

Good

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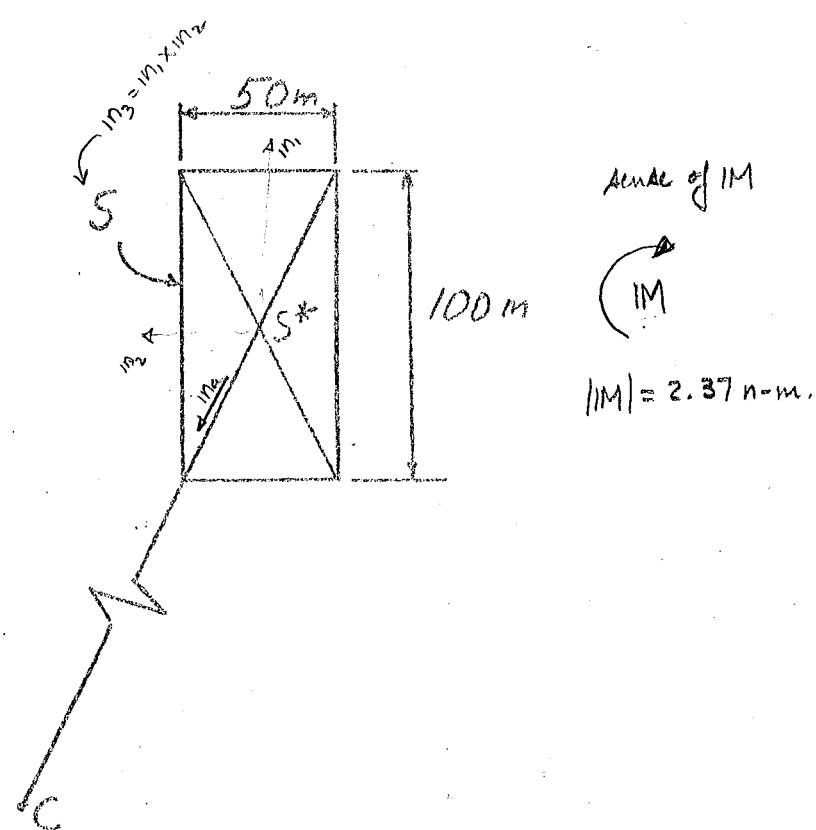
When the mass center  $S^*$  of an Earth satellite  $S$  moves on a circular orbit, the moment  $\underline{M}$  about  $S^*$  of all gravitational forces exerted on  $S$  by the Earth is given approximately by

$$\underline{M} = 3 \Omega^2 \underline{n}_a \times \underline{I}_a$$

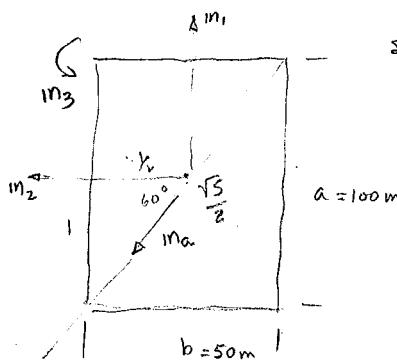
where  $\Omega$  is the angular speed of the line joining  $S^*$  to the center  $C$  of the Earth,  $\underline{n}_a$  is a unit vector parallel to this line, and  $\underline{I}_a$  is the second moment of  $S$  relative to  $S^*$  for  $\underline{n}_a$ .

The sketch shows a satellite consisting of a uniform rectangular plate having a mass of 1000 kg and having sides of length 50 m and 100 m. This satellite has a period of 90 minutes.

Determine the magnitude of  $\underline{M}$  and indicate the sense of  $\underline{M}$  on a sketch showing  $C$ ,  $S^*$ , and  $S$ .





Very good

Since  $m_1, m_2, m_3$  are unit vectors in direction of principal axes

$$\text{then } \mathbb{I} = \frac{mb^2}{12} m_1 m_1 + \frac{ma^2}{12} m_2 m_2 + \frac{m(a^2 + b^2)}{12} m_3 m_3$$

$$I_{\text{a}} = \frac{1}{\sqrt{3}} m_2 - \frac{2}{\sqrt{3}} m_1$$

$$I \cdot I_{\text{a}} = \mathbb{I}_{\text{a}} = \frac{mb^2}{12} \left( -\frac{2}{\sqrt{3}} m_1 \right) + \frac{1}{\sqrt{3}} \frac{ma^2}{12} m_2$$

$$I_{\text{a}} \times \mathbb{I}_{\text{a}} = \left( \frac{1}{\sqrt{3}} m_2 - \frac{2}{\sqrt{3}} m_1 \right) \times \left( -\frac{mb^2}{12} \cdot \frac{2}{\sqrt{3}} m_1 + \frac{1}{\sqrt{3}} \frac{ma^2}{12} m_2 \right)$$

$$= \frac{mb^2}{30} m_3 - \frac{(1/ma^2)m_3}{15}$$

$$= \frac{m(b^2 - 2a^2)}{30} m_3 = \frac{10000}{30} (-17500) m_3 = -1.75 \times 10^7 m_3 \text{ kg-m}$$

$$= \left( -1.75 \times 10^6 \text{ kg-m}^2 \right) m_3$$

$$\text{hence } M = 3S^2 \left( -\frac{1.75 \times 10^6}{3} \right) m_3 \text{ kg-sec}^2 \quad \text{where } S \text{ is in rad/sec.}$$

$$= -1.75 \times 10^6 S^2 m_3 \text{ kg-sec}^2$$

$$I_{\text{a}} = \frac{1}{\sqrt{3}} m_2 - \frac{2}{\sqrt{3}} m_1$$

$$1 \text{ period} = 90 \text{ min}$$

$$\Omega t = 2\pi \text{ rad.}$$

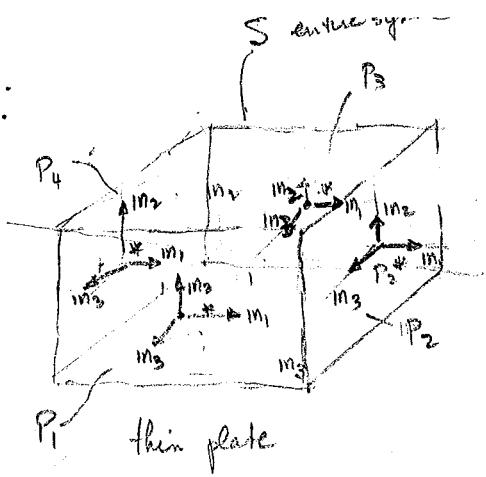
$$\Omega = \frac{2\pi}{T}$$

$$\Omega = \frac{2\pi}{90 \times 60} = \frac{2\pi}{5400} = \frac{\pi}{2700}$$

$$M = \left( \frac{2\pi}{5400} \right)^2 \left( -1.75 \times 10^6 \right) m_3 \text{ N-m} = -4\pi^2 (1.75) \cdot m_3 \text{ N-m}$$

$$|M| = \underbrace{\frac{4\pi^2 (1.75)}{(5400)^2}}_{\times} \times \underbrace{m_3}_{2.369 \text{ N-m}} \times \underbrace{2.369}_{\times} \text{ N-m}$$





Very good -  
see p. 2

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$$\text{let } m_1, m_2, m_3 \text{ be fixed in center of box}$$

$$I = \frac{ma^2}{12} (m_1m_1 + m_2m_2 + 2m_3m_3)$$

let each of these  
vectors be defined for  
each plate.  $\Rightarrow$   
 $I_1 \parallel I_2 \parallel I_3 \parallel I_4$

$$I = \frac{ma^2}{12} (2m_1m_1 + m_2m_2 + m_3m_3)$$

$$I = \frac{ma^2}{12} (m_1m_1 + m_2m_2 + 2m_3m_3)$$

$$I = \sqrt{\frac{ma^2}{12}} (2m_1m_1 + m_2m_2 + m_3m_3).$$

$$\text{now } I_{aa} = \frac{1}{\sqrt{3}} (m_2 + m_3 - m_1)$$

$$\text{then } I \cdot I_{aa} = \frac{ma^2}{12\sqrt{3}} (-m_1 + m_2 + m_3) \quad I_{aa} = \frac{ma^2}{12 \cdot 3} \cdot \frac{4}{3} = \frac{ma^2}{9}$$

$$I \cdot I_{aa} = \frac{ma^2}{12\sqrt{3}} (-2m_1 + m_2 + m_3) \quad I_{aa} = \frac{ma^2}{12 \cdot 3} \cdot \frac{4}{3} = \frac{ma^2}{9}$$

$$I \cdot I_{aa} = \frac{ma^2}{12\sqrt{3}} (-m_1 + m_2 + 2m_3) \quad I_{aa} = \frac{ma^2}{12 \cdot 3} \cdot \frac{4}{3} = \frac{ma^2}{9}$$

$$I_{aa} = I \cdot I_{aa} = \sqrt{\frac{ma^2}{9}}$$

$$\text{let } S^* \text{ center of box} \rightarrow I_{aa} = I_{aa}^{P/P_1^*} + I_{aa}^{P_1^*/S^*} \text{ w/. } P_1^*/S^* = \frac{1}{2} a m_3$$

$$\text{But } \rightarrow I_{aa}^{P_1^*/S^*} = m \left[ \frac{1}{4} a^2 - \frac{a \cdot a}{2\sqrt{3} \cdot 2\sqrt{3}} \right] = \frac{a^2}{6} = m(P \cdot P - (P \cdot I_{aa})(P \cdot I_{aa}))$$

$$\text{Also: } I_{aa}^{P_1^*/S^*} = \frac{ma^2}{9} + \frac{ma^2}{6} = \sqrt{\frac{10ma^2}{36}}$$

$$I_{aa}^{P_2/S^*} = I_{aa}^{P_2/P_2^*} + I_{aa}^{P_2^*/S^*} \text{ w/. } I_{aa}^{P_2^*/S^*} = m(P_2^* \cdot P - (P_2^* \cdot I_{aa})(P_2^* \cdot I_{aa}))$$

$$= \sqrt{\frac{10a^2 m}{36}}$$

$$\text{thus } P_2^*/S^* = \frac{1}{2} a m_1 \Rightarrow I_{aa}^{P_2^*/S^*} = m \left( \frac{a^2}{4} - \left( \frac{a}{2\sqrt{3}} \right) \left( \frac{-a}{2\sqrt{3}} \right) \right) = \frac{ma^2}{6}$$

then we can similarly show the same results for

$$I_{aa}^{P_3/S^*} = \sqrt{\frac{10ma^2}{36}}$$

$$I_{aa}^{P_4/S^*} = \sqrt{\frac{10ma^2}{36}}$$

$$I_{aa} = \sum I_{aa} = \sum I_{aa}^{P_i/S^*} = \frac{40ma^2}{36}$$

$$I_{aa}^2 = M k^2 \cdot 4m l^2 = \frac{40ma^2}{36} \Rightarrow \therefore k^2 = \frac{10a^2}{36} = \frac{a^2}{6}$$

~~S = box thus  $I_{aa} = \sum I_{aa} = \sum I_{aa}^{P_i/S^*} = \frac{40ma^2}{36}$~~

but

Technically correct answer please

$$\text{on both } I^{P/S^*} = \frac{ma^2}{12} (m_1m_1 + 2m_2m_2 + m_3m_3)$$

$$I_{ax} = \frac{ma^2}{9}$$

$$\& \text{ we can do same to show } I^{P/S^*} = \sqrt{\frac{10ma^2}{36}}$$

Let  $S = \text{box}$

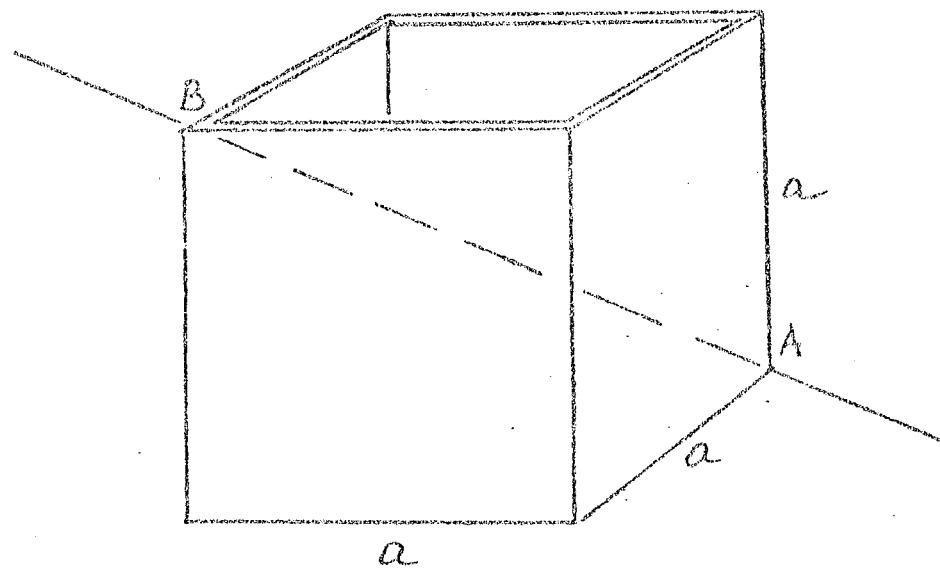
$$\text{Hence } I^{S/S^*} = \sum I = \frac{50ma^2}{36} = Mk^2$$

$$\therefore mk^2 = \frac{50ma^2}{36}$$

$$k^2 = \frac{50a^2}{144} \quad k = \frac{a\sqrt{50}}{12} \times \frac{a\sqrt{5}\sqrt{2}}{12}$$
$$\frac{\sqrt{10}}{6} a$$

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The sketch shows an open box whose base and sides consist of identical, thin, uniform plates. Determine the radius of gyration of the box with respect to line AB.





Poor

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A deformable body  $\alpha$  consists of two rigid bodies  $\beta$  and  $\gamma$  which are connected to each other at their respective mass centers but are otherwise free to move relative to each other.

At a certain instant  $t^*$ , the central principal axes of inertia of  $\beta$ , called  $Y_1, Y_2, Y_3$ , coincide respectively with  $Z_1, Z_2, Z_3$ , the central principal axes of  $\gamma$ ; and the angular velocity of  $\gamma$  relative to  $\beta$  at this instant is given by

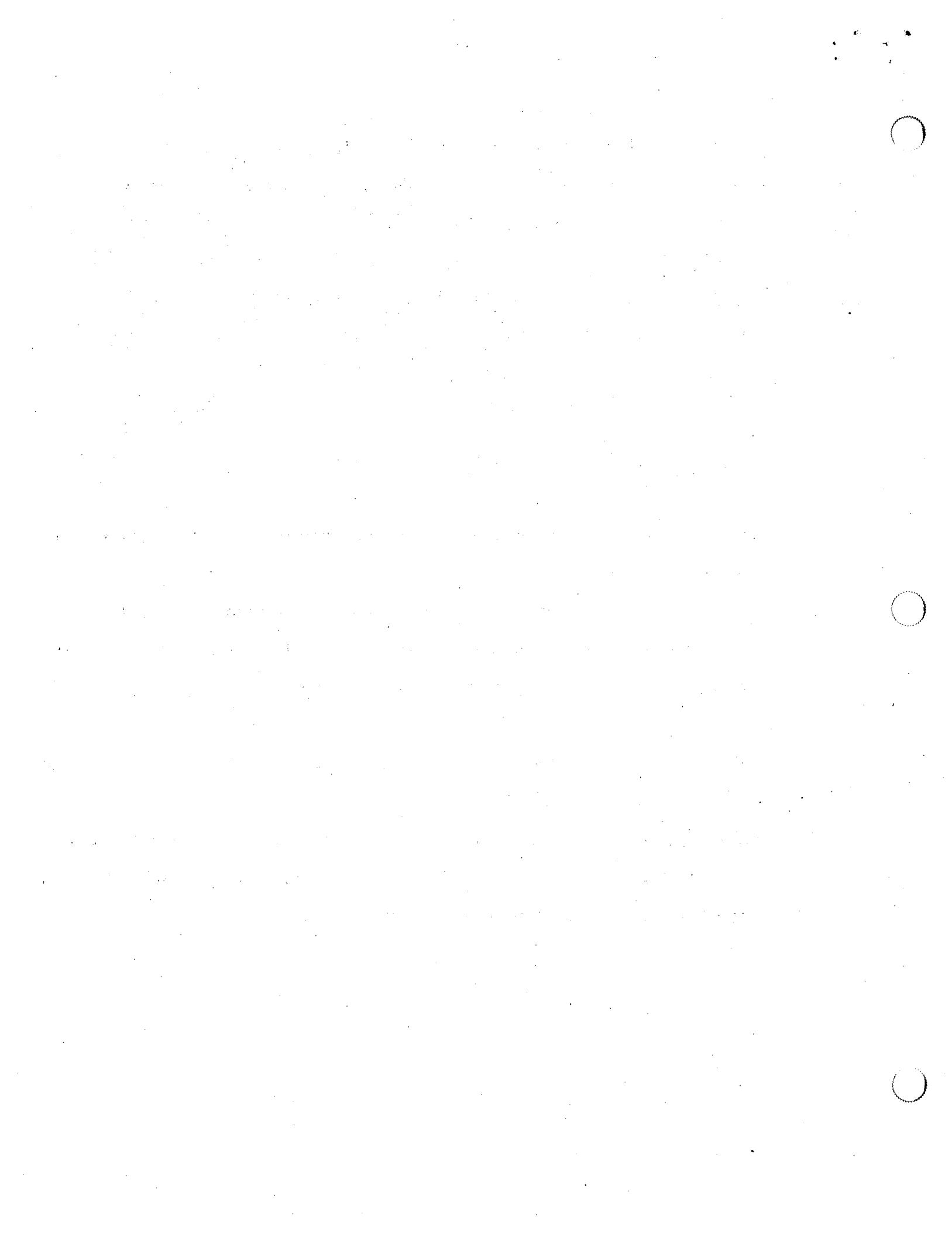
$$\overset{\beta}{\omega} \overset{\gamma}{Y} = \omega_1 \underline{n}_1 + \omega_2 \underline{n}_2 + \omega_3 \underline{n}_3$$

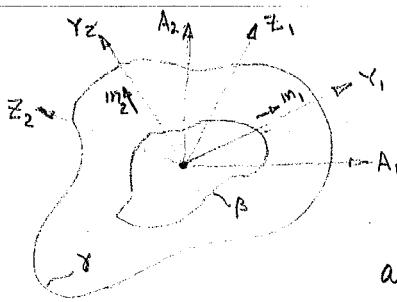
where  $\underline{n}_1, \underline{n}_2, \underline{n}_3$  are unit vectors respectively parallel to  $Y_1, Y_2, Y_3$  (and  $Z_1, Z_2, Z_3$ ).

Letting  $A$  be a reference frame in which the central principal axes of  $\alpha$  are fixed, determine the angular velocity of  $A$  relative to  $\beta$  for the instant  $t^*$ . Express the result in the form

$$\overset{\beta}{\omega} \overset{A}{A} = f_1 \underline{n}_1 + f_2 \underline{n}_2 + f_3 \underline{n}_3$$

where  $f_1, f_2, f_3$  are functions of  $\omega_1, \omega_2, \omega_3$  and of the central principal moments of inertia  $\beta_1, \beta_2, \beta_3$  and  $\gamma_1, \gamma_2, \gamma_3$  of  $\beta$  and  $\gamma$ , respectively; that is, determine these functions.





Let  $\gamma_1, \gamma_2, \gamma_3$  be the principal axes of inertia of  $\beta$ .

Let  $\mathbf{z}_1, \mathbf{z}_2, \mathbf{z}_3$  be the principal axes of inertia of  $\mathcal{S}$ .

Let  $A_1, A_2, A_3$  be the principal axes of inertia of  $\sigma$ .

in a reference frame A which is fixed

At time  $t=t^*$  the  $Z_i$ 's coincide with the  $Y_i$ 's. Furthermore

Let them coincide with the  $A_i$ 's. Let the  $m_j$ 's be unit vectors  $\parallel$  to  $y_j$ 's,

If @  $t = t^4$        $\beta_{60}^{(2)} = \omega_1 m_1 + \omega_2 m_2 + \omega_3 m_3$     &  $\beta_1's$  and  $\delta_1's$  are prime moments of inertia of  $\beta_1$ 's

$$\text{Find: } \beta_{\omega A} = f_1 m_1 + f_2 m_2 + f_3 m_3$$

In the reference frame A

$$\frac{\alpha}{\beta} = \frac{1}{1 - \left( \frac{\delta}{\beta} + \frac{\gamma}{1 - \beta/\alpha} \right)} \quad (1)$$

where  $\delta$  is a fictitious rigid body having the mass distribution of  $\alpha$  but the instantaneous motion of  $\beta$ . Hence  $I_{\beta}^{\alpha/\delta} = I_{\beta}^{\alpha/\alpha} \cdot A_{\alpha}^{\beta}$  (2)

Yet we may also write

$$\text{II-} \frac{\beta/\alpha^*}{\beta/\beta^*} = \text{II-} \frac{\beta/\beta^*}{\beta/\alpha^*} + \text{II-} \frac{\beta^*/\alpha^*}{\beta/\alpha^*}$$

$$\text{But } \prod_{i=1}^n B_i^{x_i} = m_p \prod_{i=1}^n x_i^{B_i} \times \prod_{i=1}^n B_i^* = 0 \text{ since } B_i^* \text{ and } x_i^* \text{ coincide} \quad (4)$$

$$\text{Thus } \frac{\gamma}{\| \cdot \|^{B/\alpha}} \leq \frac{\gamma}{\| \cdot \|^{B/p^*}} = \prod_{i=1}^n \frac{\gamma}{\| x_i \|^{\beta_i/p_i}} \quad (4.3)$$

Now at the instant of coincidence  $\frac{A}{H-1} \propto \alpha^* = 0$  because  $\frac{A}{\alpha} \propto 0$  (6)

Also at that instant from (2)

$$\Pi^{\delta/\delta^+} = \Pi^{\alpha/\alpha^+} \cdot \omega^\gamma = (\Pi^{\beta/\beta^+} + \Pi^{\delta/\delta^+}) \cdot \omega^\gamma$$

due to all axes being principal axes and coinciding

Now we also know that at that instant

$$IL^{P_{PM}} = \beta_1 m_1 m_1 + \beta_2 m_2 m_2 + \beta_3 m_3 m_3$$

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$$\Pi^{(V_{\alpha})} = (\beta_1 + \gamma_1) m_1 m_1 + (\beta_2 + \gamma_2) m_2 m_2 + (\beta_3 + \gamma_3) m_3 m_3 \quad (9)$$

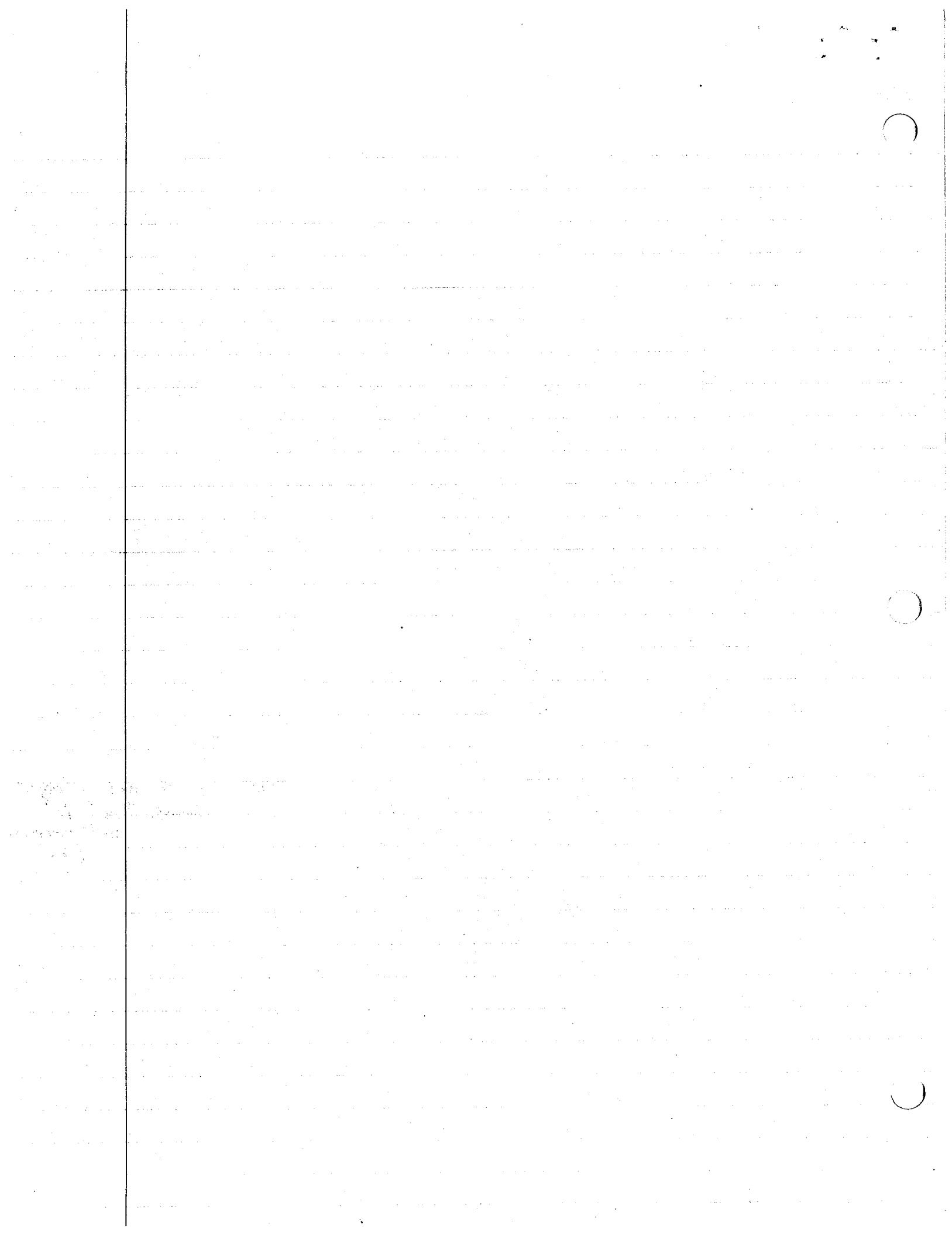
but also

$${}^A\omega^\gamma = {}^A\omega^\beta + {}^B\omega^\gamma = (-f_1 + \omega_1)m_1 + (-f_2 + \omega_2)m_2 + (-f_3 + \omega_3)m_3 \quad (10)$$

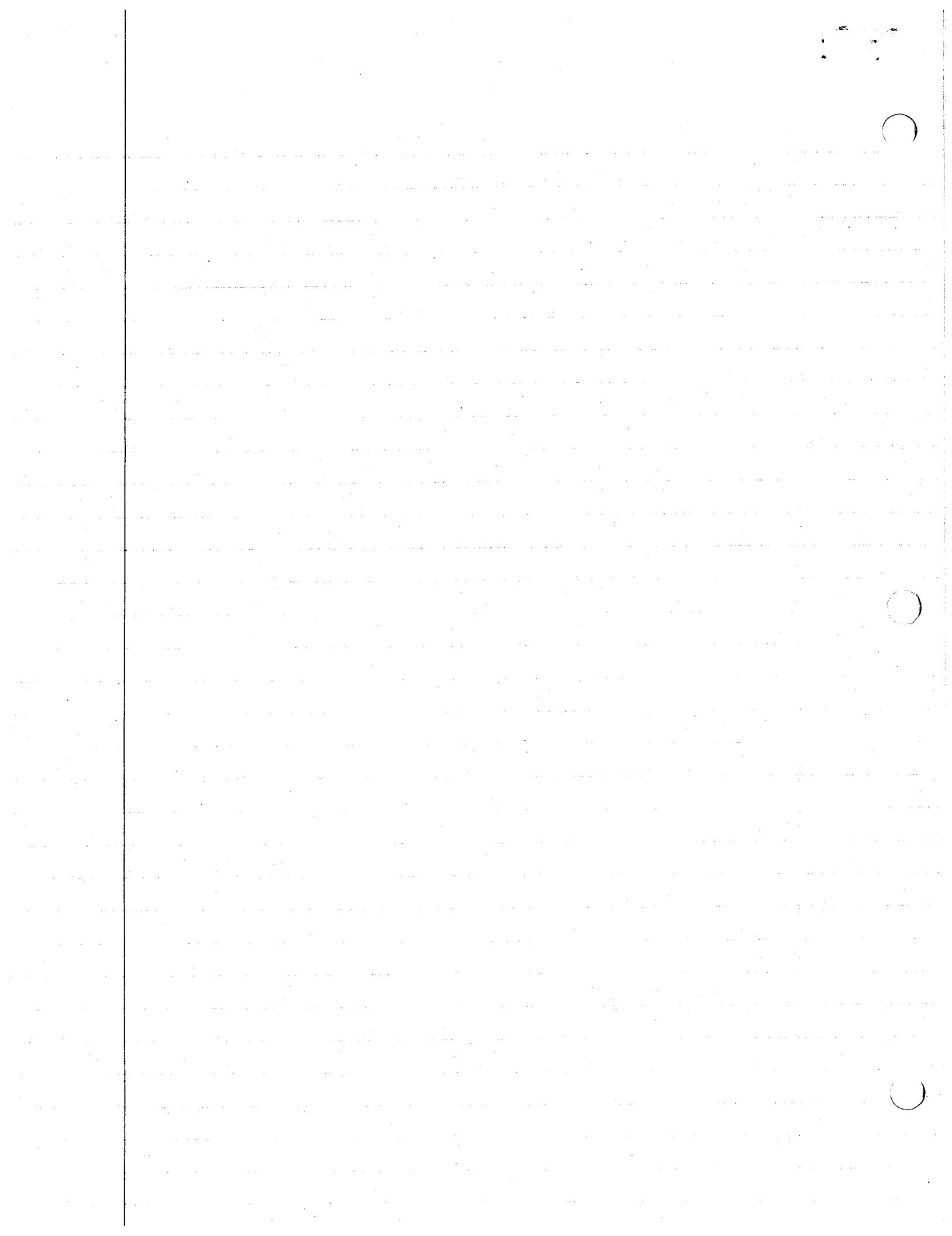
Thus using (5-10) in (3) we obtain

$$0 = (\beta_1 + \gamma_1)(-\dot{f}_1 + \omega_1)m_1 + (\beta_2 + \gamma_2)(-\dot{f}_2 + \omega_2)m_2 + (\beta_3 + \gamma_3)(-\dot{f}_3 + \omega_3)m_3 - [\beta_1\omega_1m_1 + \beta_2\omega_2m_2 + \beta_3\omega_3m_3] \\ = (\omega_1\gamma_1 - f_1[\beta_1 + \gamma_1])m_1 + (\omega_2\gamma_2 - f_2[\beta_2 + \gamma_2])m_2 + (\omega_3\gamma_3 - f_3[\beta_3 + \gamma_3])m_3 \quad (11)$$

Solving the 3 scalar equations obtained from (11) yields



$$f_1 = \frac{\alpha_1 w_1}{\alpha_1 + \beta_1} \quad f_2 = \frac{\alpha_2 w_2}{\alpha_2 + \beta_2} \quad f_3 = \frac{\alpha_3 w_3}{\alpha_3 + \beta_3} \quad X \quad (12, 13, 14)$$



## Christine Miller

Body  $\beta$  {  $y_1 \ y_2 \ y_3$  prim axes  
 $b_1 \ b_2 \ b_3$  unit vectors along prim. axes  
 $\beta_1 \ \beta_2 \ \beta_3$  prim moments

Body  $\gamma$  {  $z_1 \ z_2 \ z_3$  prim. axes  
 $c_1 \ c_2 \ c_3$  unit vectors along prim axes  
 $\gamma_1 \ \gamma_2 \ \gamma_3$  prim moments

Let A be a co-ord. system in which the central prim. moments of  $\alpha = \beta + \gamma$  are fixed, with unit vectors  $a_1, a_2, a_3$

At  $t^*$ ,  $X, Y \in A$  correspond and:

$$\beta \omega^A = \omega_1 m_1 + \omega_2 m_2 + \omega_3 m_3 = \omega_1 b_1 + \omega_2 b_2 + \omega_3 b_3$$

$$\beta \omega^A = f_1 m_1 + f_2 m_2 + f_3 m_3 = f_1 b_1 + f_2 b_2 + f_3 b_3$$

Let " $\tilde{\cdot}$ " denote quantities at  $t^*$ . We will work in a reference frame  $\beta$  where  $b_i^0$  are fixed. At a very small time  $t$  after  $t^*$ ,

$$a_i = \tilde{a}_i + \left( \frac{da_i}{dt} \right)_{t^*} t + \dots \approx \tilde{a}_i + (\beta \dot{\omega}^A \times a_i) \cdot t$$

where we will drop all terms of order two or higher in  $t$ .

$$a_1 = b_1 + [(f_1 b_1 + f_2 b_2 + f_3 b_3) \times b_1] t \\ = b_1 + f_3 t b_2 - f_2 t b_3$$

$$a_2 = b_2 + [(f_1 b_1 + f_2 b_2 + f_3 b_3) \times b_2] t \\ = -f_3 t b_1 + b_2 + f_1 t b_3$$

$$a_3 = b_3 + [(f_1 b_1 + f_2 b_2 + f_3 b_3) \times b_3] t \\ = f_2 t b_1 - f_1 t b_2 + b_3$$



For the vectors  $\alpha_i$ , for the reference frame  $\gamma$ :

$$\gamma \omega^A = \omega^B + \beta_{\omega}^A = \beta_{\omega}^A - \beta_{\omega}^{\gamma}$$

$${}^A \omega^{\gamma} = (f_1 - \omega_1) \alpha_1 + (f_2 - \omega_2) \alpha_2 + (f_3 - \omega_3) \alpha_3$$

$$\alpha_i = \tilde{\alpha}_i + \left( \frac{d\alpha}{dt} \right)_{t^*} \cdot t + \dots \equiv \tilde{\alpha}_i + (\gamma \omega^A \times \alpha_i)_{t^*} \cdot t$$

$$\alpha_1 = C_1 + (f_3 - \omega_3)t C_2 - (f_2 - \omega_2)t C_3$$

$$\alpha_2 = -(f_3 - \omega_3)t C_1 + C_2 + (f_1 - \omega_1)t C_3$$

$$\alpha_3 = (f_2 - \omega_2)t C_1 - (f_1 - \omega_1)t C_2 + C_3$$

Summarizing,

	$b_1$	$b_2$	$b_3$	$C_1$	$C_2$	$C_3$
$\alpha_1$	1	$f_3 t$	$-f_2 t$	1	$(f_3 - \omega_3)t$	$-(f_2 - \omega_2)t$
$\alpha_2$	$-f_3 t$	1	$f_1 t$	$(f_3 - \omega_3)t$	1	$(f_1 - \omega_1)t$
$\alpha_3$	$f_2 t$	$-f_1 t$	1	$(f_2 - \omega_2)t$	$-(f_1 - \omega_1)t$	1

At time  $t$  we can calculate the products of inertia of bodies  $\beta \notin \gamma$  about their mass centers for axes  $\alpha_i$ :

$$I_{ab} = \sum_{j=1}^3 \sum_{k=1}^3 \alpha_j I_{jk} b_k$$

$$I_{12}^{B*} = -\beta_1 f_3 t + \beta_2 f_3 t + \beta_3 f_2 t^2 f_1$$

$$I_{12}^{\gamma \omega^A} = -\gamma_1 (f_3 - \omega_3)t + \gamma_2 (f_3 - \omega_3)t - \gamma_3 (f_2 - \omega_2)(f_1 - \omega_1)t^2$$

$$I_{23}^{B*} = -\beta_1 f_3^2 f_2 + \beta_2 f_1 t + \beta_3 f_1 t$$

$$I_{23}^{\gamma \omega^A} = -\gamma_1 (f_3 - \omega_3)(f_2 - \omega_2)t^2 - \gamma_2 (f_1 - \omega_1)t + \gamma_3 (f_1 - \omega_1)t$$



$$I_{13}^{\beta/\beta^*} = \beta_1 f_2 t - \beta_2 f_3 t^2 f_1 - \beta_3 f_2 t$$

$$I_{13}^{\gamma/\gamma^*} = \gamma_1 (f_2 - \omega_2) t - \gamma_2 (f_3 - \omega_3) (f_1 - \omega_1) t^2 - \beta_3 (f_2 - \omega_2) t$$

Since the  $\alpha_i$  are the principal axes, the products of inertia for  $\beta + \gamma$  must disappear.

$$I_{12}^{\beta/\beta^*} + I_{12}^{\gamma/\gamma^*} = 0$$

$$-\beta_1 f_3 t + \beta_2 f_3 t - \gamma_1 (f_3 - \omega_3) t + \gamma_2 (f_3 - \omega_3) t = 0$$

$$f_3 (-\beta_1 + \beta_2 - \gamma_1 + \gamma_2) = \omega_3 (\gamma_2 - \gamma_1)$$

$$f_3 = \frac{\omega_3 (\gamma_2 - \gamma_1)}{\beta_2 - \beta_1 + \gamma_2 - \gamma_1}$$

$$I_{23}^{\beta/\beta^*} + I_{23}^{\gamma/\gamma^*} = 0$$

$$-\beta_2 f_1 t + \beta_3 f_1 t - \gamma_2 (f_1 - \omega_1) t + \gamma_3 (f_1 - \omega_1) t = 0$$

$$f_1 (-\beta_2 + \beta_3 - \gamma_2 + \gamma_3) = \omega_1 (\gamma_3 - \gamma_2)$$

$$f_1 = \frac{\omega_1 (\gamma_3 - \gamma_2)}{\beta_3 - \beta_2 + \gamma_3 - \gamma_2}$$

$$I_{13}^{\beta/\beta^*} + I_{13}^{\gamma/\gamma^*} = 0$$

$$\beta_1 f_2 t - \beta_3 f_2 t + \gamma_1 (f_2 - \omega_2) t - \gamma_3 (f_2 - \omega_2) t = 0$$

$$f_2 (\beta_1 - \beta_3 + \gamma_1 - \gamma_3) = \omega_2 (\gamma_1 - \gamma_3)$$

$$f_2 = \frac{\omega_2 (\gamma_1 - \gamma_3)}{\beta_1 - \beta_3 + \gamma_1 - \gamma_3}$$



Very nice

800130

A rigid body A and a rigid axisymmetric rotor B whose mass center  $B^*$  and axis of symmetry are fixed in A form a "gyrostat" C. Letting  $C^*$  designate the mass center of C, express the inertia torque of C in a reference frame R solely in terms of the inertia dyadics  $I_{B/B^*}^{C/C^*}$  and  $I_{A/A^*}^{B/B^*}$ , the angular velocities  $\dot{\omega}_A^R$  and  $\dot{\omega}_B^A$ , and the angular accelerations  $\ddot{\alpha}_A^R$  and  $\ddot{\alpha}_B^A$ .



## Problem #1

From Class we had been given the following:

$$\overset{R}{\text{II}} \overset{C/C^*}{=} \overset{R}{\text{II}} \overset{D/D^*}{+} \overset{A}{\text{II}} \overset{B/B^*}{\checkmark} \quad (1)$$

where C makes up the system of the rotor B and the rigid body A; D is a fictitious rigid body whose mass distribution is the same as C but whose motion at any instant is that of A. The asterisk (\*) denotes the mass center

In problem 7(l) we had showed that for a system of particles S

$$\overset{R}{\text{II}} \overset{S/S^*}{=} \overset{R}{\text{II}} \overset{S/S^*}{+} \overset{R}{\text{II}} \overset{S/S^*}{\checkmark} \quad (2a)$$

$$\text{where it was shown that } \overset{R}{\text{II}} \overset{S/S^*}{=} m_S \overset{R}{I} R \overset{S/S^*}{\times} \overset{R}{N} \overset{S/S^*}{\checkmark} \quad (2b)$$

where  $\overset{R}{\text{II}} \overset{S/S^*}{}$  is the angular momentum of S with respect to point O and  $m_S = \sum_{i=1}^N m_i$

$$\text{Thus } \overset{R}{\text{II}} \overset{B/B^*}{=} \overset{A}{\text{II}} \overset{B/B^*}{+} \overset{A}{\text{II}} \overset{B/B^*}{\checkmark} \quad (2c)$$

$$\text{where } \overset{A}{\text{II}} \overset{B/B^*}{\checkmark} = m_B \overset{R}{I} R \overset{B/B^*}{\times} \overset{A}{N} \overset{B/B^*}{\checkmark} \quad (2d)$$

But because  $B^*$  is fixed in A

$$\overset{A}{N} \overset{B/B^*}{\checkmark} = 0 \Rightarrow \overset{A}{\text{II}} \overset{B/B^*}{\checkmark} \neq 0 \quad (2e, f)$$

$$\text{Thus } \boxed{\overset{A}{\text{II}} \overset{B/B^*}{\checkmark} \overset{A}{\text{II}} \overset{B/B^*}{\checkmark}} \text{ using (2f) & (2c).} \quad (2g)$$

By problem 8(b)  $\overset{R}{\text{II}} \overset{C/C^*}{=} \overset{R}{\text{II}} \overset{D/D^*}{+} \overset{R}{\text{II}} \overset{B/B^*}{\checkmark}$  ~~using results of problem 7(a)~~

Thus by differentiation of

$$\overset{R}{\text{II}} \overset{C/C^*}{\dot{=}} \overset{R}{\text{II}} \overset{D/D^*}{+} \overset{R}{\text{II}} \overset{B/B^*}{\checkmark} \quad (3b)$$

we will find

$$\frac{d}{dt} \overset{R}{\text{II}} \overset{C/C^*}{\checkmark} = \overset{R}{\text{II}} \overset{C/C^*}{\dot{=}} = \frac{d}{dt} \overset{R}{\text{II}} \overset{D/D^*}{\checkmark} + \frac{d}{dt} \overset{R}{\text{II}} \overset{B/B^*}{\checkmark} \quad (4)$$

$$\text{Now. } \frac{d}{dt} \overset{R}{\text{II}} \overset{A}{\text{II}} \overset{B/B^*}{\checkmark} = \frac{d}{dt} \overset{R}{\text{II}} \overset{A}{\text{II}} \overset{B/B^*}{\checkmark} + \overset{R}{\omega} \overset{A}{\times} \overset{R}{\text{II}} \overset{A}{\text{II}} \overset{B/B^*}{\checkmark} = \overset{R}{\omega} \overset{A}{\times} \overset{R}{\text{II}} \overset{B/B^*}{\checkmark} + \overset{R}{\omega} \overset{A}{\times} \overset{R}{\text{II}} \overset{B/B^*}{\checkmark} \quad (5a)$$

$$= -\overset{A}{\text{II}} \overset{B}{\checkmark} - (\overset{R}{\text{II}} \overset{B/B^*}{\checkmark}, \overset{A}{\omega}) \times \overset{R}{\omega} \overset{A}{\checkmark} \quad \text{using results of problem (7i).} \quad (6)$$

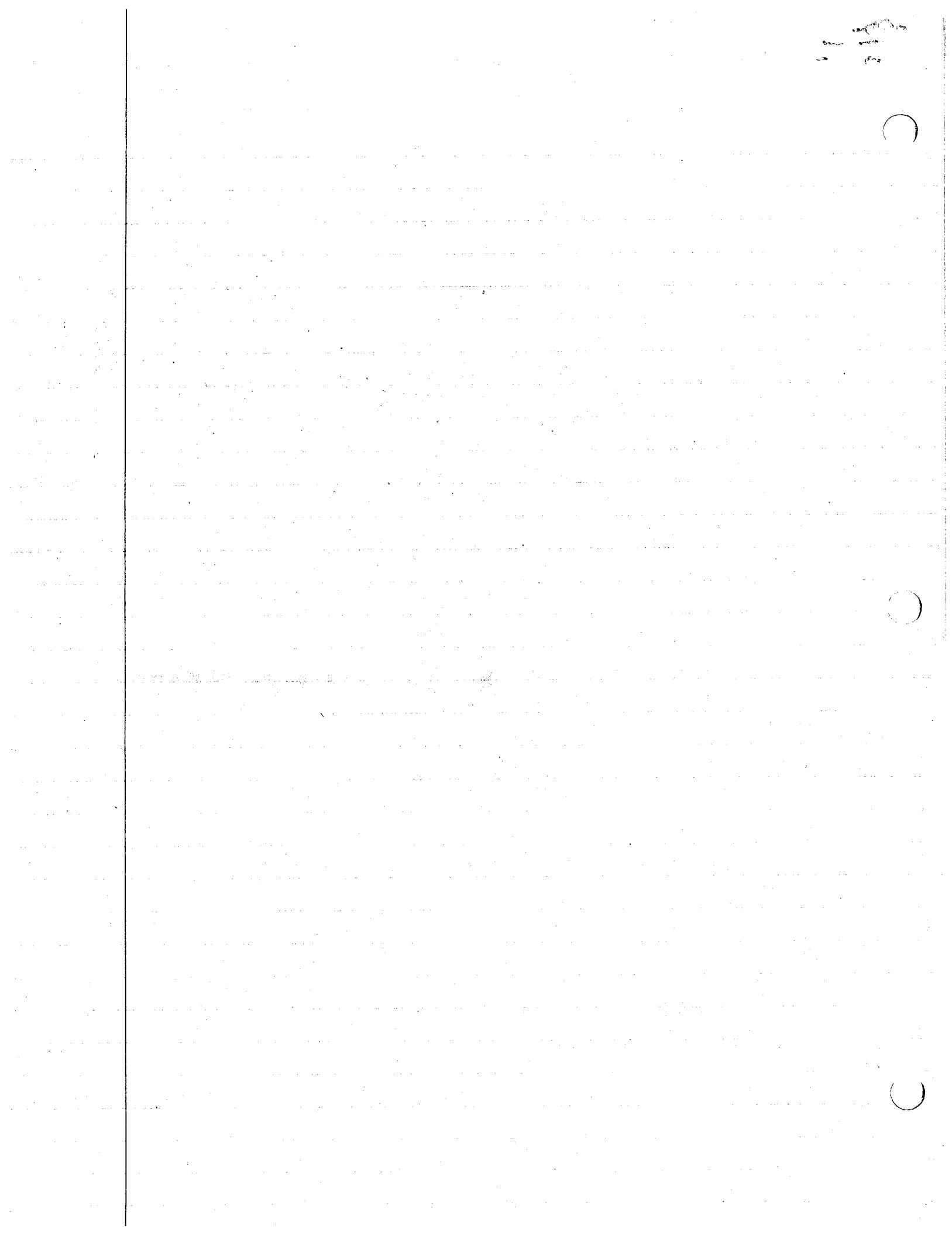
$$\text{Thus } \boxed{\frac{d}{dt} \overset{R}{\text{II}} \overset{A}{\text{II}} \overset{B/B^*}{\checkmark} = -[(\overset{R}{\text{II}} \overset{B/B^*}{\checkmark}, \overset{A}{\omega}) \times \overset{R}{\omega} \overset{A}{\checkmark} - \overset{R}{\text{II}} \overset{B/B^*}{\checkmark}, \overset{A}{\omega} \overset{B}{\checkmark} + (\overset{R}{\text{II}} \overset{B/B^*}{\checkmark}, \overset{A}{\omega} \overset{B}{\checkmark}) \times \overset{R}{\omega} \overset{A}{\checkmark}]} \quad (7)$$

$$\text{Now we look at } \frac{d}{dt} \overset{R}{\text{II}} \overset{D/D^*}{\checkmark} = -[(\overset{R}{\text{II}} \overset{D/D^*}{\checkmark}, \overset{R}{\omega} \overset{D}{\checkmark}) \times \overset{R}{\omega} \overset{D}{\checkmark} - \overset{R}{\text{II}} \overset{D/D^*}{\checkmark}, \overset{R}{\omega} \overset{D}{\checkmark}] \quad (8)$$

but since D represents a rigid body (fictitious though) whose motion is that of A at any instant and whose mass distribution is that of C then

$$\overset{R}{\omega} \overset{D}{\checkmark} = \overset{R}{\omega} \overset{A}{\checkmark} \quad \overset{R}{\omega} \overset{D}{\checkmark} = \overset{R}{\omega} \overset{A}{\checkmark} \quad \text{and } \overset{R}{\text{II}} \overset{D/D^*}{\checkmark} = \overset{R}{\text{II}} \overset{C/C^*}{\checkmark} \quad (9a, b, c)$$

Placing (9a-c) into (8) yields



$$\left| \frac{R}{dt} \frac{d^R H^{D/B}}{d\omega^A} = - \left[ (I^{C/C}, \omega^A) \times \omega^A - I^{C/C}, \omega^A \right] \right| \quad (10)$$

Now using (10) & (7) in (4). then

$$\left| R \dot{\pi}^C = (I^{C/C}, \omega^A) \times \omega^A - I^{C/C}, \omega^A + (I^{B/B}, \omega^B) \times \omega^B - I^{B/B}, \omega^B + (I^{B/B}, \omega^B) \times \omega^A \right| \quad (11)$$

which was to be obtained ✓

To show (10)

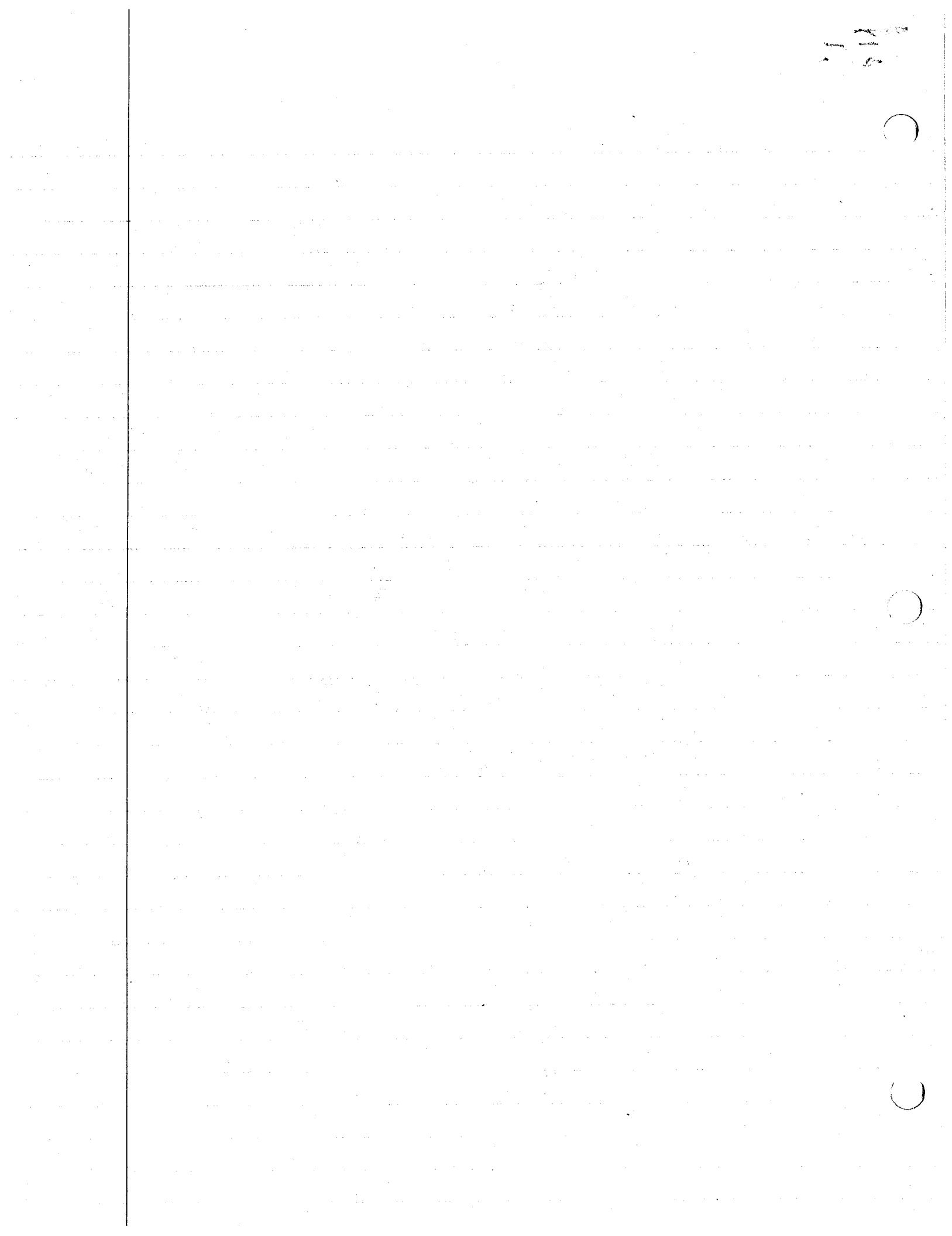
$$\frac{R}{dt} \frac{d^R H^{D/B}}{d\omega^A} = \frac{R}{dt} \sum_{i=1}^n r_i \frac{P_i/C}{\omega^A} (\omega^A \times I^P P_i/C) = \frac{R}{dt} \sum_{i=1}^n r_i \frac{P_i/C}{\omega^A} \times (\omega^A \times I^P P_i/C) + \omega^D \times \frac{R}{dt} \frac{d^R H^{D/B}}{d\omega^A}$$

$$\text{but } \frac{R}{dt} \sum_{i=1}^n r_i \frac{P_i/C}{\omega^A} \times (\omega^A \times I^P P_i/C) = \sum_{i=1}^n r_i \frac{P_i/C}{\omega^A} \times \left( \frac{R}{dt} \omega^A \right) \times I^P P_i/C \text{ since } I^P P_i/C \text{ is fixed wrt body C as well as body D.}$$

$$\text{but } \frac{R}{dt} \omega^A = \frac{R}{dt} \omega^A + \omega^D \times \omega^A = \omega^A + \omega^D \times \omega^A$$

$$\text{but } \omega^D = -\omega^A \text{ by definition of body D hence } \frac{R}{dt} \omega^A = \omega^A$$

$$\text{thus } \frac{R}{dt} \frac{d^R H^{D/B}}{d\omega^A} = I^{C/C}, \omega^A + \omega^A \times (I^{C/C}, \omega^A) \text{ as required}$$



$$A \dot{H}^{\alpha/\alpha^*} = \dot{H}^{\alpha/\alpha^*} \cdot \omega^\gamma + \dot{H}^{\beta/\beta^*} \cdot \omega^\beta$$

$$\dot{H}^{\alpha/\alpha^*} \cdot \frac{A}{\omega} \alpha$$

$$\frac{A}{\omega} \gamma$$

$$R \dot{H}^{\gamma/\gamma^*} = R \dot{H}^{\beta/\beta^*} + R \dot{H}^{\gamma/\gamma^*}$$

$$\text{since } A \dot{H}^{\beta/\beta^*} = \dot{H}^{\beta/\beta^*} + \dot{H}^{\gamma/\gamma^*}$$

$$m_B R^{\beta/\beta^*} \times \omega^{\beta/\beta^*} = 0$$

$$A \dot{H}^{\beta/\beta^*}$$

$$R \dot{H}^{\gamma/\gamma^*} = R \dot{H}^{\beta/\beta^*} + \dot{H}^{\gamma/\gamma^*} \cdot \frac{A}{\omega} \beta$$

$$R \dot{H}^{\gamma/\gamma^*} = \dot{H}^{\gamma/\gamma^*} \cdot \frac{R}{\omega} A + \dot{H}^{\gamma/\gamma^*} \cdot \frac{A}{\omega} \beta$$

if  $R = C$

$$C \dot{H}^{\gamma/\gamma^*} = 0 = \dot{H}^{\gamma/\gamma^*} \cdot \frac{C}{\omega} A + \dot{H}^{\gamma/\gamma^*} \cdot \frac{A}{\omega} \beta$$

$$\text{given } \frac{B}{\omega} A = \omega_1 m_1 + \omega_2 m_2 + \omega_3 m_3 \\ \text{need to find } \frac{B}{\omega} C$$

$$\text{Now } \dot{H}^{\gamma/\gamma^*} = (\beta_1 + \gamma_1) m_1 m_1 + (\beta_2 + \gamma_2) m_2 m_2 + (\beta_3 + \gamma_3) m_3 m_3$$

$$\frac{C}{\omega} A = \underline{\underline{m}_1 m_1 + m_2 m_2 + m_3 m_3}$$

$$\frac{C}{\omega} B + \frac{B}{\omega} A = \frac{C}{\omega} A$$

$$\dot{H}^{\beta/\beta^*} = (\beta_1) m_1 m_1 + \beta_2 m_2 m_2 + \beta_3 m_3 m_3$$

$$\frac{A}{\omega} B = (\omega_1 m_1 + \omega_2 m_2 + \omega_3 m_3)$$

hence.

$$(\beta_1 + \gamma_1) \underline{m}_1 m_1 + (\beta_2 + \gamma_2) \underline{m}_2 m_2 + (\beta_3 + \gamma_3) \underline{m}_3 m_3 + \beta_1 \omega_1 m_1 + \beta_2 \omega_2 m_2 + \beta_3 \omega_3 m_3 = 0$$

$$\text{thus } (\beta_1 + \gamma_1) \underline{m}_1 + \beta_1 \omega_1 = 0 \quad \text{thus } \underline{m}_1 = -\frac{\beta_1 \omega_1}{\beta_1 + \gamma_1} \quad \text{hence } \underline{m}_i = -\frac{\beta_i \omega_i}{\beta_i + \gamma_i} \text{ i not}$$

$$R \dot{H}^{\alpha/\alpha^*} = R \dot{H}^{\delta/\delta^*} + \dot{H}^{\beta/\beta^*}$$

$$A \dot{H}^{\alpha/\alpha^*} = A \dot{H}^{\delta/\delta^*} + \dot{H}^{\beta/\beta^*}$$

$$= \dot{H}^{\gamma/\gamma^*} \cdot \frac{A}{\omega} \gamma + \dot{H}^{\beta/\beta^*} \cdot \frac{A}{\omega} \beta$$

now if  $R = A$  & when  $\theta$ 's are zero

$$\dot{H}^{\beta/\beta^*} = \dot{H}^{\beta/\beta^*} + \dot{H}^{\beta/\beta^*}$$

$$R \dot{H}^{\beta/\beta^*} = m_B R^{\beta/\beta^*} \times \omega^{\beta/\beta^*}$$

$$\text{since } R^{\beta/\beta^*} = 0$$

$$\gamma + \beta = \alpha$$

$\delta$  is a fictitious body w.r.t.  $\alpha$  but imbalance motion of  $\gamma$

now  $A \dot{H}^{\alpha/\alpha^*} = 0$  since at the instant the momentum of  $\alpha$  wrt  $\alpha^*$  in Reference frame A = 0

$$\text{now } \gamma \omega \beta = -\frac{\beta}{\omega} \gamma = -[\omega_1 m_1 + \omega_2 m_2 + \omega_3 m_3]$$

$$\text{at that instant } \dot{H}^{\beta/\beta^*} = \beta_1 m_1 m_1 + \beta_2 m_2 m_2 + \beta_3 m_3 m_3$$

$$\dot{H}^{\alpha/\alpha^*} = (\beta_1 + \gamma_1) m_1 m_1 + (\beta_2 + \gamma_2) m_2 m_2 + (\beta_3 + \gamma_3) m_3 m_3$$

$$\text{since } \dot{H}^{\alpha/\alpha^*} = \dot{H}^{\beta/\beta^*} + \dot{H}^{\gamma/\gamma^*}$$

$$\text{and let } \overset{A}{\omega} \overset{\alpha}{\times} = - \overset{\alpha}{\omega} A = - (\overset{\alpha}{\omega} \beta + \overset{\alpha}{\omega} A) = - \overset{\alpha}{\omega} \beta - \overset{\alpha}{\omega} A \\ = (\overset{\alpha}{\omega} \beta - \overset{\alpha}{\omega} A)$$

$$S_1 m_1 + S_2 m_2 + S_3 m_3 \overset{A}{\omega} \overset{\alpha}{\times} = \overset{\alpha}{\omega} A = w_1 m_1 + w_2 m_2 + w_3 m_3 - [ \eta_1 m_1 + \eta_2 m_2 + \eta_3 m_3 ]$$

$$\overset{\alpha}{\omega} \overset{\alpha}{\times} \overset{A}{\omega} \overset{\alpha}{\times} + \overset{\alpha}{\omega} \overset{\alpha}{\times} \overset{A}{\omega} \overset{\alpha}{\times} \quad \overset{\alpha}{\omega} A = - w_1 m_1 - w_2 m_2 - w_3 m_3$$

$$(\gamma_1 + \beta_1) S_1 m_1 + (\gamma_2 + \beta_2) S_2 m_2 + (\gamma_3 + \beta_3) S_3 m_3 - \beta_1 w_1 m_1 - \beta_2 w_2 m_2 - \beta_3 w_3 m_3 = 0$$

$$(\gamma_1 + \beta_1) [w_1 - \eta_1] m_1 + (\gamma_2 + \beta_2) (w_2 - \eta_2) m_2 + (\gamma_3 + \beta_3) (w_3 - \eta_3) m_3 - \beta_1 w_1 m_1 - \beta_2 w_2 m_2 - \beta_3 w_3 m_3 = 0$$

$$[w_1 \gamma_1 + (\gamma_1 + \beta_1) \eta_1] m_1 + [w_2 \gamma_2 + (\gamma_2 + \beta_2) \eta_2] m_2 + [w_3 \gamma_3 + (\gamma_3 + \beta_3) \eta_3] m_3 = 0$$

$$\Rightarrow \frac{\gamma_i w_i}{\gamma_i + \beta_i} = \eta_i \quad \therefore \overset{\alpha}{\omega} A = \eta_i m_i$$

$$\overset{\alpha}{\omega} \overset{\alpha}{\times} A \overset{\alpha}{\times} \overset{\alpha}{\omega} \overset{\alpha}{\times} = \overset{\alpha}{\omega} \overset{\alpha}{\times} \overset{\alpha}{\omega} \overset{\alpha}{\times} A = \frac{\gamma_1 w_1}{\gamma_1 + \beta_1} m_1 + \frac{\gamma_2 w_2}{\gamma_2 + \beta_2} m_2 + \frac{\gamma_3 w_3}{\gamma_3 + \beta_3} m_3$$

$$A \overset{\alpha}{\times} \overset{\alpha}{\omega} \overset{\alpha}{\times} = (\overset{\alpha}{\omega} \overset{\alpha}{\times} \overset{\alpha}{\omega} \overset{\alpha}{\times}) \circ (\overset{\alpha}{\omega} \overset{\alpha}{\times} \overset{\alpha}{\omega} \overset{\alpha}{\times}) = \overset{\alpha}{\omega} \overset{\alpha}{\times} \overset{\alpha}{\omega} \overset{\alpha}{\times} + \overset{\alpha}{\omega} \overset{\alpha}{\times} \overset{\alpha}{\omega} \overset{\alpha}{\times}$$

$${}^R \dot{\Pi}^C = {}^R \dot{\Pi}^D + {}^A \dot{\Pi}^B$$

(1)

where C makes up the rotor B and the rigid body A

D is a fictitious rigid body whose mass distibution is equal to C but whose motion at any instant is that of A.

$$\text{in 7(c)} \quad {}^R \dot{\Pi}^S = {}^R \dot{\Pi}^{S*} + {}^R \dot{\Pi}^{S''}$$

was proven where  ${}^R \dot{\Pi}^{S''} = m_S {}^R \dot{\omega} \times {}^R S^*$

$$m_S = \sum m_i$$

$$\text{Thus } {}^A \dot{\Pi}^B = {}^A \dot{\Pi}^{B*} + {}^A \dot{\Pi}^{B''} \quad \text{where } {}^A \dot{\Pi}^{B''} = m_B {}^R \dot{\omega} \times {}^A B^*$$

now since the axis of symmetry of B is fixed in A and  $B^*$  lies on the axis of symmetry and is also fixed in A then  ${}^A \dot{\omega}^{B*} = 0$

$$\text{Thus } {}^A \dot{\Pi}^B = {}^A \dot{\Pi}^{B*} \quad (2)$$

Placing (2) in (1) leads to

$${}^R \dot{\Pi}^C = {}^R \dot{\Pi}^D + {}^A \dot{\Pi}^{B*} \quad (3)$$

$$\text{but by problem 8(b)} \quad {}^R \dot{\Pi}^D = - \frac{d}{dt} {}^R \dot{\Pi}^C$$

Thus we need to differentiate (3) wrt time in reference frame R.

$$\text{thus } \frac{d}{dt} {}^R \dot{\Pi}^C = {}^R \dot{\Pi}^{D*} = \frac{d}{dt} {}^R \dot{\Pi}^D + \frac{d}{dt} {}^A \dot{\Pi}^{B*} \quad (4)$$

$$\text{but } \frac{d}{dt} {}^A \dot{\Pi}^{B*} = \frac{d}{dt} {}^A \dot{\Pi}^{B*} + \frac{d}{dt} {}^A \dot{\omega} \times {}^A B^* \quad (5)$$

$$= - {}^A \dot{\Pi}^B + {}^A \dot{\omega} \times (\mathbb{I}^{B/B*}, {}^A B) \quad (6)$$

$$\frac{d}{dt} {}^R \dot{\Pi}^B = \mathbb{I}^{B/B*} \cdot {}^R \dot{\omega} + {}^A \dot{\omega} \times (\mathbb{I}^{B/B*}, {}^A B) + {}^A \dot{\omega} \times (\mathbb{I}^{B/B*}, {}^A B) \quad (7)$$

$$\text{Now } \frac{d}{dt} ({}^R \dot{\Pi}^D) = \mathbb{I}^{D/D*} \cdot {}^R \dot{\omega} + {}^A \dot{\omega} \times (\mathbb{I}^{D/D*}, {}^R \dot{\omega}) = - {}^R \dot{\Pi}^D \quad (8)$$

but since D represents a fictitious rigid body whose mass distribution is that of C but whose motion at any instant is that of A, then

$$\alpha^D = {}^R \dot{\omega}^A, {}^R \dot{\omega} = {}^A \dot{\omega} \text{ and } \mathbb{I}^{D/D*} = \mathbb{I}^{C/C*} \quad (9, 10, 11)$$

Placing these into (8) give

$${}^R \dot{\Pi}^D = \mathbb{I}^{C/C*} \cdot {}^R \dot{\omega} + {}^A \dot{\omega} \times (\mathbb{I}^{C/C*}, {}^R \dot{\omega}) \quad (12)$$

Using (12), (7) into (4)

$${}^R \dot{\Pi}^C = {}^A \dot{\Pi}^B + {}^R \dot{\Pi}^D + (\mathbb{I}^{B/B*}, {}^A B) \times {}^A \dot{\omega}$$

$$= (\mathbb{I}^{C/C*}, {}^R \dot{\omega}) \times {}^A \dot{\omega} + \mathbb{I}^{C/C*} \cdot {}^R \dot{\omega} + (\mathbb{I}^{B/B*}, {}^A B) \times {}^A \dot{\omega} - \mathbb{I}^{B/B*} \cdot {}^A \dot{\omega} + (\mathbb{I}^{B/B*}, {}^A B) \times {}^A \dot{\omega}$$

We can actually show that

$$\frac{d \mathbb{H}^{P/C}}{dt} = \frac{d}{dt} \sum m_i \frac{r_i/c}{\omega^A} \times \omega^A \times \frac{r_i/c}{\omega^A}$$
$$= \sum m_i \frac{d r_i}{dt} \times \omega^A + \frac{\mathbb{I}^{P/C} \cdot \omega^A}{\omega^A} + \sum m_i r_i \times \omega^A \times \frac{d r_i}{dt}$$

and that this becomes  $\sum m_i \left\{ (r_i \cdot r_i) \omega^A - (r_i \cdot \omega) r_i + (r_i \cdot r_i) \omega - (r_i \cdot \omega) r_i \right\} = \mathbb{I}^{P/C} \cdot \omega^A$

$$\sum m_i \left\{ 2(r_i \cdot r_i) \omega^A - \omega \cdot (r_i \cdot r_i + r_i \cdot r_i) \right\}$$

now  $\frac{d r_i}{dt} = \dot{r}_i = \frac{d r_i}{dt} + \frac{\omega^C}{\omega^A} \times r_i$  but  $r_i = r_i^{P/C}$  & are fixed in C thus  
 $\dot{r}_i \cdot r_i = (\omega^C \times r_i) \cdot r_i = (r_i \times r_i) \cdot \omega^C = 0$

thus we have  $-\sum m_i \frac{\omega^A}{\omega^A} \cdot \frac{d}{dt} (r_i \cdot r_i) = + \frac{\omega^A}{\omega^A} \cdot \frac{d}{dt} \sum m_i (r_i^2 U - r_i \cdot r_i)$   
 $\frac{\omega^A}{\omega^A} \cdot \frac{d}{dt} \mathbb{I}^{P/C} =$

## Numerical Solution of Initial Value Problems for Dynamics.

### INTRODUCTION

The initial value problem in dynamics consists of determining motion, space, position relative to time, of a system for time  $t \geq t_0$ , when the values of some initial conditions at time  $t_0$  are known. Frequently, the solution of this problem involves the numerical integration of a set of ordinary differential equations. For example, Fig. 1 shows a system composed of a rigid body  $B$  and a particle  $P$  that moves in a tube  $X$  and is attached to  $B$  with a spring  $\tau$  and a damper  $D$ . For  $t = 0$ , the orientation of  $B$ , the angular velocity of

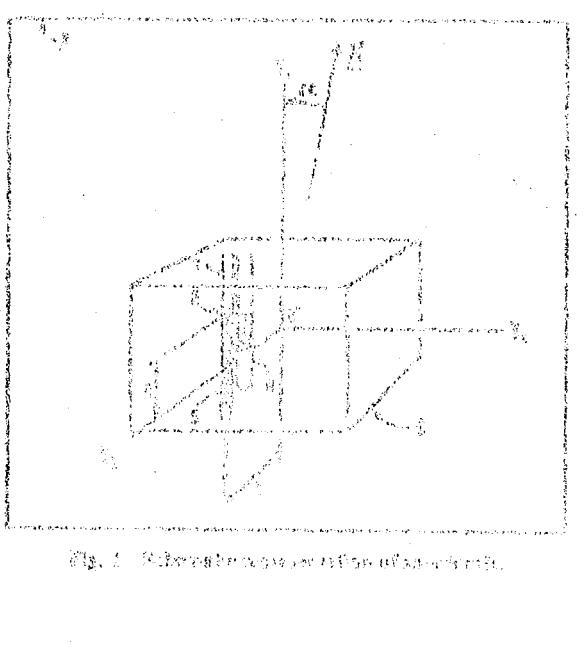
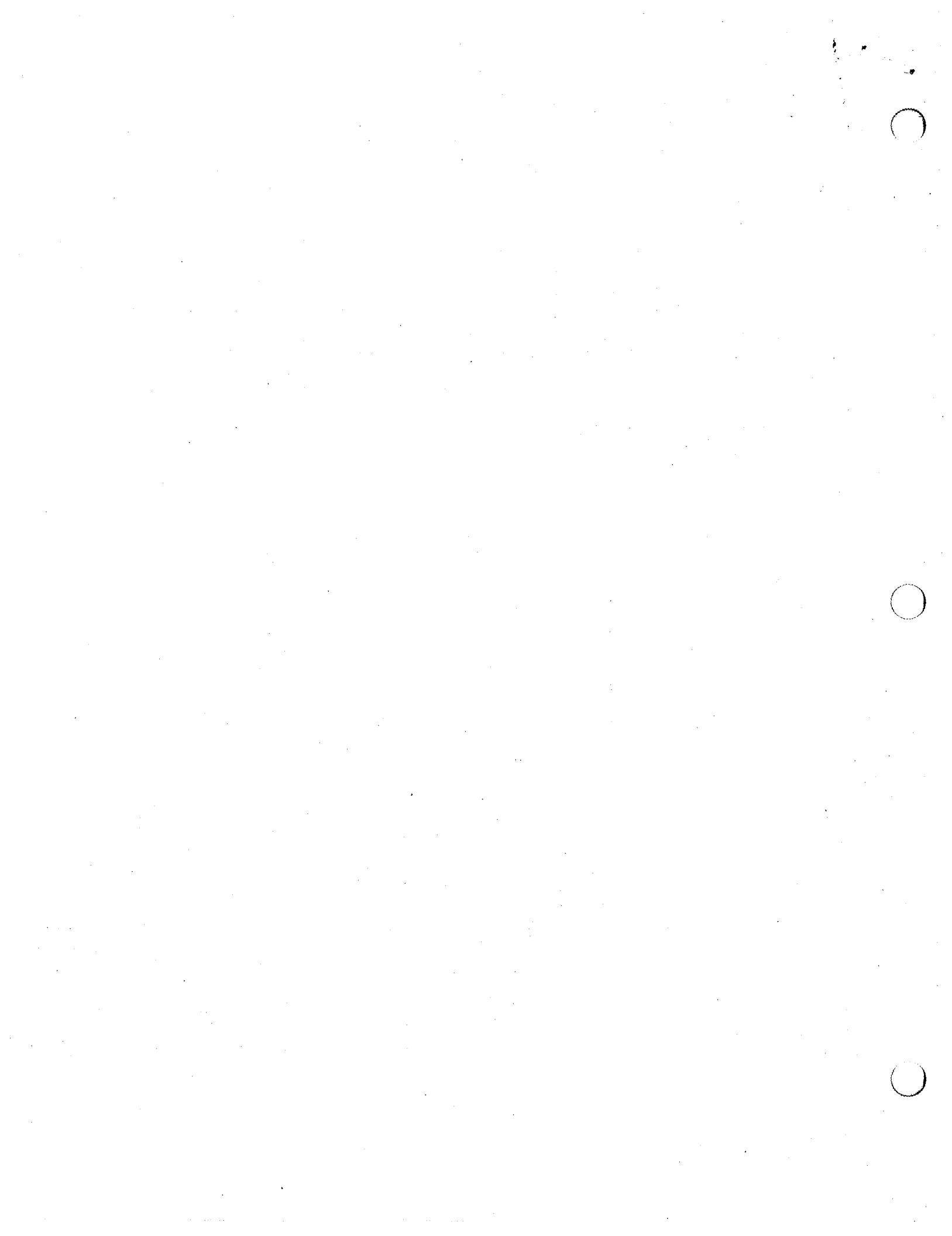


Fig. 1. Kinematic representation of a system.

$B$ , the position of  $P$  is  $\theta$ , and the velocity of  $P$  in  $X$  are known, and to determine the position of each at  $t > 0$ . Here  $\theta$  is the angle measured  $\theta_0$



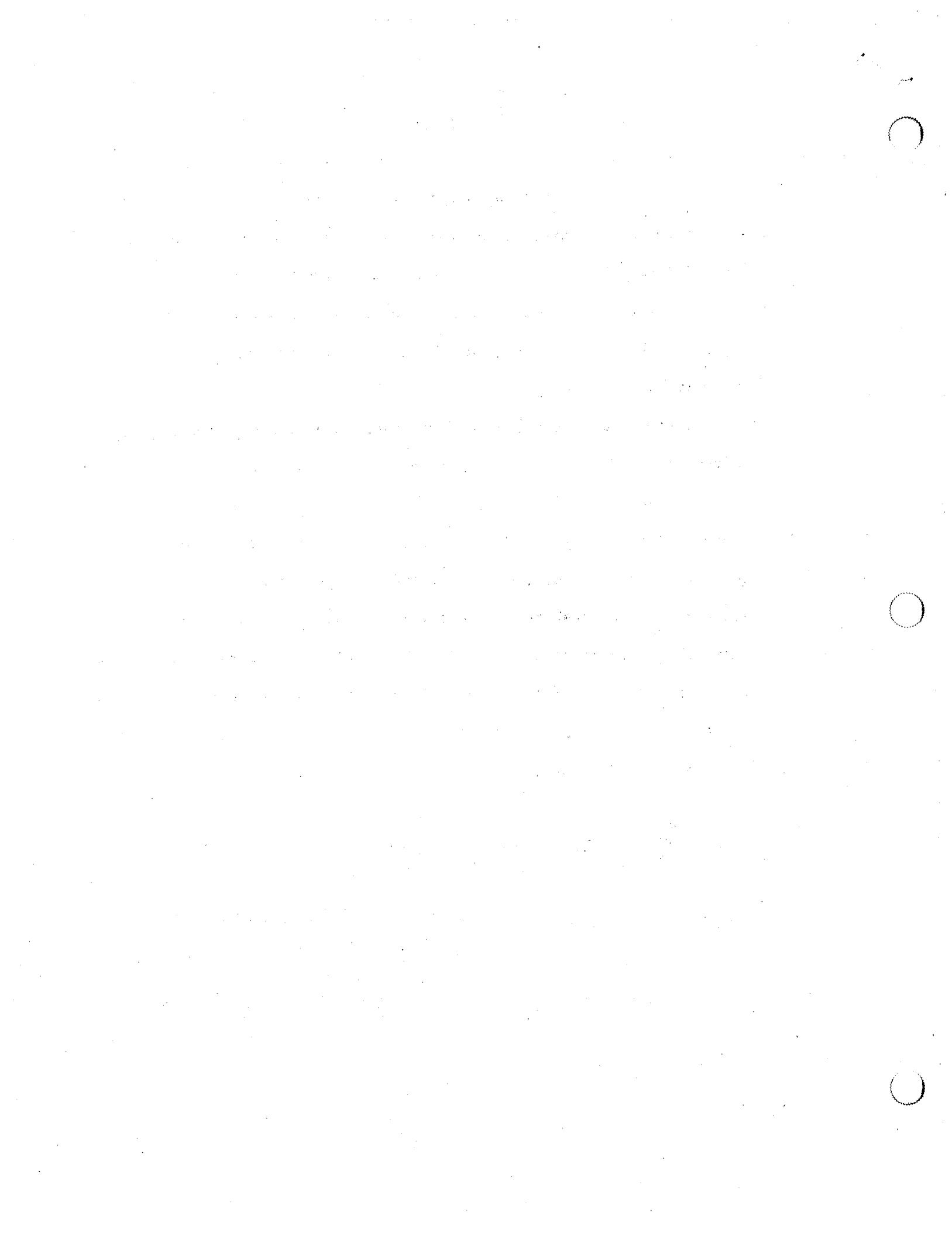
of the central principal axes of inertia of  $B$ , and an inertially fixed vector, namely  $\mathbf{H}$ , the central angular momentum of the system, and  $q$  is the distance from  $P$  to  $X_3$ , another central principal axis of inertia of  $B$ . It is the purpose of what follows to discuss the numerical solution of such problems.

The sequel is arranged as follows: Sec. 2 deals with the construction of a computational algorithm. In Sec. 3, a computer program based on the algorithm is discussed, and Sec. 4 contains a sample listing of the program.

### 2. Construction of an algorithm

Use of principles of mechanics leads to equations called dynamical equations (often differential equations). These govern certain scalar variables and involve certain system parameters. For example, use of  $\frac{d^2\mathbf{r}}{dt^2} + \mathbf{F} = 0$  (see DYNAMICS, p.177) in connection with the system mentioned in Sec. 1 leads to four differential equations governing  $v_1$ ,  $v_2$ ,  $v_3$ , and  $v_4$ , the first three of which are the  $b_1$ ,  $b_2$ ,  $b_3$  measure numbers of the angular velocity of  $B$ , where  $b_1$ ,  $b_2$ ,  $b_3$  are unit vectors parallel to  $X_1$ ,  $X_2$ ,  $X_3$ , while  $v_i$  is the time-rate of change of  $q$ , the distance between  $P$  and  $X_2$ ; the four differential equations involve the system parameters  $b$ ,  $\sigma$ ,  $\delta$ ,  $m_p$ ,  $m_B$ ,  $I_1$ ,  $I_2$ ,  $I_3$  which characterize  $S$ ,  $D$ ,  $P$ , and  $B$ . They are

$$\begin{aligned} b_1 &= \left\{ \left( -\frac{m_p m_B}{m_p + m_B} q^2 + I_2 \right) \left( \frac{m_p m_B}{m_p + m_B} b (v_1 v_3 q + v_2 v_3 b \right. \right. \\ &\quad \left. \left. - 2v_2 v_4) - v_2 v_3 (I_2 - I_3) \right) + \frac{m_p m_B}{m_p + m_B} b q \left( \frac{m_p m_B}{m_p + m_B} q (v_1 v_3 q \right. \right. \\ &\quad \left. \left. + v_2 v_3 b - 2v_2 v_4) + v_3 v_1 (I_3 - I_1) \right) \right\} \left( \frac{m_p m_B}{m_p + m_B} (q^2 I_1 \right. \\ &\quad \left. + b^2 I_2) + I_1 I_2 \right)^{-1} \end{aligned} \quad (1)$$



$$\begin{aligned}\dot{u}_2 = & \left\{ -\frac{m_p m_B}{m_p + m_B} b q \left( \frac{m_p m_B}{m_p + m_B} b (u_1 u_3 q + u_2 u_3 b - 2 u_2 u_4) \right. \right. \\ & - u_2 u_3 (I_2 - I_3) + \left( \frac{m_p m_B}{m_p + m_B} b^2 + I_2 \right) \left( \frac{m_p m_B}{m_p + m_B} q (u_1 u_3 q \right. \\ & \left. \left. + u_2 u_3 b - 2 u_2 u_4) + u_3 u_1 (I_3 - I_2) \right) \right\} \frac{m_p m_B}{m_p + m_B} (q^2 I_1 \\ & + b^2 I_2) + I_1 I_2 \end{aligned} \quad (2)$$

$$\begin{aligned}\dot{u}_3 = & [(cq + \delta u_4) b + \frac{m_p m_B}{m_p + m_B} q (2u_3 u_4 - u_3^2 b - u_1^2 b + u_1 u_2 q)] \\ & - u_1 u_2 (I_1 - I_2) \left( \frac{m_p m_B}{m_p + m_B} q^2 + I_3 \right)^{-1} \end{aligned} \quad (3)$$

$$\dot{u}_4 = - (cq + \delta u_4) \left( \frac{m_p m_B}{m_p + m_B} \right)^{-1} + \dot{u}_3 b - u_2 (u_1 b - u_2 q) + u_3^2 q \quad (4)$$

Kinematical considerations furnish additional equations, called kinematical equations, some of which may be differential equations. For example,

$$\dot{q} = u_4 \quad (5)$$

is such an equation and, after introducing Euler parameters  $e_1, \dots, e_4$ , one can write, in addition,



$$\dot{e}_1 = \frac{1}{2} (u_1 e_4 - u_2 e_3 + u_3 e_2) \quad (6)$$

$$\dot{e}_2 = \frac{1}{2} (u_1 e_3 + u_2 e_4 - u_3 e_1) \quad (7)$$

$$\dot{e}_3 = \frac{1}{2} (-u_1 e_2 + u_2 e_1 + u_3 e_4) \quad (8)$$

$$\dot{e}_4 = \frac{1}{2} (-u_1 e_1 - u_2 e_2 - u_3 e_3) \quad (9)$$

Furthermore, after defining  $\alpha$  and  $\beta$  as

$$e^2 \triangleq 2 e_2^2 + 2 e_3^2 \quad (10)$$

and

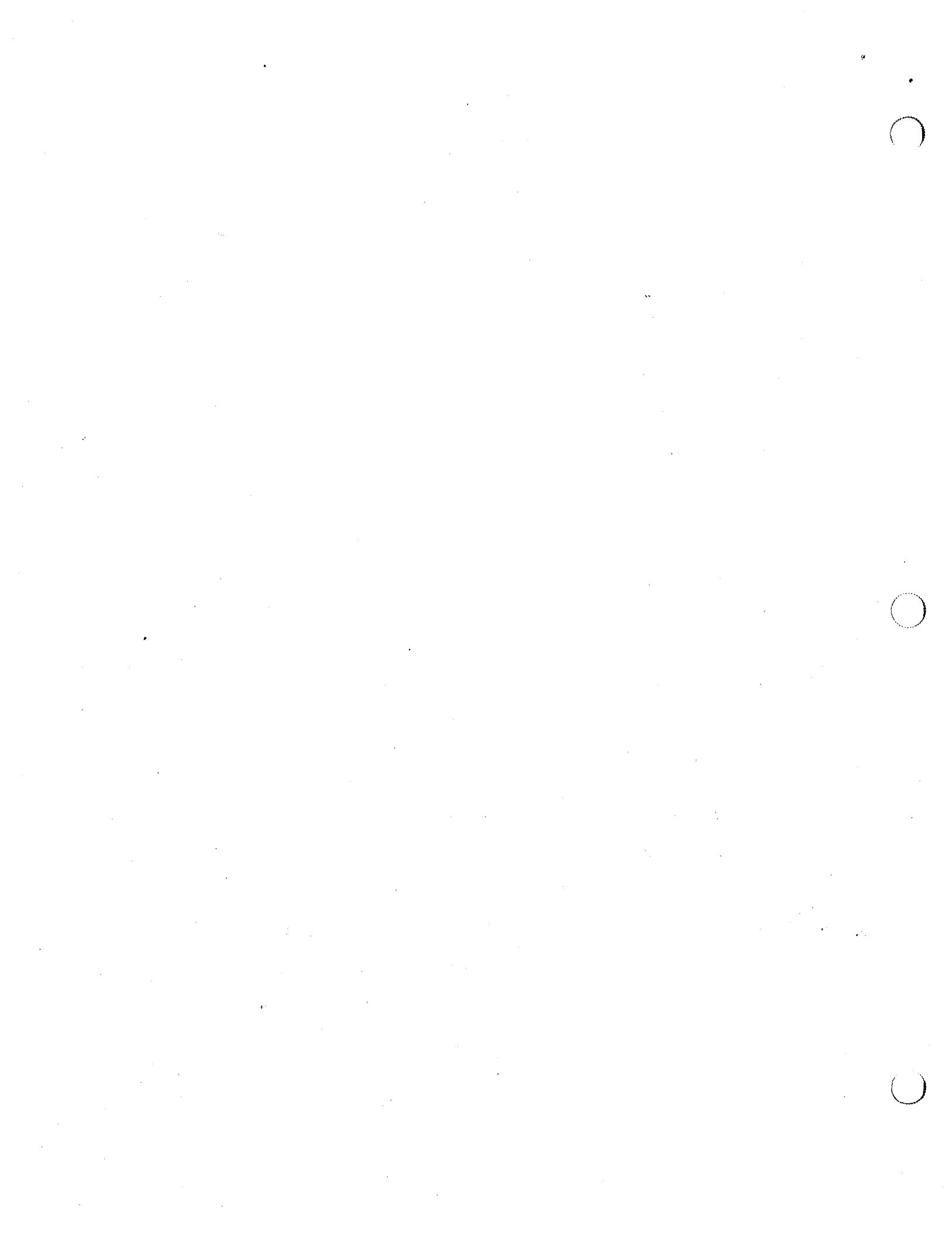
$$\beta \triangleq (1 - e^2)^{1/2} \quad (11)$$

one can express  $\alpha$  as

$$\alpha = \tan^{-1}(\beta/c) \quad (12)$$

Simultaneous solution of (1)-(9) and subsequent use of (10)-(12) leads to the desired values of  $\alpha$  and  $\beta$ . However, a program based on the equations as written would be inefficient because it would involve certain repetitive calculations. For example, since the expression  $m_p u_p / (m_p + m_p)$  appears repeatedly, it is desirable to introduce a symbol to replace this expression, i.e., to let

$$y = m_p u_p / (m_p + m_p) \quad (13)$$



Similarly, one may introduce  $x_1, \dots, x_9$  as

$$x_1 = u_1 u_3 q + u_2 u_3 b - 2 u_2 u_4 \quad (14)$$

$$x_2 = u_2 u_3 (I_2 - I_3) \quad (15)$$

$$x_3 = u_3 u_1 (I_3 - I_1) \quad (16)$$

$$x_4 = \mu b q \quad (17)$$

$$x_5 = \{\mu(q^2 I_1 + b^2 I_2) + I_1 I_2\}^{-1} \quad (18)$$

$$x_6 = \mu b x_1 - x_2 \quad (19)$$

$$x_7 = \mu q x_1 + x_3 \quad (20)$$

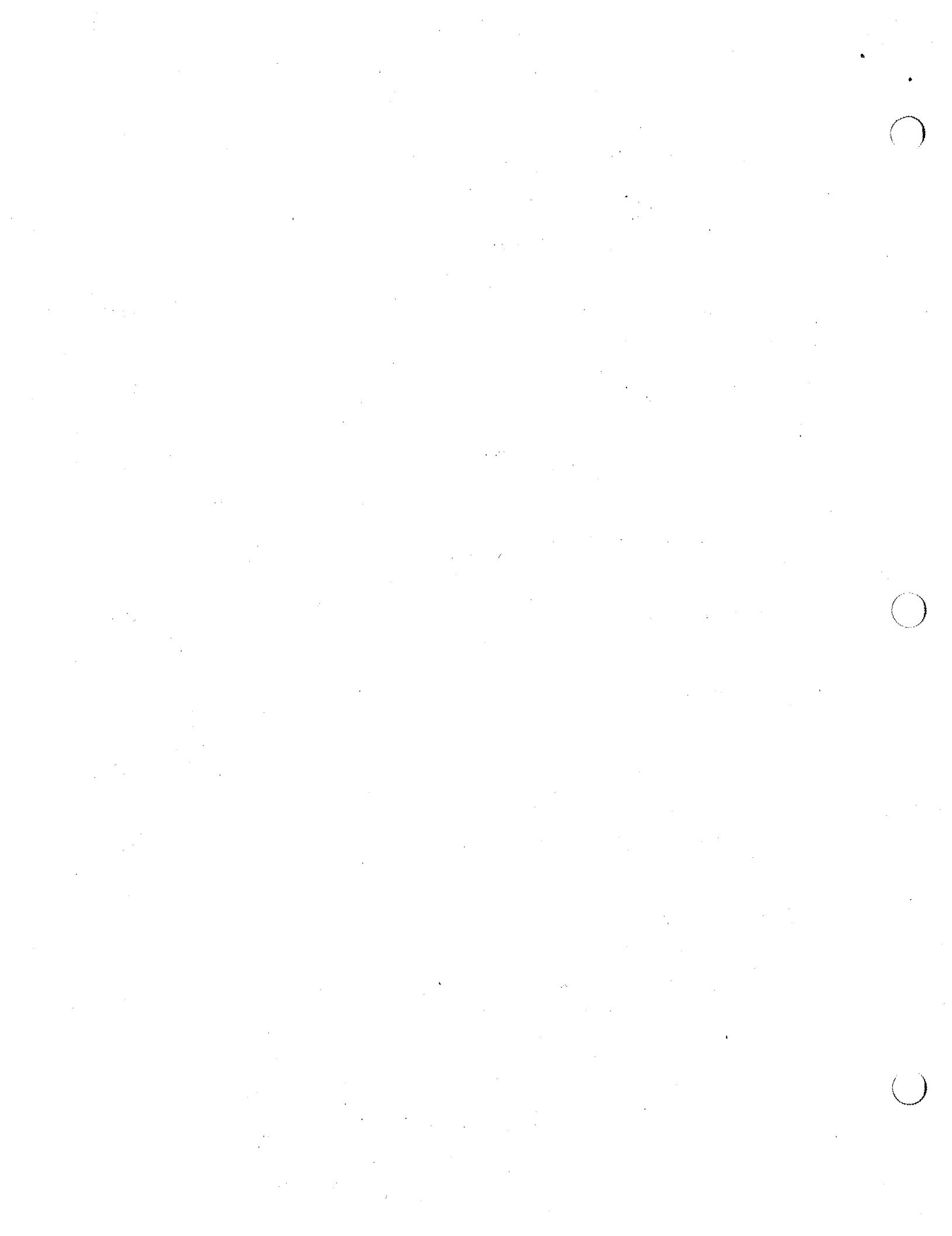
$$x_8 = \sigma q + \delta u_4 \quad (21)$$

$$x_9 = 2u_3 u_4 - u_3^2 b - u_1^2 b + u_1 u_2 q \quad (22)$$

after which one can replace (1) - (4) with

$$\dot{u}_1 = [-(\mu q^2 + I_2)x_6 + x_4 x_7]x_5 \quad (23)$$

$$\dot{u}_2 = [-x_4 x_6 + (\mu b^2 + I_1)x_7]x_5 \quad (24)$$



$$u_3 = -[z_6 b + \mu q z_9 - u_1 u_2 (I_1 - I_2)] / (\mu q^2 + I_3) \quad (25)$$

$$u_4 = -z_6/\mu + u_3 b = u_2(u_1 b - u_2 q) + u_3^2 q \quad (26)$$

### 3. Computer program

If a set of ordinary differential equations can be expressed as

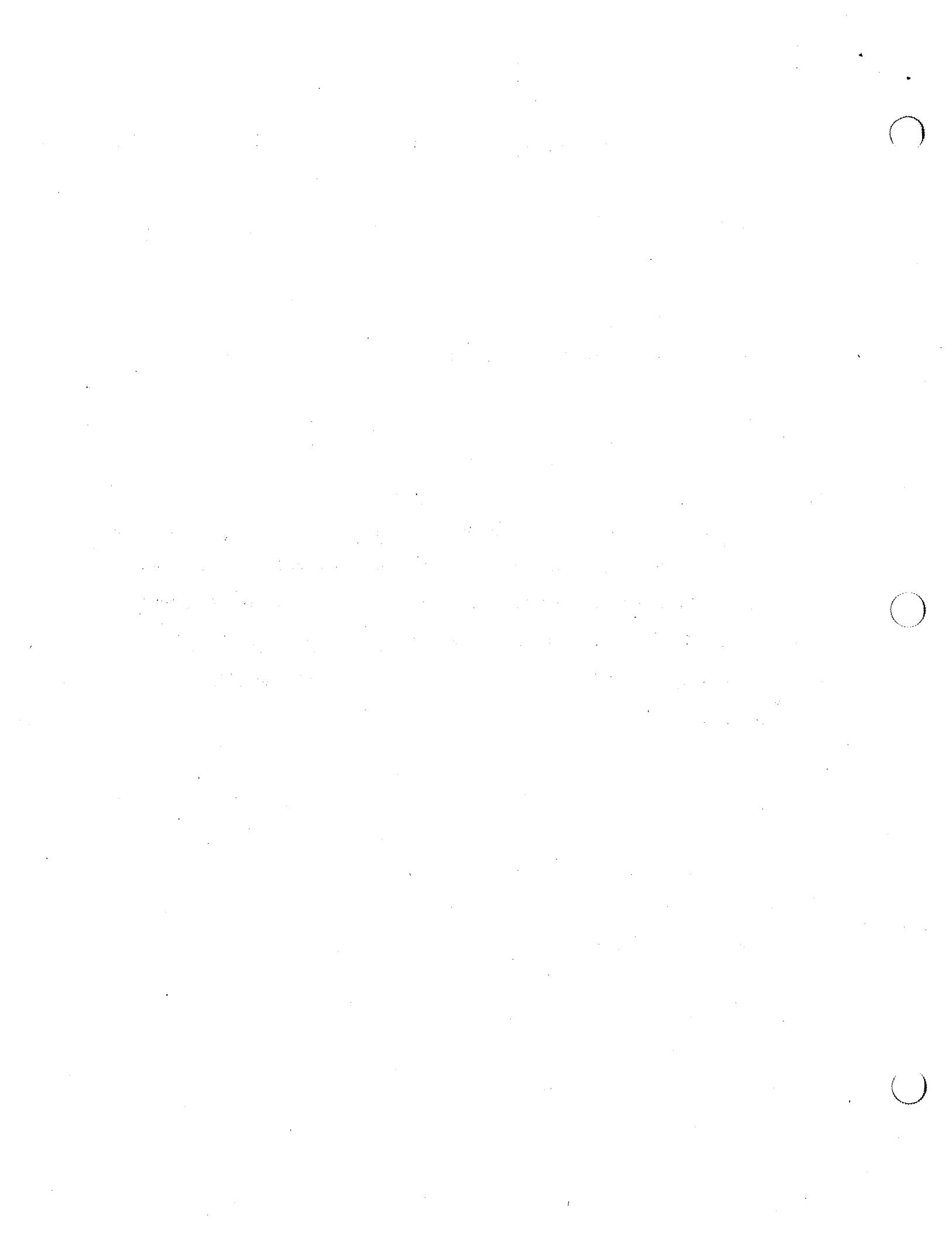
$$\frac{dy_i}{dx} = F_i(z_1, \dots, z_m; y_1, \dots, y_n, x) \quad (i=1, \dots, n)$$

where  $z_j$  ( $j=1, \dots, m$ ) is a function of  $y_1, \dots, y_n$  and  $y_k$  ( $k=1, \dots, n$ ) is a function of  $x$ , then the FORTRAN subroutine DFEQKM can be used to integrate the equations. (This subroutine is based on the KUTTA MERSON method as described in Fox, L., Numerical Solution of Ordinary and Partial Differential Equations, 1962, p. 24.) (For example, (5)-(9) and (23)-(26) form such a set with

$$y_i = \begin{cases} u_i & i = 1, \dots, 4 \\ q & i = 5 \\ \epsilon_{1-5} & i = 6, \dots, 9 \end{cases} \quad (27)$$

and

$$F_1 = [-(\mu q^2 + I_2)z_6 + z_4 z_7]z_5 \quad (28)$$



To use DFEQNM, one creates a FORTRAN program like the one in Sec. 4. An explanation of this program and a sample output can be obtained from LOGS by giving the TOPS-20 command

```
@ PRINT PS:(K.KANE)DOC..1
```

Execution of the program is initiated by giving the command

```
@ EXEC PS:(K.KANE)DYNEX.FOR, NAL:NALIB/LIB
```

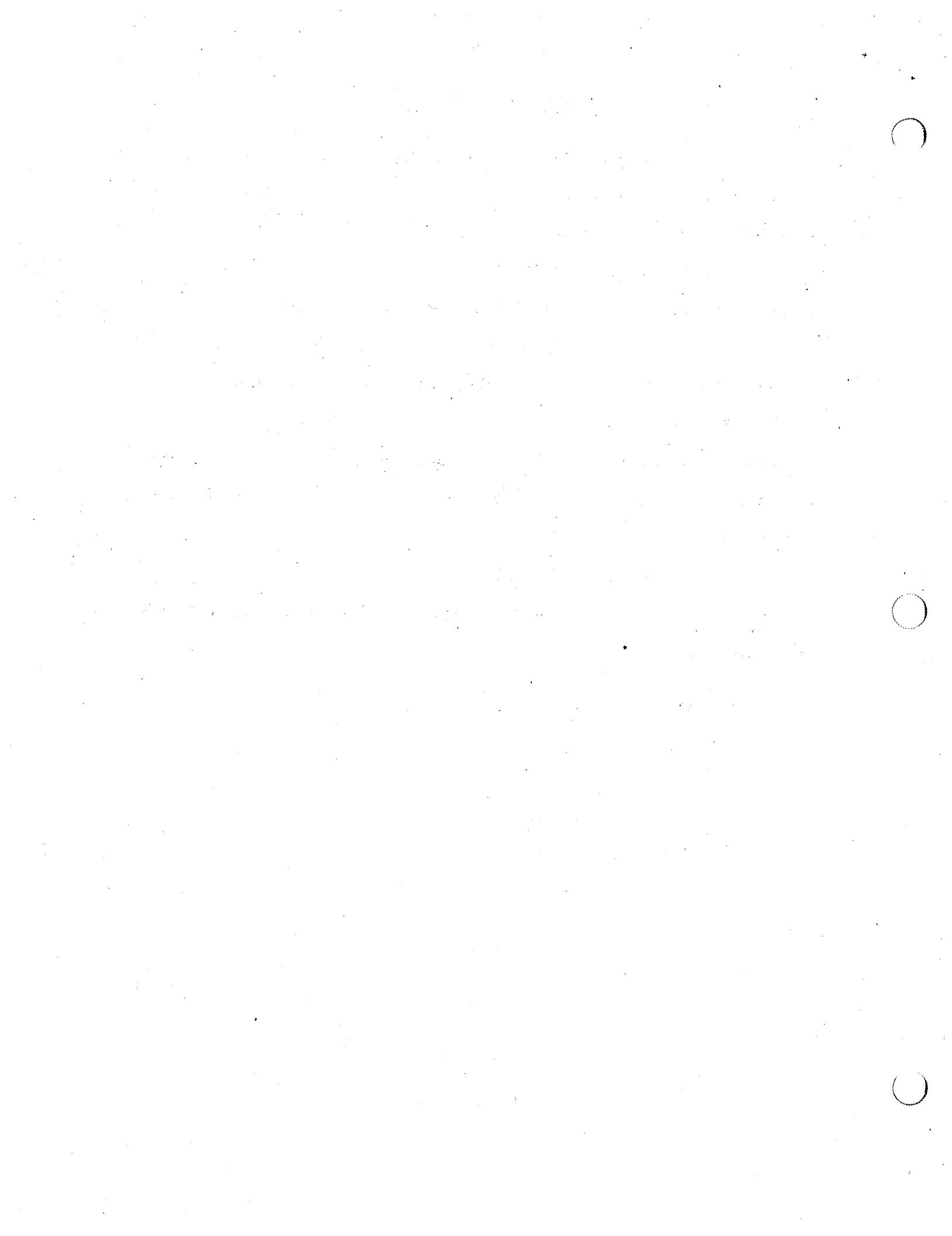
After execution, printed output is obtained by giving the command

```
@ PRINT K0R20.DAT
```

To create a program intended for the solution of an initial value problem other than the one involving (5)-(9) and (23)-(26), one can modify DYNEX.FOR by giving the commands

```
@ EDIT PS:(K.KANE)DYNEX.FOR
```

and then using editing commands to change or delete symbols underlined in the listing in Sec. 4.

