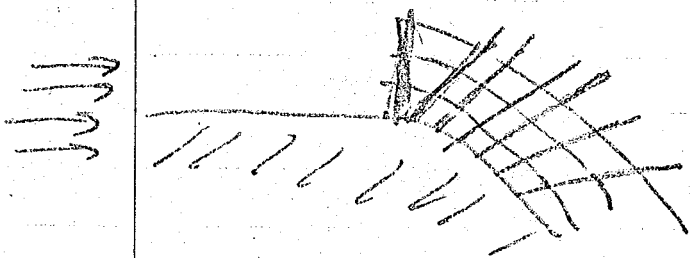
real characteristics

same theorem hold

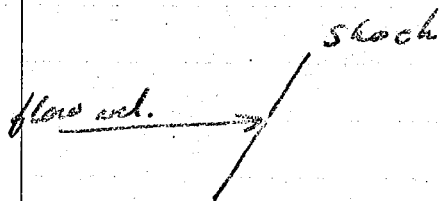
if γ a st. charac.

which is constant ~~for a region~~

Then that is embedded in a whole
field of st. charac.

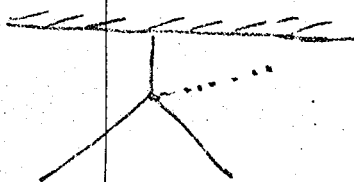


Shock waves develop here



Mach wave

need contact discontinuity

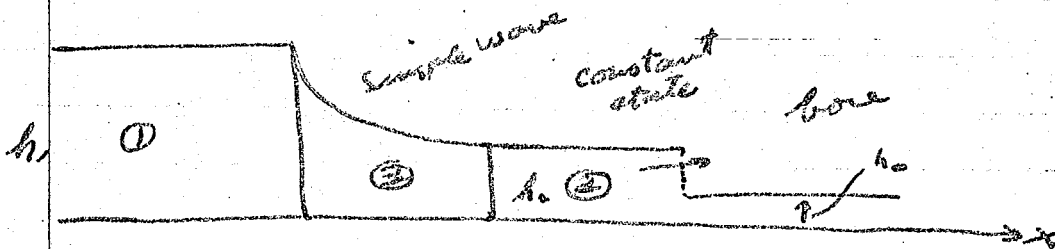
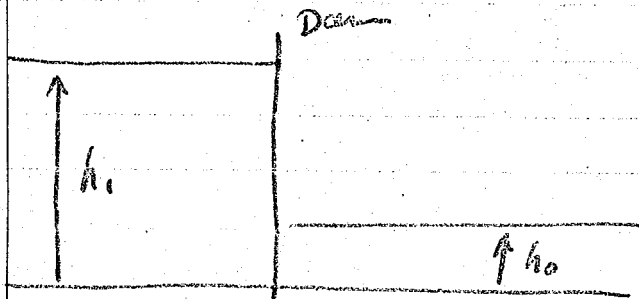


$$\frac{h_1}{h_0} = 10, \quad h_1 - h_2 = 35(h_1 - h_0)$$

$$\bar{p} \sim \left(\frac{h_1}{h_0}\right)^3$$

Breaking of a dam:

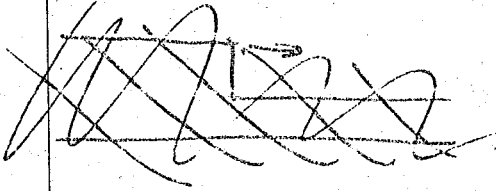
ps33



$$\frac{h_0}{h_1} = .176$$

$$h_2 \sim \frac{h_1}{3}$$

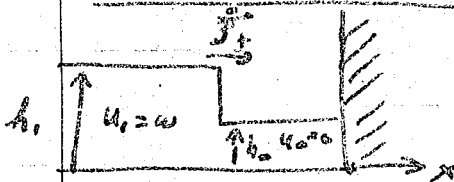
are when you have
only steady state & unif. depth & const velocity.



$$u = \frac{h}{c}$$

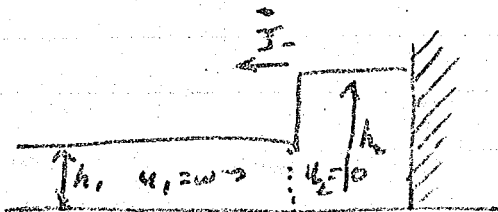
$$\frac{h^2}{c^2} + 2c^2 = h.$$

a)



what will happen when shock
comes up to wall & gets reflected?

b)

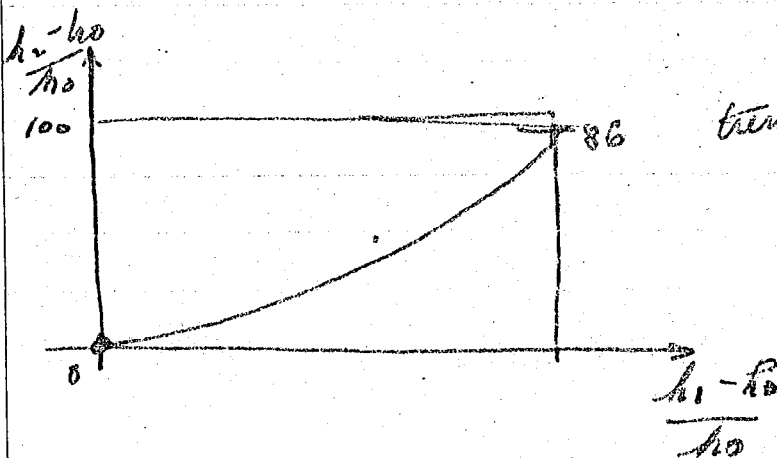


front & back
have interchanged
their roles

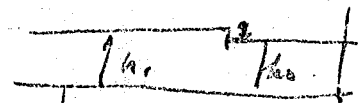
$$-\dot{s}(w - \dot{s}) = \frac{gh_0}{2} \left(1 - \frac{\dot{s}}{w - \dot{s}}\right)$$