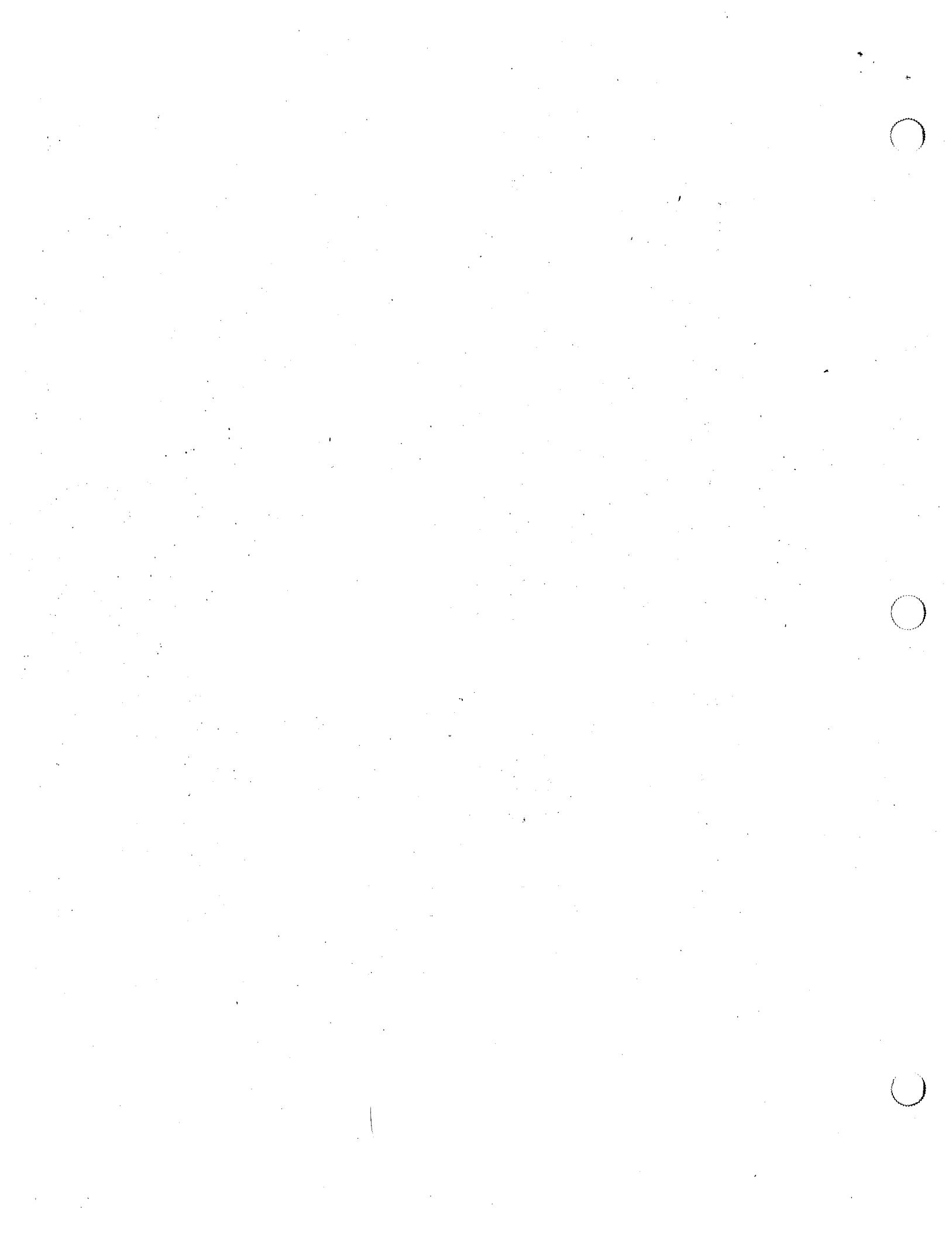


## 6. FORTRAN program

THE NAME OF THIS FILE IS DYNEX.FOR  
EXAMPLE PROBLEM

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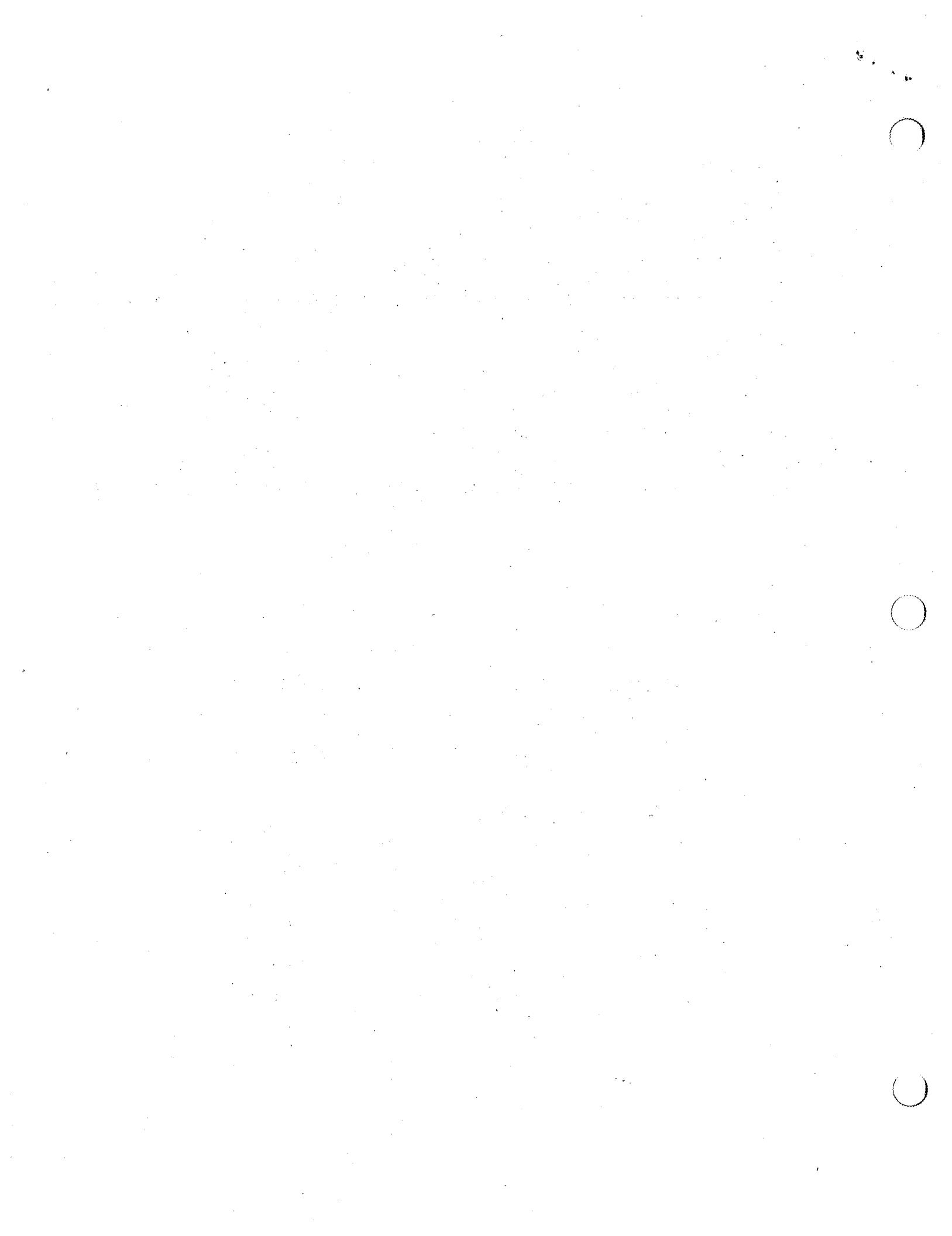


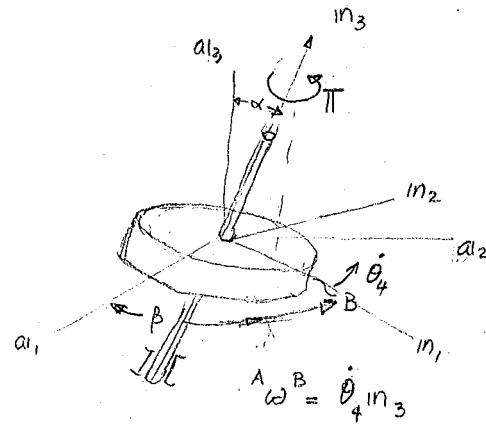
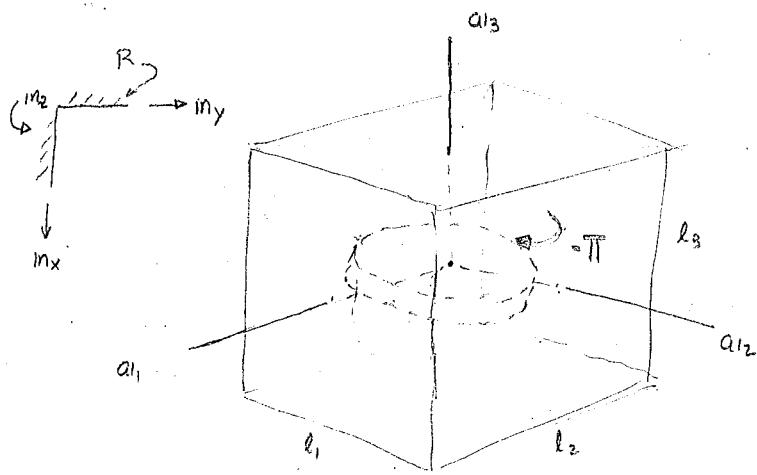
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Let  $x, y, z$  be the measure nos to the center of mass of the body A. Thus  $\overset{R}{\omega} = x \overset{m}{\omega}_x + y \overset{m}{\omega}_y + z \overset{m}{\omega}_z$ .

$$\overset{m}{\omega}^A = \dot{x} \overset{m}{\omega}_x + \dot{y} \overset{m}{\omega}_y + \dot{z} \overset{m}{\omega}_z$$

$$\overset{R}{\tau}^A = \overset{m}{\tau}^A$$

let  $a_1, a_2, a_3$  be the central axes of the body A fixed in A

let  $m_x, m_y, m_z$  be a set of axes fixed in reference frame R

thus we can define a set of intermediate reference planes  $A_1, A_2, \dots$  in which we can define simple angular motions

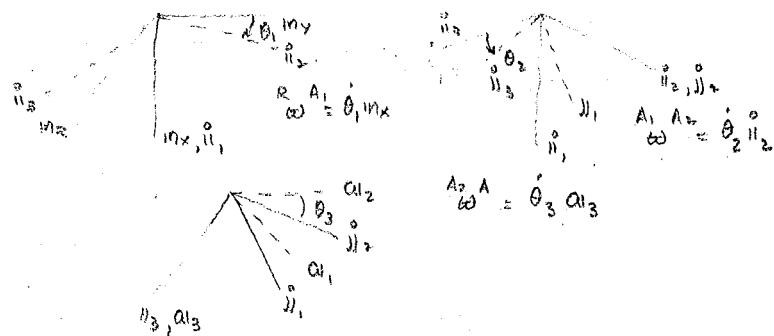
so that  $\overset{R}{\omega}^A = \overset{R}{\omega}^{A_1} + \overset{R}{\omega}^{A_2} + \overset{R}{\omega}^{A_3} = \dot{\theta}_1 \overset{m}{\omega}_x + \dot{\theta}_2 \overset{m}{\omega}_y + \dot{\theta}_3 \overset{m}{\omega}_z$

$$\overset{m}{\omega}_3 = a_{13} \cos \alpha + a_{11} \sin \alpha \cos \beta + a_{12} \sin \alpha \sin \beta$$

$$\overset{R}{\tau} = \overset{m}{\tau}_3$$

$$\overset{R}{\tau}^{B/B} = J_1 \overset{m}{\omega}_3 \overset{m}{\omega}_3 + J_2 (\overset{m}{\omega}_1 \overset{m}{\omega}_1 + \overset{m}{\omega}_2 \overset{m}{\omega}_2)$$

let us also assume that the rotor is made of the same material as the body A and that its center of mass lies at the same point of body A, ie that a cut out is made in body A of same size as body B in which we may place body B,



We will assume a parallel gravitational field. Let body A have mass  $(M-m)$  and let body B have mass  $m$ .

$$\text{thus } F_r = \overset{R}{\omega}^A \cdot \overset{R}{F} + \overset{R}{\omega}^A \cdot \overset{R}{\tau}^A + \overset{R}{\omega}^B \cdot \overset{R}{F} + \overset{R}{\omega}^B \cdot \overset{R}{\tau}^B$$

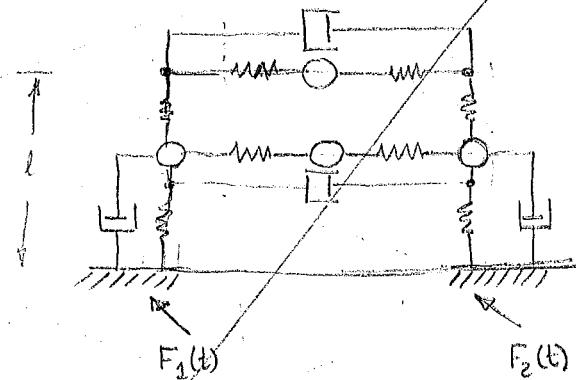
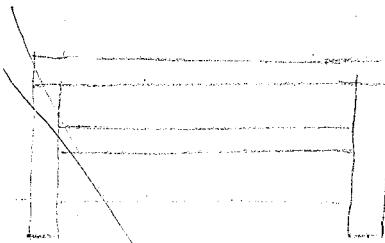
$$F_r = \overset{R}{\omega}^A \cdot \overset{R}{F} + \overset{R}{\omega}^A \cdot \overset{R}{\tau}^A + \overset{R}{\omega}^B \cdot \overset{R}{\tau}^B$$

$$\text{where } \overset{R}{\tau}^B = -M \overset{R}{\omega}^A$$

let C be the composite of A & B.

let D

$$\overset{A}{\omega}^B = \overset{A}{\omega}^B \beta = \overset{A}{\omega}^B \beta \sin \beta$$



A representation of a building by means of springs and mass to which forces at the bases are applied (representation of an earthquake wave passing the building & setting the building in motion). The attempt here is to find what values of the spring constants for the lower & upper floors necessary to minimize the motion. What I'm attempting to do is model a "shake" table with intent on finding the "building code" for this model.

This problem is quite practical in nature especially for an area such as California. The forces are modelled as semi  $\delta$ -functions which can be in fact periodic, yet of decreasing amplitude, to model reflections of P & S waves whose energy is dissipated as their travel paths through the earth's crust increase.

$$d(t) = e^{-\alpha t} \quad (\text{dissipation fn})$$

- My questions are: (1) should I model the floors by adding a light inextensible string if I want not only longitudinal but also transverse motion of the floors? As a first approximation I guess I can (to simplify model somewhat). (2) Is model too complicated for the purpose you intended?

- This is my project. This seems more like a linear vibrations problem than like an initial value problem to be solved numerically. I don't think it is suitable for our purposes.

$$O \quad \text{IF} \quad \text{A}^*$$

$$F_r = \sum_{i=1}^{N_1+N_2} W_{q_r}^{P_i} \cdot \text{IF}_i^* = \sum_{i=1}^{N_1} W_{q_r}^{P_i} \cdot \text{IF}_i^* + \sum_{i=1}^{N_2} W_{q_r}^{P_i} \cdot \text{IF}_i^*$$

$$\text{for body A } W_{q_r}^{P_i} = W_{q_r}^{A^*} + (\omega_{q_r}^A \times P_i) \quad \text{for body B} \Rightarrow W_{q_r}^{P_i} = W_{q_r}^{B^*} + (\omega_{q_r}^B \times P_i)$$

$$W_{q_r}^{A^*} \cdot \text{IF}_i^* = \sum \text{IF}_i^* (W \cdot w) \cdot m_y$$

$$W_{q_r}^{A^*} \cdot (M \cdot w) \cdot m_y$$

$$(\omega_{q_r}^A \times P_i) \cdot \text{IF}_i^* = (P_i \times \text{IF}_i^*) \cdot \omega_{q_r}^A = \omega_{q_r}^A \cdot \text{II}_i^*$$

$$W_{q_r}^{B^*} \cdot \sum m_i g m_y + \sum m_i g_{q_r}^B \times P_i \cdot m_y$$

$$W_{q_r}^{B^*} \cdot m g m_y + \sum (P_i \times \text{IF}_i^*) \cdot \omega_{q_r}^B$$

$$\sum_{i=N_1+N_2}^{N_1} W_{q_r}^{P_i} \cdot \text{IF}_i^* \Rightarrow \sum_{i=1}^{N_1} W_{q_r}^{P_i} \cdot \text{IF}_i^* \quad \text{where } \text{IF}_i^* = -m_i \text{a}_i \\ = -m_i [\alpha_i^* + \omega_{q_r}^A \times P_i + \omega_{q_r}^B \times (\omega_{q_r}^A \times P_i)]$$

$$= W_{q_r}^{A^*} \cdot \left[ (M - m) \right] \alpha_i^* + \omega_{q_r}^A \cdot \text{II}_i^* \quad \text{where } \text{II}_i^* = -\sum m_i \text{r}_i \times \alpha_i^* \\ = \left[ (\text{II}^{A/A^*}, \omega_{q_r}^A) \times \omega_{q_r}^A - \text{II}^{A/A^*}, \omega_{q_r}^A \right]$$

$$\sum_{i=1}^{N_2} W_{q_r}^{P_i} \cdot \text{IF}_i^* = -W_{q_r}^{A^*} \cdot \left[ \frac{w}{g} \alpha_i^* \right] + \omega_{q_r}^B \cdot \left[ (\text{II}^{B/B^*}, \omega_{q_r}^B) \times \omega_{q_r}^B - \text{II}^{B/B^*}, \omega_{q_r}^B \right]$$

$$\text{Thus we get } -W_{q_r}^{A^*} \cdot M \alpha_i^* + \omega_{q_r}^A \cdot \text{II}_i^* + \omega_{q_r}^B \cdot \text{II}_i^*$$

$$\text{but look at } \omega_{q_r}^A \cdot \left( \text{II}^{A/A^*} + \text{II}^{B/B^*} \right) + \frac{\text{II}^{A/A^*}, \omega_{q_r}^B}{\text{d} \text{II}^{A/A^*} / \text{dt}} \cdot \text{II}^{B/B^*} \\ = \frac{\text{d} \text{II}^{A/A^*} / \text{dt}}{\text{d} \text{II}^{B/B^*} / \text{dt}} \cdot \text{II}^{B/B^*} = -\frac{\text{d} (\text{II}^{A/A^*} + \text{II}^{B/B^*})}{\text{d} t} = \text{II}^{A/A^*} + \text{II}^{B/B^*}$$

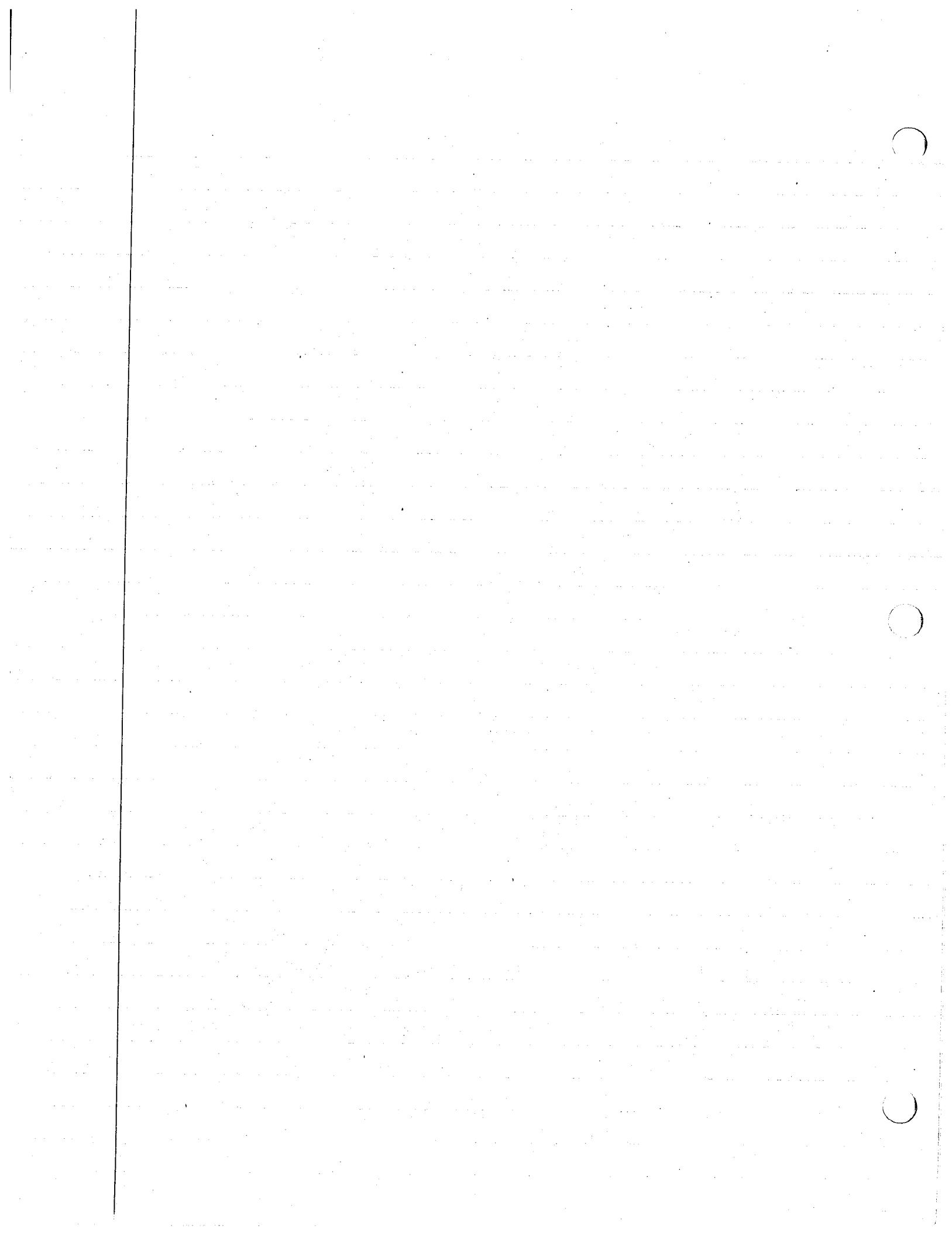
$$\text{but } \text{II}^{A/A^*} = \text{II}^{A/A^*}, \omega_{q_r}^A; \quad \text{II}^{B/B^*} = \text{II}^{B/B^*}, \omega_{q_r}^B = \text{II}^{B/B^*}, \omega_{q_r}^A + \text{II}^{B/B^*}, \omega_{q_r}^B \\ \text{II}^{A/A^*} + \text{II}^{B/B^*} = \left[ \text{II}^{A/A^*}, \omega_{q_r}^A \right] + \text{II}^{B/B^*}, \omega_{q_r}^B$$

Now define a fictitious rigid body D whose mass distribution is that of C but whose motion is that of A. Thus  $\text{II}^{D/D^*} + \text{II}^{B/B^*}$

$$\frac{\text{d} (\text{II}^{A/A^*} + \text{II}^{B/B^*})}{\text{dt}} = \frac{\text{d} \text{II}^{A/A^*}}{\text{dt}} + \frac{\text{d} \text{II}^{B/B^*}}{\text{dt}} + \frac{\text{d} \text{II}^{B/B^*}}{\text{dt}} + \omega_{q_r}^A \times \text{II}^{B/B^*} \\ = \left[ \text{II}^{A/A^*}, \omega_{q_r}^B \right] = \frac{\text{d} \text{II}^{D/D^*}}{\text{dt}} - \text{II}^{B/B^*} + \omega_{q_r}^A \times \left( \text{II}^{B/B^*}, \omega_{q_r}^B \right)$$

$$\text{but } \frac{\text{d} \text{II}^{D/D^*}}{\text{dt}} = \text{II}^{C/C^*}, \omega_{q_r}^A + \omega_{q_r}^A \times \text{II}^{C/C^*}, \omega_{q_r}^A$$

$$\text{thus } \text{II}^{A/A^*}, \omega_{q_r}^B = \left( \text{II}^{C/C^*}, \omega_{q_r}^A \right) \times \omega_{q_r}^A = \text{II}^{C/C^*}, \omega_{q_r}^A + \text{II}^{B/B^*} + \text{II}^{A/A^*}, \omega_{q_r}^A$$



$$\text{Thus } F_r^* = W_{q_r} \cdot \frac{W_{q_r}}{g} + \omega_{q_r}^A \cdot \bar{\Pi}^B + \omega_{q_r}^B \cdot \bar{\Pi}^A = \left[ (\bar{\Pi}^{C*}, \bar{\Pi}^A) \times \bar{\omega}^A + \bar{\Pi}^{C*} \cdot \bar{\omega}^A + \bar{\Pi}^B \cdot \bar{\omega}^B \right]$$

$$\text{Now back to } F_r = W_{q_r}^A \cdot \bar{\Pi}^A + W_{q_r}^B \cdot \bar{\Pi}^B + \omega_{q_r}^A \cdot \bar{\Pi}^B + \omega_{q_r}^B \cdot \bar{\Pi}^A = W_{q_r} \cdot M g m_y + \omega_{q_r}^A \cdot \bar{\Pi}^A + \omega_{q_r}^B \cdot \bar{\Pi}^B$$

$$\bar{\omega}_{\dot{x}}^C = \bar{m}_x \quad \bar{\omega}_{\dot{y}}^C = \bar{m}_y \quad \bar{\omega}_{\dot{z}}^C = \bar{m}_z \quad \bar{\omega}_{\dot{\theta}_1}^C = \bar{\omega}_{\dot{\theta}_2}^C = \bar{\omega}_{\dot{\theta}_3}^C = \bar{\omega}_{\dot{\theta}_4}^C = 0 \quad \bar{\Pi}^A = -\bar{\Pi} m_3 \quad \bar{\Pi}^B = \bar{\Pi} m_3$$

$$\bar{\omega}_{\dot{x}}^A = \bar{\omega}_{\dot{y}}^A = \bar{\omega}_{\dot{z}}^A = 0 \quad \bar{\omega}_{\dot{\theta}_1}^A = \bar{\omega}_{\dot{\theta}_2}^A = \bar{\omega}_{\dot{\theta}_3}^A = \bar{\omega}_{\dot{\theta}_4}^A = 0$$

$$\bar{\omega}_{\dot{x}}^B = \bar{\omega}_{\dot{y}}^B = \bar{\omega}_{\dot{z}}^B = 0 \quad \bar{\omega}_{\dot{\theta}_1}^B = \bar{m}_x \quad \bar{\omega}_{\dot{\theta}_2}^B = \bar{m}_2 \quad \bar{\omega}_{\dot{\theta}_3}^B = \bar{m}_3$$

$$\bar{\omega} = \bar{\omega} + \bar{\omega} = \bar{\omega}_4 m_3 = \bar{\theta}_1 \bar{m}_x + \bar{\theta}_2 \bar{m}_2 + \bar{\theta}_3 \bar{m}_3 + \bar{\theta}_4 \bar{m}_3$$

$$\bar{\omega}_{\dot{x}}^C = \bar{\omega}_{\dot{y}}^C = \bar{\omega}_{\dot{z}}^C = 0 \quad \bar{\omega}_{\dot{\theta}_1}^C = \bar{m}_3 \quad \bar{\omega}_{\dot{\theta}_2}^C = \bar{m}_y \quad \bar{\omega}_{\dot{\theta}_3}^C = \bar{m}_2 \quad \bar{\omega}_{\dot{\theta}_4}^C = \bar{m}_3$$

	$m_x$	$m_y$	$m_z$	$\bar{m}_1$	$\bar{m}_2$	$\bar{m}_3$	$\bar{m}_1$	$\bar{m}_2$	$\bar{m}_3$	$a_1$	$a_2$	$a_3$			
$\bar{m}_1$	1	0	0	$\bar{m}_1$	$c_2$	0	$-s_2$	$a_1$	$c_3$	$s_3$	0	$m_1$	$c_3 c_\alpha$	$s_3 c_\alpha$	$-s_\alpha$
$\bar{m}_2$	0	$c_1$	$s_1$	$\bar{m}_2$	0	1	0	$a_2$	$-s_3$	$c_3$	0	$m_2$	$-s_\beta$	$c_\beta$	0
$\bar{m}_3$	0	$-s_1$	$c_1$	$\bar{m}_3$	$s_2$	0	$c_2$	$a_3$	0	0	1	$m_3$	$s_\alpha c_\beta$	$s_\alpha s_\beta$	$c_\alpha$

$$q_1 = x \quad F_1 = W_{q_1} + 0 + 0 = W$$

$$q_2 = y \quad F_2 = 0 + 0 + 0 = 0$$

$$q_3 = z \quad F_3 = 0 + 0 + 0 = 0$$

$$q_4 = \theta_1 \quad F_4 = 0 + -T m_x \cdot m_3 + T \bar{m}_x \cdot \bar{m}_3 = T(c_1 \bar{m}_1 - s_3 \bar{m}_2 + s_2 \bar{m}_3) = T(c_1 \bar{m}_1 - s_1 (-s_2 \bar{m}_1 + c_2 \bar{m}_3)) = T(c_1 \bar{m}_1 + s_1 s_2 \bar{m}_1 - s_1 c_2 \bar{m}_3)$$

$$q_5 = \theta_2 \quad F_5 = 0 + -T \bar{m}_2 \cdot m_3 + T \bar{m}_2 \cdot \bar{m}_3 = T(s_3 \bar{m}_1 + c_3 \bar{m}_2) \cdot \bar{m}_3 = T(s_3 s_\alpha c_\beta + c_3 s_\alpha s_\beta)$$

$$q_6 = \theta_3 \quad F_6 = 0 + -T \bar{m}_3 \cdot m_3 + T \bar{m}_3 \cdot \bar{m}_3 = T \bar{m}_3 \cdot \bar{m}_3 = 0$$

$$q_7 = \theta_4 \quad F_7 = 0 + 0 + T \bar{m}_3 \cdot \bar{m}_3 = T$$

$$F_4 = T(c_1 \bar{m}_1 + s_1 s_2 \bar{m}_1 - s_1 c_2 \bar{m}_3) \cdot \bar{m}_3 = T(c_1 [s_3 a_1 + c_3 a_2] + s_1 s_2 [c_3 a_1 - s_3 a_2] - s_1 c_2 a_3) \cdot \bar{m}_3$$

$$= T(c_1 s_3 s_\alpha c_\beta + c_1 c_3 s_\alpha s_\beta + s_1 s_2 c_3 s_\alpha c_\beta - s_1 s_2 s_3 s_\alpha s_\beta - s_1 c_2 c_\alpha)$$

$$a_1^* = \ddot{x} m_x + \ddot{y} m_y + \ddot{z} m_z$$

$$\bar{\omega}_x^B = \dot{\theta}_4 m_3 \Rightarrow \bar{\omega}_x^B = \bar{\omega}_y^B = \bar{\omega}_z^B = \bar{\omega}_{\dot{\theta}_1}^B = \bar{\omega}_{\dot{\theta}_2}^B = \bar{\omega}_{\dot{\theta}_3}^B = \bar{\omega}_{\dot{\theta}_4}^B = 0 \quad \bar{\omega}_{\dot{\theta}_4}^B = m_3 \quad \bar{\omega}_{\dot{\theta}_4}^B = \dot{\theta}_4 m_3$$

$$\bar{\omega}_{\dot{\theta}_1}^A = c_2 (c_3 a_1 - s_3 a_2) + s_2 a_3 \quad \bar{\omega}_{\dot{\theta}_2}^A = s_3 a_1 + c_3 a_2 \quad \bar{\omega}_{\dot{\theta}_3}^A = a_3$$

$$\bar{\Pi}^{C*} = I_1 a_1 a_1 + I_2 a_1 a_2 + I_3 a_1 a_3$$

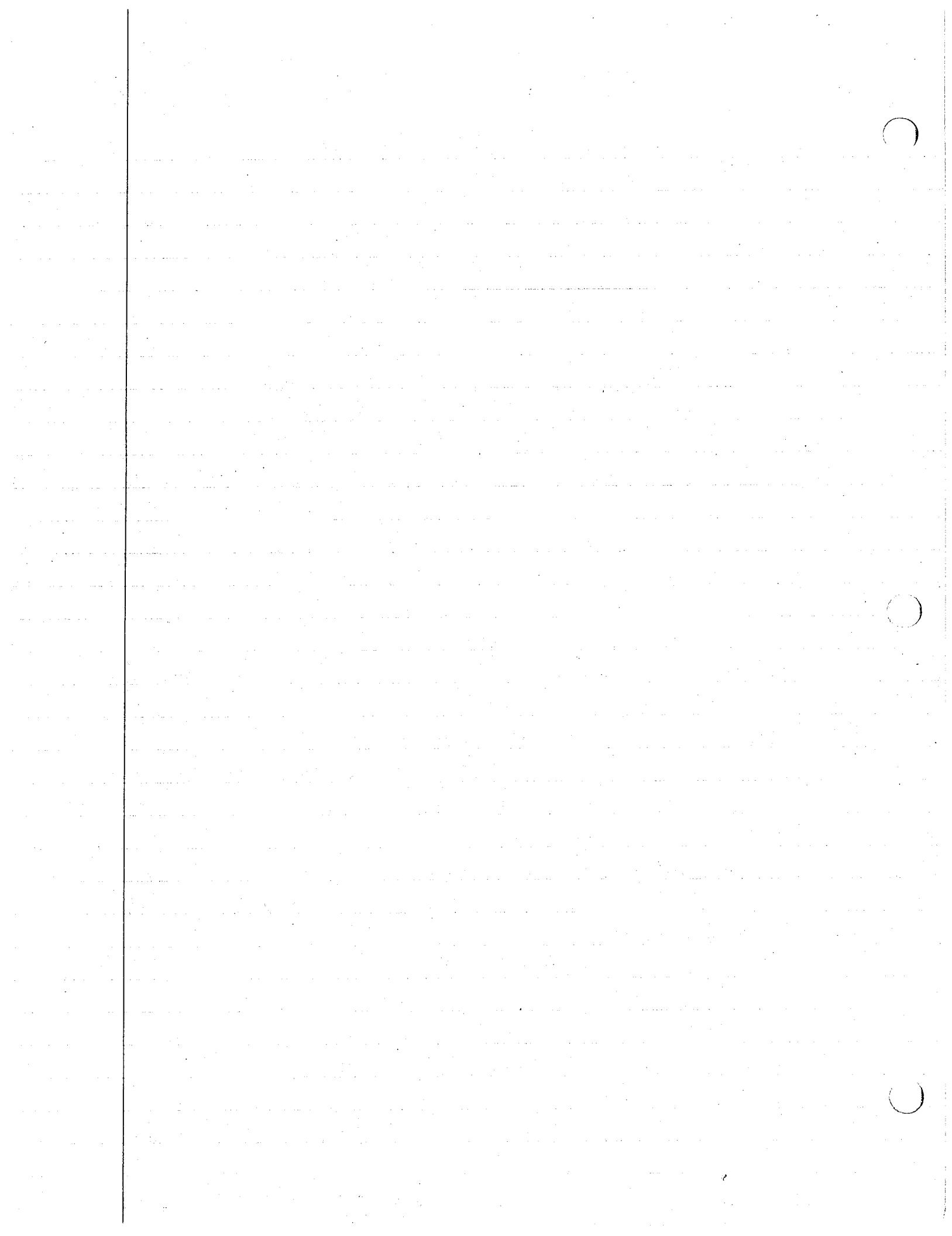
$$\bar{\omega}^A = \dot{\theta}_1 (c_2 (c_3 a_1 - s_3 a_2) + s_2 a_3) + \dot{\theta}_2 (s_3 a_1 + c_3 a_2) + \dot{\theta}_3 a_3$$

$$= (\dot{\theta}_1 c_2 c_3 + \dot{\theta}_2 s_3) a_1 + (-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) a_2 + (\dot{\theta}_1 s_2 + \dot{\theta}_3) a_3$$

$$= (\dot{\theta}_1 c_2 c_3 + \dot{\theta}_2 s_3) [c_\beta c_\alpha m_1 - s_\beta m_2 + s_\alpha c_\beta m_3] + (-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) [s_\beta c_\alpha m_1 + c_\beta m_2 + s_\alpha s_\beta m_3]$$

$$+ (\dot{\theta}_1 s_2 + \dot{\theta}_3) [-s_\alpha m_1 + c_\alpha m_3]$$

$$= [(\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) c_\beta c_\alpha + (-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) s_\beta c_\alpha - (\dot{\theta}_1 s_2 + \dot{\theta}_3) s_\alpha] m_1 + [(\dot{\theta}_1 c_2 c_3 + \dot{\theta}_2 s_3) s_\beta + (-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3)$$



$$F_r^* = -M\ddot{q}_r \cdot \frac{\overset{A}{\omega}}{g} q_r^* + \overset{A}{\omega}_{qr} \cdot \overset{B}{\pi}^B + \overset{B}{\omega}_{qr}^* \cdot \left( (\overset{A}{I})^{C*} \cdot \overset{R}{\omega}^A \right) \times \overset{R}{\omega}^A - \overset{A}{I}^{C*} \cdot \overset{R}{\alpha}^A - \overset{A}{I}^{B/B*} \cdot \overset{A}{\alpha}^B + \overset{A}{I}^{B/B*} \cdot \overset{R}{\alpha}^B \times \overset{R}{\omega}^A$$

$$F_1^* = -\frac{M\ddot{x}}{g} + 0 + 0$$

$$F_1 + F_1^* = -M\ddot{x} + Mg = 0 \quad \ddot{x} = g$$

$$F_2^* = -\frac{M\ddot{y}}{g} + 0 + 0$$

$$F_2 + F_2^* = -M\ddot{y} = 0 \quad \ddot{y} = 0$$

$$F_3^* = -\frac{M\ddot{z}}{g} + 0 + 0$$

$$F_3 + F_3^* = -M\ddot{z} = 0 \quad \ddot{z} = 0$$

$$F_4^* = 0 + 0 + 0$$

$$F_5^* = 0 + 0 + 0$$

$$F_6^* = 0 + 0 + 0$$

$$F_7^* = 0 + M_3 \cdot \overset{R}{\pi}^B + 0$$

$$F_7 + F_7^* = T + J_1(\dot{w}_3 + \ddot{\theta}_4) = 0$$

$$\overset{A}{I}^{B/B*} = \overset{A}{I}^{B/B*} \cdot \overset{R}{\omega}^A = J_1 \dot{\theta}_4 m_3$$

$$\text{Let } \overset{R}{\omega}^A = \tilde{\omega}_1 a_1 + \tilde{\omega}_2 a_2 + \tilde{\omega}_3 a_3 = w_1 m_1 + w_2 m_2 + w_3 m_3$$

$$\overset{A}{I}^{B/B*} \times \overset{R}{\omega}^A = J_1 \dot{\theta}_4 (w_1 m_2 - w_2 m_1) = J_1 \dot{\theta}_4 w_1 (-s_\beta a_1 + c_\beta a_2) - J_1 \ddot{\theta}_4 w_2 (c_\beta c_\alpha a_1 + s_\beta s_\alpha a_2 - s_\alpha a_3) =$$

$$\overset{R}{\alpha}^A = \tilde{\omega}_1 a_1 + \tilde{\omega}_2 a_2 + \tilde{\omega}_3 a_3 = (w_1 - w_2 \dot{\theta}_4) m_1 + (w_2 + \dot{\theta}_4 w_1) m_2 + \tilde{\omega}_3 m_3 \quad x_1 a_1 + x_2 a_2 + x_3 a_3$$

$$\overset{A}{I}^{C*} \cdot \overset{R}{\alpha}^A = I_1 \tilde{\omega}_1 a_1 + I_2 \tilde{\omega}_2 a_2 + I_3 \tilde{\omega}_3 a_3$$

$$(\overset{A}{I}^{C*} \cdot \overset{R}{\omega}^A) = I_1 \tilde{\omega}_1 a_1 + I_2 \tilde{\omega}_2 a_2 + I_3 \tilde{\omega}_3 a_3$$

$$\overset{R}{\pi}^B = (\overset{A}{I}^{C*} \cdot \overset{R}{\omega}^A) \times \overset{R}{\omega}^A - \overset{A}{I}^{C*} \cdot \overset{R}{\alpha}^A = [(I_2 - I_3) \tilde{\omega}_2 \tilde{\omega}_3 - I_1 \tilde{\omega}_1] a_1 + [(I_3 - I_1) \tilde{\omega}_3 \tilde{\omega}_1 - I_2 \tilde{\omega}_2] a_2 + [(I_1 - I_2) \tilde{\omega}_1 \tilde{\omega}_2 - I_3 \tilde{\omega}_3] a_3 = T_1^B a_1 + T_2^B a_2 + T_3^B a_3$$

$$\overset{A}{\alpha}^B = \frac{d}{dt} (\dot{\theta}_4 m_3) = \frac{d}{dt} \dot{\theta}_4 m_3 + \overset{A}{\omega}^B \times (\overset{A}{\omega}^B) = \ddot{\theta}_4 m_3 \quad \therefore \quad \overset{A}{I}^{B/B*} \cdot \overset{A}{\alpha}^B = J_1 \ddot{\theta}_4 m_3$$

$$\text{and } \overset{R}{I}^{B/B*} \cdot \overset{A}{\alpha}^B = J_1 \dot{\theta}_4 c_\alpha a_3 + J_1 \ddot{\theta}_4 s_\alpha s_\beta a_2 + J_1 \ddot{\theta}_4 s_\alpha c_\beta a_1$$

$$\overset{R}{\omega}^B = \overset{R}{\omega}^A + \overset{A}{\omega}^B = (\tilde{\omega}_1 + \dot{\theta}_4 s_\alpha c_\beta) a_1 + (\tilde{\omega}_2 + \dot{\theta}_4 s_\alpha s_\beta) a_2 + (\tilde{\omega}_3 + \dot{\theta}_4 c_\alpha) a_3 = x_1 a_1 + x_2 a_2 + x_3 a_3 = w_1 m_1 + w_2 m_2 + (w_3 + \dot{\theta}_4) m_3$$

$$\overset{R}{\alpha}^B = \dot{\alpha}_1 m_1 + \dot{\alpha}_2 m_2 + (\dot{\alpha}_3 + \ddot{\theta}_4) m_3 = \frac{d}{dt} \overset{R}{\omega}^B + \overset{R}{\omega}^B \times \overset{R}{\omega}^B$$

$$= (\tilde{\omega}_1 + \dot{\theta}_4 s_\alpha c_\beta) a_1 + (\tilde{\omega}_2 + \dot{\theta}_4 s_\alpha s_\beta) a_2 + (\tilde{\omega}_3 + \dot{\theta}_4 c_\alpha) a_3 + \overset{R}{\omega}^A \times \overset{R}{\alpha}^B$$

$$\overset{R}{\omega}^A \times \overset{R}{\alpha}^B = \overset{R}{\omega}^A \times \dot{\theta}_4 (s_\alpha c_\beta a_1 + s_\alpha s_\beta a_2 + c_\alpha a_3) = \dot{\theta}_4 [(\gamma_2 c_\alpha - \gamma_3 s_\alpha s_\beta) a_1 + (\gamma_3 s_\alpha c_\beta - \gamma_1 c_\alpha) a_2 + (\gamma_1 s_\alpha s_\beta - \gamma_2 s_\alpha c_\beta) a_3]$$

$$\overset{R}{\alpha}^B = [\tilde{\omega}_1 + \dot{\theta}_4 s_\alpha c_\beta + \dot{\theta}_4 (\tilde{\omega}_1 c_\alpha - \tilde{\omega}_3 s_\alpha c_\beta)] a_1 + [\tilde{\omega}_2 + \dot{\theta}_4 s_\alpha s_\beta + \dot{\theta}_4 (\tilde{\omega}_3 s_\alpha c_\beta - \tilde{\omega}_1 c_\alpha)] a_2 + [\tilde{\omega}_3 + \dot{\theta}_4 c_\alpha + \dot{\theta}_4 (\tilde{\omega}_1 s_\alpha s_\beta - \tilde{\omega}_2 s_\alpha c_\beta)] a_3$$

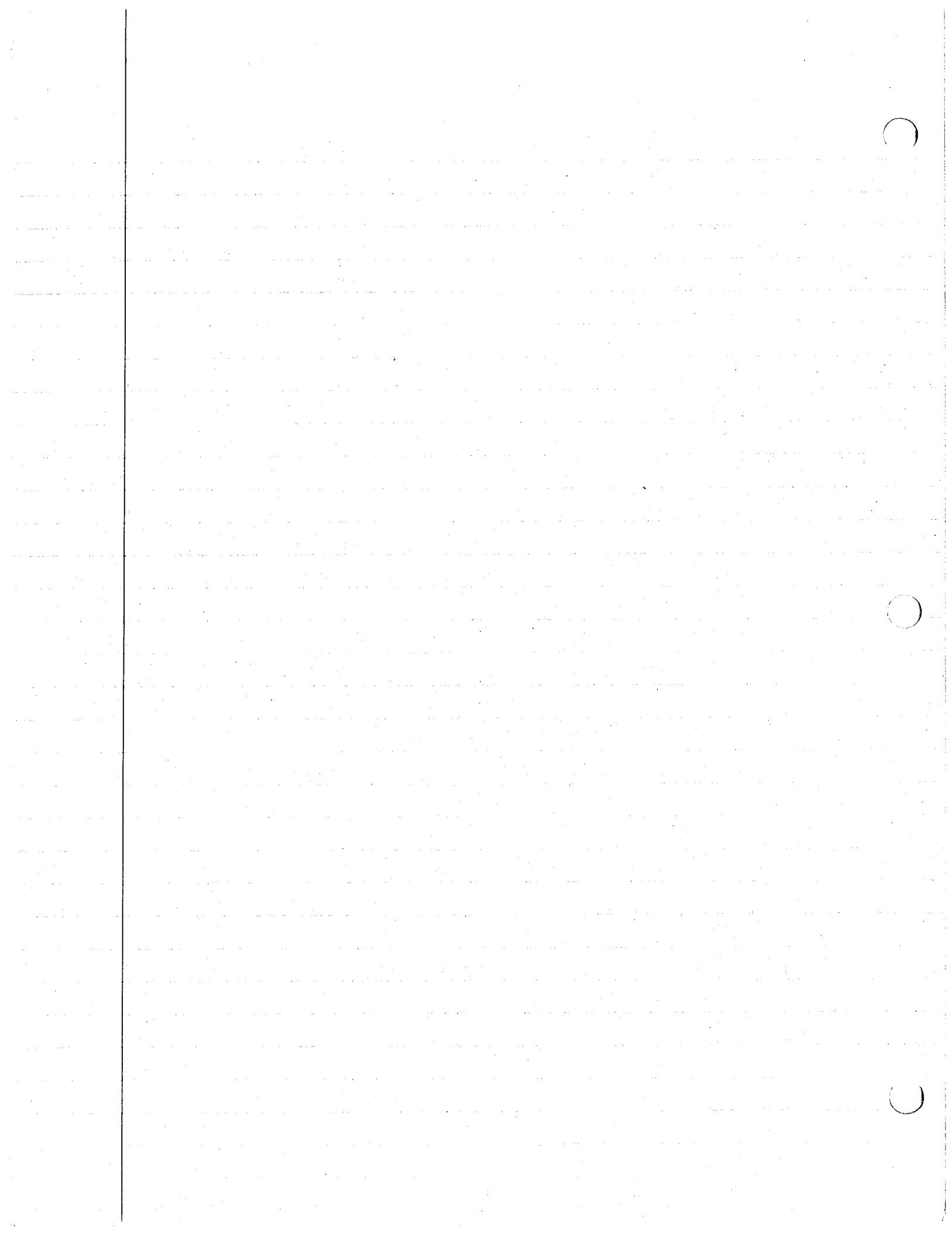
$$\overset{R}{\pi}^B = (\overset{A}{I}^{B/B*} \cdot \overset{R}{\omega}^B) \times \overset{R}{\omega}^B - \overset{A}{I}^{B/B*} \cdot \overset{R}{\alpha}^B$$

$$\overset{A}{I}^{B/B*} \cdot \overset{R}{\alpha}^B = J_1 (\dot{w}_3 + \ddot{\theta}_4) m_3 + J_2 (\dot{w}_2 m_2 + \dot{w}_1 m_1)$$

$$(\overset{A}{I}^{B/B*} \cdot \overset{R}{\omega}^B) \times \overset{R}{\omega}^B = [J_1 (\dot{w}_3 + \ddot{\theta}_4) m_3 + J_2 (\dot{w}_2 m_2 + \dot{w}_1 m_1)] \times [w_1 m_1 + w_2 m_2 + (\dot{w}_3 + \ddot{\theta}_4) m_3]$$

$$= J_1 (\dot{w}_3 + \ddot{\theta}_4) (w_1 m_2 - w_2 m_1) + J_2 w_2 (-w_1 m_3 + (w_3 + \dot{\theta}_4) m_1) + J_2 w_1 (w_2 m_3 - (w_3 + \dot{\theta}_4) m_2)$$

$$\overset{R}{\pi}^B = [(J_2 - J_1) w_3 w_2 + \dot{\theta}_4 w_2 (J_2 - J_1) + J_2 \dot{w}_1] m_1 + [(J_1 - J_2) w_3 w_1 + \dot{\theta}_4 w_1 (J_1 - J_2) - J_2 \dot{w}_2] m_2 +$$



$$[\mathcal{J}_2 \omega_1 \omega_2 - \mathcal{J}_3 \omega_3 \omega_1 \tilde{\omega}_2 (\omega_3 + \tilde{\theta}_4)] m_3$$

$$\omega_{\dot{\theta}_1} = \omega_{\dot{\theta}_2} = \omega_{\dot{\theta}_3} = 0 \quad \omega_{\dot{\theta}_4} = M_3$$

$$\omega_{\dot{\theta}_1}^A : \omega_{\dot{\theta}_1}^A = M_X, \omega_{\dot{\theta}_2}^A = P_2, \omega_{\dot{\theta}_3}^A = M_3, \omega_{\dot{\theta}_4}^A = 0$$

Only contrib to 2nd term is  $F_7$  term of  $\mathcal{J}_1(\omega_3 + \tilde{\theta}_4) = 0$

$$M_X = c_2 c_3 \alpha_1 - c_2 s_3 \alpha_2 + s_2 \alpha_3$$

$$P_2 = s_3 \alpha_1 + c_3 \alpha_2$$

$$\text{thus } M_X \cdot [ \dots ] = c_2 c_3 T_1^D - c_2 s_3 T_2^D + s_2 T_3^D - \mathcal{J}_1 \ddot{\theta}_4 (c_2 c_3 s_\alpha c_\beta - c_2 s_3 s_\alpha s_\beta + s_2 c_\alpha) \\ + c_2 c_3 \chi_1 - c_2 s_3 \chi_2 + s_2 \chi_3$$

$$\text{thus } P_2 \cdot [ \dots ] = s_3 T_1^D + c_3 T_2^D - \mathcal{J}_1 \ddot{\theta}_4 (s_3 s_\alpha c_\beta + c_3 s_\alpha s_\beta) + s_3 \chi_1 + c_3 \chi_2$$

$$\text{thus } \alpha_3 \cdot [ \dots ] = T_3^D - \mathcal{J}_1 \ddot{\theta}_4 c_\alpha + \chi_3$$

$$F_4 + F_4^* = 0 + c_2 c_3 (T_1^D + \chi_1) - c_2 s_3 (T_2^D + \chi_2) + s_2 (T_3^D + \chi_3) - \mathcal{J}_1 \ddot{\theta}_4 (c_2 s_\alpha (c_3 c_\beta - s_3 s_\beta) + s_2 c_\alpha) = 0$$

$$F_5 + F_5^* = 0 + s_3 (T_1^D + \chi_1) + c_3 (T_2^D + \chi_2) - \mathcal{J}_1 \ddot{\theta}_4 s_\alpha (s_3 c_\beta + c_3 s_\beta) = 0$$

$$F_6 + F_6^* = 0 + (T_3^D + \chi_3) - \mathcal{J}_1 \ddot{\theta}_4 c_\alpha = 0$$

$$F_7 + F_7^* = T^D - \mathcal{J}_1 (\omega_3 + \tilde{\theta}_4) = 0$$

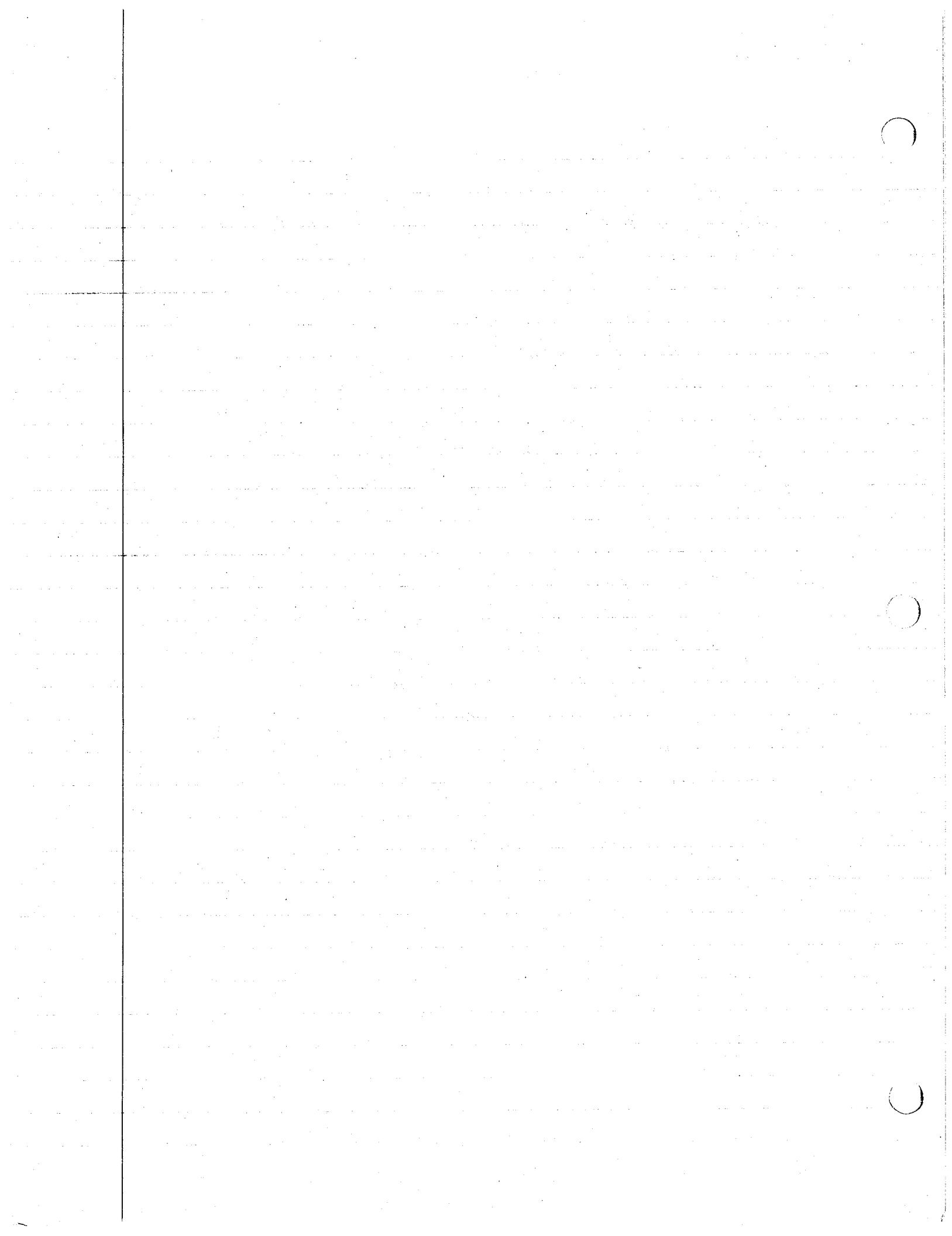
$$T_1^D = (\mathcal{I}_2 - \mathcal{I}_3) \tilde{\omega}_2 \tilde{\omega}_3 - \mathcal{I}_1 \tilde{\omega}_1$$

$$T_2^D = (\mathcal{I}_3 - \mathcal{I}_1) \tilde{\omega}_3 \tilde{\omega}_1 - \mathcal{I}_2 \tilde{\omega}_2$$

$$T_3^D = (\mathcal{I}_1 - \mathcal{I}_2) \tilde{\omega}_1 \tilde{\omega}_2 - \mathcal{I}_3 \tilde{\omega}_3$$

$$\chi_1 = - \mathcal{J}_1 \ddot{\theta}_4 (\omega_1 s_\beta + \omega_2 c_\beta c_\alpha) \quad \chi_2 = \mathcal{J}_1 \ddot{\theta}_4 (\omega_1 c_\beta - \omega_2 s_\beta c_\alpha)$$

$$\chi_3 = \mathcal{J}_1 \ddot{\theta}_4 \omega_3 s_\alpha$$



$$\gamma_1 S_\alpha S_\beta \alpha_3 + \gamma_1 C_\alpha \alpha_2 - \gamma_2 S_\alpha C_\beta \alpha_3 + \gamma_2 C_\alpha \alpha_1 + \gamma_3 S_\alpha C_\beta \alpha_2 - \gamma_3 S_\alpha S_\beta \alpha_1$$

$$(\tilde{\omega}_2 + \theta_4 S_\alpha S_\beta) c_\alpha = (\tilde{\omega}_3 + \theta_4 C_\alpha) S_\alpha S_\beta$$

$$(\tilde{\omega}_2 c_\alpha - \tilde{\omega}_3 S_\alpha C_\beta) \alpha_1$$

$$(\tilde{\omega}_3 + \theta_4 C_\alpha) S_\alpha C_\beta - (\tilde{\omega}_2 + \theta_4 S_\alpha C_\beta) c_\alpha$$

$$\tilde{\omega}_3 S_\alpha C_\beta - \tilde{\omega}_2 C_\alpha$$

$$(\tilde{\omega}_2 + \theta_4 S_\alpha C_\beta) S_\alpha S_\beta = (\tilde{\omega}_2$$

$$(\tilde{\omega}_2 + \theta_4 S_\alpha C_\beta) S_\alpha S_\beta = (\tilde{\omega}_2 + \theta_4 S_\alpha S_\beta) S_\alpha C_\beta$$

$$= J_1 (\omega_3 + \theta_4) \omega_2 + J_2 \omega_2 (\omega_3 + \theta_4) + J_2 \dot{\omega}_1$$

$$J_1 (\omega_3 + \theta_4) \omega_1 = J_2 \omega_1 (\omega_3 + \theta_4) - J_2 \dot{\omega}_2$$

$$= J_2 \omega_2 \dot{\omega}_1 + J_3 \omega_1 \omega_2$$

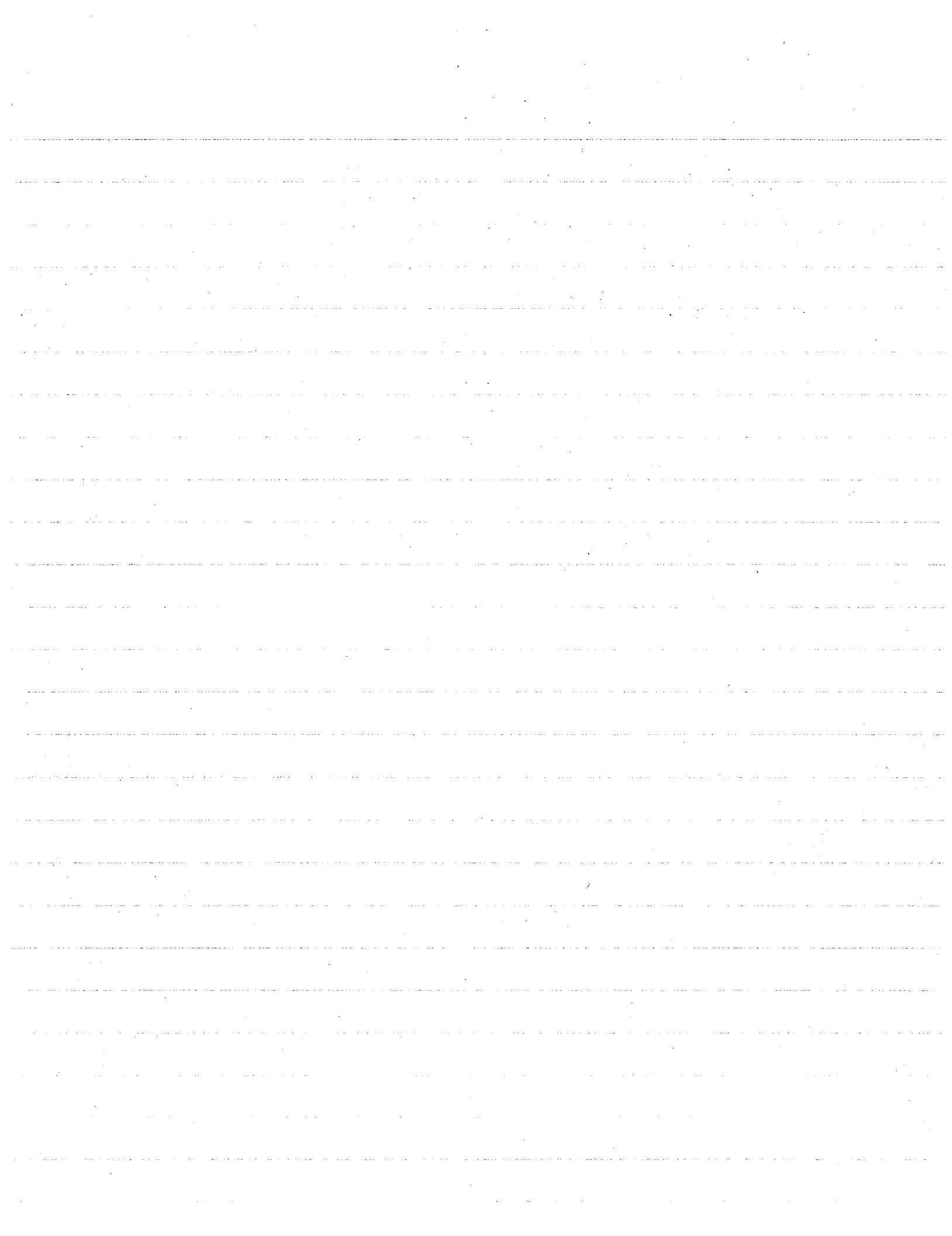
$$c_2 c_3 (c_\alpha C_\beta m_1 - S_\beta m_2 + S_\alpha C_\beta m_3) - c_2 s_3 (S_\beta C_\alpha m_1 + C_\beta m_2 + S_\alpha S_\beta m_3) + s_2 (-S_\alpha m_1 + C_\alpha m_3)$$

$$m_x = (c_2 c_3 S_\alpha C_\beta - c_2 s_3 S_\alpha S_\beta + s_2 C_\alpha) m_3 + (c_2 c_3 c_\alpha C_\beta - c_2 s_3 S_\beta C_\alpha - s_2 S_\alpha) m_1 + (-c_2 c_3 S_\beta - c_2 s_3 C_\beta) m_2$$

$$\ddot{m}_2 = s_3 (C_\beta c_\alpha m_1 - S_\beta m_2 + S_\alpha C_\beta m_3) + c_3 (S_\beta C_\alpha m_1 + C_\beta m_2 + S_\alpha S_\beta m_3) = (s_3 C_\beta C_\alpha + c_3 S_\beta C_\alpha) m_1 - (s_3 S_\alpha C_\beta + c_3 S_\alpha S_\beta) m_3$$

$$- J_1 \theta_4 (\omega_1 S_\beta + \omega_2 C_\beta C_\alpha) \alpha_1 + J_1 \theta'_4 (\omega_1 C_\beta - \omega_2 S_\beta C_\alpha) \alpha_2 + J_1 \theta'_4 \omega_2 S_\alpha \alpha_3 = M^{\text{eff}}_{\text{ext}} \omega^2$$

$$- J_1 \theta'_4 (\omega_1 S_\beta + \omega_2 C_\beta C_\alpha) c_2 c_3 + J_1 \theta'_4 (\omega_1 C_\beta - \omega_2 S_\beta C_\alpha) c_2 s_3 + J_1 \theta'_4 \omega_2 s_2 S_\alpha$$



$$\text{now } T_3^D + \kappa_3 = J_1 \ddot{\theta}_4 c_\alpha \quad \text{put into } F_4 + F_4^* = 0$$

$$c_2 c_3 (T_1^D + \kappa_1) - c_2 s_3 (T_2^D + \kappa_2) + s_2 c_\alpha J_1 \ddot{\theta}_4 - \cancel{J_1 \ddot{\theta}_4 s_2 c_\alpha} - J_1 \ddot{\theta}_4 (c_2 s_\alpha (c_3 c_\beta - s_3 s_\beta)) = 0$$

$\Rightarrow$  if  $c_2 \neq 0$

$$c_3^2 (T_1^D + \kappa_1) - s_3 c_3 (T_2^D + \kappa_2) - J_1 \ddot{\theta}_4 s_\alpha (c_3 c_\beta - s_3 s_\beta) \cancel{s_3} = 0$$

$$s_3^2 (T_1^D + \kappa_1) + c_3 s_3 (T_2^D + \kappa_2) + J_1 \ddot{\theta}_4 s_\alpha (s_3 c_\beta + c_3 s_\beta) \cancel{s_3} = 0$$

$$-(T_2^D + \kappa_2) + J_1 \ddot{\theta}_4 s_\alpha [s_3 c_3 c_\beta + c_3^2 s_\beta - s_3 c_3 c_\beta + s_3^2 s_\beta] = 0$$

$$\boxed{(T_2^D + \kappa_2) - J_1 \ddot{\theta}_4 s_\alpha s_\beta = 0}$$

$$(T_1^D + \kappa_1) - J_1 \ddot{\theta}_4 s_\alpha (c_3^2 c_\beta - s_3 c_3 s_\beta + s_3^2 c_\beta + s_3 c_3 s_\beta)$$

$$\boxed{(T_1^D + \kappa_1) - J_1 \ddot{\theta}_4 s_\alpha c_\beta = 0}$$

$$\boxed{(T_3^D + \kappa_3) - J_1 \ddot{\theta}_4 c_\alpha = 0}$$

$$T = J_1 (\dot{\omega}_3 + \ddot{\theta}_4) = 0$$

$$(T_3^D + \kappa_3) - J_1 \ddot{\theta}_4 c_\alpha = [(I_1 - I_2) \tilde{\omega}_1 \tilde{\omega}_2 - I_3 \tilde{\omega}_3] + J_1 \dot{\theta}_4 \omega_2 s_\alpha - J_1 \ddot{\theta}_4 c_\alpha = 0$$

$$[ - J_1 (\dot{\omega}^B \beta_3 - \dot{\omega}^B (\omega_2 s_\alpha)) = 0$$

$$\text{is } \omega_2 s_\alpha = \beta_1 \tilde{\omega}_2 - \beta_2 \tilde{\omega}_1 \quad \beta_1 = s_\alpha c_\beta \quad \beta_2 = s_\alpha s_\beta$$

$$\tilde{\omega}_2 = -\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3 \quad \tilde{\omega}_1 = \dot{\theta}_1 c_2 c_3 + \dot{\theta}_2 s_3 \quad \omega_2 = -(\dot{\theta}_1 c_2 c_3 + \dot{\theta}_2 s_3) s_\beta + (-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) c_\beta$$

$$\tilde{\omega}_2 \beta_1 - \tilde{\omega}_1 \beta_2 = s_\alpha [(-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) c_\beta - (\dot{\theta}_1 c_2 c_3 + \dot{\theta}_2 s_3) s_\beta] = \omega_2 s_\alpha$$

$$\rightarrow [(I_1 - I_2) \tilde{\omega}_1 \tilde{\omega}_2 - I_3 \tilde{\omega}_3] - J_1 (\dot{\omega}^B \beta_3 - \dot{\omega}^B (\tilde{\omega}_2 \beta_1 - \tilde{\omega}_1 \beta_2)) = 0$$

Now

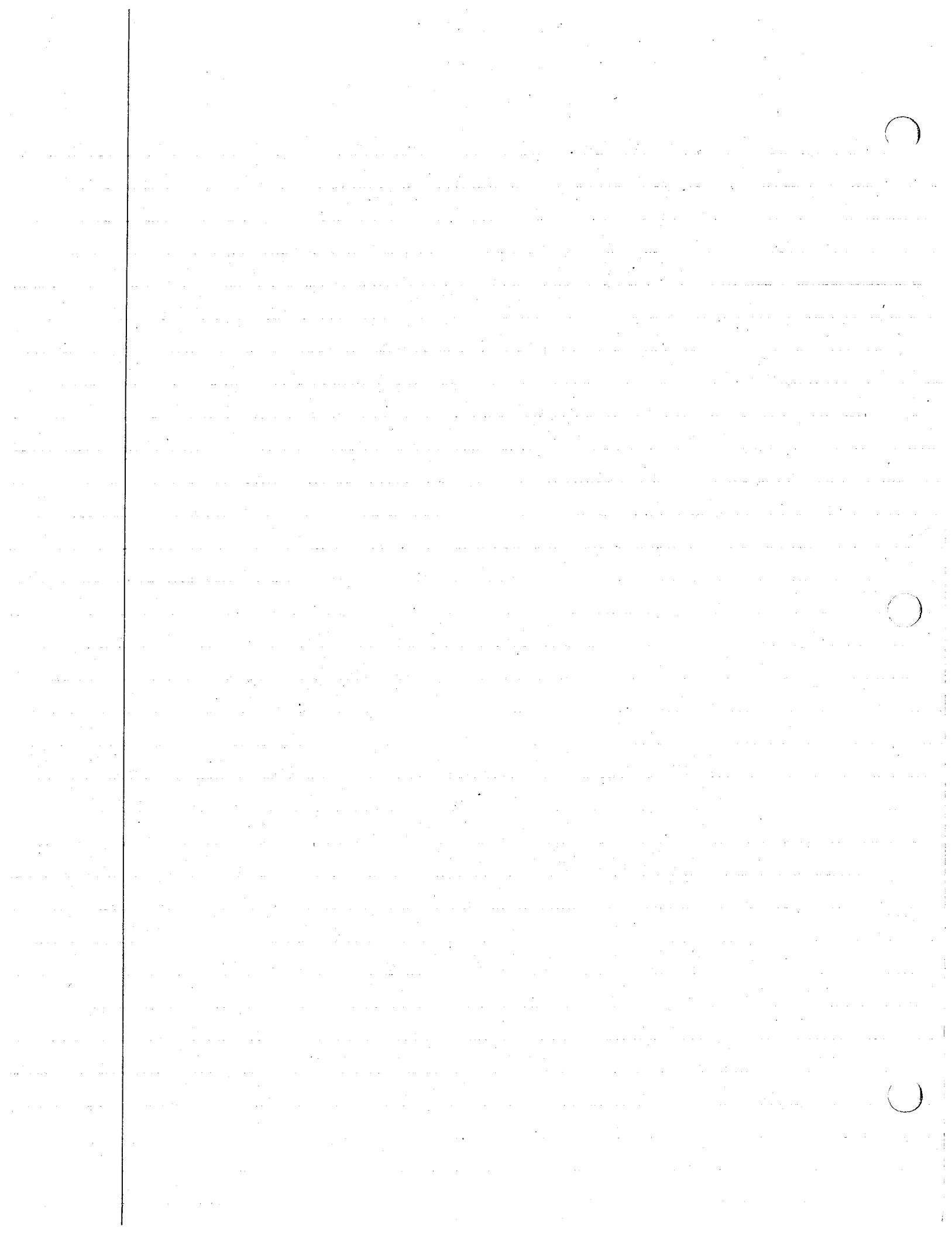
$$T_1^D + \kappa_1 - J_1 \ddot{\theta}_4 s_\alpha c_\beta = [(I_2 - I_3) \tilde{\omega}_2 \tilde{\omega}_3 - I_1 \tilde{\omega}_1] - J_1 \dot{\theta}_4 (\omega_1 s_\beta + \omega_2 c_\beta c_\alpha) - J_1 \ddot{\theta}_4 s_\alpha c_\beta = 0$$

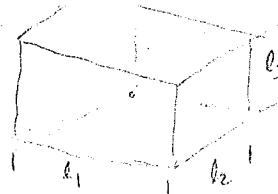
$$\rightarrow [(I_2 - I_3) \tilde{\omega}_2 \tilde{\omega}_3 - I_1 \tilde{\omega}_1] - J_1 (\dot{\omega}^B \beta_1 + \dot{\omega}^B [\beta_2 \tilde{\omega}_3 - \beta_3 \tilde{\omega}_2]) = 0$$

$$\text{is } -\omega_1 s_\beta - \omega_2 c_\beta c_\alpha = \beta_2 \tilde{\omega}_3 - \beta_3 \tilde{\omega}_2 = s_\alpha s_\beta [\dot{\theta}_1 s_2 + \dot{\theta}_3] - c_\alpha [-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3]$$

$$- s_\beta [(\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) c_\beta c_\alpha + (-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) s_\beta c_\alpha - (\dot{\theta}_1 s_2 + \dot{\theta}_3) s_\alpha] - c_\beta c_\alpha [-(\dot{\theta}_1 c_2 c_3 + \dot{\theta}_2 s_3) s_\beta + (-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3)]$$

$$- c_\alpha (-\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3) + (\dot{\theta}_1 s_2 + \dot{\theta}_3) s_\alpha s_\beta$$





$$\beta = (s_{\alpha}c_{\beta}, s_{\alpha}s_{\beta}, c_{\alpha}) \quad J_1 = \frac{mR^2}{2} \quad I_1 = \frac{M}{12}(l_1^2 + l_3^2) \quad I_2 = \frac{M}{12}(l_3^2 + l_2^2)$$

$$I_3 = \frac{(l_1^2 + l_2^2)M}{12} \quad {}^A\dot{\omega}^B = \dot{\theta}_4 \quad {}^A\ddot{\omega}^B = \ddot{\theta}_4 ; \quad \text{let } l_2 = bl_1, l_3 = cl_1, 0 < b, c < 1$$

$$\tilde{\omega}_1 = \dot{\theta}_1 c_2 c_3 + \dot{\theta}_2 s_3 \quad \tilde{\omega}_2 = -\dot{\theta}_1 c_2 s_3 + \dot{\theta}_2 c_3 \quad \tilde{\omega}_3 = \dot{\theta}_1 s_2 + \dot{\theta}_3$$

$$\tilde{\omega}_1 s_3 + \tilde{\omega}_2 c_3 = \dot{\theta}_2 \quad \text{and } (\tilde{\omega}_1 c_3 - \tilde{\omega}_2 s_3)/c_2 = \dot{\theta}_1 \quad \dot{\theta}_3 = \tilde{\omega}_3 - (\tilde{\omega}_1 c_3 - \tilde{\omega}_2 s_3) s_2/c_2$$

$$I_1 = \frac{Ml_1^2}{12}(1+c^2) \quad I_2 = \frac{Ml_1^2(b^2+c^2)}{12} \quad I_3 = \frac{Ml_1^2(1+b^2)}{12} \quad \text{let } R = \frac{dl}{2}, d \leq \min(l, c)$$

$$J_1 = \frac{ml_1^2 d^2}{4 \cdot 2} \quad \text{let } C_1 = 1 - J_1 \left( \frac{\beta_1^2}{I_1} + \frac{\beta_2^2}{I_2} + \frac{\beta_3^2}{I_3} \right) \quad 1 = \frac{1}{C_1} - \frac{J_1}{C_1}$$

$${}^A\dot{\omega}^B = \frac{I}{J_1} \cdot \frac{1}{C_1} - \frac{1}{C_1} \left[ \frac{\beta_1 F_1}{I_1} + \frac{\beta_2 F_2}{I_2} + \frac{\beta_3 F_3}{I_3} \right]$$

$$\dot{\tilde{\omega}}_3 = \frac{1}{\cancel{I_3}} \left[ F_3 - \frac{\beta_3}{C_1} \left\{ \frac{I}{I_3} - \frac{J_1}{I_3} \left[ \frac{\beta_1 F_1}{I_1} + \frac{\beta_2 F_2}{I_2} + \frac{\beta_3 F_3}{I_3} \right] \right\} \right] \quad \text{OK.}$$

$$\dot{\tilde{\omega}}_2 = \frac{1}{\cancel{I_2}} \left[ F_2 - \beta_2 \left\{ \frac{I}{C_1} - \frac{J_1}{C_1} \left[ \frac{\beta_1 F_1}{I_1} + \frac{\beta_2 F_2}{I_2} + \frac{\beta_3 F_3}{I_3} \right] \right\} \right] \quad \text{OK.}$$

$$\dot{\tilde{\omega}}_1 = \frac{1}{\cancel{I_1}} \left[ F_1 - \beta_1 \left\{ \frac{I}{C_1} - \frac{J_1}{C_1} \left[ \frac{\beta_1 F_1}{I_1} + \frac{\beta_2 F_2}{I_2} + \frac{\beta_3 F_3}{I_3} \right] \right\} \right] \quad \text{OK.}$$

$$F_1 = (I_2 - I_3) \tilde{\omega}_2 \tilde{\omega}_3 + J_1 {}^A\dot{\omega}^B [\tilde{\omega}_3 \beta_2 - \tilde{\omega}_2 \beta_3]$$

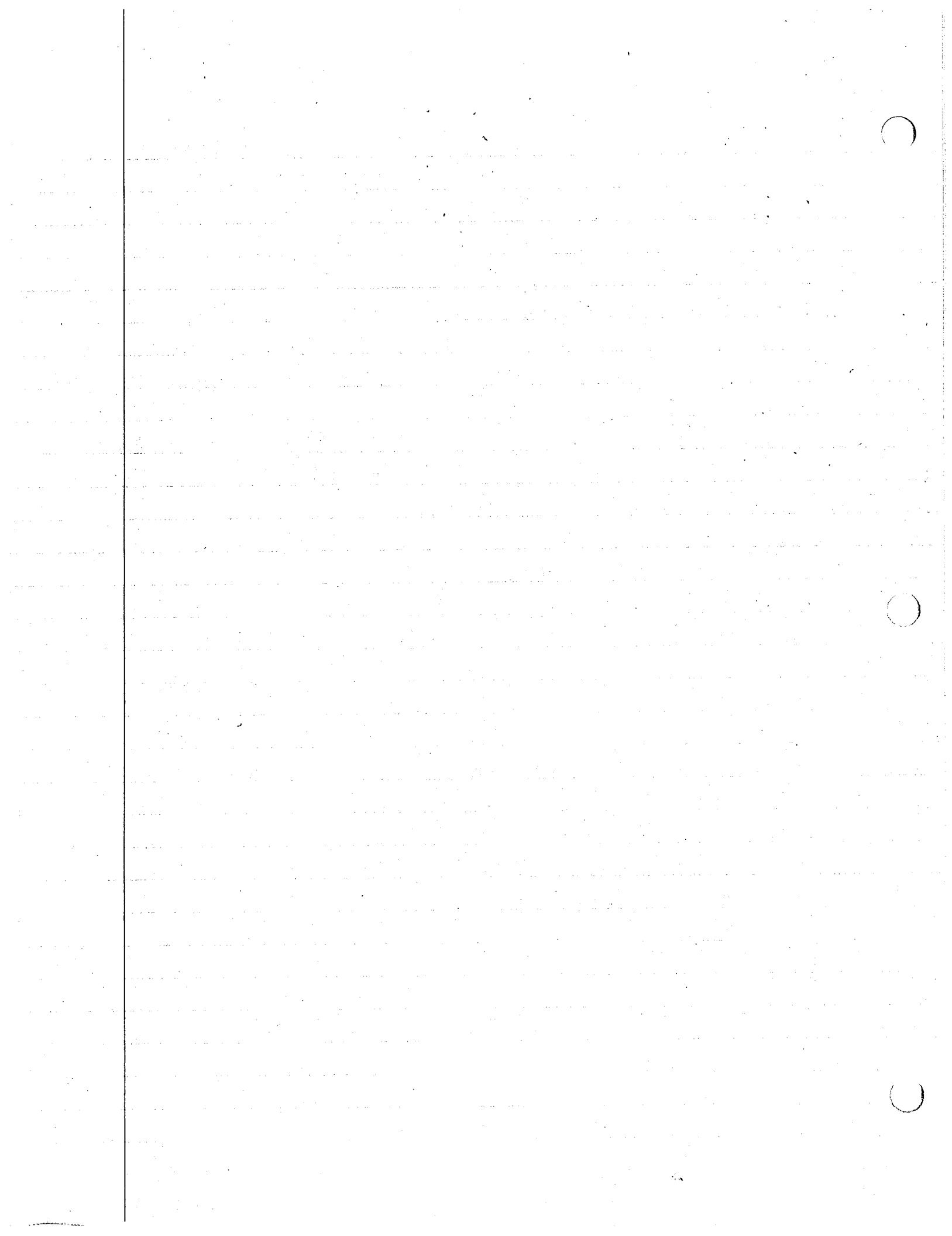
$$F_2 = (I_3 - I_1) \tilde{\omega}_3 \tilde{\omega}_1 + J_1 {}^A\dot{\omega}^B [\tilde{\omega}_1 \beta_3 - \tilde{\omega}_3 \beta_1]$$

$$F_3 = (I_1 - I_2) \tilde{\omega}_1 \tilde{\omega}_2 + J_1 {}^A\dot{\omega}^B [\tilde{\omega}_2 \beta_1 - \tilde{\omega}_1 \beta_2]$$

$$\dot{\theta}_1 = (\tilde{\omega}_1 c_3 - \tilde{\omega}_2 s_3)/c_2$$

$$\dot{\theta}_2 = \tilde{\omega}_1 s_3 + \tilde{\omega}_2 c_3$$

$$\dot{\theta}_3 = \tilde{\omega}_3 - s_2 (\tilde{\omega}_1 c_3 - \tilde{\omega}_2 s_3)/c_2$$



$$\text{Let } y_i = \begin{cases} \tilde{\omega}_i & i=1,2,3 \\ \theta_{i-3} & i=4,5,6,7 \\ \omega_B - \theta_4 & i=8 \end{cases}, \quad \begin{matrix} \tilde{\omega}_1, \tilde{\omega}_2, \tilde{\omega}_3 \\ \theta_1, \theta_2, \theta_3, \theta_4 \end{matrix}$$

define first

ALP, BET

Read L1, B, C, D, MB, MR)

A = AMIN(B, C)

IF (D.GT.A) D = A

R = A \* L1 / 2

J1 = MR + R \* R / Z

A = MB \* L1 + L2 / 12

I1 = A + (1 + C \* C)

I2 = A + (B \* B + C \* C)

I3 = A + (1 + B \* B)

PI04 = ATAN(1.00)

ALB = ALP + PI04 / 45

BEB = BET + PI04 / 45

B1 = SIN(ALB) \* COS(BEB)

B2 = SIN(ALB) \* SIN(BEB)

Z5 = COS(Y(6))

B3 = COS(ALB)

Z6 = SIN(Y(6))

Z7 = COS(Y(5))

Z8 = SIN(Y(5))

Z9 = (Y(1) \* Z5 - Y(2) \* Z6) / Z7

$$DY(1) = [Z1 - B1 * (TC - JC + Z4)] / I_1$$

$$DY(2) = [Z2 - B2 * (TC - JC + Z4)] / I_2$$

$$DY(3) = [Z3 - B3 * (TC - JC + Z4)] / I_3$$

$$DY(4) = Z9$$

$$DY(5) = Y(1) * Z6 + Y(2) * Z5$$

$$DY(6) = Y(3) - Z8 * Z9$$

$$DY(7) = Y(8)$$

$$\tilde{\omega}_B = DY(8) = TC / J1 - Z4 / C1$$



$$\beta_3 \omega_1 - \beta_1 \omega_3) \alpha_{12} + [\omega_1 \omega_2 (I_1 - I_2) - \omega_3 I_3 - J \dot{\omega}^B \beta_1 + J \dot{\omega}^B (\beta_1 \omega_2 - \beta_2 \omega_1)] \alpha_{13}$$

$$F_4^* = [\omega_2 \omega_3 (I_2 - I_3) - I_1 \dot{\omega}_1 - J \dot{\omega}^B \beta_1 + J \dot{\omega}^B (\beta_2 \omega_3 - \beta_3 \omega_2)] = F_{\omega_1}^*$$

$$F_5^* = [\omega_3 \omega_1 (I_3 - I_1) - I_2 \dot{\omega}_2 - J \dot{\omega}^B \beta_2 + J \dot{\omega}^B (\beta_3 \omega_1 - \beta_1 \omega_3)] = F_{\omega_2}^*$$

$$F_6^* = [\omega_1 \omega_2 (I_1 - I_2) - I_3 \dot{\omega}_3 - J \dot{\omega}^B \beta_3 + J \dot{\omega}^B (\beta_1 \omega_2 - \beta_2 \omega_1)] = F_{\omega_3}^*$$

$$T^{*B} = (I^{B/B^*}, \frac{R}{\omega} B) \times \frac{R}{\omega} B = I^{B/B^*} \cdot \frac{R}{\omega} B$$

$$\frac{d}{dt} \frac{R}{\omega} B = \frac{A}{dt} \frac{R}{\omega} A + \omega \times \frac{R}{\omega} A + \frac{A}{dt} \frac{R}{\omega} B + \frac{R}{\omega} A \times \frac{A}{\omega} B$$

$$R \omega^A \times \frac{A}{\omega} B = (\omega_i \alpha_i) \times (\frac{A}{\omega} \beta_j \alpha_j) = \dot{\omega}^B [(\omega_2 \beta_3 - \omega_3 \beta_2) \alpha_1 + (\omega_3 \beta_1 - \omega_1 \beta_3) \alpha_2 + (\omega_1 \beta_2 - \omega_2 \beta_1) \alpha_3]$$

$$\begin{aligned} \text{Now } \beta \cdot T^{*B} &= \beta \cdot (I^{B/B^*}, \frac{R}{\omega} B) \times \frac{R}{\omega} B = \beta \cdot I^{B/B^*} \cdot \frac{R}{\omega} B \\ &= (I^{B/B^*}, \frac{R}{\omega} B) \cdot (\frac{R}{\omega} B \times \beta) = \beta \cdot I^{B/B^*} \cdot \frac{R}{\omega} B \\ &\quad (\frac{R}{\omega} \dot{\omega}^A + \frac{A}{\omega} B) \times \beta = \beta \cdot I^{B/B^*} \cdot \frac{R}{\omega} A - \omega \beta \cdot I^{B/B^*} \cdot \beta - \beta \cdot I^{B/B^*} \cdot (\frac{R}{\omega} A \times \frac{A}{\omega} B) \\ &= (I^{B/B^*}, \frac{R}{\omega} B) \cdot (\frac{R}{\omega} A \times \beta) - J \beta \cdot \frac{R}{\omega} A - \omega \beta \cdot I^{B/B^*} - J \beta \cdot (\frac{R}{\omega} A \times \frac{A}{\omega} B) \quad \frac{A}{\omega} B = \dot{\omega}^B \beta \\ &\quad [I \cdot (\frac{R}{\omega} \dot{\omega}^A + \frac{A}{\omega} B \beta)] - J [\dot{\omega}_1 \beta_1 + \dot{\omega}_2 \beta_2 + \dot{\omega}_3 \beta_3 + \dot{\omega}^B] \\ &= (I^{B/B^*}, \frac{R}{\omega} A) + J \omega^B \beta \cdot (\frac{R}{\omega} A \times \beta) - J [\dot{\omega}_1 \beta_1 + \dot{\omega}_2 \beta_2 + \dot{\omega}_3 \beta_3 + \dot{\omega}^B] \\ \beta \cdot T^{*B} &= \uparrow \cdot (\frac{R}{\omega} A \times \beta) - J [\dot{\omega}_1 \beta_1 + \dot{\omega}_2 \beta_2 + \dot{\omega}_3 \beta_3 + \dot{\omega}^B] - J [\dot{\omega}_1 \beta_1 + \dot{\omega}_2 \beta_2 + \dot{\omega}_3 \beta_3 + \dot{\omega}^B] + T = 0 \end{aligned}$$

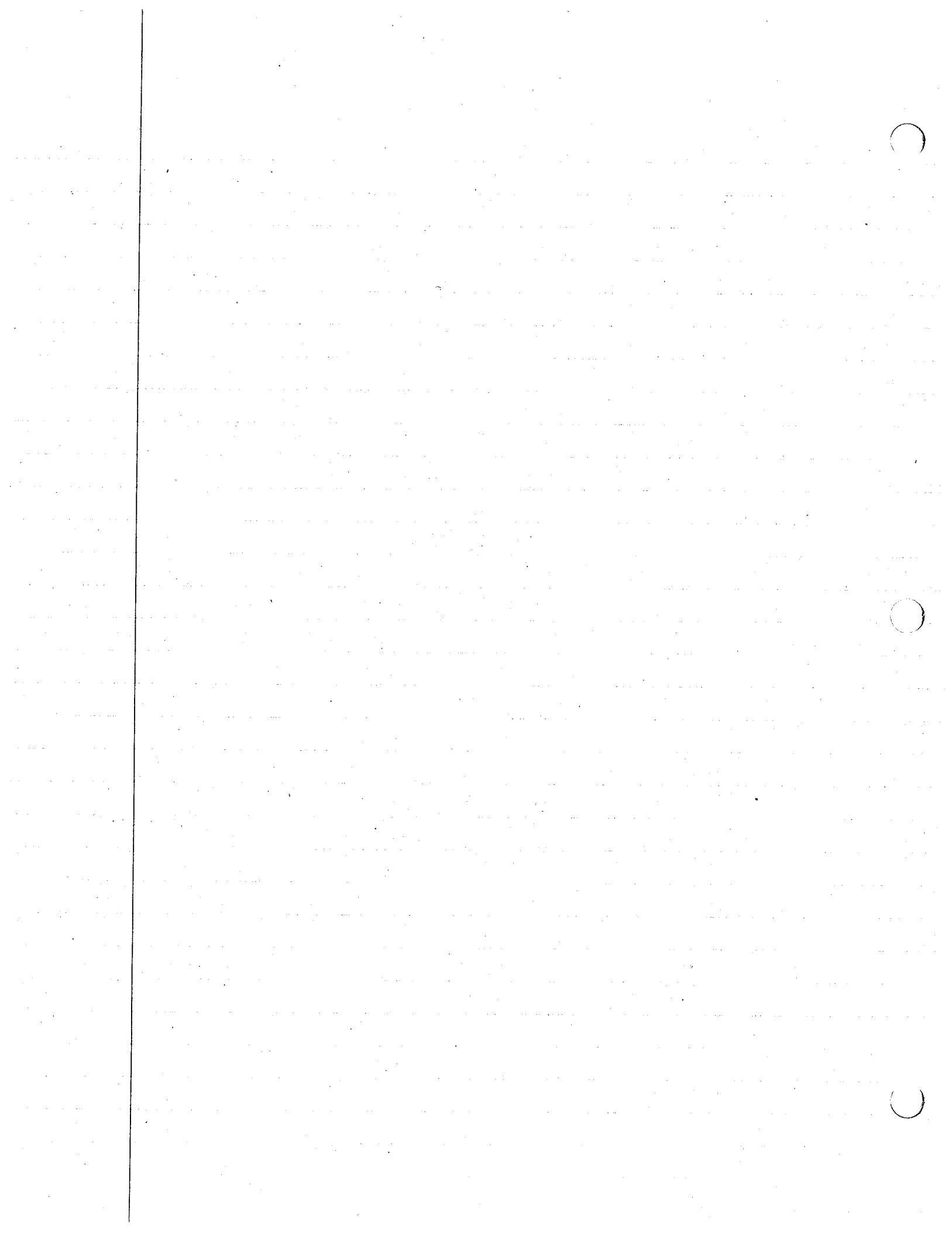
$$\frac{T}{J} - [\dot{\omega}_1 \beta_1 + \dot{\omega}_2 \beta_2 + \dot{\omega}_3 \beta_3] = \dot{\omega}^B$$

$$\beta_1 \dot{\omega}_1 = \frac{R}{I_1} [\omega_2 \omega_3 (I_2 - I_3) + J \dot{\omega}^B (\beta_2 \omega_3 - \beta_3 \omega_2)] - \frac{J \beta_1^2}{I_1} \dot{\omega}^B$$

$$= \frac{\beta_1 F_1}{I_1} - \frac{J \beta_1^2 \dot{\omega}^B}{I_1} = \frac{\beta_1 F_1}{I_1} - \frac{J \beta_1^2}{I_1} + \frac{J \beta_1^2}{I_1}$$

$$\frac{T}{J} - \left[ \frac{\beta_1 F_1}{I_1} + \frac{\beta_2 F_2}{I_2} + \frac{\beta_3 F_3}{I_3} + J \dot{\omega}^B \left( \frac{\beta_1^2}{I_1} + \frac{\beta_2^2}{I_2} + \frac{\beta_3^2}{I_3} \right) \right] = \dot{\omega}^B$$

$$\frac{T}{J} - \left[ \frac{\beta_1 F_1}{I_1} \right] = \dot{\omega}^B \left[ 1 - J \left( \frac{\beta_1^2}{I_1} \right) \right]$$



Derivation

let  $x, y, z, \omega_i, \dot{\omega}_i^A, \dot{\omega}_i^B$

$$F_r = W_{q_r}^A \cdot IF^A + \omega_{q_r}^A \cdot \overline{I}^A + W_{q_r}^B \cdot IF^B + \omega_{q_r}^B \cdot \overline{I}^B = W_{q_r}^A [IF^A + IF^B] + \omega_{q_r}^A [\overline{I}^A + \overline{I}^B] + \omega_{q_r}^B \cdot \overline{I}^B$$

$$F_r^* = W_{q_r}^A \cdot IF^* + \omega_{q_r}^A \cdot \overline{I}^{*A} + \omega_{q_r}^B \cdot \overline{I}^{*B}$$

$$\overline{I}^A = -T \beta = T \beta_i \alpha_i; \quad \overline{I}^B = +T \beta_i \alpha_i; \quad \overline{\omega}^A = \omega_i \alpha_i \quad \overline{\omega}^B = \omega_i^A + \omega_i^B = (\omega_i + \hat{\omega}_i^B \beta_i) \alpha_i$$

$$W^A = \dot{x} m_x + \dot{y} m_y + \dot{z} m_z \quad IF^A + IF^B = Mg m_x \quad \overline{\omega}_{\omega_i}^B = \alpha_i \quad \overline{\omega}_{\omega_i}^B = \beta; \quad \overline{\omega}_{\omega_i}^B = 0$$

$$\alpha_i^A = \dot{x} m_x + \dot{y} m_y + \dot{z} m_z \quad \overline{I}^A + \overline{I}^B = 0; \quad \overline{\omega}_{\omega_i}^B = \beta$$

$$F_1 = Mg + 0 + 0 \quad F_1^* = -M \ddot{x}$$

$$F_2 = 0 + 0 + 0 \quad F_2^* = -M \ddot{y}$$

$$F_3 = 0 + 0 + 0 \quad F_3^* = -M \ddot{z}$$

$$F_4 = 0 + \alpha_1 \cdot 0 + 0 \quad F_4^* = W_{\omega_1}^{A*} \cdot IF^* + \omega_{\omega_1}^A \cdot \overline{I}^{*A} + \omega_{\omega_1}^B \cdot \overline{I}^{*B}$$

$$F_5 = 0 + \alpha_2 \cdot 0 + 0 \quad 0 + \alpha_1 \cdot \overline{I}^{*A} + \alpha_1 \cdot \overline{I}^{*B}$$

$$F_6 = 0 + \alpha_3 \cdot 0 + 0 \quad F_5^* = 0 + \alpha_2 \cdot (\overline{I}^{*A} + \overline{I}^{*B})$$

$$F_7 = 0 + 0.0 + T \quad F_6^* = 0 + \alpha_3 \cdot (\overline{I}^{*A} + \overline{I}^{*B})$$

$$\overline{I}^{*A} + \overline{I}^{*B} = -\frac{dH^{A/A*}}{dt} - \frac{dH^{B/B*}}{dt} = -\frac{d}{dt} (H^{A/A*} + H^{B/B*})$$

$$H^{A/A*} = \overline{I}^{A/A*} \cdot \overline{\omega}^A \quad H^{B/B*} = \overline{I}^{B/B*} \cdot \overline{\omega}^B = \overline{I}^{B/B*} \cdot \overline{\omega}^A + \overline{I}^{B/B*} \cdot \overline{\omega}^B$$

$$\therefore H^{A/A*} + H^{B/B*} = (\overline{I}^{A/A*} + \overline{I}^{B/B*}) \cdot \overline{\omega}^A + \overline{I}^{B/B*} \cdot \overline{\omega}^B$$

Define a fictitious body (rigid D) whose mass distib. is that of G, but whose motion is that of A.

Thus  $H^{A/A*} + H^{B/B*} = H^{D/D*} + \overline{H}^{B/B*}$

$$\frac{d}{dt} ( ) = \frac{d}{dt} H^{D/D*} + \frac{d}{dt} H^{B/B*} = \frac{d}{dt} H^{D/D*} + \frac{d}{dt} H^{B/B*} + \overline{\omega}^A \times H^{B/B*} - \overline{H}^{D/D*} - \overline{\omega}^A \times \overline{H}^{B/B*} + \overline{\omega}^A \times (H^{B/B*} \cdot \overline{\omega}^B)$$

$$\therefore \overline{I}^{*A} + \overline{I}^{*B} = \overline{I}^{D/D*} + \overline{H}^{B/B*} - \overline{\omega}^A \times \overline{H}^{B/B*}$$

$$= [(I^{C/C*} \cdot \overline{\omega}^A) \times \overline{\omega}^A - I^{C/C*} \cdot \overline{\omega}^A + (I^{B/B*} \cdot \overline{\omega}^B) \times \overline{\omega}^B - I^{B/B*} \cdot \overline{\omega}^B + (I^{B/B*} \cdot \overline{\omega}^B) \times \overline{\omega}^A]$$

$$\overline{\omega}^A = \frac{d \overline{\omega}^A}{dt} = \frac{d \omega^A}{dt} + \overline{\omega}^A \times \overline{\omega}^A = \dot{\omega}_i \alpha_i$$

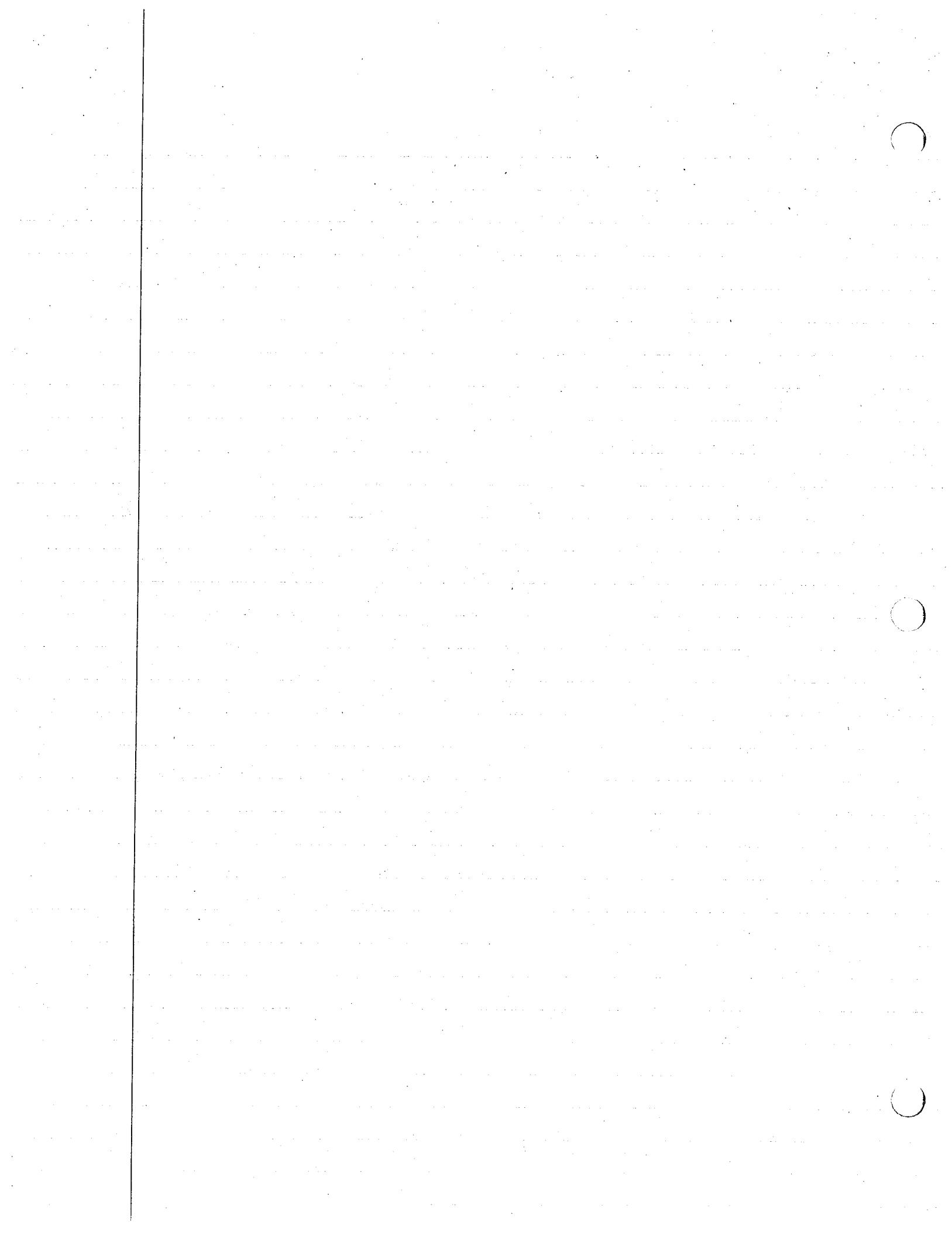
$$\overline{\omega}^B = \frac{d \overline{\omega}^B}{dt} = \overline{\omega}^B \beta$$

$$I^{C/C*} \cdot \overline{\omega}^A = I_i \omega_i \alpha_i; \quad I^{C/C*} \cdot \overline{\omega}^A = I_i \omega_i \alpha_i; \quad I^{B/B*} \cdot \overline{\omega}^B = J \omega^B \beta \quad \text{and } (I^{B/B*} \cdot \overline{\omega}^B) \times \overline{\omega}^A =$$

$$I^{B/B*} \cdot \overline{\omega}^B = J \omega^B \beta; \quad (I^{B/B*} \cdot \overline{\omega}^B) \times \overline{\omega}^A = (J \omega^B \beta \alpha_i \times \omega_j \alpha_j) = J \omega^B [(\beta_1 \omega_2 - \beta_2 \omega_1) \alpha_3 + (\beta_2 \omega_3 - \beta_3 \omega_2) \alpha_1 + (\beta_3 \omega_1 - \beta_1 \omega_3) \alpha_2]$$

$$\overline{I}^{*A} + \overline{I}^{*B} = [I_1 \omega_1 (\omega_2 \alpha_3 - \omega_3 \alpha_2) + I_2 \omega_2 (\omega_3 \alpha_1 - \omega_1 \alpha_3) + I_3 \omega_3 (\omega_1 \alpha_2 - \omega_2 \alpha_1) - I_1 \dot{\omega}_1 \alpha_1 - I_2 \dot{\omega}_2 \alpha_2 - I_3 \dot{\omega}_3 \alpha_3 - J \omega^B [\beta_1 \alpha_1 + \beta_2 \alpha_2 + \beta_3 \alpha_3] + J \omega^B (\beta_1 \omega_2 \alpha_3 - \beta_1 \omega_3 \alpha_2 + \beta_2 \omega_3 \alpha_1 - \beta_2 \omega_1 \alpha_3 + \beta_3 \omega_1 \alpha_2 - \beta_3 \omega_2 \alpha_1]]$$

$$= [I_{111111} (T_{111111} - T_{111111}) - I_{111111} - J \omega^B \beta_1 + J \omega^B (\beta_2 \omega_3 - \beta_3 \omega_2)] \alpha_1 + [I_{111111} (I_{333333} - I_{333333}) - I_{111111} - J \omega^B \beta_2 + J \omega^B (\beta_3 \omega_1 - \beta_1 \omega_3)] \alpha_2 + [I_{111111} (I_{222222} - I_{222222}) - I_{111111} - J \omega^B \beta_3 + J \omega^B (\beta_1 \omega_2 - \beta_2 \omega_1)] \alpha_3$$



Thus our 3 eqns reduce to

$$(I_1 - I_2) \tilde{\omega}_2 \tilde{\omega}_1 - J_1 (\hat{\omega}^B \beta_3 - \hat{\omega}^B [\tilde{\omega}_2 \beta_1 - \tilde{\omega}_1 \beta_2]) = I_3 \dot{\tilde{\omega}}_3 = F_3 - J_1 \beta_3 \hat{\omega}^B$$

$$(I_2 - I_3) \tilde{\omega}_2 \tilde{\omega}_3 - J_1 (\hat{\omega}^B \beta_1 - \hat{\omega}^B [\tilde{\omega}_3 \beta_2 - \tilde{\omega}_2 \beta_3]) = I_1 \dot{\tilde{\omega}}_1 = F_1 - J_1 \hat{\omega}^B \beta_1$$

$$(I_3 - I_1) \tilde{\omega}_3 \tilde{\omega}_1 - J_1 (\hat{\omega}^B \beta_2 - \hat{\omega}^B [\tilde{\omega}_1 \beta_3 - \tilde{\omega}_3 \beta_1]) = I_2 \dot{\tilde{\omega}}_2 = F_2 - J_1 \hat{\omega}^B \beta_2$$

$$\hat{\omega}^B = \dot{\theta}_4 \quad \hat{\omega}^B = \dot{\theta}_4 \quad \beta_3 = \cos \alpha \quad \beta_1 = \sin \alpha \cos \beta \quad \beta_2 = \sin \alpha \sin \beta$$

$$T + J_1 (\dot{\tilde{\omega}}_3 + \hat{\omega}^B) = 0 \quad \hat{\omega}^B = -\frac{T}{J_1} - \dot{\tilde{\omega}}_3$$

$$\text{Now } \dot{\tilde{\omega}}_1 [c_\beta c_\alpha m_1 - s_\beta m_2 + s_\alpha c_\beta m_3] + \dot{\tilde{\omega}}_2 [s_\beta c_\alpha m_1 + s_\beta m_2 + s_\alpha s_\beta m_3] + \dot{\tilde{\omega}}_3 [-s_\alpha m_1 + c_\alpha m_3] = \hat{\omega}^B$$

$$\dot{\tilde{\omega}}_1 s_\alpha c_\beta + \dot{\tilde{\omega}}_2 s_\alpha s_\beta + \dot{\tilde{\omega}}_3 c_\alpha = \dot{\tilde{\omega}}_3$$

$$\dot{\tilde{\omega}}_1 \beta_1 + \dot{\tilde{\omega}}_2 \beta_2 + \dot{\tilde{\omega}}_3 \beta_3 = \dot{\tilde{\omega}}_3$$

Solve the above for the derivatives  $\dot{\tilde{\omega}}_1, \dot{\tilde{\omega}}_2, \dot{\tilde{\omega}}_3$  & put into eqn. for  $\dot{\tilde{\omega}}_3$

$$\frac{\beta_1}{I_1} (F_1 - J_1 \hat{\omega}^B \beta_1) + \frac{\beta_2}{I_2} (F_2 - J_1 \hat{\omega}^B \beta_2) + \frac{\beta_3}{I_3} (F_3 - J_1 \hat{\omega}^B \beta_3) = \dot{\tilde{\omega}}_3$$

$$\frac{\beta_1 F_1}{I_1} + \frac{\beta_2 F_2}{I_2} + \frac{\beta_3 F_3}{I_3} = \dot{\tilde{\omega}}_3 + J_1 \hat{\omega}^B \left( \frac{\beta_1^2}{I_1} + \frac{\beta_2^2}{I_2} + \frac{\beta_3^2}{I_3} \right)$$

$$\text{Now } \hat{\omega}^B + \dot{\tilde{\omega}}_3 = \frac{T}{J_1} ; \quad \hat{\omega}^B + \frac{\beta_1 F_1}{I_1} + \frac{\beta_2 F_2}{I_2} + \frac{\beta_3 F_3}{I_3} = J_1 \hat{\omega}^B \left( \frac{\beta_1^2}{I_1} + \frac{\beta_2^2}{I_2} + \frac{\beta_3^2}{I_3} \right) = \dot{\tilde{\omega}}_3 + \hat{\omega}^B + \frac{T}{J_1}$$

$$\hat{\omega}^B \left[ 1 - J_1 \left( \frac{\beta_1^2}{I_1} + \frac{\beta_2^2}{I_2} + \frac{\beta_3^2}{I_3} \right) \right] = \frac{T}{J_1} - \left( \frac{\beta_1 F_1}{I_1} + \frac{\beta_2 F_2}{I_2} + \frac{\beta_3 F_3}{I_3} \right)$$

$$= F_1 = (I_3 - I_2) \tilde{\omega}_2 \tilde{\omega}_3 - J_1 (\tilde{\omega}_3 \beta_2 - \tilde{\omega}_2 \beta_3)$$

$$= F_2 = (I_1 - I_3) \tilde{\omega}_3 \tilde{\omega}_1 - J_1 (\tilde{\omega}_1 \beta_3 - \tilde{\omega}_3 \beta_1)$$

$$= F_3 = (I_2 - I_1) \tilde{\omega}_2 \tilde{\omega}_1 - J_1 (\tilde{\omega}_2 \beta_1 - \tilde{\omega}_1 \beta_2)$$

thus,

$$\hat{\omega}^B \left[ 1 - J_1 \left( \frac{\beta_1^2}{I_1} + \frac{\beta_2^2}{I_2} + \frac{\beta_3^2}{I_3} \right) \right] = \frac{T}{J_1} + \frac{\beta_1}{I_1} \left[ (I_3 - I_2) \tilde{\omega}_2 \tilde{\omega}_3 - J_1 (\tilde{\omega}_3 \beta_2 - \tilde{\omega}_2 \beta_3) \right]$$

$$+ \frac{\beta_2}{I_2} \left[ (I_1 - I_3) \tilde{\omega}_3 \tilde{\omega}_1 - J_1 (\tilde{\omega}_1 \beta_3 - \tilde{\omega}_3 \beta_1) \right]$$

$$+ \frac{\beta_3}{I_3} \left[ (I_2 - I_1) \tilde{\omega}_2 \tilde{\omega}_1 - J_1 (\tilde{\omega}_2 \beta_1 - \tilde{\omega}_1 \beta_2) \right]$$

with the following definitions

4300 ✓

4400 ✓

5960  $CO = 1.5 * T * \text{amu}$  ✓

6220 ✓

(1 -  $\frac{1}{\sqrt{2}}$ )  
 $b+b$ )

8750 } cont.

8775 }

9200 } cont.

9250 }

9700 }

1 }

10300 }

11900

11950

12000

W<sub>1</sub>      W<sub>2</sub>      W<sub>3</sub>      AWA  
 I<sub>1</sub>      I<sub>2</sub>      I<sub>3</sub>      J<sub>1</sub>  
 TORQUE

ROTOR MASS

T/J<sub>1</sub>

Input AMU, RM, B, C, D, AL1, ..., TJ1, ALP, BET.

Input AWB, W<sub>1</sub>, W<sub>2</sub>, W<sub>3</sub>

$$CEE = (C \cdot C - 1) / (C + C + 1)$$

$$BCE = (B \cdot B - C \cdot C) / (B \cdot B + C \cdot C)$$

$$BEE = (1 - B \cdot B) / (1 + B \cdot B)$$

$$+F - (D \cdot GT \cdot \text{ATAN}(B, C)) \quad D = \text{AMIN}(B, C)$$

$$R = D * AL1 / 2 \quad \text{radius of rotor}$$

$$ALB = C * AL1$$

I<sub>3</sub>

$$AL2 = B * AL1$$

I<sub>2</sub>

$$AJ1 = RM * R * R / 2$$

$$J1 = mR^2 / 2 \quad \text{polar mass of center of rotor}$$

$$J = \frac{m}{2} \frac{d^2 l_1^2}{4}$$

$$CO = 1.5 + AMU + D + D$$

$$\frac{3}{2} \mu d^2$$

\* \*

$$AJ1II = CO / (1 + C + C)$$

$$J1/I_1 = \frac{\frac{3}{2} \mu d^2}{1 + C^2}$$

$$AJ1I2 = CO / (B \cdot B + C + C)$$

$$J1/I_2$$

$$AJ1I3 = CO / (1 + B \cdot B)$$

$$J1/I_3$$

$$AI1 = AJ1 / AJ1II$$

$$\frac{mR^2}{2} \cdot \frac{2}{3} \frac{1+C^2 M}{d^2 m} = \frac{m}{2} \frac{d^2 l_1^2}{4} \cdot \frac{2}{3} \frac{(1+C^2)}{d^2} \frac{M}{m} = \frac{M l_1^2 (1+C^2)}{12} = I_1$$

$$AI2 = AJ1 / AJ1I2$$

I<sub>2</sub>

$$AI3 = AJ1 / AJ1I3$$

I<sub>3</sub>

$I = (AI1 + AI2 \cdot LT \cdot AI3) \text{ TYPE 618}$   
GO TO 110

$$T = TJ1 * AJ1$$

$$T = T/J_1 \cdot J_1$$

$I = (AI2 + AI3 \cdot LT \cdot AI1) \text{ TYPE 619}$   
GO TO 110

$$AL = ALP * \text{ATAN}(1.00) / 45.0$$

&

$I = (AI3 + AI1 \cdot LT \cdot AI2) \text{ TYPE 620}$   
GO TO 110

$$BE = BET * \text{ATAN}(1.00) / 45.0$$

B

$$B1 = \sin(AL) * \cos(BE)$$

B<sub>1</sub>

$$B2 = \sin(AL) * \sin(BE)$$

B<sub>2</sub>

$$B3 = \cos(AL)$$

B<sub>3</sub>

$$C1 = 1 - CO * \left( B1 * B1 / (1 + C + C) + B2 * B2 / (B \cdot B + C + C) + B3 * B3 / (1 + B \cdot B) \right)$$

$$B1C1 = B1/C1 \quad B1/C1$$

$$B3B1 = B3/C1 \quad B3/C1$$

$$B2C1 = B2/C1 \quad B2/C1$$



$$T_{I3} = T_{J1} + J_{II3}$$

$$T/I_3$$

$$\text{*** } T_{I2} = T_{J1} + J_{I2}$$

$$T/I_2$$

$$T_{II} = T_{J1} + J_{II1}$$

$$T/I_1$$

$$G_1 = CEE + Y_2 * W_3 + C_0 + (Y_3 + Y_2 - W_2 * B_3) * AWB / (1 + C + C)$$

TYPE 621 WRITE (20, 621)  
TYPE 612, AII, AI2, AI3, AJ1, T  
WRITE (20, 622) AII, AI2, AI3, AJ1, T

$$G_2 = BCE + W_3 * Y_1 + C_0 + (W_1 * B_3 - W_3 * B_1) * AWB / (B + B + C + C)$$

$$G_3 = BEE + Y_1 * W_2 + C_0 + (W_2 * B_1 - W_1 * B_2) * AWB / (1 + B + B)$$

$$H = B_1 * G_1 + B_2 * G_2 + B_3 * G_3$$

$$DY(1) \underline{W_3 DOT} = G_3 - [B_2 C_1 + (T_{I3} - AJ_{II3} + H)]$$

$$DY(2) \underline{W_2 DOT} = G_2 - [B_2 C_1 + (T_{I2} - AJ_{I2} + H)],$$

$$DY(3) \underline{W_1 DOT} = G_1 - [B_1 C_1 + (T_{II} - AJ_{II1} + H)]$$

$$DY(4) \underline{AWB DOT} = [T_{J1} - H] / C_1$$

$$DY(4) = W_1 DOT$$

$$DY(2) = W_2 DOT$$

$$DY(3) = W_3 DOT$$

$$DY(4) = AWB DOT$$

need to pass  $\check{CEE}, \check{C_0}, \check{C}, \check{B_1}, \check{B_2}, \check{B_3}, \check{BCE}, \check{B}, \check{C}, \check{BEE}$

$\check{B_2 C_1}, \check{B_2 C_1}, \check{B_1 C_1}, \check{T_{I3}}, \check{T_{I2}}, \check{T_{II}}, \check{AJ_{II3}}, \check{AJ_{II3}}, \check{AJ_{II1}}, \check{T_{J1}}, \check{C_1}$

or  $\check{T}, \check{AII}, \check{AI2}, \check{AI3} \rightarrow \check{B_1}, \check{B_2}, \check{B_3}, \check{C_1}$

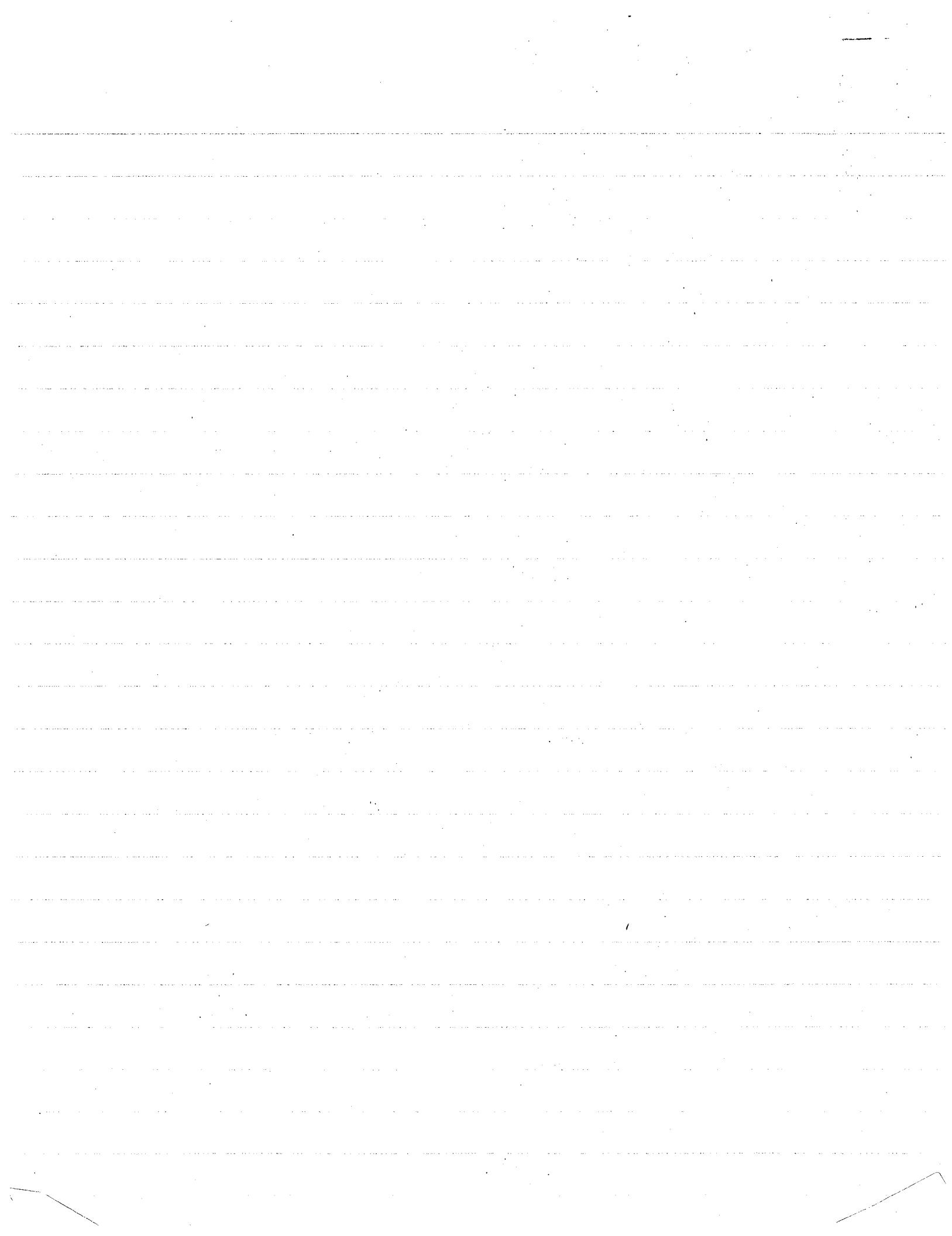
~~T<sub>J1</sub>~~

~~AII~~ ~~AI2~~ ~~AI3~~

$$G_1 = (AI_2 - AI_3) * Y(2) * Y(3) + AJ_1 * [Y(3) * B_2 - Y(2) * B_3] * Y(4).$$

$$G_2 = (AI_3 - AI_1) * Y(3) - Y(1) + AJ_1 * [Y(1) * B_3 - Y(3) * B_1] * Y(4)$$

$$G_3 = (AI_1 - AI_2) * Y(1).$$



NOME:

TRABALHO:

### 3.7 Dynamical equations for a simple gyrostat

In Fig. 3.7.1,  $G$  designates a simple gyrostat consisting of a rigid body  $A$  and an axisymmetric rotor  $B$  whose axis and mass center,  $B^*$ , are fixed in  $A$ .  $\underline{\alpha}_1, \underline{\alpha}_2, \underline{\alpha}_3$  form a dextral set of mutually perpendicular unit vectors fixed in  $A$ , each parallel to a central principal axis of inertia of  $G$ . The axis of  $B$  is parallel to  $\underline{\alpha}_3$ , and  $G$  has central principal moments of inertia  $I_1, I_2, I_3$  while  $B$  has an axial moment of inertia  $J$ .

After defining  ${}^A\omega^B$ ,  $w_i$ ,  $\beta_i$ , and  $M_i$  as

$${}^A\omega^B \triangleq {}^A\omega^B \cdot \underline{\beta} \quad (1)$$

$$w_i \triangleq {}^N\omega^A \cdot \underline{\alpha}_i \quad (i=1,2,3) \quad (2)$$

$$\beta_i \triangleq \underline{\beta} \cdot \underline{\alpha}_i \quad (i=1,2,3) \quad (3)$$

$$M_i \triangleq M \cdot \underline{\alpha}_i \quad (i=1,2,3) \quad (4)$$

where  ${}^A\omega^B$  is the angular velocity of  $B$  in  $A$ ,  ${}^N\omega^A$  is the angular velocity of  $A$  in a Newtonian reference frame  $N$ , and  $M$  is the moment about the mass center  $G^*$  of  $G$  of all forces acting on  $G$ , one can write the following set of coupled differential equations governing  ${}^A\omega^B$  and  $w_i$  ( $i=1,2,3$ ):

$$(I_3 - J) \dot{{}^A\omega^B} = -(I_1 - I_2) w_1 w_2 - M_3 + (I_3/J) \underline{\alpha}_3 \cdot \underline{M}^{AB} \quad (4)$$

where  $\underline{M}^{AB}$  is the moment about  $B^*$  of all forces exerted by  $A$  on  $B$ , and



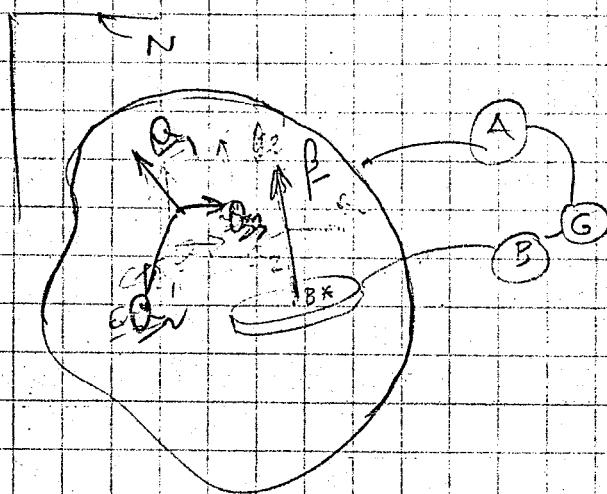


Fig. 3.7.1



NOME:

TRABALHO:

$$\begin{aligned}
 {}^A\ddot{\omega}^B \left[ 1 - J \left( \frac{\beta_1^2}{I_1} + \frac{\beta_2^2}{I_2} + \frac{\beta_3^2}{I_3} \right) \right] &= \frac{{}^T\beta \cdot M^{A/B}}{J} + \\
 &+ \frac{\beta_1}{I_1} \left[ (I_2 - I_3) w_2 w_3 + J {}^A\ddot{\omega}^B (\beta_2 w_3 - \beta_3 w_2) - M_1 \right] + \\
 &+ \frac{\beta_2}{I_2} \left[ (I_3 - I_1) w_3 w_1 + J {}^A\ddot{\omega}^B (\beta_3 w_1 - \beta_1 w_3) - M_2 \right] + \\
 &+ \frac{\beta_3}{I_3} \left[ (I_1 - I_2) w_1 w_2 + J {}^A\ddot{\omega}^B (\beta_1 w_2 - \beta_2 w_1) - M_3 \right] \quad (5)
 \end{aligned}$$

where  $M^{A/B}$  is the moment about  $B^*$  of all forces exerted by A on B, and

$$I_1 \ddot{w}_1 = (I_2 - I_3) w_2 w_3 - J [{}^A\ddot{\omega}^B \beta_1 - {}^A\ddot{\omega}^B (\beta_2 w_3 - \beta_3 w_2)] + M_1 \quad (6)$$

$$I_2 \ddot{w}_2 = (I_3 - I_1) w_3 w_1 - J [{}^A\ddot{\omega}^B \beta_2 - {}^A\ddot{\omega}^B (\beta_3 w_1 - \beta_1 w_3)] + M_2 \quad (7)$$

$$I_3 \ddot{w}_3 = (I_1 - I_2) w_1 w_2 - J [{}^A\ddot{\omega}^B \beta_3 - {}^A\ddot{\omega}^B (\beta_1 w_2 - \beta_2 w_1)] + M_3 \quad (8)$$

If the axis of B is parallel to a central principal axis of inertia of G, the complexity of these equations is reduced considerably; and further simplifications can be made if B is completely free to rotate relative to A or if B is made to rotate relative to A with a constant angular speed, say by means of a motor each of whose parts belongs either to A or to B. Under these circumstances, and with  $\beta = \alpha_3$ , the differential equations governing  $w_1$ ,  $w_2$ , and  $w_3$  can be written



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$$I_1 \ddot{w}_1 = (I_2 - \bar{I}_3) w_2 w_3 - \sqrt{\sigma} w_2 + M_1 \quad (9)$$

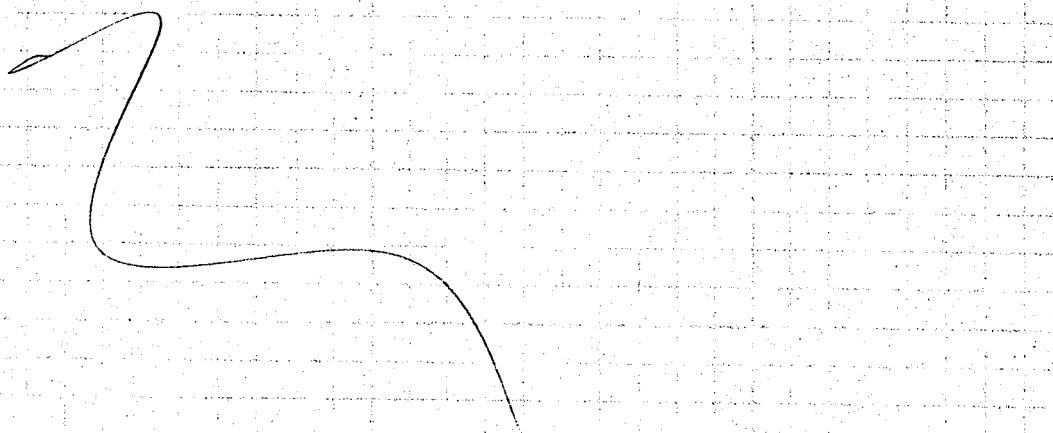
$$I_2 \ddot{w}_2 = (\bar{I}_3 - I_1) w_3 w_1 + \sqrt{\sigma} w_1 + M_2 \quad (10)$$

$$\bar{I}_3 \ddot{w}_3 = (I_1 - I_2) w_1 w_2 + M_3 \quad (11)$$

where  $\bar{I}_3$  and  $\sigma$  are given in Table 3.7.1, with  $\hat{w}^B$  and  $\hat{w}_3$  denoting the initial values of  $w^B$  and  $w_3$ , respectively.

Rotor motion	$\bar{I}_3$	$\sigma$
Free	$I_3 - \sqrt{\sigma}$	$\hat{w}_3 + \hat{w}^B$
$A^B w^B = A^A w^A$	$I_3$	$A \hat{w}^B$

Table 3.7.1





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Derivations: The inertia torque  $\underline{T}_B$  of  $B$  in  $N$  is given by

$$\underline{T}_B = -\underline{\alpha}^B \cdot \underline{I}_B - \underline{\omega}^B \times \underline{I}_B \cdot \underline{\omega}^B \quad (12) \#40$$

where  $\underline{\alpha}^B$  and  $\underline{\omega}^B$  are the angular acceleration and the angular velocity of  $B$  in  $N$ , and  $\underline{I}_B$ , the central inertia dyadic of  $B$ , can be expressed as

$$\begin{aligned} \underline{I}_B &= K(\underline{\beta}\underline{\gamma} + \underline{\delta}\underline{\delta}) + J\underline{\beta}\underline{\beta} \\ &= K\underline{U} + (J-K)\underline{\beta}\underline{\beta} \end{aligned} \quad (13)$$

Here  $K$  is the transverse central moment of inertia of  $B$ ,  $\underline{\gamma}$  and  $\underline{\delta}$  are unit vectors such that  $\underline{\beta}$ ,  $\underline{\gamma}$ , and  $\underline{\delta}$  are mutually perpendicular, and  $\underline{U}$  is the unit dyadic. Hence

$$\underline{\alpha}^B \cdot \underline{I}_B = K\underline{\alpha}^B + (J-K)\underline{\alpha}^B \cdot \underline{\beta}\underline{\beta} \quad (14) \#40$$

and

$$\underline{\omega}^B \times \underline{I}_B \cdot \underline{\omega}^B = (J-K)\underline{\omega}^B \times \underline{\beta}_3 \underline{\beta}_3 \cdot \underline{\omega}^B \quad (15) \#40$$

From which it follows that

$$\underline{T}_B \cdot \underline{\beta}_3 = -J\underline{\alpha}^B \cdot \underline{\beta}_3 = {}^A\omega_{u_4}^B \cdot \underline{\pi}^{AB} \quad (16) \#40$$

Since by D'Alembert's Principle

$$\underline{T}_B \cdot \underline{\beta}_3 + \underline{M}^{AB} \cdot \underline{\beta}_3 = 0 \quad (17) \#40$$

one thus has

$$J\underline{\alpha}^B \cdot \underline{\beta}_3 = \underline{M}^{AB} \cdot \underline{\beta}_3 \quad (18) \#40$$

Now,

$$\underline{\alpha}^B = \frac{d \underline{\omega}^B}{dt} \stackrel{(1.20.1)}{=} \frac{d}{dt} (\underline{\omega}^A + {}^A\omega^B \underline{\beta}_3) \stackrel{(16.1)}{=}$$



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Consequently,

$$\sqrt{(\underline{\alpha}^A \cdot \underline{\beta}_B + {}^A\omega^B)} = \underline{\beta} \cdot \underline{M}^{A/B} \quad (20) \text{ for } (9)$$

and since

$$\underline{\alpha}^A \cdot \underline{\beta} = \omega_1 \beta_1 + \omega_2 \beta_2 + \omega_3 \beta_3 \quad (21) \quad (20)$$

$$\dot{\alpha}^A = \frac{d \omega^A}{dt} = \frac{d}{dt} w_i \alpha_i = \dot{w}_i \alpha_i + (2)N \left( w_i \frac{d \alpha_i}{dt} = w_i \frac{d \alpha_i}{dt} + w_i \cdot B^A \times \alpha_i = w_i w_j \alpha_j \times \alpha_i \right) = \dot{w}_i \alpha_i$$

one can write

$$J(\omega_1\beta_1 + \omega_2\beta_2 + \omega_3\beta_3 + \omega^B) = \beta \cdot M^{A/B} \quad (20) \quad (21)$$

$T_R$ , the inertia torque of a rigid body R that has the same mass distribution as G, but moves like A, is given by

$$T_R = -\alpha^A \cdot I_G - \omega^A \times I_G \cdot \omega^A \quad (23) \quad (22)$$

Hence,  $T_G$ , the inertia torque of  $G$  in  $N$ , can be expressed with the aid of Eqs. (3.6.2) and (3.6.5) as

$$\frac{T}{G} = -\underline{\alpha}^A \cdot \underline{I}_G - \underline{\omega}^A \times \underline{I}_G \cdot \underline{\omega}^A - J(\underline{\omega}^B \underline{\beta}_2 + \underline{\omega}^B \underline{\omega}^A \times \underline{\beta}_2) \quad (24)$$

Carrying out the operations indicated in Eq.(24) after noting that

$$I_G = I_1 \alpha_1 \alpha_1 + I_2 \alpha_2 \alpha_2 + I_3 \alpha_3 \alpha_3 \quad (25)(26)$$

and then appealing to D'Alembert's principle to write

$$\underline{T}_G + \underline{M} = 0 \quad (26)(25)$$



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one arrives at Eqs. (6) - (8); and solution of these equations for  $\omega_1, \omega_2, \omega_3$ , followed by substitution into Eq. (22), produces Eq. (5).

Suppose now that  $\beta = \alpha_3$ , so that [see Eq. (3)]

$$\beta_1 = \beta_2 = 0, \quad \beta_3 = 1 \quad (27)$$

Then

$$\overset{A}{\omega} \left( 1 - \frac{J}{I_3} \right) = \frac{\beta \cdot M^{1/3}}{J} + \frac{I_2 - I_1}{I_3} \omega_1 \omega_2 - \frac{M_3}{I_3} \quad (28)$$

and

$$I_1 \omega_1 = (I_2 - I_3) \omega_2 \omega_3 - J \overset{A}{\omega} \omega_2 + M_1 \quad (29)$$

$$I_2 \omega_2 = (I_3 - I_1) \omega_3 \omega_1 + J \overset{A}{\omega} \omega_1 + M_2 \quad (30)$$

$$I_3 \omega_3 = (I_1 - I_2) \omega_1 \omega_2 - J \overset{A}{\omega} \omega_3 + M_3 \quad (31)$$

~~If B is completely free to rotate relative to A, that is, if  $\beta \cdot M^{1/3} = 0$ , then~~

~~$$\overset{A}{\omega} = \frac{I_2 - I_1}{I_3 - J} \omega_1 \omega_2 - \frac{M_3}{I_3 - J} \quad (32)$$~~

so that

~~$$I_3 \omega_3 = (I_1 - I_2) \omega_1 \omega_2 - \frac{J(I_2 - I_1)}{I_3 - J} \omega_1 \omega_2 + \frac{J M_3 + M_3}{I_3 - J} \quad (33)$$~~

or, equivalently,

~~$$(I_3 - J) \omega_3 = (I_1 - I_2) \omega_1 \omega_2 + M_3 \quad (34)$$~~



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If  $B$  is completely free to rotate relative to  $A$ , that is, if  $\beta, M^{AB} = 0$ , then [see Eqs. (27)]

$$\dot{\omega}_3 + \dot{\omega}^B = 0 \quad (32)$$

which implies that

$$\dot{\omega}_3 + \dot{\omega}^B = \dot{\omega}_3 + \dot{\omega}^{AB} \quad (33)$$

where  $\dot{\omega}_3$  and  $\dot{\omega}^{AB}$  are the initial values of  $\omega_3$  and  $\dot{\omega}^B$ , respectively. Hence

$$I_1 \ddot{\omega}_1 = (I_2 - I_3) \omega_2 \omega_3 - J(\dot{\omega}_3 + \dot{\omega}^{AB} - \dot{\omega}_3) \omega_2 + M_1 \quad (29)$$

$$= (I_2 - I_3 + J) \omega_2 \omega_3 - J(\dot{\omega}_3 + \dot{\omega}^{AB}) + M_1 \quad (34)$$

Similarly, from Eqs. (30) and (33),

$$I_2 \ddot{\omega}_2 = (I_3 - J - I_1) \omega_3 \omega_1 + J(\dot{\omega}_3 + \dot{\omega}^{AB}) + M_2 \quad (35)$$

Furthermore,

$$\dot{\omega}^B = \frac{I_2 - I_1}{I_3 - J} \omega_1 \omega_2 - \frac{M_3}{I_3 - J} \quad (36)$$

and substitution into Eq. (31) yields

$$(I_3 - J) \ddot{\omega}_3 = (I_1 - I_2) \omega_1 \omega_2 + M_3 \quad (37)$$

Replacing  $I_3 - J$  with  $\bar{I}_3$  and  $\dot{\omega}_3 + \dot{\omega}^B$  with  $\sigma$  in Eqs. (34), (35), and (37), one arrives at Eqs. (9), (10), and (11) respectively. On the other hand, if

$B$  is made to rotate relative to  $A$  with the constant angular speed  $\dot{\omega}^B$ , then Eqs. (29) - (31)



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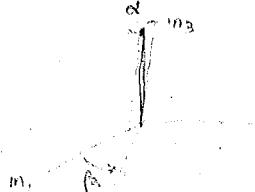
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become Eqs. (9)-(11) when  $I_3$  is replaced with  $\bar{I}_3$ ,  $\hat{\omega}^1$  is replaced with  $\sigma$ , and  $\hat{\omega}^3$  is set equal to zero. Thus Eqs. (9)-(11) apply in both cases, provided  $\bar{I}_3$  and  $\sigma$  be interpreted in accordance with Table 3.7.1.





1. Trends if all else remains the same: as  $\mu = \frac{rm}{rm+mb} \uparrow$  angular velocity of body  $\uparrow$

as well as that of the rotor.  $rm(1-\mu) = mb$  for fixed  $rm$  as  $\mu \uparrow mb \uparrow$ : effect of rotor becomes smaller. Hence the induced  $\omega_3$  of body  $\uparrow$  as expected.

2. for fixed  $\mu$   $\frac{mb}{rm} = \frac{1-\mu}{\mu}$ . Thus as the rotor mass changes for fixed  $\mu$  the body mass changes also. Hence we get no effect here. Motion here is still unstable as  $\omega$  increases as to increase

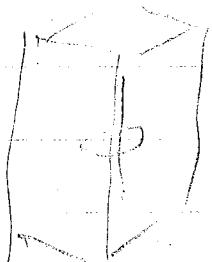
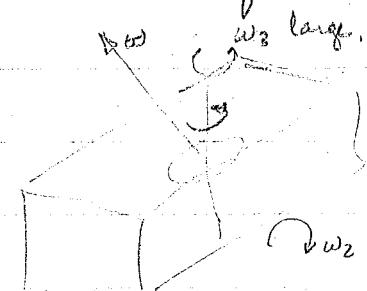
{ also the value of  $\omega_3$  remains negative for longer periods of time. Motion is quite unstable.

3. if we only change the  $T/J_1$  ratio and keep it very much below 1. then the value of  $|\omega_3| \downarrow$  as  $T/J_1 \uparrow$  and the order of magnitudes for  $|\omega_3| \uparrow$  drastically the  $\omega_3$  component decreases then starts to increase.  $\omega_3$  component is affected more than others because rotor axis is almost aligned with  $m_3$  axis. For very low values of  $T/J_1$  body rotates very slowly and  $|\omega_3|$  is very small as expected.

4.5 As  $\alpha$  increases the  $\omega$ 's affected as  $\omega_1$  and  $\omega_2$  since the rotor axes approaches the  $m_1-m_2$  plane this is noticed by the change in sign of  $\omega_1$  &  $\omega_2$ . As  $\alpha$  increases the largest effect is on  $\omega_1$ , since large changes in  $\alpha$  means that the rotor axis has a major component of itself  $\parallel$  to the  $m_1$  axis. We note that the trends in the results show that the axis upon which the major component of the rotor axis lies will always have a positive component of  $\omega$  associated with it. This is also true with  $\beta$  changes. Here the rotor's axis makes its major projection on the  $m_3$  axis when  $\beta = 90^\circ$  hence sign of  $\omega_3 = \text{sign of } \overset{\text{A}}{\omega} \overset{\text{B}}{\omega}$

6. If we keep everything else fixed and change the length of the sides  $l_2$  only we note that when  $l_2 \uparrow \text{to } 1$  Each value of  $\omega_i$  slows down & speeds up in cycles with max. of speed up for each cycle increasing with time.  $\omega_2$  component changes direction cyclically. When  $l_2 > 1$   $\omega_1$  slows down & speeds up with larger periods

as  $C$  gets larger that the motion is more stable & much less in magnitude than that of  $B$  getting larger. Reason. since  $C$  is generally in direction of axis is taken, much less to move body  $B$ . In case where  $B$  is much larger,  $B$  is  $\perp$  generally to axis of rotor  $\Rightarrow$  must increase  $\omega$  in order to move body to stay in equilibrium.



Instabilities die out quicker in case where  $C$  is larger

d. as  $d \downarrow$ ,  $\omega$ 's decrease & effect of rotor decreases.

(if direction of axis is kept fixed)

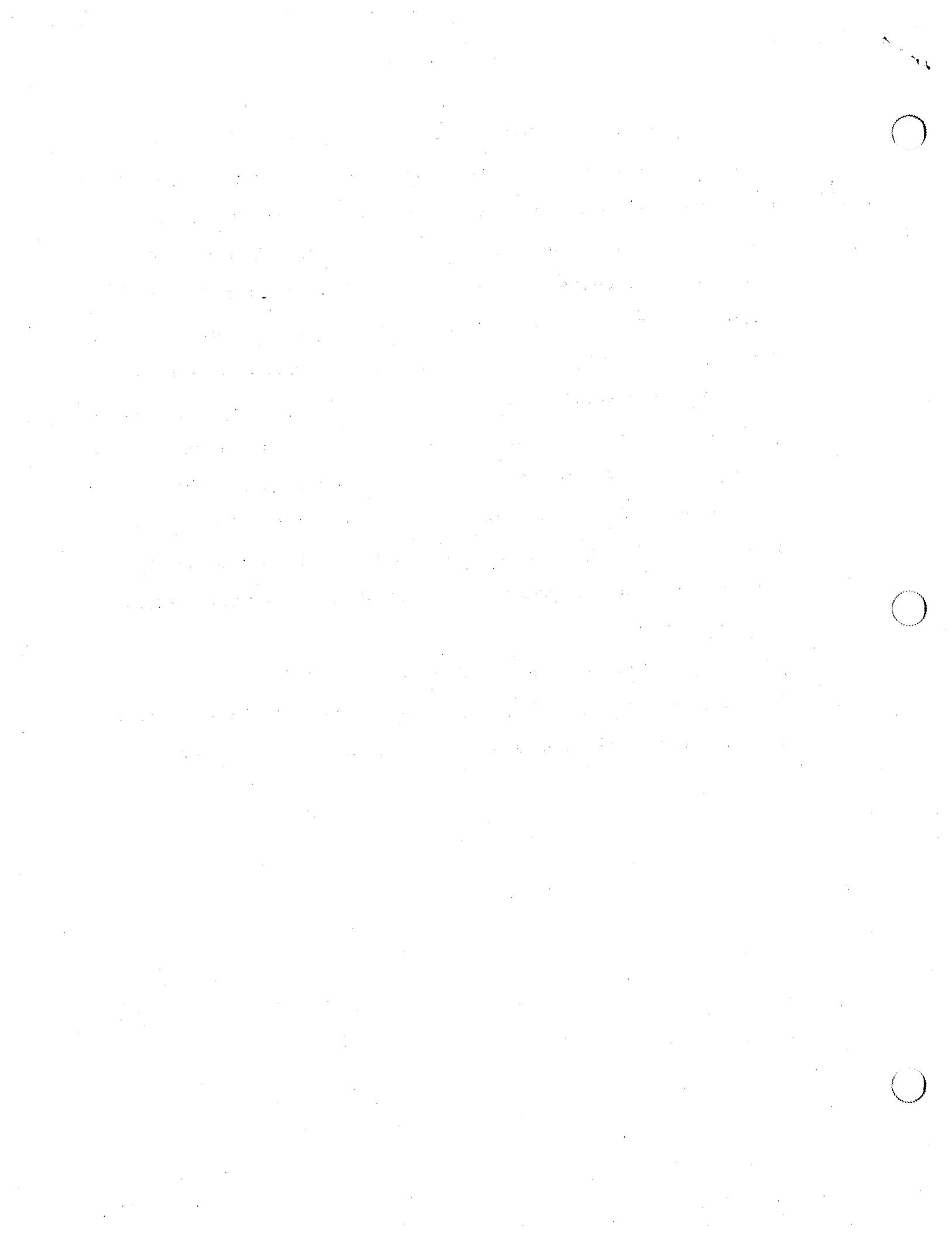
Thus for a stable system, make  $d$  small and elongate  $C_A$  over the other 2,

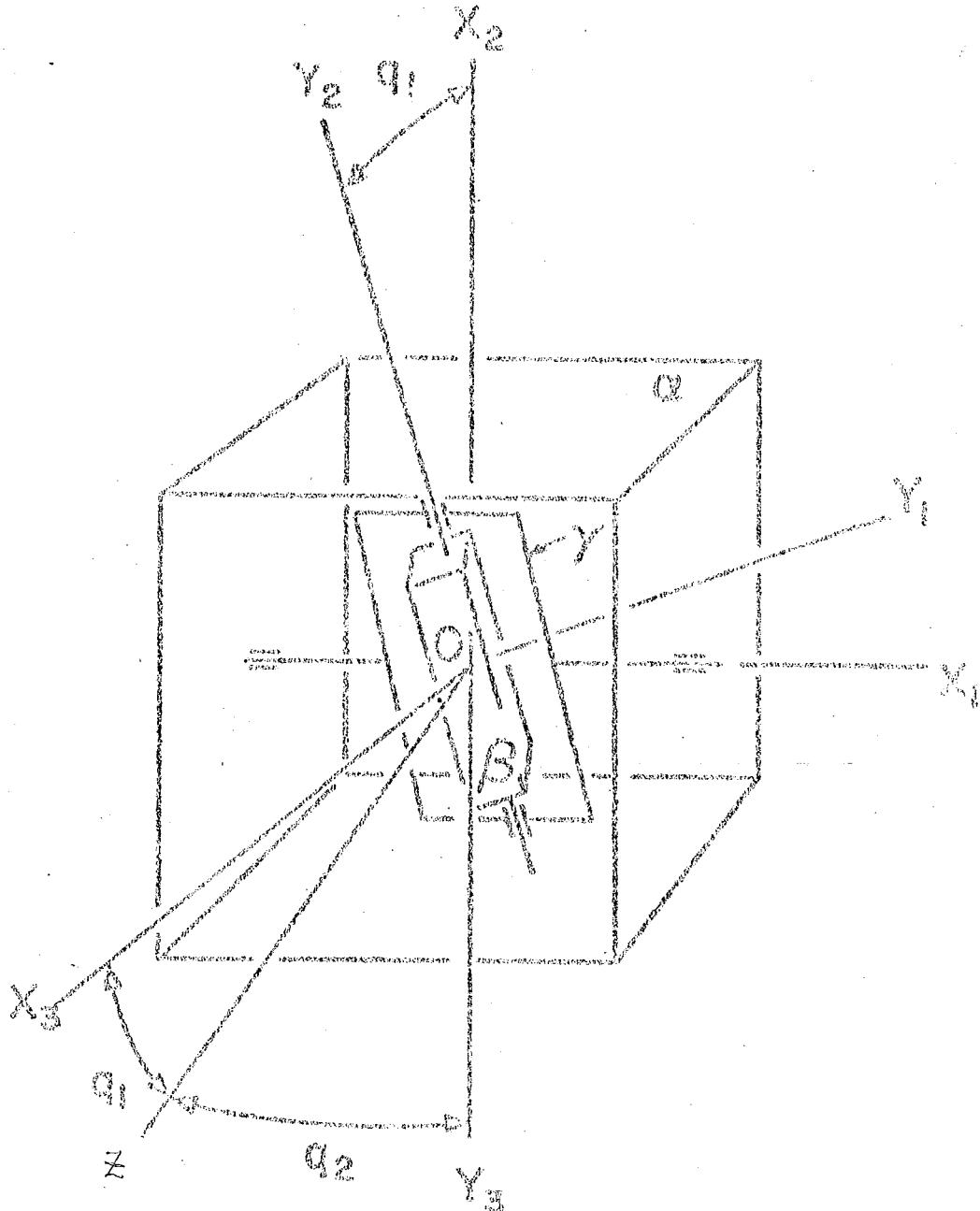
decrease  $\mu$  and  $T/J_1$

The attached sketch shows a system  $S$  composed of two rigid bodies,  $\alpha$  and  $\beta$ , which are connected to each other by means of a light gimbal  $\gamma$ . Point  $O$  is the mass center of  $\alpha$  as well as  $\beta$ ;  $X_1, X_2, X_3$  are principal axes of  $\alpha$ ;  $Y_1, Y_2, Y_3$  are principal axes of  $\beta$ ; and  $\gamma$  can rotate relative to  $\alpha$  and  $\beta$  only about  $X_1$  and  $Y_2$ , respectively, so that the relative orientation of  $\alpha$  and  $\beta$  depends solely on the angles  $q_1$  and  $q_2$ . The central principal moments of inertia of  $\alpha$  and  $\beta$  have the values  $A_1, A_2, A_3$  and  $B_1, B_2, B_3$ , respectively.

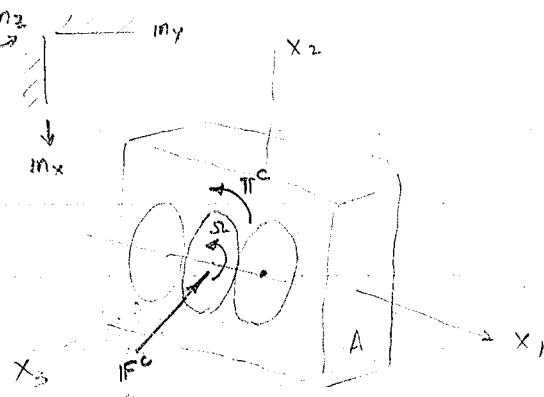
When  $q_1 = q_2 = 0$ , the central principal axes of  $S$  coincide with those of  $\alpha$  and  $\beta$ . Letting  $Z$  be the central principal axis of  $S$  that coincides with  $X_3$  and  $Y_3$  when  $q_1 = q_2 = 0$ , and taking  $A_1 = 4$ ,  $A_2 = 2$ ,  $A_3 = 3$ ,  $B_1 = 1.5$ ,  $B_2 = 3 \text{ kg cm}^2$ , find a value of  $B_3$  such that the moment of inertia of  $S$  about  $Z$  has a (local) minimum value when  $q_1 = q_2 = 0$ .

Note:  $f(x, y)$  has a local minimum at  $x = y = 0$  if  $f_x = f_y = 0$ ,  $f_{xx} > 0$ , and  $f_{xx}f_{yy} - (f_{xy})^2 > 0$ , where subscripts denote partial differentiations and all derivatives are evaluated at  $x = y = 0$ .









this is a 7 degree of freedom system.

to describe motion of body A in R

to describe motion of body C in A

motions of  $B_1$  and  $B_2$  are known since their motion depend on motion of C

$$\text{Assume That } \Pi^C = T_1 m_1 + T_2 m_2 + T_3 m_3$$

$$IF^C = f_1 m_1 + f_2 m_2 + f_3 m_3$$

$$R\omega^A = \omega_1 m_1 + \omega_2 m_2 + \omega_3 m_3 \\ = u_1 m_1 + u_5 m_2 + u_6 m_3$$

Assume that  $C^*$  and  $A^*$  are same and rotors have radii R

$$u_1 m_1 + u_2 m_2 + u_3 m_3 = W^{A^*} = v_1 m_1 + v_2 m_2 + v_3 m_3 = W^{C^*}$$

$$u_1 m_1 + u_2 m_2 + u_3 m_3 = W^{C^*} = v_1 m_1 + v_2 m_2 + v_3 m_3$$

$$W^{C^*} + 2r[u_6 m_2 - u_5 m_3] = W^{B_1^*} = W^{C^*} + 2r[\omega_3 m_2 - \omega_2 m_3]$$

$$W^{C^*} + 2r[-u_6 m_2 + u_5 m_3] = W^{B_2^*} = W^{C^*} + 2r[-\omega_3 m_2 + \omega_2 m_3]$$

$$A\omega^C = \dot{s}_2 m_3 = u_7 m_3$$

$$A\omega^C = \dot{s}_2 m_3 \\ = u_7 m_3$$

$$A\omega^{B_1} = \dot{s}_2 m_3 \\ = u_7 m_3$$

$$A\omega^{B_2} = -\dot{s}_2 m_3 \\ = -u_7 m_3$$

$$IR^{B_1^*} = IR^{C^*} + 2r m_1$$

$$R\omega^{B_1^*} = W^{C^*} + 2r \frac{d\omega}{dt} m_1$$

$$\frac{d\omega}{dt} m_1 = \frac{d\omega}{dt} + \omega \times R^A \times m_1 \\ = W^{C^*} + 2r \omega^A \times m_1 = W^{C^*} + 2r [-\omega_2 m_3 + \omega_3 m_2]$$

$$R\omega^{B_2^*} = W^{C^*} - 2r \omega^A \times m_1 = W^{C^*} + 2r [\omega_2 m_3 - \omega_3 m_2]$$

Now let the generalized coordinates of entire system be  $u_i (i=1,2,3)$ ,  $w_i (i=1,2,3)$  and  $\theta$

now  $u_1, u_2, u_3$  equations of motion will decouple from rest.

$$r=1,2,3 \quad F_r = W_{u_r}^A \cdot IF^A + W_{u_r}^C \cdot IF^C + W_{u_r}^{B_1} \cdot IF^{B_1} + W_{u_r}^{B_2} \cdot IF^{B_2} + \dot{\theta} \omega_{u_r}^A \cdot \Pi^A + \dot{\theta} \omega_{u_r}^C \cdot \Pi^C + \dot{\theta} \omega_{u_r}^{B_1} \cdot \Pi^{B_1} + \dot{\theta} \omega_{u_r}^{B_2} \cdot \Pi^{B_2}$$

since only body & contact forces acting are gravitational forces  $IF_B^{B_2} = IF_B^{B_1} = IF_B^C = mg m_x$   $IF_B^A = Mg m_x$

the contact forces between the rotors contribute nothing

$$IF^A = Mg m_x - f_1 m_1 \quad IF^{B_1} = IF^{B_2} = mg m_x \quad IF^C = mg m_x + f_3 m_3$$

$$\Pi^C = \Pi^A \quad \Pi^{B_1} = \Pi^{B_2} = 0$$

$$\text{let } u_i = v_i \quad (i=1,2,3)$$

$$u_i = w_{i-3} \quad (i=4,5,6)$$

$$u_i = \dot{s}_2 \quad i=7$$

$$F_1 = m_1 \cdot IF^A + m_1 \cdot IF^C + m_1 \cdot IF^{B_1} + m_1 \cdot IF^{B_2} + 0 + 0 + 0 + 0 \quad \text{since } \omega_{u_1} \neq \text{fn}(u_1, u_2, u_3)$$

$$F_2 = m_2 \cdot IF^A + m_2 \cdot IF^C + m_2 \cdot IF^{B_1} + m_2 \cdot IF^{B_2} + 0 + 0 + 0 + 0$$

$$F_3 = m_3 \cdot IF^A + m_3 \cdot IF^C + m_3 \cdot IF^{B_1} + m_3 \cdot IF^{B_2} + 0 + 0 + 0 + 0$$

$$F_4 = 0 + 0 + 0 + 0 + m_1 \cdot \Pi^A + m_1 \cdot \Pi^C + m_1 \cdot \Pi^{B_1} + m_1 \cdot \Pi^{B_2} \quad \text{since } W_{u_1} \neq \text{fn of } u_4$$

$$F_5 = 0 + 0 + 2r m_3 \cdot IF^{B_1} + 2r m_3 \cdot IF^{B_2} + m_2 \cdot \Pi^A + m_2 \cdot \Pi^C + m_2 \cdot \Pi^{B_1} + m_2 \cdot \Pi^{B_2}$$

$$F_6 = 0 + 0 + 2r m_2 \cdot IF^{B_1} - 2r m_3 \cdot IF^{B_2} + m_3 \cdot \Pi^A + m_3 \cdot \Pi^C + m_3 \cdot \Pi^{B_1} + m_3 \cdot \Pi^{B_2}$$



$$F_7 = 0 + 0 + 0 + 0 + 0 + m_3 \cdot \pi^c - m_3 \cdot \pi^{B_1} - m_3 \cdot \pi^{B_2} \quad \text{since } \pi_{u_7}^{(A, C, B_1, B_2)} \neq f_n(u_7)$$

Question: if motor exerts a force & a torque on C does it exert a reverse force & reverse torque to

$$A? \text{ Assume yes } \Rightarrow \pi^A = \pi^C \text{ and } \pi^A = \pi_{AB}^A - \pi_{T}^A \quad \pi^C = \pi_{AB}^C + \pi_{T}^C$$

$$\therefore F_1 = m_1 \cdot [\pi_{AB}^A + \pi_{AB}^C + 2\pi_{T}^A] = m_1 \cdot m_x [(M+3m)g]$$

$$F_2 = m_2 \cdot [\pi_{AB}^A + \pi_{AB}^C + 2\pi_{T}^A] = m_2 \cdot m_x [(M+3m)g]$$

$$F_3 = m_3 \cdot [\pi_{AB}^A + \pi_{AB}^C + 2\pi_{T}^A] = m_3 \cdot m_x [(M+3m)g]$$

$$F_4 = 0$$

$$F_5 = 0$$

$$F_6 = 0$$

$$F_7 = T_3$$

$$\text{To obtain } F_1^* = \pi_{u_1}^{(A, C, B_1, B_2)} = \pi_{u_1}^{(A, C)} + \pi_{u_1}^{(C, C)} + \pi_{u_1}^{(B_1, B_1)} + \pi_{u_1}^{(B_2, B_2)} + \omega_{u_1}^A \cdot \pi^{(A, A)} + \omega_{u_1}^C \cdot \pi^{(C, C)} + \omega_{u_1}^{B_1} \cdot \pi^{(B_1, B_1)} + \omega_{u_1}^{B_2} \cdot \pi^{(B_2, B_2)}$$

$$F_1^* = m_1 \cdot (-M\alpha_1^A) + m_1 \cdot (-m\alpha_1^C) + m_1 \cdot (-m\alpha_1^{B_1}) + m_1 \cdot (-m\alpha_1^{B_2}) + 0 + 0 + 0 + 0 \quad \text{since } \pi_{u_1}^{(A, C, B_1, B_2)} \neq f_n(u_1)$$

$$m_1 \cdot \alpha_1^A = m_1 \frac{d}{dt} \pi_{u_1}^{(A, C, B_1, B_2)} + m_1 \cdot \omega_1^A \times \pi_{u_1}^{(A, C, B_1, B_2)} = \dot{u}_1 + m_1 \cdot [u_4 m_1 + u_5 m_2 + u_6 m_3] \times [u_1 m_1 + u_2 m_2 + u_3 m_3]$$

$$m_1 \cdot \alpha_1^C = m_1 \cdot \frac{d}{dt} \pi_{u_1}^{(C, C)} + m_1 \cdot \omega_1^C \times \pi_{u_1}^{(C, C)} = \dot{u}_1 + u_5 u_3 - u_6 u_2$$

$$m_1 \cdot \alpha_1^{B_1} = m_1 \cdot \frac{d}{dt} \pi_{u_1}^{(B_1, B_1)} + m_1 \cdot \omega_1^{B_1} \times \pi_{u_1}^{(B_1, B_1)} = \dot{u}_1 - 2r [u_5^2 + u_6^2] + 2ru_1 [u_5 u_3 - u_6 u_2]$$

$$m_1 \cdot \alpha_1^{B_2} = m_1 \cdot \frac{d}{dt} \pi_{u_1}^{(B_2, B_2)} + m_1 \cdot \omega_1^{B_2} \times \pi_{u_1}^{(B_2, B_2)} = \dot{u}_1 + 2r [u_5^2 + u_6^2] - 2ru_1 [u_5 u_3 - u_6 u_2]$$

$$[F_1^* = -(M+3m)\dot{u}_1 - (M+m)(u_5 u_3 - u_6 u_2)] = -(M+3m)[\dot{u}_1 + u_5 u_3 - u_6 u_2] \quad \checkmark$$

$$F_2^* = m_2 \cdot (-M\alpha_2^A - m\alpha_2^C - m\alpha_2^{B_1} - m\alpha_2^{B_2}) \quad \text{since } \pi_{u_2}^{(A, C, B_1, B_2)} \neq f_n(u_2)$$

$$= -(M+3m)\dot{u}_2 - (M+m)(u_6 u_1 - u_4 u_3) = -(M+3m)[\dot{u}_2 + u_6 u_1 - u_4 u_3] \quad \checkmark$$

$$F_3^* = m_3 \cdot (-M\alpha_3^A - m\alpha_3^C - m\alpha_3^{B_1} - m\alpha_3^{B_2})$$

$$F_3^* = -(M+3m)\dot{u}_3 - (M+m)(u_4 u_2 - u_5 u_1) = -(M+3m)[\dot{u}_3 + u_4 u_2 - u_5 u_1]$$

$$F_4^* = 0 + 0 + 0 + 0 + m_1 \cdot \pi^{(A, C)} + m_1 \cdot \pi^{(C, C)} + m_1 \cdot \pi^{(B_1, B_1)} + m_1 \cdot \pi^{(B_2, B_2)} \quad \text{since } \pi_{u_4}^{(A, C, B_1, B_2)} \neq f_n(u_4)$$

$$[F_4^* = m_1 \cdot [\pi^{(A, C)} + \pi^{(C, C)} + \pi^{(B_1, B_1)} + \pi^{(B_2, B_2)}]]$$

$$F_5^* = 0 + 0 - 2rm_3 [-m\alpha_3^A] + 2rm_3 [-m\alpha_3^{B_2}] + m_2 \cdot [\pi^{(A, C)} + \pi^{(C, C)} + \pi^{(B_1, B_1)} + \pi^{(B_2, B_2)}]$$

$$= +2rm [m_3 (\alpha_3^{B_1} - \alpha_3^{B_2})] + m_2 \cdot [$$

$$[F_5^* = 2rm [-4r\dot{u}_5 + 4ru_4u_6] + m_2 \cdot [\pi^{(A, C)} + \pi^{(C, C)} + \pi^{(B_1, B_1)} + \pi^{(B_2, B_2)}]]$$



$$\alpha_1^{B_1, B_2} = \frac{d\omega^c}{dt} + \omega^A \times \omega^c \pm 2r \frac{d\omega^A}{dt} \times m_1 \pm 2r \frac{\omega^A}{r} \times (\omega^A \times m_1)$$

$$\alpha_1^{B_1} - \alpha_2^{B_2} = 4r \frac{d\omega^A}{dt} \times m_1 + 4r \left[ (\omega^A \cdot m_1) \omega^A - \omega^A \cdot m_1 \right] \quad \text{CHECK THIS}$$

$$F_6^* = 0 + 0 + 2r m_2 [ -m \alpha_1^{B_1} ] - 2r m_2 [ -m \alpha_2^{B_2} ] + m_3 \cdot [ \pi^{*A} + \pi^{*B} + \pi^{*C} + \pi^{*D} ]$$

$$= -2rm [ \beta(\alpha_1^{B_1} - \alpha_2^{B_2}) ] + m_3 \cdot [ \pi^{*A} + \pi^{*B} + \pi^{*C} + \pi^{*D} ]$$

$$\left. F_6^* = -2rm [ 4r \dot{u}_6 + 4r u_4 u_5 ] + m_3 \cdot [ \pi^{*A} + \pi^{*B} + \pi^{*C} + \pi^{*D} ] \right\}$$

$$F_7^* = 0 + 0 + 0 + 0 + m_3 \cdot \pi^{*C} - m_3 \cdot \pi^{*B_1} - m_3 \cdot \pi^{*B_2} \quad \text{since } \pi^{(A, B_1, C, B_2)} \neq f(u_7) \text{ and } \pi^{(A, B_1)} \neq f(u_7)$$

Since  $C, B_1, B_2$  have simple angular velocities in reference frame A let us rewrite

$$\pi^{*C} = \frac{d}{dt} \pi^{*C} = -\frac{d}{dt} \pi^{*C} - \omega^A \times \pi^{*C} \neq \pi^{*C} + \pi^{*C} \times \omega^A = \pi^{*C} + (\pi^{*C} \cdot \omega^A) \times \omega^A$$

$$m_3 \cdot \pi^{*C} = m_3 \cdot \pi^{*C} + m_3 \cdot (\pi^{*C} \cdot \omega^A) \times \omega^A ; m_3 \cdot (\pi^{*C} \cdot \omega^A) \times \omega^A - m_3 \cdot \pi^{*C} \times \omega^A = -J \dot{u}_7$$

$$= -J \ddot{u}_7 + m_3 \cdot (\pi^{*C} \cdot \omega^A) \times \omega^A + m_3 \cdot (\pi^{*C} \cdot \omega^A) \times \omega^A \quad \pi^{*C} = \pi^{B/B_1} = \pi^{B/B_2}$$

$$= -J \ddot{u}_7 + 0 + 0$$

$$\pi^{*B_1} = \pi^{B/B_1} + (\pi^{B/B_1} \cdot \omega^A) \times \omega^A = \pi^{B/B_1} + (\pi^{B/B_1} \cdot \omega^A) \times \omega^A + (\pi^{B/B_1} \cdot \omega^A) \times \omega^A - J \ddot{u}_7 m_3$$

$$m_3 \cdot \pi^{*B_1} = +J \ddot{u}_7 + 0 + 0$$

$$m_3 \cdot \pi^{*B_2} = J \ddot{u}_7 \quad \text{by same token.}$$

$$\therefore \boxed{F_7^* = -3J \ddot{u}_7}$$

$$\begin{aligned} \pi^{*C, R^A} &= [Jm_3 n_3 + K(m_1 n_1 + m_2 n_2)] - [u_4 m_1 + u_5 m_2] (u_6 + u_7) \\ &\quad + [K(u_6 + u_7) m_3 + K(u_4 m_1 + u_5 m_2)] / 2 (u_4 m_1 + u_5 m_2 + u_6 m_3 + u_7 m_4) \\ &= J(u_6 + u_7) u_4 m_2 - J(u_6 + u_7) u_5 m_1 \\ &\quad + K u_4 u_5 m_3 + u_4 u_5 m_3 - u_4 u_6 m_2 \end{aligned}$$

$$\begin{aligned} \pi^{*C} &= -\frac{d}{dt} \pi^{*C} = -\frac{d}{dt} (\pi^{*C} \cdot \omega^A) = -\frac{d}{dt} (\pi^{*C} \cdot \omega^A) - \frac{d}{dt} (\pi^{*C} \cdot \omega) \\ &\quad - \frac{d}{dt} (\pi^{*C} \cdot \omega) - \omega^A \times (\pi^{*C} \cdot \omega) \\ &\quad = (\pi^{*C} \cdot \omega^A) \times \omega^A - \pi^{*C} \cdot \omega^A \times \omega^A + \pi^{*C} \cdot \omega \times \omega^A \end{aligned}$$

$$m_3 \cdot \pi^{*C} = -Jm_3 \cdot \omega^A + m_3 \cdot \pi^{*C} = -J \dot{u}_6 - J \dot{u}_7 = -J(\dot{u}_6 + \dot{u}_7)$$

$$m_3 \cdot \pi^{*B_1} = -J(\dot{u}_6 - \dot{u}_7) = m_3 \cdot \pi^{*B_2}$$

$$m_3 \cdot \pi^{*B_1} = (-) \pi^{*B_1 \cdot R^A} \cdot \pi^{*B_1} = -J \dot{u}_6 + J \dot{u}_7$$

$$F_7^* = -J(\dot{u}_6 + \dot{u}_7) + 2J(\dot{u}_6 - \dot{u}_7) = -3J \dot{u}_7 + J \dot{u}_6$$

$$\boxed{F_7^* = -3J \dot{u}_7 + J \dot{u}_6}$$



$$V^A = V^C = u_1 m_1 + u_2 m_2 + u_3 m_3$$

$$\omega^A = u_4 m_1 + u_5 m_2 + u_6 m_3$$

$$V^{B_1} = V^C + \frac{d}{dt} 2r m_2 = V^C + 2r \omega^A m_1 = V^C + 2r (u_6 m_2 - u_5 m_3) ; V^{B_2} = V^C - 2r \left( \frac{u_6 m_2 + u_5 m_3}{\omega^A \times V^C} \right)$$

$$\alpha^{B_1} = \frac{dV^{B_1}}{dt} = \frac{dV^C}{dt} + \omega^A \times V^C = u_1 m_1 + u_2 m_2 + u_3 m_3 + 2r(u_6 m_2 - u_5 m_3) + \frac{\omega^A \times V^C}{\omega^A \times 2r (\omega^A \times m_1)} + 2r \left[ \frac{\omega^A \cdot m_1}{\omega^A} \right] \omega^A = \omega^{A^2} m_1$$

$$\alpha^{B_2} = u_1 m_1 + u_2 m_2 + u_3 m_3 - 2r(u_6 m_2 - u_5 m_3) + \omega^A \times V^C - 2r \omega^A \times (\omega^A \times m_1)$$

$$\alpha^A = \alpha^C = u_1 m_1 + u_2 m_2 + u_3 m_3 + \frac{\omega^A \times V^C}{\omega^A \times V^C} = [(u_5 u_3 - u_6 u_2) m_1 + (u_6 u_1 - u_4 u_3) m_2 + (u_1 u_2 - u_5 u_1) m_3]$$

$$\text{Now } -M[m_1 \alpha^A] + m/m_1 \cdot \alpha^C + m_1 \cdot \alpha^{B_1} + m_1 \cdot \alpha^{B_2}$$

$$-M u_1 - M (u_5 u_3 - u_6 u_2) - m[u_1 + (u_5 u_3 - u_6 u_2)] - m[u_1 + (u_5 u_3 - u_6 u_2) + 2r(u_4^2 - u_4 u_5^2 - u_6^2)]$$

$$-m[u_1 + (u_5 u_3 - u_6 u_2) - 2r(u_4^2 - u_4 u_5^2 - u_5^2 - u_6^2)]$$

$$-(M+3m)(u_1 + u_5 u_3 - u_6 u_2) - m[2r(-u_5^2 - u_6^2) - 2r(-u_5^2 - u_6^2)]$$

$$-M[m_2 \cdot \alpha^A] - m[m_2 \cdot \alpha^C + m_2 \cdot \alpha^{B_1} + m_2 \cdot \alpha^{B_2}]$$

$$= -M[u_2 + (u_6 u_1 - u_4 u_3)] - m[u_2 + (u_6 u_1 - u_4 u_3) + u_2 + 2ru_6 + (u_6 u_1 - u_4 u_3) + 2r u_4 u_5 + u_2 - 2ru_6 + (u_6 u_1 - u_4 u_3) - 2ru_4 u_5] = -(M+3m)[u_2 + (u_6 u_1 - u_4 u_3)]$$

$$-2r m_3 \omega^{B_1} + 2r m_3 \omega^{B_2} = -2r m_3 (-m \alpha^{B_1}) + 2r m_3 (-m \alpha^{B_2})$$

$$= 2rm[u_3 + 2ru_5 + (u_4 u_2 - u_5 u_1) + 2r u_4 u_6] - 2rm[u_3 + 2ru_5 + (u_4 u_2 - u_5 u_1) - 2r u_4 u_6] = 2rm[-4ru_5 + 4ru_4 u_6]$$

$$-2rm[m_2 \cdot \alpha^{B_1}] + 2rm[m_2 \cdot \alpha^{B_2}] = -2rm[u_2 + 2ru_6 + (u_6 u_1 - u_4 u_3) + 2ru_4 u_5] + 2rm[u_2 - 2ru_6 + (u_6 u_1 - u_4 u_3) - 2ru_4 u_5] = -2rm[4ru_6 + 4ru_4 u_5]$$