case a) Choose $\dot{s}_+$, $h_1(w - \dot{s}_+) = -h_0 \dot{s}_+$

case b) Choose $\dot{s}_-$, $h_1(w - \dot{s}_-) = -h_0 \dot{s}_-$



tremendous inc.
in amplitude



Ex: $h_1 - h_0$ small, then $h_2 - h_0 \approx 2(h_1 - h_0)$

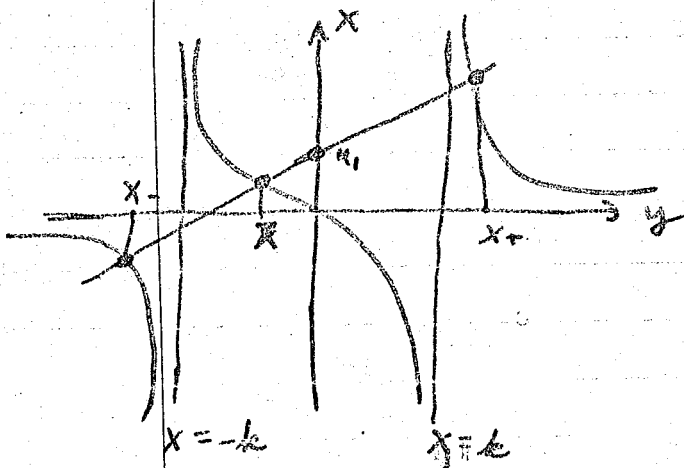
Eliminate $\bar{p}_0$ and

$$-\dot{s}(u, -\dot{s}) = \frac{g \bar{p}_1}{2\bar{p}} \left(1 - \frac{u_1 - \dot{s}}{\dot{s}}\right)$$

$$x = -\dot{s}, \quad y = u_1 - \dot{s}, \quad k^2 = \frac{g h_1}{2} = \frac{g \bar{p}_1}{2\bar{p}}$$

$$xy = k^2 \left(1 + \frac{y}{\dot{s}}\right)$$

$$y(\dot{s}^2 - k^2) = k^2 x \Rightarrow \begin{cases} y = \frac{k^2 x^2}{x^2 - k^2} \\ y = u_1 + x \end{cases}$$



x not physically significant
since $\dot{s} = -x$

$$\bar{p}_1 (u_1 + x) = \bar{p}_0 x$$

$$\bar{p}_1 y = \bar{p}_0 x$$

true only when

y and x have same sign

$\Rightarrow$ only 2 physically signif. roots.

$u_1 = 0 \Rightarrow$ no shock

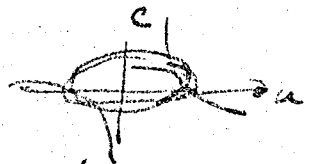
$\dot{s}_+, \dot{s}_-$ only roots useable

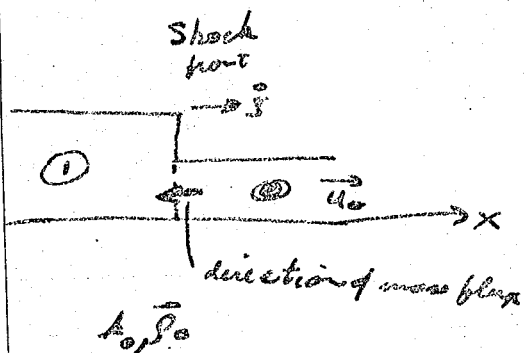
$$\begin{cases} u_t + u u_x + 2c c_x = 0 \\ 2c x + 2u c_x + c u_x = 0 \end{cases}$$

are there any steady state solutions

$$\begin{cases} u \frac{du}{dx} + 2c \frac{dc}{dx} = 0 \\ 2u \frac{dc}{dx} + c \frac{du}{dx} = 0 \end{cases}$$

$$\begin{cases} u^2 + 2c^2 = \text{const} \\ u c^2 = \text{const} \quad c^2 = \frac{\text{const}}{u} \end{cases}$$





$\dot{S}$ speed of propagation

$$\bar{p} = p(\eta + h)$$

$$\bar{p} = \frac{g}{2p} \bar{p}^2$$

adiabatic relation

relative flow velocity

$$V_1 = u - \dot{S}$$

$$V_0 = u_0 - \dot{S}$$

Shock conditions

$$\textcircled{1} \quad \bar{p}_1 V_1 = \bar{p}_0 V_0 = m$$

— cons. of mass

represents mass flux through shock

$$\textcircled{2} \quad \bar{p}_1 u_1 V_1 - \bar{p}_0 u_0 V_0 = \bar{p}_0 - \bar{p}_1$$

— cons. of momentum

$$\frac{dE}{dt} = \frac{mg}{p} \frac{(\bar{p}_0 - \bar{p}_1)^2}{4 \bar{p}_1 \bar{p}_0} < 0$$

not zero due to turbulence

$$m(V_1 - V_0) = \bar{p}_0 - \bar{p}_1$$

$$V_1 V_0 = \frac{\bar{p}_0 - \bar{p}_1}{\bar{p}_0 - \bar{p}_1}$$

$$\text{at } t: u_0 = 0, V_0 = -\dot{S}, V_1 = u_1 - \dot{S}$$

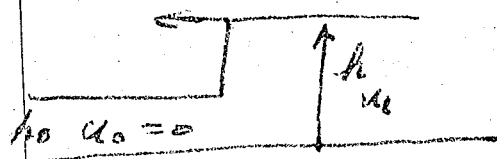
$$-\dot{S}(u_1 - \dot{S}) = \frac{g}{2p} (\bar{p}_0 + \bar{p}_1)$$

$$\bar{p}_1 (u_1 - \dot{S}) = -p_0 \dot{S} \quad \text{or} \quad u_1 = \dot{S} - \left(1 - \frac{\bar{p}_0}{\bar{p}_1}\right)$$

eliminate u_1 yields

$$\dot{S}^2 = \frac{2\bar{p}_1}{2p} \left(1 + \frac{\bar{p}_1}{\bar{p}_0}\right), \quad \dot{S} = -\sqrt{g h_0 \left(\frac{h_1 + h_0}{2}\right)}$$

~~At $t=0$~~



velocity of shock always less
velocity of wave let's behind

$$\dot{S} < \sqrt{gh}$$

7/22/74

Prof AS. Peters on Singular Integral Equations.