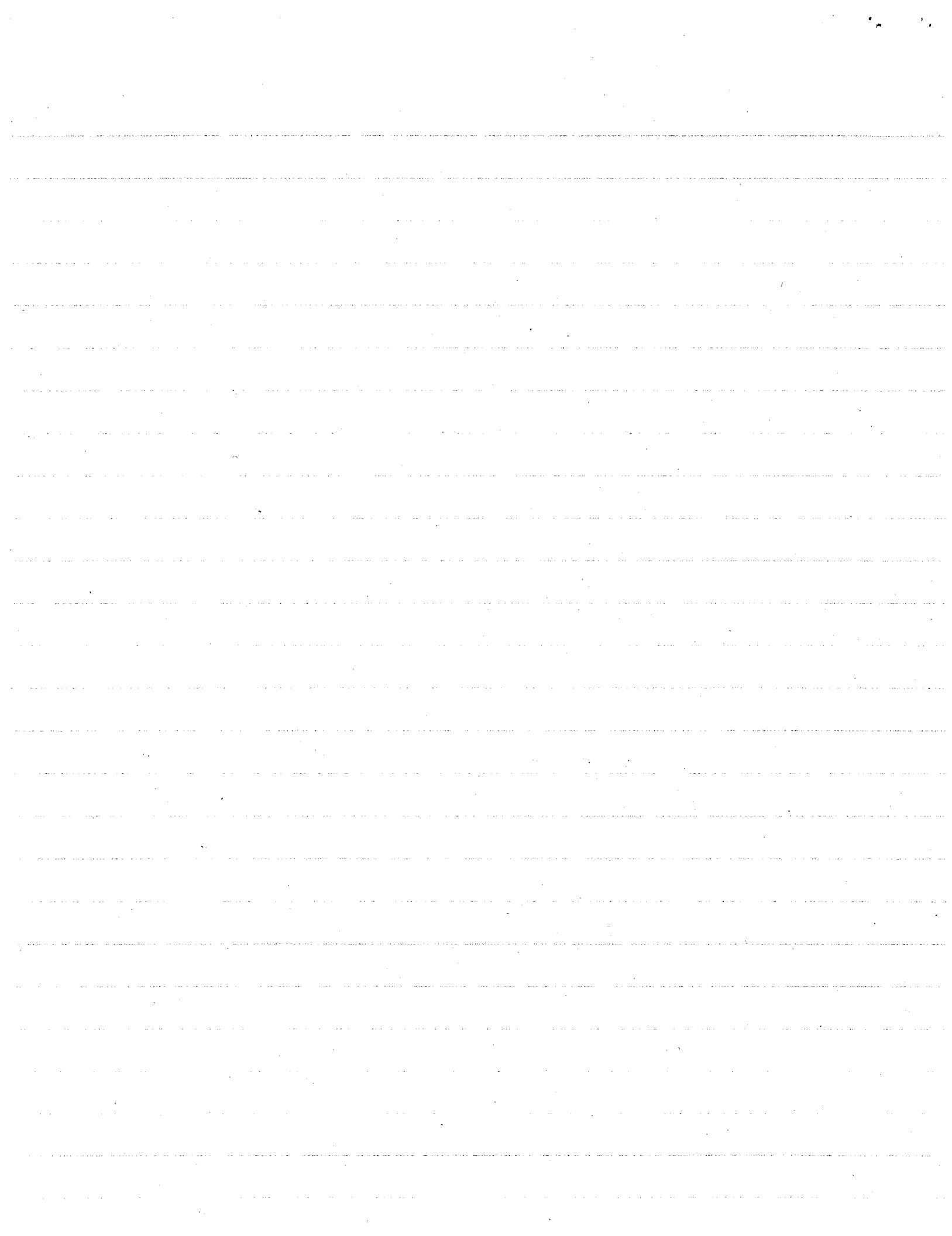
Now  $K_1, K_2, K_3$  are the moments of inertia of entire system about  $X_1, X_2, X_3$

$$\begin{aligned} \Pi^A + \Pi^{B_1} + \Pi^C + \Pi^{B_2} &= -\frac{d}{dt} \left[ I^A A^A + I^C C^A + I^{B_1} B_1^A + I^{B_2} B_2^A \right] \\ &= -\frac{d}{dt} \left[ I^A A^A \cdot \omega^A + I^C C^A \cdot \omega^C + I^{B_1} B_1^A \cdot \omega^{B_1} + I^{B_2} B_2^A \cdot \omega^{B_2} \right] \\ &\quad \left[ (I^A A^A + I^C C^A + I^{B_1} B_1^A + I^{B_2} B_2^A) \cdot \omega^A + I^C C^A \cdot \omega^C + I^{B_1} B_1^A \cdot \omega^{B_1} + I^{B_2} B_2^A \cdot \omega^{B_2} \right] \end{aligned}$$

$$\text{Now } \bar{I}^{B/A} = \bar{I}^{B/B_1} + \bar{I}^{B/A^2}$$

$$\text{Now } \bar{\omega}^C = \bar{\omega}^{B_1} = -\bar{\omega}^{B_2} \text{ and } \bar{I}^{C/A} = \bar{I}^{B_1/A} = \bar{I}^{B_2/A}$$

$$\therefore (\Pi^{A^A} + \Pi^{C/A} + \bar{I}^{B_1/A} + \bar{I}^{B_2/A} + \bar{I}^{B_2/A^2}) = \Pi^{(A+C+B_1+B_2)/A^2} - \bar{I}^{B_1/A^2} - \bar{I}^{B_2/A^2}$$



$$\pi^A + \pi^B + \pi^C + \pi^D = -\frac{d}{dt} \left\{ \begin{array}{l} \pi^{D/D^*} \cdot \omega^A - (\pi^{B/A^*} + \pi^{B_2/A^*}) \cdot \omega^A - \pi^{C/C^*} \cdot \omega^A \\ \pi^D - \pi^C + \frac{d}{dt} (Br^2 m [m_2 m_2 + m_3 m_3] \cdot \omega^A) + \frac{R^A}{\omega} \times \pi^C \end{array} \right\}$$

only moment of inertia  $m_2 m_2$   
 $m_3 m_3$

$$\begin{aligned} \pi^{B/A^*} &= m(\pi^2 U - \pi \dot{P}) & \pi^B = 2r m_1 & \pi^2 = 4r^2 \\ \pi^{B_2/A^*} &= m(4r^2 U - 4r^2 m_1 m_1) = m \cdot 4r^2 (m_2 m_2 + m_3 m_3) \\ \pi^{B_2/A^*} &= m(4r^2 U - 4r^2 m_1 m_2) ; \quad \pi^B = -2r m_1 & \pi^2 = 4r^2 \\ &= 4r^2 m (m_2 m_2 + m_3 m_3) & \pi^D = \pi^C + Br^2 m \left\{ (u_5 m_2 + u_6 m_3) + \frac{R^A}{\omega} \times \mathbf{w} \right\} \\ &= \pi^D - \pi^C + \frac{R^2 m}{\omega} \frac{d}{dt} [u_5 m_2 + u_6 m_3] & + \frac{R^A}{\omega} \times \mathbf{w} \\ &= - (u_4 m_1 + u_5 m_2 + u_6 m_3) \times u_4 m_1 \\ \text{now } \mathbf{w} &= \frac{R^A}{\omega} - u_4 m_1 & \frac{R^A}{\omega} \times \mathbf{w} = (u_5 \times \frac{R^A}{\omega} - \frac{R^A}{\omega} \times u_4 m_1) = - (u_4 m_1 + u_5 m_2 + u_6 m_3) \times u_4 m_1 \\ &= + u_5 u_4 m_3 - u_6 u_4 m_2 & = + u_5 u_4 m_3 - u_6 u_4 m_2 \\ &= + u_5 u_4 m_3 - u_6 u_4 m_2 & = + u_5 u_4 m_3 - u_6 u_4 m_2 \\ \therefore \pi^A + \pi^B + \pi^C + \pi^D &= \pi^D - \pi^C + Br^2 m [(u_5 + u_6 u_4) m_2 + (u_6 + u_5 u_4) m_3] + \frac{R^A}{\omega} \times \pi^C \end{aligned}$$

$$\begin{aligned} F_4^* &= m_1 \cdot [\pi^D - m_1 \cdot \pi^C + m_1 \cdot \frac{R^A}{\omega} \times \pi^C] & \alpha^C = u_7 m_3 \\ F_5^* &= -8r^2 m [u_5 + u_4 u_6] + m_2 \cdot \pi^D - m_2 \cdot \pi^C + 8r^2 m [u_5 - u_6 u_4] & + m_2 \cdot \frac{R^A}{\omega} \times \pi^C \\ F_6^* &= 8r^2 m [u_6 + u_4 u_5] + m_3 \cdot \pi^D - m_3 \cdot \pi^C + 8r^2 m [u_6 - u_5 u_4] & = m_2 \cdot (\pi^D - \pi^C) + m_2 \cdot \frac{R^A}{\omega} \times \pi^C \\ \text{now } \pi^D &= (\pi^{D/D^*}, \omega^A) \times \omega^A = \pi^{D/D^*}, \frac{R^A}{\omega} & + m_3 \cdot (\pi^D - \pi^C) + m_3 \cdot \frac{R^A}{\omega} \times \pi^C \\ \frac{R^A}{\omega} &= \frac{d \omega^A}{dt} = \frac{d}{dt} \frac{R^A}{\omega} + \frac{R^A}{\omega} \times \frac{R^A}{\omega} = u_4 m_1 + u_5 m_2 + u_6 m_3 & \pi^{D/D^*} = K_1 m_1 m_1 + K_2 m_2 m_2 + K_3 m_3 m_3 \\ \pi^{D/D^*} \cdot \frac{R^A}{\omega} &= K_1 u_4 m_1 + K_2 u_5 m_2 + K_3 u_6 m_3 & \pi^{D/D^*}, \frac{R^A}{\omega} = K_1 u_4 m_1 + K_2 u_5 m_2 + K_3 m_3 \\ \text{Now } \frac{R^A}{\omega} &= (\pi^{D/D^*}, \omega^A) \times \omega^A = [(K_2 - K_3) u_5 u_6 - K_1 u_4] m_1 + ((K_3 - K_1) u_4 u_6 - K_2 u_5) m_2 + \\ &\quad ((K_1 - K_2) u_4 u_5 - K_3 u_6) m_3 & \frac{R^A}{\omega} \times \pi^C = \frac{R^A}{\omega} \times [J u_7 m_3] = J u_7 (u_5 m_1 - \\ &\quad u_7 m_3) & \end{aligned}$$

$$F_4^* = (K_2 - K_3) u_5 u_6 - K_1 u_4 + J u_7 u_5$$

$$F_5^* = (K_3 - K_1) u_4 u_6 - K_2 u_5 - J u_7 u_4$$

$$F_6^* = (K_1 - K_2) u_4 u_5 - K_3 u_6 + J u_7$$

$$F_7^* = -3 J u_7 + J u_6$$

$$= 3 J u_7 - J u_6 = T$$

$$u_7 + \frac{u_6}{3} = \frac{T}{3J}$$

$$u_7 = \frac{T}{3J} - \frac{u_6}{3}$$



$$\dot{u}_4 = \frac{T}{K_1} u_7 u_5 + \frac{K_2 - K_3}{K_1} u_5 u_6 \Rightarrow \frac{T}{K_1} \omega_2 + \frac{K_2 - K_3}{K_1} \omega_2 \omega_3 = \dot{\omega}_1$$

$$\dot{u}_5 = -\frac{T}{K_2} u_7 u_4 + \frac{K_3 - K_1}{K_2} u_4 u_6 \Rightarrow -\frac{T}{K_2} \omega_1 + \frac{K_3 - K_1}{K_2} \omega_1 \omega_3 = \dot{\omega}_2$$

$$\dot{u}_6 = \frac{T u_7}{K_3} + \frac{K_1 - K_2}{K_3} u_4 u_5 \Rightarrow \frac{T u_6}{3K_3} + \frac{T_3}{3K_3} + \frac{K_1 - K_2}{K_3} u_4 u_5 \Rightarrow \dot{u}_6 = \left[ \frac{T_3}{3K_3} + \frac{K_1 - K_2}{K_3} u_4 u_5 \right] / \left( 1 - \frac{T}{3K_3} \right)$$

$$\dot{u}_3 = \left[ \frac{T_3}{3K_3} + \frac{K_1 - K_2}{K_3} \omega_1 \omega_2 \right] / \left( 1 - \frac{T}{3K_3} \right)$$

$$\dot{u}_7 = \frac{u_6}{3} + \frac{T_3}{3T} = \frac{T u_7}{3K_3} + \frac{T_3}{3T} + \frac{K_1 - K_2}{3K_3} u_4 u_5 \Rightarrow \dot{u}_7 = -\frac{T}{K_3} + \frac{T_3}{3K_3} + \frac{K_1 - K_2}{3K_3} u_4 u_5$$

$$\dot{u}_2 = \left( 1 - \frac{T}{3K_3} \right) \dot{u}_2$$

$$\dot{u}_7 = \left[ \frac{T_3}{3T} + \frac{K_1 - K_2}{3K_3} + \frac{K_1 - K_2}{3K_3} u_4 u_5 \right] / \left( 1 - \frac{T}{3K_3} \right)$$

$$\dot{u}_2 = \left[ \frac{T_3}{3T} + \frac{K_1 - K_2}{3K_3} + \frac{K_1 - K_2}{3K_3} \omega_1 \omega_2 \right] / \left( 1 - \frac{T}{3K_3} \right)$$

$$\dot{u}_6 = \left[ \frac{T_3}{3K_3} + \frac{K_1 - K_2}{3K_3} T u_4 u_5 \right] / \left( 1 - \frac{T}{3K_3} \right) + \frac{K_1 - K_2}{K_3} u_4 u_5 \cdot \left( 1 - \frac{T}{3K_3} \right) / \left( 1 - \frac{T}{3K_3} \right)$$

$$\dot{u}_6 = \left[ \frac{T_3}{3K_3} + \frac{K_1 - K_2}{K_3} u_4 u_5 \right] / \left( 1 - \frac{T}{3K_3} \right)$$

$$\omega_2 \omega_3 (T_2, T_3) = \dot{\omega}^B \Rightarrow T_2 \dot{\omega}_1$$

$$\omega_3 \omega_1 (T_3, T_1) \Leftrightarrow \dot{\omega}^B \omega_3 = T_3 \dot{\omega}_2$$

$$\omega_1 \omega_2 (T_1, T_2) + \dot{\omega}^B \omega_1 = T_2 \dot{\omega}_3$$

$$(T_2 - \frac{F_1}{T_1}) / (1 - \frac{T}{T_1}) = \dot{\omega}^B = (T_2 - \frac{\omega_1 \omega_3 (T_2, T_3)}{T_1}) / (1 - \frac{T}{T_1})$$

$$F_1 = \omega_2 \omega_3 (T_2, T_3)$$

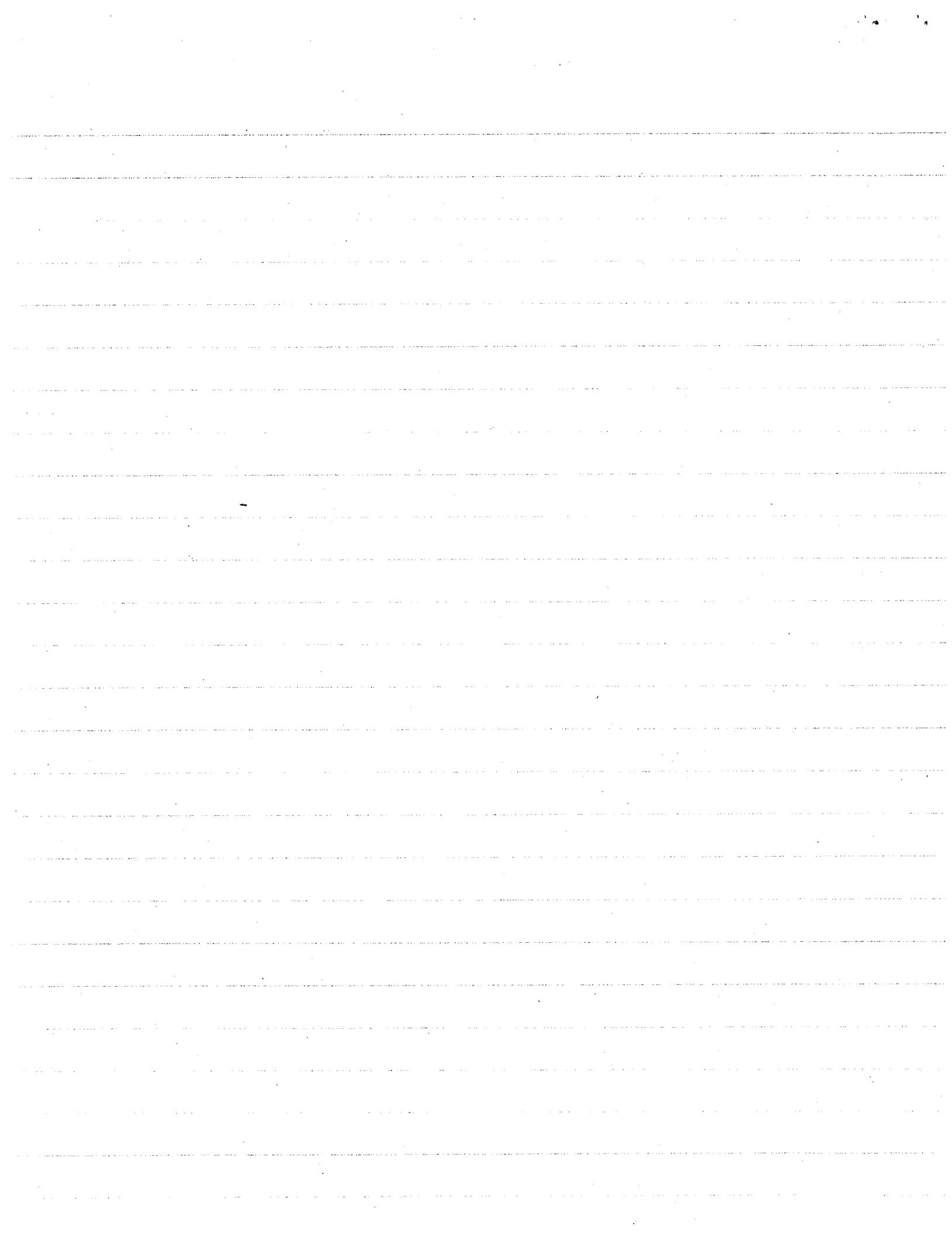
$$+ \omega_1 \omega_3 (T_2, T_3) + \left( -\frac{T}{T_1} + \frac{\omega_2 \omega_3 (T_2, T_3) T}{T_1} \right) / (1 - \frac{T}{T_1}) = -\dot{\omega}^B + \omega_2 \omega_3 (T_2, T_3)$$

$$= T_1 \dot{\omega}_1 + \omega_2 \omega_3 (T_2, T_3) - \omega_2 \omega_3 (T_2, T_3) \frac{T}{T_1} = T + \omega_2 \omega_3 (T_2, T_3) \frac{T}{T_1}$$

$$1 - \frac{T}{T_1}$$

$$\dot{\omega}_1 = \omega_2 \omega_3 (T_2, T_3) - T_1$$

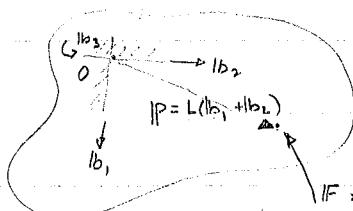
$$1 - \frac{T}{T_1}$$



From problem 2a we obtain the following if  $\overset{A}{\omega}^B = \overset{A}{\omega}^{I_1} + \overset{A}{\omega}^{I_2} + \overset{A}{\omega}^{I_3} = \dot{q}_1 \overset{A}{l} l_{11} + \dot{q}_2 \overset{A}{l} l_{22} + \dot{q}_3 \overset{A}{l} l_{33}$

where initially  $a_{ij} + lb_i$  correspond before the successive rotations

$$\begin{array}{c|ccc} & a_{11} & a_{12} & a_{13} \\ \hline l_{11} & 1 & 0 & 0 \\ l_{12} & 0 & c_1 & s_1 \\ l_{13} & 0 & -s_1 & c_1 \end{array} \quad \begin{array}{c|ccc} & l_{11} & l_{12} & l_{13} \\ \hline l_{21} & c_2 & 0 & -s_2 \\ l_{22} & 0 & 1 & 0 \\ l_{23} & s_2 & 0 & c_2 \end{array} \quad \begin{array}{c|ccc} & l_{11} & l_{12} & l_{13} \\ \hline l_{31} & c_3 & s_3 & 0 \\ l_{32} & -s_3 & c_3 & 0 \\ l_{33} & 0 & 0 & 1 \end{array}$$

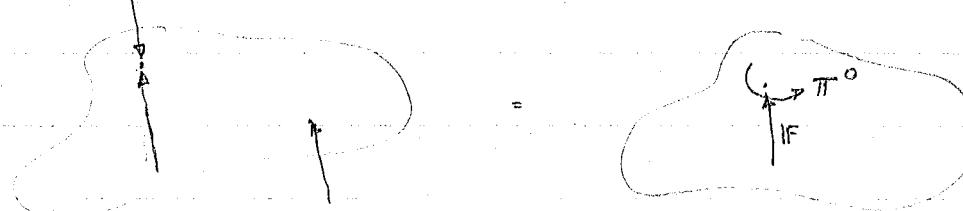


find a potential function for  $\text{IF}$ . To find a potential fn.

for  $\text{IF}$  we must first find  $\text{IF}'$ 's contribution to the generalized active forces

Since  $O$  is fixed let us find an equivalent Moment & force syste. about point  $O$

$\therefore$  at  $O$  there will be a force  $\text{IF}$  and a couple  $= \overline{\Pi} = \text{p} \times \text{IF}$



$$\begin{aligned} \overline{\Pi} &= L(lb_1 + lb_2) \times F_1 lb_1 + F_2 lb_2 + F_3 lb_3 \\ &= L(F_2 lb_3 - F_3 lb_2 - F_1 lb_3 + F_3 lb_1) \\ &= L[(F_2 - F_1)lb_3 + F_3 lb_1 - F_3 lb_2] \end{aligned}$$

$$\overline{\Pi} = KL^2 [c_2(s_3 + c_3)lb_3 + s_2lb_1 + s_2lb_2]$$

Now  $F_r'$  (being the contribution of  $\text{IF}'$  to the generalized active forces) =  $N \dot{q}_r \cdot \text{IF}' + \overset{A}{\omega} \dot{q}_r \cdot \overline{\Pi}^O$

$$\text{but since } O \text{ is fixed } N^O = 0 \quad \therefore F_r' = \overset{A}{\omega} \dot{q}_r \cdot \overline{\Pi}^O \quad \text{and} \quad F_r' = -\frac{\partial P}{\partial \dot{q}_r}$$

$$\overset{A}{\omega} \dot{q}_1 = a_{11}, \quad \overset{A}{\omega} \dot{q}_2 = l_{12}, \quad \overset{A}{\omega} \dot{q}_3 = l_{13}$$

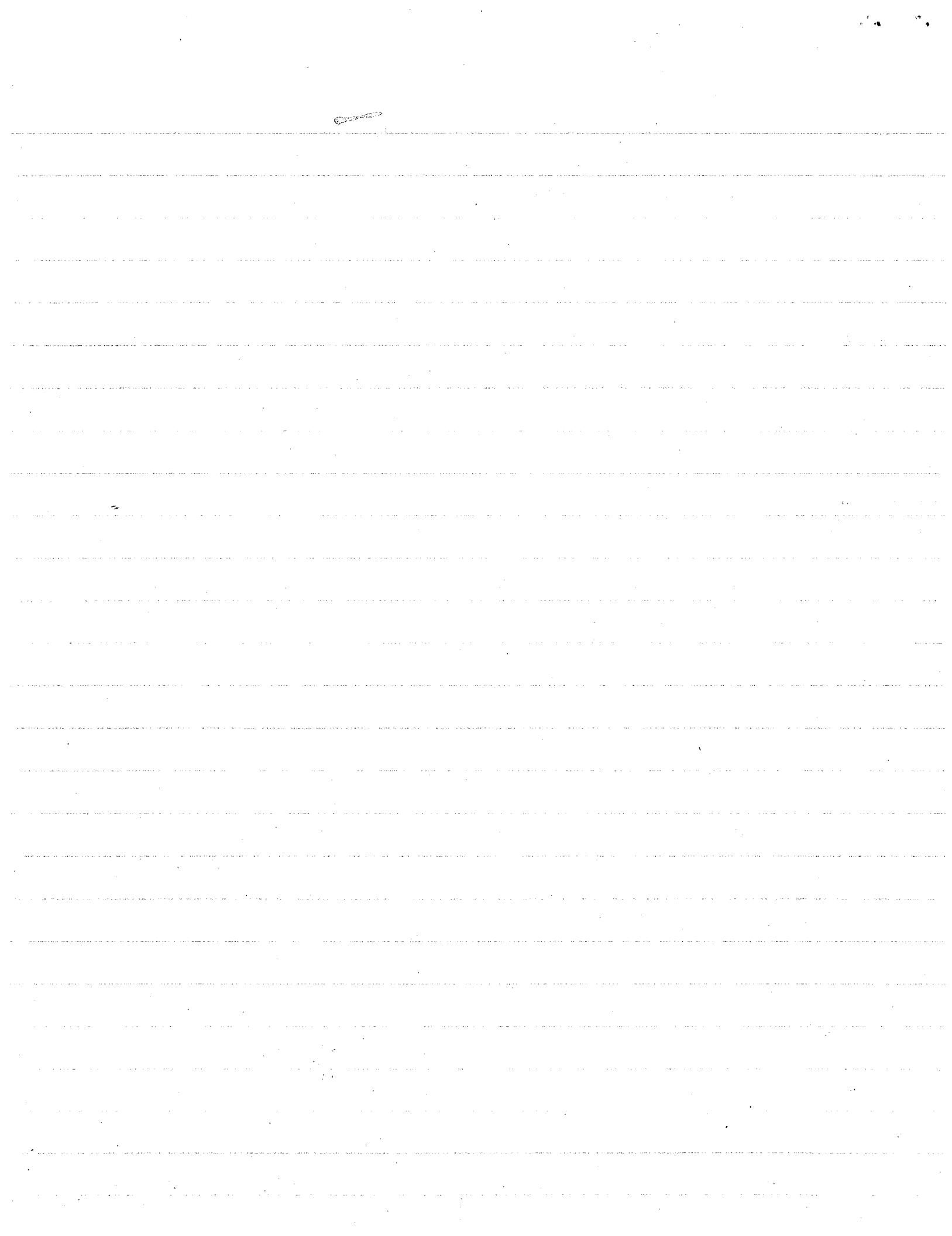
$$\begin{aligned} \therefore F_1' &= a_{11} \cdot \overline{\Pi}^O = l_{11} \cdot \overline{\Pi}^O = (c_2 l_{11} + s_2 l_{22}) \cdot \overline{\Pi}^O = (c_2 [c_3 l_{11} - s_3 l_{22}] + s_2 l_{33}) \cdot \overline{\Pi}^O \\ &= KL^2 [(c_2 c_3) l_{11} - c_2 s_3 l_{22} + s_2 l_{33}] \cdot [-s_2 l_{11} + s_2 l_{22} + c_2 (s_3 + c_3)] \\ &= KL^2 [-c_2 c_3 s_2 - c_2 s_3 s_2 + s_2 c_3 s_3 + s_2 c_3 c_3] = 0 \quad = -\frac{\partial P}{\partial \dot{q}_1} \quad (\alpha) \end{aligned}$$

$$\begin{aligned} F_2' &= l_{12} \cdot \overline{\Pi}^O = l_{12} \cdot \overline{\Pi}^O = (s_3 l_{11} + c_3 l_{22}) \cdot KL^2 [-s_2 l_{11} + s_2 l_{22} + c_2 (s_3 + c_3) l_{33}] \\ &= KL^2 [-s_3 s_2 + c_3 s_2] = -\frac{\partial P}{\partial \dot{q}_2} \quad (\beta) \end{aligned}$$

$$F_3' = l_{13} \cdot \overline{\Pi}^O = l_{13} \cdot KL^2 [-s_2 l_{11} + s_2 l_{22} + c_2 (s_3 + c_3) l_{33}] = KL^2 [c_2 s_3 + c_3 c_3] = -\frac{\partial P}{\partial \dot{q}_3} \quad (\gamma)$$

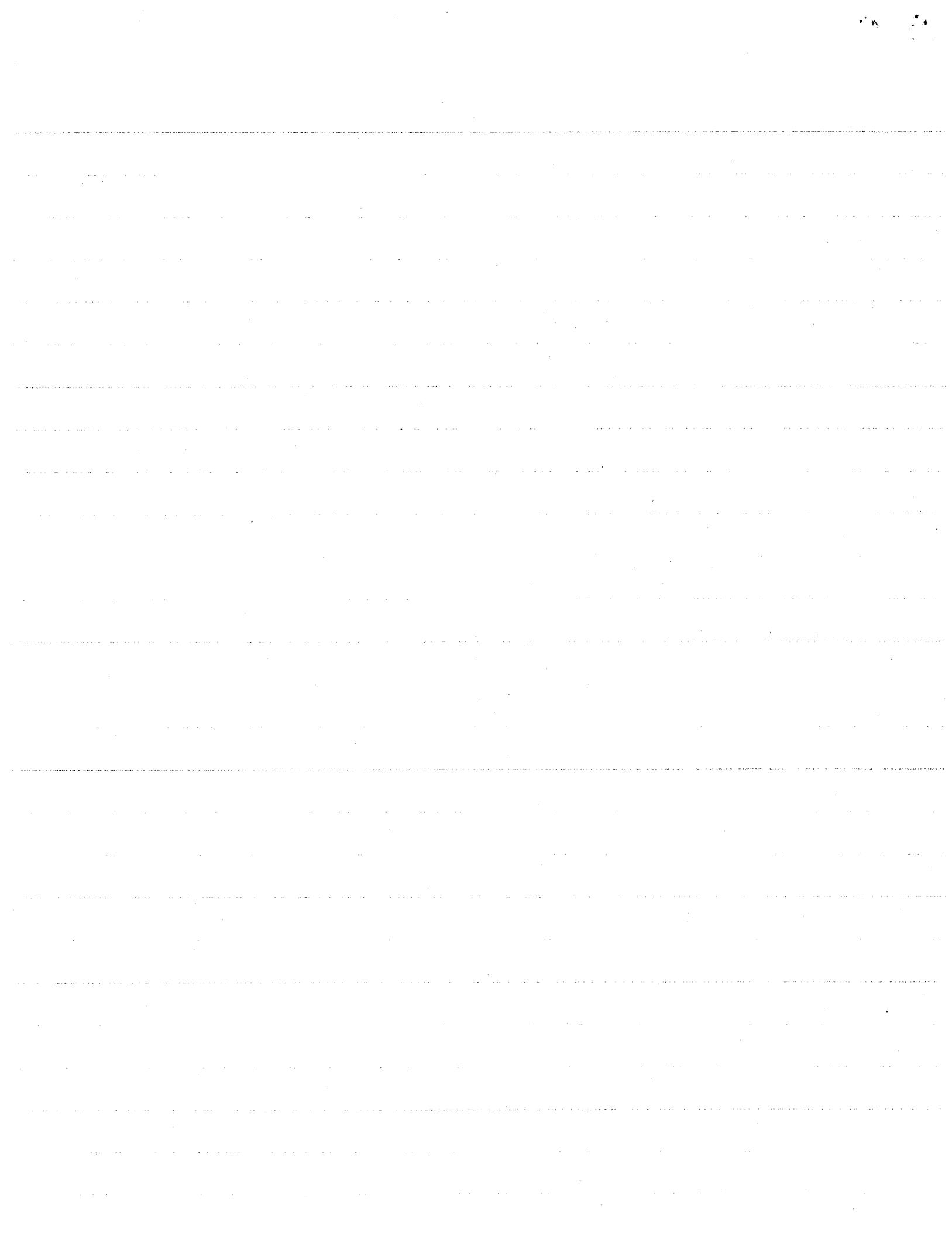
$\therefore$  integrating (a)  $P = f(\dot{q}_1, \dot{q}_3, t)$

$$(b) -P = KL^2 (s_3 c_2 + c_3 c_2) * f(\dot{q}_2, t) \Rightarrow P = KL^2 (c_3 - s_3) c_2 + f(\dot{q}_2, t)$$



$$-\frac{\partial P}{\partial q_3} = KL^2(c_3c_2 + s_3c_2) - \frac{\partial f}{\partial q_3} = KL^2(c_3c_2 + c_2s_3) \Rightarrow \frac{\partial f}{\partial q_3} = 0 \text{ and } f = f(t) \text{ only}$$

$$\therefore P = KL^2c_2(c_3 - s_3) + f(t) \quad \text{for } \mathbb{R}$$



# Excellent

A

#23  
LEVY

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A rigid body A carries three identical rotors  $B_1$ ,  $B_2$ , and C on parallel axes fixed in A, as shown in the attached sketch. By means of a motor attached to A, C is made to rotate relative to A at a rate  $\Omega$  (not necessarily constant), and  $B_1$  and  $B_2$  are driven by C at the same rate, but with opposite sense.

$x_1$ ,  $x_2$ ,  $x_3$  (see sketch) are principal axes of inertia of A for the mass center of A, and the moments of inertia of the entire system about these axes have the values  $K_1$ ,  $K_2$ ,  $K_3$ , respectively. The mass center of each of the three rotors lies on  $x_1$ , and each rotor has moment of inertia  $J$  about its axis of symmetry.

Let  $n_i$  be a unit vector pointing in the direction of the positive  $x_i$ -axis ( $i = 1, 2, 3$ ); replace the system of forces exerted on C by A (by means of the motor) with a couple of torque

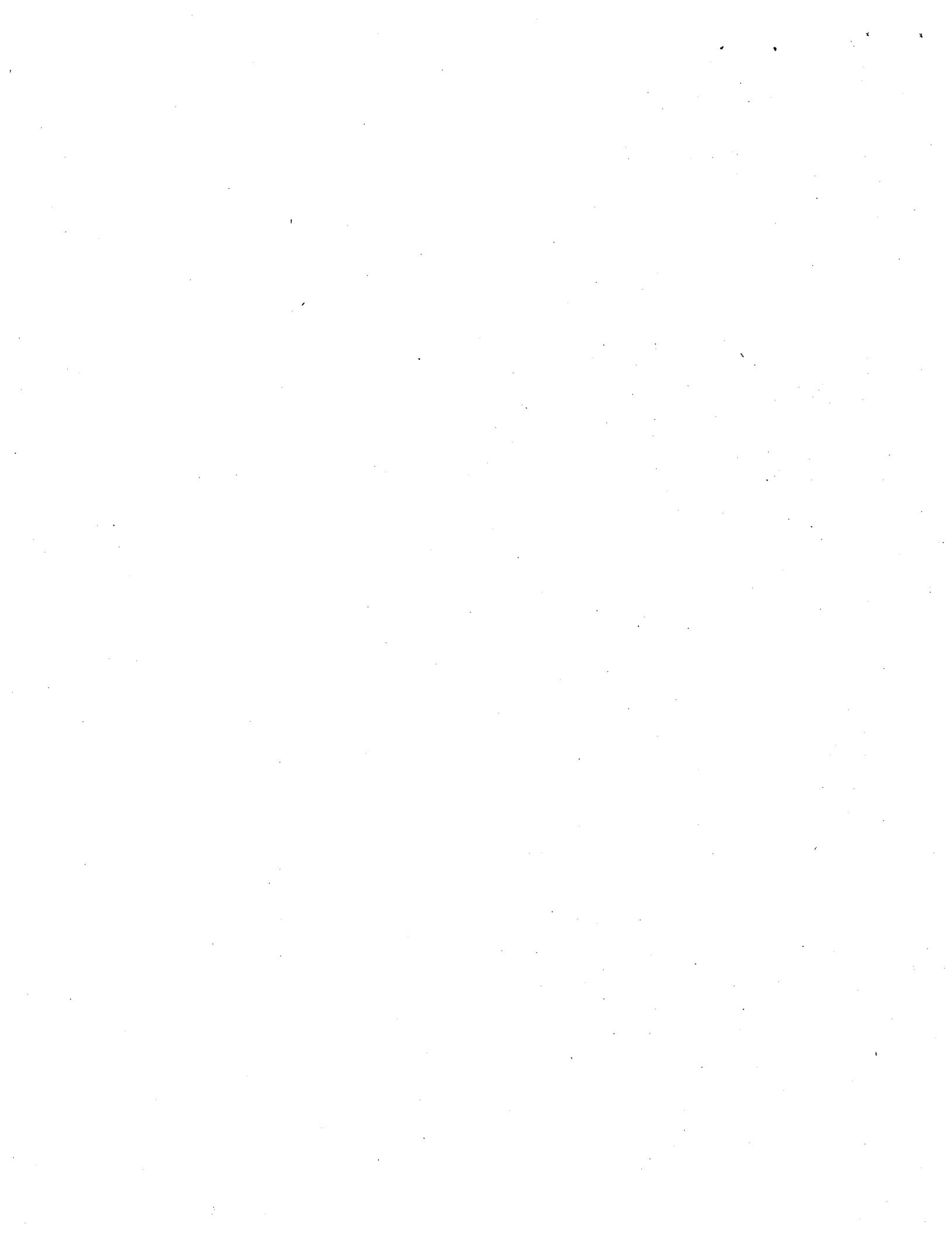
$$T_1 n_1 + T_2 n_2 + T_3 n_3$$

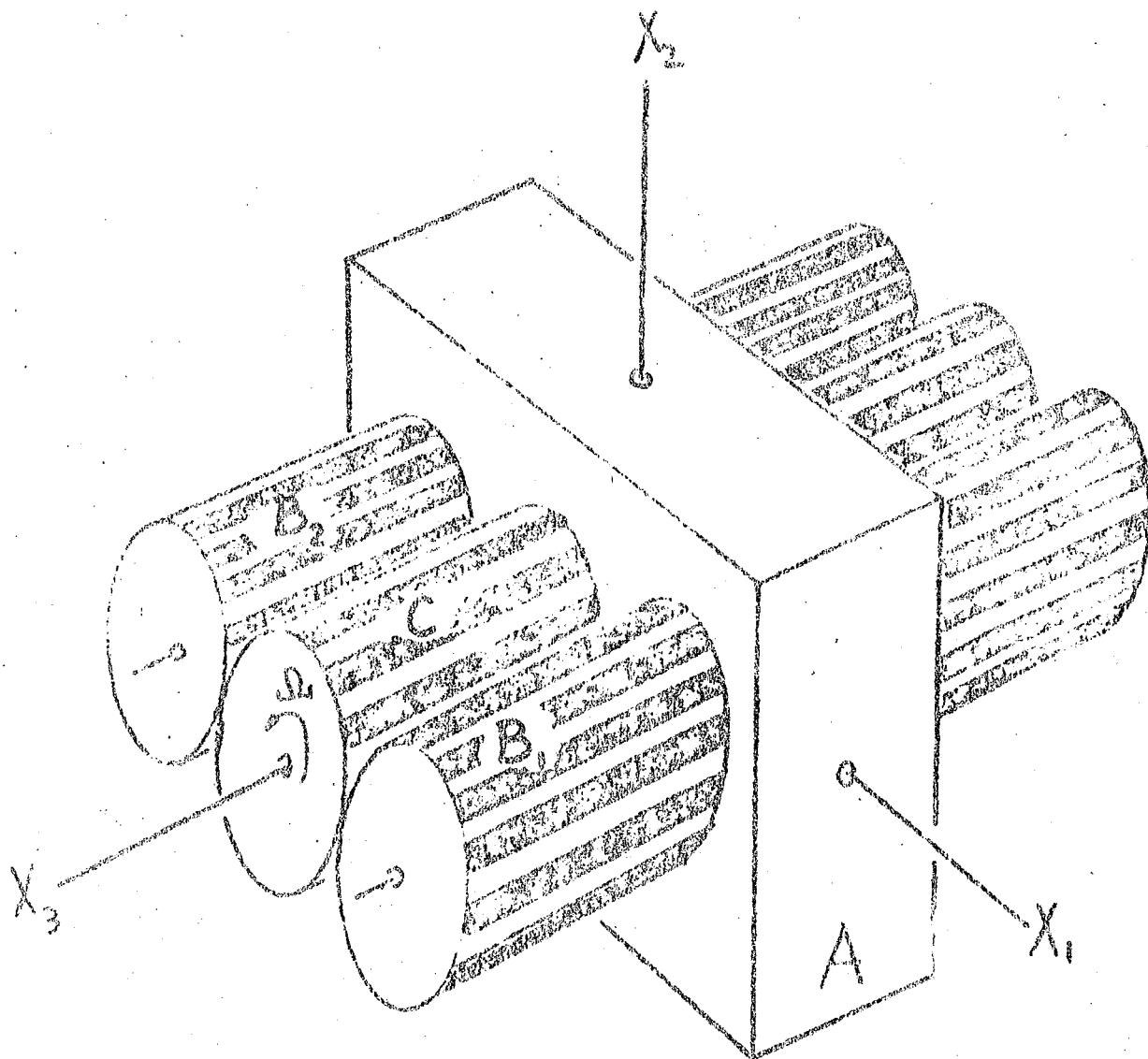
together with a force applied at the mass center of C; and assume that the only body and contact forces acting on the system are gravitational forces.

The differential equations of motion of this system can be expressed as

$$\ddot{\Omega} = F; \quad \ddot{a}_i = G_i \quad (i = 1, 2, 3)$$

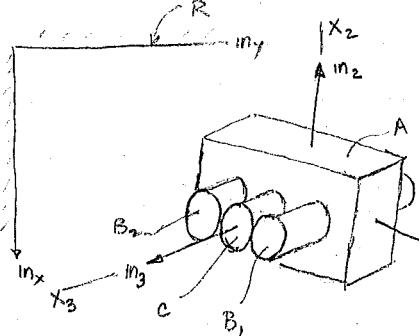
where  $F$  and  $G_i$  are functions of  $\Omega$ ,  $\omega_1$ ,  $\omega_2$ ,  $\omega_3$ ,  $J$ ,  $K_1$ ,  $K_2$ ,  $K_3$ ,  $T_1$ ,  $T_2$ ,  $T_3$ . Determine  $F$  and  $G_i$  ( $i=1,2,3$ ).







We assume a reference frame  $R$  (with mutually perpendicular vectors  $m_x, m_y, m_z$ ) which is fixed and in which the system in question moves. The system in question has 7 degrees of freedom, 6 to describe the motion of body  $A$  and 1 to describe the motion of body  $C$  relative to body  $A$  (similarly this same degree of freedom describes the motion of bodies  $B_1$  and  $B_2$  relative to  $A$ ). Since a motor causes a



set of forces to act on body  $C$  and we replace them by a torque  $\Pi^C$  and a resultant force  $\text{IF}_T^C$  acting at  $C^*$ . Similarly by the law of action and reaction a torque  $-\Pi^C$  and a resultant force  $-\text{IF}_T^C$  acting at  $A^*$  exists. Since the only body and contact forces involved are  $Mg m_x$  on body  $A$  and

$mg m_x$  on each of the bodies  $C, B_1, B_2$ . Because the translational motion can be decoupled from the rotational motion, we need only look at the rotational equations to obtain our results. However I will set up the general results.

If we assume  $(\cdot)^*$  represent the values of the center of mass of the body and

$$W^{A^*} = V_1 m_1 + V_2 m_2 + V_3 m_3 \quad (1)$$

then we can define the following generalized coordinates

$$\begin{aligned} u_i = & \begin{cases} V_i & (i=1,2,3) \\ \omega_{i-3} & (i=4,5,6) \\ \Omega & i=7 \end{cases} \end{aligned}$$

where

$${}^R \omega^A \cdot m_i = \omega_i \quad (2)$$

$${}^A \omega^C = - {}^A \omega^{B_1} = - {}^A \omega^{B_2} = \Omega m_3 \quad (3)$$

Also

$$\text{IF}^A = - \text{IF}_T^C + Mg m_x \quad (4)$$

$$\text{IF}_T^C = mg m_x + \text{IF}_T^C \quad (5)$$

$$\text{IF}^{B_1} = mg m_x \quad (6)$$

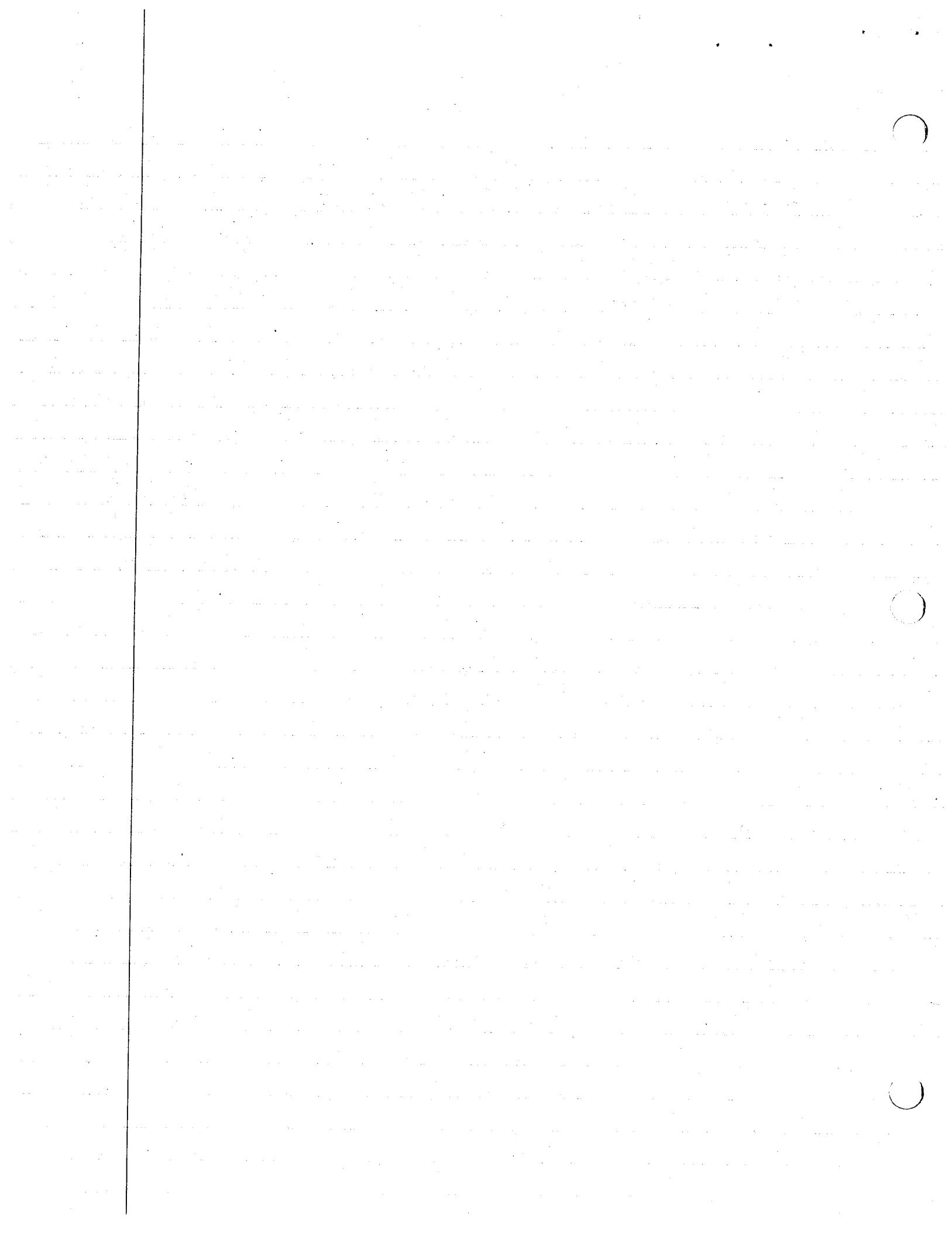
$$\text{IF}^{B_2} = mg m_x \quad (7)$$

Similarly

$$\Pi^C = T_1 m_1 + T_2 m_2 + T_3 m_3 \quad (8)$$

$$\Pi^A = - \Pi^C \quad (9)$$

$$\Pi^{B_1} = \Pi^{B_2} = 0 \quad (10)$$



Now we can define

$$W^{A^*} = W^{C^*} = u_1 m_1 + u_2 m_2 + u_3 m_3 \quad (11)$$

$$W^{B_1} = W^{C^*} + l \cdot \overset{R}{\omega}^A \times m_1 = W^{C^*} + l [u_6 m_2 - u_5 m_3] \quad (12)$$

$$W^{B_2} = W^{C^*} - l \cdot \overset{R}{\omega}^A \times m_1 = W^{C^*} - l [-u_6 m_2 + u_5 m_3] \quad (13)$$

Where  $l$  is the magnitude of the distance between  $A^*$  and  $B_1^*$ , and between  $A^*$  and  $B_2^*$ . We now define the generalized active force  $F_r$  by

$$F_r = W_{u_r}^{A^*} \cdot IF^{A^*} + W_{u_r}^{C^*} \cdot IF^C + W_{u_r}^{B_1^*} \cdot IF^{B_1^*} + W_{u_r}^{B_2^*} \cdot IF^{B_2^*} + \overset{R}{\omega}_{u_r} \cdot \Pi^A + \overset{R}{\omega}_{u_r}^C \cdot \Pi^C + \overset{R}{\omega}_{u_r}^{B_1} \cdot \Pi^{B_1} + \overset{R}{\omega}_{u_r} \cdot \Pi^{B_2} \quad \text{for } r=1, \dots, 7 \quad (14)$$

We note certain items: all the  $\omega$ 's are not functions of  $u_1, u_2, u_3$ ; all the  $W$ 's are not functions of  $u_4$  and  $u_7$ , and  $W^{A^*}, W^{C^*}$  are not functions of  $u_5$  and  $u_6$ . Using these and (1-4) we obtain

$$F_1 = m_1 \cdot m_x [(M+3m)g] \quad (15)$$

$$F_2 = m_2 \cdot m_x [(M+3m)g] \quad (16)$$

$$F_3 = m_3 \cdot m_x [(M+3m)g] \quad (17)$$

$$F_4 = F_5 = F_6 = 0 \quad (18-20)$$

$$F_7 = T_3 \quad (21)$$

To obtain the generalized inertia forces  $F_r^*$

$$F_r^* = W_{u_r}^{A^*} \cdot IF^{A^*} + W_{u_r}^{C^*} \cdot IF^C + W_{u_r}^{B_1^*} \cdot IF^{B_1^*} + W_{u_r}^{B_2^*} \cdot IF^{B_2^*} + \overset{R}{\omega}_{u_r} \cdot \Pi^{A^*} + \overset{R}{\omega}_{u_r}^C \cdot \Pi^C + \overset{R}{\omega}_{u_r}^{B_1} \cdot \Pi^{B_1} + \overset{R}{\omega}_{u_r} \cdot \Pi^{B_2} \quad (22)$$

$$\text{where } IF^{A^*} = -M \alpha^{A^*} \quad \Pi^{A^*} = -\frac{d}{dt} \overset{R}{H}^{A/A^*} \quad (23-30)$$

$$IF^{C^*} = -M \alpha^{C^*} \quad \Pi^{C^*} = -\frac{d}{dt} \overset{R}{H}^{C/C^*}$$

$$IF^{B_1^*} = M \alpha^{B_1^*} \quad \Pi^{B_1^*} = -\frac{d}{dt} \overset{R}{H}^{B_1/B_1^*}$$

$$IF^{B_2^*} = -M \alpha^{B_2^*} \quad \Pi^{B_2^*} = -\frac{d}{dt} \overset{R}{H}^{B_2/B_2^*}$$

Now

$$\alpha^{A^*} = \frac{d}{dt} W^{C^*} + \overset{R}{\omega}^A \times W^C = \alpha^{C^*} \quad (31, 32)$$

$$\alpha^{B_1^*} = \alpha^{C^*} + l \frac{d}{dt} \overset{R}{\omega}^A \times m_1 + l [(\overset{R}{\omega}^A \cdot m_1) \overset{R}{\omega}^A - \overset{R}{\omega}^{A^2} m_1] \quad (33)$$

$$\alpha^{B_2^*} = \alpha^{C^*} - l \frac{d}{dt} \overset{R}{\omega}^A \times m_1 - l [(\overset{R}{\omega}^A \cdot m_1) \overset{R}{\omega}^A - \overset{R}{\omega}^{A^2} m_1] \quad (34)$$

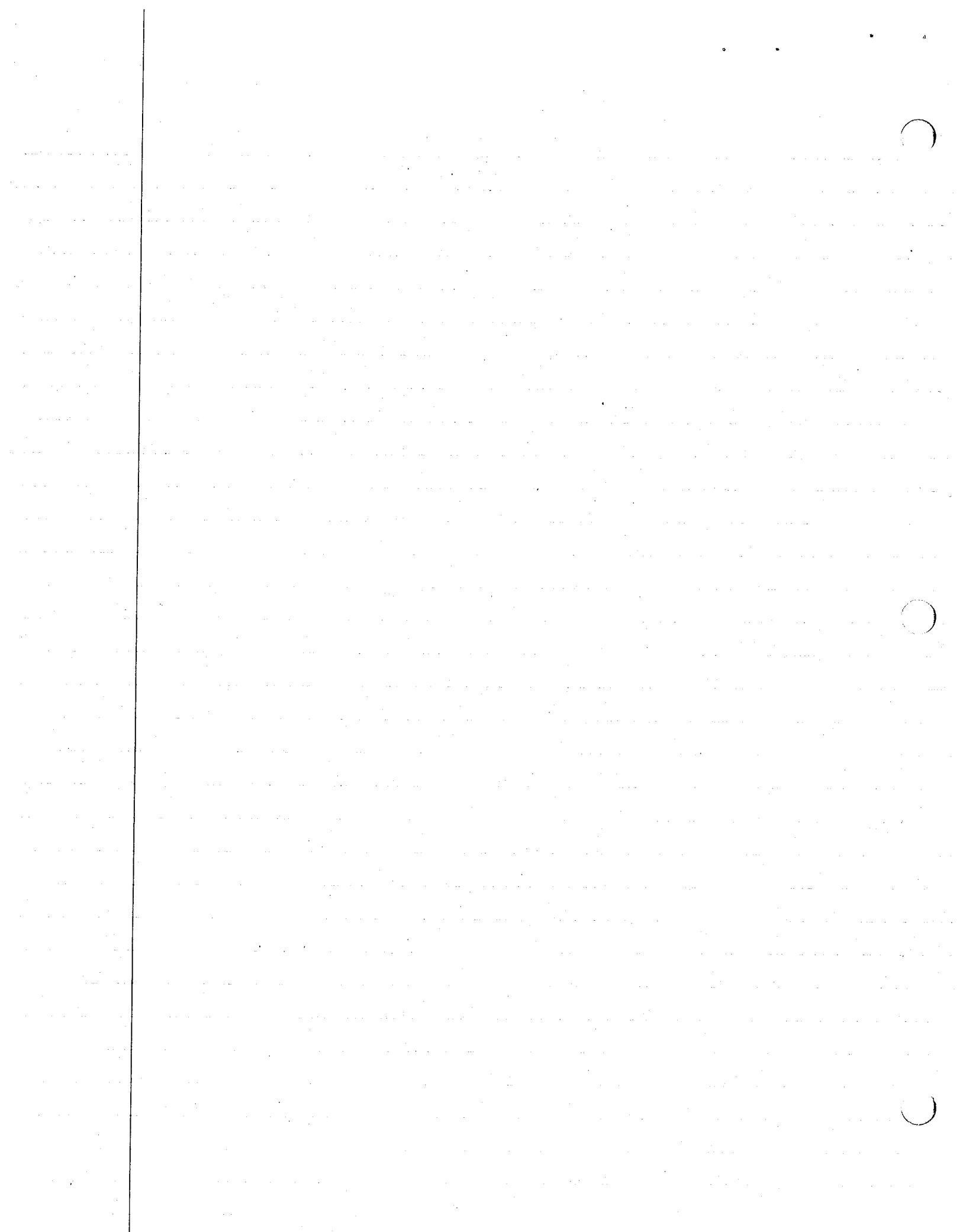
Using these results we find

$$F_1^* = -(M+3m)[u_1 + u_5 u_3 - u_6 u_2] \quad (35)$$

$$F_2^* = -(M+3m)[u_2 + u_6 u_1 - u_4 u_3] \quad (36)$$

$$F_3^* = -(M+3m)[u_3 + u_4 u_2 - u_5 u_1] \quad (37)$$

$$F_4^* = m_1 \cdot (\Pi^{A^*} + \Pi^{C^*} + \Pi^{B_1^*} + \Pi^{B_2^*}) \quad (38)$$



$$F_5^* = -2\ell^2 m [\dot{u}_5 - u_4 u_6] + m_2 \cdot [\Pi^{*A} + \Pi^{*C} + \Pi^{*B_1} + \Pi^{*B_2}] \quad (39)$$

$$F_6^* = -2\ell^2 m [\dot{u}_6 + u_4 u_5] + m_3 \cdot [\Pi^{*A} + \Pi^{*B} + \Pi^{*B_1} + \Pi^{*B_2}] \quad (40)$$

$$F_7^* = m_3 \cdot [\Pi^{*C} - \Pi^{*B_1} - \Pi^{*B_2}] \quad (41)$$

Now

$$\Pi^{*C} = -\frac{d}{dt} \Pi^{*C/C^*} = -\Pi^{*C/C^*} \cdot \alpha^A + \alpha^C = -\Pi^{*C/C^*} (\alpha^A + \alpha^C) \quad (42)$$

$$\Pi^{*B_1} = -\frac{d}{dt} \Pi^{*B_1/B_1^*} = -\Pi^{*C/C^*} \cdot \alpha^A + \alpha^{\Pi^{*B_1}} \text{ since } \Pi^{*C/C^*} = \Pi^{*B_1/B_1^*} = \Pi^{*B_2/B_2^*} \quad (43)$$

$$\Pi^{*B_2} = -\frac{d}{dt} \Pi^{*B_2/B_2^*} = -\Pi^{*C/C^*} \cdot \alpha^A + \alpha^{\Pi^{*B_2}} = -\Pi^{*C/C^*} (\alpha^A + \alpha^B) = \Pi^{*B_2} \quad (44)$$

Using these in (41) leads to

$$F_7^* = -J(\dot{u}_6 + \dot{u}_7) + 2J(\dot{u}_6 - \dot{u}_7) = -3J\dot{u}_7 + J\dot{u}_6 \quad (45)$$

Now

$$\begin{aligned} \Pi^{*A} + \Pi^{*C^*} + \Pi^{*B_1^*} + \Pi^{*B_2^*} &= -\frac{d}{dt} \left[ \Pi^{*A/A^*} + \Pi^{*C/C^*} + \Pi^{*B_1/B_1^*} + \Pi^{*B_2/B_2^*} \right] \\ &= -\frac{d}{dt} \left[ (\Pi^{*A/A^*} \cdot \Pi^{*C/C^*} + \Pi^{*B_1/B_1^*} + \Pi^{*B_2/B_2^*}) \cdot \omega^A + \Pi^{*C/C^*} \cdot \omega^C + \Pi^{*B_1/B_1^*} \cdot \omega^{B_1} + \Pi^{*B_2/B_2^*} \cdot \omega^{B_2} \right] \end{aligned} \quad (46)$$

$$\text{Using } \Pi^{*B/A^*} = \Pi^{*B_1/B_1^*} + \Pi^{*B_2/B_2^*} \text{ and } \Pi^{*B_2/A^*} = \Pi^{*B_2/B_2^*} + \Pi^{*B_2/A^*} \quad (47 \& 48)$$

then (46) becomes:

$$\begin{aligned} &= -\frac{d}{dt} \left[ \Pi^{*(A+C+B_1+B_2)/A^*} \cdot \omega^A - (\Pi^{*B_1/A^*} + \Pi^{*B_2/A^*}) \cdot \omega^A - \Pi^{*C/C^*} \cdot \omega^C \right] \\ &= \Pi^{*D} - \Pi^{*C} + 2\ell^2 m \frac{d}{dt} [(m_2 m_2 + m_3 m_3) \cdot \omega^A] + \omega^A \times \Pi^{*C/C^*} \end{aligned} \quad (49)$$

This is obtained since

$$\Pi^{*B/A^*} = m L^2 (W - m_1 m_1) = m L^2 (m_2 m_2 + m_3 m_3) = \Pi^{*B_2/A^*} \text{ where } W \text{ is the unit dyadic} \quad (50)$$

If we note that  $\frac{d}{dt} [(m_2 m_2 + m_3 m_3) \cdot \omega^A] = (\dot{u}_5 - u_6 u_4) m_2 + (\dot{u}_6 + u_5 u_4) m_3$  then

$$\Pi^{*A} + \Pi^{*C^*} + \Pi^{*B_1^*} + \Pi^{*B_2^*} = \Pi^{*D} - \Pi^{*C} + 2\ell^2 m \{ (\dot{u}_5 - u_6 u_4) m_2 + (\dot{u}_6 + u_5 u_4) m_3 \} + \omega^A \times \Pi^{*C/C^*} \quad (51)$$

Here  $D$  is a fictitious rigid body whose mass distribution is the same as  $A + B_1 + C + B_2$  and whose motion is that of body A. (i.e.  $\Pi^{*D} = (\Pi^{*D/A^*} \cdot \omega^A) \times \omega^A - (\Pi^{*D/A^*} \cdot \alpha^A)$ ). (52)

Using (51) in (38-40) and noting that  $\Pi^{*D/A^*} = K_1 m_1 m_1 + K_2 m_2 m_2 + K_3 m_3 m_3$  (53)

$$F_4^* = (K_2 - K_3) u_5 u_6 - K_1 \dot{u}_4 + J \dot{u}_7 u_5 \quad (54)$$

$$F_5^* = (K_3 - K_1) u_6 u_6 - K_2 \dot{u}_5 - J \dot{u}_7 u_4 \quad (55)$$

$$F_6^* = (K_1 - K_2) u_4 u_5 - K_3 \dot{u}_6 + J \dot{u}_7 \quad (56)$$

These are obtained by observing that  $\Pi^{*C} = -J \dot{u}_7 m_3$  and  $\omega^A \times \Pi^{*C/C^*} = J \dot{u}_7 (u_5 m_1 - u_4 m_2)$  (57, 58)

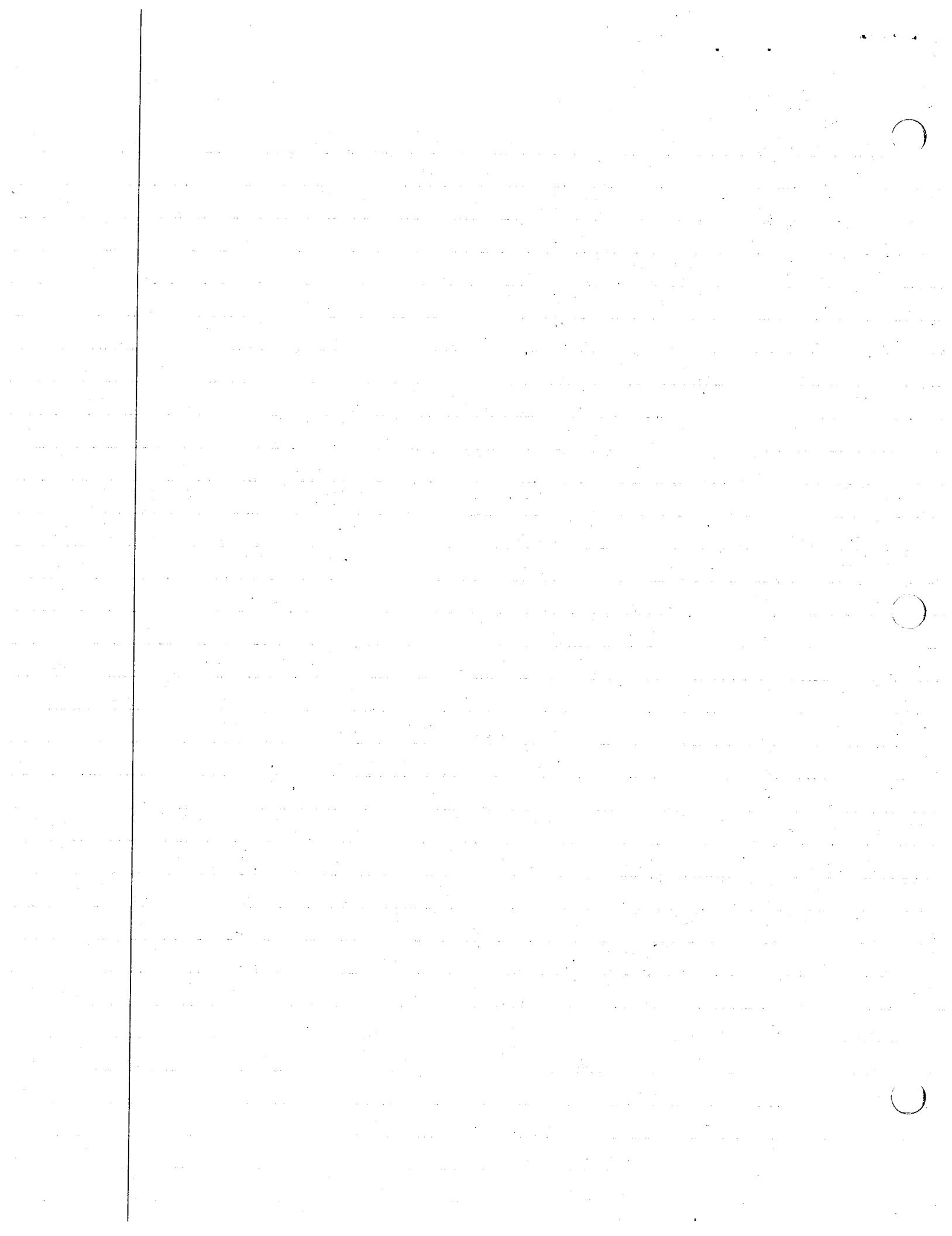
Now using the equations of motion  $F_r + F_r^* = 0$  for  $r = 4, \dots, 7$  i.e. eqns (57-58), (51) we obtain

$$K_1 \dot{u}_4 = (K_2 - K_3) u_5 u_6 + J \dot{u}_7 u_5 \quad (59)$$

$$K_2 \dot{u}_5 = (K_3 - K_1) u_4 u_6 - J \dot{u}_7 u_4 \quad (60)$$

$$K_3 \dot{u}_6 = J \dot{u}_7 + (K_1 - K_2) u_4 u_5 \quad (61)$$

$$3J \dot{u}_7 = \dot{T}_{11} + T_{22} \quad (62)$$



$$K_3 \dot{u}_6 = \frac{Ju_6}{3} + \frac{T_3}{3} + (K_1 - K_2) u_4 u_5 \Rightarrow \left[ \dot{u}_6 = \left( \frac{T_3}{3K_3} + \frac{K_1 - K_2}{K_3} u_4 u_5 \right) / \left( 1 - \frac{J}{3K_3} \right) \right] \quad (63)$$

$$\left[ \dot{u}_7 = \frac{\dot{u}_6}{3} + \frac{T_3}{3J} = \left( \frac{T_3}{3J} + \frac{K_1 - K_2}{3K_3} u_4 u_5 \right) / \left( 1 - \frac{J}{3K_3} \right) \right] \quad (64)$$

$$\left[ \dot{u}_4 = \left( \frac{K_2 - K_3}{K_1} \right) u_5 u_6 + \frac{J}{K_1} u_7 u_5 \right] \quad (65)$$

$$\left[ \dot{u}_5 = \left( \frac{K_3 - K_1}{K_2} \right) u_4 u_6 - \frac{J}{K_2} u_7 u_4 \right] \quad (66)$$

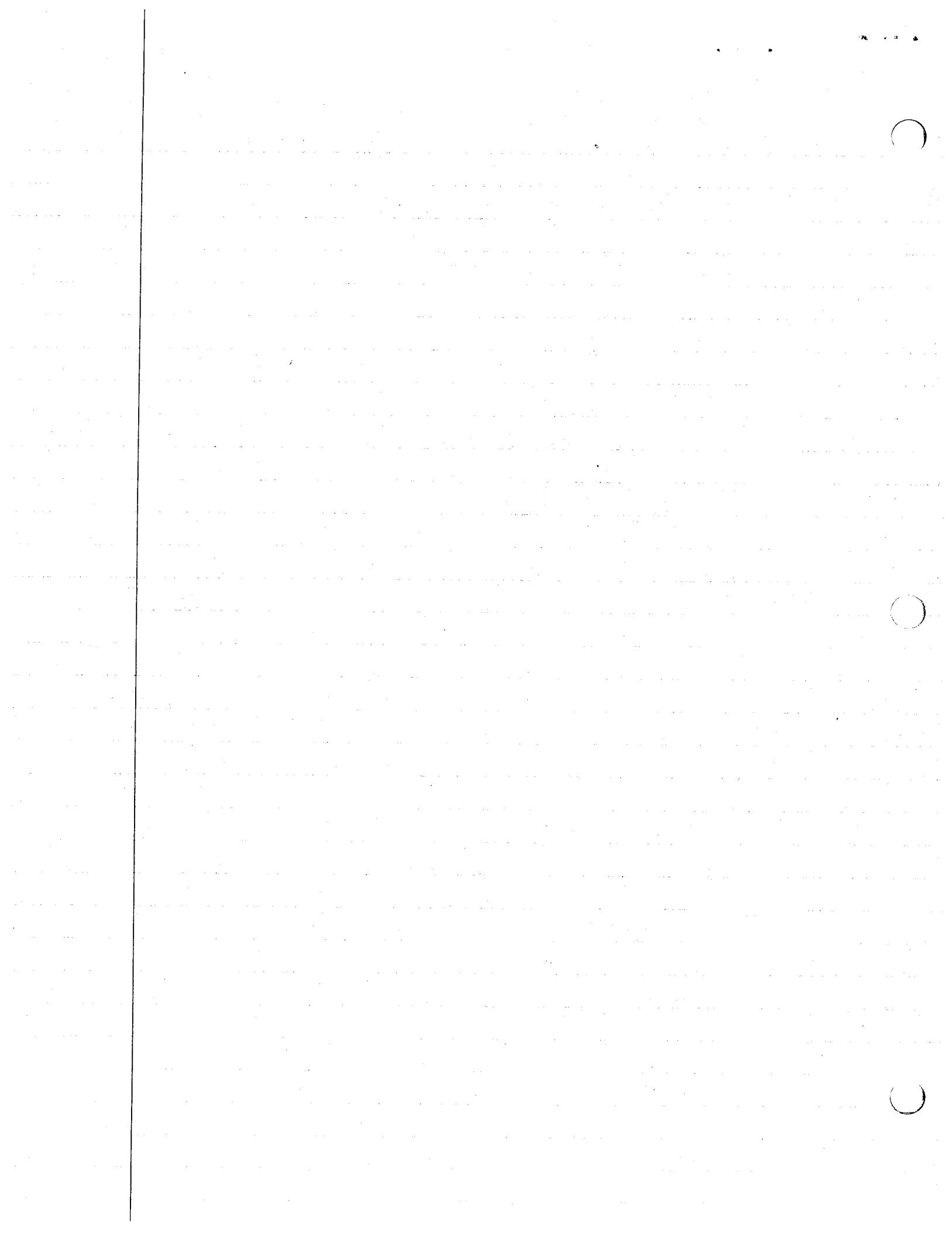
Using the definitions for the  $u_i$ 's to obtain

$$\dot{\omega}_1 = \left[ \frac{K_2 - K_3}{K_1} \omega_3 + \frac{J}{K_1} \Omega \right] \omega_2 \quad (67)$$

$$\dot{\omega}_2 = \left[ \frac{K_3 - K_1}{K_2} \omega_3 - \frac{J}{K_2} \Omega \right] \omega_1 \quad (68)$$

$$\dot{\omega}_3 = \left[ \frac{T_3}{3K_3} + \frac{K_1 - K_2}{K_3} \omega_1 \omega_2 \right] / \left( 1 - \frac{J}{3K_3} \right) \quad (69)$$

$$\dot{\Omega} = \left[ \frac{T_3}{3J} + \frac{K_1 - K_2}{3K_3} \omega_1 \omega_2 \right] / \left( 1 - \frac{J}{3K_3} \right) \quad (70)$$



Very nice

800310

A point  $O$  of a rigid body  $B$  is fixed. Mutually perpendicular axes  $B_1, B_2, B_3$  are fixed in  $B$  and intersect at  $O$ . Initially,  $B_1, B_2, B_3$  are aligned with fixed axes  $A_1, A_2, A_3$ , respectively, and successive rotations of  $B$  of amounts  $q_1, q_2, q_3$  are then performed about  $B_1, B_2, B_3$ , respectively, to bring  $B$  into a general position.

A force  $\underline{F}$  is applied to  $B$  at a point  $P$ .  $\underline{F}$  depends on the orientation of  $B$  as follows:

$$\underline{F} = kL[(1 - c_2c_3)\underline{b}_1 + (1 + c_2s_3)\underline{b}_2 - s_2\underline{b}_3]$$

where  $k$  and  $L$  are constants,  $\underline{b}_1$  is a unit vector parallel to  $B_1$ , and  $s_j = \sin q_j$ ,  $c_j = \cos q_j$ . The position vector  $\underline{p}$  of  $P$  relative to  $O$  is given by

$$\underline{p} = L(\underline{b}_1 + \underline{b}_2)$$

Find a potential function for  $\underline{F}$ .



From problem 2a: If we denote fixed reference frame A having axes  $A_1, A_2, A_3$  with unit vectors  $a_{11}, a_{12}, a_{13}$  along these axes respectively; and the body B having axes  $B_1, B_2, B_3$  which are fixed in the body and mutually perpendicular with unit vectors  $b_{11}, b_{12}, b_{13}$  along these axes respectively, then by introducing intermediate reference frames I<sub>1</sub> and I<sub>2</sub> with unit vectors  $(\hat{a}_1, \hat{a}_2, \hat{a}_3)$  and  $(\hat{j}_1, \hat{j}_2, \hat{j}_3)$  which are fixed in each reference frame and mutually perpendicular we may write

$$\stackrel{A}{\omega}^B = \stackrel{A}{\omega}^{I_1} + \stackrel{I_1}{\omega}^{I_2} + \stackrel{I_2}{\omega}^B = \dot{q}_1 a_{11} + \dot{q}_2 \hat{a}_2 + \dot{q}_3 b_{13} \quad (1)$$

where

$$a_{11} = c_2 c_3 b_{11} - c_2 s_3 b_{12} + s_2 b_{13} \quad (2)$$

$$\hat{a}_2 = s_3 b_{11} + c_3 b_{12} \quad (3)$$

In order to find a potential function for  $\Pi^o$  we must first form the contribution of  $\Pi^o$  to the generalized active forces  $F_r'$  and then use

$$F_r' = - \frac{\partial P}{\partial q_r} \quad r=1,2,3 \quad (4)$$

$$\text{It should be noted that because point } O \text{ is fixed } \Pi^o = 0 \quad (5)$$

and it would be advantageous to replace  $\Pi^o$  acting at P by an equivalent force at O and moment about point O. This leads to

$$\Pi^o = kL [(1 - c_2 c_3) b_{11} + (1 + c_2 s_3) b_{12} - s_2 b_{13}] \quad (6)$$

$$\text{and } \Pi^o = |P \times \Pi^P| = L(b_{11} + b_{12}) \times |F^o| = kL^2 [-s_2 b_{11} + s_2 b_{12} + c_2 (s_3 + c_3) b_{13}] \quad (7)$$

To form  $F_r'$

$$F_r' = \nabla_{q_r}^\circ \cdot \Pi^o + \stackrel{A}{\omega}_{q_r}^B \cdot \Pi^o = \stackrel{A}{\omega}_{q_r}^B \cdot \Pi^o \quad r=1,2,3 \quad (8)$$

$$\text{Thus } F_1' = q_1 \cdot \Pi^o = 0 = - \frac{\partial P}{\partial q_1} \quad (9)$$

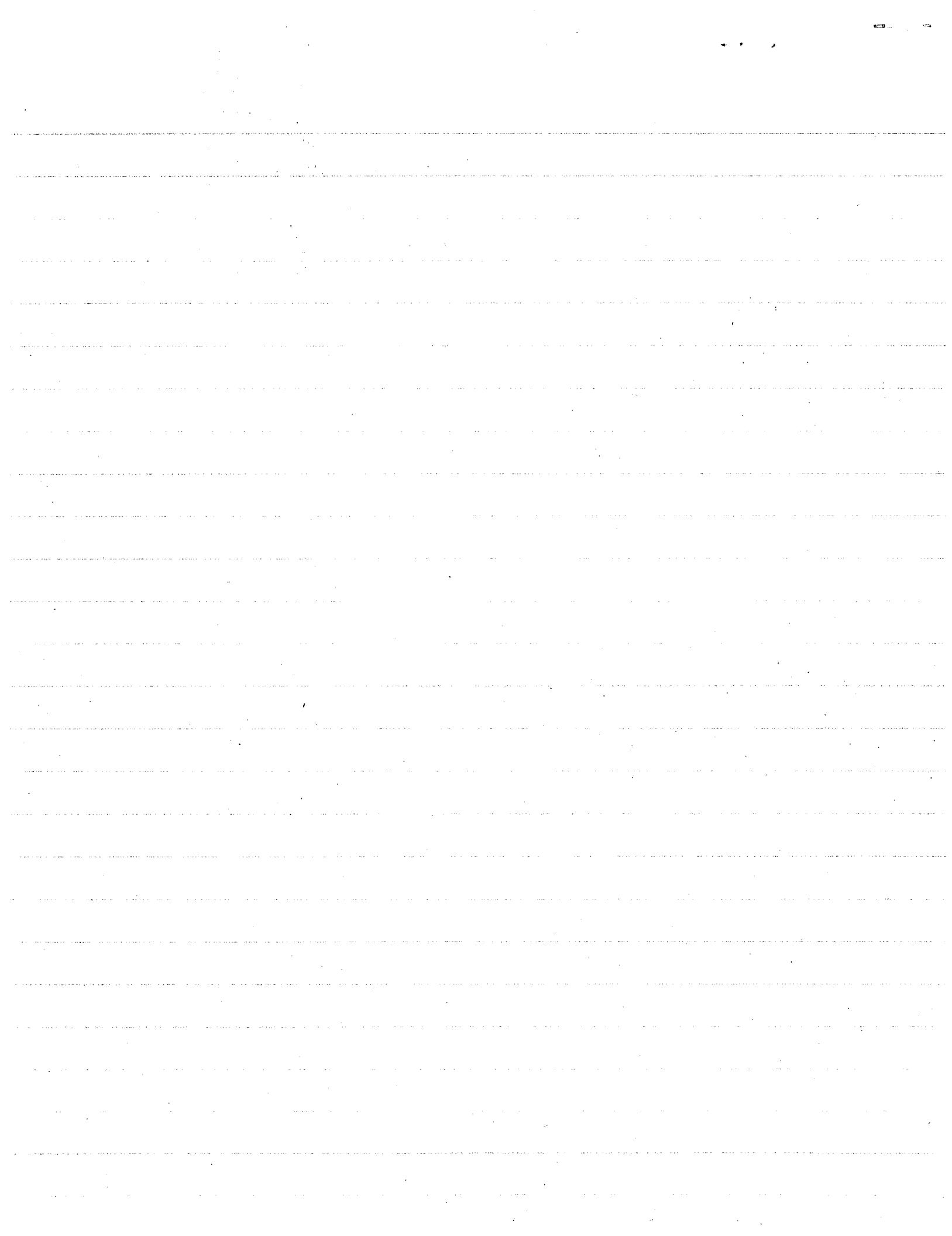
$$F_2' = \hat{a}_2 \cdot \Pi^o = kL^2 [s_2(c_3 - s_3)] = - \frac{\partial P}{\partial q_2} \quad (10)$$

$$F_3' = b_{13} \cdot \Pi^o = kL^2 [c_2(c_3 + s_3)] = - \frac{\partial P}{\partial q_3} \quad (11)$$

$$\text{from (9): } P = f(q_2, q_3, t) \text{ only} \quad (12)$$

$$\text{from (10): } -P = kL^2 [-c_2(c_3 - s_3)] = g(q_3, t) \quad (13)$$

$$\text{from (13): } -\frac{\partial P}{\partial q_3} = kL^2 [c_2(s_3 + c_3)] = \frac{\partial g}{\partial q_3} = kL^2 [c_2(c_3 + s_3)] \Rightarrow \frac{\partial f}{\partial q_3} = 0 \text{ and } g = g(t) \text{ only}$$



therefore using the results of (12), (13), (4) we obtain

$$P = k L^2 [c_2(c_3 - s_3)] + g(t) \quad \text{for force } F \quad (15)$$





DYNAMICS PROJECT:

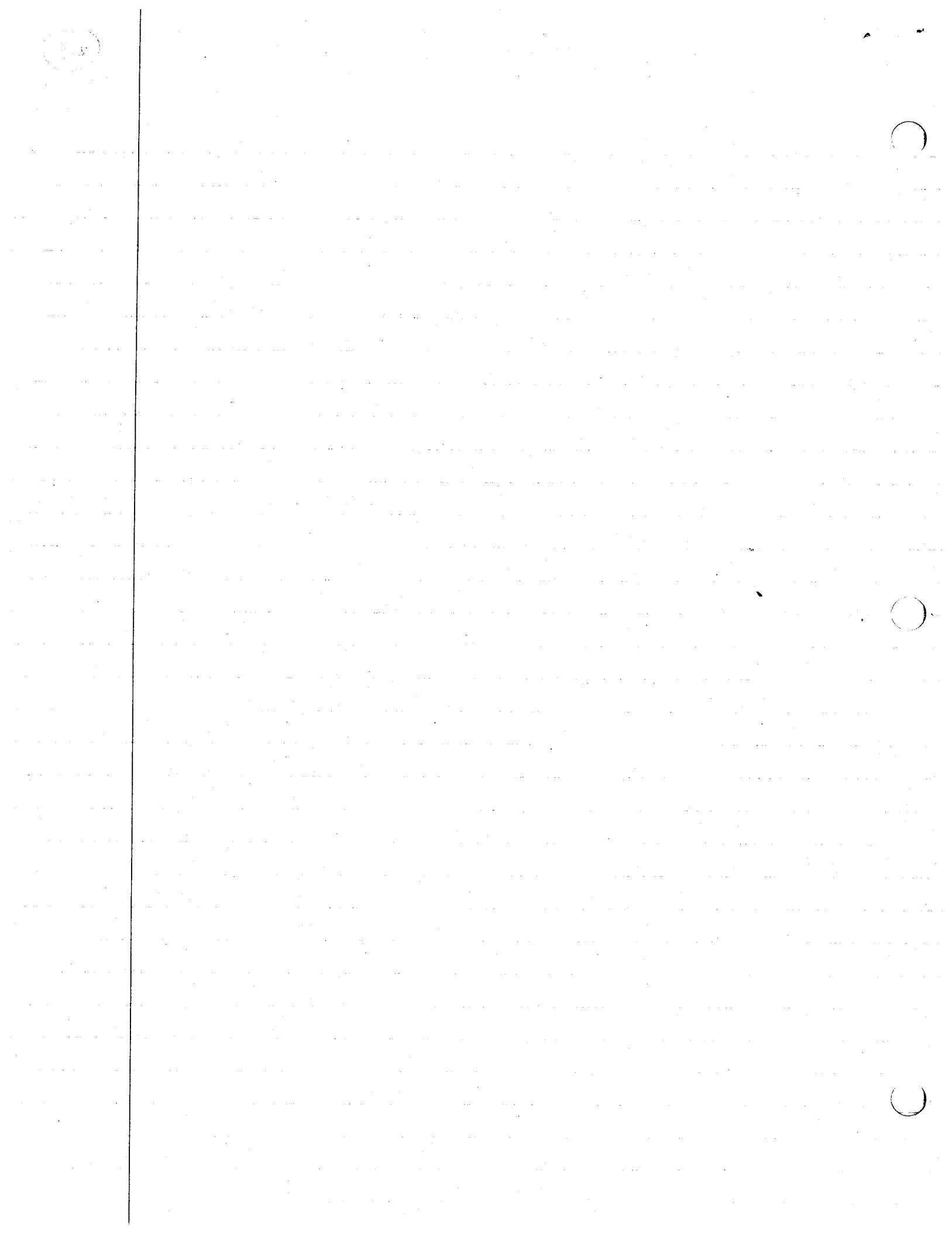
STUDY  
OF A  
SIMPLE GYROSTAT

CESAR LEVY

ME 231C

Winter 1980

Prof. Kane



## Study of a Simple Gyrostat

The study of a simple gyrostat  $G$  consisting of an axisymmetric rotor  $B$  whose axis and mass center  $B^*$  are fixed in a rectangular body  $A$  is performed with the additional condition that the mass centers of both bodies are to coincide (see Figure 1). A set of unit vectors  $a_1, a_2, a_3$  form a right-handed mutually perpendicular triad fixed in  $A$ , each vector parallel to a central principal axis of inertia of  $G$ . The axis of  $B$  is parallel to a unit vector  $\beta$ , and the rotor-rectangular body system is such that the central principal moments of inertia of  $G$  are equal to those of the rectangular body. These central principal moments of inertia are denoted by  $I_1, I_2, I_3$  and  $B$  has a polar moment of inertia about  $\beta$  equal to  $J$ .

If we define  $\overset{A}{\omega}{}^B$ ,  $w_i$  and  $\beta_i$  by as

$$\overset{A}{\omega}{}^B \triangleq \overset{A}{\omega}{}^B \cdot \beta \quad (1)$$

$$w_i \triangleq \overset{R}{\omega}{}^A \cdot a_{1i} \quad (i=1,2,3) \quad (2)$$

$$\beta_i \triangleq \beta \cdot a_{1i} \quad (i=1,2,3) \quad (3)$$

with  $\overset{A}{\omega}{}^B$  being the angular velocity of  $B$  in  $A$  and  $\overset{R}{\omega}{}^A$  being the angular velocity of  $A$  in a "fixed" reference frame  $R$ , we may then write the following set of differential equations that govern  $\overset{A}{\omega}{}^B$  and  $w_i$  ( $i=1,2,3$ ):

$$\overset{A}{\dot{\omega}}{}^B = (\frac{I}{J} - H)/c_1; \quad (4)$$

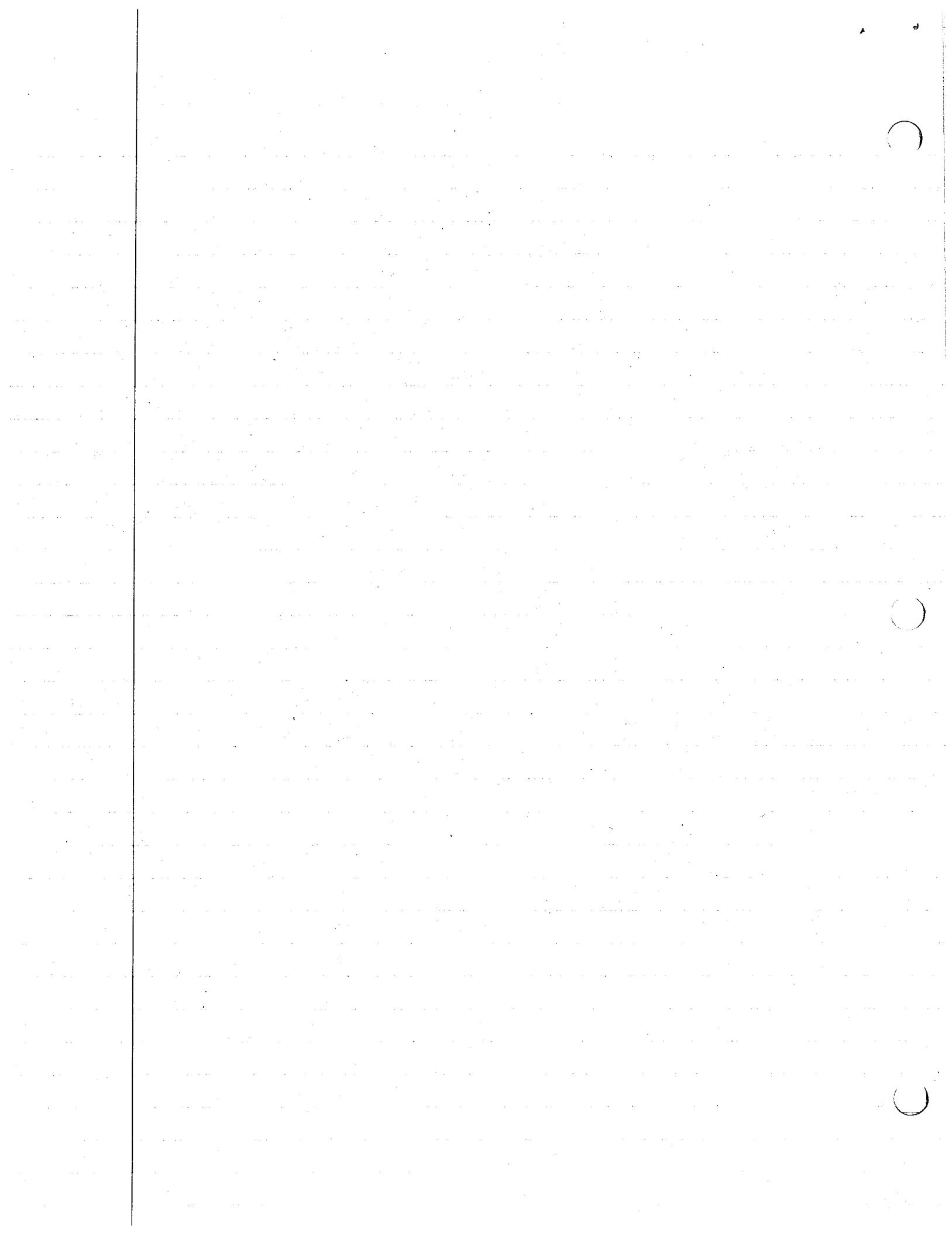
$$\dot{w}_1 = [F_1 - \frac{\beta_1 J}{c_1} (T_J - H)]/I_1; \quad (5)$$

$$\dot{w}_2 = [F_2 - \frac{\beta_2 J}{c_1} (T_J - H)]/I_2; \quad (6)$$

$$\text{and } \dot{w}_3 = [F_3 - \frac{\beta_3 J}{c_1} (T_J - H)]/I_3. \quad (7)$$

In the preceding equations  $c_1, F_i$  ( $i=1,2,3$ ),  $H$  and  $T$  are defined as follows:

$$c_1 \triangleq 1 - J \left[ \frac{\beta_1^2}{I_1} + \frac{\beta_2^2}{I_2} + \frac{\beta_3^2}{I_3} \right] \quad (8)$$



$$H \triangleq \beta_1 F_1 / I_1 + \beta_2 F_2 / I_2 + \beta_3 F_3 / I_3 \quad (9a)$$

$$F_1 \triangleq (I_2 - I_3) \omega_2 \omega_3 + J^A \dot{\omega}^B [\omega_3 \beta_2 - \omega_2 \beta_3] \quad (9b)$$

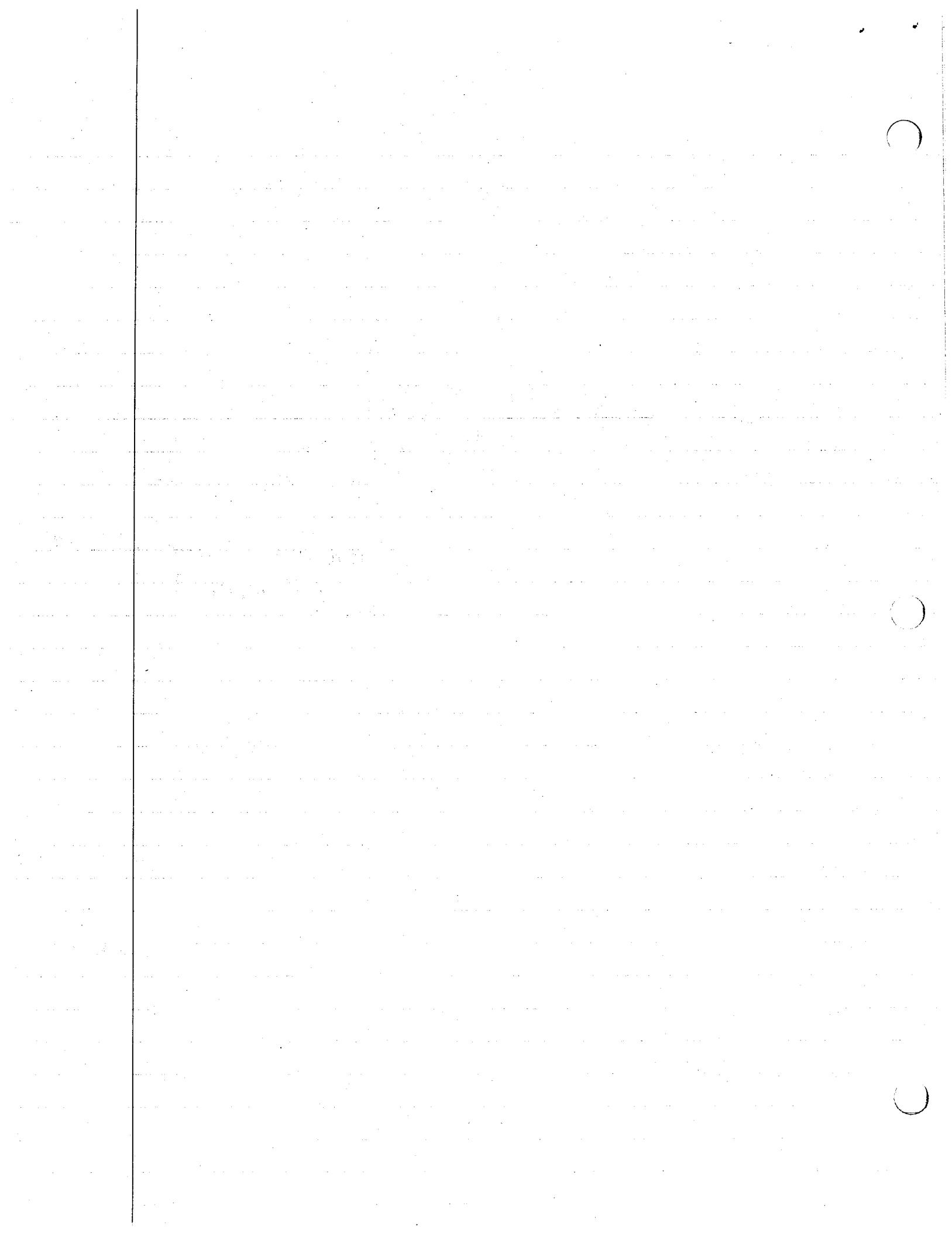
$$F_2 \triangleq (I_3 - I_1) \omega_3 \omega_1 + J^A \dot{\omega}^B [\omega_1 \beta_3 - \omega_3 \beta_1] \quad (9c)$$

$$F_3 \triangleq (I_1 - I_2) \omega_1 \omega_2 + J^A \dot{\omega}^B [\omega_2 \beta_1 - \omega_1 \beta_2] \quad (9d)$$

and T is the induced torque on B by A when B is free to rotate relative to A.