Wiener - Hopf

$$y \frac{\partial w(x, y)}{\partial x} + w(x, y) = \lambda \int_{-1}^1 w(x, y') dy' \equiv \lambda \varphi(x)$$

$$\text{subject to } w(0, y) = \begin{cases} I(y) & y > 0 \\ 0 & y < 0 \end{cases}$$

Arose in radiative equilibrium problem

Equivalent system

$$\varphi(x) + g(x) = \lambda \int_0^{\infty} K(x-\xi) \varphi(\xi) d\xi$$

Example

Suppose we have potential

$$\Delta \varphi = 0 \quad y < 0$$

$$\varphi_y(x, 0) = 0 \quad x < 0$$

$$\varphi_x - c \varphi_y = f(x) \quad x > 0, y = 0$$

Assume poisson representation we can show

$$\varphi_x(x, y) = \frac{1}{\pi} \int_{-\infty}^{\infty} \frac{(\xi - x) \varphi_y(\xi, 0) d\xi}{(\xi - x)^2 + y^2}$$

use bdy cond & let $y \rightarrow 0$ with $x > 0$

$$\varphi_x(x, y) = \frac{1}{\pi} \int_0^{\infty}$$

$$\varphi_x(x, 0) = \frac{1}{\pi} \int_0^{\infty} \frac{\varphi_y(\xi, 0) d\xi}{\xi - x}$$

using bdy cond $x > 0, y = 0$

$$c \varphi_y(x, 0) + f(x) = \frac{1}{\pi} \int_0^{\infty} \frac{\varphi_y(\xi, 0) d\xi}{\xi - x}$$

This is a Wiener - Hopf equation singular in 2 ways infinite interval & infinite kernel

Example 2

dock at an infinite ocean

$$\varphi_y(x, 0) = 0$$

$$\varphi_y - c \varphi = f(x)$$

$$\Delta \varphi = 0$$

$$\Rightarrow x > 0$$

$$\varphi_x(x, 0) = \frac{1}{\pi} \int_0^{\infty} \frac{f(\xi) + c \varphi(\xi) d\xi}{\xi - x}$$

Wiener Hoff comes up when solving some PDE. in a domain D which has as ~~one~~^{part of} bdy the half line & data on the bdy.

Can circumvent WH if we are willing to deal with "barrier eqns"
we note that $K(t)$ must be defined $\forall$ values of its argument
 $\nexists x > 0$

$$\text{Define } \varphi(x) = 0 \quad x < 0$$

then

$$\varphi(x) + g(x) = \lambda \int_{-\infty}^{\infty} K(x-\xi) \varphi(\xi) d\xi$$

this is convol kernel in fourier transform.

LHS unknown for $x < 0 \therefore$

let for $x < 0$

$$\varphi(x) = \lambda \int_{-\infty}^{\infty} K(x-\xi) \varphi(\xi) d\xi$$

$$\text{define RH fourier transform } \Phi_+(\xi) = \int_0^{\infty} e^{i\xi x} \varphi(x) dx$$

$$\text{" " " " } G_+(\xi) = \int_0^{\infty} e^{i\xi x} g(x) dx$$

$$\text{define } K(\xi) = \int_{-\infty}^{\infty} e^{i\xi t} K(t) dt$$

$$\text{define LH transp } \Phi_-(\xi) = \int_{-\infty}^0 e^{i\xi x} \varphi(x) dx$$

$$\Phi_+(z) + \Psi_-(z) + G_+(z) = K(z) \Phi_+(z)$$

What conditions are satisfied by K, Φ assuming $\Im a > 0$.

The untransformed eqns are equiv to transformed eqn. Assume this holds for $\Im m \zeta = a$. $\therefore$ we have a fun eqn to be solved

$$[1 - K(\zeta)] \Phi_+(\zeta) + \Psi_-(\zeta) = -G_+(\zeta)$$

define
$$F(z) = \begin{cases} \Phi_+(z) & \Im m z > \Im m \zeta \\ -\Psi_-(z) & \Im m z < \Im m \zeta \end{cases}$$

Analytic in upper half plane & lower half plane & the above eq must hold on the line $\Im m \zeta = a$.

$[1 - K(\zeta)] F^+(\zeta) - F^-(\zeta) = -G_+(\zeta)$ barrier eq
find $F(z)$ which is sectionally holomorphic for which has transform properties.

$$f(z) = \int_{-\infty + ia}^{\infty + ia} \frac{h(w) dw}{w - z}$$

Representation for sectionally holomorphic.

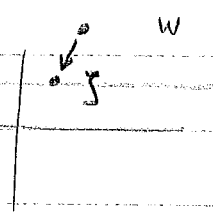
$$f^+(z) = \pi i h(\zeta) + \int_L \frac{h(w) dw}{w - z}$$

Plemelj

$$f^-(z) = -\pi i h(\zeta) + \int_L \frac{h(w) dw}{w - z}$$

$$[f^+ - f^-] = 2\pi i h(\zeta)$$

The barrier eq looks like this to get it in same form
Let $F(z) = F_1(z) F_2(z)$ choose



$$[1 - K(s)] F_1^+ F_2^+ - F_1^- F_2^- = -G_+$$

Choose $F_1 \Rightarrow [1 - K(s)] F_1^+(s) = F_1^-(s)$

F_1 soln to homog eq then

$$F_2^+ - F_2^- = \frac{-G_+(s)}{[1 - K(s)] F_1^+(s)}$$

now use polemely

$$F_2(z) = -\frac{1}{2\pi i} \int_L \frac{G_+(s)}{[1 - K(s)] F_1^+(s) (s-z)} ds + p(z)$$

rational

where $p_1^+(s) - p_1^-(s) = 0$

$p(z)$ can be picked so that the transf. properties of F are satisfied

$$\log F_1^+ - \log F_1^- = -\log[1 - K(s)]$$

$$\ln F_1(z) = -\frac{1}{2\pi i} \int_L \frac{\log[1 - K(s)] ds}{s-z}$$

Example: $\varphi(x) = \lambda \int_0^\infty e^{-|x-\xi|} \varphi(\xi) d\xi$ are these solns.

$$K(s) = \int_{-\infty}^\infty e^{ist} e^{-|t|} dt = \frac{2\lambda}{s^2 + 1} \quad -1 < s < 1$$

$$\therefore \left[1 - \frac{2\lambda}{s^2 + 1}\right] F^+(s) - F^-(s) = 0 \quad -1 < s < 1 \quad \text{exactly a strip}$$

$$[1 - K(s)] = \left[\frac{(s-s_1)(s-s_2)}{s^2 + 1} \right] \quad s_1 = \sqrt{2\lambda} - 1 \quad s_2 = -\sqrt{2\lambda} - 1$$

$$\log F^+ - \log F^- = \frac{\ln(s^2 + 1)}{(s-s_1)(s-s_2)}$$

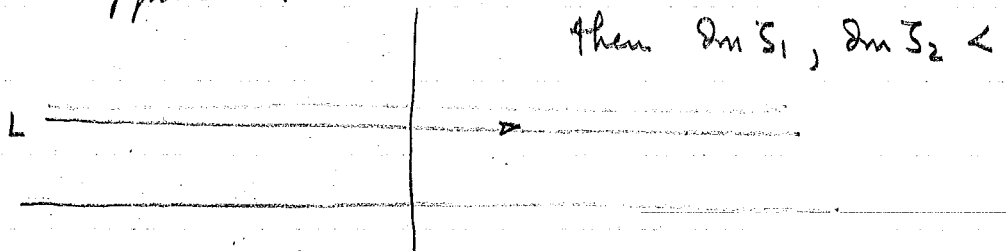
Take $\operatorname{Im} z = 1 - \epsilon$ $\epsilon > 0$

$$\ln F(z) = \frac{1}{2\pi i} \int_{\operatorname{Im} \zeta = 1 - \epsilon} \frac{1}{\zeta - z} \ln \frac{(\zeta^2 + 1)}{(\zeta - \zeta_1)(\zeta - \zeta_2)} d\zeta + p(z)$$

with $p^+ - p^- = 0$

This integral doesn't exist by Cauchy Residue th. But we will fix it. Suppose $\lambda > 0$

then $\operatorname{Im} \zeta_1, \operatorname{Im} \zeta_2 < 1$



we can rewrite (!)

$$\ln F(z) = \frac{1}{2\pi i} z \int \frac{1}{\zeta} \frac{1}{\zeta - z} \ln \frac{\zeta^2 + 1}{(\zeta - \zeta_1)(\zeta - \zeta_2)} d\zeta + p(x)$$

as $\zeta \rightarrow z$ then $\ln F(z) \rightarrow \frac{1}{2\pi i} \int \frac{1}{\zeta - z} d\zeta$

for $\operatorname{Im} z > \operatorname{Im} \zeta$

$$\Phi_+(z) = p(z) \frac{z+i}{(z-\zeta_1)(z-\zeta_2)} \quad p^+ - p^- = 0$$

Φ_+ vanishes for large $z \Rightarrow p(z) = \text{const}$

since $\frac{z+i}{(z-\zeta_1)(z-\zeta_2)} \rightarrow 0$ as $z \rightarrow \infty$ & has no sing in upper half plane.

for $\lambda \leq 0$ ζ_1 is in upper half plane

$$\Phi_+(z) = p(z) \frac{z+i}{z-\zeta_2}$$

$p(z)$ must be $\Rightarrow p^+ - p^- = 0$ & $p(z) \rightarrow 0$ as $z \rightarrow \infty$ $p(z)$ cannot have zero

$\Rightarrow p(z) = 0 \quad \therefore$ we have trivial sol.

Ex $e^{-x} = \int_0^{\infty} K_0(|x-\xi|) \varphi(\xi) d\xi$ Bessel fn of second kind

Sommerfeld diffraction probl

$$K(\zeta) = \int_{-\infty}^{\infty} e^{i\zeta t} K_0(|t|) dt = \frac{\pi}{\sqrt{1+\zeta^2}} \quad \text{analytic } -1 < \operatorname{Im} \zeta < 1$$

$$\frac{\pi}{\sqrt{\zeta+1}} F^+(\zeta) + F^-(\zeta) = G_+(\zeta) = \frac{1}{1-i\zeta}$$

$$\frac{\pi}{\sqrt{\zeta+1}} F^+(\zeta) + \sqrt{\zeta-1} F^-(\zeta) = \frac{\sqrt{\zeta-1}}{-i(\zeta+1)} \quad -1 < \operatorname{Im} \zeta < 1$$

$$\det F(\zeta) = \begin{cases} \frac{\sqrt{\zeta+1}}{\pi} F_2(\zeta) & \operatorname{Im} \zeta > \operatorname{Im} \zeta \\ -\frac{1}{\sqrt{\zeta-1}} F_2(\zeta) & \operatorname{Im} \zeta < \operatorname{Im} \zeta \end{cases}$$

$$F_2^+(\zeta) - F_2^-(\zeta) = \frac{i\sqrt{\zeta-1}}{\zeta+1}$$

$$F_2(\zeta) = \frac{1}{2\pi} \int_L \frac{\sqrt{\zeta-1}}{(\zeta-1)(\zeta-z)} d\zeta + p(z)$$

for $\text{Im } z > 0$

$$\Phi^+ = \frac{\sqrt{z}}{\pi \sqrt{1-iz}} + p(z) \frac{\sqrt{z+i}}{\pi} \quad \text{choose } p(z) = 0$$

inverse gives solution

At home $g(x) = \int_0^\infty \frac{\varphi(\xi)}{x-\xi} d\xi$ Cauchy Singular Integral Eqn.

$$K(\zeta) = \frac{\pi i |\zeta|}{\zeta} \quad \text{Im } \zeta = 0 \quad \text{must be}$$

$$\varphi(\lambda) = \frac{1}{\pi^2} \sqrt{\lambda} \int_0^\infty \frac{g(x) dx}{\sqrt{x}(\lambda-x)} + \frac{C}{\sqrt{\lambda}} \quad \text{Most general}$$

A problem to show equivalence of W-H to go straight to barrier eqn.

$$\varphi(x, 1) = 0$$

$$\Delta \varphi = 0 \quad 0 < y < 1$$

$$\varphi_y(x, 0) = 0 \quad \varphi(x, 0) = 1$$

$$\Phi(\zeta, y) = \int_{-\infty}^{\infty} e^{i\zeta x} \varphi(x, y) dx$$

$$\Phi_{yy}(\zeta, y) - \zeta^2 \Phi = 0 \quad \Phi(\zeta, 1) = 0$$

$$\Phi(\zeta, y) = A(\zeta) \sinh \zeta(y-1)$$

$$\Phi_y(z, y) = \zeta A(\zeta) \coth \zeta (y-1)$$

$$\lim_{\zeta \rightarrow 0} A(\zeta) \Rightarrow$$

$$\Phi_y = \zeta \coth \zeta (y-1) \Phi(z, y)$$

$$\text{let } y \rightarrow 0 \quad \Phi_y(z, 0_+) = -\zeta \coth \zeta \Phi(z, 0_+)$$