With the equations (4-7) the significant parameters that affect the stability of motion of body A may be studied based upon the induced torque T. We note that  $\beta$  is defined in the derivation section.

What is "based upon ... T"?  
Body A? Motion?



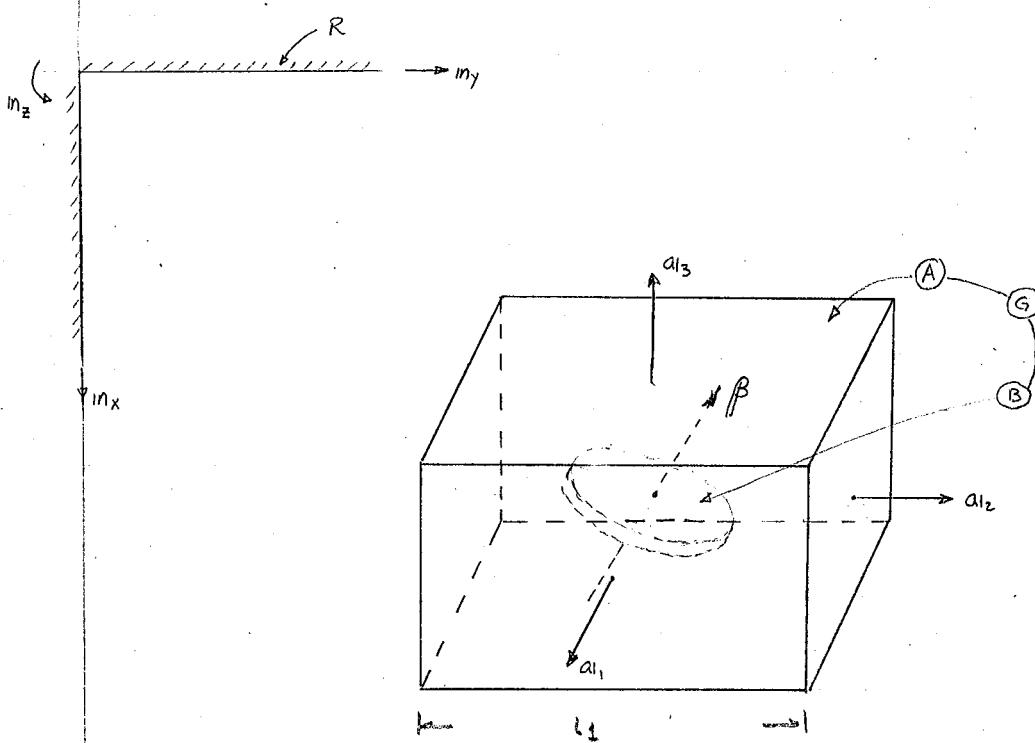
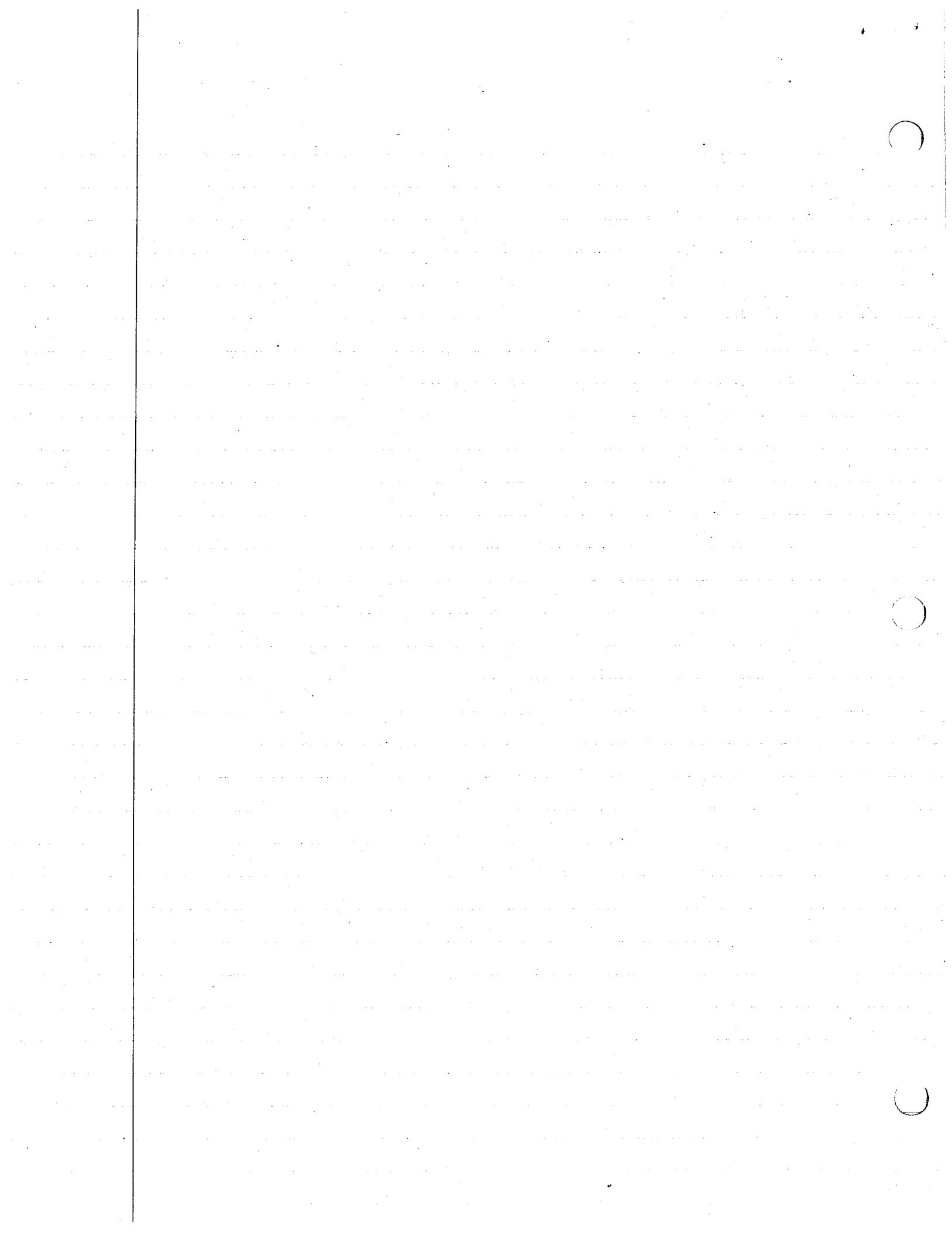


FIG. 1.

SIMPLE GYROSTAT SYSTEM



Where is the program?

## 2. Results

An initial value problem was solved for starting values of  $w_i$  ( $i=1, 2, 3$ ) and  $\dot{w}^B$  equal to zero by varying the following parameters one at a time:

$\mu = \frac{m_b}{m_b + m_a}$ , the ratio of the rotor mass to that of the rotor+body mass;

$\mu$  does not appear in the equations of motion. How can changing  $\mu$  affect the motion?

$T/J$ , the ratio of induced torque to rotor polar moment of inertia;

$\alpha, \beta$  the cant angle of the rotor axis; how defined?

$b = L_2/L_1$ ,  $c = L_3/L_1$ , where  $L_1, L_2, L_3$  are the rectangular body's lengths;

$d = r/2L_1$  where  $r$  is the rotor radius; and

$b_1$  being the length of the side of the rectangular body whose normal is  $\alpha$ ,

Figures (2-6) present the results for the variables indicated on the figures for  $|^R w^A|$  versus  $T$  (time) and  $^B w^B$  versus  $T$  for times up to 40 seconds.  
same symbol as for torque!

Figures (2a,b) indicate the variation of  $\mu$  with the remaining variables being unchanged, we note the linear characters of the plots and that increase in  $\mu$  causes corresponding increases in both  $|^R w^A|$  and  $^B w^B$ . We also observe that  $^B w^B$  is very much larger than  $|^R w^A|$  but that the magnitude of the changes in  $|^R w^A|$  between  $\mu=0.03$  and  $\mu=0.1$  is larger than the magnitude of  $^B w^B$ . The trend of the plots can be explained because the dominance of the rotor becomes significant.

We also find that by keeping  $\mu$  fixed and varying the rotor mass  $m_b$  there is no change in the plots and the results for, lets say  $\mu=0.03$ , are characterized by the  $\mu=0.03$  plots in Figures 2a and 2b. We see this when we note that  $m_b/m_a = (1-\mu)/\mu$  and that for fixed  $\mu$ , changes in  $m_b$  require changes in  $m_a$  such that the ratio of moments of inertia to rotor polar moment of inertia is unchanged.

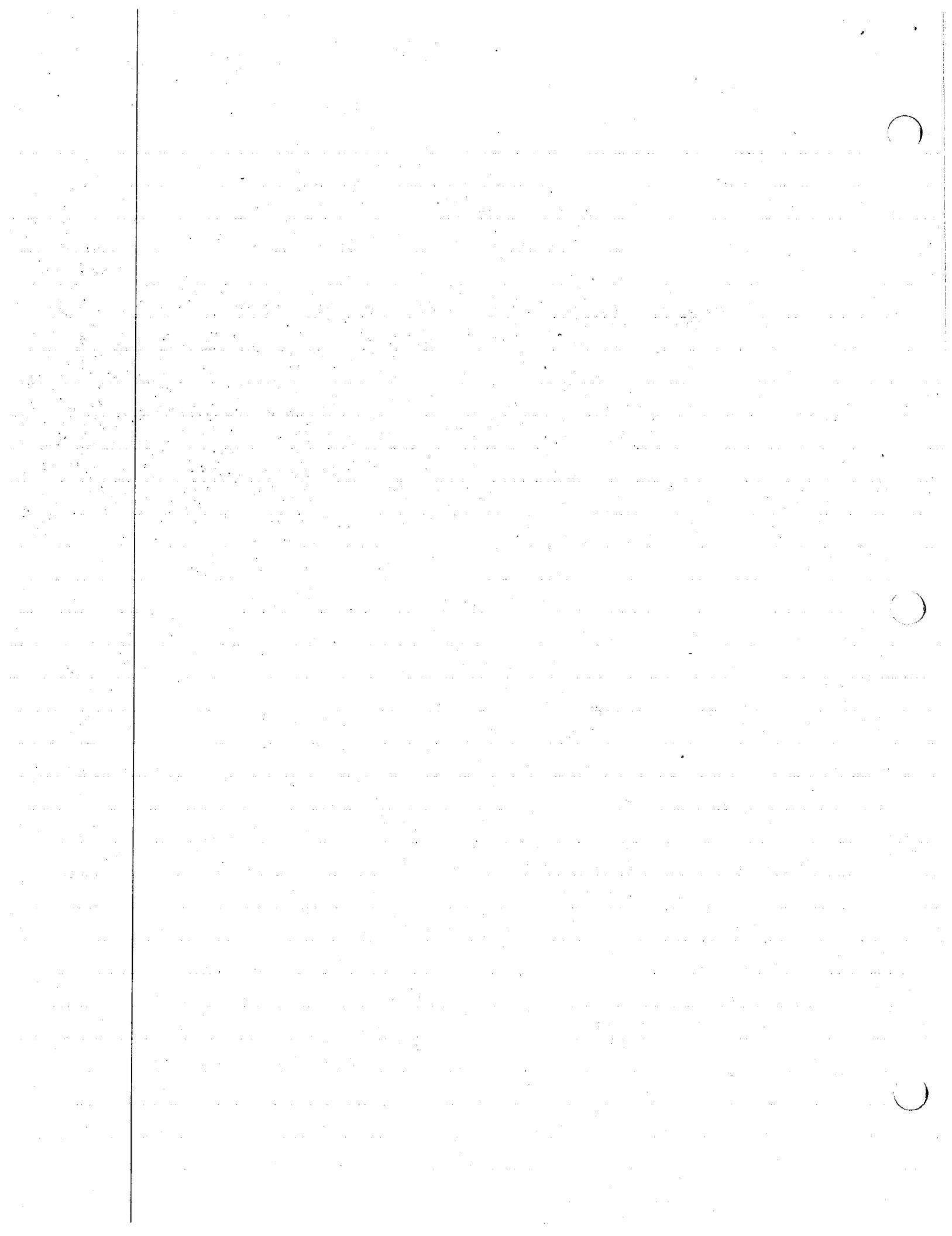


Figure 3 indicates the variations of  $T/J$ , all else being fixed; again we note the linear character of the plots and that  $\hat{w}^B$  is still larger than  $|^R w^A|$ . As  $T/J$  increases tenfold we note that  $\hat{w}^B$  and  $|^R w^A|$  does so, too. This is so because the dominant term in our differential equations is the  $T/J$  term.

Figure 4 provides a plot for variations in the cant angles  $\alpha$  and  $\beta$ . We note that the same results occur no matter what the variation in  $\alpha$  or no matter what the variations of  $\beta$ . The plots are such that each angle is varied separately while the other is kept fixed. This is due to the cubic nature of body A. From the numerical results obtained when only varying  $\alpha$  (not plotted) show that for  $\alpha = 90^\circ$  the  $w_3$  component is zero and that both  $w_1$  and  $w_2$  double when  $\alpha$  changes from  $30^\circ$  to  $90^\circ$ . Similarly when  $\beta$  is the only parameter varied, from the numerical results (not plotted) the  $w_1$  component is zero for  $\beta = 90^\circ$ ,  $w_2$  doubles when  $\beta$  changes from  $30^\circ$  to  $90^\circ$ , but  $w_3$  remains unchanged. These phenomena can be explained due to the fact that the components of the rotor axis ( $\beta_1, \beta_2, \beta_3$ ) are close to being zero or one and therefore suppress terms in the differential equations. If the starting values of  $\alpha$  and  $\beta$  were not  $30^\circ$  and  $10^\circ$  respectively, these effects would not be as apparent.

Figure 5 is a plot of the variations in both  $b$  and  $c$ . Again we note the linear character of all these plots and that as either  $b$  or  $c$  increases,  $|^R w^A|$  decreases; this is due to conservation of momentum. As  $b$ , for instance, increases then correspondingly  $I_2$  and  $I_3$  increase. Since initially all the values of  $w_i$  and  $\hat{w}^B$  were zero, the initial momentum was zero. Because of the increase in  $\hat{w}^B$  when the rotor is spun up, then the pertinent momentum equations

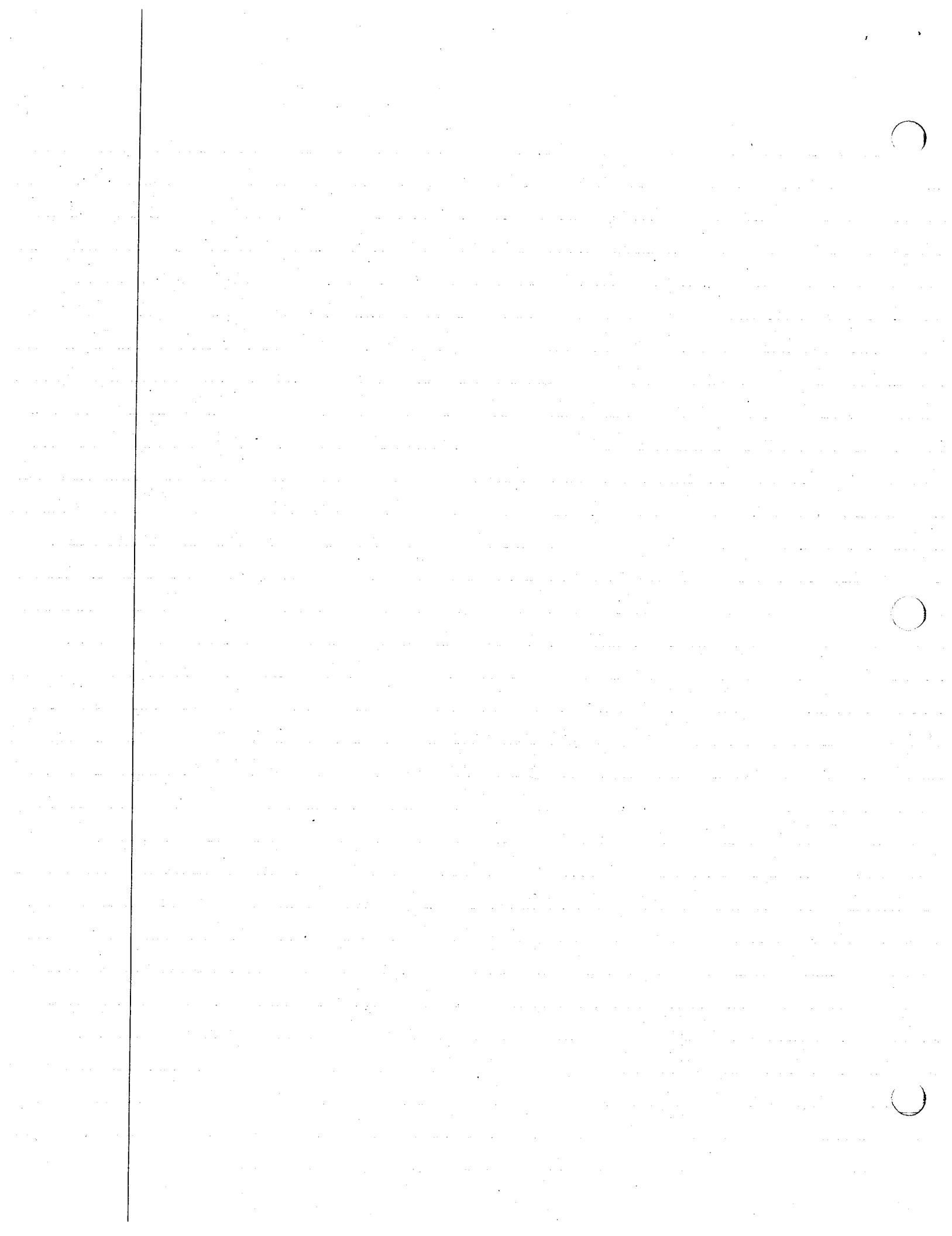
$$I_2 w_2 + \beta_2 J \hat{w}^B = 0$$

$$I_3 w_3 + \beta_3 J \hat{w}^B = 0$$

show that for fixed  $I_2, I_3, \beta_2, \beta_3$  and  $J$ , that  $w_2$  and  $w_3$  must increase. Also that for fixed  $\beta_3, J$  and  $\hat{w}^B$ , an increase in  $I_2$  or  $I_3$  (ie when  $b$  increases) requires a corresponding decrease in  $w_2$  and  $w_3$ .

Since  $|^R w^A| = \sum_{i=1}^3 (w_i \dot{w}_i)^{1/2}$  and  $w_i$  being close to the same value (numerical results not shown) then  $|^R w^A|$  must decrease.

Figure 6 provides us with the plots of the variations in  $d$ , the rotor diameter non-dimensionalization factor. We note again the linear character of the plots and as in all cases that  $|^R w^A|$  is less than its corresponding  $\hat{w}^B$  plot.



Here, as in the case of the  $\mu$  variations, as  $d$  increases  $|^R w^A|$  also increases. This is due to the rotor size being "felt" by the rectangular body (ie  $J \propto d^2$ ).

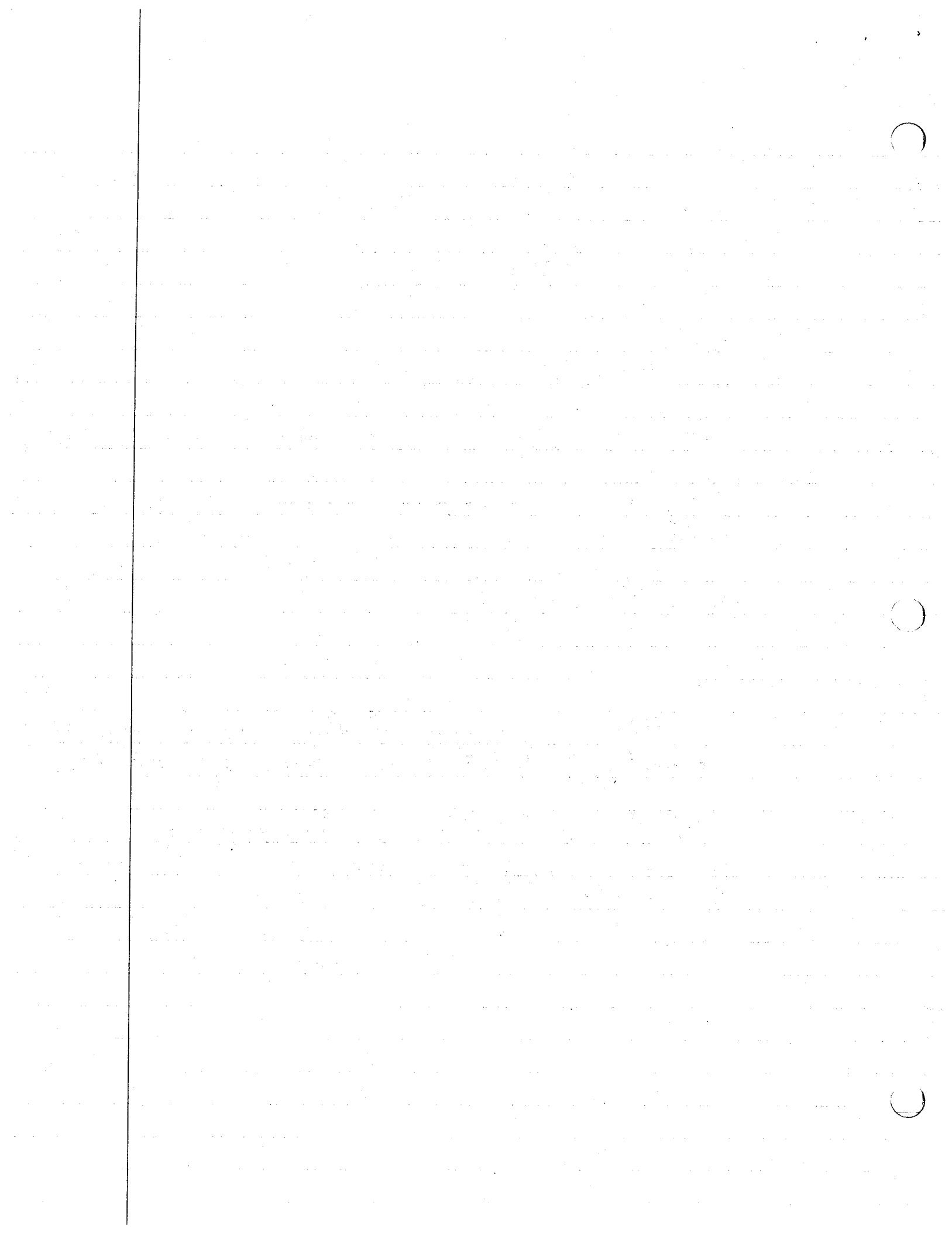
It is interesting to note that even though there are large variations in all the plots of  $|^R w^A|$ , the variations of  $^A w^B$  is still close even at time  $T = 40$  sec. Of course we expect this to change over longer periods of time.

Based on the results obtained we may then recommend the following in order to keep the rectangular body's motion stable:

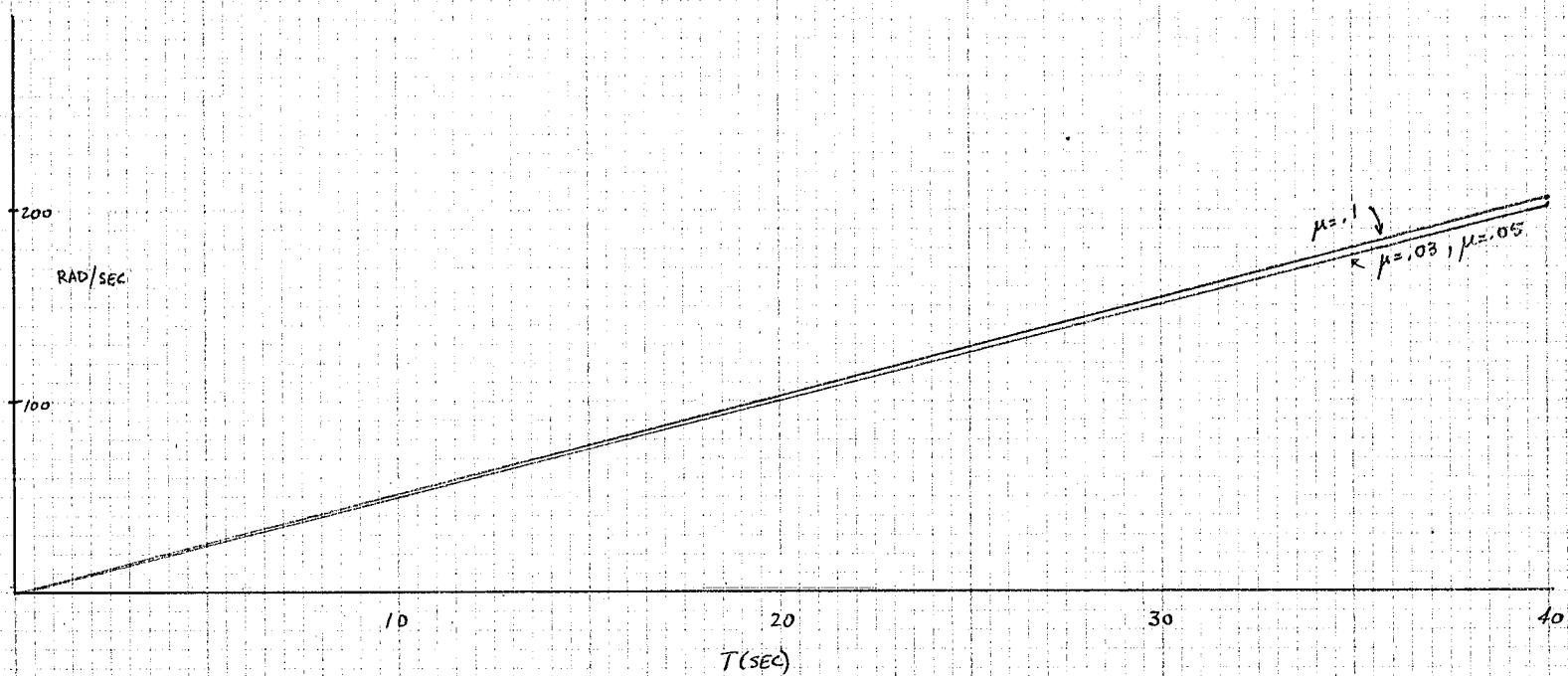
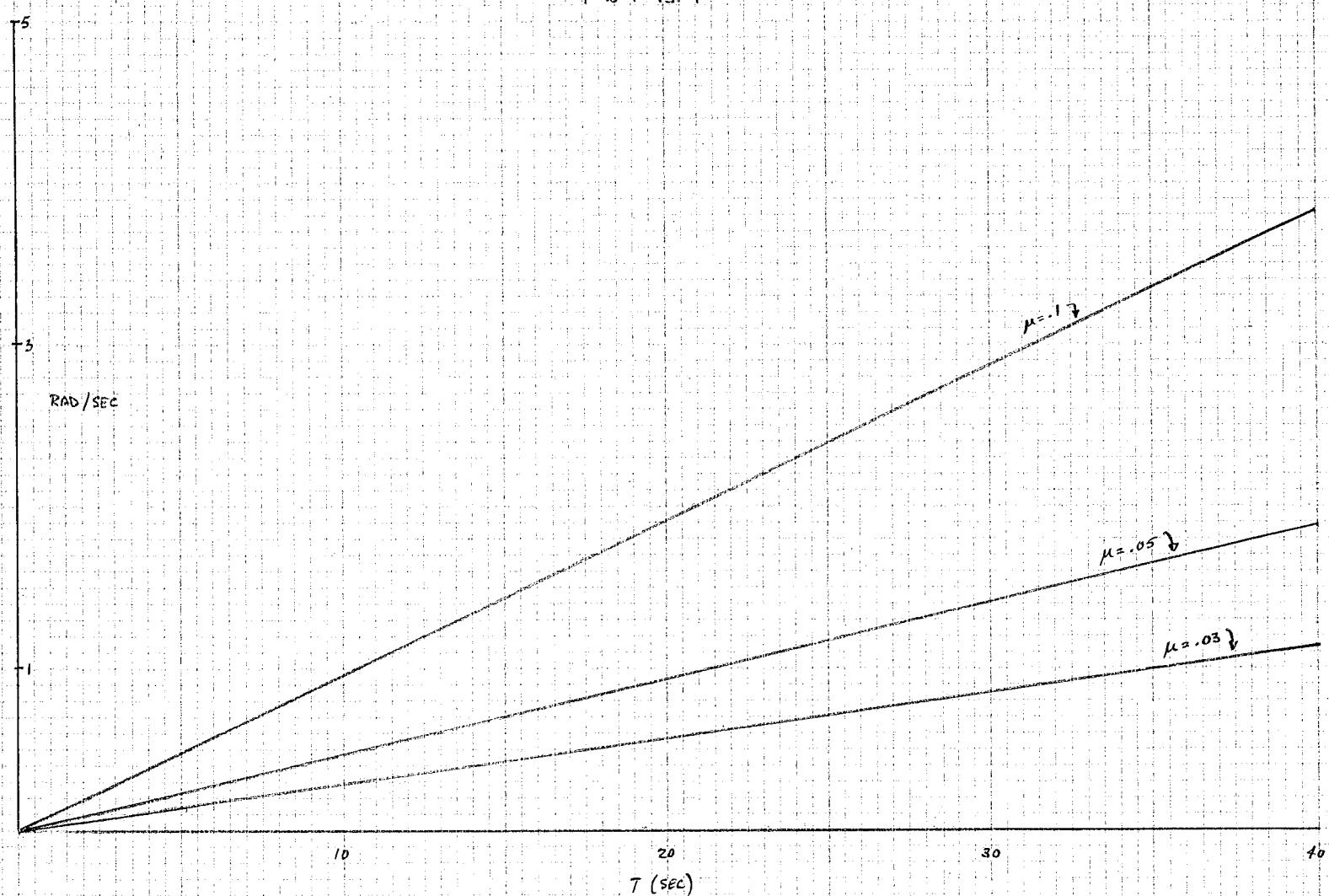
1. Keep  $\mu$  as small as possible yet large enough for the rotor to do its job;
2. Keep  $T/J$  small or cycle the rotor so that  $T$  varies sinusoidally over a short period;
3. If the cant angle of the rotor axis is to lie along one of the central principle axes, increase the length of sides "perpendicular" to the rotor axis (ie if axis is parallel to  $L_3$ , increase  $L_1$  and  $L_2$ );
4. Keep the rotor diameter as small as possible.

What does any of this have to do with stability? What do even mean by stability?

So far as I can see, the dynamics aspect is o.k. The use you made of the simulation is questionable.



FIGURES (2a, b)

 $M_b = 10 \text{ KG}$  $T/J = 5 \text{ RAD/SEC}^2$  $\alpha = 5^\circ, \beta = 10^\circ$  $b = 1.0, c = 1.9, d = 0.5$  $LL = 10 \text{ M}$  $|^A w^B| \text{ vs. } T$  $|^B w^A| \text{ vs. } T$ 

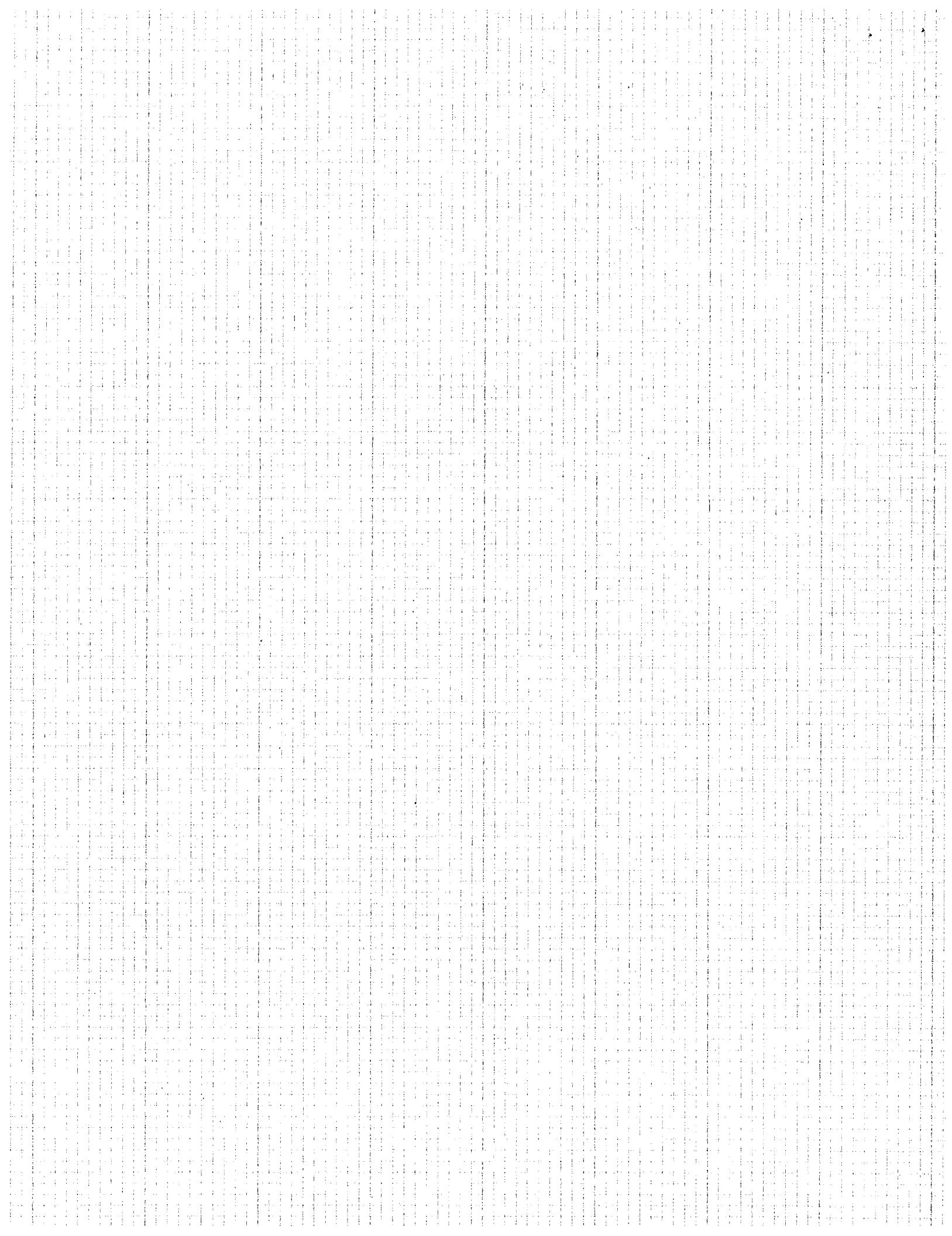


FIGURE 3  
 $\hat{\omega}^B$  and  $|^P\hat{\omega}^A|$  vs. T

$\mu = 0.8$

$m_B = 10 \text{ KG}$

$\alpha = 5^\circ, \beta = 10^\circ$

$b = c = 1.0, d = 0.5$

$L_1 = 10.0 \text{ M}$

+40

+30

RAD/SEC

+20

+10

0

T (sec)

20

30

40

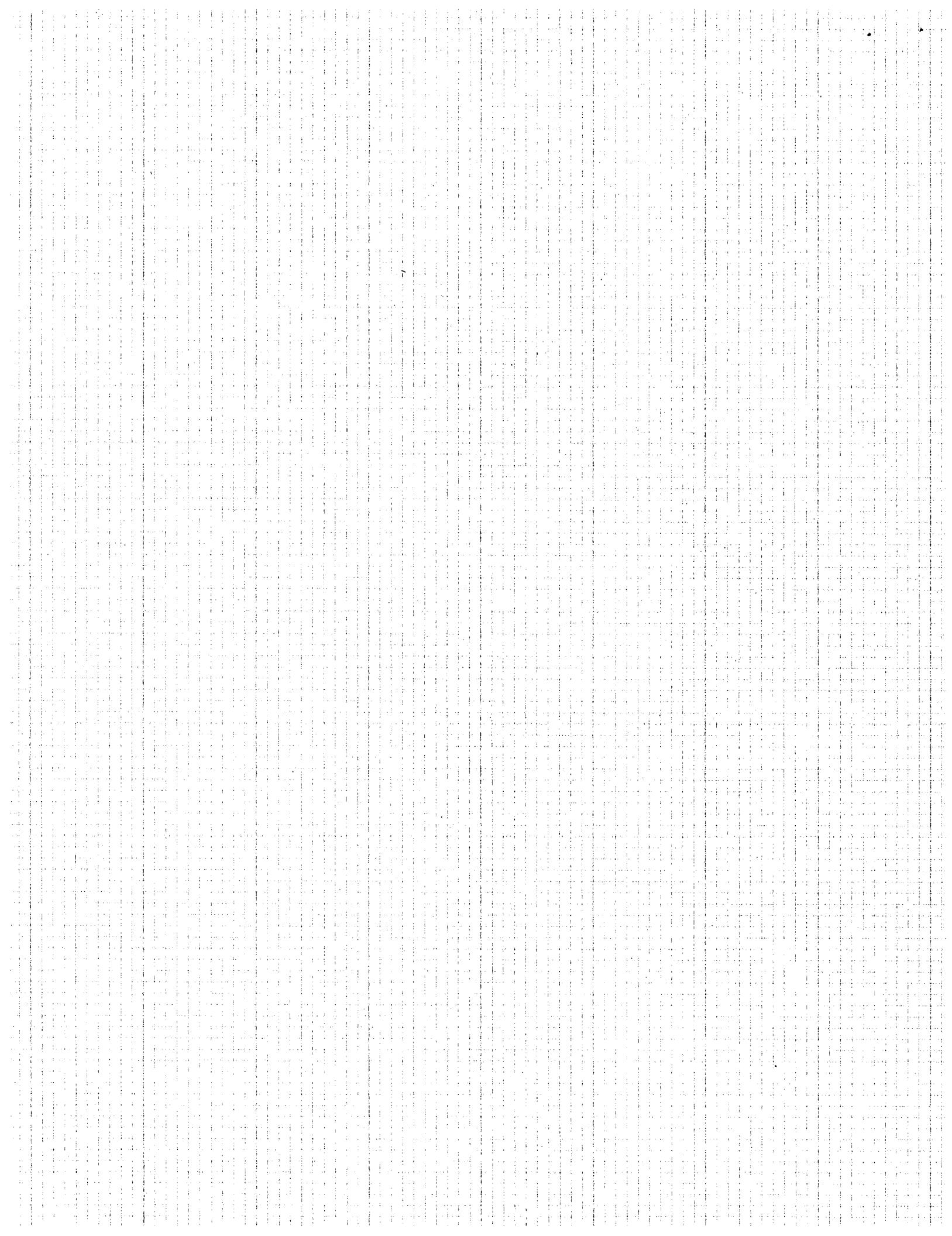
$T/J = 1.0$

$|^P\hat{\omega}^A| \text{ for } T/J = \frac{1.0}{0.01}$

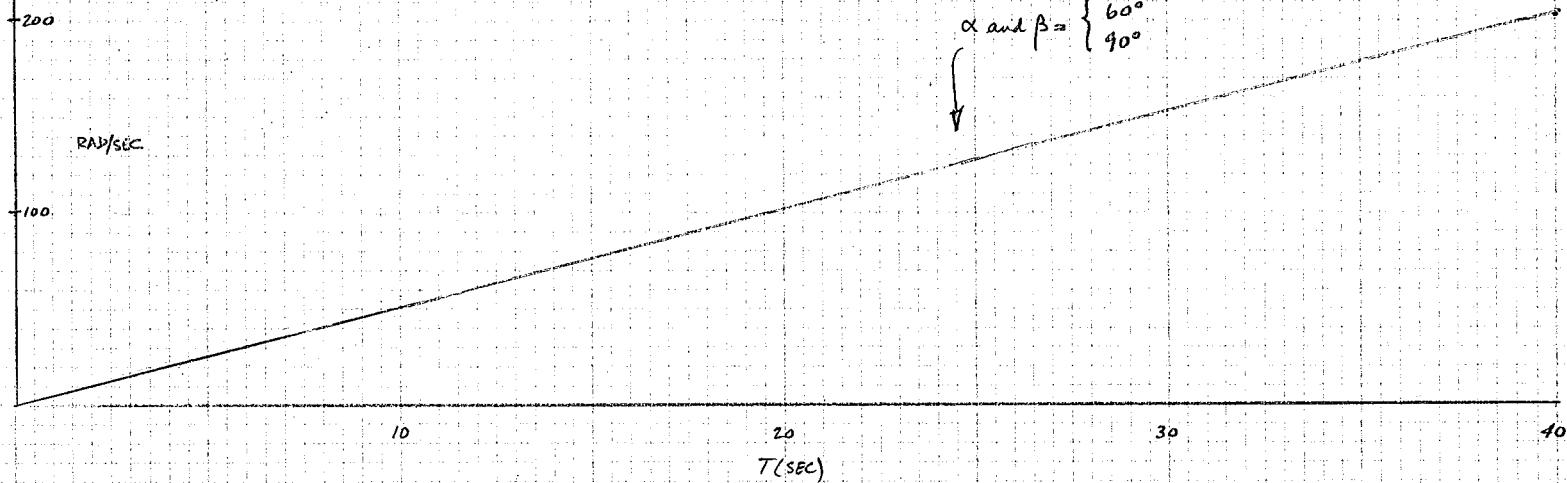
$T/J = 0.01$

$\hat{\omega}^B$

$\hat{\omega}^B$



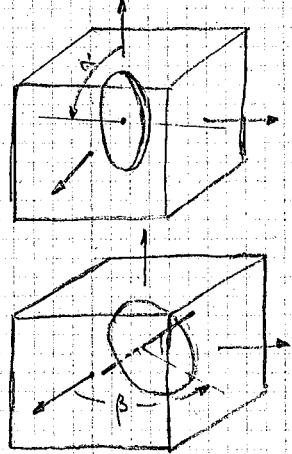
$\mu = .03$   
 $m_b = 10 \text{ KG}$   
 $T/J = 5 \text{ RAD/SEC}^2$   
 $b = c = 1.0, d = 0.5$   
 $LL = 10 \text{ M}$



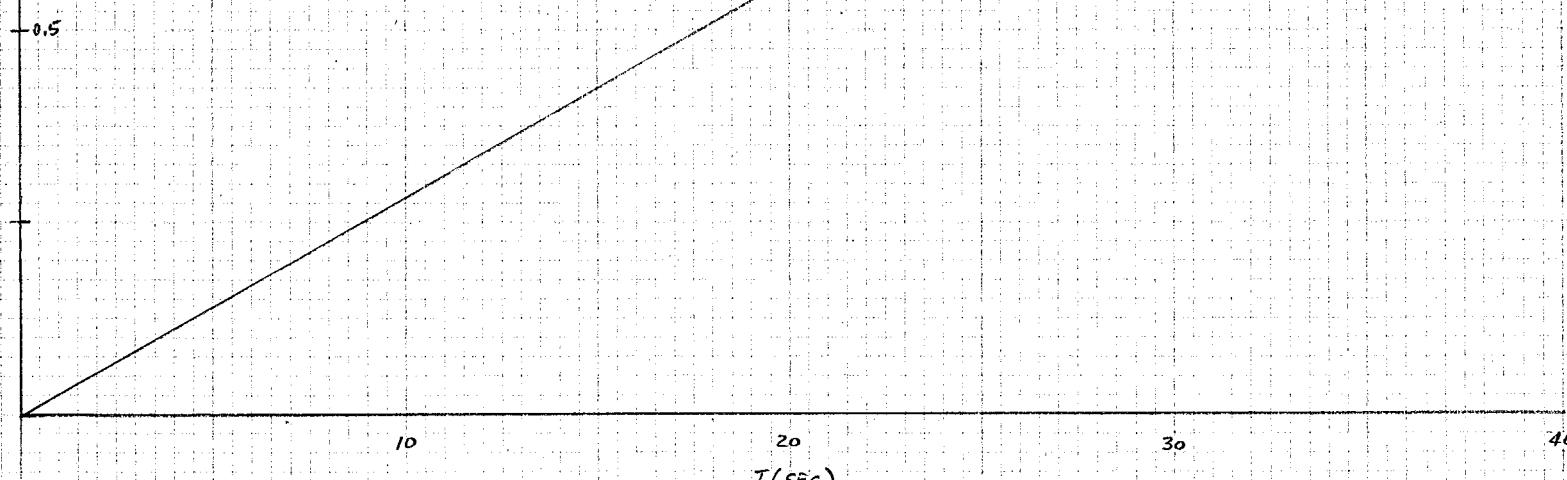
$\omega_B$  vs.  $T$

$$\alpha \text{ and } \beta = \begin{cases} 30^\circ \\ 60^\circ \\ 90^\circ \end{cases}$$

\*



RAD/SEC



$|\omega_A|$  vs.  $T$

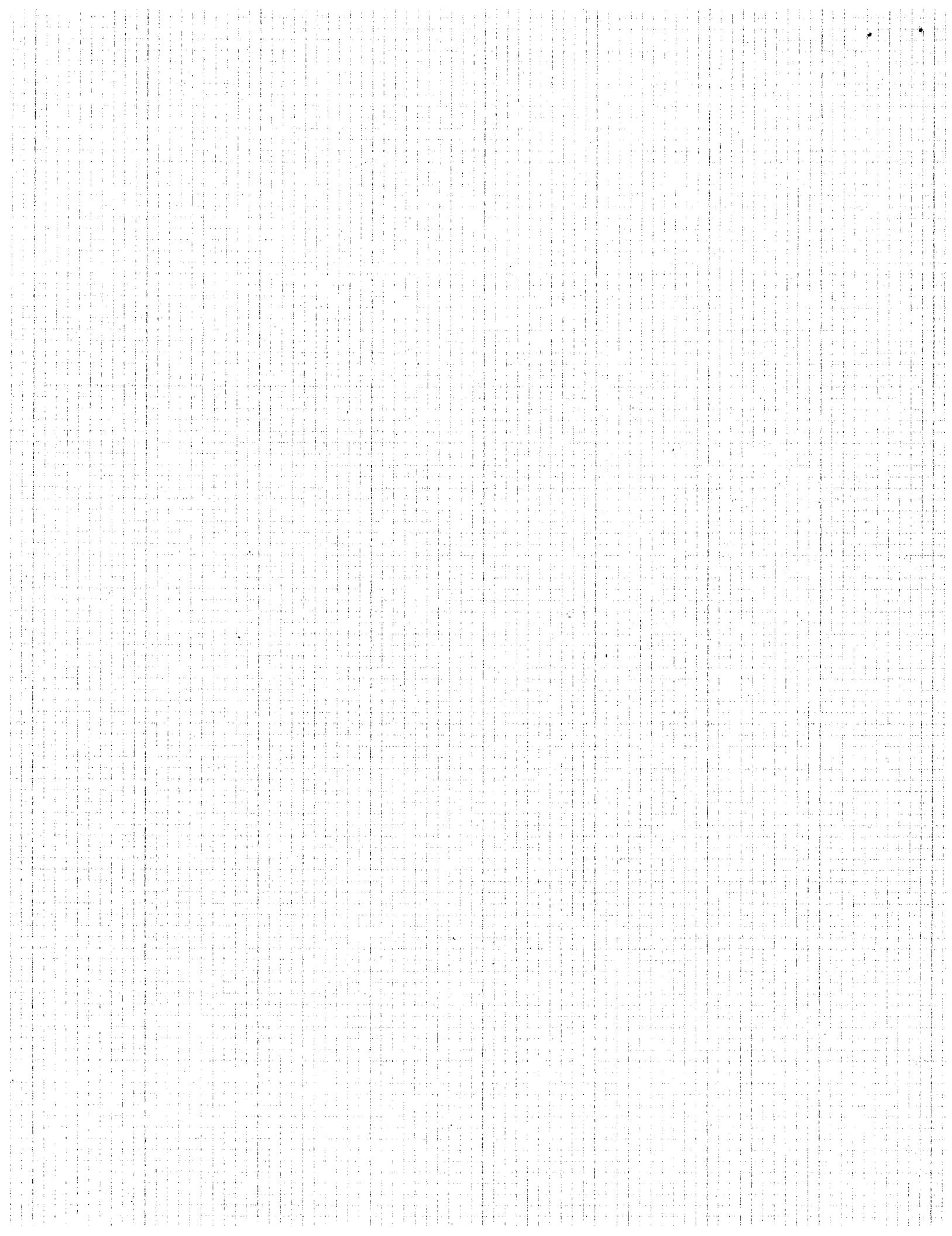
$$\alpha \text{ and } \beta = \begin{cases} 30^\circ \\ 60^\circ \\ 90^\circ \end{cases}$$

\*

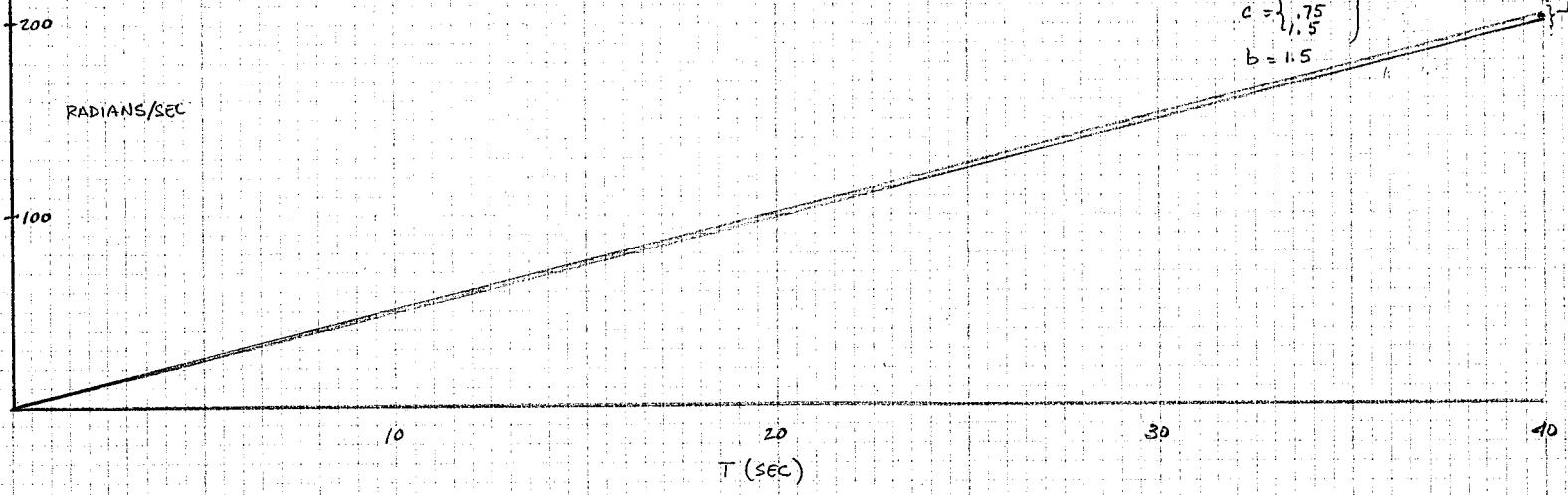
FIGURES (Aa, b)

\* WHEN  $\alpha$  IS VARIED  $\beta$  IS KEPT FIXED AT  $10^\circ$

WHEN  $\beta$  IS VARIED  $\alpha$  IS KEPT FIXED AT  $5^\circ$



FIGURES (5a, b)

 $\omega_B$  vs  $T$ 

$$\mu = .03$$

$$m_b = 10 \text{ kg}$$

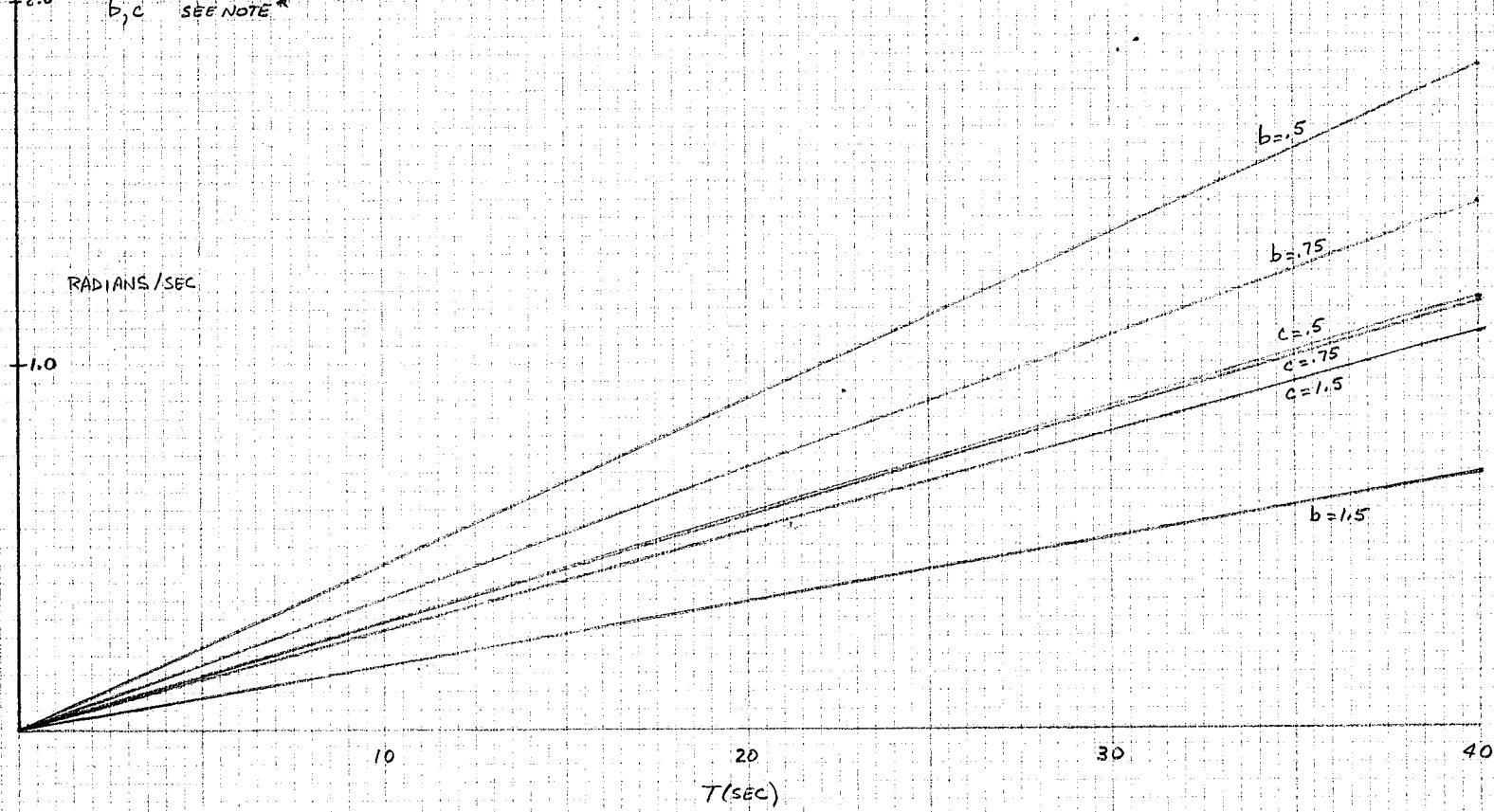
$$T/J = 5 \text{ rads/sec}^2$$

$$\alpha = 5^\circ, \beta = 10^\circ$$

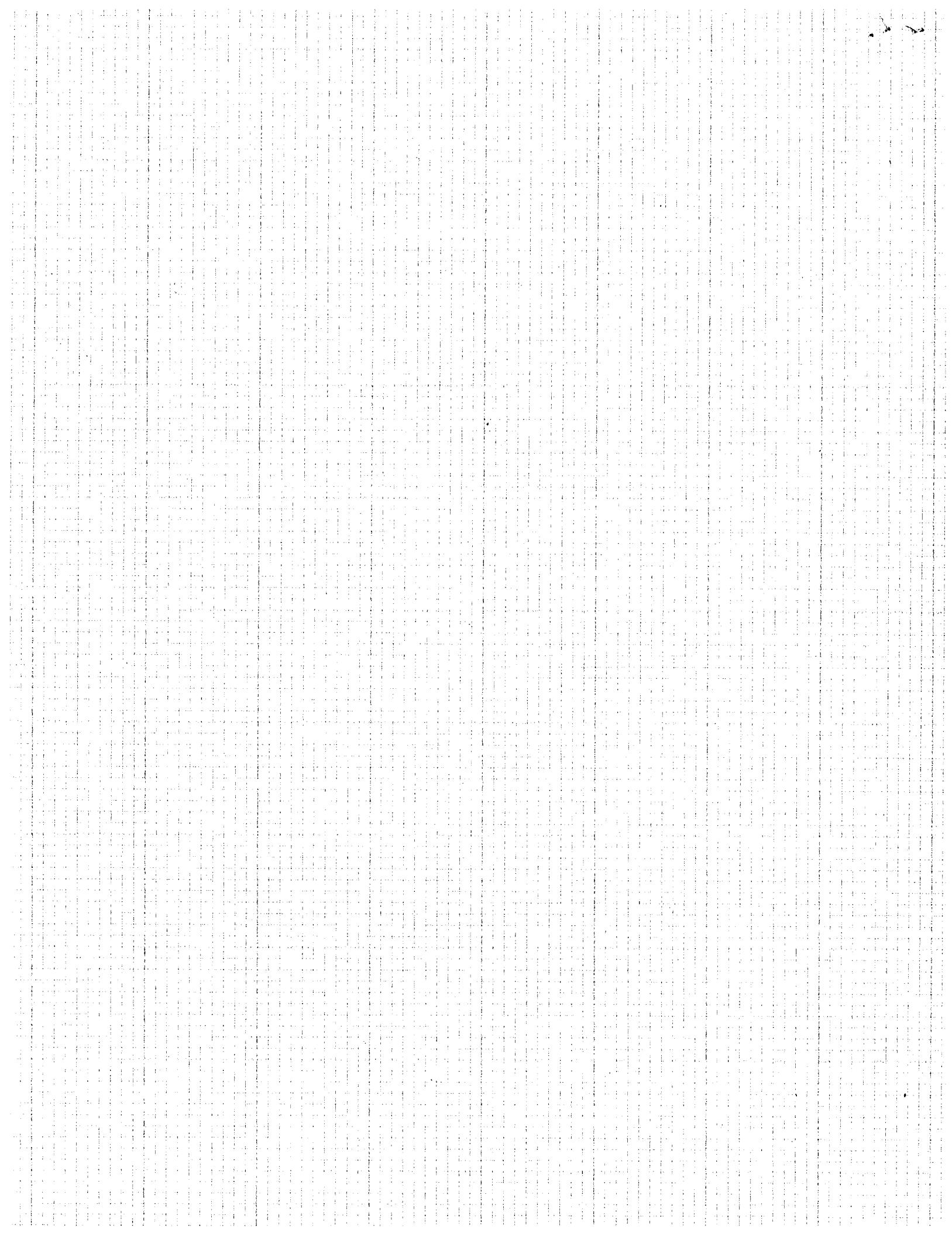
$$d = 0.5$$

$b, c$  SEE NOTE\*

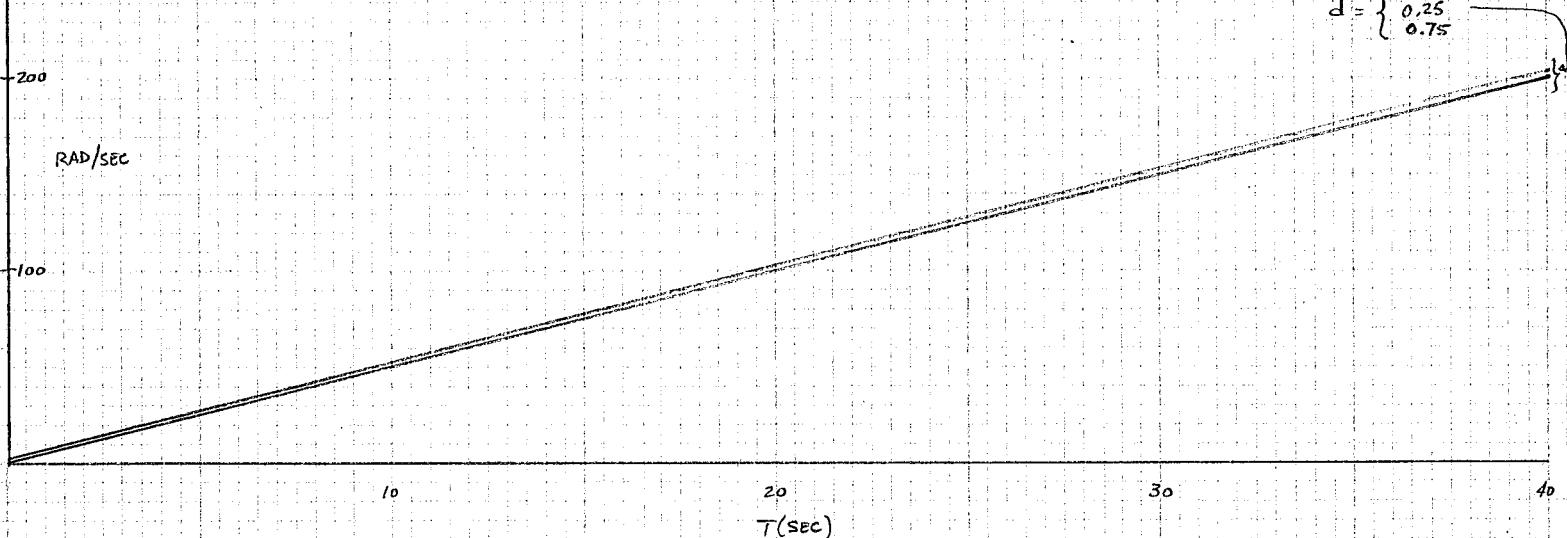
$|r_{WA}|$  vs.  $T$



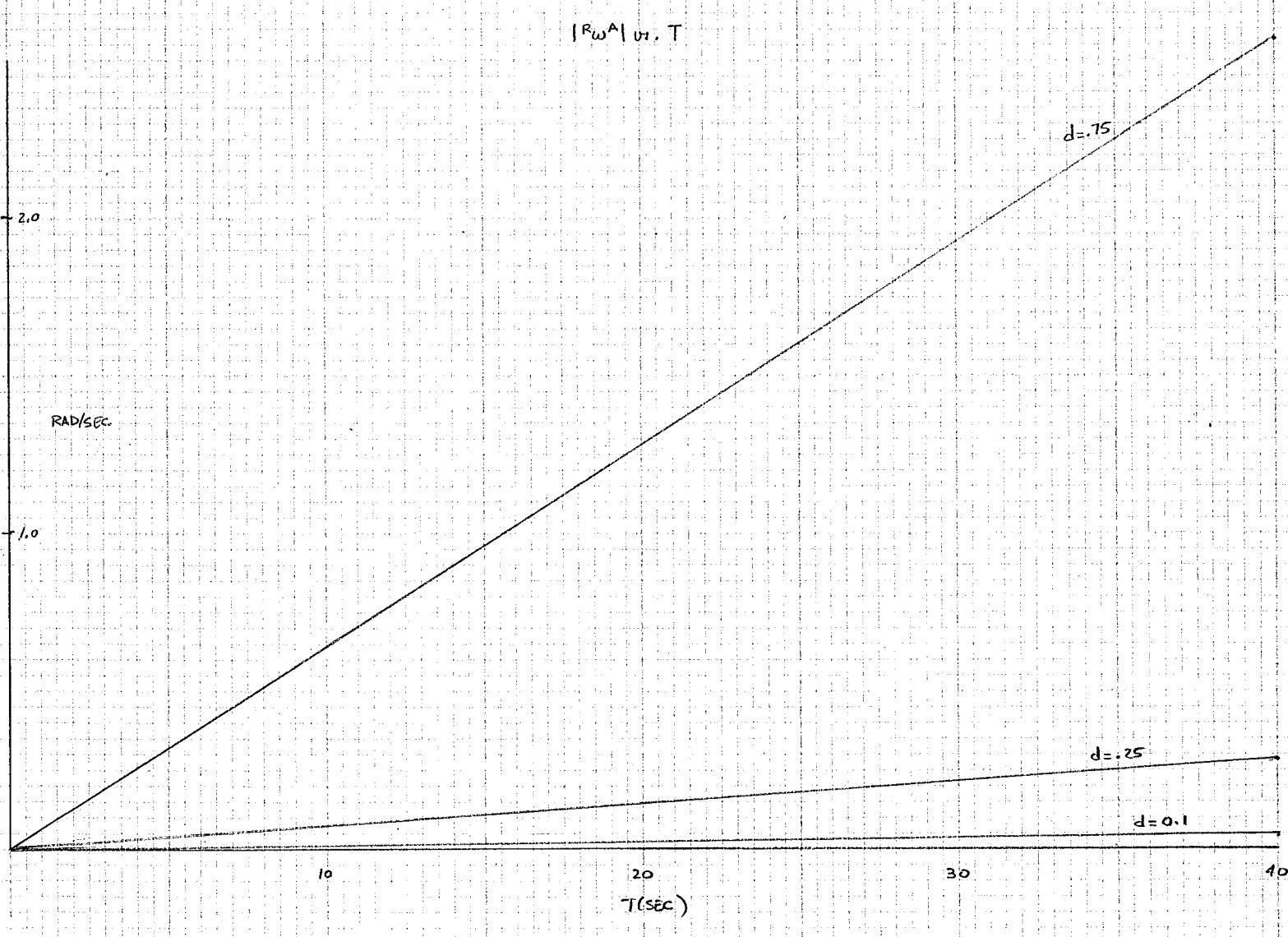
FOR VARIATIONS IN  $b$ ,  $c=1.0$  ALWAYS  
FOR VARIATIONS IN  $c$ ,  $b=1.0$  ALWAYS



FIGURES (6a, b)

 $\omega^B$  vs.  $T$ 

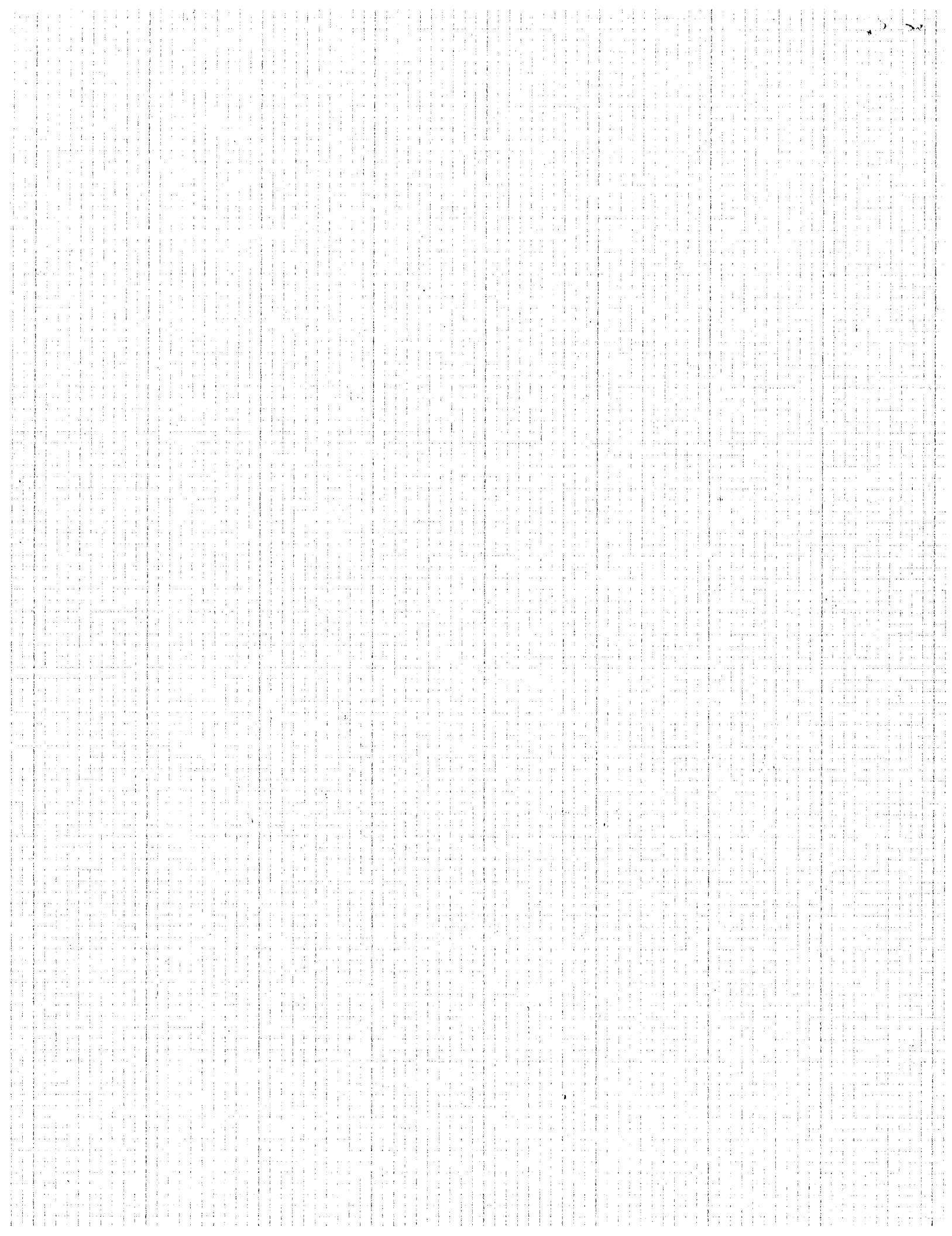
$$d = \{ \begin{array}{l} 0.1 \\ 0.25 \\ 0.75 \end{array}$$

 $|R\omega^A|$  vs.  $T$ 

$$d=0.1$$

$$d=0.25$$

$$d=0.75$$



### 3. Derivation

In deriving equations (4-7) we note that our system has seven degrees of freedom, namely:  $(x, y, z)$  the coordinates of the center of mass of G (also of A and B);  $w_i$  ( $i=1, 2, 3$ ) and  $\hat{w}^B$  as defined previously. We then use Lagrange's Form of D'Alembert's Principle

$$F_r + F_r^* = 0 \quad r=1, \dots, 7 \quad (1)$$

For  $x, y, z$  we may define  $(x, y, z) = (q_1, q_2, q_3)$  coordinates and

$$F_r = V_{q_r}^A \cdot IF^A + V_{q_r}^B \cdot IF^B + \omega_{q_r}^A \cdot IT^A + \omega_{q_r}^B \cdot IT^B \quad (2)$$

where  $V_{q_r}^A$  and  $V_{q_r}^B$  are the partial velocities of the points of application of resultant of all applied forces  $IF^A$  and  $IF^B$  on bodies A and B in R;  $\omega_{q_r}^A$  and  $\omega_{q_r}^B$  are the partial angular velocities of the points of application of the resultant of all applied torques  $IT^A$  and  $IT^B$  on bodies A and B in R.

We may also define

$$F_r^* = V_{q_r}^{A*} \cdot IF^{*A} + V_{q_r}^{B*} \cdot IF^{*B} + \omega_{q_r}^{A*} \cdot IT^{*A} + \omega_{q_r}^{B*} \cdot IT^{*B} \quad (3)$$

where  $V_{q_r}^{A*}$  and  $V_{q_r}^{B*}$  are the partial velocities of the centers of mass of bodies A and B in R;  $IF^{*A} = -m_A \hat{a}_{cm}^{*A}$  and  $IF^{*B} = -m_B \hat{a}_{cm}^{*B}$ ,  $m_A$  and  $m_B$  are the masses of bodies A and B and  $\hat{a}_{cm}^{*}$  is the acceleration of the center of mass;  $IT^{*A}$  and  $IT^{*B}$  are the inertia torques of bodies A and B.

For  $w_i$  ( $i=1, 2, 3$ ) and  $\hat{w}^B$  we may define  $(w, w_2, w_3, \hat{w}^B) = (u_1, u_2, u_3, u_4)$  generalized coordinates;

$$\text{thus } F_r = V_{u_r}^A \cdot IF^A + V_{u_r}^B \cdot IF^B + \omega_{u_r}^A \cdot IT^A + \omega_{u_r}^B \cdot IT^B \quad (4)$$

$$\text{and } F_r^* = V_{u_r}^{A*} \cdot IF^{*A} + V_{u_r}^{B*} \cdot IF^{*B} + \omega_{u_r}^{A*} \cdot IT^{*A} + \omega_{u_r}^{B*} \cdot IT^{*B} \quad (5)$$

where  $(\ )_{u_r}$  are the partial rates with respect to the generalized coordinates.

$$\text{Since } V^A = V^B = V^{*A} = V^{*B} = \dot{x}m_x + \dot{y}m_y + \dot{z}m_z \quad \text{and} \quad (6)$$

$$\hat{a}_{cm}^{*A} = \hat{a}_{cm}^{*B} = \ddot{x}m_x + \ddot{y}m_y + \ddot{z}m_z \quad \text{and because} \quad (7)$$

$$R\omega^A = w_1 a_1 + w_2 a_2 + w_3 a_3 \quad \text{and} \quad R\omega^B = \omega^A + \omega^B = \omega^A + \hat{w}^B \beta \quad (8, 9)$$

we may then write, using  $IT^A = -T\beta$  and  $IT^B = T\beta$ ,  $IF^A = m_A g m_x$ ,  $IF^B = m_B g m_x$

$$F_1 = Mg$$

$$F_5 = 0 \quad F_2^* = -M\ddot{y}$$

$$F_2 = 0$$

$$F_6 = 0 \quad F_3^* = -M\ddot{z}$$

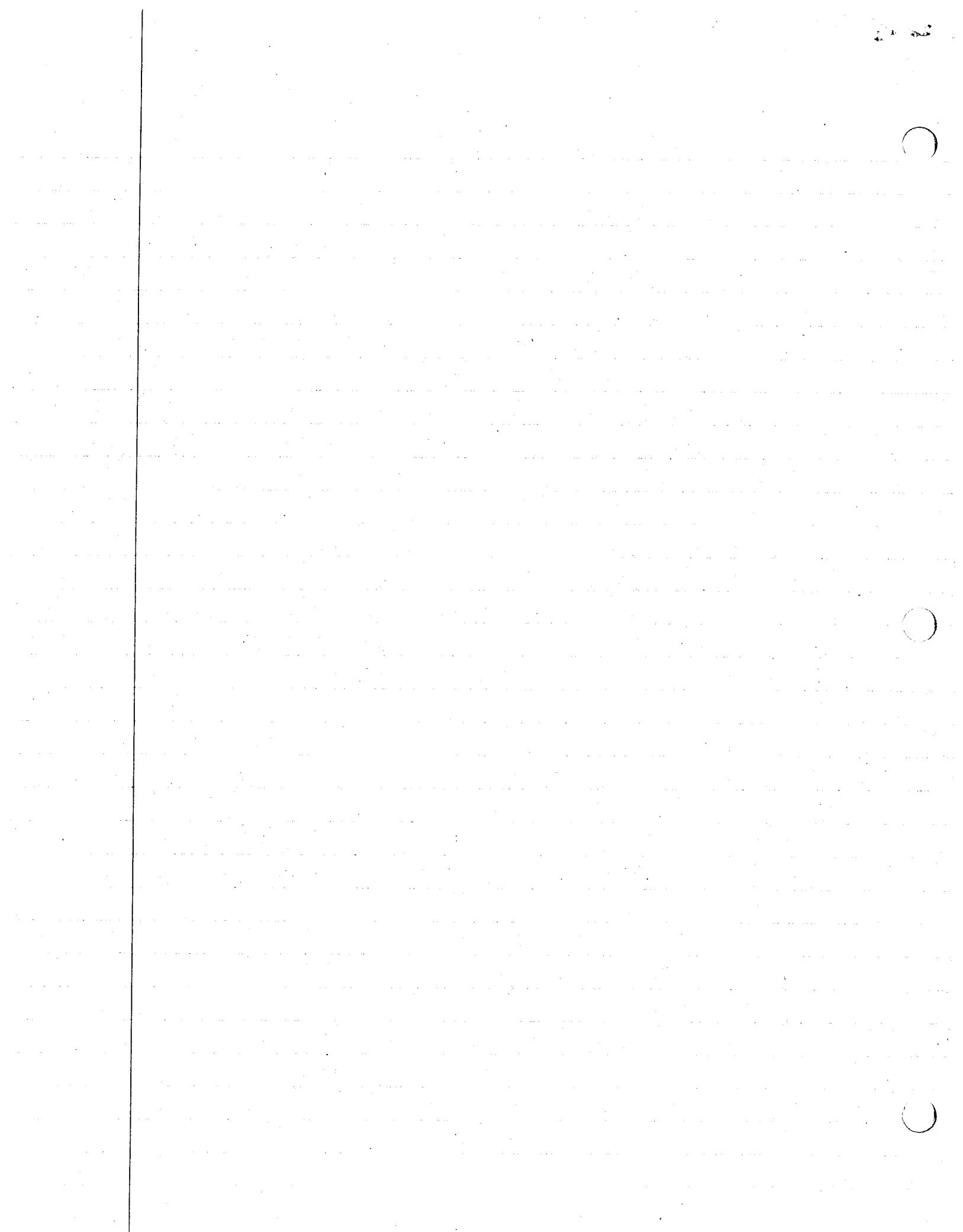
$$F_3 = 0$$

$$F_7 = T$$

$$\text{with } M = m_A + m_B$$

$$F_4 = 0$$

$$F_1^* = -M\ddot{x}$$



Now to obtain  $F_4 - F_7$  we may rewrite (6) as

$$\begin{aligned} F_r^* &= \dot{\omega}_{ur}^A \cdot (I\Gamma^{*A} + I\Gamma^{*B}) + \omega_{ur}^A \cdot (\Gamma^{*A} + \Gamma^{*B}) + \dot{\omega}_{ur}^B \cdot \Gamma^{*B} \\ &= \dot{\omega}_{ur}^A \cdot (\Gamma^{*A} + \Gamma^{*B}) + \dot{\omega}_{ur}^B \cdot \Gamma^{*B} \end{aligned} \quad (19)$$

since  $\dot{\omega}_{ur}^A = 0$

$$\begin{aligned} \text{Now } \Gamma^{*A} + \Gamma^{*B} &= -\frac{d}{dt}(I\Gamma^{*A} + I\Gamma^{*B}) = -\frac{d}{dt}(I\Gamma^{*A}, \omega^A + I\Gamma^{*B}, \omega^B) \\ &= -\frac{d}{dt}([I\Gamma^{*A}, I\Gamma^{*B}], \omega^A + I\Gamma^{*B}, \omega^B) \\ &= (I\Gamma^{*A}, \omega^A) \times \omega^A - I\Gamma^{*A}, \omega^A + (I\Gamma^{*B}, \omega^B) \times \omega^B - I\Gamma^{*B}, \omega^B + (I\Gamma^{*B}, \omega^A) \times \omega^A \end{aligned} \quad (20)$$

Using (8,9).

$$\alpha^A = \dot{\omega}_1 \alpha_1 + \dot{\omega}_2 \alpha_2 + \dot{\omega}_3 \alpha_3 \quad \alpha^B = \dot{\omega}^B \beta \quad (21,22)$$

$$\text{Remembering that } I\Gamma^{*G} = I_1 \alpha_1 \alpha_1 + I_2 \alpha_2 \alpha_2 + I_3 \alpha_3 \alpha_3 \text{ and } I\Gamma^{*B} = J \beta \beta + \dots \quad (23,24)$$

and using these in (19) along with (20) and (8,9) we find

$$F_4^* = [\omega_2 \omega_3 (I_2 - I_3) - I_1 \dot{\omega}_1 - J \dot{\omega}^B \beta_1 + J \dot{\omega}^B (\beta_2 \omega_3 - \beta_3 \omega_2)] \quad (25)$$

$$F_5^* = [\omega_3 \omega_1 (I_3 - I_1) - I_2 \dot{\omega}_2 - J \dot{\omega}^B \beta_2 + J \dot{\omega}^B (\beta_3 \omega_1 - \beta_1 \omega_3)] \quad (26)$$

$$F_6^* = [\omega_1 \omega_2 (I_1 - I_2) - I_3 \dot{\omega}_3 - J \dot{\omega}^B \beta_3 + J \dot{\omega}^B (\beta_1 \omega_2 - \beta_2 \omega_1)] \quad (27)$$

$$F_7^* = -J [\dot{\omega}_1 \beta_1 + \dot{\omega}_2 \beta_2 + \dot{\omega}_3 \beta_3 + \dot{\omega}^B] \quad (28)$$

Using (10-18, 25-28) and (1) we obtain after rearranging

$$\ddot{x} = g \quad (29)$$

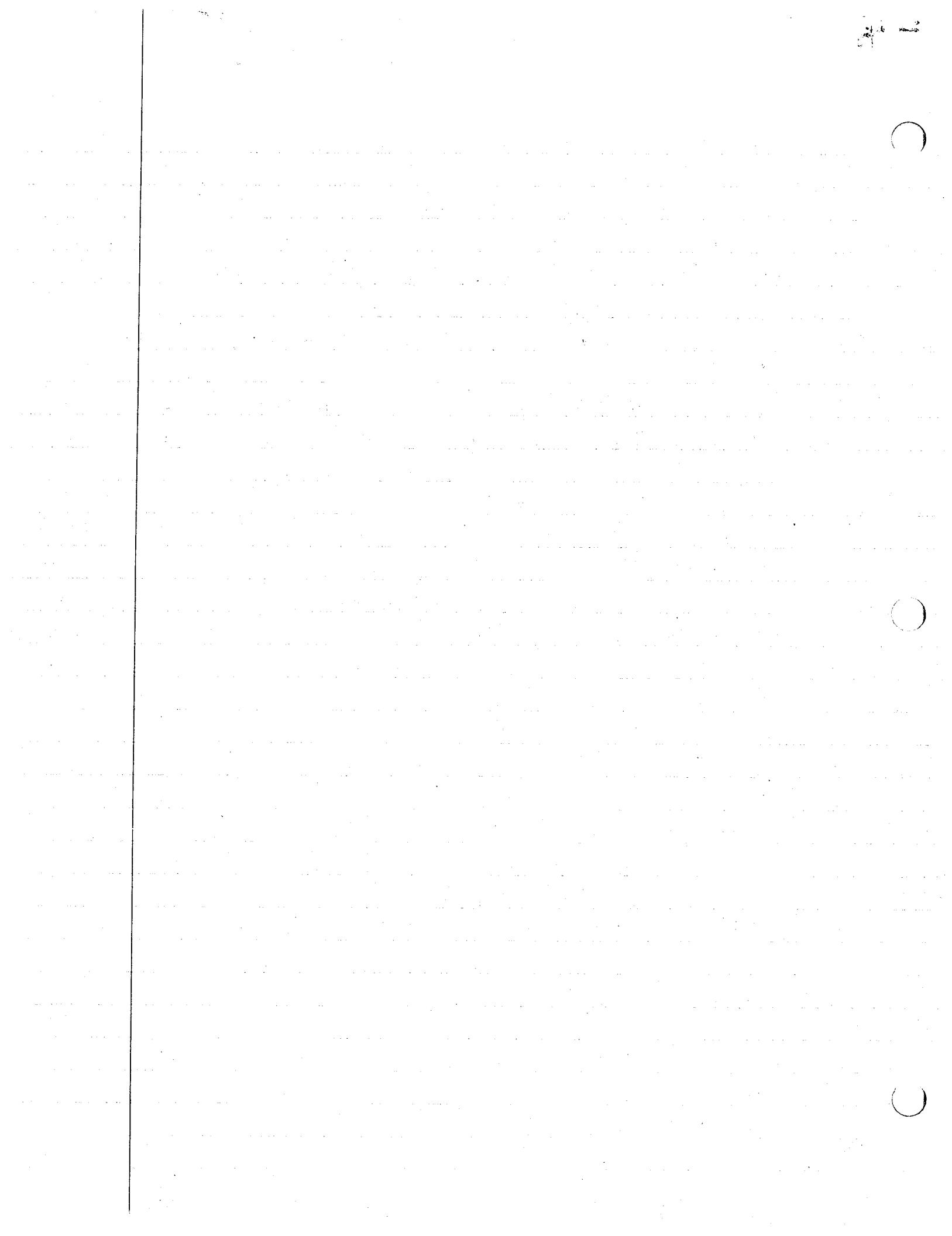
$$\ddot{y} = 0 \quad (30)$$

$$\ddot{z} = 0 \quad (31)$$

and the equations shown in section 1.

$\beta$  is defined by

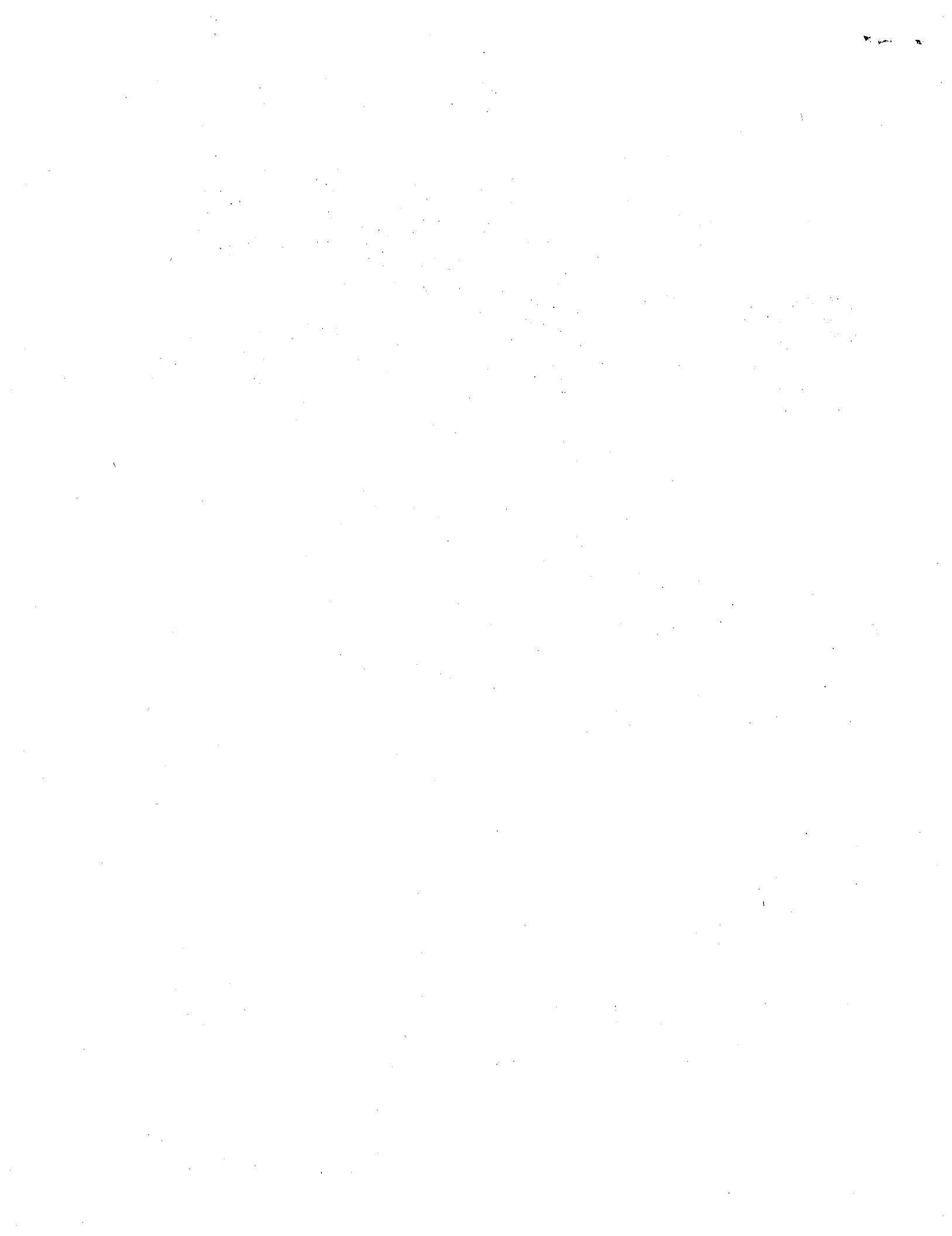
$$\beta = \cos \alpha \alpha_3 + \sin \alpha \sin \beta \alpha_2 + \sin \alpha \cos \beta \alpha_1, \quad (31)$$



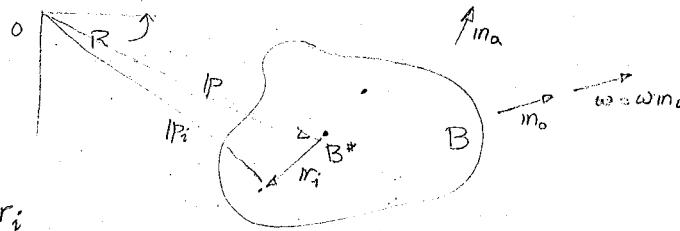
For a set  $S$  of particles  $P_1, \dots, P_N$ , a quantity  $G$ , known as a Gibbs function, is defined as

$$G \triangleq \frac{1}{2} \sum_{i=1}^N m_i \underline{a}_i^2$$

where  $m_i$  and  $\underline{a}_i$  are, respectively, the mass and the acceleration of  $P_i$ . Regarding a rigid body  $B$  as a set  $S$  of a large number of particles, express  $G$  as a function of the acceleration  $\underline{a}$  of the mass center of  $B$ , the angular velocity  $\underline{\omega}$  of  $B$ , the angular acceleration  $\underline{\alpha}$  of  $B$ , the mass  $m$  of  $B$ , and the central inertia dyadic  $\underline{I}$  of  $B$ .



$$\alpha = \alpha_{in_a}$$



$$G \triangleq \frac{1}{2} \sum_{i=1}^N m_i (\alpha_i \cdot \alpha_i)$$

$$\text{Let } ||P_i|| = ||P + m_i r_i||$$

$$v_i = v^* + \frac{R_B}{\omega} \times \omega \times m_i r_i$$

$$\alpha_i = \alpha^* + \alpha \times m_i r_i + \frac{R_B}{\omega} \times (\omega \times m_i r_i)$$

$$\alpha_i \cdot \alpha_i = \alpha^* \cdot \alpha^* + 2\alpha^* \cdot \alpha \times m_i r_i + 2\alpha^* \cdot \frac{R_B}{\omega} \times (\omega \times m_i r_i) + \dots + (\alpha \times m_i r_i)^2$$

$$+ 2(\alpha \times m_i r_i) \cdot [\frac{R_B}{\omega} \times (\omega \times m_i r_i)] + [\frac{R_B}{\omega} \times (\omega \times m_i r_i)]^2$$

$$\text{now } \sum_{i=1}^N m_i (\alpha_i \cdot \alpha_i) = \left( \sum_{i=1}^N m_i \alpha^* \cdot \alpha^* = m \alpha^*{}^2 \right) + \left( \sum m_i \alpha^* \cdot \alpha \times m_i r_i = \sum m_i r_i \cdot (\alpha \times \alpha) = 0 \right)$$

$$+ 2 \left( \sum m_i \alpha^* \cdot \frac{R_B}{\omega} \times (\omega \times m_i r_i) = \alpha^* \cdot \frac{R_B}{\omega} \times (\frac{R_B}{\omega} \times \sum m_i r_i) = 0 \right)$$

$$+ \left( \sum m_i (\alpha \times m_i r_i) \cdot (\alpha \times m_i r_i) = \alpha^2 m_i \sum m_i r_i \times (m_i \times m_i r_i) = \alpha^2 m_i \cdot I_{\alpha}^{B/B^*} \right)$$

$$+ 2 \left( \sum m_i (\alpha \times m_i r_i) \cdot [\frac{R_B}{\omega} \times (\omega \times m_i r_i)] = \alpha \omega^2 \sum m_i (m_i \times m_i r_i) \cdot [m_i \times (m_i \times m_i r_i)] = \right.$$

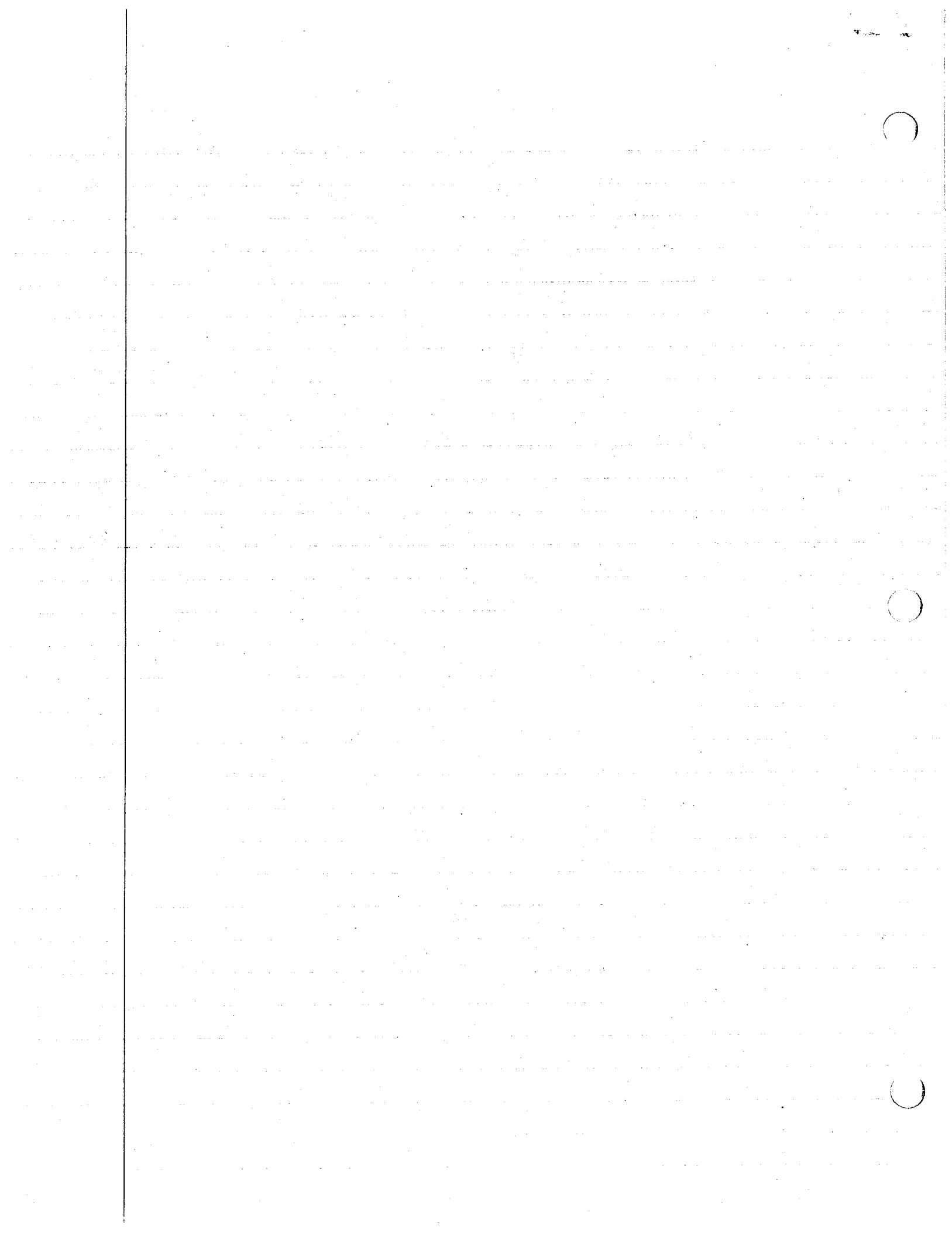
$$\left. \alpha \omega^2 \sum m_i [r_i \times (m_i \times (m_i \times m_i r_i))] \cdot m_i = \alpha \omega^2 m_i \cdot m_i \times [\sum m_i r_i \times (m_i \times m_i r_i)] = \right.$$

$$\left. \omega \cdot [\alpha \cdot \{\frac{R_B}{\omega} \times I_{\alpha}^{B/B^*}\}] + \left( \sum m_i [\frac{R_B}{\omega} \times (\frac{R_B}{\omega} \times m_i r_i)] \right)^2 = \sum m_i \left[ \frac{R_B}{\omega} \right]^2 [\omega \times m_i r_i]^2 - [\omega \cdot (\omega \times m_i r_i)] \right)$$

$$= \sum m_i \frac{R_B}{\omega}^2 [r_i \times (\omega \times m_i r_i)] \cdot \omega = (\frac{R_B}{\omega})^2 \omega \cdot \sum m_i r_i \times (\omega \times m_i r_i) = (\frac{R_B}{\omega})^2 \omega \cdot \omega \cdot I_{\alpha}^{B/B^*}$$

$$\text{thus } G \triangleq \frac{1}{2} \left[ m \alpha^*{}^2 + \alpha \cdot I_{\alpha}^{B/B^*} + 2 \alpha \cdot \left\{ \frac{R_B}{\omega} \times \left[ \frac{R_B}{\omega} \cdot I_{\alpha}^{B/B^*} \right] \right\} + \omega^2 (\omega \cdot I_{\alpha}^{B/B^*} \cdot \omega) \right]$$

$$G \triangleq \frac{1}{2} \left[ m \alpha^*{}^2 + \alpha \cdot I_{\alpha}^{B/B^*} + 2 \omega \cdot \left[ (I_{\alpha}^{B/B^*} \cdot \omega) \times \alpha \right] + \omega^2 \omega \cdot I_{\alpha}^{B/B^*} \cdot \omega \right]$$



6a. let P+Q in 3(j) has equal weight W find  $F_{\theta_i}^*$  assoc. with  $\theta_i$

$$F_{\theta_i}^* = \sum_{i=1}^N \tilde{W}_{\theta_i}^{P_i}$$

If we assume that the joints are smooth then



$$\text{From previous } \tilde{W}_{\theta_i}^{P_i} = L \ddot{\theta}_i \quad \tilde{W}_{\theta_i}^Q = L \ddot{\theta}_i \frac{\sin(\theta_3 - \theta_1)}{\sin(\theta_3 - \theta_2)} = L \frac{\sin(\theta_3 - \theta_1)}{\sin(\theta_3 - \theta_2)} [\cos \theta_2 m_y - \sin \theta_2 m_x]$$

$$\text{Also } \dot{W}^P = \dot{\theta}_1 L \ddot{\theta}_1 \quad \text{and } \dot{W}^Q = \dot{\theta}_1 L \ddot{\theta}_1 + \dot{\theta}_3 [-3L \sin \theta_3 m_x + 3L \cos \theta_3 m_y] = L \ddot{\theta}_1 \frac{\sin(\theta_3 - \theta_1)}{\sin(\theta_3 - \theta_2)}$$

$$\ddot{a}^P = \frac{d \dot{W}^P}{dt} = \ddot{\theta}_1 L \ddot{\theta}_1 + \dot{\theta}_1 L [-8 \sin \theta_1 \dot{\theta}_1 m_y - \cos \theta_1 \dot{\theta}_1 m_x] = \ddot{\theta}_1 L \ddot{\theta}_1 - \dot{\theta}_1^2 L \ddot{\theta}_1$$

$$\text{thus } [\tilde{W}_{\theta_i}^P \cdot \ddot{a}^P] = L \ddot{\theta}_1 ($$

$$\ddot{a}^Q = \frac{d \dot{W}^Q}{dt} = \frac{d}{dt} (2 \ddot{\theta}_2 L \ddot{\theta}_2) = 2 \ddot{\theta}_2^2 L \ddot{\theta}_2 + 2 \dot{\theta}_2 L \frac{d \ddot{\theta}_2}{dt} + 2 \dot{\theta}_2 L \frac{d \dot{\theta}_2}{dt} = 2 \ddot{\theta}_2^2 L \ddot{\theta}_2 - 2 \dot{\theta}_2^2 L \ddot{\theta}_2$$

$$\tilde{W}_{\theta_i}^P \cdot \ddot{a}^Q = L \ddot{\theta}_2 \frac{\sin(\theta_3 - \theta_1)}{\sin(\theta_3 - \theta_2)} \cdot 2 \dot{\theta}_2 L \ddot{\theta}_2 = 2 \dot{\theta}_2 L^2 \frac{\sin(\theta_3 - \theta_1)}{\sin(\theta_3 - \theta_2)}$$

$$\text{thus } F_{\theta_i}^* = -m [L^2 \ddot{\theta}_1 + 2L^2 \ddot{\theta}_2 \frac{\sin(\theta_3 - \theta_1)}{\sin(\theta_3 - \theta_2)}] = -\frac{WL^2}{g} [\ddot{\theta}_1 + 2\dot{\theta}_2 \frac{\sin(\theta_3 - \theta_1)}{\sin(\theta_3 - \theta_2)}]$$

6b. Example on pg 107  $\mathbb{I}_a^{Pla}$  is not  $\parallel m_a$

also in most cases  $\|p_i = a_i m_a + b_i m_b$  when  $a_i = m_a / p_i$   $b_i = m_b / p_i$   $m_a, m_b \neq 0$

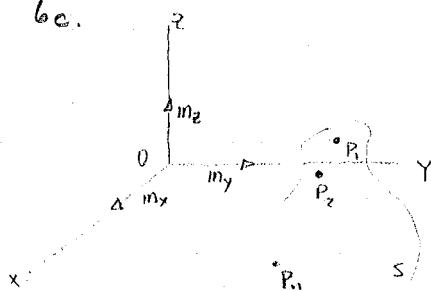
$$\therefore \max \|p_i = b_i \max m_b \quad \text{let } \max m_b = m_c$$

$$\text{now } \mathbb{I}_a = \sum_{i=1}^n m_i [a_i m_a + b_i m_b] \times b_i m_c$$

$$= \sum_{i=1}^n m_i a_i b_i (-m_b) + \sum_{i=1}^n m_i b_i^2 m_a$$

in most case  $\sum_{i=1}^n m_i a_i b_i \neq 0 \quad \therefore \mathbb{I}_a \text{ will not be parallel to } m_a$

6c.



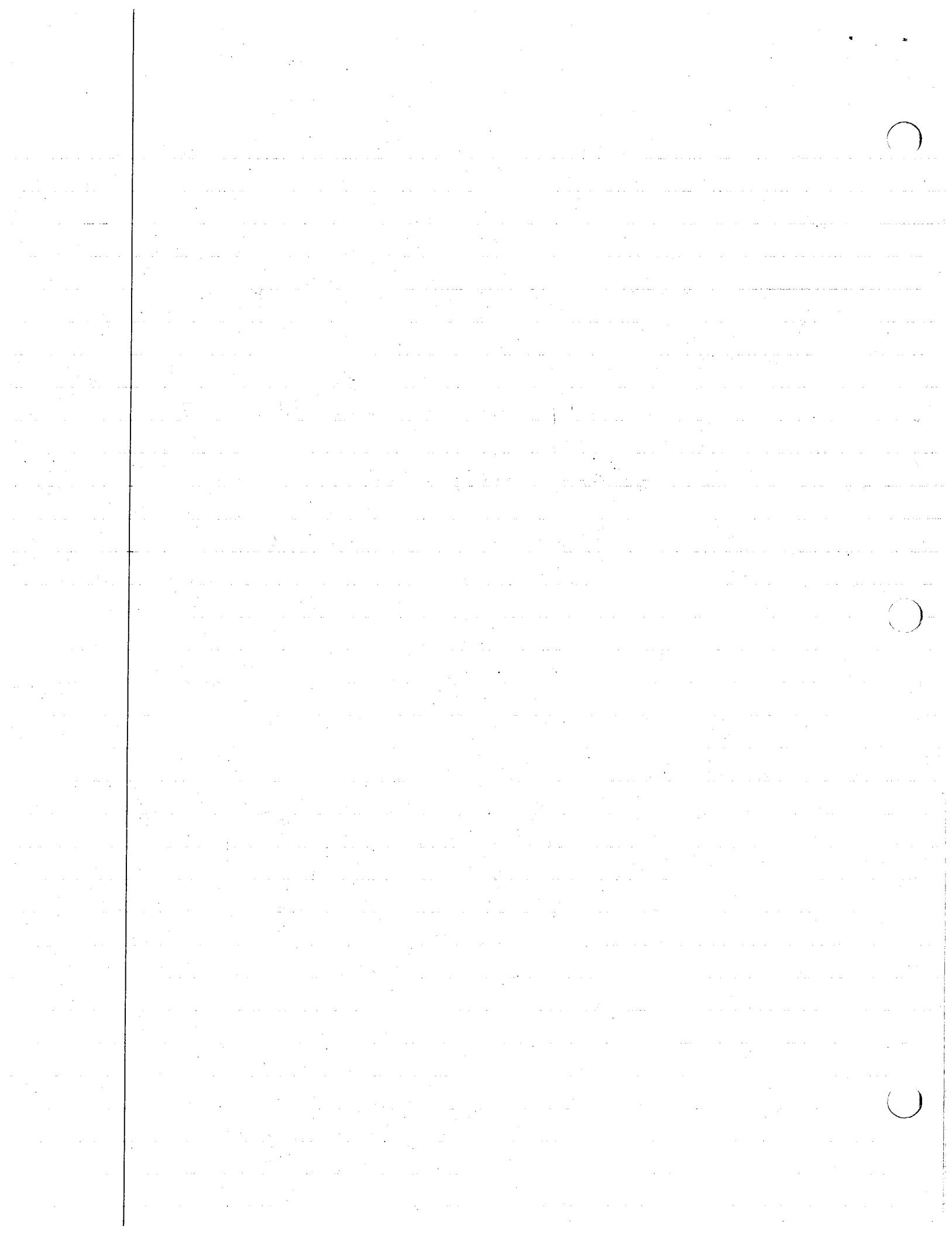
$$\mathbb{I}_x = \sum m_i \|p_i \times (m_x \times \|p_i)\)$$

$$\text{Let } \|p_i = x_i m_x + y_i m_y + z_i m_z$$

$$m_x \times \|p_i = y_i m_z - z_i m_y$$

$$\begin{aligned} \mathbb{I}_x &= \sum m_i (x_i m_x + y_i m_y + z_i m_z) \times (y_i m_z - z_i m_y) \\ &= \sum m_i [-x_i y_i m_y - x_i z_i m_z + y_i^2 m_x + z_i^2 m_x] \end{aligned}$$

$$\mathbb{I}_x = \mathbb{I}_x m_x = \sum m_i (y_i^2 + z_i^2)$$



$$I_y = \sum m_i p_i \times (m_y \times p_i)$$

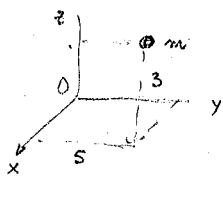
$$(m_y \times p_i) = -x_i m_z + z_i m_x$$

$$p_i \times (-x_i m_z + z_i m_x) = +x_i^2 m_y - y_i x_i m_x - y_i z_i m_z + z_i^2 m_y$$

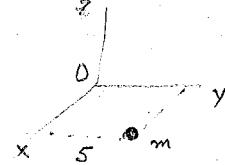
$$I_{yz} = I_y \cdot m_z = -\sum m_i y_i z_i$$

- RHS or LHS doesn't matter since you're taking the double cross product. This produces two minus signs which cancel each other.
- For non mutually  $\perp$  coordinate systems we will find that results are not as simple as those shown, because we will get components that involve the sines of the angles between these vectors. In a  $\perp$  system the sines are  $\approx 0$  or  $1$ .
- The minus sign comes in from the definition of  $I_{yz} = I_y \cdot m_z$

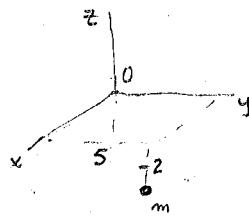
6(d)



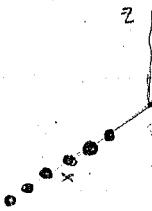
$$1 \text{ particle } I_{yz} = -m \cdot 15 = -15m < 0$$



$$1 \text{ particle } I_{yz} = -m \cdot 5 \cdot 0 = 0$$

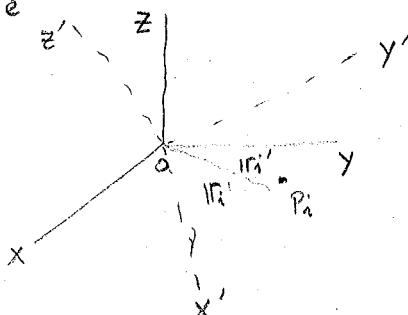


$$I_{yz} = m(5)(-z) = 10m > 0$$



$$\begin{aligned} I_x &= \sum_{i=1}^6 m_i (z_i^2 + y_i^2) = 0 \\ &= M k^2 = 0 \\ k &= 0 \quad M = \sum_{i=1}^6 m_i \end{aligned}$$

6e

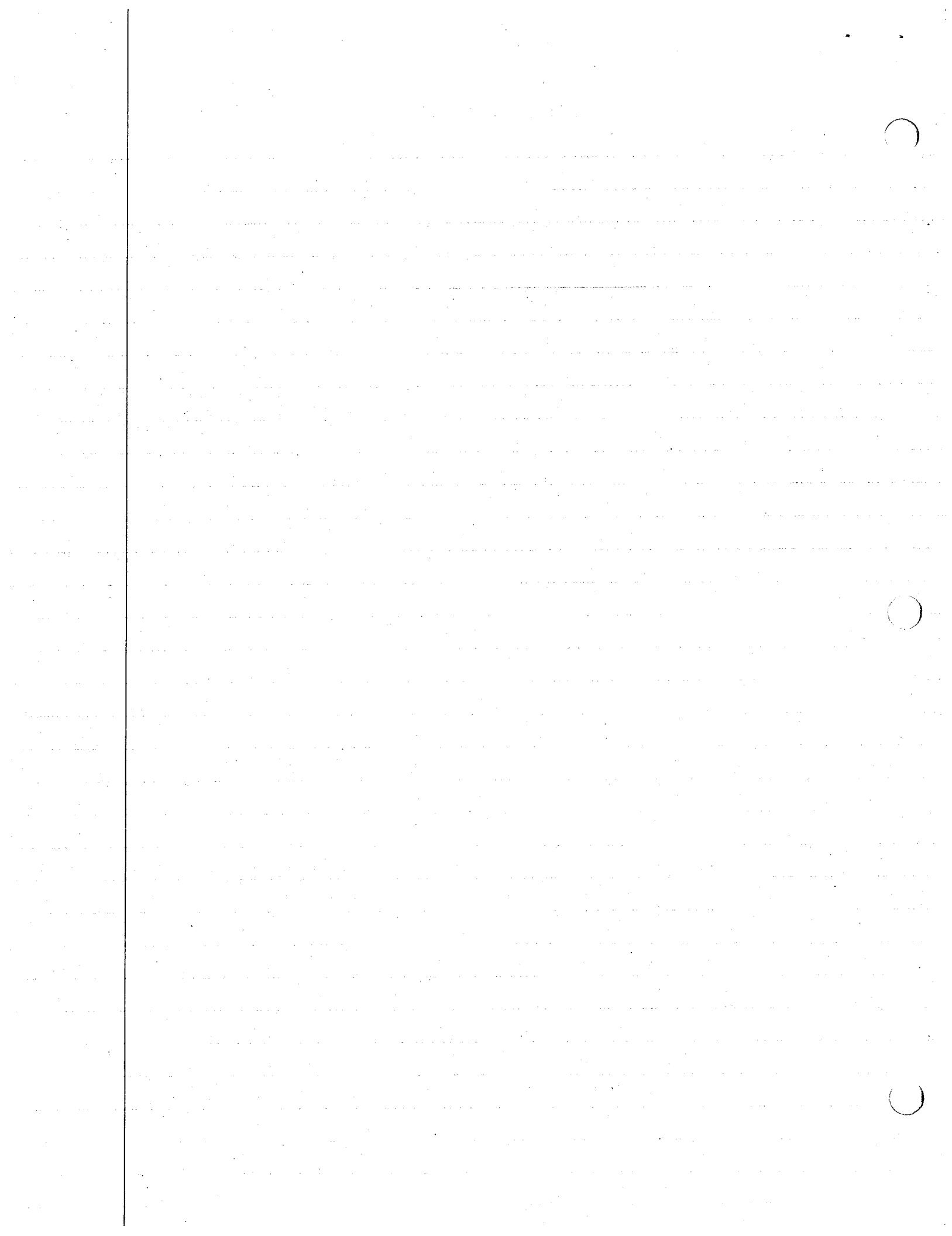


$$I_x = A \quad I_y = B \quad I_z = C$$

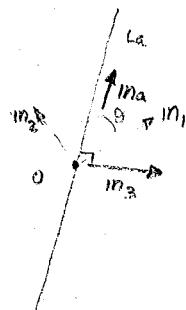
$$I_{x'} = A' \quad I_{y'} = B' \quad I_{z'} = C'$$

$$\text{now } I_x + I_y + I_z = \sum_{i=1}^N 2m_i (x_i^2 + y_i^2 + z_i^2) = \sum 2m_i r_i \cdot r_i$$

but  $x_i^2 + y_i^2 + z_i^2 =$  the distance squared from  $O$  to the point  $P_i$ . Distances are the same under transformation hence  $I_{x'} + I_{y'} + I_{z'} = \sum 2m_i r'_i \cdot r'_i = \sum 2m_i r_i \cdot r_i$   
thus  $I_x + I_y + I_z = I_{x'} + I_{y'} + I_{z'}$



6f.



Since  $m_3$  is  $\perp$  to  $l_a$  then  $l_a$  must lie in the plane

$$\text{of } m_1 \times m_2 \text{ hence } m_a = (m_a \cdot m_1)m_1 + (m_a \cdot m_2)m_2$$

$$I_a = \sum_{j=1}^3 \sum_{k=1}^3 m_j I_{jk} m_k \quad m_a \cdot m_3 = 0$$

$$m_a \cdot l_a = I_a = \sum_{j=1}^3 \sum_{k=1}^3 m_a \cdot m_j I_{jk} m_k \cdot m_a \quad m_a = \frac{m_1 \cdot m_2}{\cos \theta \sin \theta}$$

$$\text{now } I_a = I_1 \cos^2 \theta + I_2 \sin^2 \theta + I_{12} \sin 2\theta$$

$$\frac{\partial I_a}{\partial \theta} = 0 \Rightarrow -I_1 \cdot 2 \cos \theta \sin \theta + 2 I_2 \sin \theta \cos \theta + 2 I_{12} \cos 2\theta = 0$$

$$-(-I_2 + I_1) \sin 2\theta + 2 I_{12} \cos 2\theta = 0 \text{ or } \tan 2\theta = \frac{2 I_{12}}{I_1 - I_2}$$

$$\frac{2 I_{12}}{I_1 - I_2}$$

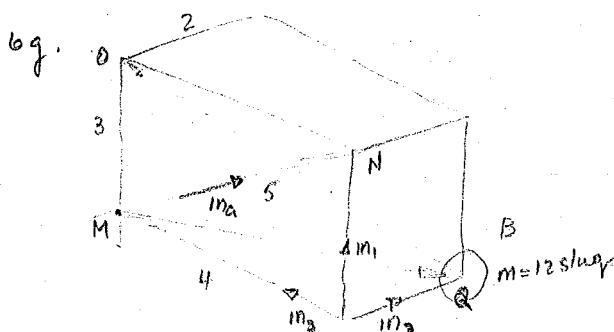
$$I_a = I_1 \left( \frac{\cos 2\theta + 1}{2} \right) + I_2 \left( \frac{1 - \cos 2\theta}{2} \right) + I_{12} \sin 2\theta \quad \frac{I_1 - I_2}{I_1 - I_2}$$

$$= \frac{I_1 + I_2}{2} + \frac{I_1 - I_2}{2} \cdot \frac{I_1 - I_2}{\sqrt{I_1 - I_2}} + 2 \frac{I_{12}^2}{\sqrt{I_1 - I_2}}$$

$$\frac{(I_1 - I_2)^2}{2\sqrt{I_1 - I_2}} + \frac{4 I_{12}^2}{2\sqrt{I_1 - I_2}} = \left( \pm \left[ \quad \right] \right)^2$$

$$\sqrt{=} = \sqrt{(I_1 - I_2)^2 + 4 I_{12}^2}$$

$$I_a = \frac{I_1 + I_2}{2} \pm \left[ \left( \frac{I_1 - I_2}{2} \right)^2 + I_{12}^2 \right]^{\frac{1}{2}} \quad \begin{array}{l} \text{given max} \\ \text{given min} \end{array}$$



$I_{jk}^{B/M}$	1	2	3
1	260	72	-144
2	72	32.5	96
3	-144	96	169

$$\text{find } I_a^{B/M} = m_a I_a \cdot m_a$$

$$m_a = (3m_1 - 4m_3) \frac{1}{5}$$

$$I_a^{B/M} = \sum_{j=1}^3 \sum_{k=1}^3 m_j I_{jk}^{B/M} m_k \quad m_a \cdot m_1 = \frac{3}{5}, \quad m_a \cdot m_2 = 0, \quad m_a \cdot m_3 = -\frac{4}{5}$$

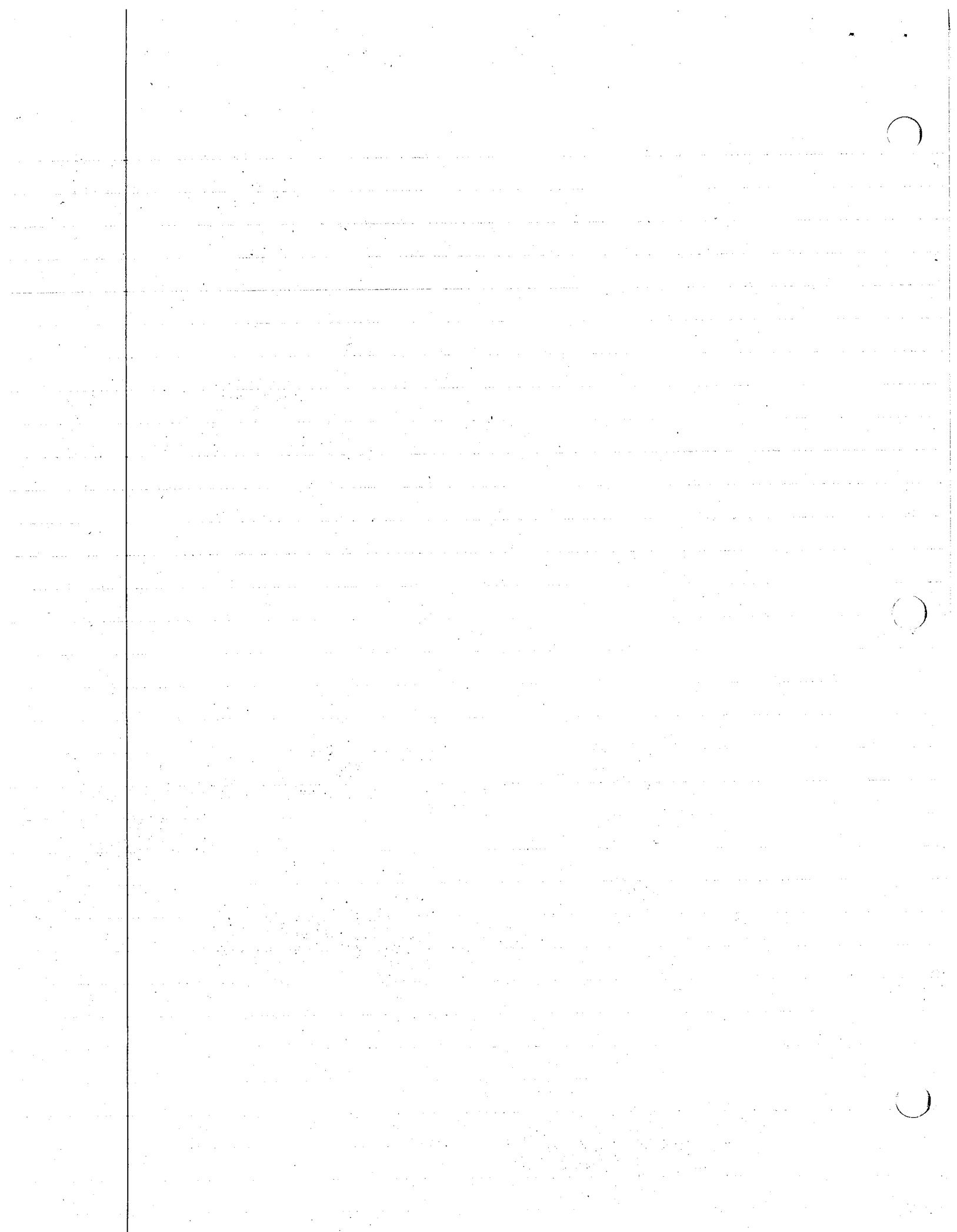
$$I_a^{B/M} = \frac{3}{5} \left[ I_{11} \cdot \frac{3}{5} + I_{12} \cdot 0 + I_{13} \cdot -\frac{4}{5} \right] + 0 \left[ \quad \right] - \frac{4}{5} \left[ I_{31} \cdot \frac{3}{5} + I_{33} \cdot -\frac{4}{5} \right]$$

$$= \frac{9}{25} I_{11} - \frac{12}{25} I_{13} - \frac{12}{25} I_{31} + \frac{16}{25} I_{33}$$

$$I_a^{B/M} = \frac{1}{25} \left[ 9 \cdot 260 + 24 \cdot 144 + 169 \cdot 16 \right] = \frac{8500}{25}$$

$$I_a^{B/M} = I_a^{B/A} + I_a^{Q/M} \quad I_a^{Q/M} = m_a \cdot [P^{Q/M} \times (m_a \times P^{Q/M})] m$$

$$I_a^{B/A} = I_a^{B/M} + I_a^{Q/A} ; \quad I_a^{Q/A} = I_a^{Q/M} \cdot m_a = m_a \cdot m \cdot [P^{Q/A} \times (m_a \times P^{Q/A})]$$



$$\begin{vmatrix} m_1 & m_2 & m_3 \\ \frac{3}{5} & 0 & -\frac{4}{5} \\ 0 & 2 & 4 \end{vmatrix}$$

$$\begin{aligned} P^{Q/A} &= -2m_2 + 3m_1 + 4m_3 & m_a \times P^{Q/A} &= -\frac{6}{5}m_3 - \frac{12}{5}m_2 - \frac{8}{5}m_1 - 12m_2 = \left[ +\frac{6}{5}m_3 + \frac{24}{5}m_2 + \frac{8}{5}m_1 \right] \\ &= P^{Q/A} \times \left[ +\frac{8}{5}m_1 + \frac{24}{5}m_2 + \frac{6}{5}m_3 \right] & \begin{vmatrix} m_1 & m_2 & m_3 \\ 0 & 2 & 4 \\ +\frac{8}{5} & +\frac{24}{5} & +\frac{6}{5} \end{vmatrix} &= +\frac{12}{5}m_1 - \frac{32}{5}m_2 - \frac{72}{5}m_3 = \frac{16}{5}m_3 + \frac{12}{5}m_2 + \frac{96}{5}m_1 \\ &\therefore m_a \cdot [P^{Q/A} \times (m_a \times P^{Q/A})] & &= \frac{108}{5}m_1 - \frac{14}{5}m_2 - \frac{88}{5}m_3 \\ &= \frac{3m_1 - 4m_3}{5} \cdot \left( \frac{108}{5}m_1 - \frac{14}{5}m_2 - \frac{88}{5}m_3 \right) \end{aligned}$$

$$\frac{324}{25} + \frac{352}{25} = \frac{676}{25}$$

$$\therefore I_a^{Q/A} = m \cdot \frac{676}{25}$$

$$I_a^{Q/A} = \frac{676m}{25}$$

$$P^{Q/M} = (2m_2 - 4m_3)$$

$$m_a \times P^{Q/M} = \begin{vmatrix} m_1 & m_2 & m_3 \\ \frac{3}{5} & 0 & -\frac{4}{5} \\ 0 & 2 & 4 \end{vmatrix} = \frac{6}{5}m_3 + \frac{12}{5}m_2 + \frac{8}{5}m_1$$

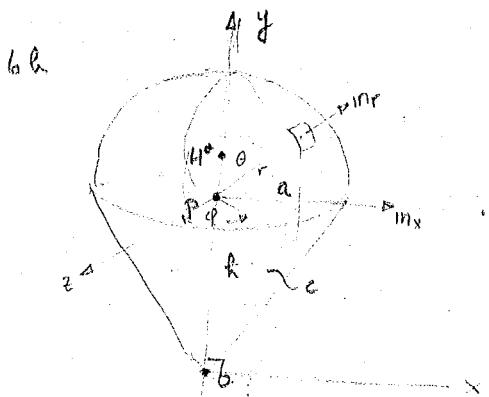
$$P^{Q/M} \times [m_a \times P^{Q/M}] = \begin{vmatrix} m_1 & m_2 & m_3 \\ 0 & 2 & -4 \\ +\frac{8}{5} & +\frac{12}{5} & +\frac{6}{5} \end{vmatrix} = \frac{12}{5}m_1 - \frac{32}{5}m_2 - \frac{16}{5}m_3 + \frac{48}{5}m_1 = \frac{60}{5}m_1 - \frac{32}{5}m_2 - \frac{16}{5}m_3$$

$$m_a \cdot [P^{Q/M} \times (m_a \times P^{Q/M})] = \frac{(3m_1 - 4m_3)}{5} \cdot \left( \frac{60m_1}{5} - \frac{32}{5}m_2 - \frac{16}{5}m_3 \right) = \frac{180 + 64}{25} = \frac{244}{25}$$

$$I_a^{Q/M} = m \cdot \frac{244}{25}$$

$$\text{now } I_a^{B/M} = I_a^{b/o} - I_a^{Q/A} + I_a^{B/Q} = \frac{8500 + m(244 - 676)}{25}$$

$$I_a^{B/M} = \frac{8500}{25} + \left( -\frac{676}{25} + \frac{244}{25} \right)m = 3316$$



$$I_x^{c/o} = \frac{3}{5}m\left(\frac{1}{4}a^2 + h^2\right) \quad \text{if } r = a \\ = \frac{3}{5}m\left(\frac{1}{4}r^2\right) = \frac{3}{4}mr^2$$

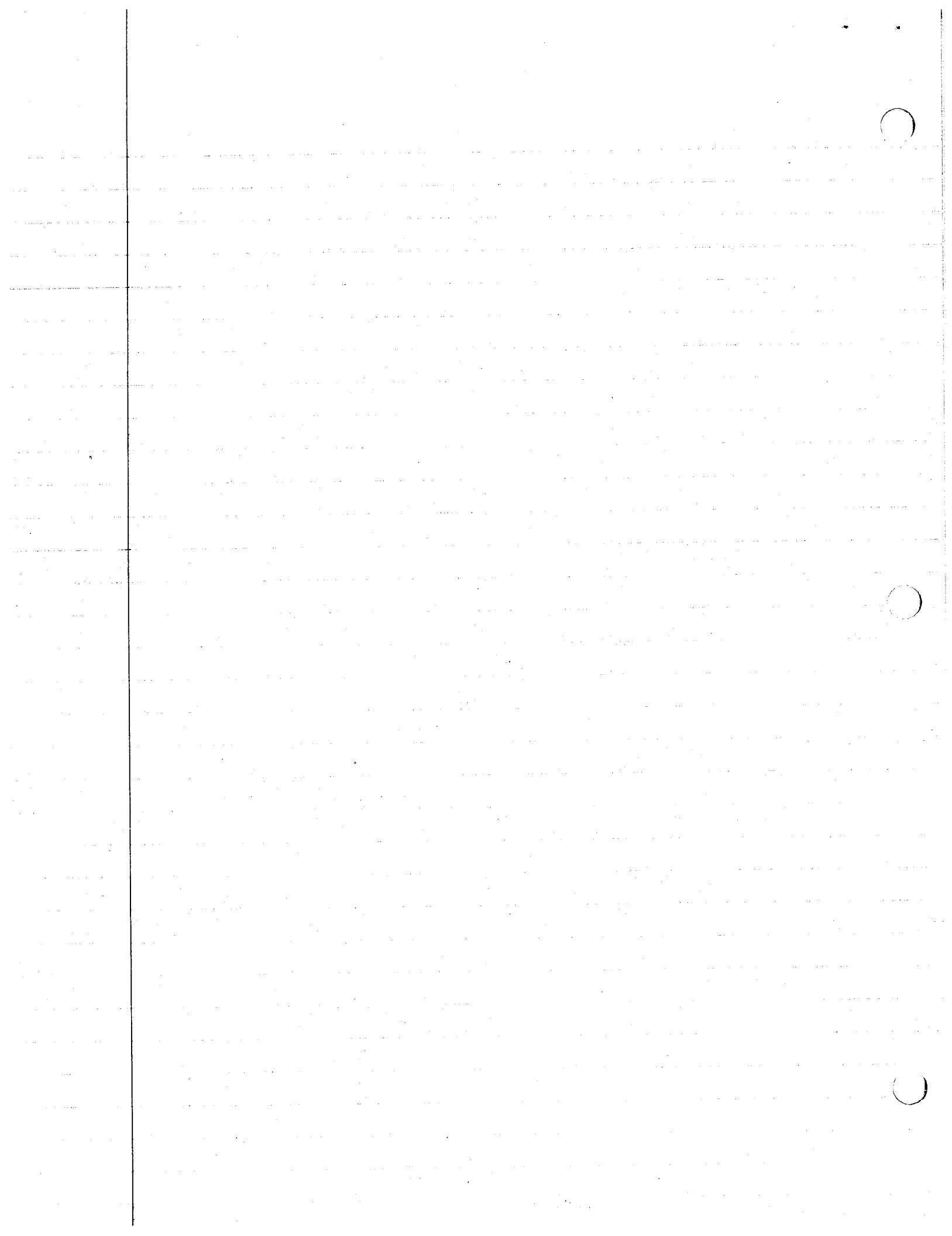
$$I_x^{H/O} = I_x^{H/H} + I_x^{H/O} \\ = \frac{83}{320} \cdot 2mR^2 + 2m \cdot \left(\frac{3}{8}R\right)^2 = \frac{86}{20}mR^2$$

$$I_x^{S/O} = I_x^{H/O} + I_x^{c/o} = \left(\frac{15}{20} + \frac{86}{20}\right)mR^2 = \frac{101}{20}mR^2$$

$$I_x^{P/H} = \frac{1}{2}M(R^2 + r^2)$$

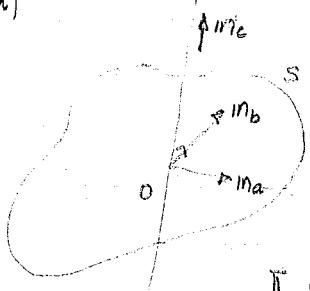
$$I_x^{P/H} = \frac{1}{2}M(R^2 + r^2) + m\left(\frac{1}{4}a^2 + h^2\right) + m\left(\frac{1}{4}r^2\right) + m\left(\frac{1}{4}R^2\right) + mR^2 = \frac{1}{2}M(R^2 + r^2) + mR^2 + m\left(\frac{1}{4}R^2 + \frac{1}{4}r^2 + \frac{1}{4}a^2 + h^2\right)$$

Ans.



$$\begin{aligned} I_x^{H/H^*} &= I_x^{H/P} + I_x^{H/P} = \frac{16mR^2}{9\pi^2} + \frac{4mR^2}{S} \\ I_x^{H/A} &= I_x^{H/H^*} + I_x^{H/A} = \frac{16mR^2}{9\pi^2} + \frac{4mR^2}{S} + 2m \left[ \frac{4R}{3\pi} + R \right]^2 = \left( \frac{8R^2}{3\pi} + R^2 \right) 2m + \frac{4mR^2}{S} \\ I_x &\approx \frac{15}{20} mR^2 + \frac{16mR^2}{20} + \frac{40R^2}{20} \end{aligned}$$

6i)



$$I_c \cdot m_c = I_c = \sum m_i [\|p_i \times (m_c \times p_i)\|]$$

$$\text{where } \|p_i\| = a_i m_a + b_i m_b$$

$$m_c \times \|p_i\| = a_i m_b + b_i m_a$$

$$(a_i m_a + b_i m_b) \times (a_i m_b - b_i m_a) = a_i a_i m_c + b_i b_i m_c = (a_i^2 + b_i^2) m_c$$

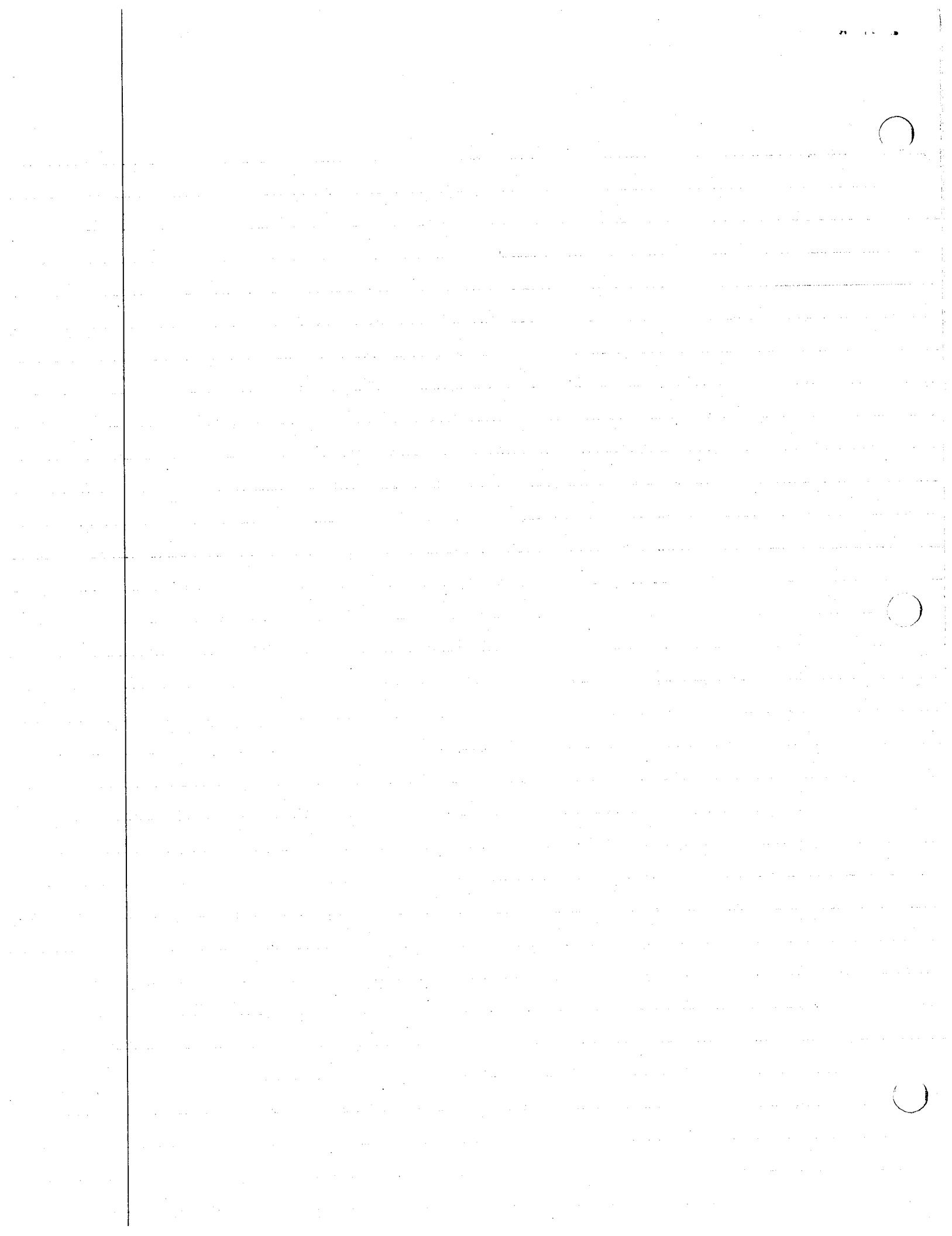
$$I_c \cdot m_c = I_c = \sum m_i (a_i^2 + b_i^2)$$

$$I_a \cdot m_a = I_a = \sum m_i [\|p_i \times (m_a \times p_i)\| \cdot m_a]$$

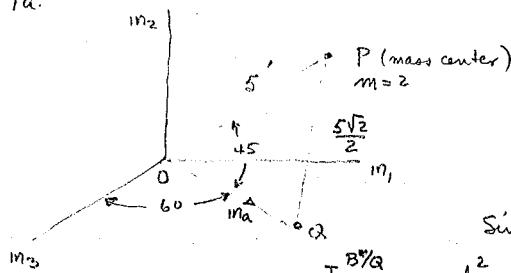
$$m_a \times \|p_i\| = b_i m_c \quad \|p_i \times (m_a \times p_i)\| = -a_i b_i m_b + b_i^2 m_a$$

$$I_a = \sum m_i b_i^2 \quad ; \quad I_b = \sum m_i a_i^2 \quad \text{similarly}$$

$$\therefore I_c = I_a + I_b$$



7a.



$$I_1 = 20 \quad I_2 = 30 \quad I_3 = 40$$

find  $I_a^{B/B^*}$

$$I_{ma} = \frac{1}{2} I_{m3} + \frac{\sqrt{3}}{2} I_{m1}$$

$$\text{Since we want } I_a^{B/B^*} = I_a^{B/B^*} + I_a^{B/B^*}$$

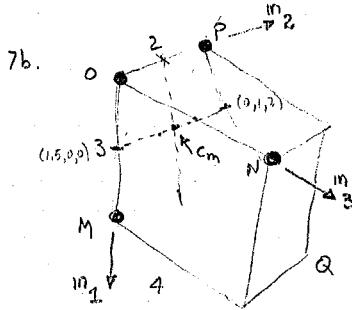
$$I_a^{B/B^*} = m \cdot l_{pq}^2 = 2 \cdot \left(\frac{5\sqrt{2}}{2}\right)^2 = 2 \cdot \frac{25}{2} = 25 \text{ slug-ftr}^2$$

$$I_a^{B/B^*} = I_{ma} \cdot I_a^{B/B^*} \cdot I_{ma}$$

$$\text{where } I_a^{B/B^*} = I_1 I_{m1} I_{m1} + I_2 I_{m2} I_{m2} + I_3 I_{m3} I_{m3}; \text{ since } I_{ma} \text{ has no } m_2 \text{ component}$$

$$\text{then } I_a^{B/B^*} = (I_1 I_{m1}) \frac{\sqrt{3}}{2} + (I_3 I_{m3}) \cdot \frac{1}{2} \quad I_a = I_1 \left(\frac{\sqrt{3}}{2}\right)^2 + I_3 \left(\frac{1}{2}\right)^2 = I_1 \cdot \frac{3}{4} + I_3 \cdot \frac{1}{4} \\ = 20 \cdot \frac{3}{4} + 40 \cdot \frac{1}{4} = 25$$

$$I_a^{B/B^*} = I_a^{B/B^*} + I_a^{B/B^*} = 25 + 25 = 50 \text{ slug-ftr}^2$$



To find min radius of gyration we need to find the min moment of inertia, which will be the minimum principal moment of inertia. Thus if we call it  $I_{min}$  then  $k = \sqrt{\frac{I_{min}}{4m}}$ . Now the min principal moment of inertia will be at the mass center. By construction we can find the mass center. To this end, define a set of coordinates and let us find the moments of inertia about the given directions  $m_1, m_2, m_3$  for point O for the system.

$$\text{Using (problem 6c)} \quad I_1^{S/O} = \sum m_i ((x_2)_i^2 + (x_3)_i^2) \quad I_2^{S/O} = \sum m_i [(x_3)_i^2 + (x_1)_i^2] \\ I_3^{S/O} = \sum m_i [(x_1)_i^2 + (x_2)_i^2] \quad I_{12}^{S/O} = -2 \sum m_i (x_1)_i (x_2)_i \quad I_{23}^{S/O} = -2 \sum m_i (x_2)_i (x_3)_i$$

$$\text{and } I_{31}^{S/O} = -2 \sum m_i (x_3)_i (x_1)_i$$

$$P_1 \text{ located at } (0, 0, 0) \quad P_2 (3, 0, 0) \quad P_3 (0, 2, 0) \quad P_4 (0, 0, 4). \text{ Since all have same mass } m \\ \text{thus } \sum (x_2)_i^2 = 4 \quad \sum (x_1)_i^2 = 9 \quad \sum (x_3)_i^2 = 16. \quad \sum (x_1 x_2)_i = 0 \quad \sum [(x_2)(x_3)]_i = 0$$

$$I_1^{S/O} = m \cdot 20 \quad I_2^{S/O} = m \cdot 25 \quad I_3^{S/O} = m \cdot 13 \quad \sum (x_3 x_1)_i = 0$$

$\frac{I_{jk}^{S/O}}{I_{jk}^{S/O}}$	1	2	3
1	20m	0	0
2	0	25m	0
3	0	0	13m

$J_{jk}^{S/O}$

$$\Rightarrow cm = \left(\frac{3}{4}, \frac{1}{2}, 1\right) \text{ obtained by } \bar{x}_i = \frac{\sum (x_i m_i)_i}{\sum m_i} \\ \Rightarrow I_{jk}^{S/S} = I_{jk}^{S/O} - I_{jk}^{S/O} w/ I_{jk}^{S/O} = 4m \left\{ p_j^{S/O} p_k^{S/O} - (p_j^{S/O} p_k^{S/O}) \right\} \\ (p^{S/O})^2 = \frac{9}{16} + \frac{4}{16} + \frac{16}{16} = \frac{29}{16}$$

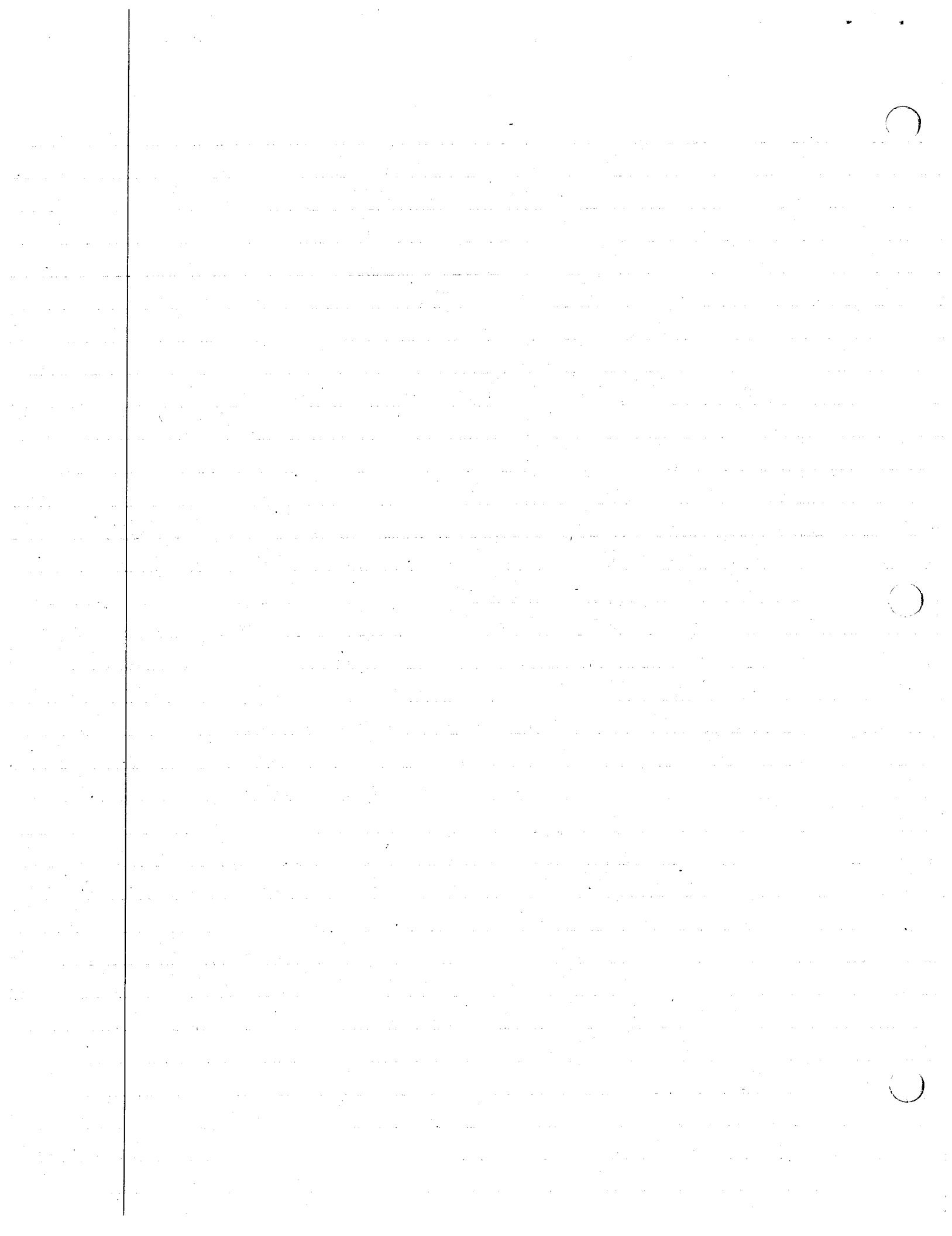
$J_{jk}^{S/O}$

$\frac{J_{jk}^{S/O}}{J_{jk}^{S/O}}$	1	2	3
1	5m	$-\frac{3}{2}m$	-3m
2	$-\frac{3}{2}m$	$\frac{25}{4}m$	-2m
3	-3m	-2m	$\frac{13}{4}m$

$$\text{NOTE: } I_{jk}^{S/O} = 4m \left[ p_j^{S/O} (m_j \times p_k^{S/O}) \right] \cdot m_k \\ 4m \left[ p_j^{S/O} (m_j \times p_k^{S/O}) \right] \cdot m_k$$

$$\text{but } p \times (m_j \times p) = (p \cdot p) m_j - (p \cdot m_j) p \\ \text{dot w/ } m_k = (p \cdot p) \delta_{jk} - (p \cdot m_j) (p \cdot m_k)$$

$\frac{J_{jk}^{S/O}}{J_{jk}^{S/O}}$	1	2	3
1	15m	1.5m	3m
2	1.5m	18.75m	2m
3	3m	2m	9.75m



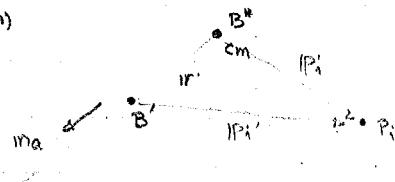
Aside: we note that when  $j=k$   $I_{jj}^{S/S} = (P^2 - (P \cdot m_j)^2) \cdot 4m$  but  $(P \cdot m_j)$  is the proj onto the  $m_j$  axis  
 $= 4m[P^2 - P^2 \cos^2(P \cdot m_j)] = [P^2 \sin^2(P \cdot m_j)]4m = (4m)l^2$  where  $l=1$  distance  
from  $S^*$  to  $m_j$  axis.

thus to minimize  $I_{jk}$   $\Rightarrow \det \frac{S/S}{m^3} \begin{bmatrix} 15 - \tilde{I} & 1.5 & 3 \\ 1.5 & 18.75 - \tilde{I} & 2 \\ 3 & 2 & 9.75 - \tilde{I} \end{bmatrix} = 0$  where  $\tilde{I} = I/m$

thus  $(15 - \tilde{I})(18.75 - \tilde{I})(9.75 - \tilde{I}) + 18 - 9(18.75 - \tilde{I}) - 2.25(9.75 - \tilde{I}) - 4(15 - \tilde{I}) = 0$  Solution to this  
for min  $I$  must be  $< 9.75m$ .  $I_{\min} = 8.243m$  is obtained after solution

$$\text{now } k = \sqrt{\frac{I_{\min}}{4m}} = \sqrt{2.061} = 1.436 \text{ ft}$$

7c. (i)  $I_a^{B/B^*} = \sum m_i P_i \times (m_a \times P_i)$ . for a principal axis  $I_a = \lambda m_a$



$$\text{thus } I_a^{B/B^*} \times m_a = 0$$

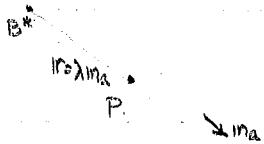
$$I_a^{B/B^*} = I_a^{B/B^*} + I_a^{B^*/B^*} \text{ if } P^{B/B^*} = \lambda m_a$$

$$= I_a^{B/B^*} + M(\lambda m_a \times (m_a \times \lambda m_a))$$

$$\text{thus } I_a^{B/B^*} = I_a^{B/B^*} \text{ but } I_a^{B/B^*} \times m_a = I_a^{B/B^*} \times m_a = 0$$

Hence, principal axis for  $B^*$  is a principal axis for any point on the axis.

7c(ii)



$$\text{now } I_a^{B/P} \times m_a = 0 \text{ since } m_a \text{ is a principal axis for } P.$$

but it also passes through  $B^*$  hence we can form

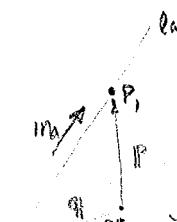
$$I_a^{B/B^*}. \text{ yet } I_a^{B/B^*} = I_a^{B/P} + I_a^{P/B^*} = I_a^{B/P} - I_a^{B/B^*}$$

$$\text{now } I_a^{B/B^*} \times m_a = (I_a^{B/P} \times m_a = 0 \text{ by hypothesis}) - I_a^{B/P} \times m_a$$

$$\text{but } I_a^{B/P} \times m_a = M[\lambda m_a \times (m_a \times \lambda m_a)] \times m_a = 0 \Rightarrow I_a^{B/B^*} \times m_a = 0 \text{ which is only true if}$$

$$I_a^{B/B^*} = \mu m_a \Rightarrow \text{the axis is a principal axis for } B^*$$

7c(iii)



$$\text{if } I_a^{B/P_1} \times m_a = 0 \text{ and } I_a^{B/P_2} \times m_a = 0 \Rightarrow I_a^{B/B^*} \times m_a = 0.$$

$$I_a^{B/B^*} = I_a^{B/B^*} + I_a^{B^*/B^*} \text{ and } I_a^{B^*/B^*} \times m_a = I_a^{B^*/B^*} \times m_a$$

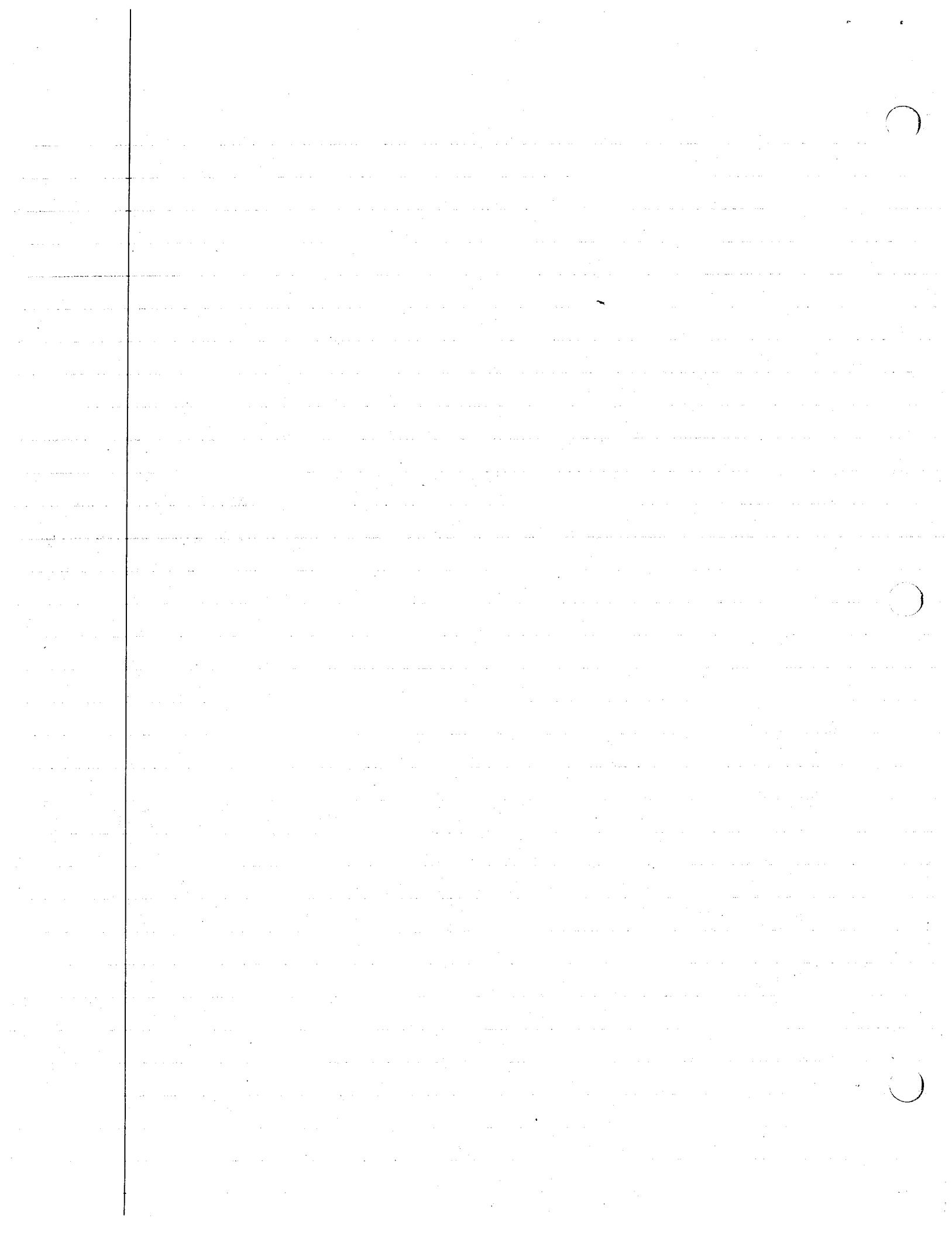
$$I_a^{B/P_2} = I_a^{B/B^*} + I_a^{B^*/B^*} = I_a^{B/B^*} \times m_a + I_a^{B^*/B^*} \times m_a$$

$$\text{then } I_a^{B/P_1} - I_a^{B/P_2} = I_a^{B^*/P_1} - I_a^{B^*/P_2} \text{ and}$$

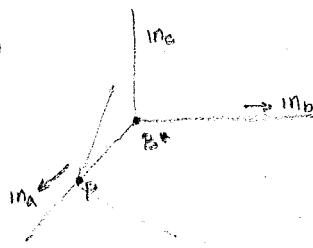
$$I_a^{B^*/P_1} \times m_a = I_a^{B^*/P_2} \times m_a \text{ but since } P_2, P_1 \text{ are arbitrary}$$

$$\text{then } \Rightarrow \text{either } P = q \text{ or } P \neq q \text{ must be } \parallel \text{ to } m_a \Rightarrow I_a^{B^*/P_1} \times m_a = 0 \text{ since } P \neq q \text{ implies}$$

$$\Rightarrow I_a^{B/B^*} \times m_a = 0 \Rightarrow I_a^{B/B^*} \parallel m_a \Rightarrow b_a \text{ is a principal axis for } B^* \text{ by 7c(i)}$$



7c(v)



since P lies on a principal axis of the

mass center then  $I_a^{B/P} \times m_a = 0$  thus  $I_a^{B/B} \cdot m_b = 0$

$$\text{and } I_a^{B/P} \cdot m_c = 0$$

$$\text{but } I_a^{B/P} = I_a^{B/B} + I_a^{B''/P}$$

$$I_a^{B/P} \cdot m_b = 0 + I_{ab}^{B''/P} \text{ but } I_{ab} = m \cdot 0 = 0 \Rightarrow I_{ab}^{B''/P} = 0$$

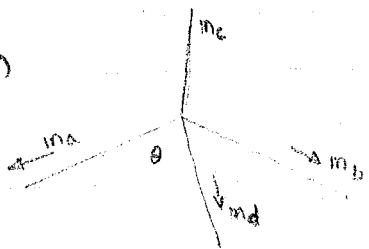
similarly we can show that  $I_a^{B/P} \cdot m_c = I_{ac}^{B/P} = 0$

$$\text{also } I_b^{B/P} = I_b^{B/B} + I_b^{B''/P} \text{ but } I_b^{B''/P} = m (\lambda m_a \times (m_b \times \lambda m_a)) = +m\lambda^2 m_b \text{ where } \frac{B''/P}{B/P} = \lambda m_a$$

hence  $I_b^{B/P} \times m_b = I_b^{B/B} \times m_b + m\lambda^2 m_b \times \lambda m_b = 0 \Rightarrow I_b^{B/P} \parallel m_b$  and hence is a principal axis of  $I_b^{B/P}$ . We can do this too for  $I_c^{B/P}$  & show that it is  $\parallel$  to  $m_c$ . Similarly  $I_{ac}^{B/P} = I_{bc}^{B/P}$

hence the principal axes for any point on a principal axis for the mass center are  $\parallel$  to the principal axes for the mass center.

7c(v)



$$\text{if } I_a = I_b \Rightarrow I_d = I_a = I_b$$

$$I = I(m_a m_a + m_b m_b) + J m_c m_c$$

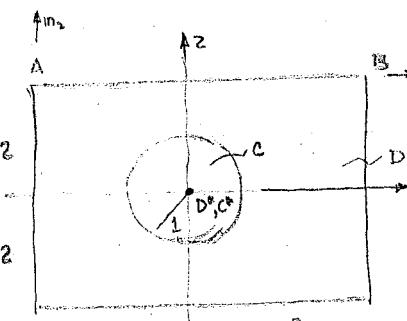
$$m_d \cdot I m_d = m_d (m_a m_a + m_b m_b) \cdot m_d$$

$$I_d = I(\cos^2 \theta + \sin^2 \theta) = I$$

$$\text{thus } I_d = I_a = I_b$$

since  $m_a$  &  $m_b$  are principal axes  $\perp$  to  $m_c$ ,  $m_c$  also principal  
 $\perp m_a, m_b$  also principal  
 $m_d \cdot m_a = \cos \theta$   
 $m_d \cdot m_b = \sin \theta$

7(d)



$$\text{let } S = D - C.$$

$I_{jk}^{B/B}$	1	2	3
1	$4m/3$	0	0
2	0	$3m$	0
3	0	0	$13m/3$

$I_{jk}^{S/A}$	1	2	3
1	$m/4$	0	0
2	0	$m/4$	0
3	0	0	$m/2$

$I_{jk}^{S/B}$	1	2	3
1	$13m/12$	0	0
2	0	$11m/4$	0
3	0	0	$23m/6$

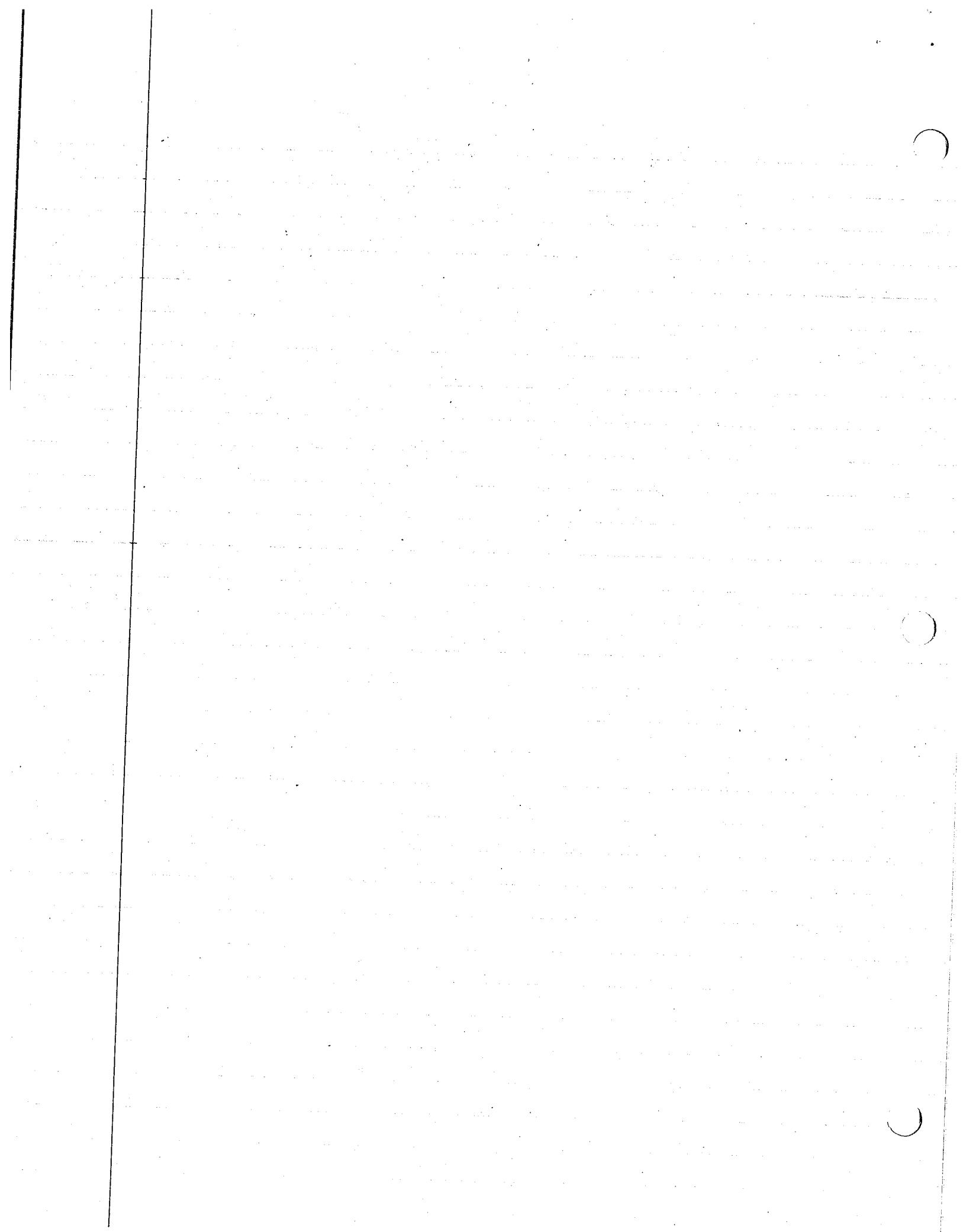
$$I_{jk}^{S/A} = I_{jk}^{S/S} + I_{jk}^{S/A}$$

$$I_{jk}^{S/A} = m \left[ \frac{S}{A} \times (m_j \times \frac{S/A}{P}) \right] \cdot m_k = (P \cdot P) \delta_{jk} - (P \cdot m_j)(P \cdot m_k)$$

$$P = -2m_2 + 3m_1, \quad P \cdot P = 13$$

$I_{jk}^{S/A}$	1	2	3
1	$4m/6m/0$		
2	$6m/9m/0$		
3	0 0 13m		

$I_{jk}^{S/A}$	1	2	3
1	$61m/12$	$6m/0$	
2	$6m/47m/0$		
3	0 0 $101m/6$		



now form the determinant  $(I_{jk} - I\Delta) = 0$   $\Delta_{ii} = 1$   
 $i, j = 0 \text{ or } i \neq j$   
 thus  $I_3 = \frac{101}{6} m$  and  $\left(\frac{61}{12}m - I\right)\left(\frac{47}{4}m - I\right) - 36m^2 = 0$  for the other 3.

$$\frac{61 \cdot 47}{12}m^2 - \frac{202}{12}mI + I^2 - 36m^2 = 0 \quad \text{or} \quad I^2 - 16.83mI + 23.73 = 0$$

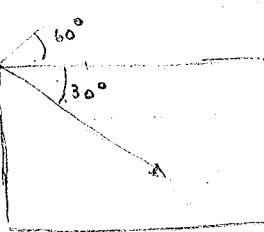
$$I_{1,2} = \frac{16.83 \pm 13.73}{2} \Rightarrow I_1 = 15.28m \quad I_2 = 1.55m$$

$$\begin{bmatrix} 5.08m - I & 6m & 0 \\ 6m & 11.75m - I & 0 \\ 0 & 0 & 16.83m - I \end{bmatrix} \begin{pmatrix} a \\ b \\ c \end{pmatrix} = 0$$

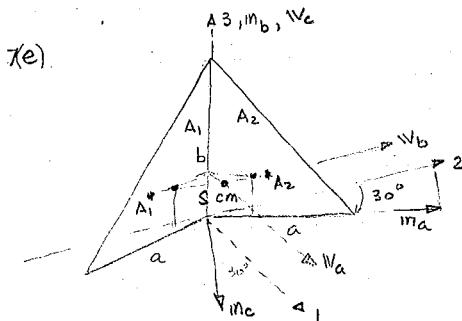
for  $I_3 = 16.83m \Rightarrow \hat{m}_3 = (0, 0, 1)$   
 for  $I_1 = 15.28m \Rightarrow \hat{m}_1 = (.5, .86, 0)$   
 for  $I_2 = 1.55m \Rightarrow \hat{m}_2 = (.86, .5, 0)$

$$\text{for } I_1 \quad -10.2m \cdot a + 6m \cdot b = 0 \quad \text{let } a = 1 \quad b = \frac{10.2}{6} = 1.7 \quad \therefore \sqrt{a^2+b^2} = 1.97 ; \quad \bar{a} = \frac{1}{1.97} \quad \bar{b} = \frac{1.7}{1.97}$$

$$\text{for } I_2 \quad 3.53m \cdot a + 6m \cdot b = 0 \quad \text{let } a = 1 \quad b = -\frac{3.53}{6} = -0.588 \quad \therefore \sqrt{a^2+b^2} = 1.16 ; \quad \bar{a} = \frac{1}{1.16} \quad \bar{b} = \frac{-0.588}{1.16}$$



thus the  $\min$  angle is  $30^\circ$



To prove the above statement:

$I_{jk}^{A_1/A_2^*}$	$m_a$	$m_b$	$m_c$
$m_a$	$\frac{mb^2}{18}$	$mab$	0
$m_b$	$\frac{mb}{18}$	$\frac{ma^2}{18}$	0
$m_c$	0	0	$m(a^2+b^2)$

$$\text{now } I_{jk}^{A_2/A_1^*} = I_{jk}^{A_2/A_2^*} + I_{jk}^{A_1^*/A_1^*} = I_{jk}^{A_2/A_2^*} - I_{jk}^{S^*/A_2^*}$$

$$I_{jk}^{S^*/A_2^*} = -\frac{a}{2\sqrt{3}}m_2 \quad m_2 = \frac{1}{2}m_b + \frac{\sqrt{3}}{2}m_a$$

$$I_{jk}^{S^*/A_2^*} = \frac{a}{2\sqrt{3}}\left(\frac{1}{2}m_b - \frac{\sqrt{3}}{2}m_a\right) = \frac{a}{4\sqrt{3}}m_b - \frac{a}{4}m_a$$

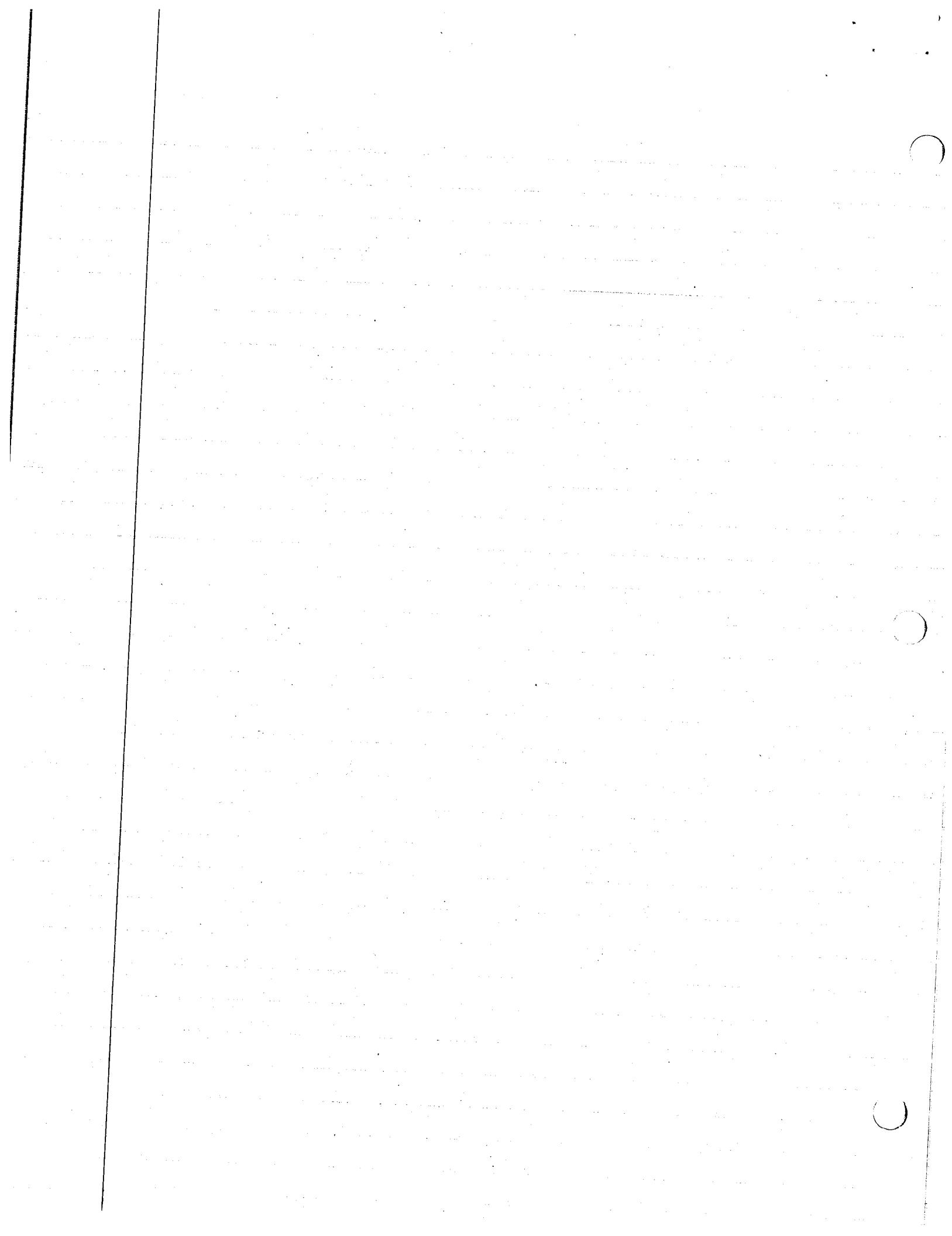
$$\text{now } I_{jk}^{S^*/A_2^*} = m \left[ p \cdot p \delta_{jk} - (p \cdot m_j)(p \cdot m_k) \right]$$

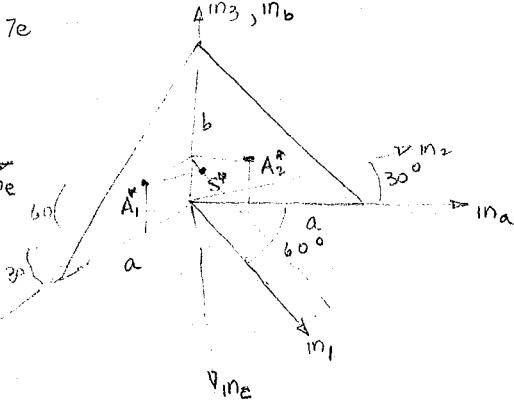
$$p \cdot p = \frac{a^2}{48} + \frac{a^2}{16} = \frac{a^2}{12}$$

$$I_{jk}^{M_{jk}^*} = \frac{m_a}{m_a} \quad \frac{m_b}{m_b} \quad \frac{m_c}{m_c}$$

$I_{jk}^{S^*/A_2^*}$	$m_a$	$m_b$	$m_c$
$m_a$	$\frac{m_a^2}{48}$	$\frac{m_a^2}{16\sqrt{3}}$	0
$m_b$	$\frac{m_a^2}{16\sqrt{3}}$	$\frac{m_a^2}{16}$	0
$m_c$	0	0	$\frac{m_a^2}{12}$

$I_{jk}^{M_{jk}^*}$	$m_a$	$m_b$	$m_c$
$m_a$	$\frac{m(b^2-a^2)}{18\sqrt{3}}$	$\frac{m(ab-a^2)}{18\sqrt{3}}$	0
$m_b$	$\frac{m(ab-a^2)}{18\sqrt{3}}$	$\frac{m(a^2-b^2)}{18}$	0
$m_c$	0	0	$\frac{m(a^2-b^2)}{18}$





$$\text{cm of } A_2 = \left( \frac{a}{6}, \frac{2c}{3\sqrt{3}}, \frac{1}{3}b \right)$$

$$\text{cm of } A_1 = \left( \frac{a}{6}, -\frac{2c}{3\sqrt{3}}, \frac{1}{3}b \right)$$

$$\text{cm of } S = \left( \frac{a}{6}, 0, \frac{1}{3}b \right)$$

$$\begin{array}{c|ccc} & I_{jk} & I_{jk}^* \\ \hline I_{lm} & I_{lm} & I_{lm} & I_{lm} \\ I_{ma} & \frac{mb^2}{18} & \frac{mab}{36} & 0 \\ I_{mb} & \frac{mab}{36} & \frac{mc^2}{18} & 0 \\ I_{mc} & 0 & 0 & m(a^2+b^2) \end{array}$$

$$\text{now } I_{lm} = \sum_{j=1}^c \sum_{k=1}^c a_{kj} I_{jk} I_{km}$$

$$\text{where } a_{kj} \Rightarrow \begin{cases} l=k, 1, 2, 3 \\ j=a, b, c \end{cases}$$

$$b_{km} \Rightarrow \begin{cases} k=a, b, c \\ m=1, 2, 3 \end{cases}$$

$$\begin{array}{c|ccc} & I_{ma} & I_{mb} & I_{mc} \\ \hline I_{m1} & \frac{1}{2} & 0 & \frac{\sqrt{3}}{2} \\ I_{m2} & \frac{\sqrt{3}}{2} & 0 & -\frac{1}{2} \\ I_{m3} & 0 & 1 & 0 \end{array}$$

$$\begin{array}{c|ccc} & I_{m1} & I_{m2} & I_{m3} \\ \hline I_{ma} & \frac{1}{2} & \frac{\sqrt{3}}{2} & 0 \\ I_{mb} & 0 & 0 & 1 \\ I_{mc} & \frac{\sqrt{3}}{2} & -\frac{1}{2} & 0 \end{array}$$

$A_{ij}$

$b_{km}$

$I_{lm}^{A_2/A_2^*}$

thus

$$\begin{pmatrix} \frac{1}{2} & 0 & \frac{\sqrt{3}}{2} \\ \frac{\sqrt{3}}{2} & 0 & -\frac{1}{2} \\ 0 & 1 & 0 \end{pmatrix} \begin{pmatrix} \frac{mb^2}{18} & \frac{mab}{36} & 0 \\ \frac{mab}{36} & \frac{mc^2}{18} & 0 \\ 0 & 0 & m(a^2+b^2) \end{pmatrix} \begin{pmatrix} \frac{1}{2} & \frac{\sqrt{3}}{2} & 0 \\ 0 & 0 & 1 \\ \frac{\sqrt{3}}{2} & -\frac{1}{2} & 0 \end{pmatrix} = \begin{pmatrix} \frac{mb^2}{18} + m(a^2+b^2)/3 & \frac{mb^2\sqrt{3}}{72} - m(a^2+b^2)\sqrt{3}/3 & \frac{mab}{72} \\ \frac{mb^2\sqrt{3}}{72} - m(a^2+b^2)\sqrt{3}/3 & \frac{mb^2/3}{72} + m(a^2+b^2)/3 & \frac{mab\sqrt{3}}{72} \\ \frac{mab}{72} & \frac{mab\sqrt{3}}{72} & \frac{mab^2}{18} \end{pmatrix}$$

$$\text{now } \rightarrow I_{lm}^{A_2/S^*} = I_{lm}^{A_2/A_2^*} + I_{lm}^{A_2^*/S^*}$$

but first for  $A_1$ ,

$$\begin{array}{c|ccc} & I_{md} & I_{mb} & I_{mc} \\ \hline I_{m1} & \frac{1}{2} & 0 & -\frac{\sqrt{3}}{2} \\ I_{m2} & -\frac{\sqrt{3}}{2} & 0 & -\frac{1}{2} \\ I_{m3} & 0 & 1 & 0 \end{array}$$

$$\begin{array}{c|ccc} & I_{m1} & I_{m2} & I_{m3} \\ \hline I_{md} & \frac{1}{2} & -\frac{\sqrt{3}}{2} & 0 \\ I_{mb} & 0 & 0 & 1 \\ I_{mc} & -\frac{\sqrt{3}}{2} & -\frac{1}{2} & 0 \end{array}$$