$$\Phi_y = 0 \text{ for } x < 0$$

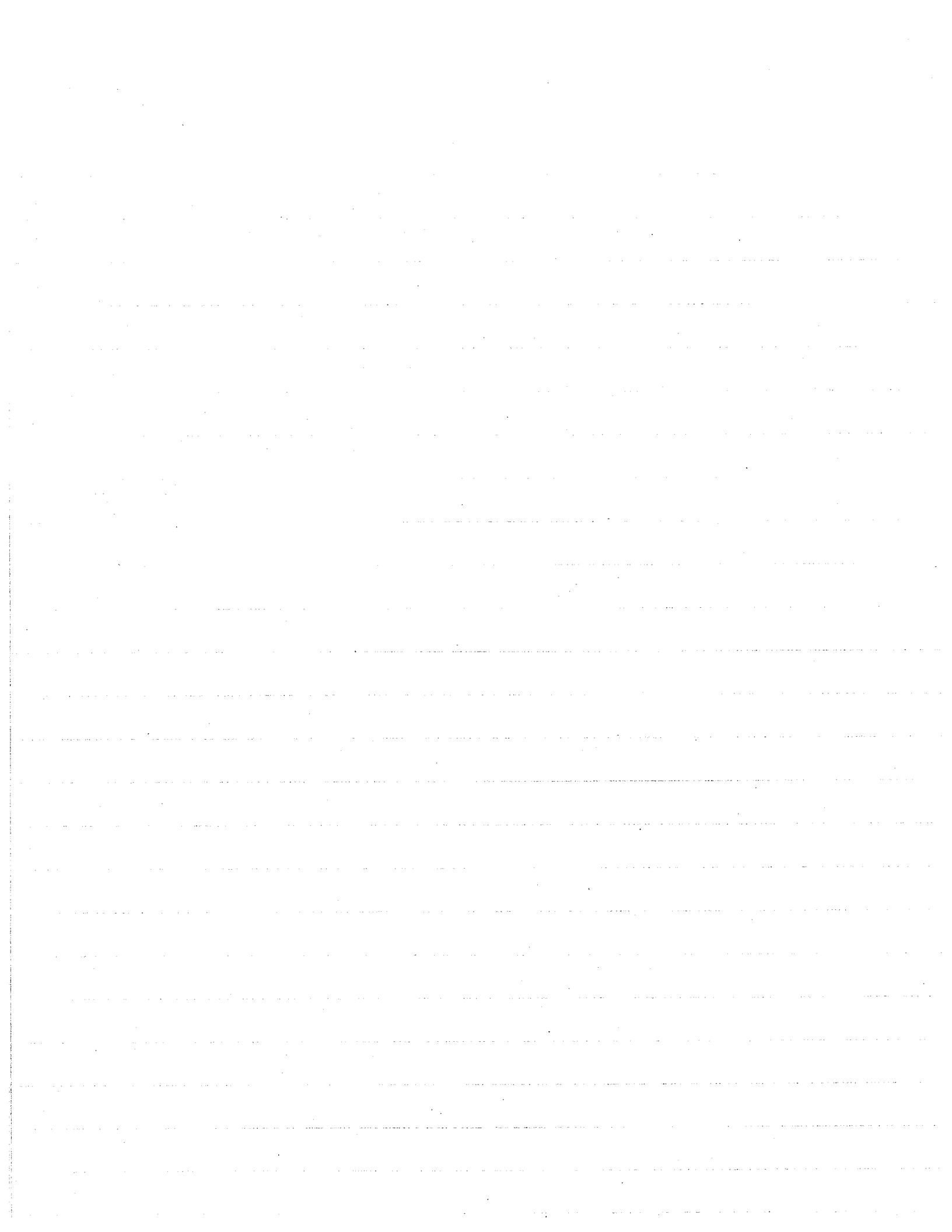
$$\int_0^\infty e^{i\zeta x} \varphi_y(x, 0) dx = -\zeta \coth \zeta \left[\int_{-\infty}^0 e^{i\zeta x} \varphi(x, 0) dx + \int_0^\infty e^{i\zeta x} \varphi(x, 0) dx \right]$$

$$\Phi_y^+ = -\zeta \coth \zeta \left\{ \Phi_- + \frac{1}{i\zeta} \right\}$$

$$\text{or } \frac{\tanh \zeta}{\zeta} \int_0^\infty e^{i\zeta x} \varphi_y(x, 0) dx + \int_{-\infty}^0 e^{i\zeta x} \varphi(x, 0) dx = \frac{1}{i\zeta}$$

$$F(z) = \begin{cases} \int_0^\infty e^{izx} \varphi_y(x, 0) dx & \operatorname{Im} z > \operatorname{Im} \zeta = \epsilon \\ \int_{-\infty}^0 e^{izx} \varphi(x, 0) dx & \end{cases}$$

$$\text{Bernier eq } \frac{\tanh \zeta}{\zeta} F^+ + F^- = \frac{1}{i\zeta}$$



Decaying Shocks by Anneli Lax

In a paper by K.O. Friedrichs "Formation & Decay of Shock Waves" an approximate method was developed to determine the behavior of decaying shock waves in a gas. Prof. Lax tests this method on a problem in surface waves in shallow water; this approximate method was then compared to a finite difference scheme, applicable to shocks of any strength, limited only by the mesh size of the scheme.

The Problem

A bore of constant speed U moved into still water of uniform depth h_0 . Behind the bore the water moves with a velocity u_2 and has a uniform height h_1 . This part in itself can be solved by using the shock jump conditions of gas dynamics. Now suppose that at time $t=0$ a wall is suddenly inserted into the moving water at position $x=0$. This creates a depression wave which will catch up to the bore, since its velocity is greater than the bore velocity, and will start to decay the bore. The problem is to find the position, velocity and strength as functions of time of the bore. Because the bores weaken a combination finite difference and approximate method is used, i.e. the finite difference is used until the bore is weakened and the continuation of the solution is given by the Friedrichs' approximation.

1. The Solution - First the bore jump conditions

Looking at the governing equations and non dimensionalizing

the velocities with respect $\sqrt{gh_0}$, the distance with respect to the distance between the wall and the bore x_0 , and the time with respect to $x_0/\sqrt{gh_0}$ and defining $c^2 = h/h_0$, we obtain equations similar to the shock equations in gas dynamics, viz

$$\tilde{p}_0 v_0 = \tilde{p}_1 v_1 \quad : \quad \tilde{p} = \rho g h \quad (1a, b)$$

$$\tilde{p}_0 + \tilde{p}_0 v_0^2 = \tilde{p}_1 + \tilde{p}_1 v_1^2 \quad : \quad p = (g/2) \rho h^2$$

thus defining the adiabatic equation of state not involving entropy:

$$p = \tilde{p}^2 \frac{1}{2\rho g} \quad (2)$$

For a barotropic gas it is noted that

$$p = K \rho^\gamma \quad \gamma = 1.4 \quad ;$$

however in our case $\gamma = 2$. With the aid of (1) and (2) the shock velocity U and particle velocity u behind the shock can be found in terms of c behind the shock.