$A_{ij}$

$b_{km}$

$I_{lm}^{A_1/A_1^*}$

$$\begin{pmatrix} \frac{mb^2}{36} & \frac{mab}{72} & -m(a^2+b^2)\sqrt{3}/36 \\ -m(b^2\sqrt{3})/36 & -mab\sqrt{3}/72 & -m(a^2+b^2)/36 \\ mab/36 & mab^2/18 & 0 \end{pmatrix}$$

$$= \begin{pmatrix} \frac{mb^2}{72} + m(a^2+b^2)/3 & -\frac{mb^2\sqrt{3}}{72} + m(a^2+b^2)\sqrt{3}/72 & \frac{mab}{72} \\ -\frac{mb^2\sqrt{3}}{72} + m(a^2+b^2)\sqrt{3}/72 & \frac{mb^2/3}{72} + m(a^2+b^2)/3 & \frac{mab\sqrt{3}}{72} \\ \frac{mab}{72} & \frac{mab\sqrt{3}}{72} & \frac{mab^2}{18} \end{pmatrix}$$

$$\text{Now we want translate this to } I_{lm}^{A_1/S^*} = I_{lm}^{A_1/A_1^*} + I_{lm}^{A_1^*/S^*} =$$

$$I_P^{A_2^*/S^*} = +\frac{a}{2\sqrt{3}} I_{m2}$$

$$I_P^{A_1^*/S^*} = -\frac{a}{2\sqrt{3}} I_{m2}$$

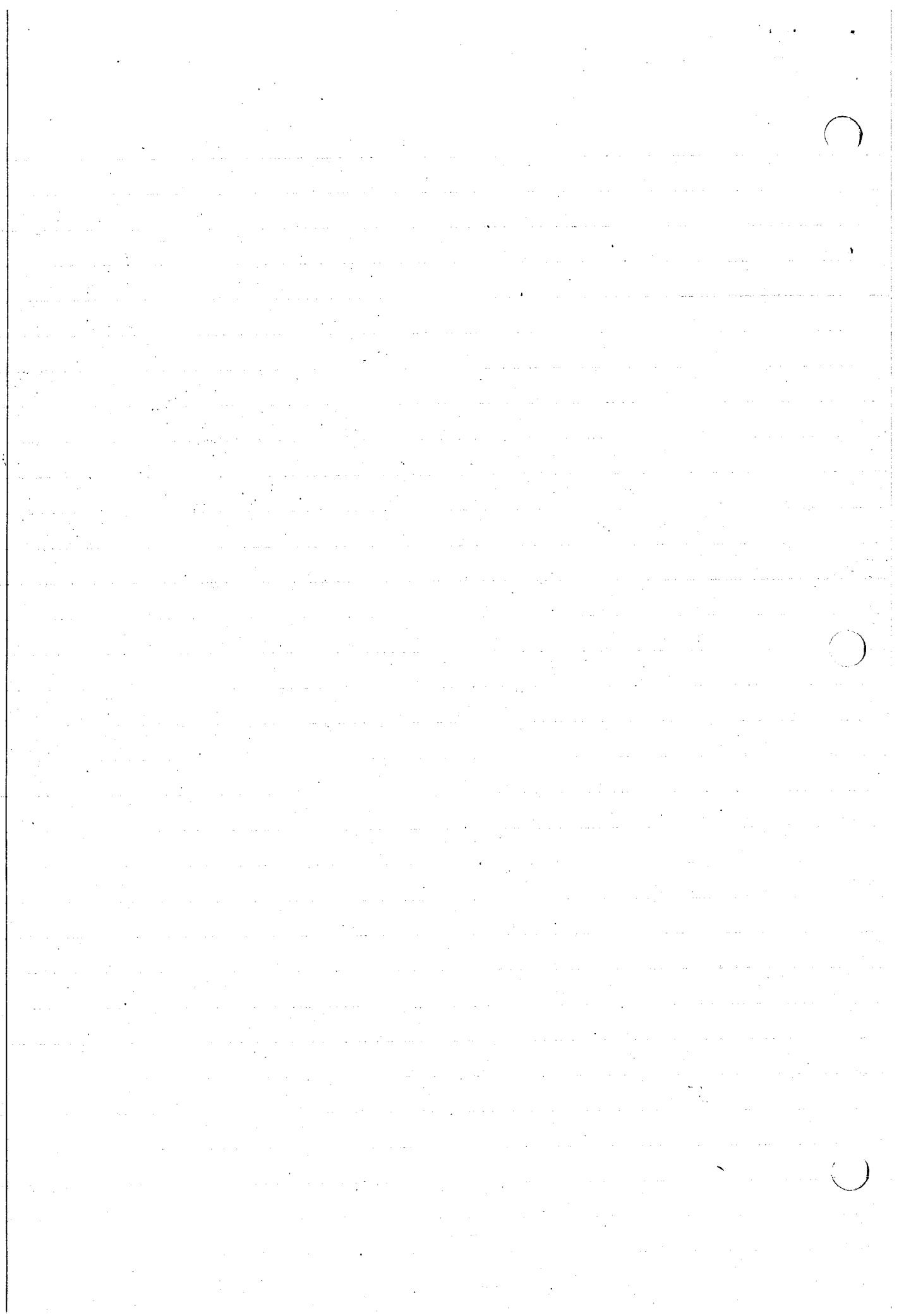
$$I_P^2 = \frac{a^2}{12}$$

$$I_P^2 = \frac{a^2}{12}$$

$$\text{now } I_{lm}^{A_2/S^*} = m [I_P^2 S_{lm} - (I_P \cdot I_{m1})(I_P \cdot I_{m3})]$$

$$I_P \cdot I_{m2} = +\frac{a}{2\sqrt{3}}$$

$$I_P \cdot I_{m1} = I_P \cdot I_{m3} = 0$$



Thus  $I_{dm}^{A_2/S^*}$  and the same is true for  $I_{dm}^{A_1/S^*}$

$$\begin{matrix} & m_1 & m_2 & m_3 \\ \begin{array}{c} \frac{ma^2}{12} \\ 0 \\ 0 \\ 0 \end{array} & \left| \begin{array}{ccc} 0 & 0 \\ 0 & 0 \\ \frac{ma^2}{12} \end{array} \right. \end{matrix}$$

Thus  $I_{dm}^{A_1/S^*}$

$$\left( \begin{array}{ccc} \frac{mb^2}{72} + 3m(a^2+b^2) + ma^2 & -\frac{mb^2\sqrt{3}}{72} & \frac{mab}{72} \\ -\frac{mb^2\sqrt{3}}{72} + m(a^2+b^2)\sqrt{3} & \frac{3mb^2}{72} + m(a^2+b^2) & -\frac{mab\sqrt{3}}{72} \\ \frac{mab}{72} & -\frac{mab\sqrt{3}}{72} & \frac{ma^2}{18} + \frac{mb^2}{12} \end{array} \right)$$

and  $I_{dm}^{A_2/S^*}$

$$\left( \begin{array}{ccc} \frac{mb^2}{72} + 3m(a^2+b^2) + ma^2 & \frac{mb^2\sqrt{3}}{72} - m(a^2+b^2)\sqrt{3} & \frac{mab}{72} \\ \frac{mb^2\sqrt{3}}{72} - m(a^2+b^2)\sqrt{3} & \frac{3mb^2}{72} + m(a^2+b^2) & \frac{mab\sqrt{3}}{72} \\ \frac{mab}{72} & \frac{mab\sqrt{3}}{72} & \frac{ma^2}{18} + \frac{mb^2}{12} \end{array} \right)$$

Now

$$I_{dm} = I_{dm}^{A_1/S^*} + I_{dm}^{A_2/S^*}$$

$$\left( \begin{array}{ccc} \frac{mb^2}{36} + \frac{m(a^2+b^2)}{12} + \frac{ma^2}{6} & 0 & \frac{mab}{36} \\ \frac{ma^2}{4} + \frac{mb^2}{9} & 0 & 0 \\ \frac{mab}{36} & 0 & \frac{ma^2}{9} + \frac{mb^2}{6} \\ 0 & \frac{mab}{36} & \frac{10ma^2}{36} \end{array} \right)$$

for  $a=2b$  &  $b=1$   $a=2$

$$\left( \begin{array}{ccc} m[1+\frac{1}{4}] = m\frac{5}{4}b^2 & 0 & \frac{m}{18}b^2 \\ 0 & \frac{2mb^2}{9} & 0 \\ \frac{2mb^2}{18} & 0 & \frac{m \cdot 10b^2}{9} \end{array} \right)$$

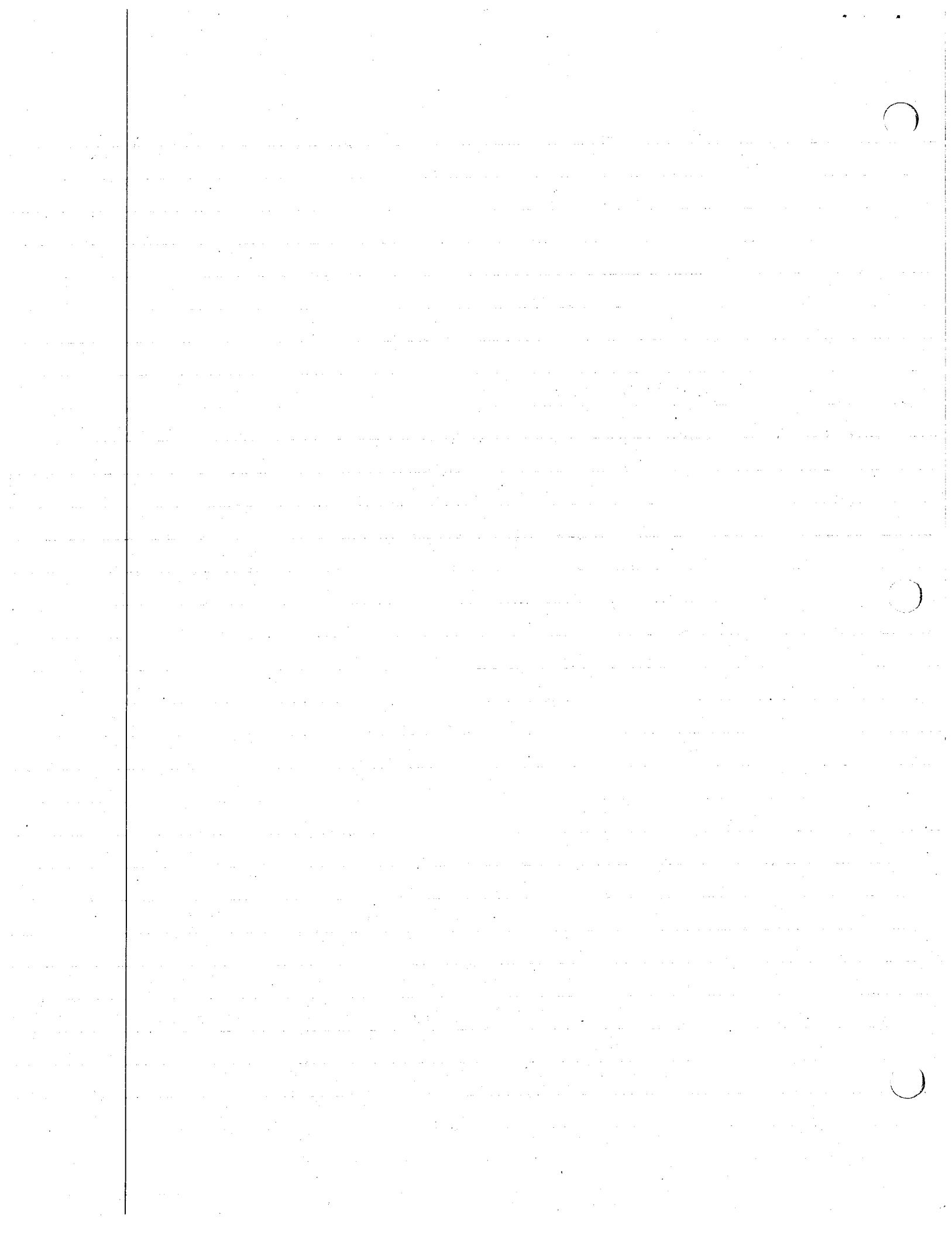
thus  $(\frac{10}{9}m - I)(\frac{2mb}{9} - I)(\frac{10}{9}m - I) - \frac{m^2}{18}(\frac{2m}{9} - I)^2 = 0 \quad \therefore (\frac{2m}{9} - I) \left[ \left( \frac{10}{9}m - I \right)^2 - \frac{m^2}{324} \right] = 0$

①  $I = \frac{2mb^2}{9}$     ②  $b^2(\frac{100}{81}m^2 - \frac{20m}{9}I + I^2 - \frac{m^2}{324}) = b(\frac{399}{324}m^2 - \frac{20m}{9}I + I^2) = 0$

$$I = b^2 \left( \frac{20}{18}m \pm \frac{1}{2} \sqrt{\frac{m^2}{81}} \right) = \frac{20}{18}m \pm \frac{1}{2} \sqrt{\frac{m^2}{81}} = \frac{20}{18}mb \pm \frac{mb}{18} \Rightarrow \frac{21}{18}mb^2 \text{ or } \frac{19}{18}mb^2$$

$$I = 2mk^2 = \frac{21}{18}mb^2 \text{ or } \frac{19}{18}mb^2 \quad k^2 = \sqrt{\frac{21}{36}b^2} \text{ or } \sqrt{\frac{19}{36}b^2}$$

$$\frac{2mb^2}{9} = I = 2mk^2 \Rightarrow k = \frac{mb}{\sqrt{3}}$$



for  $b = 2a$  w/  $A = 1$   $b = 2$  thus

$$J_{lm}^{sys} = \begin{pmatrix} m\left[\frac{1}{4} + \frac{4}{9}\right] = \frac{25}{36}ma^2 & 0 & \frac{ma^2}{18} \\ 0 & m\left(\frac{4}{9} + \frac{1}{36}\right) = m\frac{17}{36}a^2 & 0 \\ \frac{ma^2}{18} & 0 & \frac{10ma^2}{36} \end{pmatrix}$$

$$\text{thus } \left[ \left( \frac{25m}{36} - I \right) \left( \frac{17m}{36} - I \right) \left( \frac{10m}{36} - I \right) - \frac{m^2}{324} \left( \frac{17m}{36} - I \right)^2 \right] a^2 = 0$$

$$\therefore I = \frac{17m}{36}a^2 \quad \Rightarrow \left[ \left( \frac{25m}{36} - I \right) \left( \frac{17m}{36} - I \right) - \frac{m^2}{324} \right] a^4 = 0$$

$$k = \sqrt{\frac{17}{72}}a = \frac{1}{12}\sqrt{\frac{17}{2}}b \quad a^4 \left[ \frac{250m^2}{36^2} - \frac{35}{36}mI + I^2 - \frac{m^2}{(18)^2} \right] = 0$$

$$a^4 \left[ \frac{246m^2}{(36)^2} - \frac{35}{36}mI + I^2 \right] = 0$$

$$I = a^2 \left[ \frac{35m}{72} \pm \frac{m}{2} \sqrt{\left( \frac{35}{36} \right)^2 - 4 \cdot \frac{246}{(36)^2}} \right] = \left[ \frac{35m}{72} \pm \frac{m}{72} \sqrt{(35)^2 - 4(246)} \right] a^2$$

$$= \left[ \frac{35m}{72} \pm \frac{m}{72} \sqrt{241} \right] a^2$$

$$k = \sqrt{35 \pm \sqrt{241}} \cdot \frac{1}{12}a = \frac{1}{24} [35 \pm (241)^{1/2}]^{1/2} b$$

$$35+16 \approx 51 \quad \sqrt{16} \approx 4 \quad \sqrt{51} = \frac{4.5}{24} \approx .18 \dots 2$$

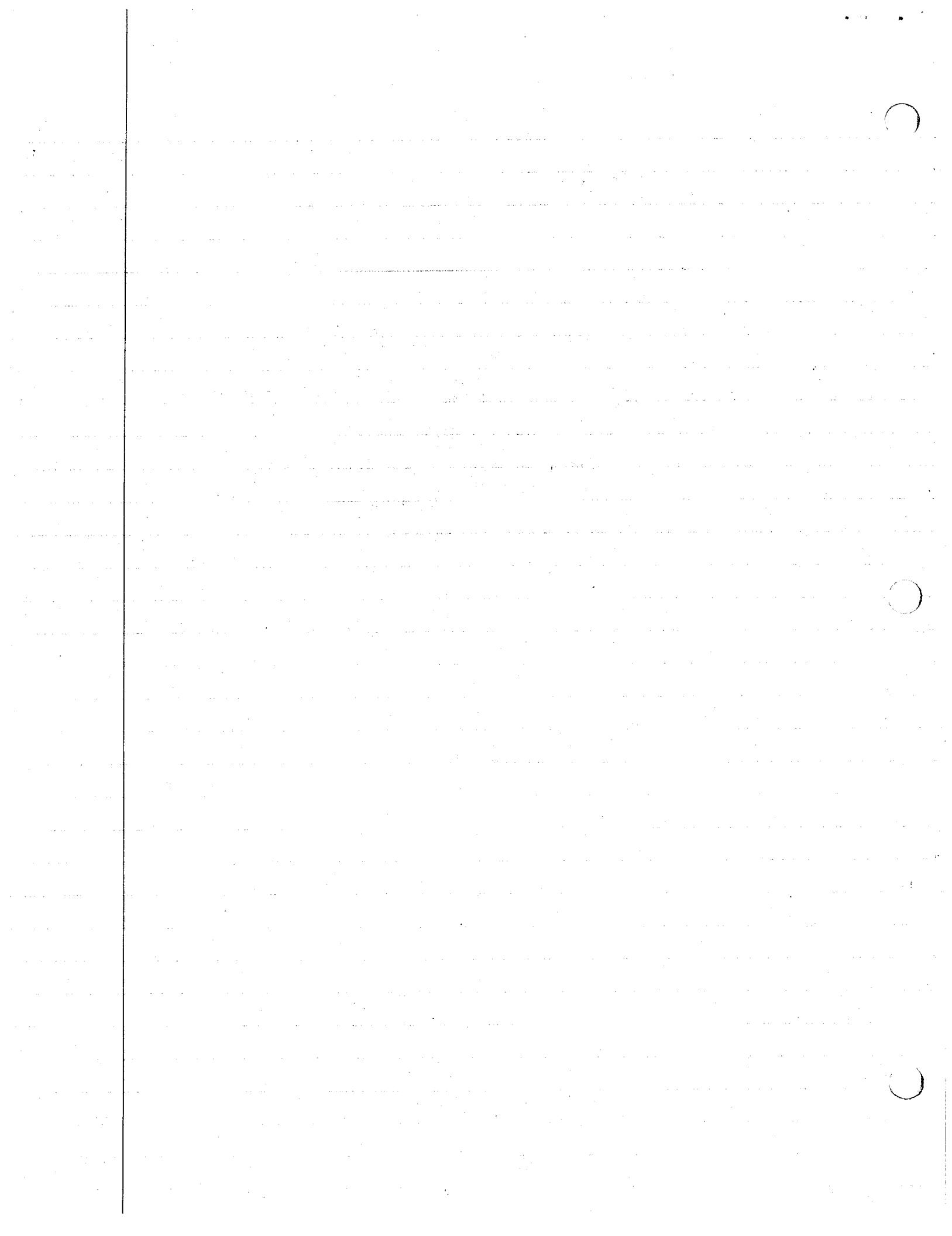
$$35+16 \approx 51 \quad \sqrt{16} \approx 4 \quad \frac{4.5}{24} \approx .18 \dots 2$$

$$\frac{1}{12}\sqrt{\frac{17}{2}} = \frac{13}{12 \cdot 4} = \frac{1}{9} \cdot 2.5$$

$$\text{thus } k_{\min} = \frac{1}{24} [35 - (241)^{1/2}]^{1/2} b$$

984

$\frac{35}{72}$   
 $\frac{35}{108}$   
 $\frac{175}{108}$   
 $\frac{125}{108}$   
 $\frac{984}{241}$



now  $m_a = \frac{\sqrt{3}}{2}m_2 + \frac{1}{2}m_1$ ,  $m_b = \frac{\sqrt{3}}{2}m_1 - \frac{1}{2}m_2$ ,  $m_c = m_2$   
 thus  $I_{lm} = \sum_{l=1}^3 \sum_{m=1}^3 a_{lj} I_{jkl} a_{km}$

7f  $I_{jkl}^{B/N} = I_{jkl}^{B/Q} + I_{jkl}^{Q/N}$   $I_{jkl}^{B/Q} = I_{jkl}^{B/Q} + I_{jkl}^{Q/Q}$   
 $\|P\|^{Q/N} = -3m_1 + 2m_2$   
 $\|P\|^{B/Q} = 13$   $I_{jkl}^{Q/N} = m [ 13\delta_{jk} - (\|P\| \cdot m_j)(\|P\| \cdot m_k) ]$   
 now  $I_{jkl}^{B/N} = I_{jkl}^{B/Q} - I_{jkl}^{Q/Q} + I_{jkl}^{Q/N}$   
 $\|P\|^{Q/Q} = -3m_1 - 4m_3 + 2m_2$   
 $\|P\|^{B/Q} = 29$   $I_{jkl}^{Q/Q} = m [ 29\delta_{jk} - (\|P\| \cdot m_j)(\|P\| \cdot m_k) ]$

$I_{jkl}^{Q/N}$	1	2	3
1	48	72	0
2	72	108	0
3	0	0	156

$I_{jkl}^{Q/Q}$	1	2	3
1	240	72	-144
2	72	300	96
3	-144	96	156

$I_{jkl}^{B/N}$	1	2	3
1	68	72	0
2	72	133	0
3	0	0	169

$$\text{II} = 68m_1m_1 + 72(m_1m_2 + m_2m_1) + 133m_2m_2 + 169m_3m_3$$

7g  $m = \frac{3}{5}m_1 - \frac{4}{5}m_3$ ,  $I_n = m \cdot \text{II} \cdot m = 68 \cdot \frac{9}{25} + 169 \left( \frac{16}{25} \right) = \frac{3316}{25}$  Same result  
 hence  $I_n = m \cdot \text{II} \cdot m$

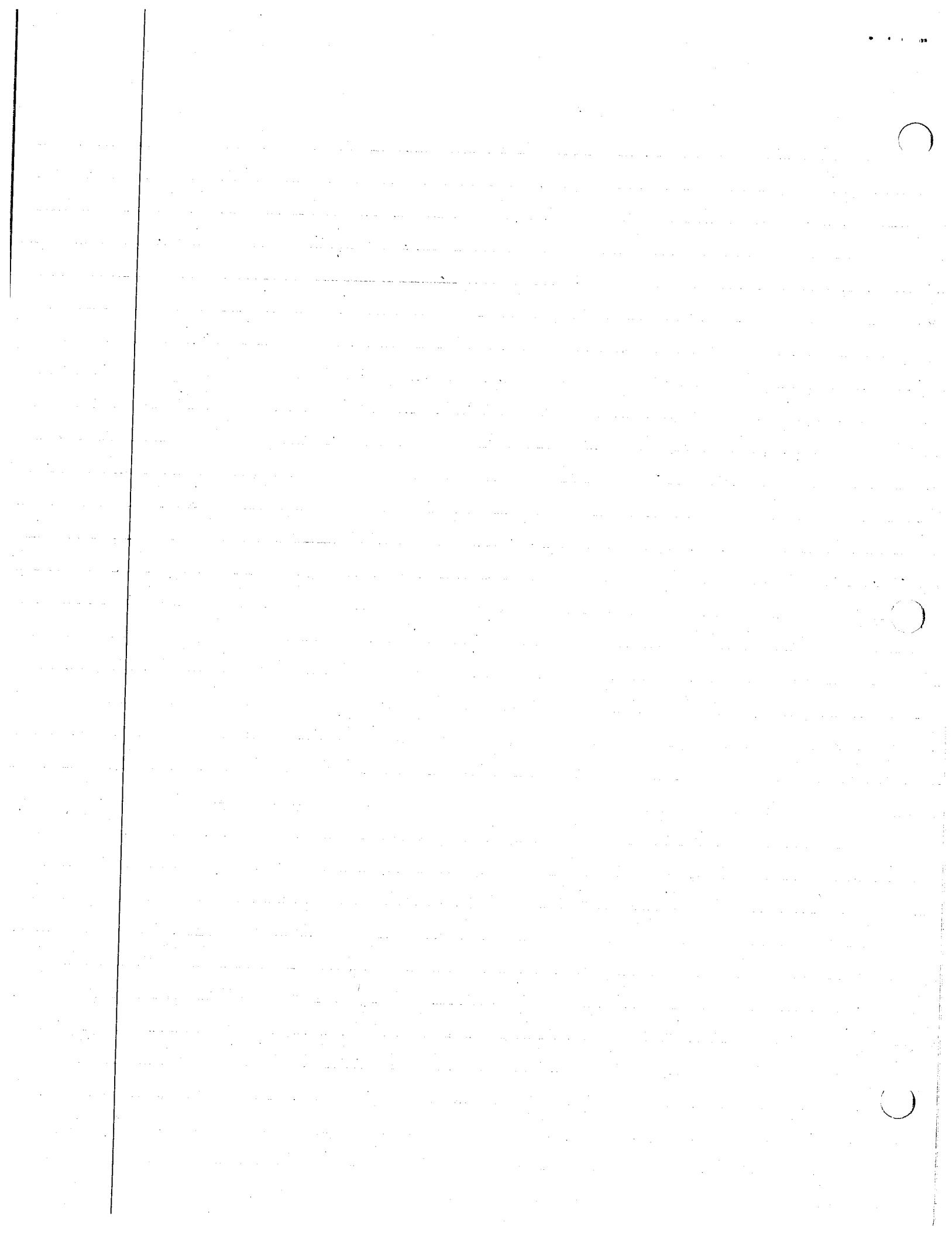
7h. Since  $\text{II}_a = \text{II} \cdot m_a = \sum m_i \|p_i\| \times (m_a \times \|p_i\|)$

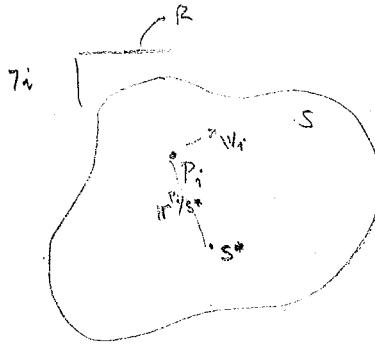
$$= \sum m_i [ (\|p_i\| \cdot p_i) m_a - (m_a \cdot p_i) \|p_i\| ] = \sum m_i [ \|p_i\|^2 ((m_a \cdot m_1)m_1 + (m_a \cdot m_2)m_2 + (m_a \cdot m_3)m_3) - (m_a \cdot \|p_i\|) \|p_i\| ]$$

$$= \sum m_i \{ m_a \cdot [ m_1m_1 + m_2m_2 + m_3m_3 ] \|p_i\|^2 - m_a \cdot ( \|p_i\| \cdot p_i ) \} = m_a \cdot \sum m_i ( \|p_i\|^2 - \|p_i\| p_i )$$

thus since  $\text{II} \cdot m_a = m_a \cdot \text{II} = m_a \cdot \sum m_i ( \|p_i\|^2 - \|p_i\| p_i ) \Rightarrow$

$$\text{II} = \sum m_i ( \|p_i\|^2 - \|p_i\| p_i )$$

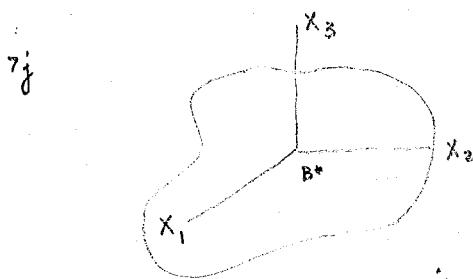




$${}^R\text{H}^{S/S^*} = \sum_{i=1}^N m_i r_i \times w_i$$

$$\text{now } w_i = w^S + \omega \times r_i$$

$$\begin{aligned} {}^R\text{H}^{S/S^*} &= \sum m_i r_i \times w^S + \sum m_i r_i \times (\omega \times r_i) \quad \text{Since } S^* \text{ is mass center} \\ {}^R\text{H}^{S/S^*} &= \sum m_i r_i \times (\omega \times r_i) = I \cdot \omega^B = I_w \end{aligned}$$

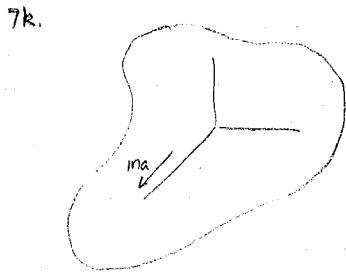


$$\omega^B_{\text{cm}} = w_i m_i \quad \text{where } w_i = \omega \cdot n_i$$

$${}^R\text{H}^{B/B^*} = I \cdot \omega$$

$$\text{since } X_i \text{'s are principal axes} \quad I = I_1 m_1 + I_2 m_2 + I_3 m_3$$

$$\therefore I \cdot \omega = I_1 w_1 m_1 + I_2 w_2 m_2 + I_3 w_3 m_3 = {}^R\text{H}^{B/B^*}$$



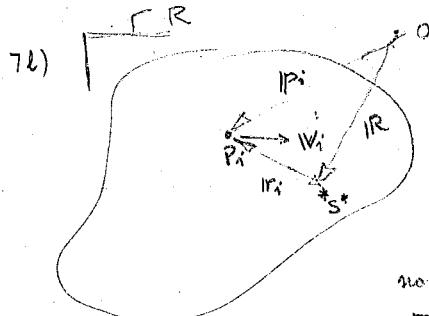
Show that if  ${}^R\text{H}^{B/B^*}$  is // to  $\omega^B$   $\Rightarrow$   $\omega^B$  is // to  $I_a$

and if  $\omega^B$  is // to  $I_a$   $\Rightarrow$   ${}^R\text{H}^{B/B^*}$  is // to  $\omega^B$

lets  $\omega = \omega m_a$  then lets  $m_a \cdot n_i = a_i$

if  $\omega$  is // to  $m_a$  then  $I_a = I \cdot m_a = \omega I \cdot m_a \Rightarrow I_a$  is // to  $\omega$

If  ${}^R\text{H}^{B/B^*}$  is // to  $\omega^B$  then  $I_a = H \cdot m_a = I \cdot \omega = \omega I \cdot m_a = \omega I_a \Rightarrow I_a = \frac{H}{\omega} m_a \Rightarrow I_a$  is // to  $m_a$   $\Rightarrow$   $m_a$  must be a vector in principal direction  $\Rightarrow \omega$  is // to a principal axis



$$\begin{aligned} {}^R\text{H}^{S/O} &= \sum m_i p_i \times w_i \quad \text{let } p_i = r_i^S + I^R \text{H}^{S/O} \\ &= \sum m_i (r_i + IR) \times (w^S + \omega^B \times r_i) \\ &= \sum m_i r_i \times w^S + \sum m_i r_i \times (\omega^B \times r_i) + \sum m_i R \times w^S \\ &\quad + \sum m_i R \times (\omega^B \times r_i) \end{aligned}$$

$$\text{now } \sum m_i r_i = 0 \quad \& \quad \sum m_i r_i \times (\omega \times r_i) = {}^R\text{H}^{S/S^*}$$

$$\sum m_i R \times w^S + \sum m_i R \times (\omega^B \times r_i) = mIR \times w^S + \sum m_i ((R \cdot r_i) \omega - (\omega \cdot IR))$$

$$\sum m_i (R \cdot r_i) \omega = [IR \cdot (\sum m_i r_i)] \omega = 0; \quad \sum m_i (\omega \cdot IR) r_i = (\omega \cdot IR) \sum m_i r_i = 0$$

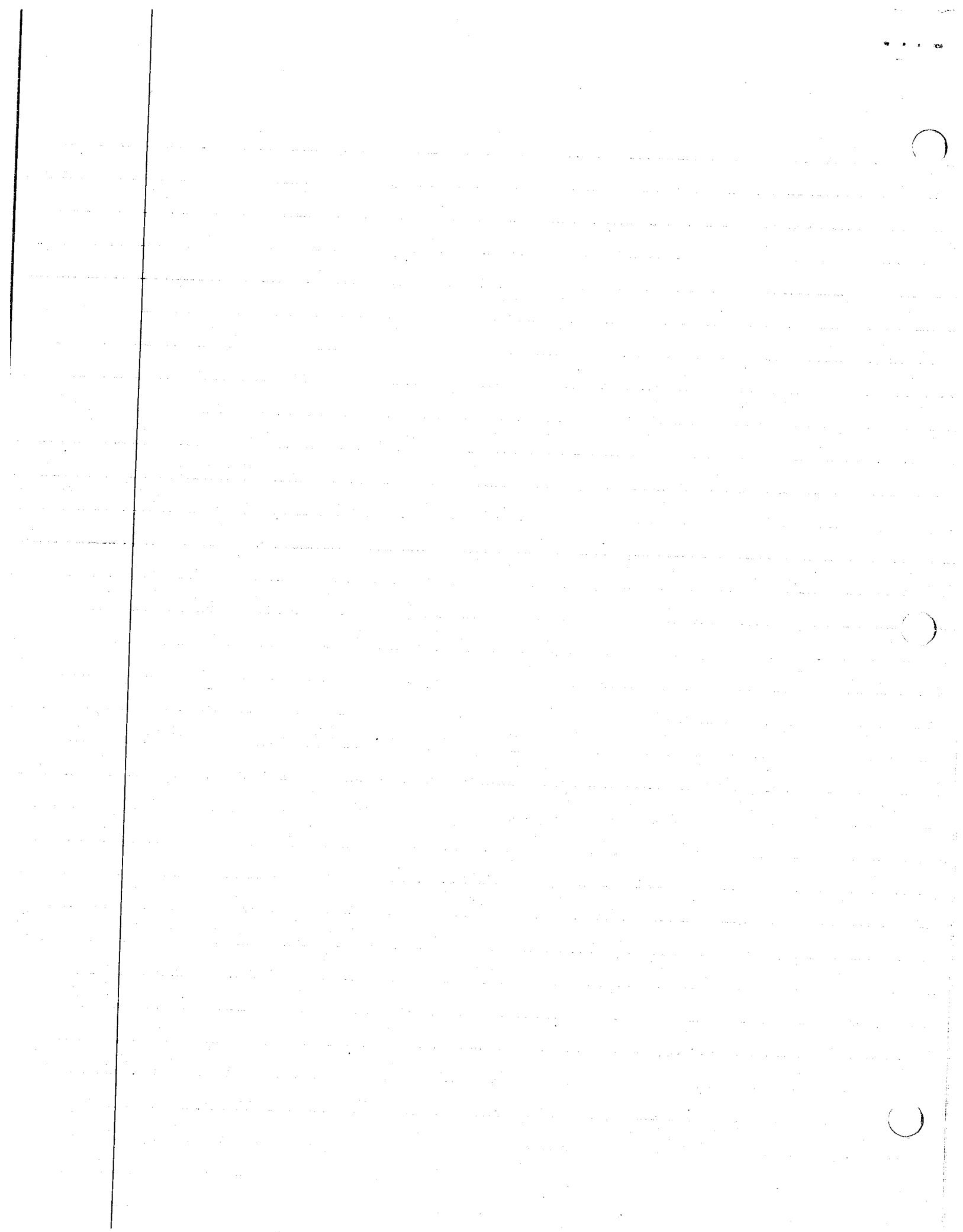
Thus

$${}^R\text{H}^{S/O} = {}^R\text{H}^{S/S^*} + mIR \times w^S$$

$$m = \sum m_i$$

$$\frac{mIR \times w^S}{mIR \times w^S} = \frac{{}^R\text{H}^{S/S^*}}{IR \times (m w^S)}$$

Angular momentum  
of a



8b) From problem 7b) we showed that  $\overset{R}{I} \cdot \overset{B}{\omega} = \overset{R}{I} \cdot \overset{B}{\alpha}$

$$\text{Now } \frac{d}{dt} (\overset{R}{I} \cdot \overset{B}{\omega}) = \frac{d}{dt} (\overset{R}{I} \cdot \overset{B}{\omega}) + \overset{R}{\omega} \times (\overset{R}{I} \cdot \overset{B}{\omega}) \quad \text{but what is } \frac{d}{dt} (\overset{R}{I} \cdot \overset{B}{\omega})$$

$$\overset{R}{I} \cdot \overset{B}{\omega} = \overset{R}{I}_{co} = \sum m_i \overset{R}{r}_i \times (\overset{R}{\omega} \times \overset{R}{r}_i) \quad \text{thus } \frac{d}{dt} (\overset{R}{I} \cdot \overset{B}{\omega}) = \sum m_i \overset{R}{r}_i \times (\overset{R}{\omega} \times \overset{R}{r}_i) + \\ \sum m_i \overset{R}{r}_i \times \left( \frac{d \overset{R}{\omega}}{dt} \times \overset{R}{r}_i \right) + \sum m_i \overset{R}{r}_i \times (\overset{R}{\omega} \times \overset{R}{r}_i)$$

but since  $\overset{R}{r}_i$  are fixed wrt body B hence  $\overset{R}{r}_i = \frac{d}{dt} \overset{R}{r}_i = 0$ . Thus 2nd & 3rd sums are zero.  $\frac{d \overset{R}{\omega}}{dt} = \overset{R}{\alpha} + \overset{R}{\omega} \times \overset{R}{\omega} = \overset{R}{\alpha}$  since  $\overset{R}{\omega} \times \overset{R}{\omega} = 0$

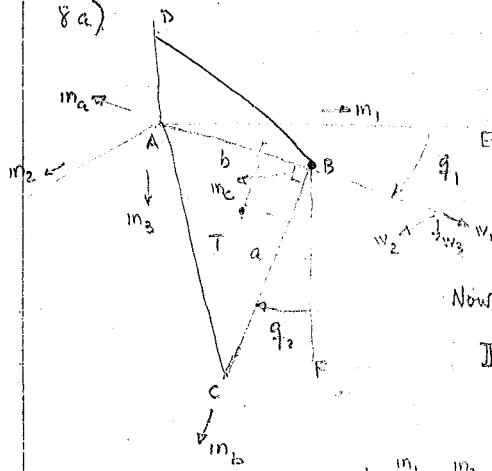
hence,

$$\frac{d}{dt} (\overset{R}{I} \cdot \overset{B}{\omega}) = \sum m_i \overset{R}{r}_i \times (\overset{R}{\alpha} \times \overset{R}{r}_i) = \overset{R}{I} \cdot \overset{R}{\alpha} \quad \text{thus}$$

$$\overset{R}{I} \cdot \overset{B}{\omega} = \overset{R}{I} \cdot \overset{R}{\alpha} + \overset{R}{\omega} \times (\overset{R}{I} \cdot \overset{B}{\omega}) = \overset{R}{I} \cdot \overset{R}{\alpha} - (\overset{R}{I} \cdot \overset{B}{\omega}) \times \overset{R}{\omega} = -\overset{R}{I}^*$$

$$\text{hence } \overset{R}{I}^* = -\overset{R}{I} \cdot \overset{B}{\omega} \quad \text{as required}$$

8a)



$$\overset{R}{\omega} = \dot{q}_1 m_3 + \dot{q}_2 m_a$$

$$F_2^* = W_2^* \cdot IF^* + \overset{R}{\omega} \dot{q}_2 \overset{R}{I}^*$$

$$\text{where } IF^* = m_a l^* \quad \text{and } \overset{R}{I}^* = (\overset{R}{I} \cdot \overset{B}{\omega}) \times \overset{R}{\omega} = \overset{R}{I} \cdot \overset{R}{\alpha}$$

and  $W^*$  is the velocity of the mass center

Now for the plate

$$\overset{R}{I} = \frac{m_a^2}{18} m_a m_a + \frac{mab}{36} (m_a m_b + m_b m_a) + \frac{mb^2}{18} m_b m_b + \frac{m(a^2+b^2)}{18} m_c m_c$$

$w_1$	$m_1$	$m_2$	$m_3$	$m_a$	$w_2$	$w_3$
	$c_1$	$s_1$	0	$m_a$	-1	0
$w_2$	$-s_1$	$c_1$	0	$m_b$	0	$s_2$
$w_3$	0	0	1	$m_c$	0	$c_2$

$$m_3 = w_3 = c_2 m_b + s_2 m_a$$

$$\overset{R}{\omega} = \dot{q}_1 (c_2 m_b - s_2 m_a) + \dot{q}_2 m_a$$

$$\overset{R}{\alpha} = \dot{q}_1 c_2$$

$$\overset{R}{\alpha} = \overset{R}{\alpha} + \overset{R}{\omega} \times \overset{R}{\omega} = \frac{d \overset{R}{\omega}}{dt}$$

$$\overset{R}{I} \cdot \overset{R}{\omega} = \overset{R}{I} \cdot \overset{R}{\omega} = \frac{m_a^2}{18} \dot{q}_1 m_a + \frac{mab}{36} \dot{q}_2 m_b + \frac{mab}{36} \dot{q}_1 c_2 m_a + \frac{mb^2}{18} \dot{q}_2 c_2 m_b + m(a^2+b^2) \left( -\dot{q}_1 s_2 m_c \right)$$

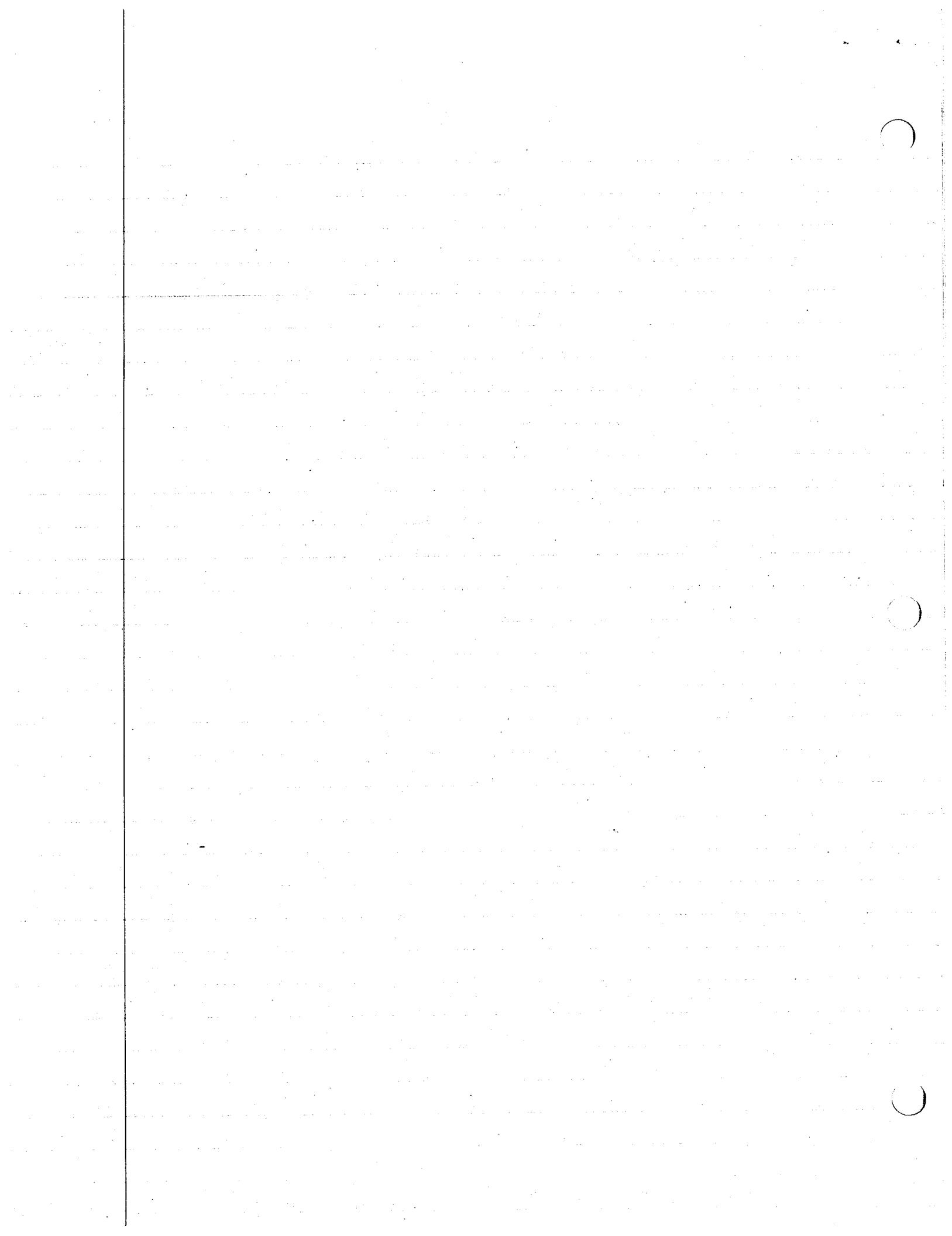
$$= \left( \frac{m_a^2}{18} \dot{q}_1 + \frac{mab}{36} \dot{q}_2 c_2 \right) m_a + \left( \frac{mab}{36} \dot{q}_2 + \frac{mb^2}{18} \dot{q}_1 c_2 \right) m_b + m(a^2+b^2) \dot{q}_1 s_2 m_c$$

$$\overset{R}{I} \cdot \overset{R}{\omega} \times \overset{R}{\omega} = -\dot{q}_2 \left[ \frac{mab}{36} \dot{q}_2 + \frac{mb^2}{18} \dot{q}_1 c_2 \right] m_c - m(a^2+b^2) \dot{q}_1 \dot{q}_2 s_2 m_b + \dot{q}_1 c_2 \left( \frac{ma^2}{18} \dot{q}_1 + \frac{mab}{36} \dot{q}_1 c_2 \right) m_c +$$

$$\dot{q}_1 s_2 \left( \frac{ma^2}{18} \dot{q}_2 + \frac{mab}{36} \dot{q}_1 c_2 \right) m_b = \dot{q}_1 s_2 \left( \frac{mab}{36} \dot{q}_2 + \frac{mb^2}{18} \dot{q}_1 c_2 \right) m_a + \dot{q}_1 c_2 \left[ m \left( \frac{a^2+b^2}{18} \right) \dot{q}_1 s_2 \right] m_a$$

$$\overset{R}{\alpha} = \frac{d \overset{R}{\omega}}{dt} = \dot{q}_1 (c_2 m_b - s_2 m_a) + \dot{q}_1 (-s_2 \dot{q}_2 m_b - c_2 \dot{q}_2 m_a) + \dot{q}_2 m_a = \dot{q}_2 m_a + (\dot{q}_1 c_2 - \dot{q}_1 \dot{q}_2 s_2) m_b - (\dot{q}_1 s_2 + \dot{q}_1 \dot{q}_2 c_2) m_a$$

$$\rightarrow \overset{R}{I} \cdot \overset{R}{\alpha} = \frac{ma^2}{18} \dot{q}_2 m_a + \frac{mab}{36} \dot{q}_2 m_b + \frac{mab}{36} (\dot{q}_1 c_2 - \dot{q}_1 \dot{q}_2 s_2) m_a + \frac{mb^2}{18} (\dot{q}_1 c_2 - \dot{q}_1 \dot{q}_2 s_2) m_b + m(a^2+b^2) (\dot{q}_1 s_2 + \dot{q}_1 \dot{q}_2 c_2) m_a$$



$$\therefore \omega_{q_2}^* \cdot \Pi^* = m_a \cdot \Pi^* = -\dot{q}_1 s_2 \left( \frac{mab}{36} \dot{q}_2 + \frac{mb^2}{18} \dot{q}_1 c_2 \right) + \dot{q}_1 c_2 \left[ \frac{m(a^2+b^2)}{18} \dot{q}_1 s_2 \right] \\ - \frac{ma^2}{18} \dot{q}_2 - \frac{mab}{36} (\ddot{q}_1 c_2 - \dot{q}_1 \dot{q}_2 s_2)$$

now  $\text{IR}^T = -\frac{2}{3} b m_a + \frac{2}{3} m_b$

$$R^T = \text{IV}^T + G^T \times \text{IR}^T = \dot{q}_1 (c_2 m_b - s_2 m_a) + \dot{q}_2 m_a \left( -\frac{2b}{3} m_a + \frac{2}{3} m_b \right)$$

$$= \frac{2b}{3} \dot{q}_1 c_2 m_a + \dot{q}_1 s_2 \cdot \frac{2b}{3} m_b + \dot{q}_2 s_2 m_a + \dot{q}_2 c_2 m_a$$

$$= \dot{q}_1 s_2 m_a + \dot{q}_1 s_2 \cdot \frac{2b}{3} m_b + \left( \frac{2b}{3} \dot{q}_1 c_2 + \dot{q}_2 c_2 \right) m_a$$

$$AI^T = \text{AI}^T + \text{IW}^T \times W$$

$$AI^T = \frac{2b}{3} m_a (\ddot{q}_1 c_2 + \dot{q}_1 s_2 \dot{q}_2) + \frac{2b}{3} m_b (\ddot{q}_1 s_2 + \dot{q}_1 \dot{q}_2 c_2) + (\ddot{q}_1 s_2 + \dot{q}_1 \dot{q}_2 c_2) m_a + \dot{q}_2 \dot{q}_2 m_a$$

$$\text{IW} \times W = \dot{q}_2 \dot{q}_1 s_2 \frac{2b}{3} m_a - \dot{q}_2 \left( \frac{2b}{3} \dot{q}_1 c_2 + \dot{q}_2 c_2 \right) m_b - \dot{q}_1 c_2 (\frac{2b}{3} \dot{q}_1 c_2 + \dot{q}_2 c_2) m_a + \dot{q}_1 c_2 \left( \frac{2b}{3} \dot{q}_1 c_2 + \dot{q}_2 c_2 \right) m_a \\ - \dot{q}_1 s_2 (\dot{q}_1 s_2) m_b + \dot{q}_1 s_2 \left( \dot{q}_1 s_2 \cdot \frac{2b}{3} \right) m_a$$

$$\text{then } W_{q_2}^* \cdot AI^T = \frac{2ba}{9} (\ddot{q}_1 c_2 - \dot{q}_1 s_2 \dot{q}_2) + \dot{q}_2 \dot{q}_2 + \dot{q}_2 \dot{q}_1 s_2 \frac{2ba}{9} - \dot{q}_1 \dot{q}_1 c_2 s_2 \frac{a^2}{9}$$

$$\text{hence } F_2^* = -\frac{2bma}{9} (\ddot{q}_1 c_2 + \dot{q}_1 \dot{q}_2 s_2) - \frac{m \dot{q}_2^2}{9} - m \dot{q}_2 \dot{q}_1 s_2 \cdot \frac{2ba}{9} + m \dot{q}_1^2 c_2 s_2 a^2 \\ = \dot{q}_1 \dot{q}_2 \frac{mab}{36} s_2 - \dot{q}_1^2 \frac{mb^2}{18} s_2 c_2 + \dot{q}_1^2 c_2 s_2 \frac{m(a^2+b^2)}{18} - \frac{ma^2}{18} \dot{q}_2 - \frac{mab}{36} \dot{q}_1 c_2 + \frac{mab}{36} \dot{q}_1 \dot{q}_2 s_2 \\ = \dot{q}_1 \left[ -\frac{2bma}{9} c_2 - \frac{mab}{36} c_2 \right] + \dot{q}_2 \left[ -\frac{ma^2}{9} - \frac{ma^2}{18} \right] + \dot{q}_1^2 \left[ \frac{ma^2}{9} c_2 s_2 - \frac{mb^2}{18} s_2 c_2 + \frac{m(a^2+b^2)}{18} c_2 s_2 \right] \\ + \dot{q}_1 \dot{q}_2 \left[ \frac{2mba}{9} s_2 - \frac{2mba}{9} s_2 - \frac{mabs_2}{36} + \frac{mabs_2}{36} \right]$$

$$= \dot{q}_1 \left[ -\frac{1}{4} mba c_2 \right] + \dot{q}_2 \left[ -\frac{ma^2}{6} \right] + \dot{q}_1^2 \left[ \frac{ma^2}{6} c_2 s_2 \right]$$

$$= \frac{ma}{12} \left[ 2a (c_2 s_2 \dot{q}_1^2 - \dot{q}_2) - 3 \dot{q}_1 b c_2 \right]$$



$m_c, m_b, m_a$  are fixed in plane of ring

$$F_0^* = W_0^* \cdot I^* + \omega_0^* \cdot \Pi^*$$

$$\text{where } I^* = -m a^2 \quad \omega = \theta m_c = SLIK \quad R = (R^2 - l^2)^{1/2} m_a$$

$$\text{from 5a we found } W^P = (R^2 - l^2)^{1/2} [\theta \cos \theta m_b - \theta \sin \theta IK - SLIK]$$

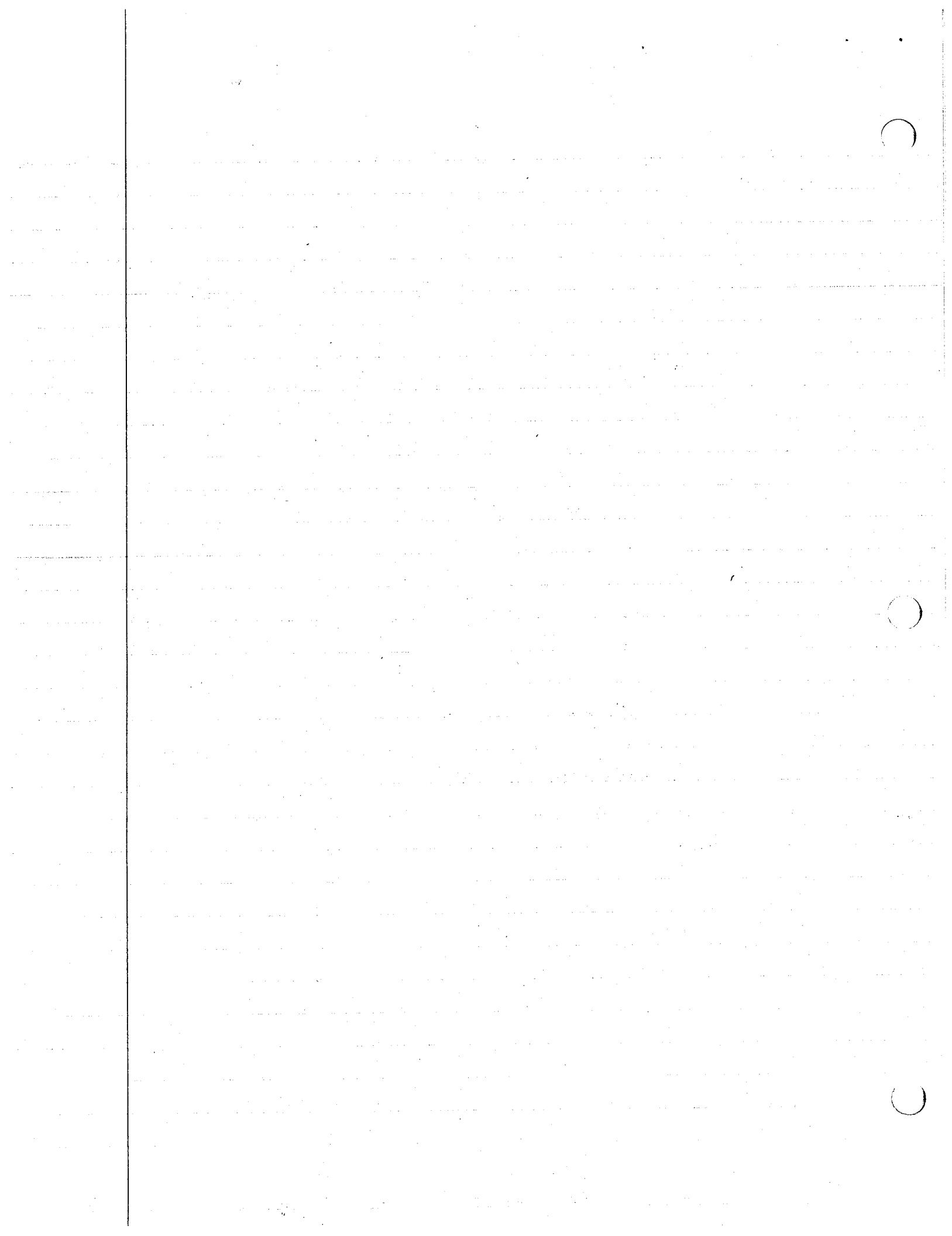
$$\text{then, } W_0^* = (R^2 - l^2)^{1/2} [\cos \theta m_b - \sin \theta IK] = (R^2 - l^2)^{1/2} [\cos \theta m_b - S' \theta m_a]$$

$$\text{now, } AI^* = \frac{dW^P}{dt} = \frac{dW^P}{dt} + \omega \times W^P \quad \frac{dW^P}{dt} = SLIK$$

$$\frac{dW^P}{dt} = (R^2 - l^2)^{1/2} \left[ \theta \sin \theta m_b - \theta \cos \theta m_b - \theta \sin \theta IK - \theta \cos \theta IK - SLIK \right]$$

$$\omega \times W^P = (R^2 - l^2)^{1/2} \left[ -\theta \sin \theta m_c - \theta \cos \theta m_c - SLIK \right]$$

$$AI^* = (R^2 - l^2)^{1/2} \left[ (-\theta^2 \sin^2 \theta - \theta^2 \cos^2 \theta) IK + (\theta^2 S - \theta^2 L - \theta^2 C) m_b - 2SLIK m_a \right]$$



$$\text{Thus, } W_g \cdot I F^* = +m(R^2 L^2) \left[ -S(\dot{\theta}^2 c + \ddot{\theta} s) + (\dot{\theta}^2 s + \ddot{\theta} s - \ddot{\theta} c)c \right] = m(R^2 L^2) \left[ +\ddot{\theta}^2 c s - \ddot{\theta}(s^2 + c^2) \right] \\ \omega_g = m_c$$

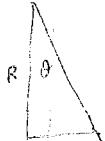
Now

$\ddot{\theta}$	$m_a$	$m_b$	$m_c$
$m_a$	0	0	0
$m_b$	0	$m_b L^2$	0
$m_c$	0	0	$\frac{m_c L^2}{3}$

$$I = \frac{m L^2}{3} (m_a m_b + m_b m_c) \quad m_a = c m_b R K + s \theta m_b \\ m_a m_b = c \theta^2 R m_b R K + \frac{s^2 \theta}{2} (m_a m_b + m_b m_c) \\ I \propto \frac{R \text{ rod}}{dt} = \frac{m L^2}{3} (c^2 R K + s c [k m_b + m_b R K] + s^2 m_b m_b + m_c m_c) \\ (\ddot{\theta} m_c = \ddot{\theta} m_b)$$

$$(I \cdot \omega) \times \omega = \frac{m L^2}{3} \{ \ddot{\theta} m_a \cdot c^2 R K + S c c m_b \} \times \{ \ddot{\theta} m_c - S R K \} = \frac{m L^2}{3} \{ \ddot{\theta} \ddot{\theta} m_b + c^2 S \ddot{\theta} m_b - S c c \ddot{\theta} m_a - S^2 c \ddot{\theta} m_c \}$$

$$\alpha \propto \frac{R \text{ rod}}{dt} = \frac{R \text{ rod}}{dt} = \frac{d^2 \omega}{dt^2} \text{ ROD} + \frac{R \text{ rod}}{dt} \times \frac{R \text{ rod}}{dt} \text{ ROD} = \ddot{\theta} m_c + (-S R K) \times (\ddot{\theta} m_c - S R K) = \ddot{\theta} m_c + S R \ddot{\theta} m_b \\ \text{since } m_b, m_a, m_c \text{ are fixed in ring}$$



$$\text{now } I \cdot \alpha = \frac{m L^2}{3} \{ c^2 R K + s c [k m_b + m_b R K] + s^2 m_b m_b + m_c m_c \} \cdot (\ddot{\theta} m_c + S R \ddot{\theta} m_b) \\ = \frac{m L^2}{3} \{ \ddot{\theta} m_c + S \theta S c R K + S \theta S^2 m_b \}$$

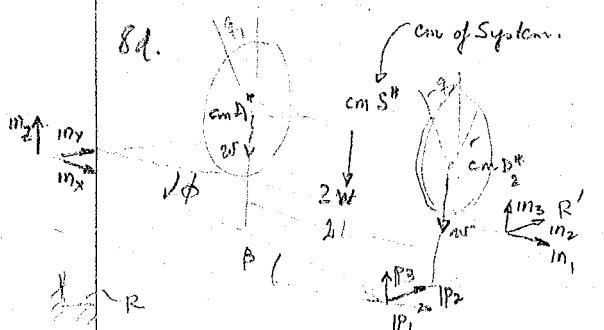
$$I^* = (I \cdot \omega) \times \omega - I \cdot \alpha = \frac{m L^2}{3} \{ (-2 S c \dot{\theta}) m_a + (-\dot{\theta} S m_b + c^2 S \dot{\theta} - S \dot{\theta} S^2) m_b + (-S^2 c - \ddot{\theta}) m_c \}$$

$$\text{now } \omega_g \cdot I^* = -(\ddot{\theta}^2 c + \ddot{\theta}) \frac{m L^2}{3}$$

Hence

$$F_\theta^* = W_g \cdot I F^* + \omega_g \cdot I^* = \frac{3m(R^2 L^2)}{3} (\ddot{\theta}^2 c + \ddot{\theta}) - \frac{m L^2}{3} (\ddot{\theta}^2 c + \ddot{\theta}) \\ = -\frac{W}{3g} \left[ \frac{(3R^2 - 3L^2 + L^2)}{(3R^2 - 2L^2)} \ddot{\theta} + S^2 c (4L^2 - 3R^2) \right] \\ = -\frac{W}{3g} \left[ (3R^2 - 2L^2) \ddot{\theta} + S^2 c (4L^2 - 3R^2) \right] \quad \text{when } ?$$

8d. Cm of System.



$$\text{now } (F_r)^* = W_{qr}^* \cdot I F^* + \omega_{qr} \cdot I^* \quad \text{where } I F^* = -m_a \alpha^*$$

$$\text{now } W_{q_1}^{D_2} = 0 \quad W_{q_2}^{D_2} = r ( \sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2 ) = r q_2 (\sin \phi m_x - \cos \phi m_y)$$

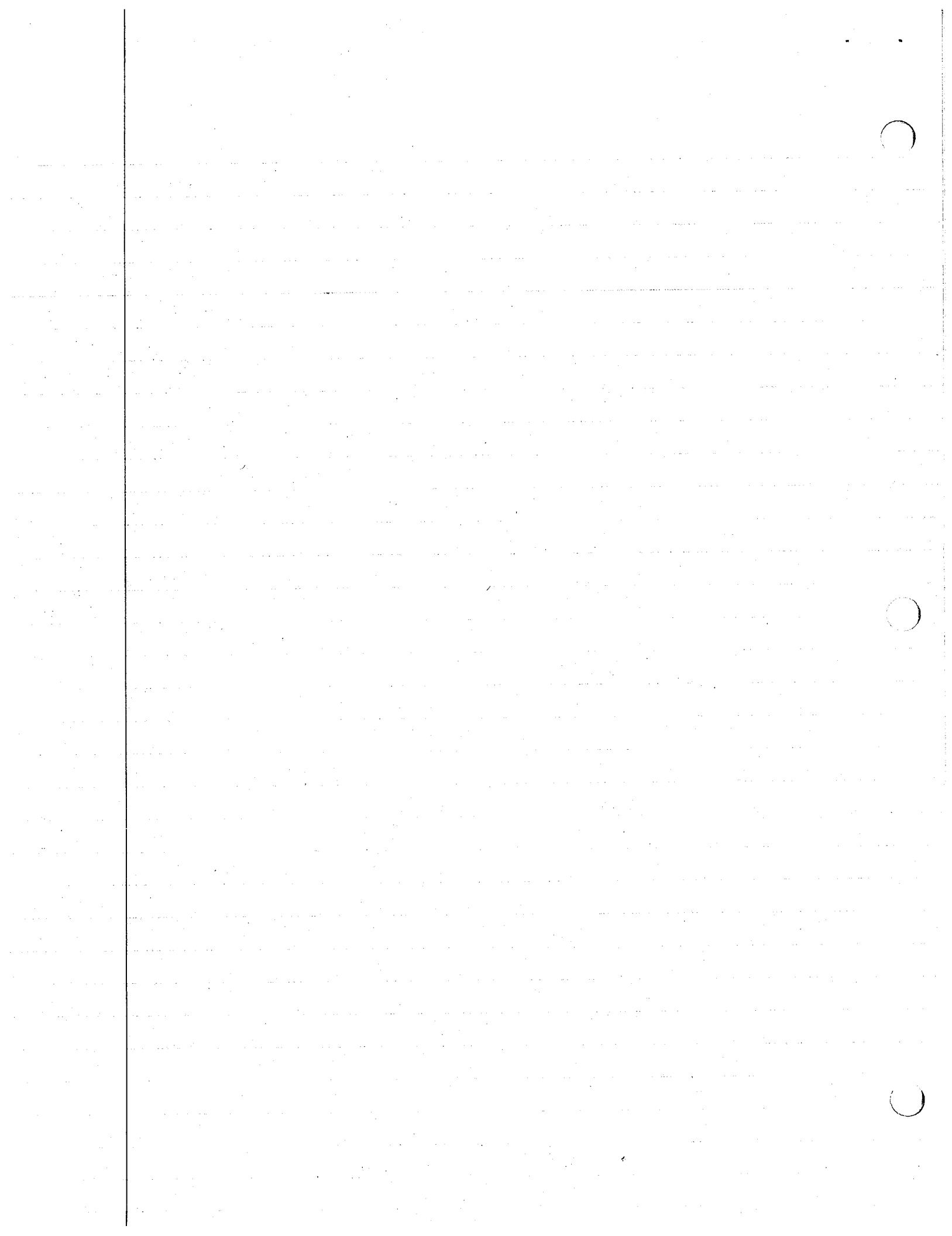
$$W_{q_1}^{D_1} = r q_1 (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) \quad W_{q_2}^{D_1} = 0$$

$$W_{q_1}^{S^*} = \frac{r}{2} (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) = W_{q_2}^{S^*}$$

$$W_{q_1}^{D_2} = r q_2 (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) + r q_2 \phi (\cos \phi \cos \beta p_1 - \cos \phi \sin \beta p_3 + \sin \phi p_2)$$

$$W_{q_1}^{D_1} =$$

$$W_{q_1}^{D_2} = r q_1 (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) = r q_1 (\sin \phi m_x - \cos \phi m_y) \\ W_{q_2}^{D_1} = r q_2 (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) = r q_2 (\sin \phi m_x - \cos \phi m_y) \\ W_{q_1}^{S^*} = \frac{r}{2} (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) = W_{q_2}^{S^*} \\ W_{q_1}^{D_1} = r q_1 (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) = r q_1 (\sin \phi m_x - \cos \phi m_y) \\ W_{q_2}^{D_1} = r q_2 (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) = r q_2 (\sin \phi m_x - \cos \phi m_y) \\ W_{q_1}^{S^*} = \frac{r}{2} (\sin \phi \cos \beta p_1 - \sin \phi \sin \beta p_3 - \cos \phi p_2) = W_{q_2}^{S^*}$$



$$\begin{aligned} \alpha'^{D_2} &= -r\ddot{q}_2 m_2 - r\dot{q}_2 (-\dot{\phi}m_1 + \dot{q}_2 m_3) \\ \alpha'^{D_1} &= -r\ddot{q}_1 m_2 - r\dot{q}_1 (-\dot{\phi}m_1 + \dot{q}_1 m_3) \\ \alpha'^{S^*} &= -r_2(\ddot{q}_1 + \ddot{q}_2)m_2 - r_2(\dot{q}_1 + \dot{q}_2)(-\dot{\phi}m_1) \\ \Pi^S &= \frac{2ML^2}{3}(m_1m_2 + m_2m_3) = \frac{2WL^2}{3g}(m_1m_2 + m_2m_3). \end{aligned}$$

$$\begin{aligned} W^{D_2} &= -r\dot{q}_2 m_2 & \dot{\phi} &= \frac{r}{2L} [\ddot{q}_1 + \ddot{q}_2] \\ W^{D_1} &= -r\dot{q}_1 m_2 & \\ W^{S^*} &= -r_2(\dot{q}_1 + \dot{q}_2)m_2 & \end{aligned}$$

$$\Pi^{D_2} = \Pi^{D_1} = \frac{Wr^2}{4g} [2m_1m_1 + m_2m_2 + m_3m_3]$$

$$\begin{aligned} \omega'^{D_2} &= \dot{\phi}m_3 + \dot{q}_2m_1 & \omega'^{D_1} &= \dot{\phi}m_3 + \dot{q}_1m_1 \\ \omega'^S &= \dot{\phi}m_3; \quad \alpha'^{D_2} = \dot{\phi}m_3 + \dot{q}_2\dot{\phi}m_2 & \\ \alpha'^{D_1} &= \dot{\phi}m_3 + \dot{q}_1m_1 + \dot{q}_1\dot{\phi}m_2; \quad \alpha'^{S^*} = \dot{\phi}m_3 & \end{aligned}$$

$$(\Pi^D \cdot \omega) = \frac{Wr^2}{4g} [2\dot{q}_1m_1 + \dot{\phi}m_3]$$

$$(\Pi^D \cdot \omega) \times \omega = \frac{Wr^2}{4g} [2\dot{q}_1\dot{\phi}m_2 + \dot{\phi}\dot{q}_1m_2] = -\frac{Wr^2}{4g} \dot{q}_1\dot{\phi}m_2$$

$$\Pi^{D_2} \cdot \omega = \frac{Wr^2}{4g} [2\dot{q}_2m_1]$$

$$(\Pi^D \cdot \omega) \times \omega = -\frac{Wr^2}{4g} [\dot{q}_2\dot{\phi}m_2]$$

$$\Pi^D \cdot \alpha = \frac{Wr^2}{4g} [2\dot{q}_1m_1 + \dot{q}_1\dot{\phi}m_2 + \dot{\phi}m_3]$$

$$\Pi^{D_2} \cdot \alpha = \frac{Wr^2}{4g} [2\dot{q}_2m_1 + \dot{q}_2\dot{\phi}m_2 + \dot{\phi}m_3]$$

$$\omega_{\dot{q}_1}^{D_1} = \left( \frac{r}{2L} m_3 + m_1 \right)$$

$$\omega_{\dot{q}_2}^{D_1} = -\frac{r}{2L} m_3$$

$$\omega_{\dot{q}_1}^{D_2} = \frac{r}{2L} m_3$$

$$\omega_{\dot{q}_2}^{D_2} = -\frac{r}{2L} m_3 + m_1$$

$$\omega_{\dot{q}_1}^S = \frac{r}{2L} m_3$$

$$\omega_{\dot{q}_2}^S = \frac{r}{2L} m_3; \quad (\Pi^S \cdot \omega) = \frac{2WL^2}{3g} (\dot{\phi}m_3); \quad (\Pi^S \cdot \omega) \times \omega = 0 \quad \Pi^S \alpha = \frac{2WL^2}{3g} \dot{\phi}m_3$$

$$\text{Thus } \omega_{\dot{q}_1}^{D_1} \cdot \Pi^{D_1} = \frac{r}{2L} \left( -\dot{\phi} \frac{Wr^2}{4g} \right) = \frac{Wr^2}{4g} (2\ddot{q}_1)$$

$$\omega_{\dot{q}_2}^{D_1} \cdot \Pi^{D_1} = -\frac{r}{2L} \left[ \frac{-Wr^2 \dot{\phi}}{4g} \right]$$

$$\omega_{\dot{q}_1}^{D_2} \cdot \Pi^{D_2} = +\frac{r}{2L} \left[ \frac{-Wr^2 \dot{\phi}}{4g} \right]$$

$$\omega_{\dot{q}_2}^{D_2} \cdot \Pi^{D_2} = -\frac{r}{2L} \left[ \frac{-Wr^2 \dot{\phi}}{4g} \right] - \frac{Wr^2}{4g} \cdot 2\ddot{q}_2$$

$$\omega_{\dot{q}_1}^S \cdot \Pi^S = \frac{r}{2L} \cdot \frac{2WL^2}{3g} \dot{\phi}$$

$$\omega_{\dot{q}_2}^S \cdot \Pi^S = +\frac{r}{2L} \cdot \frac{2WL^2}{3g} \dot{\phi}$$

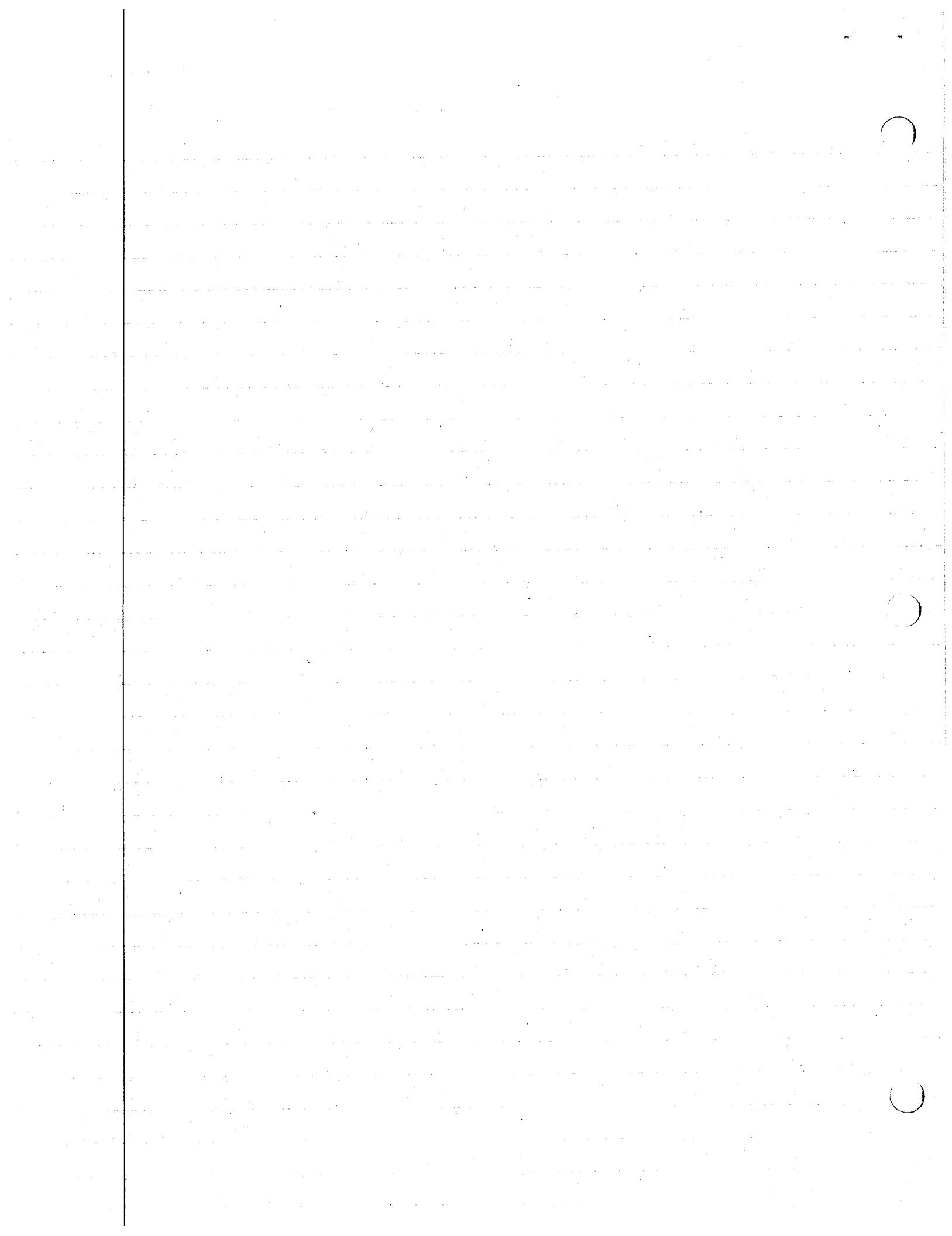
$$\text{Now } W_{\dot{q}_1}^{D_2} = 0 \quad W_{\dot{q}_2}^{D_2} = -r m_2 \quad W_{\dot{q}_1}^{D_1} = -r m_2 \quad W_{\dot{q}_2}^{D_1} = 0 \quad W_{\dot{q}_1}^S = -r_2 m_2 \quad W_{\dot{q}_2}^S = -r_2 m_2.$$

$$-MW_{\dot{q}_1}^{D_2} \cdot \alpha_1^{D_2} = 0; \quad -MW_{\dot{q}_2}^{D_2} \cdot \alpha_1^{D_2} = -Mr^2 \ddot{q}_2; \quad -MW_{\dot{q}_1}^{D_1} \cdot \alpha_1^{D_1} = Mr^2 \ddot{q}_1; \quad -MW_{\dot{q}_2}^{D_1} \cdot \alpha_1^{D_1} = 0;$$

$$-MW_{\dot{q}_1}^S \cdot \alpha_1^S = -Mr^2 \ddot{q}_1 (\ddot{q}_1 + \ddot{q}_2); \quad -MW_{\dot{q}_2}^S \cdot \alpha_1^S = -Mr^2 \ddot{q}_2 (\ddot{q}_1 + \ddot{q}_2)$$

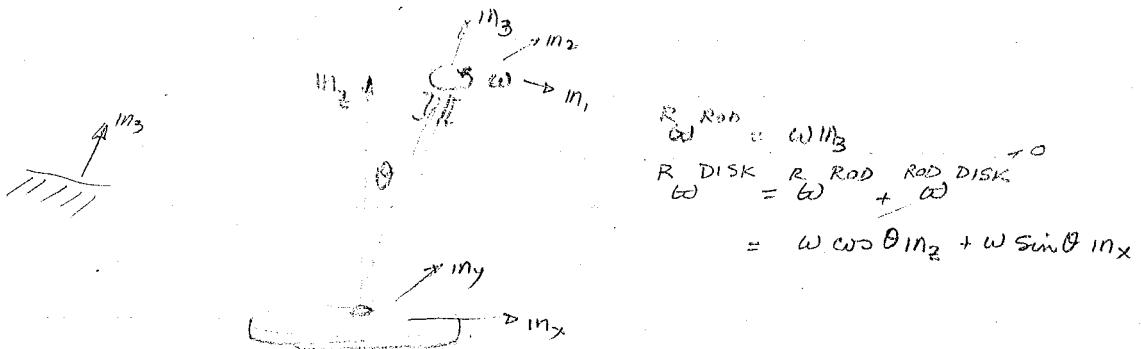
Now

$$\begin{aligned} F_1^* &= \left[ -\frac{2Wr^2}{4g} (\ddot{q}_1 + \ddot{q}_2) - \frac{Mr^2}{8} \ddot{q}_1 + \frac{r}{2L} \cdot \frac{2WL^2}{3g} \cdot \frac{r}{2L} (\ddot{q}_1 + \ddot{q}_2) - \frac{r}{2L} \left[ \frac{Wr^2}{4g} \cdot \frac{r}{2L} (\ddot{q}_1 + \ddot{q}_2) \right] - \frac{Wr^2}{4g} \cdot 2\ddot{q}_1 \right. \\ &\quad \left. - \frac{r}{2L} \left[ \frac{Wr^2}{4g} \cdot \frac{r}{2L} (\ddot{q}_1 + \ddot{q}_2) \right] \right] = -\frac{3Wr^2}{2g} \ddot{q}_1 + \frac{Wr^4}{8gL^2} (\ddot{q}_1 + \ddot{q}_2) + \frac{Wr^2}{6g} (\ddot{q}_1 + \ddot{q}_2) - \frac{Wr^2}{2g} (\ddot{q}_1 + \ddot{q}_2) \\ &= -\ddot{q}_1 \left( +\frac{3Wr^2}{2g} + \frac{Wr^4}{8gL^2} + \frac{2Wr^2}{3g} \right) - \ddot{q}_2 \left( \frac{Wr^2}{3g} + \frac{Wr^4}{8gL^2} \right) - \frac{Wr^2}{g} \left( -\frac{2}{3} \ddot{q}_1 - \frac{1}{3} \ddot{q}_2 \right) \end{aligned}$$



$$\begin{aligned}
 F_2^* &= \left[ -\frac{2Wr^2}{g} (\ddot{q}_1 + \ddot{q}_2) - \frac{Wr^2}{g} \ddot{q}_2 + \frac{r^2}{2L^2} \cdot \frac{2WL^2}{3g} (\ddot{q}_1 - \ddot{q}_2) + \frac{r^4}{4L^2} \frac{W}{4g} (\ddot{q}_1 - \ddot{q}_2) + \frac{r^4}{4L^2} \frac{W}{4g} (\ddot{q}_1 - \ddot{q}_2) \right. \\
 &\quad \left. - \frac{Wr^2}{2g} \ddot{q}_2 \right] = -\frac{3r^2}{2g} \ddot{q}_2 + \frac{r^4}{8L^2g} (\ddot{q}_1 - \ddot{q}_2) + \frac{Wr^2}{4g} \left( \frac{1}{3}\ddot{q}_1 - \frac{1}{3}\ddot{q}_2 \right) \\
 &= -\ddot{q}_1 \left( -\frac{r^4 W}{8gL^2} + \frac{1}{3} \frac{Wr^2}{g} \right) + \ddot{q}_2 \left( \frac{r^4 W}{8L^2g} + \frac{3r^2 W}{2g} - \frac{2}{3} \frac{Wr^2}{g} \right)
 \end{aligned}$$

8e.



$$\begin{aligned}
 \dot{\omega}_{RAD} &= \omega M_3 \\
 \dot{\omega}_{DISK} &= \dot{\omega}_{RAD} + \dot{\omega}_{DISK} \\
 &= \omega \cos \theta m_2 + \omega \sin \theta m_3
 \end{aligned}$$

what is  $\Pi^*$  disk  $M_2 = \cos \theta m_1$

$$\Pi = \frac{mR^2}{2} m_2 m_2 + \frac{mR^2}{4} (m_x m_x + m_y m_y)$$

$$\Pi \cdot \omega = \frac{mR^2}{2} \omega \cos \theta m_2 + \frac{mR^2}{4} \omega \sin \theta m_x$$

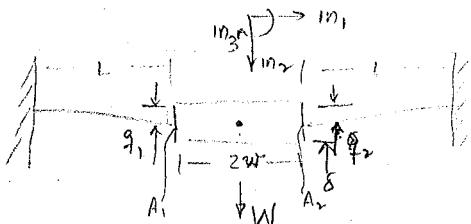
$$\begin{aligned}
 (\Pi \cdot \omega) \times \omega &= \frac{mR^2}{2} \omega^2 \sin \theta \cos \theta m_y - \frac{mR^2}{4} \omega^2 \sin \theta \cos \theta m_y \\
 &= \frac{mR^2 \omega^2}{4} \sin \theta \cos \theta m_y = \frac{mR^2 \omega^2}{8} \sin 2\theta m_y
 \end{aligned}$$

$$\alpha^{RAD} = \dot{\omega} m_3 \text{ but } \omega = \text{const} \therefore \alpha = 0 \Rightarrow \Pi \cdot \alpha = 0$$

$$\Pi^* = (\Pi \cdot \omega) \times \omega - \Pi \cdot \alpha = \frac{mR^2 \omega^2}{8} \sin 2\theta m_y$$

D'Alembert Principal  $\bar{T} + \Pi^* = 0 \Rightarrow \Pi^* = -\Pi = -\Pi = \text{torque applied by bearings}$

8f.



Assume  $\delta \ll 2w$  thus block is like a thin rod.