II. The Solution - Now add the depression wave

We note that a depression wave in fluids is like a rarefaction wave in gas dynamics. Hence $\exists$ a centered simple wave region having straight line characteristics emanating from the origin. The depression wave moves with sound speed c_1 into a medium whose velocity is u_1 . The straight line characteristics C^+ are given by $\frac{dx}{dt} = u_1 + c_1$ and the simple wave solution is $\frac{x}{t} = u_1 + c_1$. Since $c_1 > U$ the depression wave catches up to the bore at $x = x^* = l$ & $t = t^* = \kappa$; κ and l can be found as follows:

Since we have non-dimensionalized, at $t=0$ the bore is at $x=1$. Hence in time $t=\kappa$ the bore is at $x=x^*=l=U\kappa+1$. In time κ the depression wave has moved a distance l ; therefore from $\frac{x}{t} = u_1 + c_1$, $\frac{l}{\kappa} = u_1 + c_1$,

These two equations can be solved for l & κ . From these coordinates a new coordinate system $\bar{x}, \bar{t}$ is defined so that $\bar{x} = x - l$, $\bar{t} = t - \kappa$. (It is noted that the reflection of the depression wave from the bore is ignored since this effect enters in terms of higher than second order.) Since the only common intersection point in a centered simple wave region of the C^+ characteristics is the origin then for $(\bar{x}, \bar{t}) = (\xi, 0)$

$$\frac{\bar{x}}{\bar{t}} = u + c \Rightarrow \frac{\xi + l}{\kappa} = u(\xi) + c(\xi) \quad (3)$$

It is assumed that the Riemann invariant s is constant across the bore which is consistent with the approximate theory, since changes in s appear in third order or higher terms; thus for a depression wave, on the C^- characteristic

$$\frac{u}{2} - \frac{c}{\gamma - 1} = s \quad (4)$$

and since $\gamma = 2$ we have

$$u - 2c = 2s. \quad (4')$$

Across the bore: $h = h_0$, $u = 0 \Rightarrow c = 1$ $u = 0 \therefore$

$$u - 2c = -2. \quad (5)$$

Behind the bore $u - 2c = u_1 - 2c_1$ (6)

Using (3) and (6) and (5)

$$\begin{aligned} u(\xi) &= \frac{2}{3} \left[\frac{\xi + l}{\kappa} - 1 \right] \\ c(\xi) &= \frac{1}{3} \left[\frac{\xi + l}{\kappa} + 2 \right]. \end{aligned} \quad (7a, b)$$

The bore curve $U = f[\bar{x}(\xi), \bar{t}(\xi)]$ can be found by obtaining a differential equation for $\bar{t}(\xi)$ and solving it. It is known that behind the shock

$$\frac{\bar{x}}{\bar{t}} = \frac{\bar{x} - \xi}{\bar{t}} = u(\xi) + c(\xi); \quad (8)$$

thus by differentiating (8) with respect to ξ the following is obtained

$$\bar{\xi}_{\xi} [U - c(\xi) - u(\xi)] = \bar{\xi} [u'(\xi) + c'(\xi)] + 1 \quad (9)$$

using $\bar{x}_{\xi} = U$.

The homogeneous solution is

$$\ln \bar{\xi}_h(\xi) = \int_0^{\xi} \frac{u'+c'}{U-(c+u)} d\check{\xi} = I(\xi)$$

and the particular solution is

$$\bar{\xi}_p(\xi) = e^{I(\xi)} \int_0^{\xi} \frac{e^{-I(\check{\xi})}}{U-(u+c)} d\check{\xi} \quad (10)$$

From (8)

$$\bar{x}(\xi) = \xi + (u+c) \bar{\xi}(\xi) \quad (11)$$

It is found that only part of the depression wave interacts with the bore; this part of the wave has height greater than h_0 , i.e. that region in the x, t space whose characteristics satisfy $x/t = u+c \geq 1$. Hence if we define $\sigma(\xi) = u+c-1$ a simpler expression for $\bar{\xi}(\xi)$ can be obtained when u & U are written as functions of $\sigma(\xi)$. In doing so it is seen that the forward moving part decays the bore for an infinitely long time, i.e. $(\xi+l)/\kappa = 1$.

If the wall moves with a given velocity $u_1 > w > u_1 - 2c_1$, the depression wave is narrowed, since the last characteristic has particle velocity $u=w$, and a constant state appears to the left of the variable zone. If $w \leq 0$ the depression wave is still wide enough to include that part which decays the shock. If $0 < w < u_1$, then $\exists$ a finite time period in which the interaction causes the bore to decay from one constant state to another. From the results shown the height of the

bore decreases rapidly at first and then approaches its asymptotic value slowly.

Solution by Finite Differences

The governing equations are one dimensional hyperbolic equations

$$u_t + u u_x + 2 c c_x = 0 \quad (12a, b)$$

$$2 c_t + c u_x + 2 u c_x = 0$$

which give the following:

$$\left. \begin{aligned} X_\alpha &= (u+c) t_\alpha \\ u+2c &= \text{const} \end{aligned} \right\} \text{ along } C^+ \text{ characteristics} \quad (13a, b)$$

$$\left. \begin{aligned} X_\beta &= (u-c) t_\beta \\ u-2c &= \text{const} \end{aligned} \right\} \text{ along } C^- \text{ characteristics} \quad (14a, b)$$

We replace 13a, 14a by

$$\Delta_\alpha x = (\tilde{u} + \tilde{c}) \Delta_\alpha t \quad \text{along } \beta = \text{const } (C^+) \quad (13a')$$

$$\Delta_\beta x = (\bar{u} - \bar{c}) \Delta_\beta t \quad \text{along } \alpha = \text{const } (C^-) \quad (14a')$$

$$\text{where } \Delta_\alpha () = ()_{i,k} - ()_{i-1,k} \quad \tilde{()} = [()_{i,k} + ()_{i-1,k}] / 2$$

$$\Delta_\beta () = ()_{i,k} - ()_{i,k-1} \quad \bar{()} = [()_{i,k} + ()_{i,k-1}] / 2$$

as noted in Courant & Friedrichs. From (13a') and (14a') we can solve for t_{ik} and x_{ik} .

Upon using (14b) in the equations for t_{ik} and x_{ik} , u_{ik} and c_{ik} can be obtained.

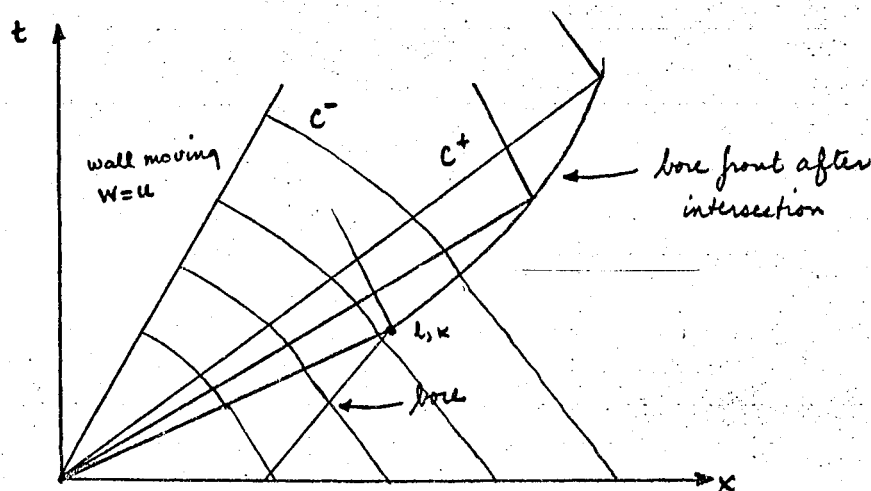
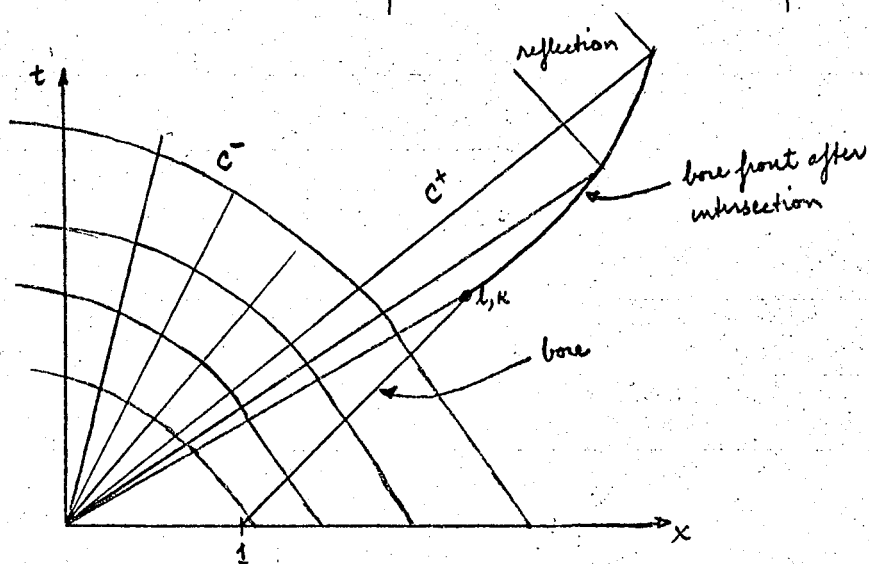
With these results and the jump conditions the velocity of the bore U_{ii} can be found from $U_{ii} = \sqrt{(c_{ii}^2 + 1)/2} \cdot c_{ii}$

Results & Conclusions

Prof. Lax shows results in x, t space for bore strength $h_1/h_0 = 1.8, 3.0$; we note that the agreement is very good, even for such a high value of h_1/h_0 .

between the finite difference and approximate method. For $x < 7$ both methods give coincident results on both graphs. However as x, t become large the bore curves $x = x(t)$ obtained no longer coincide, probably due the large mesh size of the Finite Difference scheme in that region. We note that Friedrichs' Method should become more accurate since the bore weakens when x, t become large.

Prof. Iax concludes that even higher than second order terms in the expansion of quantities across the bore differ little from what they would be in the case of a continuous wave and it seems reasonable to apply the approximate method in cases where the pressure ratios are high.



write up tomorrow

Tchaikovsky 1812 Overture
Sibelius #1 & Finlandia
Rimsky Korsakov - Russian
Easter Outh
Saint Saens Symphony #3

Dvorak #6, Carnival Outh
Mendelssohn - Hebrides
King Stephen - Beethoven
Jupiter - Planets - Holst
Beethoven #6, 7, 3

Decaying Shocks by Ansell Lap

In Paper by K.O. Friedrichs Formation & Decay of Shock Waves
an approx method was developed to determine the behavior of decaying
shock wave in a gas. To test this method on problem i surface
waves - shallow water this approx method was used as well
as a finite difference scheme applicable to shocks of any strength,
which is only limited by the mesh size.

The physical problem: A shock wave (or bore) moves with constant speed
 V into still water of unifo depth h_0 . The height of the water
behind the shock is h_1 , and its velocity is u_1 . @ $t=0$ a
vertical wall is suddenly inserted i to the moving water @ $x=0$
so that a rarefaction (or depression) wave is created which will
catch up with the shock, since such disturbances are propagated
with a speed greater than that of the shock; thus shock will decay

Since strong shocks weaken due to the above then the following
treatment of them is justified: a combination of Finite Difference
& Friedrichs approximation; F.D. until the shock is weakened
and then continuation of solution by K.O.F. Approx.

We non dimensionalize the velocities with respect to $(gh_0)^{1/2}$
the distance with respect to x_0 the distance between wall
& discont & the time w.r.t $\frac{x_0}{(gh_0)^{1/2}}$.
Problem is to find the strength of the shock as well as the
veloc of shock & its position as fun of time.

The approx method applicable to weak shocks uses the same shock equations as those for a gas

$$\tilde{p}_0 \tilde{u}_0 = \tilde{p}_1 \tilde{u}_1$$

$$\tilde{p}_0 + \tilde{\rho}_0 \tilde{u}_0^2 = \tilde{p}_1 + \tilde{\rho}_1 \tilde{u}_1^2$$

where the velos are relative to shock & $\tilde{p} = \rho g h$ $\rho = \text{density of water}$

$$\& \quad p = (g/2) \rho h^2 \quad \text{then} \quad p = \frac{1}{2\rho g} \tilde{p}^2$$

$$\frac{1}{2} \rho_0 h_0^2 + \rho_0 g h_0 \tilde{u}_0^2$$

This is the adiabatic eq of state which doesn't involve entropy & adiabatic exponent is 2.

From these eqs the shock velocity U & particles velocity behind the shock can be found in terms of c behind the shock.

Since the rarefaction wave moves with sound speed c_1 into medium ^{where} ~~with~~ velocity u_1 . The wave front is given by $\frac{dx}{dt} = u_1 + c_1$ [this is the characteristic] & catches up @ $x = l, t = K$. We assume that when the shock is at $x = 1$ ^{$t=0$} the wall is placed into the flow.