We have found  $\omega^B = (-\dot{q}_1 + \dot{q}_2)/2w m_3$  and that  $N^{A_1} = \dot{q}_1 m_2$ ,  $N^{A_2} = \dot{q}_2 m_2$ .

$$\begin{aligned}
 \theta_2 &= \frac{q_2 - q_1}{2w} \\
 \theta_1 &= \frac{q_1 - q_2}{2w}
 \end{aligned}$$

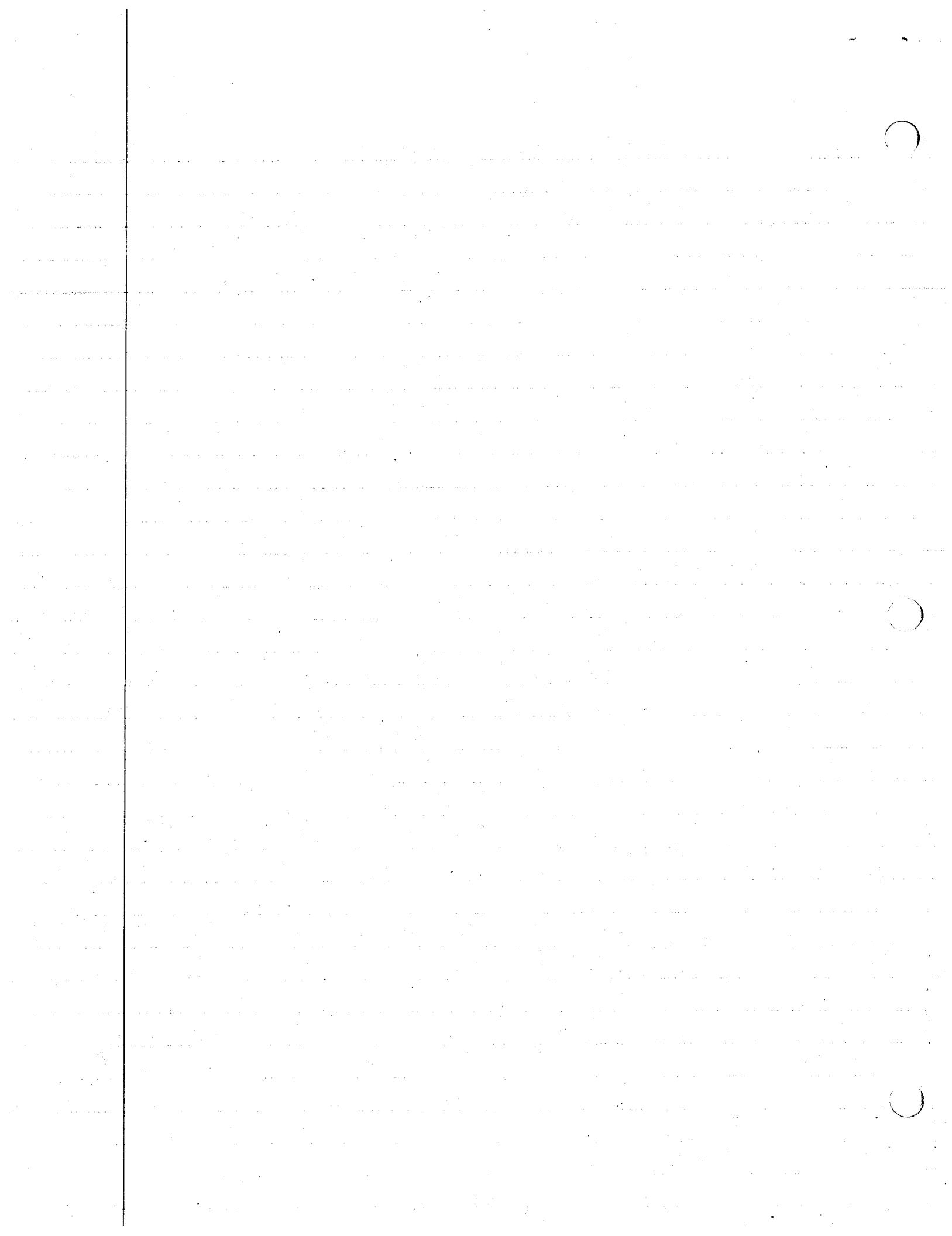
$$\omega = \dot{\theta} m_3 \quad \omega_{q_1} = -\frac{1}{2w} m_3 \quad \omega_{q_2} = \frac{1}{2w} m_3$$

$$N^{B*} = N^{A_1} + \omega \times \omega m_1 = \frac{q_2 - q_1}{2} m_2$$

$$N_{q_1}^{B*} = \frac{1}{2} m_2 \quad N_{q_2}^{B*} = \frac{1}{2} m_2$$

$$\alpha^{B*} = \frac{\dot{q}_2 + \dot{q}_1}{2} m_2$$

$$\therefore -m N_{q_1} \cdot \alpha^{B*} = -\frac{m}{4} (\ddot{q}_2 + \ddot{q}_1) \quad -m N_{q_2} \cdot \alpha^{B*} = -\frac{m}{4} (\ddot{q}_2 + \ddot{q}_1)$$



$$\text{II}^B = \frac{m l^2}{12} (m_1 m_2 + m_2 m_3) \quad m = \frac{W}{g} \quad l = 2w$$

$$= \frac{W w^2}{3g} (m_1 m_2 + m_2 m_3)$$

$$\text{II} \cdot \omega = \frac{W w^2}{3g} \left( \frac{\ddot{q}_2 - \ddot{q}_1}{2w} \right) m_3 \quad (\text{II} \cdot \omega) \times \omega = 0 \quad \alpha = \frac{\ddot{q}_2 - \ddot{q}_1}{2w} m_3$$

$$\text{II} \cdot \alpha = \frac{W w^2}{3g} \frac{\ddot{q}_2 - \ddot{q}_1}{2w} m_3$$

$$\text{II}^* = -\frac{W w}{6g} (\ddot{q}_2 - \ddot{q}_1) m_3$$

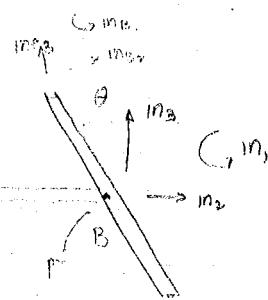
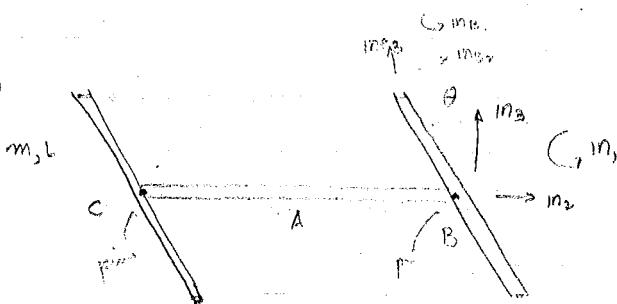
$$\omega_{q_1} \cdot \text{II}^* = \frac{W}{12g} (\ddot{q}_2 - \ddot{q}_1) \quad \omega_{q_2} \cdot \text{II}^* = -\frac{W}{12g} (\ddot{q}_2 - \ddot{q}_1)$$

$$F_1^* = -\frac{W}{4g} (\ddot{q}_2 + \ddot{q}_1) + \frac{W}{12g} (\ddot{q}_2 - \ddot{q}_1) = -\ddot{q}_1 \left( \frac{1}{3} \frac{W}{g} \right) - \ddot{q}_2 \left( \frac{1}{6} \frac{W}{g} \right)$$

$$= -\frac{W}{6g} [2\ddot{q}_1 + \ddot{q}_2]$$

$$F_2^* = -\frac{W}{4g} (\ddot{q}_2 + \ddot{q}_1) - \frac{W}{12g} (\ddot{q}_2 - \ddot{q}_1) = -\ddot{q}_1 \left( \frac{1}{6} \frac{W}{g} \right) - \ddot{q}_2 \left( \frac{1}{3} \frac{W}{g} \right) = \frac{W}{6g} [\ddot{q}_1 + 2\ddot{q}_2]$$

8g)



$$\text{let } u_i = \begin{cases} \omega \cdot m_i & i = 1, 2, 3 \\ 0 & i = 4 \\ W \cdot m_{i-4} & i = 5, 6, 7 \end{cases}$$

drop all 2nd & higher degrees of  $\theta$   
& 2nd & higher degrees of  $\dot{\theta}$ ,  $\ddot{\theta}$  etc  
find  $F_1^*, F_2^*, F_3^*$

$$\text{now } {}^R V^A = (W \cdot m_{i-4}) m_{i-4} = u_5 m_1 + u_6 m_2 + u_7 m_3 \quad (W_{u_1}^A = 0, W_{u_4}^A = 0, W_{u_7}^A = m_3)$$

$$\omega^A = u_1 m_1 + u_2 m_2 + u_3 m_3 ; \quad \begin{cases} \omega_{u_7}^A = 0 & \omega_{u_4}^A = m_1 \\ \omega_{u_4}^A = 0 & \omega_{u_1}^A = m_1 \end{cases} \quad \begin{matrix} \rightarrow \\ \rightarrow \\ \rightarrow \end{matrix} \quad \begin{matrix} \theta \\ \theta \\ \theta \end{matrix} \quad \begin{matrix} \ddot{\theta} \\ \ddot{\theta} \\ \ddot{\theta} \end{matrix} \quad \begin{matrix} \ddot{\theta} \\ \ddot{\theta} \\ \ddot{\theta} \end{matrix}$$

$$\text{II}^{A/A^*} = \frac{m L^2}{12} (m_3 m_1 + m_1 m_3)$$

$$\text{for } m_1 \text{ fixed in A only} \quad \omega^A = u_5 m_1 + u_6 m_2 + u_7 m_3 + (u_1 u_2 + u_4 u_1) m_3 + (u_5 u_3 - u_2 u_1) m_2 + (-u_6 u_3 + u_2 u_1) m_1$$

$$- m W_{u_4}^A \cdot \omega^A = - m m_3 \cdot \omega^A = - m \dot{u}_2 = m (u_3 u_1 - u_2 u_1)$$

$$\text{II}^A = (\text{II} \cdot \omega) \times \omega - \text{II} \cdot \alpha$$

$$\alpha^A = u_1 m_1 + u_2 m_2 + u_3 m_3 + u_1 (u_2 m_3 + u_3 m_2) + u_2 (u_1 m_3 - u_3 m_1) + u_3 (-u_1 m_2 + u_2 m_1)$$

$$(\text{II}^A \cdot \omega^A) = \frac{m L^2}{12} (u_3 m_3 + u_1 m_1) \quad (\text{II}^A \cdot \omega^A) \times \omega^A = \frac{m L^2}{12} (u_3 u_1 m_2 - u_3 u_2 m_1)$$

$$\text{II} \cdot \alpha^A = \frac{m L^2}{12} [(u_1 - u_2 u_3 + u_3 u_1) m_1 + (u_3 - u_1 u_2 + u_2 u_1) m_3] \quad + u_1 u_2 m_3 - u_1 u_3 m_2$$

$$\therefore \text{II}^A = \frac{m L^2}{12} [(u_3 u_2 - u_1) m_1 + (u_1 u_2 - u_3) m_3]$$

$$W_{u_7}^A \cdot \text{II}^A = 0$$

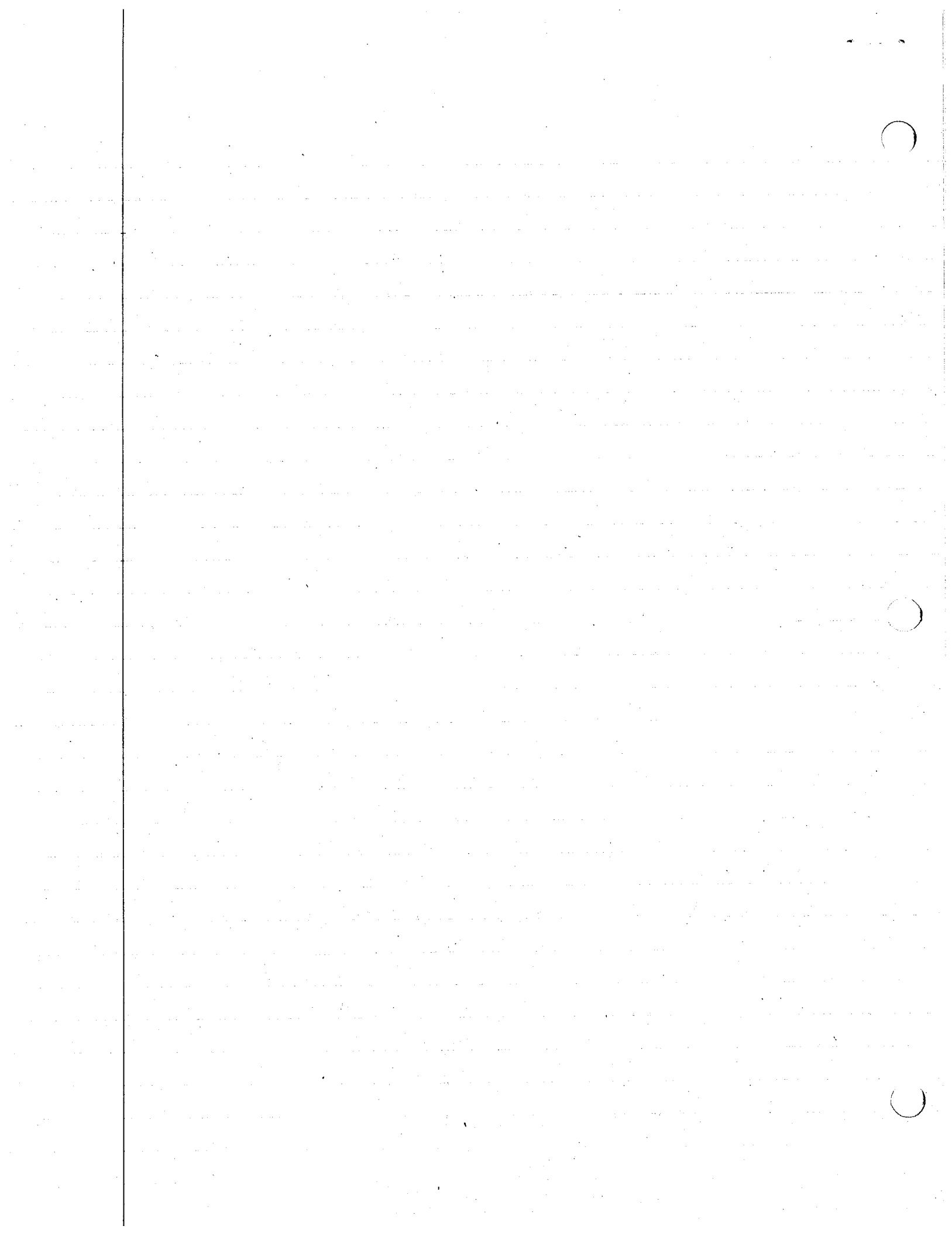
$$\omega^B = \theta m_1 = u_4 m_1, \quad \omega^B = u_4 m_1 + \omega^A; \quad \omega_{u_3}^B = 0 \quad \begin{cases} \theta_{u_2} = m_1 \\ \theta_{u_1} = m_1 \end{cases}$$

$$W^B = W^A + \omega^A \times \frac{L}{2} m_2 = W^A + [(u_4 + u_1) m_1 + u_2 m_2 + u_3 m_3] \times \frac{L}{2} m_2 = u_5 m_1 (u_6 - u_3 u_1) m_2 + (u_7 + u_2 u_1) m_3$$

$$\alpha^B = u_5 m_1 + (u_6 - u_3 u_1) m_2 + (u_7 + (u_4 + u_1)/2) m_3 + u_5 (u_6 u_2 - u_3 u_2) + (u_6 - u_3 u_2) (u_4 u_3 + u_3 u_1)$$

$$+ (u_7 + (u_4 + u_1) u_1) (-u_4 u_2 + u_2 u_1)$$

$$W_{u_2} = m_3; \quad W_{u_4} = u_2 m_3; \quad W_{u_1} = \frac{L}{2} m_2$$



$$\begin{aligned}
& \text{to } \omega^C = \dot{\theta} m_1 = u_4 m_1, \quad \omega^c = \omega^A + u_4 m_1 \\
& \omega^c = (u_4 + u_1) m_1 + u_2 m_2 + u_3 m_3 \\
& \omega^C = \omega^A + \omega^A \times -\frac{l}{2} m_2 = u_5 m_1 + (u_6 + u_3 l/2) m_2 + (u_7 + (u_1 + u_4) l/2) m_3 \\
& \left| \begin{array}{l} \omega_{u_7}^c = 0 \quad \omega_{u_4}^c = m_1 \quad \omega_{u_1}^c = m_1 \\ \omega_{u_2}^c = m_2 \quad u_{u_4}^c = \frac{l}{2} m_3 \quad \omega_{u_1}^c = -\frac{l}{2} m_3 \end{array} \right| \rightarrow u_5 (\omega^c \times m_1) + (u_6 + u_3 l/2) (\omega^c \times m_2) + \\
& \alpha^C = u_5 m_1 + (u_6 + u_3 l/2) m_2 + [u_7 - (u_1 + u_4) l/2] m_3 \quad \left| \begin{array}{ccc} m_1 & m_2 & m_3 \\ m_3 & 1 & 0 & 0 \\ m_2 & 0 & -s_0 & +s_0 \\ m_1 & 0 & s_0 & -s_0 \end{array} \right. \\
& + [u_7 - (u_1 + u_4) l/2] (m_3), \quad \ddot{\Pi}^{B/A} = \frac{m l^2}{12} (m_{u_2} m_{u_2} + m_{u_3} m_{u_1}) \\
& = \frac{m l^2}{12} [(c_0 m_2 + s_0 m_3)(c_0 m_2 + s_0 m_3) + m_1 m_1] \\
& = \frac{m l^2}{12} [c_0^2 m_2 m_2 + c_0 s_0 (m_2 m_3 + m_3 m_2) + s_0^2 m_3 m_3 + m_1 m_1] \\
& \omega^B = (u_4 u_1) m_1 + u_2 m_2 + u_3 m_3 \\
& (\ddot{\Pi}^{B/A}, \omega^B) = [(u_4 + u_1) m_1 + u_2 (c_0^2 m_2 + c_0 s_0 m_3) + u_3 (s_0^2 m_3 + c_0 s_0 m_2)] \frac{m l^2}{12}, \\
& (\ddot{\Pi} \cdot \omega^B) \times \omega^B = [(u_4 + u_1) u_2 m_3 - (u_4 + u_1) u_3 m_2 + u_2 c_0^2 (u_4 + u_1) m_3 + u_2 u_3 c_0^2 m_1 + u_2 (u_4 + u_1) c_0 s_0 m_2 \\
& \rightarrow u_2 c_0^2 s_0 u_2 m_1 + u_3 s_0^2 (u_4 + u_1) m_2 - u_2 u_3 s_0^2 m_1 - u_3 c_0 s_0 (u_4 + u_1) m_3 + u_3 c_0 s_0 m_1] \\
& = [(u_4 + u_1) u_2 s_0^2 m_3 - u_3 (u_4 + u_1) c_0^2 m_2 + u_2 u_3 c_{20} m_1 + u_2 (u_4 + u_1) c_0 s_0 m_2 \\
& - u_2^2 c_0 s_0 m_1 - u_3 (u_4 + u_1) c_0 s_0 m_3 + u_3^2 c_0 s_0 m_1] \frac{m l^2}{12} \\
& \omega^B = (u_4 + u_1) m_1 + u_2 m_2 + u_3 m_3 + (u_4 + u_1) (-u_2 m_3 + u_3 m_2) + u_2 (u_4 m_3 + u_3 m_2) + u_3 (u_1 m_2 + u_2 m_1) \\
& \ddot{\Pi} \cdot \omega^B = \frac{m l^2}{12} [(u_4 + u_1) u_2 m_3 + u_2 (u_4 + u_1) u_3 m_2 + (u_4 + (u_4 + u_1) u_2 - u_3 u_1) m_3 + (u_3 - u_2 (u_4 + u_1) + u_2 u_1) m_1] \\
& \omega_{u_7}^B = 0 \quad \omega_{u_7}^B \cdot \ddot{\Pi}^A = \omega_{u_7}^B \left[ (\ddot{\Pi} \cdot \omega^B) \times \omega^B \right] = 0
\end{aligned}$$

Since  $\ddot{\Pi}^A$  has same  $\ddot{\Pi}^{C/A}$  as  $\ddot{\Pi}^{B/A}$ ,  $\omega_{u_7}^c \times \omega_{u_7}^B = 0$

then  $(\ddot{\Pi} \cdot \omega^c) \times \omega^c = \ddot{\Pi} \cdot \omega^c \times \ddot{\Pi}^C = \ddot{\Pi}^B$  hence  $\omega_{u_7}^c \cdot \ddot{\Pi}^B = 0$  also

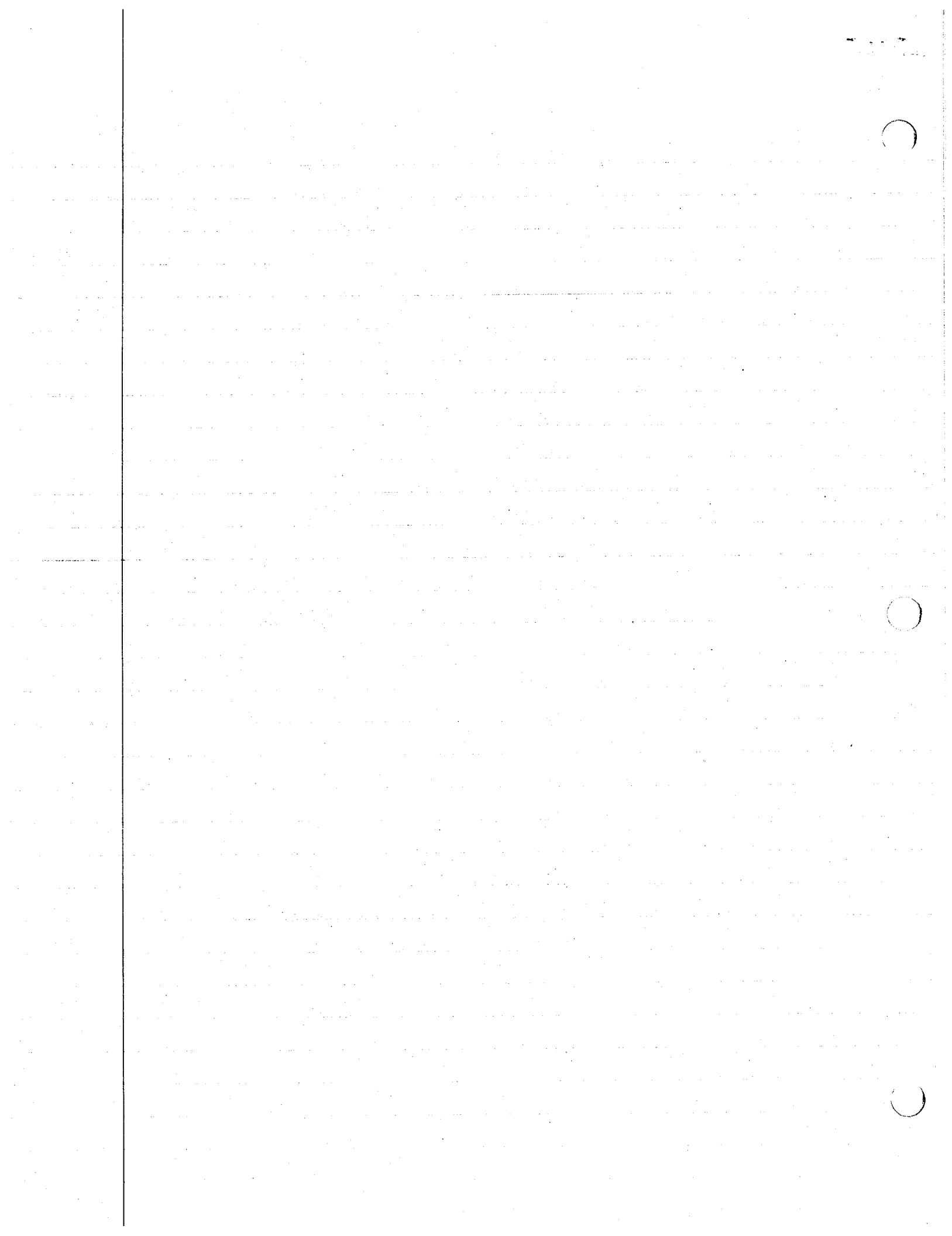
thus along compatibility,

$$\begin{aligned}
\text{to: } F_7^A &= m \omega_{u_7}^A \cdot \alpha^A + m \omega_{u_7}^B \cdot \alpha^B + m \omega_{u_7}^C \cdot \alpha^C \\
&= m(u_1 u_2 - u_3 u_2) - m(u_7 + u_4 + u_1 l/2) - m(u_1 u_6 - u_2 u_5) \\
&= m(u_7 - (u_1 + u_4) l/2) - m(u_1 u_6 - u_2 u_5) + 3m[u_7 + (u_6 u_1 - u_5 u_2)]
\end{aligned}$$

$$\text{to } F_4^A = +\omega_{u_4}^c \cdot \ddot{\Pi}^C + \omega_{u_4}^c \cdot \ddot{\Pi}^B + \omega_{u_4}^B \cdot \ddot{\Pi}^B + \omega_{u_4}^B \cdot \ddot{\Pi}^{A/B} \text{ since } \omega_{u_4}^A = \omega_{u_4}^c = 0$$

$$\text{to } F_1^A = (\omega_{u_1}^c \cdot \ddot{\Pi}^{A/B} + \omega_{u_1}^B \cdot \ddot{\Pi}^B) + (\omega_{u_1}^c \cdot \ddot{\Pi}^B + \omega_{u_1}^B \cdot \ddot{\Pi}^B) + \omega_{u_1}^A \cdot \ddot{\Pi}^{A/B}$$

In obtaining  $\ddot{\Pi}^A$  &  $\ddot{\Pi}^B$  we must use  $m_1 = \omega \times m_1$ ,  $m_2 = \omega \times m_1$ ,  $m_3 = \omega \times m_1$



Set #9

9a.



$$F_1 = \frac{Gm_1 m_2}{r^2}, \quad F_2 = \frac{Gm_1 m_2}{r^2}$$

$$W_{u_i}^{P_1} = u_{i1} m_i$$

$$F_1 = \frac{Gm_1 m_2}{r^2}, \quad F_2 = \frac{Gm_1 m_2}{r^2}$$

$$W_{u_i}^{P_2} = u_{i2} m_i$$

$$\begin{aligned} W_{u_i}^{P_2} &= W_{u_i}^{P_1} + w(t), \\ W_{u_i}^{P_2} &= W_{u_i}^{P_1} + \dot{w} \end{aligned}$$

$$\therefore W_{u_i}^{P_1} = m_i$$

$W_{u_i}^{P_2} = m_i$  since it is not a fn of  $u_i$

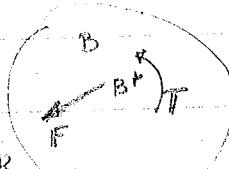
$$\therefore F'_i = W_{u_i}^{P_1} \cdot F_1 + W_{u_i}^{P_2} \cdot F_2 = 0 \quad i=1,2,3; \quad F'_q = \frac{W_{u_i}^{P_2}}{r} \cdot F_2 = -\frac{Gm_1 m_2}{r^2} m_i \cdot m_i = -\frac{Gm_1 m_2}{r^2}$$

$$\text{but } F'_q = -\frac{Gm_1 m_2}{r^2} = -\frac{\partial P}{\partial r} \quad \therefore \frac{\partial Gm_1 m_2}{\partial r} = \frac{\partial P}{\partial r} \Rightarrow P = -\frac{Gm_1 m_2}{r} + f(u_1, u_2, u_3, t)$$

$$\text{but } \frac{\partial P}{\partial u_1} = 0, \quad \frac{\partial P}{\partial u_2} = 0, \quad \frac{\partial P}{\partial u_3} = 0 \Rightarrow f(u_1, u_2, u_3, t) = w(t) \quad \therefore P = -\frac{Gm_1 m_2}{r} + w(t)$$

$$R_B^B = \omega^B \times I^B \quad \omega^B = \Omega m_a + \dot{\theta}^B \quad \text{let } \dot{\theta}^B = \omega_{q_r}^B \text{ where}$$

$m_i$ 's are  $\parallel$  to principal axes of  $B$



$$F_r = -F_{na} \quad \text{and} \quad \tau = \frac{3GM}{R^2} m_a \times I_a$$

$$W_{q_r}^{B^*} = \omega^B \times R_{na} = R(\omega^B \times m_a) \quad W_{q_r}^{B^*} = \omega_{q_r}^B \times m_a$$

$$\text{now } F_r = W_{q_r}^{B^*} \cdot F + \omega_{q_r}^B \cdot \tau = -F_{na} \cdot (m_a \times I_a) + \omega_{q_r}^B \cdot \frac{3GM}{R^2} m_a \times I_a$$

$$= \frac{3GM}{R^2} \omega_{q_r}^B \cdot (I_a \times I_a) = \frac{3GM}{R^2} I_a \cdot (\omega_{q_r}^B \times m_a)$$

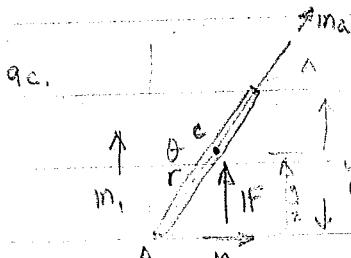
$$\text{but } \frac{\partial m_a}{\partial q_r} = \frac{\partial m_a}{\partial q_r} + \omega_{q_r}^B \times m_a \quad \text{and} \quad \frac{\partial m_a}{\partial q_r} = 0 \quad \text{since } m_a \text{ is not a fn of } q_r \text{ in } F_r$$

$$F_r = \frac{3GM}{R^2} I_a \cdot \left( -\frac{\partial m_a}{\partial q_r} \right) = -\frac{3GM}{R^2} I_a \cdot \frac{\partial m_a}{\partial q_r}$$

$$\text{if } m_a = a_i m_i \text{ then } I_a = I_i a_i m_i \quad \text{then } I_a \cdot \frac{\partial m_a}{\partial q_r} = I_i (a_i \frac{\partial a_i}{\partial q_r}) = I_i + \frac{1}{2} \frac{\partial a_i^2}{\partial q_r}$$

$$\therefore F_r = -\frac{3GM}{2R^2} \frac{\partial (a_i^2 I_i)}{\partial q_r} \quad \text{but } a_i^2 I_i = I = m_a^2 I \cdot m_a$$

$$F_r = -\frac{\partial}{\partial q_r} \left( \frac{3GM}{2R^2} I \right) \quad \therefore P = \frac{3GM}{2R^2} I + w(t)$$



$$w = \rho g \quad \text{displaced volume is } \Delta y \quad \therefore \text{by Archimedes principle}$$

The buoyant force = displaced volume  $\times$  weight density

$$F = \Delta y \cdot \rho g m_A \text{ acting upward at mass center}$$

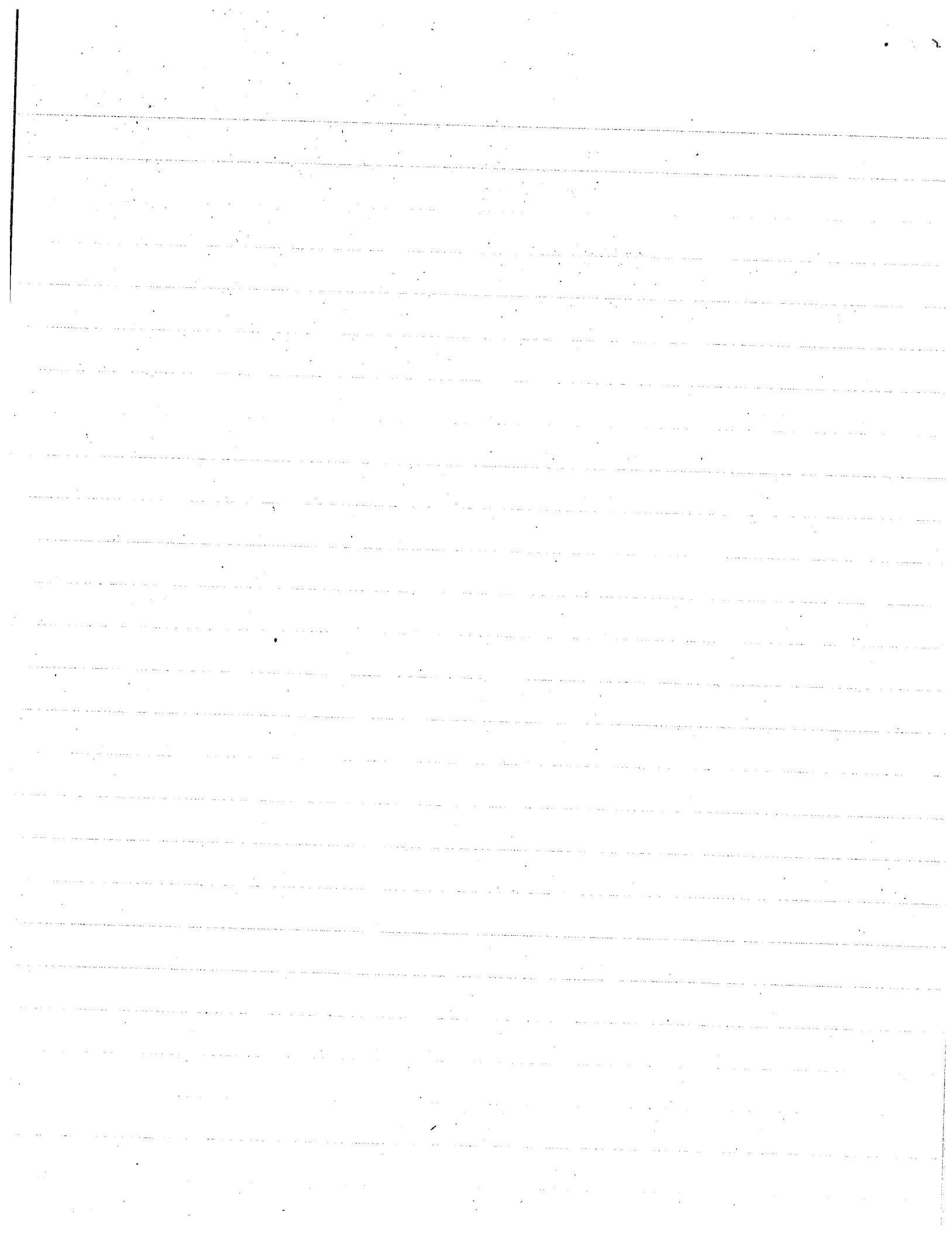
$$\text{now } W_e = W^A + \rho V_A m_A$$

of submerged portion

$$\text{now } W_e = \rho V_A m_A + W_A^A$$

$$W_A^A = 0$$

$$\therefore (F_B)_{\text{buoyant force}} = \frac{\Delta y \rho V_A m_A}{m_A \Delta y} = \frac{\Delta y \rho V_A m_A}{m_A}$$



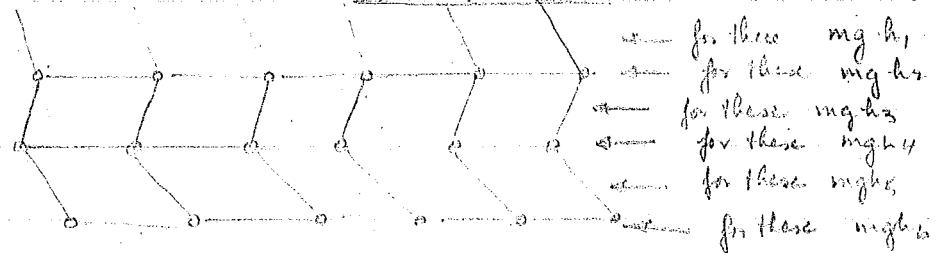
but  $m_a = m_1 \cos\theta + m_2 \sin\theta$ ,  $m_3 \times m_a = \cos\theta m_2 - m_1 \sin\theta$  and  $m_3 \times m_2 \cdot m_1 = \sin^2\theta$

$$\therefore F_\theta = -rAy w \tan\theta \quad \text{but } r = \frac{y}{2 \cos\theta} \quad \therefore F_\theta = -\frac{Ay^2 w \sin\theta}{2 \cos^2\theta}$$

$$\text{but } F_\theta = -\frac{\partial P}{\partial \theta} = -\frac{Ay^2 w \sin\theta}{2 \cos^2\theta} \quad P = \frac{Ay^2 w}{2 \cos\theta} + f(t)$$

$$\text{or } P = \frac{Ay^2 w}{2} \sec\theta + f(t)$$

9d Now we have



deflection line

$$P = \sum mg_i h_i$$

$$\text{for these } mg_i h_i$$

$$h_1 = \frac{l}{2} \cos q_1, \quad h_2 = -l \sin q_1, \quad h_3 = -l(\cos q_1 + \frac{1}{2} \sin q_2), \quad h_4 = -l(\cos q_1 + \cos q_2)$$

$$h_5 = -l(\cos q_1 + \cos q_2 + \frac{1}{2} \cos q_3), \quad h_6 = -l(\cos q_1 + \cos q_2 + \cos q_3)$$

$$\therefore 6mgh_1 + 5mgh_2 + 6mgh_3 + 5mgh_4 + 6mgh_5 + 5mgh_6$$

$$= 6mg \left[ + 3\cos q_1 + 5\cos q_1 + 6\cos q_1 + 3\cos q_2 + 5\cos q_1 + 5\cos q_2 + 6\cos q_1 + 6\cos q_2 + 3\cos q_3 + 5\cos q_3, 15\cos q_1 + 5\cos q_2 \right]$$

$$P = -Wl [30\cos q_1 + 19\cos q_2 + 8\cos q_3]$$

9e Since we found in 5(a)  $F_\theta = -W(R^2 - L^2)^{1/2} \sin\theta = -\frac{\partial P}{\partial \theta}$

$$\therefore \frac{\partial P}{\partial \theta} = W(R^2 - L^2)^{1/2} \sin\theta \quad \therefore P = -W(R^2 - L^2)^{1/2} \cos\theta + f(t)$$

9f From 5(d) we found

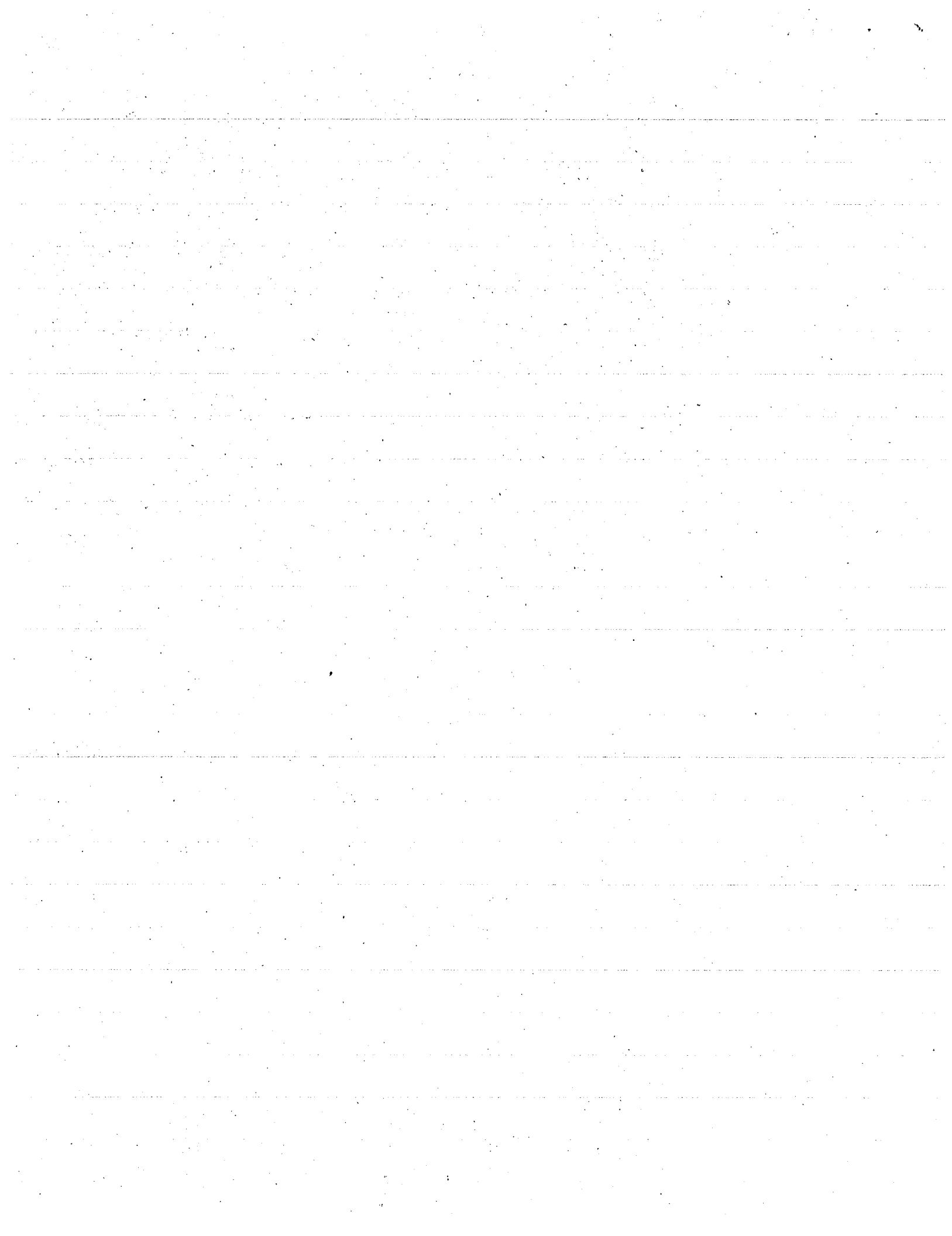
$$F_1 = a_1 q_1 + b_1 q_2 + \frac{w}{2} = -\frac{\partial P}{\partial q_1} \quad \therefore P = -a_1 q_1^2/2 + b_1 q_2 q_1 + \frac{w}{2} q_1 + f(q_2, t)$$

$$F_2 = b_1 q_1 + a_1 q_2 + \frac{w}{2} = -\frac{\partial P}{\partial q_2} \quad \frac{\partial P}{\partial q_2} = b_1 q_1 + \frac{\partial f}{\partial q_2} = b_1 q_1 + q_1 q_2 + \frac{w}{2}$$

$$\frac{\partial f}{\partial q_2} = q_1 q_2 + \frac{w}{2} \quad \therefore f = a_1 q_1^2/2 + \frac{w}{2} q_2 + g(t)$$

$$\therefore P = -a_1 q_1^2/2 + b_1 q_2 q_1 + \frac{w}{2} (q_1 + q_2) + q_1 q_2^2/2 + g(t)$$

$$= \left( \frac{q_1^2 - q_2^2}{2} \right) \left[ \frac{12EI}{L^3} \left( 1 + \frac{L}{2w} + \frac{L^2}{w^2} \right) \right] + \frac{6EI}{L^2 w} \left( 1 + \frac{L}{3w} \right) q_2 q_1 + \frac{w}{2} (q_1 + q_2) + g(t)$$



9(g) look at one corner



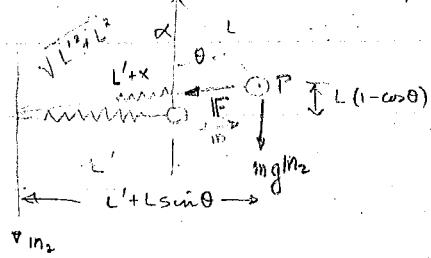
Movement of the corner of the cube : By looking the movement of the cube corner center as the cube undergoes displacement to a new point; the cube corner undergoes an extension in the  $-x_3$  direction. Now if the cube is rotated about the  $x_1$  axis will give rise to a displacement  $a\theta_1$ . Now if the cube is rotated about the  $x_2$  direction gives rise to an  $a\theta_2$  displacement. Rotation about the  $x_3$  axis gives rise to displacements in the  $x_1$  &  $x_2$  direction thus the overall displacement of the spring is  $-x_3 + a\theta_1 + a\theta_2 = x$

Now  $P$  for a spring is  $\frac{1}{2}kx^2 \therefore P_i = \frac{1}{2}k\alpha^2(\theta_1 + \theta_2 - x_3/a)^2$ . Similarly we can do the

same for the other springs to get  $P_2$  &  $P_3$ . Thus  $P = 2P_i$ . What we've done here is good for small displacements ie displacements for terms of 0 or 1st degree in  $\theta_i$  and  $x_i$ .

DATUM LINE

9(h)



$$r^P = L\cos\theta \mathbf{i}_2 + L\sin\theta \mathbf{i}_1$$

$$\mathbf{w}^P = (-L\sin\theta \mathbf{i}_2 + L\cos\theta \mathbf{i}_1)\dot{\theta}$$

$$\text{now } (L'+x)^2 = L^2 + (L'^2 + L^2) - 2L\sqrt{L^2 + L'^2} \cos(\theta + \alpha)$$

$$L'^2 + 2L'x + x^2 - 2L^2 + L'^2 = 2L\sqrt{L^2 + L'^2} \cos(\theta + \alpha)$$

$$\therefore 2L^2 + 2L\sqrt{L^2 + L'^2} \left[ \cos\theta \frac{1}{\sqrt{L^2 + L'^2}} - \sin\theta \frac{L'}{\sqrt{L^2 + L'^2}} \right]$$

$$\mathbf{F} = -k\mathbf{x} \therefore \mathbf{i}_1 = (L' + L\sin\theta)\mathbf{i}_1, -L(1-\cos\theta)\mathbf{i}_2$$

$$2L'x + x^2 + L'^2 = 2L^2 + 2L^2\cos\theta + 2LL'\sin\theta + L'^2$$

$$\mathbf{F}_\theta = -kx \{ L\cos\theta(L' + L\sin\theta) + L^2\sin\theta(1-\cos\theta) \} = mgL\sin\theta$$

$$x^2 + 2L'x + 2L^2(1-\cos\theta) + 2LL'\sin\theta + L'^2 = 0$$

the pot. fn due to gravity is  $P = mg h \quad h = -L\cos\theta \quad \therefore P = -mgL\cos\theta$  from the datum line

the potential fn due to the spring  $= \frac{1}{2}kx^2$

$$\text{Now } (x+L')^2 = 2L^2(1-\cos\theta) + L'(L' + 2L\sin\theta)$$

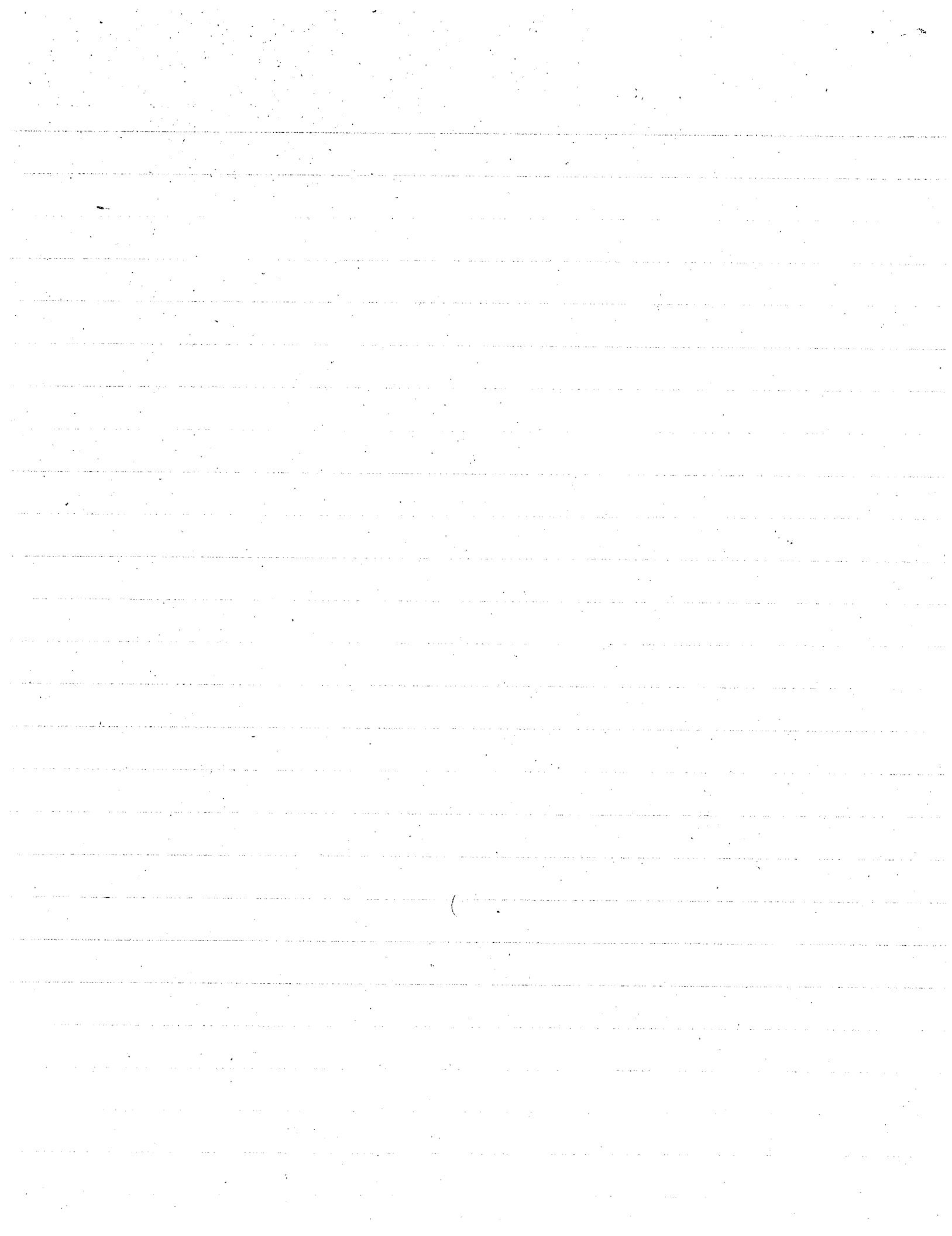
$$(x+L') = L' \sqrt{2L^2(1-\cos\theta)} + \left( 1 + \frac{L'}{L} \sin\theta \right)$$

$$\text{use expansion } (1+p)^{1/2} = 1 + \frac{1}{2}p - \frac{1}{8}p^2 + \frac{1}{16}p^3 \quad \text{where } p = \frac{2L\sin\theta}{L} + \frac{2L^2(1-\cos\theta)}{L^2}$$

$$\therefore p \approx \frac{2L}{L} (\theta - \frac{\theta^3}{3!} + \dots) + \frac{2L^2}{L^2} (\frac{\theta^2}{2!} - \frac{\theta^4}{4!}) = \frac{2L}{L} (\theta + \frac{L}{2L}\theta^2 - \frac{\theta^3}{3!} - \frac{L}{4L}\theta^4)$$

$$= 1 + \frac{L}{L} (\theta - \frac{\theta^3}{3!}) + \frac{L^2}{L^2} (\frac{\theta^2}{2!} - \frac{\theta^4}{4!}) - \frac{1}{8} \cdot \frac{4L^2}{L^2} \left[ \theta^2 + \frac{L^2}{4L^2}\theta^4 + \frac{2L}{3L}\theta^3 - \frac{2\theta^4}{3!} \right] + \dots$$

$$\therefore x = L(\theta - \frac{\theta^3}{3!}) + O(\theta^4) + \frac{1}{16} \left[ \frac{8L^3}{L^3} (\theta^2 + \frac{3L}{2L}\theta^4) + \dots \right] \Rightarrow x^2 = L^2 (\theta^2 - \frac{20\theta^4}{6} + \text{h.o.t.})$$



$$P_{\text{TOT}} = -mgL \left( 1 - \frac{\theta^2}{2!} + \frac{\theta^4}{4!} + \dots \right) + \frac{1}{2} \cdot \frac{5mg}{L} \cdot L^2 \left[ \theta^2 - \frac{\theta^4}{3!} + \dots \right]$$

$$= -mgL + mgL \frac{\theta^2}{2!} - mgL \frac{\theta^4}{4!} + \frac{5}{2} mgL \theta^2 - \frac{5}{6} mgL \theta^4 + \dots$$

$$P_{\text{TOT}} - mgL = mgL \left[ \theta^2 \cdot 3 \right] - mgL \theta^4 \left[ \frac{1}{24} + \frac{5}{6} \right] = mgL \left[ 3\theta^2 - \frac{7}{8}\theta^4 \right]$$

$$a_1 = 0, a_2 = 3, a_3 = 0, a_4 = -\frac{7}{8}$$

$$\text{Now } F_\theta = -\frac{kx}{x+L} \{ Ll' \cos\theta + l^2 \sin\theta \} - mgL \sin\theta = -\frac{kl(\theta - \theta^3/3!)}{x+L} \left\{ (1 - \frac{\theta^2}{2!} + \frac{\theta^4}{4!} + \dots) + \frac{l}{L} (\theta - \theta^3/3!) \right\} \left\{ 1 - \frac{l}{L} (\theta - \theta^3/3!) + \frac{l^2}{L^2} (\theta - \theta^3/3!)^2 + \dots \right\} \\ - mgL \left\{ \theta - \theta^3/3! \right\} = -kl^2 (\theta - \theta^3/3!) \left\{ 1 + \frac{l}{L} \theta - \frac{\theta^2}{2!} - \frac{l}{L} \theta^3 \right\} \left\{ 1 - \frac{l}{L} \theta + \frac{l^2}{L^2} \theta^2 + \frac{l^3}{L^3} \theta^3 + \dots \right\} = -kl^2 (\theta - \theta^3/3!) \left\{ 1 + \theta^2 \frac{l^2}{2!} - \frac{\theta^2}{2!} \right\} - mgL \left\{ \theta - \theta^3/3! \right\} \\ = -6mgL \theta + \frac{5}{6} mgL \theta^3 + \frac{5mg}{6} \theta^5 + mgL \frac{\theta^3}{6}$$

q: since  $P = mgL \sum a_n \theta^n$  then  $\frac{\partial P}{\partial \theta} = mgL \sum a_n n \theta^{n-1} = mgL [6\theta + 3.5\theta^3 \frac{5}{2}]$

$$F_\theta = -\frac{\partial P}{\partial \theta} = -mgL [a_1 + a_2 \cdot 2\theta + a_3 \cdot 3\theta^2 + \dots]$$

$$mgL [b_1 \theta + b_2 \theta^2 + \dots] = mgL [-a_1 - a_2 \cdot 2\theta - a_3 \cdot 3\theta^2 + \dots]$$

$$a_1 = 0, b_1 = -2a_2 = -6, b_2 = -3a_3 = 0, b_3 = -4a_4 = +4 \cdot \frac{7}{8} = +3.5$$

easier to find the potential energy then differentiate. Both methods are cumbersome.

$$F_\theta = \omega N^B$$

$$N^B = (l_1 q_1 + l_2 q_2) m$$

$$N^A = (l_1 + q_1) m$$

$$W^B = (q_1 + q_2) m$$

$$W^B - W^A = q_2 m$$

$$W^A = q_1 m$$

$$F_\theta = \beta (q_1 + q_2) m$$

$$F_\theta = -\alpha (q_1 + q_2) m$$

$$F_\theta = -\beta (W^B, W^A)$$

$$N = 0$$

$$-\beta F_\theta$$

$$F'_1 = F_\alpha \cdot N^B q_1 + F_\beta \cdot N^A q_2 + F_\alpha \cdot 0 + F_\beta \cdot N^A$$

$$= -\alpha [q_1 + q_2] - \beta q_2 + \beta q_2 = -\alpha [q_1 + q_2]$$

$$F'_2 = F_\alpha \cdot N^B q_2 + F_\beta \cdot N^A q_1 + F_\alpha \cdot 0 + F_\beta \cdot N^A$$

$$= -\alpha [q_1 + q_2] + \beta q_1 = -(\alpha + \beta) q_1 - \alpha q_1$$

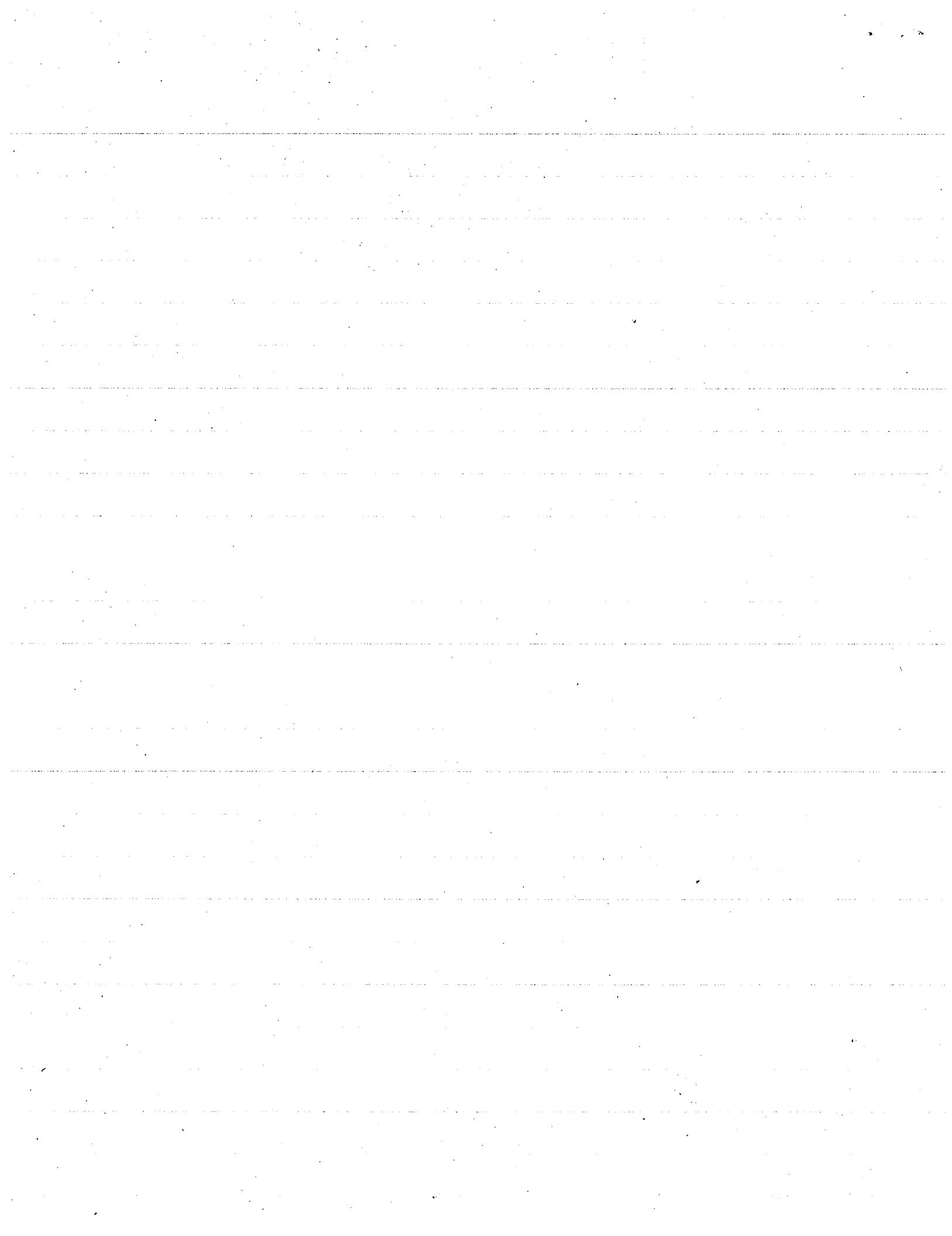
$$+\frac{\partial f}{\partial q_1} = F'_1 + \alpha [q_1 + q_2] \quad f = \alpha [q_1^2 - \frac{1}{2} q_1 q_2] + f(l_1, t)$$

$$\frac{\partial f}{\partial q_2} = \alpha q_1 + f' = -F'_2 = (\alpha + \beta) q_2 + \alpha q_1$$

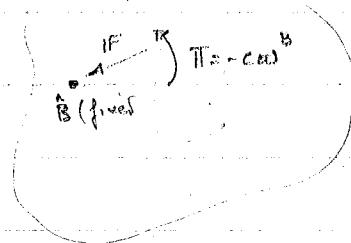
$$f' = (\alpha + \beta) q_2$$

$$f = (\alpha + \beta) \frac{q_1^2}{2} + g(t)$$

$$\therefore f = \alpha q_1^2 + \alpha q_1 q_2 + (\alpha + \beta) \frac{q_2^2}{2}$$



9k



$$F_r = \omega_{q_r} \cdot \vec{r} + \omega_{q_r} \cdot \vec{\pi} \quad \vec{\pi} = -c\omega \hat{z}$$

$$\text{since } \vec{v} = 0 \quad (\vec{B} \text{ is fixed})$$

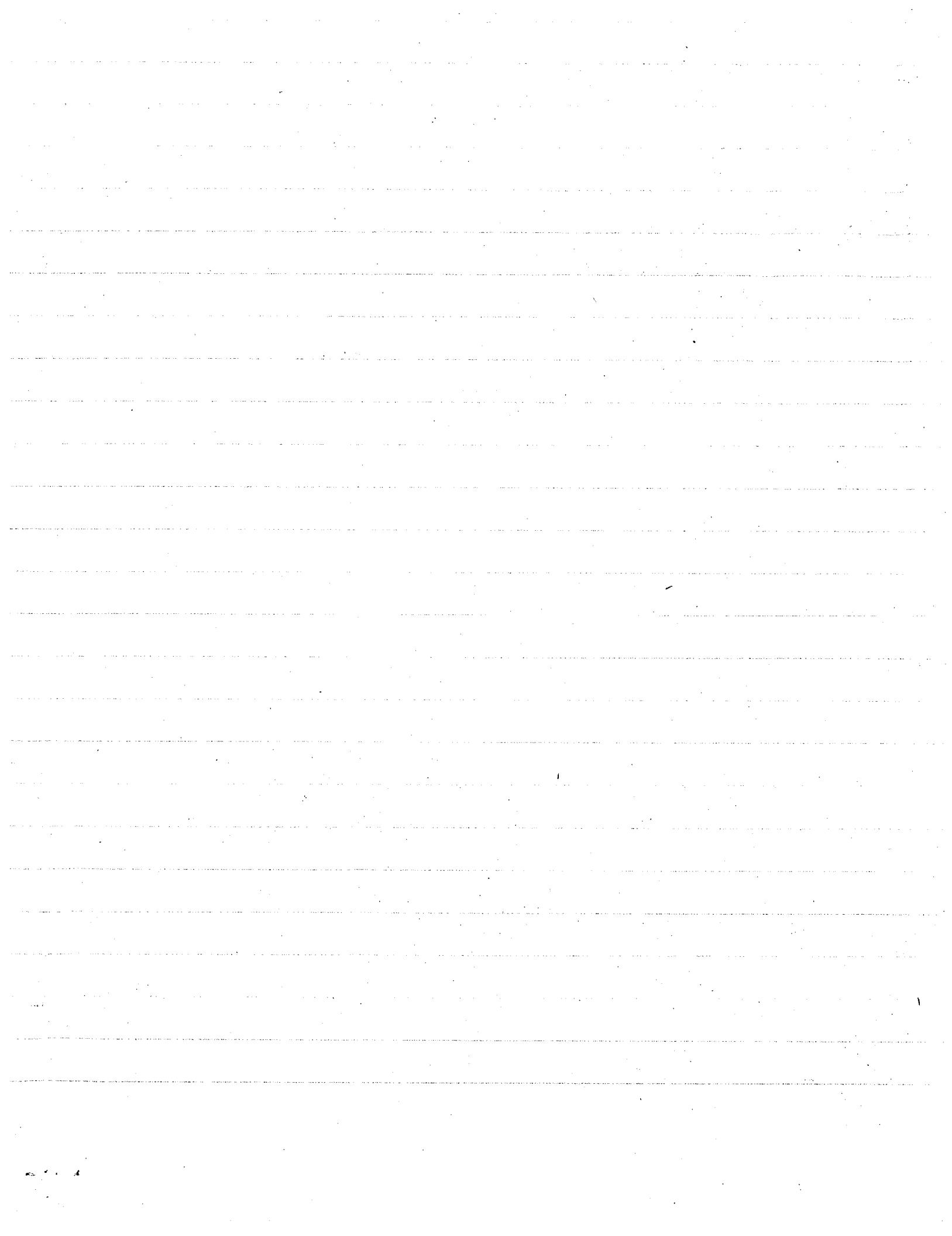
$$F_r = \omega_{q_r} \cdot \vec{\pi} = \omega_{q_r} \cdot (-c\omega)$$

$$\text{now if } \omega_{q_r} = \frac{\partial \omega}{\partial q_r} \text{ then } \omega = \omega_{q_r} q_r + \omega_t$$

$$\text{hence } F_r = -c \cdot (\omega \cdot \omega'_{q_r}) = c (\omega_{q_r} \cdot \omega_t)$$

$$F_r = -c \left( \frac{\partial \omega^2}{\partial q_r} \right) = c \frac{\partial}{\partial q_r} \left( \frac{d \omega^2}{dt} \right)$$

$$\text{Now } F_r = -\frac{dJ}{dq_r} = -c \frac{\partial}{\partial q_r} \frac{\omega^2}{2} \Rightarrow J = \frac{c\omega^2}{2} + \text{fm of time.}$$

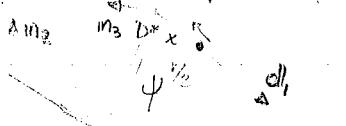


10a

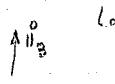
$$\mathbf{v}^P = (\dot{\phi} m_2 - \dot{\theta} m_1 + \dot{\psi} m_3) \times r \mathbf{m}_2 = r \dot{\phi} \cdot 0 - r \dot{\theta} m_3 - r \dot{\psi} m_1$$

$$\mathbf{r}^P = \mathbf{r}^{D*} + \mathbf{x} \times \mathbf{d}_1, \quad \mathbf{w}^P = \mathbf{w}^{D*} + \mathbf{v} \times \mathbf{d}_1 + \mathbf{x} \times \omega^D \times \mathbf{d}_1$$

$$\mathbf{R}_{\text{rod}}^D = \dot{\phi} \mathbf{i}_3^0 - \dot{\theta} \mathbf{i}_1 + \dot{\psi} \mathbf{i}_3; \quad \mathbf{R}_{\text{rod}}^D \times \mathbf{d}_1 = -\dot{\phi} m_3 + \dot{\psi} m_2$$



$$\text{when } \theta=0, \psi=0: \quad \mathbf{i}_3 \times \mathbf{d}_1 = \mathbf{m}_2; \quad \mathbf{m}_1 \times \mathbf{d}_1 = 0; \quad \mathbf{i}_3 \times \mathbf{d}_1 = -\mathbf{m}_3 \quad \mathbf{d}_1 = \mathbf{m}_1$$



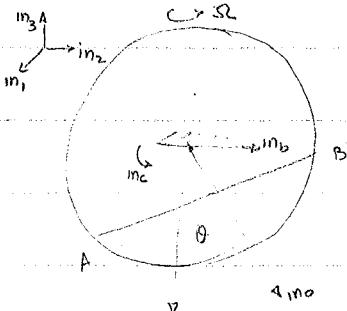
$$\begin{aligned} \mathbf{w}^P &= \mathbf{w}^{D*} + \mathbf{v} \times \mathbf{d}_1 + r \mathbf{w}^{D*} (\omega^D \times \mathbf{d}_1) + v^2 + 2\omega r \frac{d_1 \cdot (\omega^D \times \mathbf{d}_1)}{2} + \frac{r^2}{4} (\omega \times \mathbf{d}_1)^2 \\ &= r^2 \dot{\theta}^2 + r^2 \dot{\psi}^2 - 2\omega (-r \dot{\psi}) + r [\dot{r} \dot{\theta} \dot{\phi}] + v^2 + \frac{r^2}{4} [\dot{\phi}^2 + \dot{\psi}^2] \\ &= r^2 \left[ \dot{\theta}^2 + \frac{5}{4} \dot{\psi}^2 + \frac{\dot{\phi}^2}{4} + 2\omega \frac{\dot{\psi}}{r} + \dot{\theta} \dot{\phi} + \frac{v^2}{r^2} \right] \\ \mathbf{w}^P &= \left[ r^2 (\dot{\theta}^2 + \frac{5}{4} \dot{\psi}^2 + \frac{\dot{\phi}^2}{4} + \dot{\theta} \dot{\phi}) + \omega (v + 2r \dot{\psi}) \right] \end{aligned}$$

$$K = \frac{1}{2} m \mathbf{w}^P = \frac{1}{2} m \left[ r^2 (\dot{\theta}^2 + \frac{5}{4} \dot{\psi}^2 + \frac{\dot{\phi}^2}{4} + \dot{\theta} \dot{\phi}) + \omega (v + 2r \dot{\psi}) \right]$$

$$10b. \quad K = K_v + K_w$$

$$K_v = \frac{1}{2} m v^2 \quad K_w = \frac{1}{2} \omega \cdot \mathbf{I} \cdot \omega$$

$$\omega_{\text{rod}} = \dot{\theta} m_c - \omega I_K$$



$$v^2 = (R^2 - L^2) [\dot{\theta}^2 + \Omega^2 \sin^2 \theta]$$

$$\mathbf{I} = \frac{m L^2}{3} m_0 m_0 + \frac{m L^2}{3} m_c m_c \quad I_K = c I_K + S m_b$$

$$m_0 m_0 = c^2 I_K + \frac{S^2}{3} m_b$$

$$\mathbf{I} \cdot \omega = \frac{m L^2}{3} \{-S c^2 I_K - S c \sin \theta + \theta \sin c\}$$

$$\frac{1}{2} \omega \cdot \mathbf{I} \cdot \omega = S^2 c^2 m_L^2 + \frac{m L^2 \theta^2}{3}$$

$$K = K_v + K_w = \frac{1}{2} m \left[ (R^2 - L^2) \{ \dot{\theta}^2 + \Omega^2 \sin^2 \theta \} + S^2 c^2 \theta \frac{L^2}{3} + \frac{L^2 \theta^2}{3} \right]$$

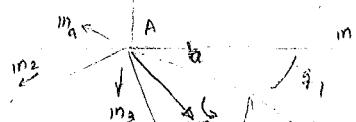
$$F_{\theta}^* = \frac{\partial K}{\partial \theta} = \frac{d}{dt} \frac{\partial K}{\partial \dot{\theta}} = \frac{1}{2} m [(R^2 - L^2) \Omega^2 \cdot 2 \sin \theta \cos \theta + 2 S^2 c^2 \cos \theta \sin \theta \frac{L^2}{3}] - \frac{1}{2} m [(R^2 - L^2) \cdot 2\ddot{\theta} + \frac{2L^2 \ddot{\theta}}{3}]$$

$$= -\frac{m}{3} [ \Omega^2 \sin \theta \cos \theta (4L^2 - 3R^2) + (3R^2 - 2L^2) \ddot{\theta} ]$$

Same result. Much easier tho.

$$10c. \quad \mathbf{v}^T = \dot{q}_1 s_2 \frac{a}{3} \mathbf{m}_a + \dot{q}_1 s_2 \frac{2b}{3} \mathbf{m}_b + \left( \frac{2b}{3} \dot{q}_1 c_2 + \dot{q}_2 a_3 \right) \mathbf{m}_a$$

$$\mathbf{w}^T = \dot{q}_1 \left[ \mathbf{m}_b \cos \theta_2 - \mathbf{m}_a \sin \theta_2 \right] + \dot{q}_2 \mathbf{m}_a$$



$$\mathbf{I} = \frac{m a^2}{18} m_a m_a + \frac{m a b}{36} (m_a m_b + m_b m_a) + \frac{m b^2}{18} m_b m_b + \frac{m (a^2 + b^2)}{18} m_c m_c$$

$$P \cdot \mathbf{m}_a = -\frac{2}{3} b \mathbf{m}_a + \frac{1}{3} a \mathbf{m}_b$$

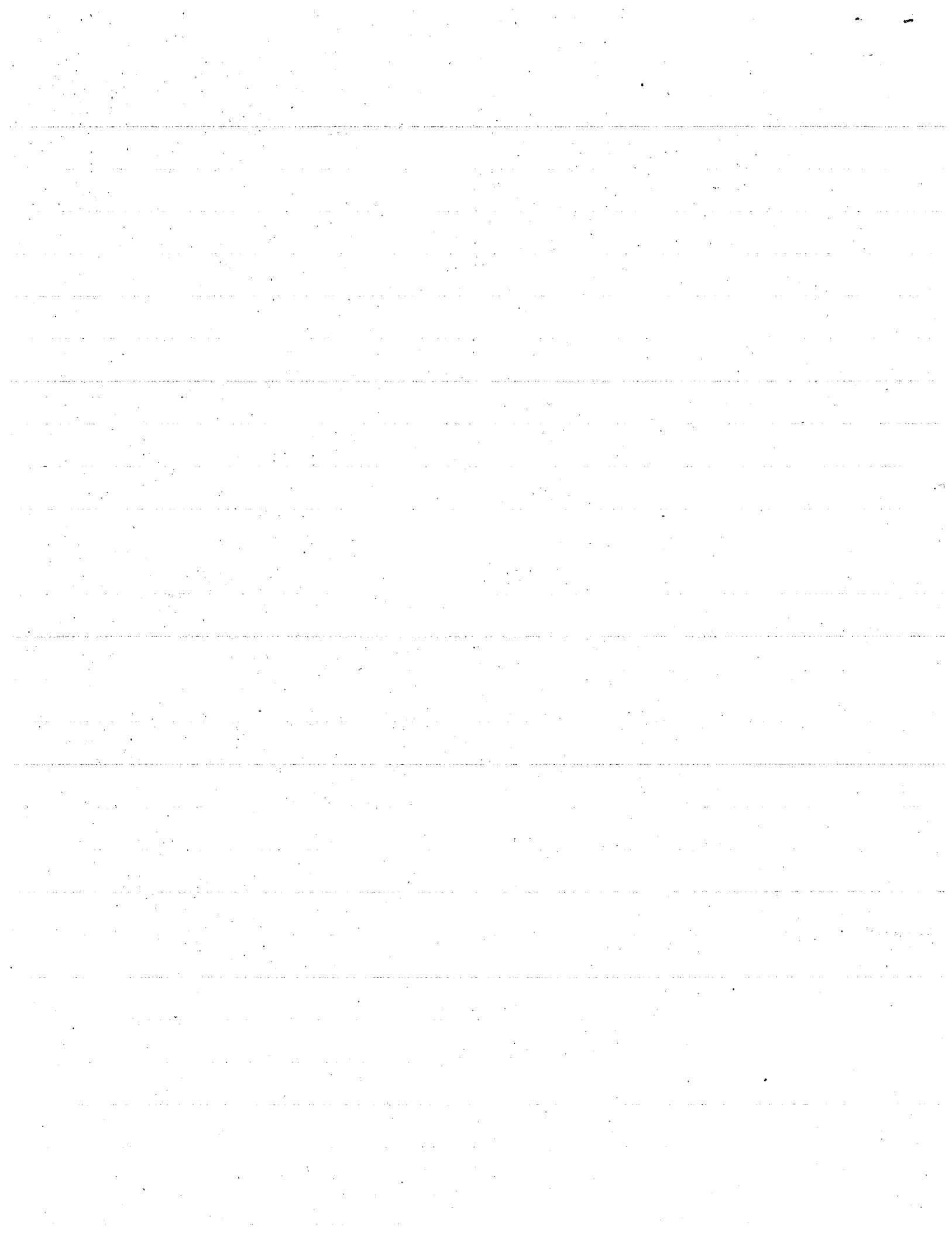
$$P \cdot \mathbf{m}_b = \frac{1}{3} a \mathbf{m}_a$$

$$P \cdot P = \frac{4}{9} b^2 + \frac{b^2}{9}$$

$$\mathbf{I}^{T/A} = \begin{bmatrix} \frac{m a^2}{18} + \frac{m a^2}{9} & \frac{m a b}{18} & 0 \\ \frac{m a b}{18} & \frac{m b^2}{18} + \frac{8 m^2 b^2}{18} & 0 \\ 0 & 0 & m (a^2 + b^2) \end{bmatrix}$$

$$I_{lm} = m [ P^2 \delta_{lm} - (P \cdot \mathbf{m}_l)(P \cdot \mathbf{m}_m) ]$$

$$F_2^* = \frac{\partial K}{\partial q_2} = \frac{d}{dt} \frac{\partial K}{\partial \dot{q}_2}$$



$$\text{II} \cdot \omega = \frac{ma^2}{6} \dot{q}_2 m_a + \frac{mab}{4} [q_1 c_2 m_a + q_2 m_b] + \frac{mb^2}{2} \dot{q}_1 c_2 m_b - q_1 s_2 m \left( \frac{3a^2 + 9b^2}{18} \right) m_c$$

$$\frac{1}{2} \text{II} \cdot \omega = \frac{ma^2}{12} \dot{q}_2^2 + \frac{mab}{8} [2q_1 q_2 c_2] + \frac{mb^2}{4} \dot{q}_1^2 c_2^2 + q_1^2 s_2^2 m \left( \frac{3a^2 + 9b^2}{36} \right)$$

$$\frac{\partial K}{\partial \dot{q}_2} = -\frac{mab}{4} \dot{q}_1 \dot{q}_2 s_2 - \frac{9mb^2}{18} \dot{q}_1^2 c_2 s_2 + \dot{q}_1^2 m \left( \frac{3a^2 + 9b^2}{18} \right) s_2 c_2 = \dot{q}_1^2 \frac{ma^2}{6} s_2 c_2$$

$$\frac{\partial K}{\partial \dot{q}_2} = \frac{ma^2}{6} \dot{q}_2 + \frac{mab}{4} \dot{q}_1 c_2 - \frac{d}{dt} \left( \frac{\partial K}{\partial \dot{q}_2} \right) = -2 \frac{ma^2}{12} \dot{q}_2 + \frac{mab}{4} q_1 s_2 \dot{q}_2 - \frac{3mab}{12} \dot{q}_1 c_2$$

$$F_2^* = \frac{ma}{12} [2a(s_2 c_2 \dot{q}_2^2 - \dot{q}_2) - 3b c_2 \dot{q}_1]$$

### Method of Solution

10d. Find the velocity of the mass centers;  $\text{II} \cdot \omega = I_3 \omega_3$  where  $I_3, \omega_3$  are components in direction out of paper. Thus for each rod form  $\sum \frac{1}{2} I_3 \omega_{3i}^2 = K_w$   $I_3 = ml^2/12$  for rods length  $l$ .

Square velocity & form  $\sum \frac{1}{2} m_i v_i^2 = K_v$  now that  $\frac{\partial K_T}{\partial \dot{q}_1} - \frac{d}{dt} \frac{\partial K_T}{\partial \dot{q}_1} = F_1^*$

$$\text{where } K_T = K_w + K_v$$

$$\text{Better to use } \text{IF}^* \cdot \nabla \dot{q}_1 + \text{II}^* \cdot \omega \dot{q}_1 = F_1^*$$

10e. (8d) is a non-holonomic system.

(8e) Because of generalized speeds we have problem

$$10f. \text{ must form } KE_w + KE_v \quad KE_v = \frac{1}{2} m V_{sp}^{*2} \cdot 4 + \frac{1}{2} m V_c^{*2} \stackrel{?}{=} 0$$

$$KE_w = \frac{1}{2} \omega^c \cdot \text{II}^c \omega^c + \frac{4}{2} \omega^s \cdot \text{II}^s \omega^s = \frac{1}{2} J \omega^2 +$$

$$\text{II}^s = \frac{2}{5} mr^2 (m_1 m_3 + m_2 m_2 + m_3 m_3), \quad \text{II}^c = J m_2 m_2, \quad \omega^c = \omega m_2$$

$$V_s^* \cdot m_3 = \frac{r}{2} V^s = \frac{\omega r \sin \theta r}{\cos \theta + \sin \theta} \quad \text{from problem (3d)} \quad \text{at } 30^\circ \quad \frac{\omega \cdot \sqrt{2} r}{\sqrt{3} r + r} = \frac{\omega r}{\sqrt{3} + 1}$$

$$\omega^s = \omega^c + \omega^s = \omega m_2 + \left( -\frac{r}{2} \omega - \omega \right) m_2 + \left( \frac{r}{2} \omega \right) m_1 = (m_1 - m_2) \frac{r \omega}{r}$$

$$\omega^s = (m_1 - m_2) \frac{\omega}{\sqrt{3} + 1} \quad \omega^c = \omega m_2 \quad V^* = \frac{\omega^2 r^2}{4 - 2\sqrt{3}}$$

$$\therefore K = \frac{1}{2} m \frac{\omega^2 r^2}{4 - 2\sqrt{3}} + \frac{1}{2} J \omega^2 + \frac{4}{2} \frac{\omega^2}{4 - 2\sqrt{3}} (m_1 - m_2) \cdot \left[ \frac{2}{5} mr^2 (m_1 m_3 + m_2 m_2 + m_3 m_3) \right] (m_1 - m_2)$$

$$= \frac{4}{2} \frac{mr^2 \omega^2}{4 - 2\sqrt{3}} + \frac{1}{2} J \omega^2 + \frac{4}{2} \frac{\omega^2}{4 - 2\sqrt{3}} \cdot \frac{2}{5} mr^2 = \frac{2mr^2 \omega^2}{4 - 2\sqrt{3}} + \frac{8}{5} \frac{mr^2 \omega^2}{4 - 2\sqrt{3}} + \frac{1}{2} J \omega^2$$

$$= \frac{m \omega^2 r^2}{2 - \sqrt{3}} + \frac{4}{5} \frac{m \omega^2 r^2}{2 - \sqrt{3}} + \frac{1}{2} J \omega^2$$

$$= \frac{9}{5} m \omega^2 r^2 (2 + \sqrt{3}) + \frac{1}{2} J \omega^2$$



10(g)

$$\omega^c = \omega m_3 \quad \omega^B = \omega m_3 \quad \omega^P_3 = \omega m_3$$

$$W^c = 0 \quad W^B = 0 \quad W^P_2 = (\dot{\theta} + \alpha_2 \sin \theta) m_3 + \frac{g}{2} (1 - \cos \theta) m_1$$

$$W^P_2 = \omega^2 r_2 \times W = \omega \frac{r_2}{2} (1 - \cos \theta) m_2$$

$$K_{TOT} = \frac{1}{2} m W^P_2^2 + \frac{1}{2} \omega^P_1 \cdot \mathbf{I} \cdot \omega^P_1 + \frac{1}{2} \omega^P_2 \cdot \mathbf{I} \cdot \omega^P_2$$

$$= \frac{1}{2} m \frac{a^2}{9} (1 - \cos \theta)^2 + \frac{1}{2} \omega^2 \cdot \frac{m a^2}{12} + \frac{1}{2} \omega^P_2 \cdot \mathbf{I} \cdot \omega^P_2$$

$$I_{xx} = \frac{m a^2}{12}, \quad I_{yy} = \frac{m a^2}{18}, \quad I_{zz} = \frac{m a^2}{12}$$

$$\begin{array}{|ccc|} \hline & P_2 & \\ \hline & m_1 & m_2 & m_3 \\ m_1 & \frac{m a^2}{12} & 0 & 0 \\ m_2 & 0 & \frac{m a^2}{6} & 0 \\ m_3 & 0 & 0 & \frac{m a^2}{12} \\ \hline \end{array} \quad \begin{array}{|ccc|} \hline & m_1 & m_2 & m_3 \\ m_1 & c & g & 0 \\ m_2 & 0 & -\theta & 0 \\ m_3 & -s & 0 & 0 \\ \hline \end{array} \quad \text{only need } I_{33}$$

$$\mathbf{I}^P_2 = \frac{m a^2}{12} (m_1 m_a + 2 m_2 m_b + m_3 m_c)$$

$$\mathbf{I}^P_1 = \frac{m a^2}{12} s^2 m_3 m_3 + 0 + \frac{m a^2}{12} c^2 m_2 m_2 = \frac{m a^2}{12} m_3 m_3 + \dots \quad \therefore \frac{1}{2} \frac{m a^2}{12} (s^2 + 2c^2) = \frac{1}{2} \omega^P_2 \cdot \mathbf{I}^P_2 \cdot \omega^P_2$$

$$\therefore K_{TOT} = \frac{m a^2}{12} s^2 + \frac{1}{2} m \frac{a^2}{12} \theta^2 (1 - \cos \theta)^2 + \frac{1}{2} \omega^2 \frac{m a^2}{12} (s^2 + 2c^2) + \frac{1}{2} \omega^2 \frac{m a^2}{12}$$

$$\frac{\partial K}{\partial \theta} = 0 = 6(1 - \cos \theta)s + 2sc + 4cs$$

$$= 6s - 6cs + 2sc - 4cs = 6s - 8cs = 0 \quad \therefore \frac{3}{4} = \cos \theta \quad \theta = \arccos(\frac{3}{4})$$

$$\frac{\partial^2 K}{\partial \theta^2} = 6c - 8c^2 + 8s^2 = 6c - 8(c^2 - s^2) = 6c - 8(\cos 2\theta)$$

since  $\theta < 90^\circ$  then  $\cos 2\theta < 0 \quad \therefore \frac{\partial^2 K}{\partial \theta^2} > 0 \quad \text{min}$

10(h)

for the fixed point  $\Rightarrow (M_{rs})_B = \omega \dot{q}_r \cdot \mathbf{I} \cdot \omega \dot{q}_s$   
of a rigid body

let  $I_1, I_2, I_3$  be the value of the principal moments of inertia

$$\omega = \dot{q}_1 x_1 + \dot{q}_2 x_2 + \dot{q}_3 x_3$$

$$= (\dot{q}_1 c_2 c_3 + \dot{q}_2 c_1 + \dot{q}_3 \dot{q}_1) a_1 + (-\dot{q}_1 s_3 c_2 + \dot{q}_2 s_1 + \dot{q}_3 \dot{q}_2) a_2 + (s_2 \dot{q}_1 + \dot{q}_3 \dot{q}_2) a_3$$

from the form of  $\omega_i = (c_2 c_3 a_1 - s_3 c_2 a_1 + s_2 a_3)$

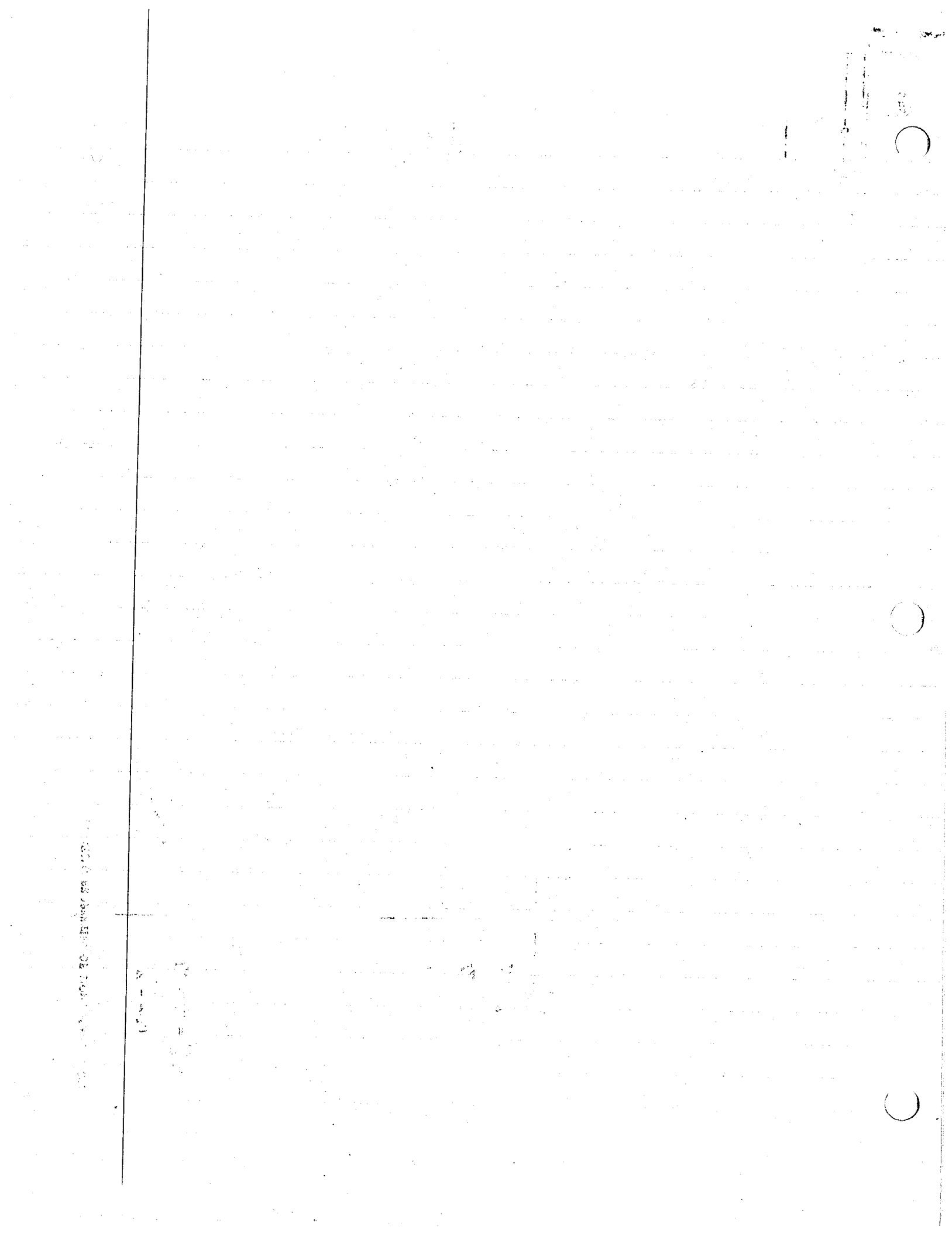
$$\omega \dot{q}_1 = (s_2 a_1 + c_3 a_2)$$

$$\omega \dot{q}_2 = c_3 a_3$$

$$\therefore m_{12} = c_2 c_3 s_1 I_1 - s_3 c_2 c_3 I_2 \quad m_{13} = s_2 I_3 \quad m_{23} = 0$$

$\therefore q_1$  &  $q_2$  and  $q_1$  and  $q_3$  are coupled dynamically

Note that i, j, k, l are virtual work problems



## APPENDIX A

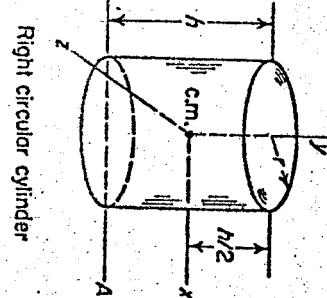
INERTIAL PROPERTIES OF  
HOMOGENEOUS BODIES

$$V = \pi r^2 h$$

$$I_{xx} = I_{yy} = \frac{m}{12} (3r^2 + h^2)$$

$$I_{yy} = \frac{1}{2} mr^2$$

$$I_{zz} = \frac{m}{12} (3r^2 + 4h^2)$$



Right circular cylinder

$$V = \frac{1}{3} \pi r^2 h$$

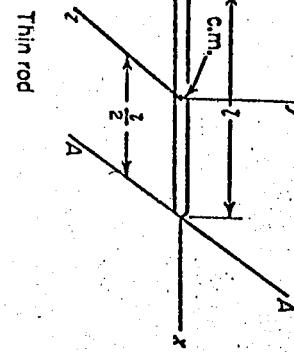
$$I_{xx} = I_{yy} = \frac{3m}{80} (4r^2 + h^2)$$

$$I_{yy} = \frac{3}{10} mr^2$$

$$I_{zz} = \frac{3m}{20} (r^2 + 4h^2)$$

Right circular cone

$$I_{zz} = \frac{m}{20} (3r^2 + 2h^2)$$



Thin rod



Thin circular disk

$$A = \pi r^2$$

$$I_{xx} = I_{yy} = \frac{mr^2}{4}$$

$$I_{zz} = \frac{mr^2}{2}$$

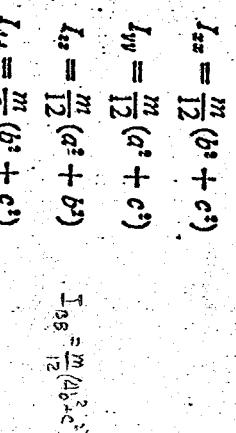
$$A = ab \quad \text{see Fig. 108}$$

$$I_{xx} = \frac{mb^2}{12}$$

$$I_{yy} = \frac{ma^2}{12}$$

$$I_{zz} = \frac{m}{12} (a^2 + b^2)$$

$$T_{xx} = m \frac{b^2}{2}$$



$$V = abc$$

$$I_{xx} = \frac{m}{80} (4b^2 + 3h^2)$$

$$I_{yy} = \frac{m}{20} (a^2 + b^2)$$

$$I_{zz} = \frac{m}{80} (4a^2 + 3h^2)$$

$$I_{zz} = \frac{m}{20} (b^2 + 12h^2)$$

Right rectangular pyramid

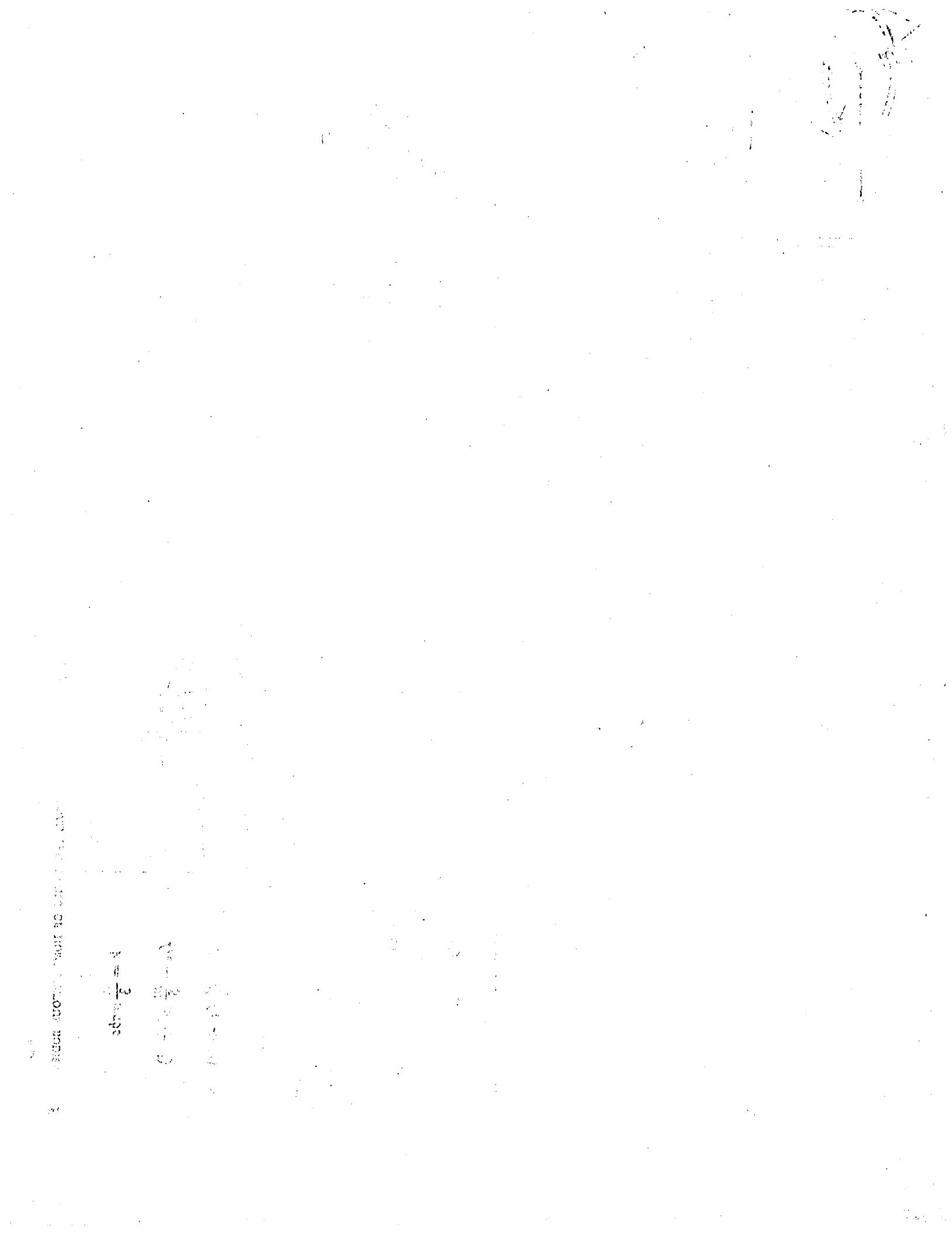
$$I_{zz} = \frac{m}{20} (b^2 + 2h^2)$$

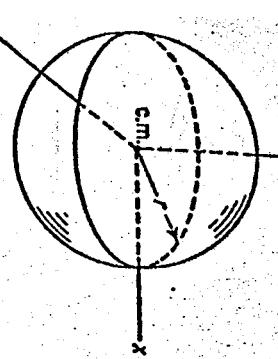
$$I_{xx} = \frac{m}{12} (b^2 + c^2)$$

$$I_{yy} = \frac{m}{12} (a^2 + c^2)$$

$$I_{zz} = \frac{m}{12} (a^2 + b^2)$$

$$T_{xx} = \frac{m}{12} (4b^2 + c^2)$$

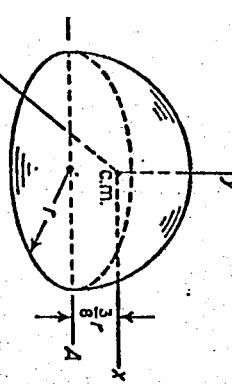




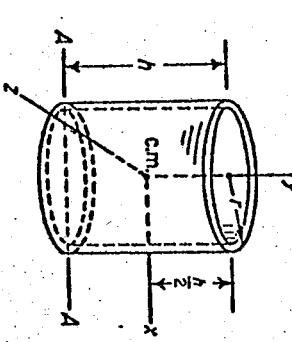
$$V = \frac{4}{3}\pi r^3$$

$$I_{xx} = I_{yy} = I_{zz} = \frac{2}{5}mr^2$$

Sphere



Hemisphere



Thin circular cylindrical shell

$$A = 4\pi r^2$$

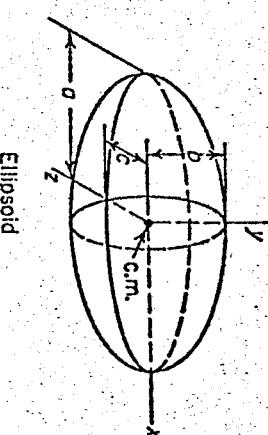
$$I_{xx} = I_{yy} = I_{zz} = \frac{2}{3}mr^2$$

$$V = \frac{2}{3}\pi r^3$$

$$I_{xx} = I_{yy} = \frac{83}{320}mr^2$$

$$I_{yy} = \frac{2}{5}mr^2$$

$$I_{zz} = \frac{2}{5}mr^2$$



Ellipsoid

$$I_{xx} = I_{yy} = I_{zz} = \frac{4}{3}\pi abc$$

$$I_{xx} = \frac{m}{5}(b^2 + c^2)$$

$$I_{yy} = \frac{m}{5}(a^2 + c^2)$$

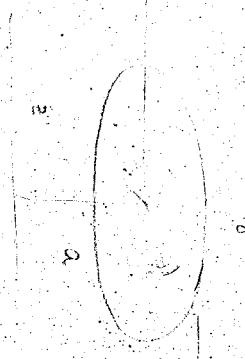
$$I_{zz} = \frac{m}{5}(a^2 + b^2)$$

$I_{ij}$	1	2	3
$I_{11}$	$\frac{mb^2}{18}$	$\frac{mc^2}{36}$	0
$I_{22}$	$\frac{ma^2}{18}$	$\frac{mb^2}{36}$	0
$I_{33}$	0	0	$\frac{m(a^2 + b^2)}{18}$

$$I_{xx} = m(a^2 + b^2)$$

$$I_{yy} = m(b^2 + c^2)$$

$$I_{zz} = \frac{m}{4}(a^2 + b^2 + c^2)$$



Sphere