Hence @ distance $x^* = UK + 1 = l$; from the wall we get a centered simple wave $\frac{x}{t} = u_1 + c_1$ (rarefaction) $\therefore x^* = l = K(u_1 + c_1)$

$$\Rightarrow K = \frac{l}{u_1 + c_1 - U}$$

assume that $\bar{x}, \bar{t}$ are new coord from (l, K) $\therefore \bar{x} = x - l \quad \bar{t} = t - K$

We assume that reflection of the rarefaction wave from the shock is ignored since effect enters in terms of higher than second order.

Since centered simple wave $\exists$ a unique characteristic intersection emanating from the origin $(x, t) = (0, 0)$, with the negative $\bar{x}$ axis

hence from $\frac{x}{t} = u + c$ then if intersection is at $\bar{x}, \bar{t} = (\xi, 0)$
then

$$\frac{\xi + l}{k} = u(\xi) + c(\xi)$$

using the Riemann invariant across the rarefaction wave

$$\frac{u}{2} - \frac{c}{\gamma-1} = \frac{s}{2} \quad \text{and since } \gamma=2 \quad \text{we get } u - 2c = \xi$$

@ the front of the rarefaction $\xi = u_1 - 2c_1$

$$\therefore u(\xi) - 2c(\xi) = u_1 - 2c_1$$

$$\Rightarrow u(\xi) = \frac{1}{3} \left[2 \frac{\xi + l}{k} + (u_1 - 2c_1) \right]$$

$$c(\xi) = \frac{1}{3} \left[\frac{\xi + l}{k} - (u_1 - 2c_1) \right]$$

We assume that the Riemann invariant ξ is a const across the shock
which is consistent with the approx theory since changes in ξ appear
in third or higher order terms. $\therefore u - 2c = -2$

To find the shock curve $U = f[\bar{x}(\xi), \bar{t}(\xi)]$ a diff eq for $\bar{t}(\xi)$ is set
up and solved. Behind the shock

$$\frac{\bar{x} - \xi}{\bar{t}} = u(\xi) + c(\xi) \quad \Rightarrow \quad \bar{x}_{\xi} = (u+c)\bar{t}_{\xi} + \bar{t}(u'(\xi) + c'(\xi)) + 1$$

$$\hookrightarrow \bar{x}_{\bar{t}} \bar{t}_{\xi} = U \bar{t}_{\xi}$$

$$\text{or } \bar{t}_{\xi} [U - (c(\xi) + u(\xi))] = \bar{t} [u'(\xi) + c'(\xi)] + 1$$

$$\text{which has H.O.M Sol } \frac{\bar{t}_{\xi}}{\bar{t}} = \frac{u' + c'}{U - (c+u)} \quad \ln \bar{t} = \int_0^{\xi} \frac{u' + c'}{U - (c+u)} d\xi = I(\xi)$$

$$\therefore \bar{t}_{\xi} = e^{I(\xi)} \int_0^{\xi} \frac{e^{-I(\xi)}}{U - (u+c)} d\xi \quad \bar{t}_h = e^{I(\xi)}$$

$$\& \quad \bar{x}(\xi) = \xi + (u+c)\bar{t}$$

to get a simple expression for $t(\xi)$ we write u & U in terms of $\sigma(\xi) = u+c=1$

to get the eqs; why pick $\sigma(\xi)$ to be this, see below.

Find that only a part of the depression wave interacts with the shock;
this part is that part of the wave whose height is $> h_0$ i.e.

that region in x, t space whose characteristics satisfy $x/t = u+c \geq 1$

this is so if we look at

$$t(\xi) = K \left[\frac{(4/5 - \frac{\xi+l}{K})(5-1)}{(4/5-5)(\frac{\xi+l}{K}-1)} \right]$$

when $\frac{\xi+l}{K} = 1$ x & t are ∞ these are the coord of intersection between

the shock & characteristic $\Rightarrow u=0$ & $c=1 \Rightarrow$ forward moving part
decays shock for an infinitely long time.

If the wall moves with a given velocity $|u| > w > u_1 - 2c_1$. The rarefaction wave
is narrowed & a const state appears to left of variable zone. $u=w$

If $w \leq 0$ rarefaction wave is still wide enough to include that part
which decays the shock. If $0 < w < u_1$, there's a finite time
interaction in which shock decays from one constant state to another.
from the results indicated the height decreases rapidly at first & then
approaches its asymptotic value slowly

Finite Diff eqs.

Governing D.E. $u_t + uu_x + 2cc_x = 0$

$$2c_t + cu_x + 2uc_x = 0$$

with chara eq

$$\left. \begin{aligned} dx/dt &= u+c \\ u+2c &= \text{const} \end{aligned} \right\} \text{ along } C^+ \text{ char}$$

$$dx/dt = u-c \quad \left. \right\} \text{ along } C^- \text{ char}$$

The reason for the formulae as given is we look at

$$\frac{x_{i,k} - x_{i-1,k}}{t_{i,k} - t_{i-1,k}} = \frac{\Delta x}{\Delta t} \bigg|_{i=\frac{1}{2},k} = \frac{1}{2} [S_{i,k} + S_{i-1,k}] \quad \text{Centered difference on forward characteristic}$$

$$\frac{x_{i,k} - x_{i,k-1}}{t_{i,k} - t_{i,k-1}} = \frac{\Delta x}{\Delta t} \bigg|_{i=k=\frac{1}{2}} = \frac{1}{2} [T_{i,k} + T_{i,k-1}] \quad \text{centered diff on backward characteristic}$$

these can be solved for $u_{i,k} + c_{i,k}$, $u_{i,k} - c_{i,k}$, $t_{i,k}$, $x_{i,k}$

then from the jump conditions $v_{i,k}$ can be found.

Curves in x,t space are given for shock strength ~~for~~ $p_1/p_0 = 1.8, 3.0$ and ^{it is shown} ~~we show~~ that even for 3.0 results are in good agreement between

Finite Diff & Approx method; For $x \leq 7$ both methods give coinciding results. for large x,t they no longer coincide, probably due to the large mesh size of the F.D. Scheme in that region. Also in this region the Approx method ~~is better~~ should become more accurate since the shock becomes weaker.

Conclusion even higher than second order terms in the expansion of quantities across the shock differ little from what they would be in the case of a continuous wave. It therefore seems reasonable to apply the method of Friedrichs in cases where the pressure ratios are high.

